



Outline



- Heritage & History
- Level 1 Requirements
- Top Level Overview of the Observatory
- Development Challenges
- Highlight Photos





The Great Observatories

- Mt Wilson
- Mt Palomar
- Keck (Hawaii)





Hubble



National Aeronautics and Space

Ames Research Center

Administration





National Aeronautics and Space Administration Ames Research Center

Hubble Discoveries







1965-1985

- 1965-1969 First use of NASA aircraft for infrared (IR) astronomy(CV-990 & Lear Jet)
- 1969 Planning started for installation of 36-inch telescope in a CV-990
- **1969** First astronomy community interest in a Boeing 747
- 1971 National Academy of Sciences (NAS) Decade Survey (Greenstein)
 Report recommended study of Large Airborne Telescope (LAT)
- 1974 First research flight of Kuiper Airborne Observatory; Initial studies on larger system
- 1983-1984 Success of Infrared Astronomy Satellite (IRAS) shows need for follow-up/exploitation
- 1984 "Stratospheric Observatory For Infrared Astronomy (SOFIA) Preliminary Feasibility Study" report issued (10th KAO Anniversary)
- 1985 SOFIA study funding provided partnership with Germany started





1986

- JANUARY Ames Research Center (ARC) establishes SOFIA Study Office
- JANUARY Challenger accident
- MAY SOFIA technology workshop at Ames;
- MAY-NOVEMBER Boeing-Military Aircraft Company Phase I Study; Confirms feasibility of installing a 2.5 meter telescope in a 747SP
- **JULY** Draft of the Memorandum of Understanding (MOU) for Telescope System study established with Germany
- OCTOBER Ames in-house Conceptual System study begins
- **NOVEMBER** Collaborative agreement made with DFVLR (Deutsche Forschungsanstalt für Luft und Raumfahrt)
- NOVEMBER German Phase A Telescope System studies kickoff

<u>1987</u>

- **FEBRUARY** Telescope System Phase A Study midterm review at ARC
- **FEBRUARY-SEPTEMBER** Boeing-MAC Phase II Study
- MAY German Phase A studies completed
- JULY SOFIA concept review held at ARC
- SEPTEMBER Ames Conceptual System study finished; SOFIA "Phase A System Concept Description" (The Red Book) published





1988

- JUNE Space and Earth Sciences Advisory Committee (SESAC) recommends that SOFIA proceed into definition phase
- JUNE Began planning for wind tunnel tests
- OCTOBER Phase B (Definition Study) kickoff for Aircraft System at Ames
- OCTOBER Phase B (Definition Study) kickoff for Telescope Assembly at Zeiss

<u> 1989</u>

- **JANUARY** Telescope fixed at 2.5 meters by NASA HQ/DFVLR agreement
- **FEBRUARY** Wind tunnel model design complete and fabrication begins
- MAY Project Definition Review completed at ARC; Found SOFIA well planned and defined and approved the project to proceed into the development phase contingent on a successful completion of wind tunnel test
- JUNE Draft MOU for development & operations phases reviewed by Ames & DLR
- JULY Non-Advocate Cost Review successfully completed, Affirmation of project readiness for 1991 start; FRG listed as responsible for telescope assembly
- JULY Definition studies completed by NASA
- **SEPTEMBER** Telescope and Aircraft System Phase B final reviews are completed and reports published
- OCTOBER Boeing re-organizes; No longer interested in "one-off" mods like SOFIA
- NOVEMBER Berlin wall falls; Reunification of East and West Germany considered





1990

- MARCH SOFIA I wind tunnel model tests start
- MAY DARA budget cuts begin
- **JUNE** Non-Advocate Review is held for SOFIA in accordance with the agency's new start-gate policy; SOFIA deemed ready to proceed to development again and recommended for 1992 start
- **JUNE** Preliminary engineering study of SOFIA Ground Support Facility
- JUNE Aircraft System modification procurement activities underway, Source Evaluation Board (S.E.B.) established
- **IULY** Wind tunnel tests successfully completed; A low drag passive shear layer control device derived that exceeds performance expectations
- **OCTOBER** Reunification of Germany, requires reduction of German government agencies' budgets

