

Abstract:

Until recently, the design of jet engines for quiet operation was limited by engineers' ability to predict the jet noise generated by a nozzle exhaust system. More importantly, the 'intermediate steps' between nozzle design and noise had not been understood, limiting the design process to small variations around existing solutions. In recent years NASA's Quiet Aircraft Technology (QAT) Program has advanced the understanding and modeling of jet noise to give engineers the tools they need to design quiet nozzle systems for subsonic exhaust systems. The presentation discusses the approach followed for QAT and argues that a similar effort aimed at supersonic jet exhaust systems will be needed to allow designs of quiet military aircraft in the future.

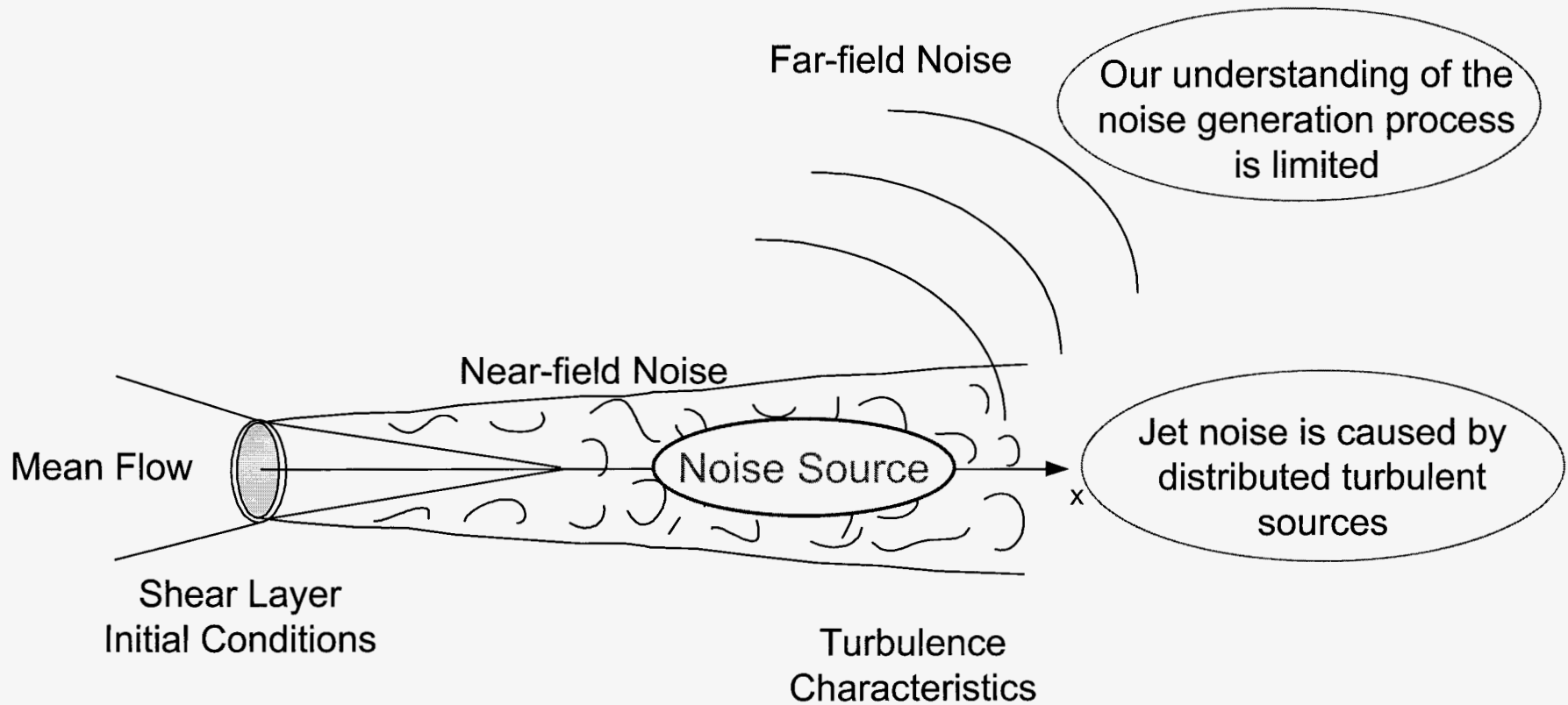


NASA's Vision for Jet Noise Engineering

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NASA Langley Research Center

Dr. James E. Bridges
NASA Glenn Research Center

Jet Noise Problem



A comprehensive understanding of jet flowfield is at the heart of any jet noise reduction strategy

