

# Comparison of the Atomic Oxygen Erosion Depth and Cone Height of Various Materials at Hyperthermal Energy

*Deborah L. Waters*  
*QSS Group, Inc., Cleveland, Ohio*

*Bruce A. Banks*  
*Glenn Research Center, Cleveland, Ohio*

*Stephen D. Thorson*  
*University of Wisconsin, Madison, Madison, Wisconsin*

*Kim K. de Groh and Sharon K. Miller*  
*Glenn Research Center, Cleveland, Ohio*

## NASA STI Program . . . in Profile

Since its founding, NASA has been dedicated to the advancement of aeronautics and space science. The NASA Scientific and Technical Information (STI) program plays a key part in helping NASA maintain this important role.

The NASA STI Program operates under the auspices of the Agency Chief Information Officer. It collects, organizes, provides for archiving, and disseminates NASA's STI. The NASA STI program provides access to the NASA Aeronautics and Space Database and its public interface, the NASA Technical Reports Server, thus providing one of the largest collections of aeronautical and space science STI in the world. Results are published in both non-NASA channels and by NASA in the NASA STI Report Series, which includes the following report types:

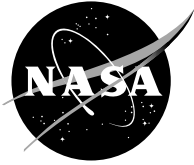
- **TECHNICAL PUBLICATION.** Reports of completed research or a major significant phase of research that present the results of NASA programs and include extensive data or theoretical analysis. Includes compilations of significant scientific and technical data and information deemed to be of continuing reference value. NASA counterpart of peer-reviewed formal professional papers but has less stringent limitations on manuscript length and extent of graphic presentations.
- **TECHNICAL MEMORANDUM.** Scientific and technical findings that are preliminary or of specialized interest, e.g., quick release reports, working papers, and bibliographies that contain minimal annotation. Does not contain extensive analysis.
- **CONTRACTOR REPORT.** Scientific and technical findings by NASA-sponsored contractors and grantees.

- **CONFERENCE PUBLICATION.** Collected papers from scientific and technical conferences, symposia, seminars, or other meetings sponsored or cosponsored by NASA.
- **SPECIAL PUBLICATION.** Scientific, technical, or historical information from NASA programs, projects, and missions, often concerned with subjects having substantial public interest.
- **TECHNICAL TRANSLATION.** English-language translations of foreign scientific and technical material pertinent to NASA's mission.

Specialized services also include creating custom thesauri, building customized databases, organizing and publishing research results.

For more information about the NASA STI program, see the following:

- Access the NASA STI program home page at <http://www.sti.nasa.gov>
- E-mail your question via the Internet to [help@sti.nasa.gov](mailto:help@sti.nasa.gov)
- Fax your question to the NASA STI Help Desk at 301-621-0134
- Telephone the NASA STI Help Desk at 301-621-0390
- Write to:  
NASA STI Help Desk  
NASA Center for AeroSpace Information  
7115 Standard Drive  
Hanover, MD 21076-1320



# Comparison of the Atomic Oxygen Erosion Depth and Cone Height of Various Materials at Hyperthermal Energy

*Deborah L. Waters*  
*QSS Group, Inc., Cleveland, Ohio*

*Bruce A. Banks*  
*Glenn Research Center, Cleveland, Ohio*

*Stephen D. Thorson*  
*University of Wisconsin, Madison, Madison, Wisconsin*

*Kim K. de Groh and Sharon K. Miller*  
*Glenn Research Center, Cleveland, Ohio*

Prepared for the  
10th International Symposium on “Materials in a Space Environment” (ISMSE)  
sponsored by the Office National d’Etudes et de Recherches Aérospatiales (ONERA), Centre National  
d’Etudes Spatiales (CNES), and the European Space Agency (ESA)  
Collioure, France, June 19–23, 2006

National Aeronautics and  
Space Administration

Glenn Research Center  
Cleveland, Ohio 44135

## Acknowledgments

The authors gratefully acknowledge the assistance with the facility by Michael J. DePauw of NASA Glenn Research Center and Frank Lam of Jacobs Sverdrup.

Trade names and trademarks are used in this report for identification only. Their usage does not constitute an official endorsement, either expressed or implied, by the National Aeronautics and Space Administration.