1991

- NAS Decade Survey (Bahcall) Report recommends SOFIA as the top priority moderate new missions for NASA
- MAY With the realization of DARA budget cuts, SOFIA plans FY92 to prepare for an all U.S. program with optional help from DARA in FY94
- MAY Aircraft Modification Contractors road trip to find companies w/interest/capability to perform the SOFIA aircraft modification
- JULY SEPT In-house descope studies, to reduce total cost; 5 cases considered, one considers an aft cavity location to reduce aircraft modification costs
- OCTOBER Aft cavity location adopted as new baseline for the Aircraft System Presenter: Nans Kunz 650-604-5988 Nans.Kunz@nasa.gov Al Bowers 661-276-3716 Al.Bowers@nasa.gov





1992

- JANUARY IR measurements made of the Shuttle Carrier Aircraft (SCA) engine plumes using IR cameras mounted in Lear jet
- **DECEMBER** Final reports of Aircraft Systems NRA concur with Ames in-house study regarding feasibility and cost savings for the aft cavity configuration

<u>1993</u>

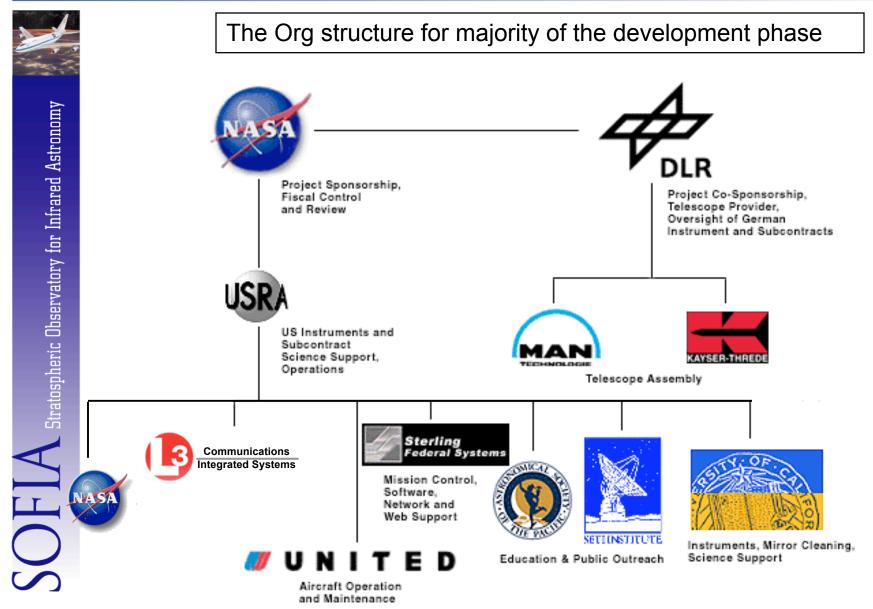
- **JUNE** ARC Code R agrees to de-mothball 14 ft wind tunnel for SOFIA test; Test entry planned for 1994
- AUGUST Headquarters OSS proposes SOFIA as an FY95 new start to Administration/Comptroller

1994-1995

- SOFIA New Start approved
- Headquarters mandates Privatization concept:
 - "Government owned, contractor operated"
 - "Better-Faster-Cheaper"
- Procurement proceeds for development & operations phase
 - Science organization prime Government work packages

<u> 1996</u>

- December SEB process complete NASA <u>contract awarded</u> to USRA-UAL-Chrysler Tech Team
- December DLR <u>awards contract</u> to team of MAN-G, MAN-T, & KT



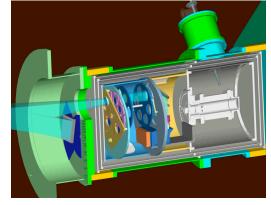




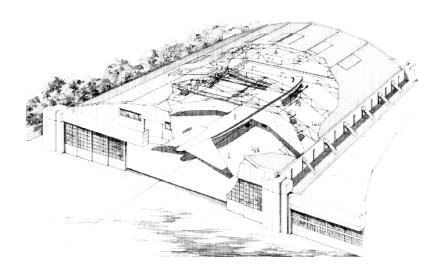
Major Components of SOFIA







Science Instruments



Science and Mission Operations Center







- System requirement reviews completed (2)
- Baseline Flight test completed
- SOFIA V Wind Tunnel test completed
- 1998
 - TA PDR completed
 - AS PDR completed
- 1999
 - Schedule slips
 - 3% S&C wind tunnel tests completed
- 2000
 - TA CDR completed
 - AS CDR completed
 - Schedule continues to slip
- 2001
 - TA ground I&T begins
 - September 11 attack impacts US airlines