Global Scaling Laws

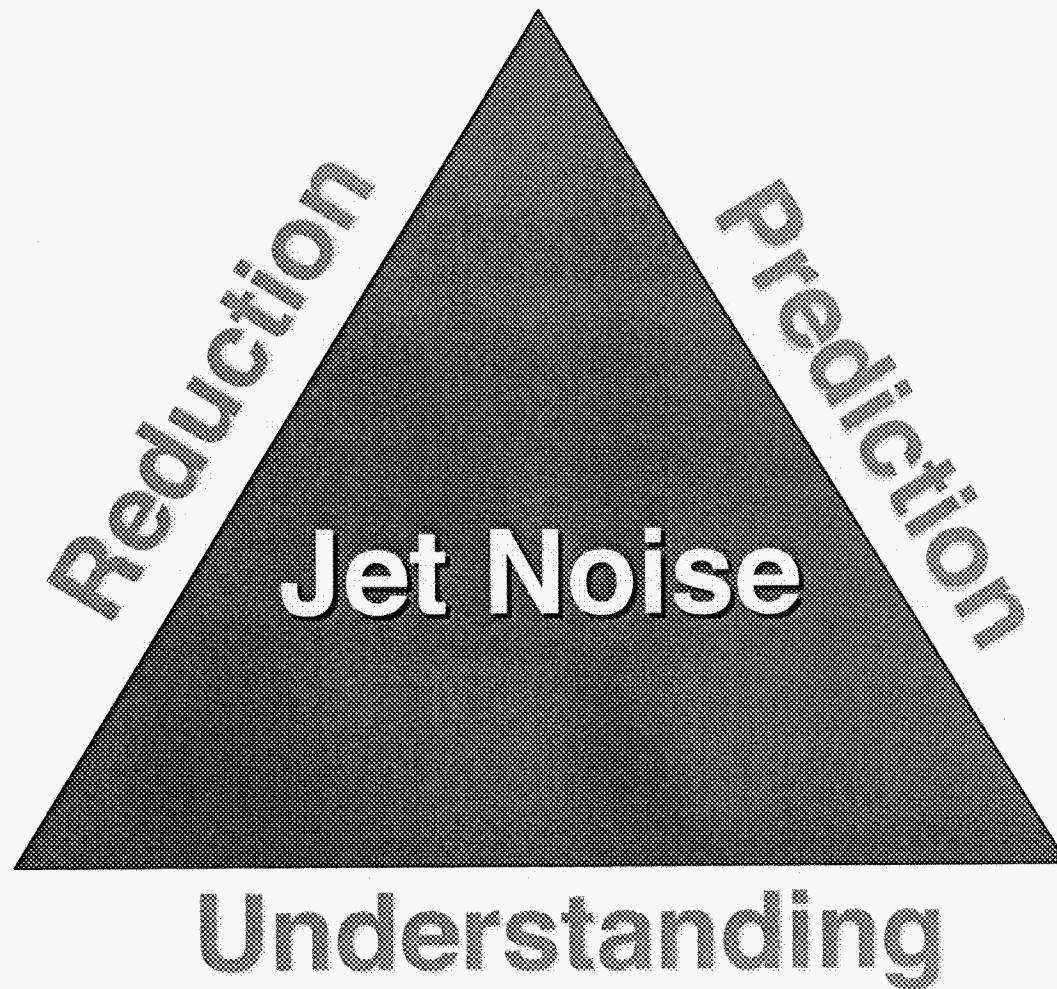


Single most effective jet noise reduction technique is to reduce the jet velocity

$$\text{SPL} \propto 10 \log \left(\frac{\rho_j^2 D_j^2 U_j^8}{\rho_\infty C_\infty^5} \right)$$
A diagram showing the equation for SPL. The term U_j^8 in the numerator is circled in black. A black arrow points from the text above to the circled term.

This level of knowledge will not allow us to design a quiet aircraft

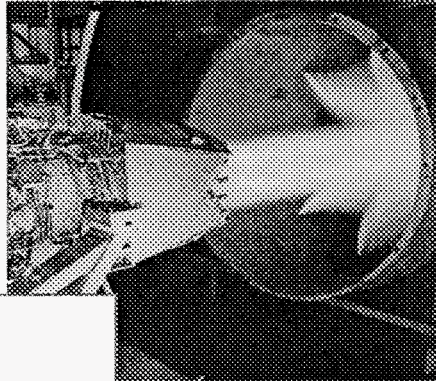
Jet Noise Focus Areas



Jet Noise Research Paradigm



Thrust



Reduction

Engine Cycle
Geometry
Active control

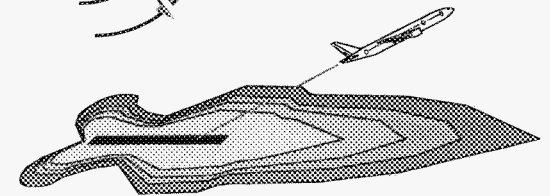
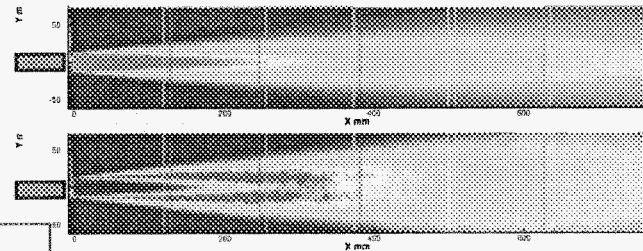
Turbulence

Anisotropy
Spectral shift
Modal content

Acoustic Source Strength

Directionality
Shielding
Atmospheric atten

Noise



NASA Jet Noise Objectives



1. Develop diagnostic and analytical tools to Understand jet noise mechanisms
2. Create physics-based Predictive tools for general subsonic jets—flow and noise.
3. Use Understanding to create noise Reduction concepts and use Predictive tools to guide experiments.

We are pursuing these objectives in the subsonic Quiet Aircraft Technology program