*Level of Review:* This material has been technically reviewed by technical management.

Available from

NASA Center for Aerospace Information  
7115 Standard Drive  
Hanover, MD 21076-1320

National Technical Information Service  
5285 Port Royal Road  
Springfield, VA 22161

Available electronically at <http://gltrs.grc.nasa.gov>

# Comparison of the Atomic Oxygen Erosion Depth and Cone Height of Various Materials at Hyperthermal Energy

Deborah L. Waters  
QSS Group, Inc.  
Cleveland, Ohio 44135

Bruce A. Banks  
National Aeronautics and Space Administration  
Glenn Research Center  
Cleveland, Ohio 44135

Stephen D. Thorson  
University of Wisconsin, Madison  
Madison, Wisconsin 53711

Kim K. de Groh, and Sharon K. Miller  
National Aeronautics and Space Administration  
Glenn Research Center  
Cleveland, Ohio 44135

## Abstract

Atomic oxygen readily reacts with most spacecraft polymer materials exposed to the low Earth orbital (LEO) environment. If the atomic oxygen arrival comes from a fixed angle of impact, the resulting erosion will foster the development of a change in surface morphology as material thickness decreases. Hydrocarbon and halopolymer materials, as well as graphite, are easily oxidized and textured by directed atomic oxygen in LEO at energies of ~4.5 eV. What has been curious is that the ratio of cone height to erosion depth is quite different for different materials. The formation of cones under fixed direction atomic oxygen attack may contribute to a reduction in material tensile strength in excess of that which would occur if the cone height to erosion depth ratio was very low because of greater opportunities for crack initiation. In an effort to understand how material composition affects the ratio of cone height to erosion depth, an experimental investigation was conducted on 18 different materials exposed to a hyperthermal energy directed atomic oxygen source (~70 eV). The materials were first salt-sprayed to provide microscopic local areas that would be protected from atomic oxygen. This allowed erosion depth measurements to be made by scanning microscopy inspection. The polymers were then exposed to atomic oxygen produced by an end Hall ion source that was operated on pure oxygen. Samples were exposed to an atomic oxygen effective fluence of  $1.0 \times 10^{20}$  atoms/cm<sup>2</sup> based on Kapton H (DuPont) polyimide erosion. The average erosion depth and average cone height were determined using field emission scanning electron microscopy (FESEM). The experimental ratio of average cone height to erosion depth is compared to polymer composition and other properties.

## 1. Introduction

Most spacecraft polymer materials exposed to the low Earth orbital (LEO) environment will erode due to atomic oxygen

exposure. If the atomic oxygen arrival comes from a fixed angle of impact, a change in surface texture will develop as the material oxidizes and becomes thinner.

Surface texture can change the optical reflectance of a material from specular to diffuse and increase the solar absorptance of opaque materials. Surface texture can also be the cause of crack initiation or tearing of thin film polymers that have become embrittled as a result of ionizing radiation. As a thin film polymer develops texture, the valleys of the texture can erode through the polymer before the polymer is completely eroded thus making the polymer only partially protective of materials beneath it. Thus, there is merit in understanding the degree of surface texturing that will occur for materials that erode by atomic oxygen attack. The amount of surface texturing increases as the atomic oxygen fluence increases. Thus, it is of interest to examine the cone height to erosion depth ratio for various materials to better understand this phenomenon. Table 1 shows that the ratio of cone height to erosion depth is quite different for different materials in LEO (ref. 1).

TABLE 1.—RATIO OF CONE LENGTH TO EROSION DEPTH AS A FUNCTION OF FLUENCE FOR VARIOUS MATERIALS EXPOSED TO DIRECTED LEO ATOMIC OXYGEN

Material	Ratio of average cone length to erosion depth	Space mission	Atomic oxygen fluence, atoms/cm <sup>2</sup>
Pyrolytic graphite	0.60	EOIM III	$2.3 \times 10^{20}$
Kapton H	0.28	EOIM III	$2.3 \times 10^{20}$
FEP Teflon	0.07	LDEF	$8.43 \times 10^{21}$

These changes in surface morphology, depending on atomic oxygen arrival, could have an impact on tensile properties possibly leading to increased surface crack initiation. Various material compositional properties are analyzed to determine their potential effect on the cone height versus erosion depth of the polymer samples exposed at hyperthermal energy.