Contract Milestones



- TA ground testing & Project Final Review completed
- TA shipped to Waco September 2002

2003

- TA integration into aircraft begins
- Columbia accident
- UAL departs SOFIA program under bankruptcy protection (9/11)

2004

- TA functional, SI mounted, First Light August 2004
- Aircraft Proof pressure test completed
- DSI selected in Germany to support SOFIA Ops

2005

- Push for flight leads to multiple mishaps then work stoppage
- Mod audit conducted
- Per ICSMR recommedations Re-baseline & new approach begins





Contract Milestones

- 2006
 - SOFIA Budget zeroed for FY07 budget
 - SORT commissioned to consider options for future
 - GVT conducted on aircraft in June per IMS established 8/2005
 - HQ dropped requirement for FAA certification (Public Use)
 - Budget rebaselined & program office transferred to Dryden
 - Flight Readiness Review started in Oct
- 2007
 - Airworthiness Flight Safety Review Board on 15 Mar 07
 - First Flight on 26 Apr 07
 - Ferry Flight of SOFIA to Dryden 31 May 07
 - Begin Phase 1 flights (door closed envelope expansion) Sep 25 07?



Heritage



- Modified C-141 with 36" Diameter Telescope
- Flew w/open port cavity 1974-1995
- Cavity in forward fuselage
- Porous fence was primary Shear Layer Control device
- Aft Ramp augmentation based on SOFIA development wind tunnel test results was implemented in 1993
 - Flow attachment significantly improved
 - Internal Cabin noise significantly reduced for Open cavity flight
 - Cavity Environment significantly improved
 - Allowed fence position to be lowered from 30° to 10°
 - Reduced drag improved flight performance

Administration

National Aeronautics and Space **Kuiper Airborne Observatory (KAO)**





1974-1995

Lockheed C-300 (Modified C-141)

36" Telescope



KAO Aft Ramp - Passive Flow Fairing

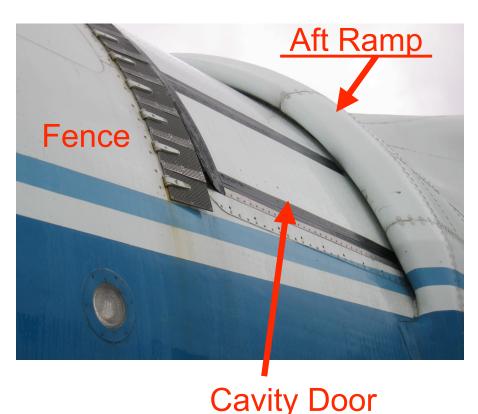


- Installed in 1993
- Developed from wind tunnel test data and research performed during initial development of the SOFIA Shear Layer Control System
- KAO design represents a compromise due to existing OML & cavity door constraint



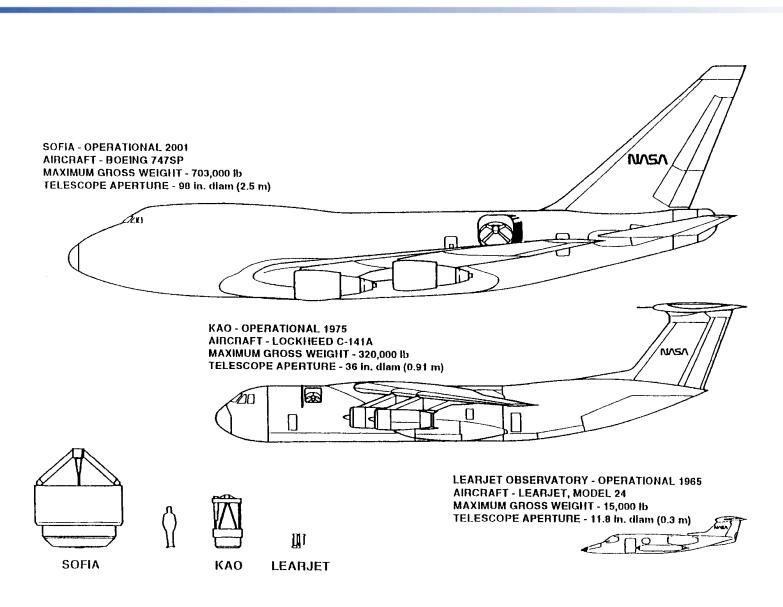


KAO Aft Ramp - Passive Flow Fairing



- Designed to stabilize the shear layer re-attachment downstream of the open cavity.
- Enabled KAO to fly with the cavity fence at 10° instead of 30°
- Reduced Shear layer thickness
- Significant improvements in "Seeing"
- Reduced cavity aero-acoustics
- Reduced structural fatigue in and around cavity
- Pilot noticed improvements in open door flight