Fundamental Understanding



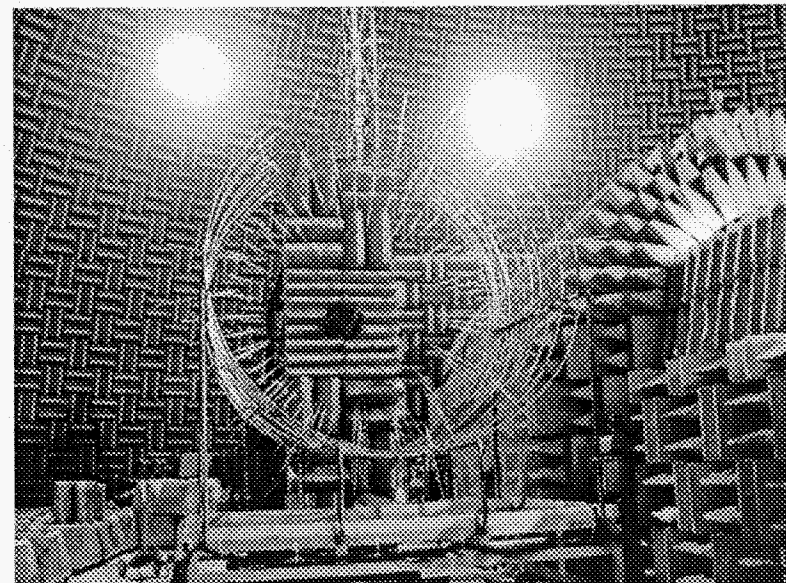
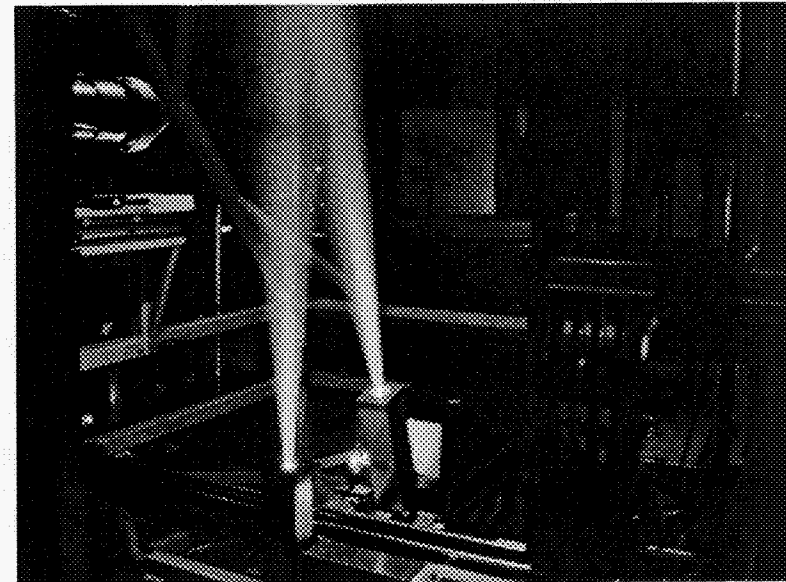
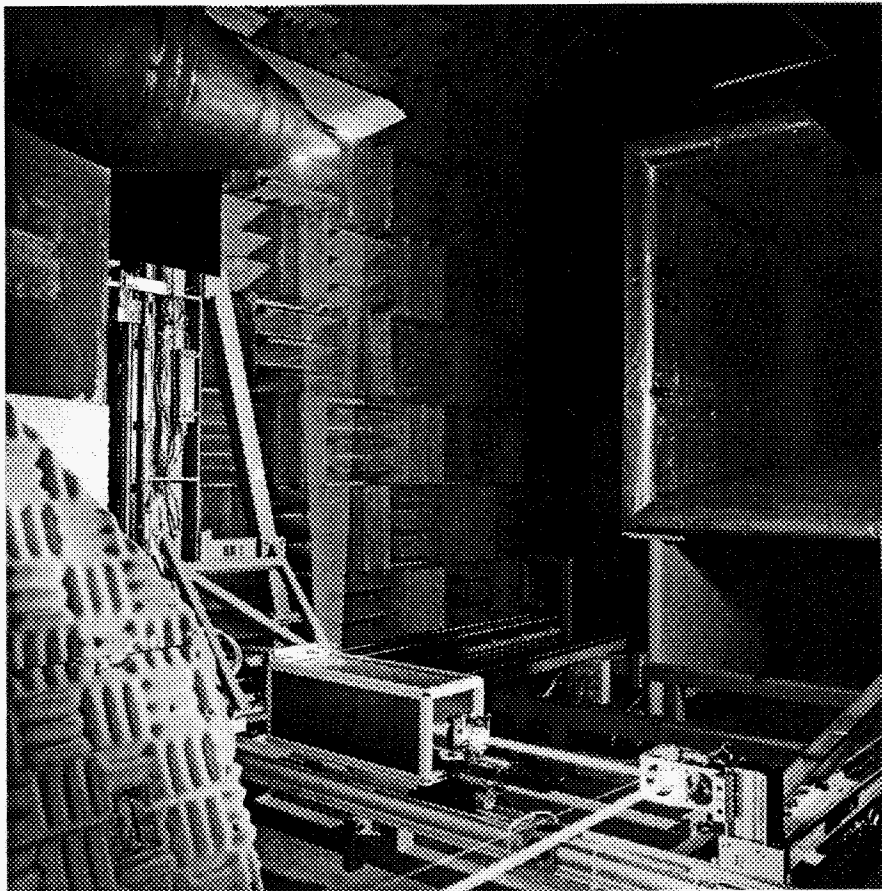
Understanding

- Fundamental Understanding
 - How do jets make noise?