## 2. Experimental Methods

Table 2 lists the materials, abbreviations, trade names, densities, LEO erosion yields, and atomic fractional composition that were tested (refs. 1 and 2). Table 2 also lists the hyperthermal (~70 eV) erosion yield relative to Kapton H for fifteen materials (ref. 2). The densities listed are from

either literature values or actual density gradient column measurements. The LEO erosion yields listed in table 2 are those determined from in-space exposures on the Long Duration Exposure Facility (for polyimide Kapton H), or recently retrieved results from a four year exposure on the International Space Station (for all the remaining polymers) (refs. 3 and 4).

TABLE 2.—MATERIALS TESTED AND THEIR ABBREVIATIONS, TRADE NAMES, DENSITIES, AND EROSION YIELDS

Material	Abbrev.	Trade name	Density, gm/cm <sup>3</sup>	LEO erosion yield, × 10 <sup>-24</sup> cm <sup>3</sup> /atom	~70 eV Erosion yield, relative to Kapton H (ref. 2)	Atomic percent						
						H	C	N	O	F	S	Cl
Polyimide H (PMDA)	PI-H	Kapton H	1.427	3.0	1.00	26	56	5	13	0	0	0
Polyimide-HN (PMDA)	PI-HN	Kapton HN	1.435	2.81	1.03	26	56	5	13	0	0	0
Black polyimide	Black PI	Kapton CB	1.42	-----	0.90	--	--	--	--	0	0	0
Pyrolytic Graphite normal to beam	PG	-----	2.22	0.415	0.58	0	100	0	0	0	0	0
Highly-Oriented Pyrolytic Graphite	HOPG	-----	2.26	-----	-----	0	100	0	0	0	0	0
Fluorinated ethylene propylene	FEP	Teflon FEP	2.144	0.200	1.89	0	33	0	0	67	0	0
Ethylene Chlorotrifluoro-ethylene	ECTFE	Halar	1.676	1.79	1.59	33	33	0	0	25	0	8
Polycarbonate	PC	Lexan	1.123	4.29	1.07	42	48	0	9	0	0	0
Chlorotrifluoro-ethylene	CTFE	Aclar	2.133	0.831	2.86	0	33	0	0	50	0	17
Polyethylene	PE	Lupolen	0.918	3.97	1.06	67	33	0	0	0	0	0
Polyethylene Terephthalate	PET	Mylar A200	1.393	3.01	1.30	38	48	0	14	0	0	0
Polyoxymethylene	POM	Delrin	1.398	9.14	10.87	50	25	0	25	0	0	0
Polytetrafluoro-ethylene	PTFE	Teflon PTFE	2.150	0.142	1.86	0	33	0	0	67	0	0
Polyvinyl fluoride	PVF	Tedlar	1.379	3.19	1.10	50	33	0	0	17	0	0
Polyvinylidene Fluoride	PVDF	Kynar	1.762	1.29	1.42	33	33	0	0	33	0	0
Polytetrafluoro-ethylene ethylene Copolymer	ETFE	Tefzel ZM	1.740	0.961	1.08	33	33	0	0	33	0	0
Polymethyl methacrylate	PMMA	Lucite; Plexiglass	1.163	>5.60	-----	53	33	0	13	0	0	0
Polystyrene	PS	Styron; Lustrex; Polystyrol	1.503	3.74	-----	50	50	0	0	0	0	0

The eighteen samples were salt-sprayed with a fine mist of saturated sodium chloride which dried and provided sites protected from atomic oxygen erosion. This allowed the erosion depth to be distinguished from the cone development. The samples were held in place in the facility with a circular wire copper holder with attached mounting clips which held the samples at approximately equal distance radially around the axis of the beam. This assured that they all received

approximately the same Kapton effective fluence (fig. 1). Various polymers and graphite were chosen because of potential spacecraft interest and usage. These materials have been studied in space (refs. 3 and 4), which allows for potential conclusions to be drawn.