National Aeronautics and Space Administration Ames Research Center OFIA - Airborne Astronomy Size Comparison

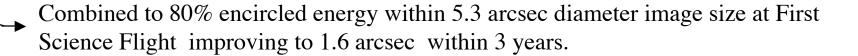




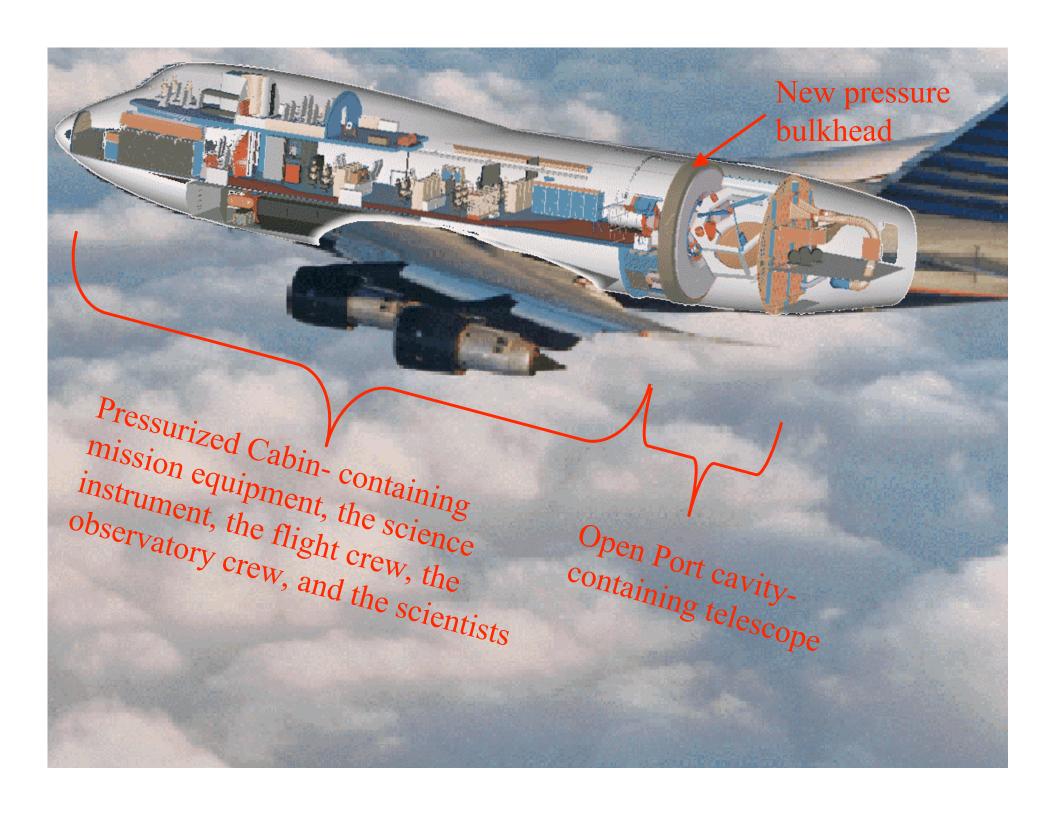
SOFIA - Requirements/Specifications



- Wavelength Range 0.3 1600 microns
- Unvignetted elevation range 20° to 60° above the horizon
- Configuration: Instrument Access in Cabin
- Telescope effective Aperture Diameter 2.5 meters
- Time at $\geq 41,000$ feet ≥ 6 hours
- Observing hours per year ≥ 960
- Lifetime≥ 20 years
- PI Teams per year capability ≥ 40
- Education Goals: NASA OSS Guidelines
- Airworthiness: FAR FAA Certification
- IR functional capabilities: chopping, nodding, & scanning
- Image quality 80% encircled energy within 1.5 arcsec at visible wavelength
- Image stability at focal plane 0.2 arcsec rms