Advanced Measurement Techniques



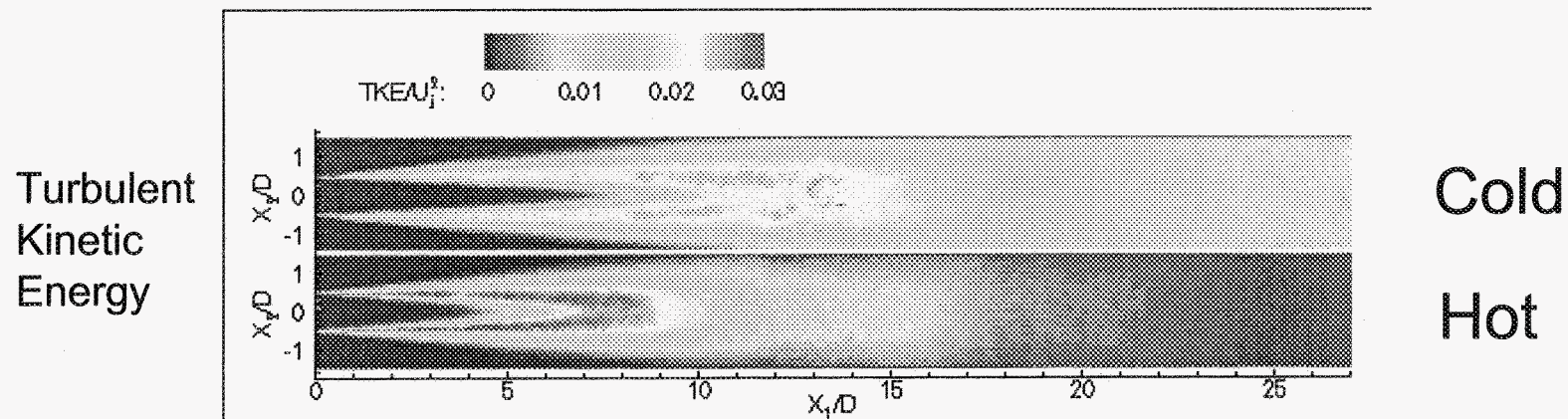
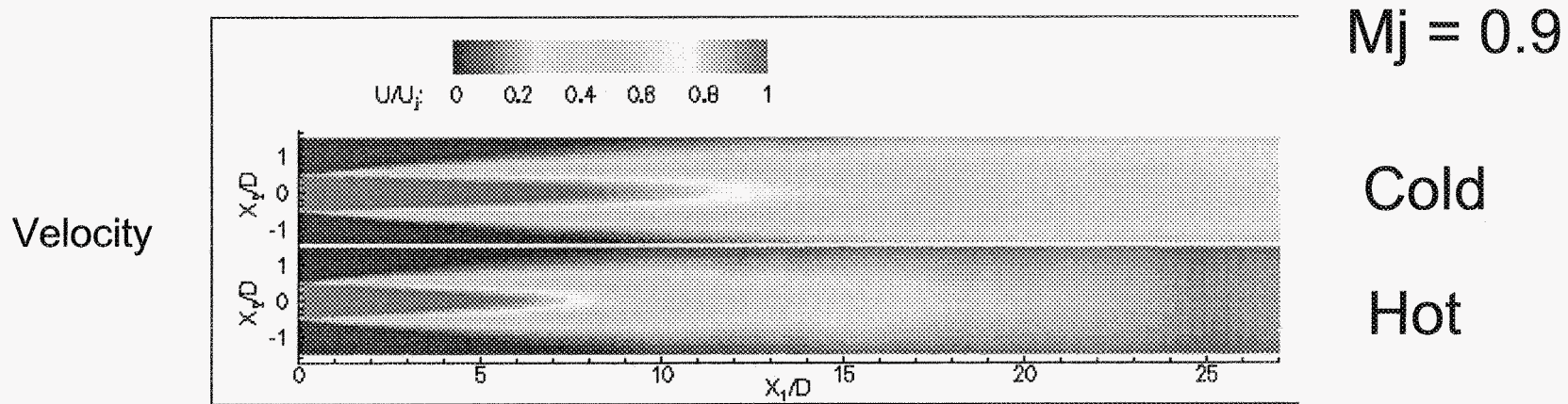
Understanding



Effect of Temperature on Jet Flow



Understanding



Important to account for temperature effects in modeling and experiment



- Jet Noise Prediction
 - What is the flow and acoustic fields produced by a given nozzle with specified cycle conditions?

Levels of Prediction Codes



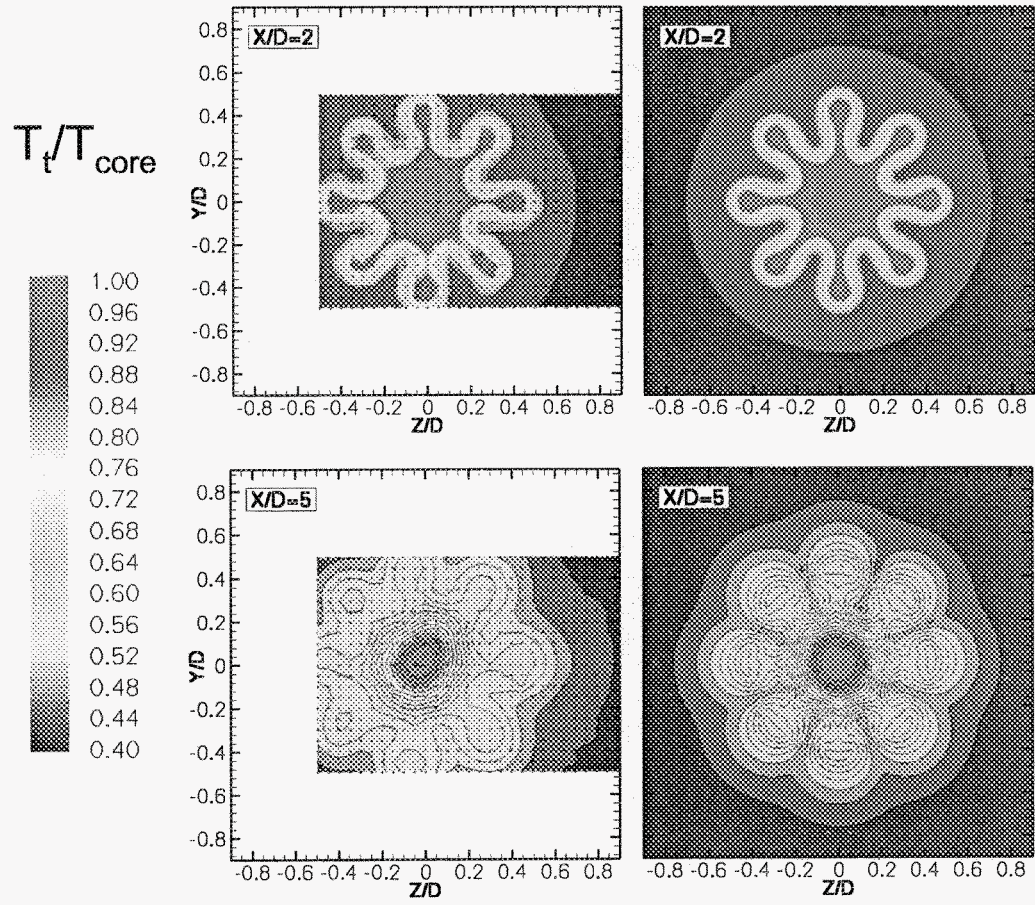
- Empirical correlation with nozzle conditions
 - NOISEMAP, ARP, SAE, ANOPP
 - Database interpolation (Noise, Power, Distance)
 - All cases approximated as simple round nozzle
- Time-averaged physics-based codes
 - Jet3D, MGBK
 - Reynolds-averaged NS + acoustic analogy
 - Can handle unique configurations
- Time-dependent full-physics codes
 - LES, DNS
 - Discrete versions of Navier-Stokes
 - Info overload; infinite cost