The atomic oxygen effective fluence was measured using Kapton H polyimide based on a density of 1.42 gm/cm<sup>3</sup> and a LEO erosion yield of 3.0×10<sup>-24</sup> cm<sup>3</sup>/atom for fluence witness



TABLE 3.—RATIO OF CONE HEIGHT TO EROSION DEPTH  
FOR THE VARIOUS MATERIALS

Materials	Average cone height (μm)	Standard deviation of cone height (μm)	Average erosion depth (μm)	Standard deviation of erosion depth (μm)	Ratio of cone height to erosion depth	Standard deviation of ratio
Fluorinated ethylene Propylene, Teflon FEP	0.478	0.094	2.853	0.238	0.168	0.213
Polyoxymethylene, Delrin	3.742	0.779	20.597	1.918	0.182	0.228
Chlorotrifluoroethylene, Aclar	0.758	0.099	3.732	0.417	0.203	0.172
Polytetrafluoroethylene, Teflon PTFE	0.723	0.156	3.547	0.465	0.204	0.252
Polyvinylidene fluoride, Kynar	0.530	0.085	2.164	0.221	0.245	0.190
Ethylene chlorotrifluoroethylene, Halar	0.711	0.096	2.841	0.265	0.250	0.164
Polycarbonate, Lexan	0.523	0.065	2.023	0.157	0.259	0.146
Polyethylene terephthalate, Mylar	0.568	0.104	2.113	0.232	0.269	0.213
Polytetrafluoroethylene ethylene copolymer, Tefzel	0.523	0.077	1.928	0.360	0.271	0.237
Polymethyl methacrylate, Lucite	0.927	0.141	3.191	0.266	0.290	0.173
Polyvinyl fluoride, Tedlar	0.656	0.149	2.238	0.327	0.293	0.270
Polystyrene, Styron	0.524	0.092	1.728	0.214	0.303	0.215
Polyimide H (PMDA), Kapton H	0.595	0.101	1.935	0.180	0.307	0.194
Black polyimide, Kapton CB	0.565	0.121	1.824	0.192	0.310	0.239
Polyimide-HN (PMDA), Kapton HN	0.545	0.089	1.724	0.239	0.316	0.214
Polyethylene, Lupolen	0.430	0.074	1.352	0.171	0.318	0.214
Pyrolytic graphite, normal to beam, PG	0.250	0.049	0.453	0.066	0.553	0.243
Highly-oriented pyrolytic graphite, normal to beam, HOPG	0.131	0.021	0.183	0.022	0.716	0.203

generally show a ratio of cone height to erosion depth that is less than non-halogenated and non-oxygen containing hydrocarbons and carbon but greater than the halogenated polymers. Delrin, a polymer with oxygen functionality has the highest ratio for polymers with oxygen functionality, actually in the same range as the halogenated polymers. In comparing the various polymer materials, it appears to be a trend that the ratio of cone height to erosion depth correlates with the polymer functionality, being lowest for halogenated polymers, intermediate for oxygen functionalities and highest for simple hydrocarbons and carbon. Density and LEO (~4.5 eV) atomic oxygen erosion yield values, shown in table 2, were also examined to determine if they correlate with the ratio of cone height to erosion length. However, no consistent trends are evident in correlating the cone height to erosion depth ratio with either of these material properties.

Banks et al. (ref. 2) also utilized fifteen of the materials used in this report in a study of erosion yields at hyperthermal energies (~70 eV). The ratios of cone height to erosion depth shown in table 3 were compared to hyperthermal energy atomic oxygen erosion yield values (ref. 2), shown in table 2, and are plotted in figure 4 indicating a gradual reduction cone height to erosion depth with increasing end Hall erosion yield. Atomic oxygen erosion is a spatially independent erosion process similar to snow accumulation or the fluence of cosmic rays arriving at a surface. As such, it should obey Poisson statistics where the standard deviation in erosion (or cone height) should increase as the square root of the erosion (fluence times erosion yield) (ref. 7). Thus there should be a linear relationship between cone height/erosion depth and  $1/(\text{end Hall erosion yield})^{0.5}$  for a constant fluence.