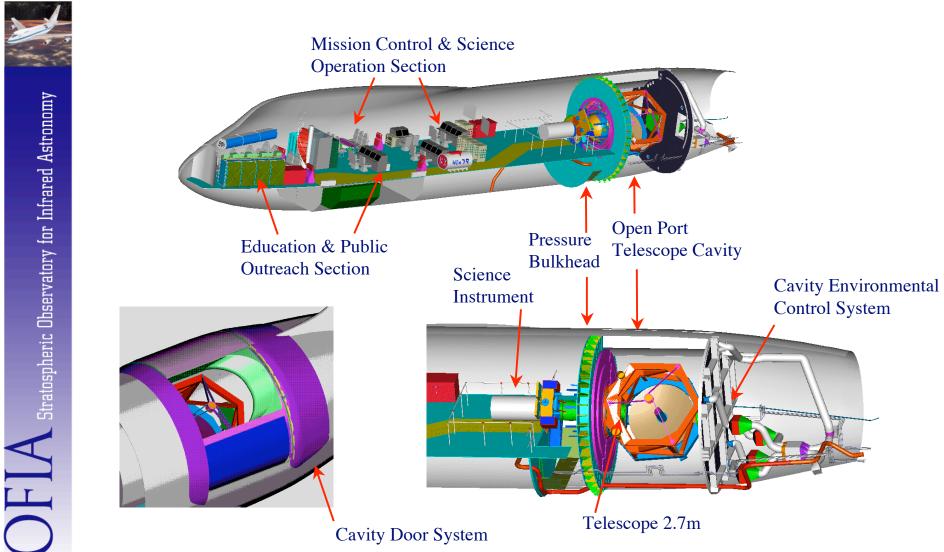








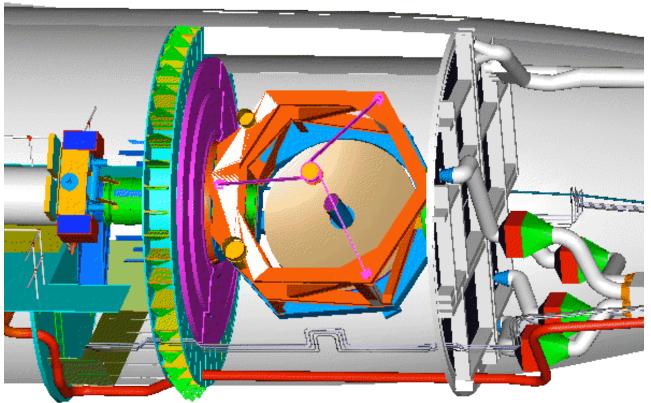
Airborne Observatory Layout





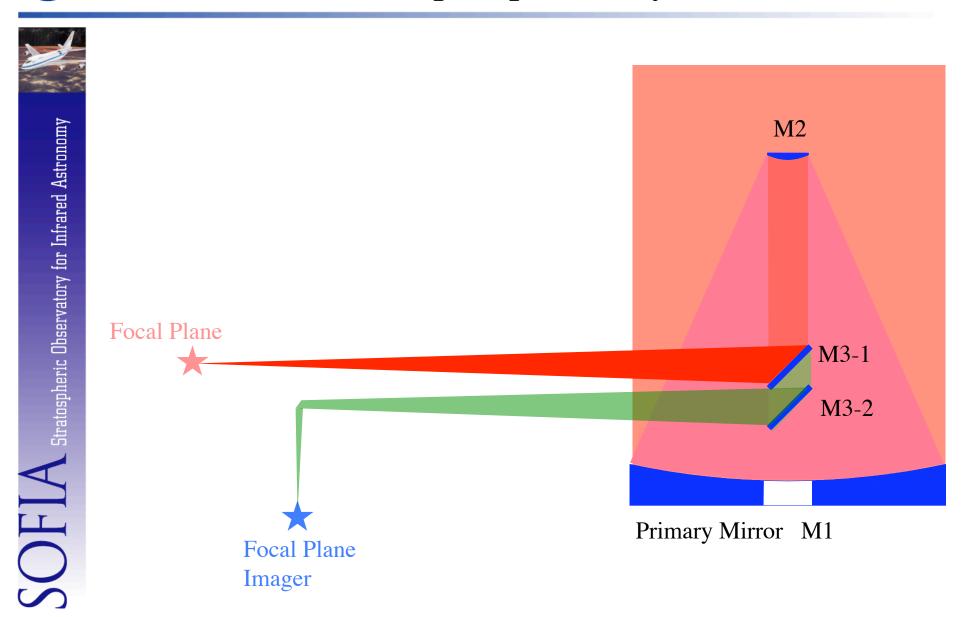
2.5 Meter effective aperture

- Aircraft Size
- Large 2.7 Meter Primary Mirror
- "Fast" Mirror to fit within aircraft
 - Drives alignment/stiffness requirements