CFD Flowfield Prediction

Prediction

Cross-section of Chevron Flow Experiment Prediction



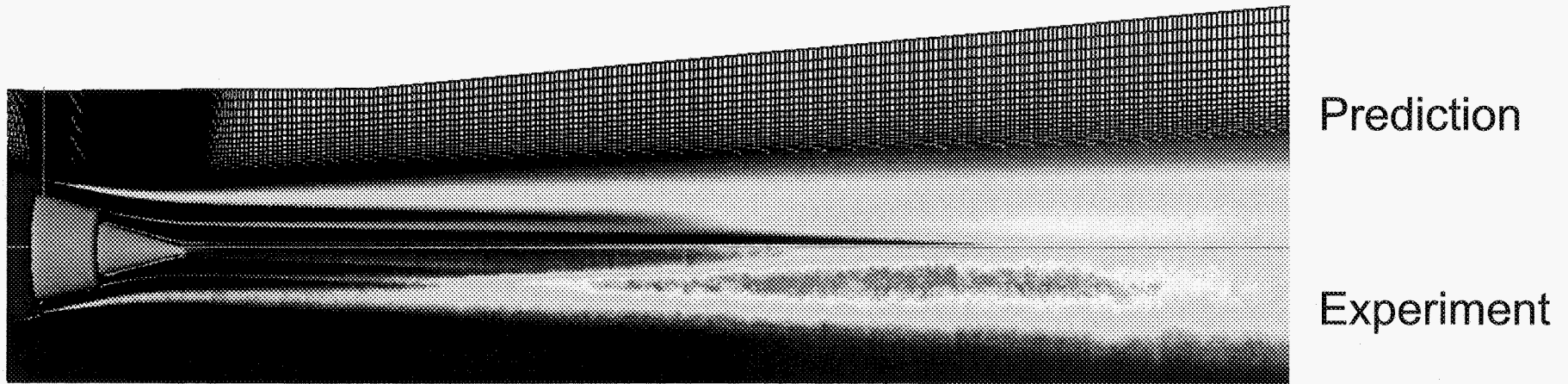
Good mean flow predictions

CFD Flowfield Prediction

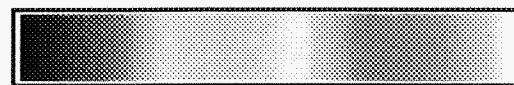


Prediction

Turbulent Kinetic Energy of BPR 5 Separate Flow Nozzle



Turbulent Kinetic Energy (m^2/s^2)



0

2500

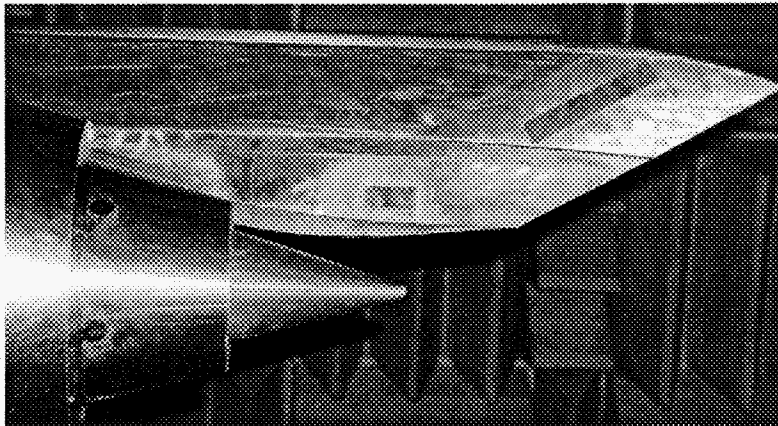
5000

Prediction of turbulence quantities still deficient

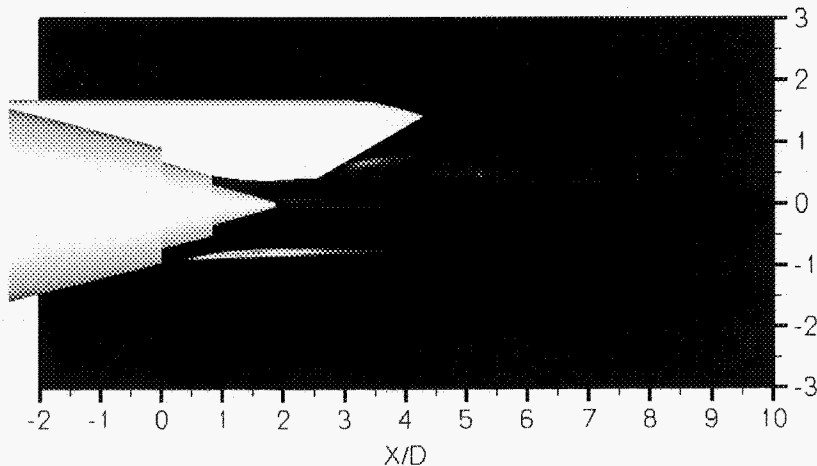
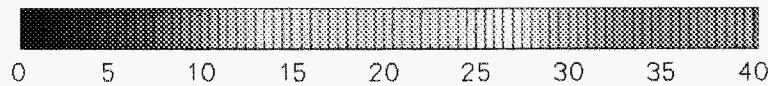
Aeroacoustic Noise Prediction