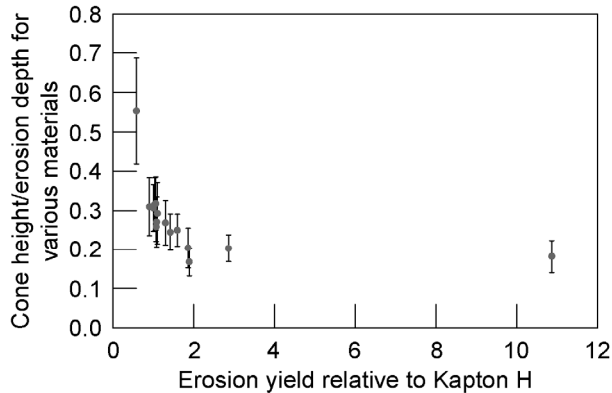


Figure 4.—Graph of the cone height/erosion depth versus the hyperthermal energy erosion yield relative to Kapton H for 15 different materials.

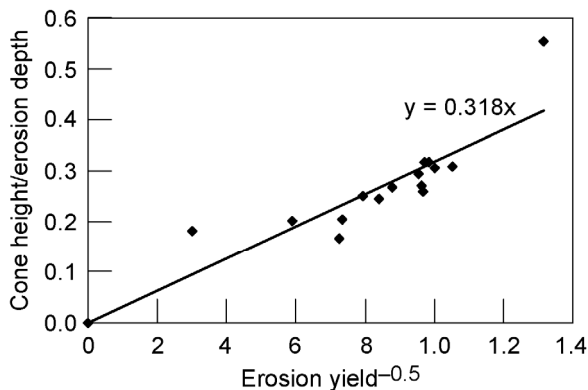


Figure 5.—Cone height/erosion depth versus  $1/(\text{erosion yield})^{-0.5}$  for end Hall ion source data.

Using this end Hall data relative to Kapton H, a graph was constructed with cone height/erosion depth versus  $1/(\text{end Hall erosion yield})^{0.5}$ . This graph with a trend line is shown in figure 5. It indicates a linear relationship for most polymers with some exception for Delrin and pyrolytic graphite. Based on figure 5, the cone height can be predicted as:

$$\text{Cone height} = (0.318) (\text{erosion depth}) (\text{erosion yield})^{-0.5} \quad (1)$$

or

$$\text{Cone height} = (0.318) (\text{fluence}) (\text{erosion yield})^{0.5} \quad (2)$$

where all quantities are in cm units. When using the above formulas, it is important to use the erosion yield of the material in its relevant environment (such as hyperthermal, LEO or thermal energy atomic oxygen).

The slope of the trend line in figure 5 was transferred over to known data (ref. 1) from LEO for three materials to produce the graph shown in figure 6. This plot also shows good

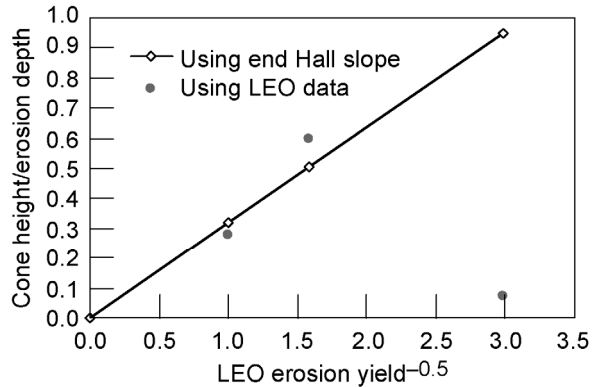


Figure 6.—Cone height/erosion depth versus  $1/(\text{erosion yield})^{-0.5}$  for LEO data for Kapton H, pyrolytic graphite and FEP Teflon.

correlation in predicting the cone height over erosion depth with the exception of FEP where very small cones formed in spite of reasonable fluence. The exception for FEP may be caused by an erosion yield for fluoropolymers that is solar flux and temperature dependent unlike most other polymers.