Telescope Size is Maximum that can fit Available Volume

Telescope Optical Layout





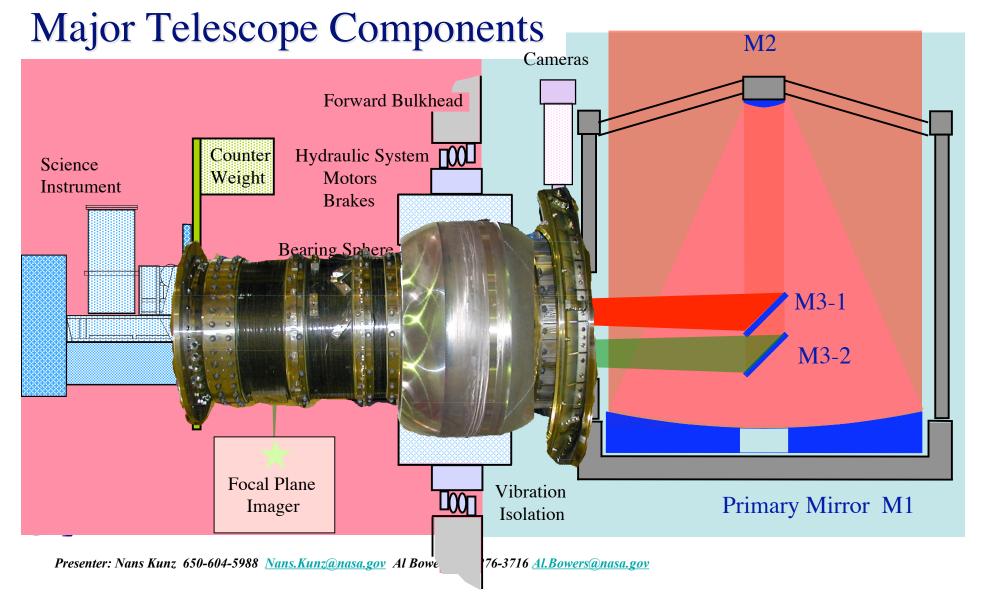
2.5 Meter effective aperture





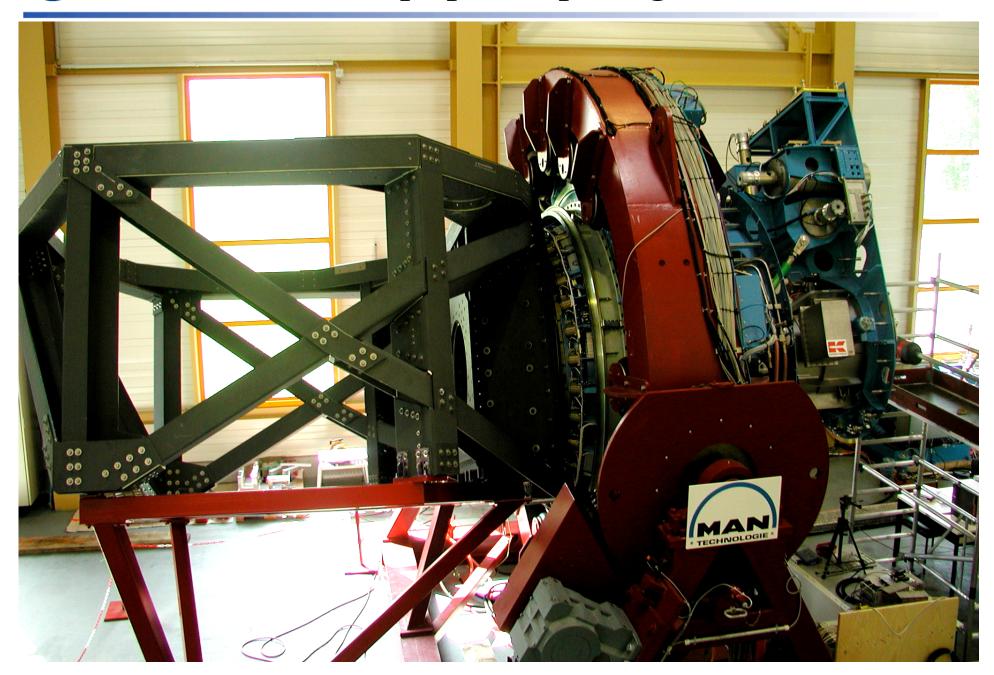