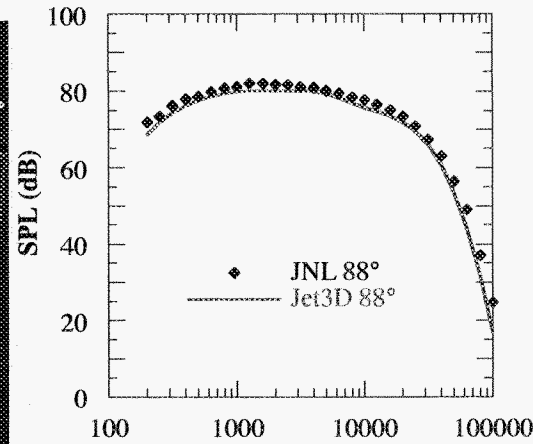
Prediction



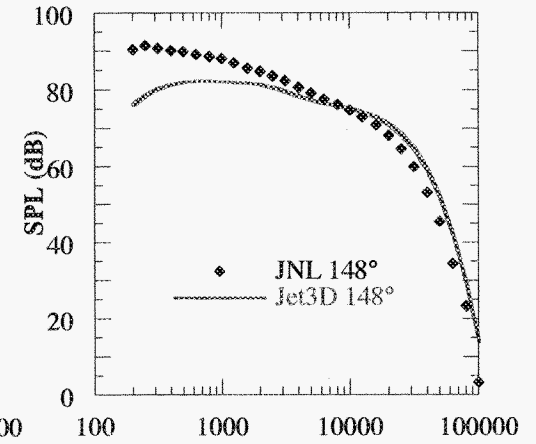
Mean-square Acoustic Pressure
8000Hz Spatial Source Density $(N/m^2)^2/m^3$



Acoustic Source Density Prediction



1/3 Octave Band Center Frequency (Hz)



1/3 Octave Band Center Frequency (Hz)

Far-field Acoustic Prediction

BPR = 5

Takeoff Conditions

NASA is currently developing several CAA type codes

Noise Reduction Concepts



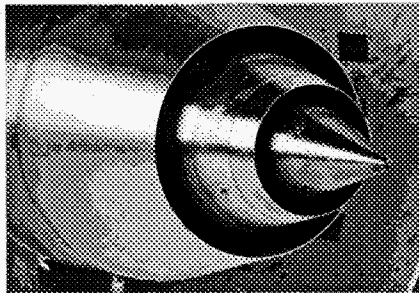
Reduction

- Noise Reduction Concepts
 - How should jet be modified to reduce noise?