#### 4. Summary

Predicting in-space atomic oxygen erosion phenomena based on ground laboratory testing is valuable in understanding expected LEO changes in optical, mechanical, and atomic oxygen protection properties of materials. The development of surface texture expressed as the ratio of cone height to erosion depth is not a simple constant for all materials. The ratio of cone height to erosion depth was evaluated for eighteen different polymers exposed to a hyperthermal oxygen ion beam using an end Hall oxygen ion source operated at an average energy of 70 eV. Several compositional factors were examined for trends. In correlating the trend with polymer composition, it was observed that the halogenated polymers showed the lowest ratios of cone height to erosion depth, simple hydrocarbons and carbon showed the highest ratio, and polymers with oxygen functionalities generally showed ratios higher than halogenated hydrocarbons and lower than simple hydrocarbons and carbon. Directed atomic oxygen texturing appears to obey Poisson statistics where the texture grows as the square root of the erosion. The hyperthermal erosion yield of fifteen materials was found to yield a linear relationship between cone height/erosion depth and  $1/(\text{end Hall erosion yield})^{0.5}$  for most materials. This linear relationship was found to be consistent with ratios of cone height to erosion depth data for Kapton H and pyrolytic graphite with the exception of FEP Teflon. Thus, for most all polymers, with the exception of fluoropolymers such as FEP Teflon, prediction of texture development with atomic oxygen fluence appears possible if one knows the erosion yield of the material.

## References

1. Banks, Bruce A., de Groh, Kim K., and Miller, Sharon K., "Low Earth Orbital Atomic Oxygen Interactions With Spacecraft Materials," NASA/TM—2004-213400 and paper presented at the Fall 2004 Meeting of the Materials Research Society, Boston, Massachusetts, November 29—December 3, 2004.
2. Banks, Bruce A., Waters, Deborah L., Thorson, Steven D., de Groh, Kim K., Snyder, Aaron, Miller, Sharon, "Comparison of Atomic Oxygen Erosion Yields of Materials at Various Energies and Impact Angles," Paper presented at the ISMSE-ICPMSE 2006, Collioure, France, June 19–23, 2006.
3. Banks, B.A., Chapter 4, entitled, "The Use of Fluoropolymers in Space Applications," in *Modern Fluoropolymers: High Performance Polymers for Diverse Applications*, edited by J. Scheirs, John Wiley & Sons, Ltd., 1997.
4. de Groh, K. Banks, B., McCarthy, C., Rucker, R., Berger, L., and Roberts, L., "Analysis of the MISSE PEACE Polymers International Space Station Environmental Exposure Experiment," Paper presented at the 10th International Symposium on Materials in a Space Environment, Collioure, France, June 19–23, 2006.
5. ASTM E 2089–00, "Standard Practices for Ground Laboratory Atomic Oxygen Interaction Evaluation of Materials for Space Applications, June 2000.
6. James R. Kahn and Raymond S. Robinson, NASA contract report, Front Range Research, Fort Collins, CO, Sept. 28, 1989.
7. Banks, B.A., Rutledge, S.K., Chan, A., Sahota, M., "The Development of Surface Roughness and Implications for Cellular Attachment in Biomedical Applications," NASA/TM—2001-211288, presented at the 2001 Fall Meeting of the Materials, Research Society, Boston, MA, November 26–30, 2001.

# REPORT DOCUMENTATION PAGE

*Form Approved*  
*OMB No. 0704-0188*

Public reporting burden for this collection of information is estimated to average 1 hour per response, including the time for reviewing instructions, searching existing data sources, gathering and maintaining the data needed, and completing and reviewing the collection of information. Send comments regarding this burden estimate or any other aspect of this collection of information, including suggestions for reducing this burden, to Washington Headquarters Services, Directorate for Information Operations and Reports, 1215 Jefferson Davis Highway, Suite 1204, Arlington, VA 22202-4302, and to the Office of Management and Budget, Paperwork Reduction Project (0704-0188), Washington, DC 20503.