Major Telescope Components



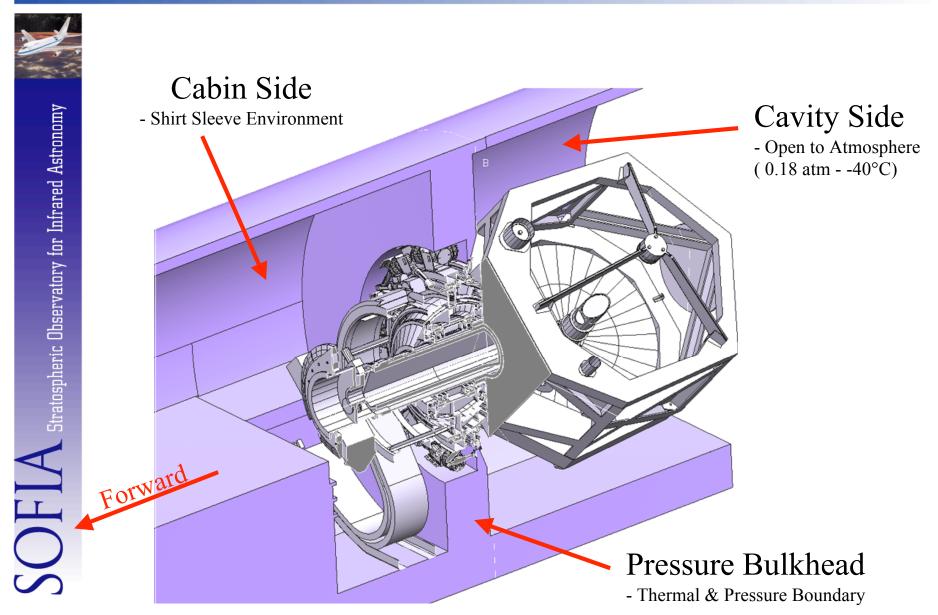




Telescope pre-ship integration



Configuration: Instrument Access in Cabin

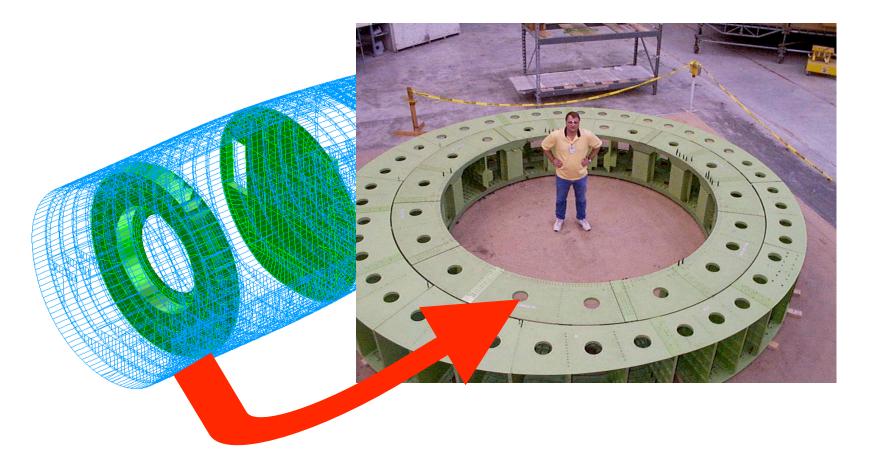