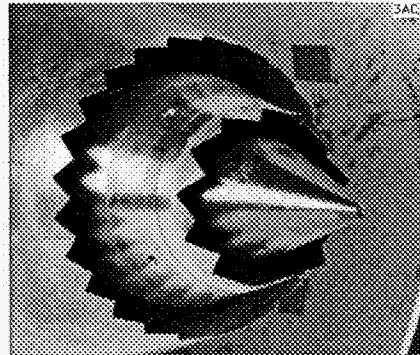
Chevron Mixing Devices



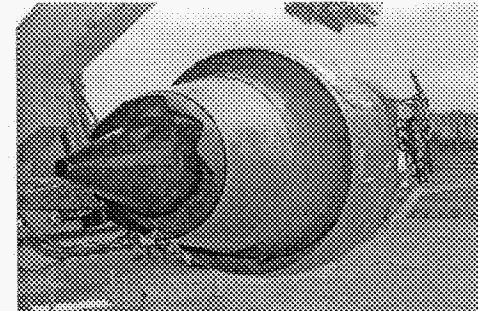
Reduction



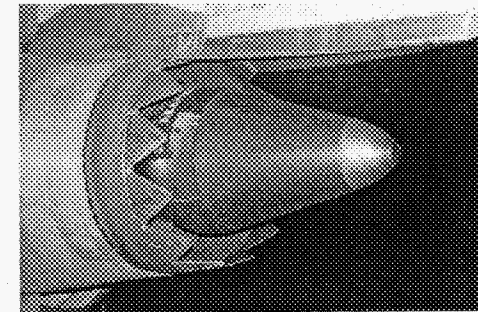
**Baseline Nozzles
(Flying Now)**



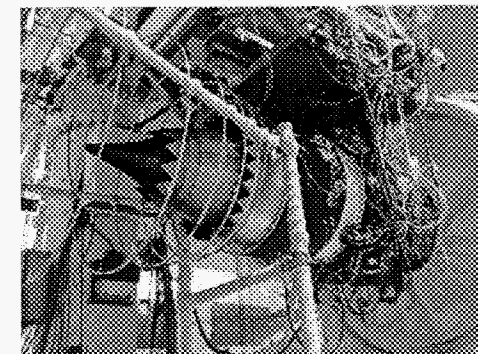
**1997 Model Tests
Chevron Nozzle**



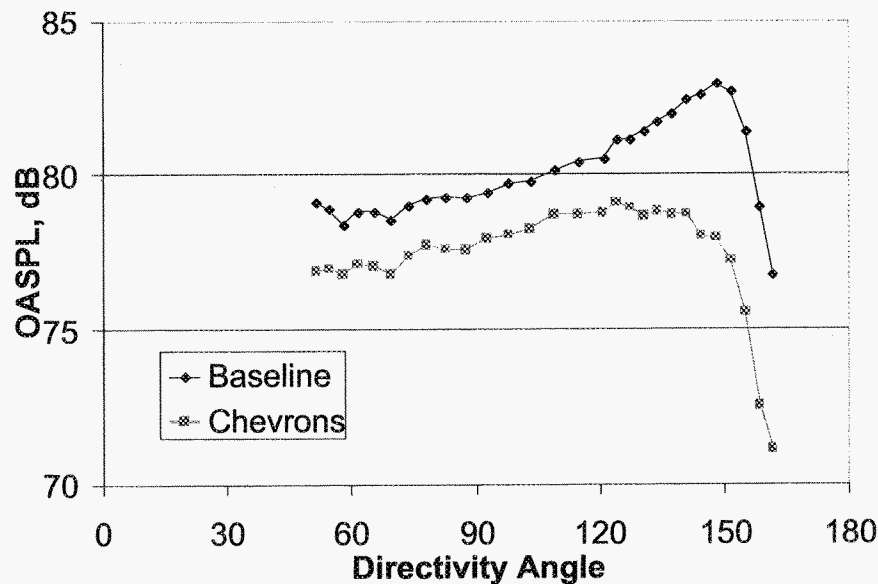
GE CF34 Engine



GE CF6 Engine



Honeywell TFE731-60 Engine

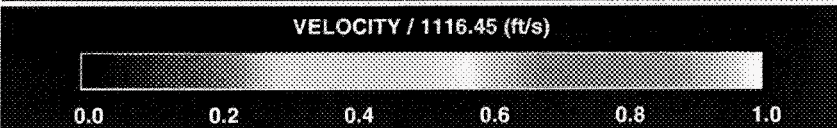
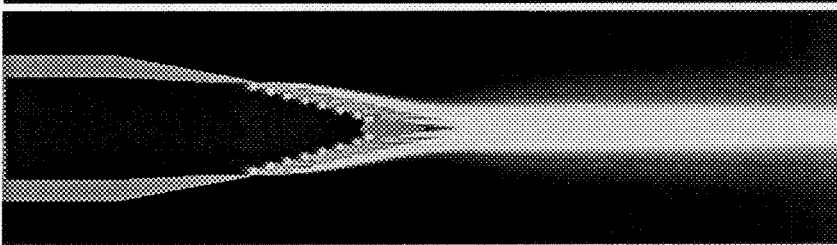
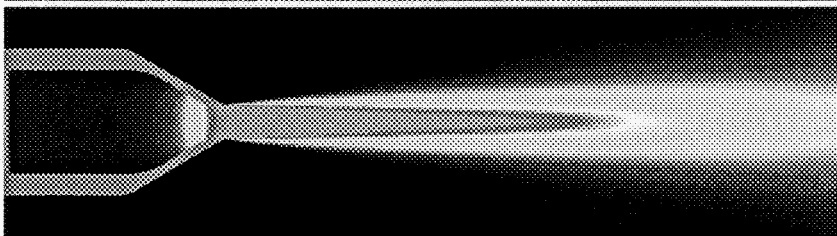
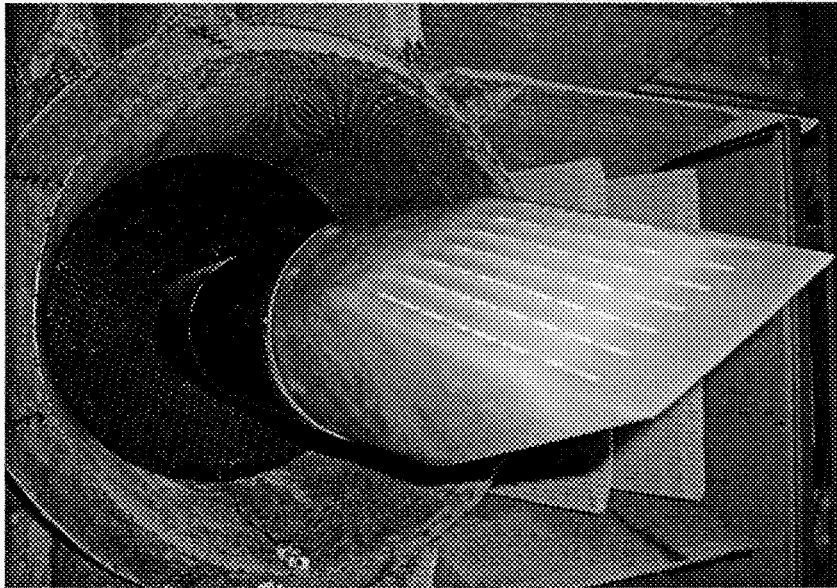