<b>1. AGENCY USE ONLY</b> ( <i>Leave blank</i> )		<b>2. REPORT DATE</b> March 2007	<b>3. REPORT TYPE AND DATES COVERED</b> Technical Memorandum	
<b>4. TITLE AND SUBTITLE</b> Comparison of the Atomic Oxygen Erosion Depth and Cone Height of Various Materials at Hypothermal Energy			<b>5. FUNDING NUMBERS</b>  WBS 843515.01.15.03	
<b>6. AUTHOR(S)</b> Deborah L. Waters, Bruce A. Banks, Stephen D. Thorson, Kim K. de Groh, and Sharon K. Miller				
<b>7. PERFORMING ORGANIZATION NAME(S) AND ADDRESS(ES)</b> National Aeronautics and Space Administration John H. Glenn Research Center at Lewis Field Cleveland, Ohio 44135-3191			<b>8. PERFORMING ORGANIZATION REPORT NUMBER</b>  E-15655	
<b>9. SPONSORING/MONITORING AGENCY NAME(S) AND ADDRESS(ES)</b> National Aeronautics and Space Administration Washington, DC 20546-0001			<b>10. SPONSORING/MONITORING AGENCY REPORT NUMBER</b>  NASA TM-2007-214374	
<b>11. SUPPLEMENTARY NOTES</b> Prepared for the 10th International Symposium on "Materials in a Space Environment" (ISMSE) sponsored by the Office National d'Etudes et de Recherches Aerospatiales (ONERA), Centre National d'Etudes Spatiales (CNES), and the European Space Agency (ESA), Collioure, France, June 19-23, 2006. Deborah L. Waters, e-mail: Deborah.L.Waters@nasa.gov, QSS Group, Inc., 21000 Brookpark Road, Cleveland, Ohio 44135; Bruce A. Banks, e-mail: Bruce.A.Banks@nasa.gov, Kim K. de Groh, and Sharon K. Miller, Glenn Research Center; Stephen D. Thorson, e-mail: sdthorson@wisc.edu, University of Wisconsin, Madison, 1633 Monroe Street, Madison Wisconsin 53711. Responsible person, Deborah L. Waters, organization code RPY, 216-433-5371.				
<b>12a. DISTRIBUTION/AVAILABILITY STATEMENT</b> Unclassified - Unlimited Subject Category: 31  Available electronically at <a href="http://gltrs.grc.nasa.gov">http://gltrs.grc.nasa.gov</a> This publication is available from the NASA Center for AeroSpace Information, 301-621-0390.			<b>12b. DISTRIBUTION CODE</b>	
<b>13. ABSTRACT</b> ( <i>Maximum 200 words</i> ) Atomic oxygen readily reacts with most spacecraft polymer materials exposed to the low Earth orbital (LEO) environment. If the atomic oxygen arrival comes from a fixed angle of impact, the resulting erosion will foster the development of a change in surface morphology as material thickness decreases. Hydrocarbon and halopolymer materials, as well as graphite, are easily oxidized and textured by directed atomic oxygen in LEO at energies of ~4.5 eV. What has been curious is that the ratio of cone height to erosion depth is quite different for different materials. The formation of cones under fixed direction atomic oxygen attack may contribute to a reduction in material tensile strength in excess of that which would occur if the cone height to erosion depth ratio was very low because of greater opportunities for crack initiation. In an effort to understand how material composition affects the ratio of cone height to erosion depth, an experimental investigation was conducted on 18 different materials exposed to a hyperthermal energy directed atomic oxygen source (~70 eV). The materials were first salt-sprayed to provide microscopic local areas that would be protected from atomic oxygen. This allowed erosion depth measurements to be made by scanning microscopy inspection. The polymers were then exposed to atomic oxygen produced by an end Hall ion source that was operated on pure oxygen. Samples were exposed to an atomic oxygen effective fluence of 1.0x10 <sup>20</sup> atoms/cm <sup>2</sup> based on Kapton H polyimide erosion. The average erosion depth and average cone height were determined using field emission scanning electron microscopy (FESEM). The experimental ratio of average cone height to erosion depth is compared to polymer composition and other properties.				
<b>14. SUBJECT TERMS</b>  Atomic oxygen; Low earth orbit; Erosion yield			<b>15. NUMBER OF PAGES</b> 12	
			<b>16. PRICE CODE</b>	
<b>17. SECURITY CLASSIFICATION OF REPORT</b> Unclassified	<b>18. SECURITY CLASSIFICATION OF THIS PAGE</b> Unclassified	<b>19. SECURITY CLASSIFICATION OF ABSTRACT</b> Unclassified	<b>20. LIMITATION OF ABSTRACT</b>	