3 EPNdB Jet Noise Reduction!

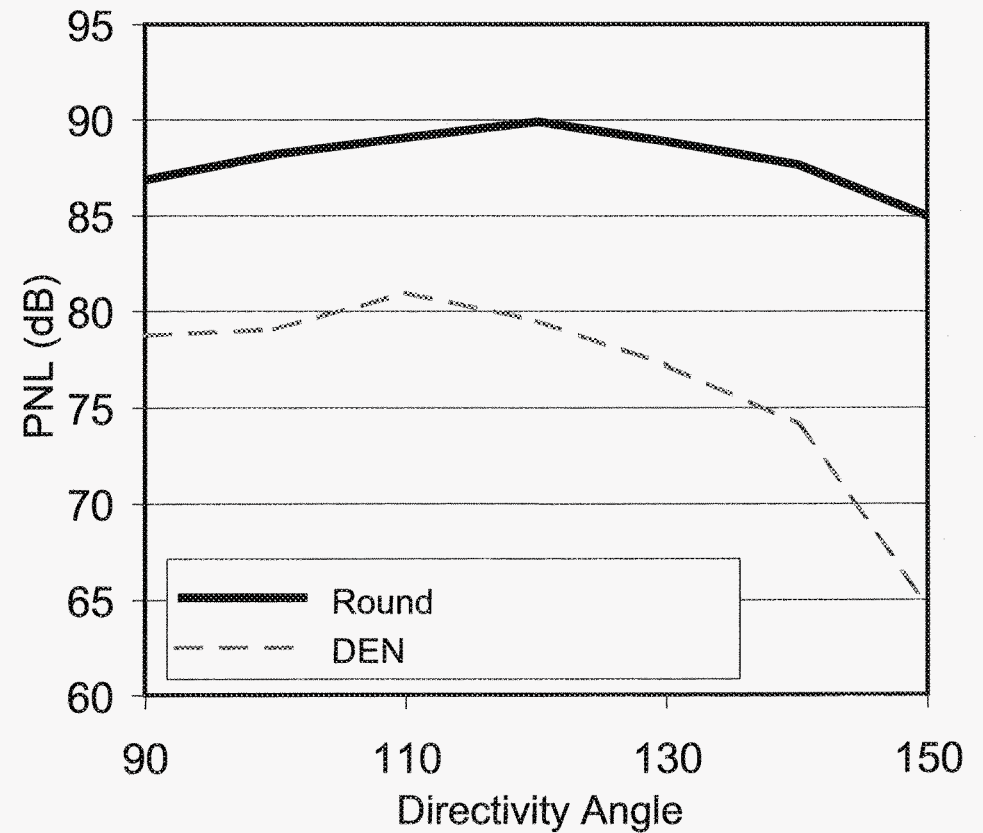
Distributed Exhaust Nozzle



Reduction



Use CFD and testing to optimize DEN thrust and acoustic performance





How Would NASA Approach Supersonic Jet Noise Reduction?

Extending Approach to Supersonics

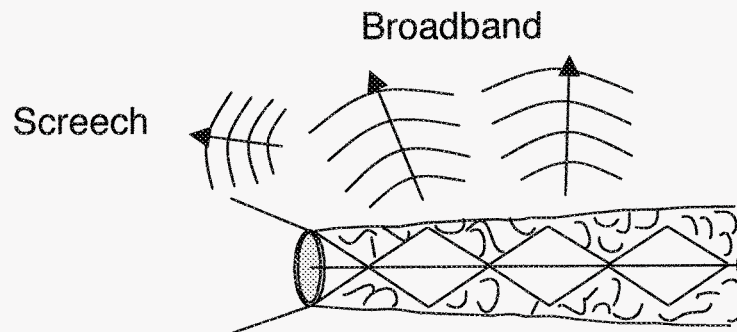


- Develop an understanding of unique supersonic noise sources and propagation
- Include additional sources in prediction tools
- Extend experimental facilities and validate techniques at higher pressure and temperature flows to test noise reduction concepts.

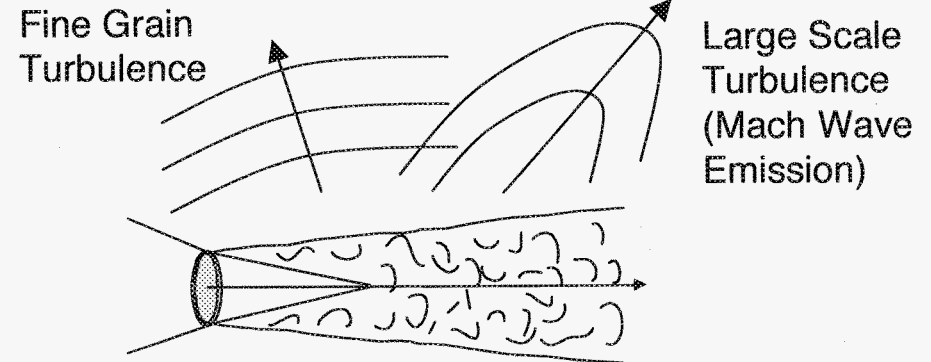
High Speed Jet Noise Sources



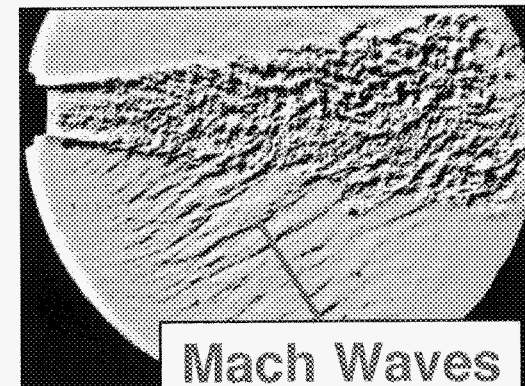
Shock Noise



Mixing Noise



- Mixing noise
- Mach wave radiation
- Shock associated noise
- STOVL noise/tones



Courtesy of D. Papamoschou

Prediction and noise reduction technology must address each of these depending on flight regime

Critical Military Jet Characteristics



Exhaust Temperature – **HOT !!!**

Velocity Profile – **Non-Uniform**

These effects **MUST** be correctly represented to provide research relevant to military aircraft

Suggested Plan for Noise Reduction



- Identify specific areas of need as they relate to aircraft mission
- Develop tools to predict shock associated and Mach wave emission noise
- Couple aeroacoustic modeling and experiments to identify and validate noise reduction concepts

Implement balanced research tasks that address understanding, predicting, and reducing supersonic jet noise



Backup Charts

Chevron Mixing Devices



Reduction

Plume Turbulence is the Main Source of Jet Noise