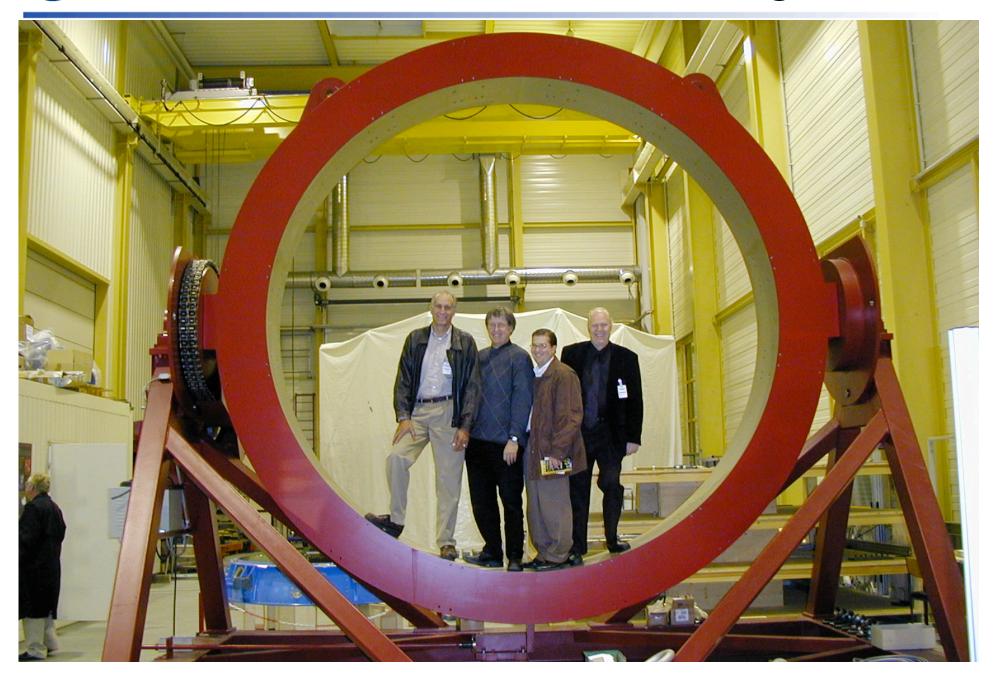
Bulkhead - Flight Hardware

New Pressure Bulkhead



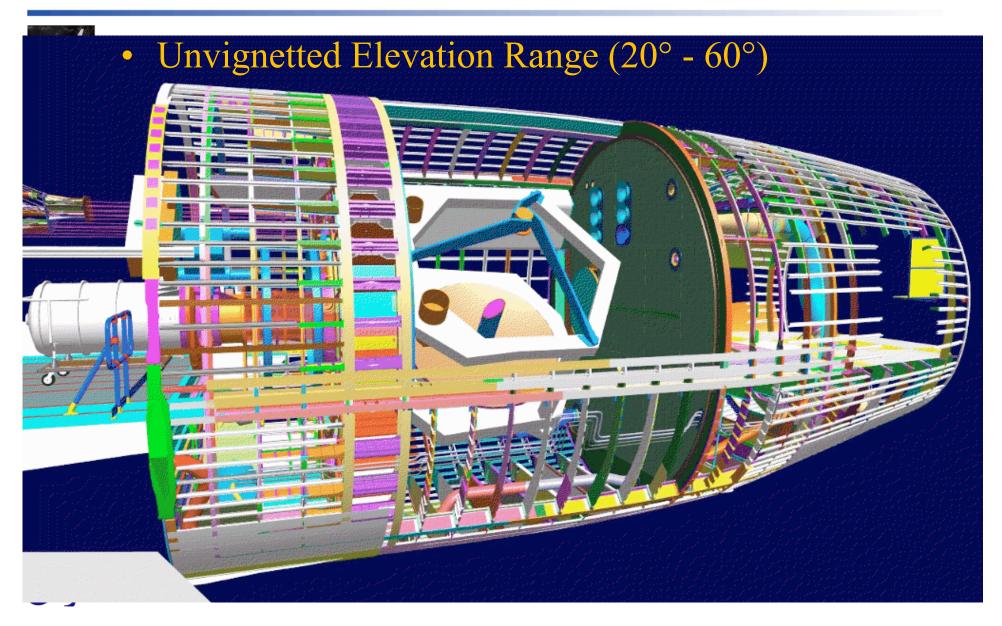


Bulkhead Simulator for TA Integration

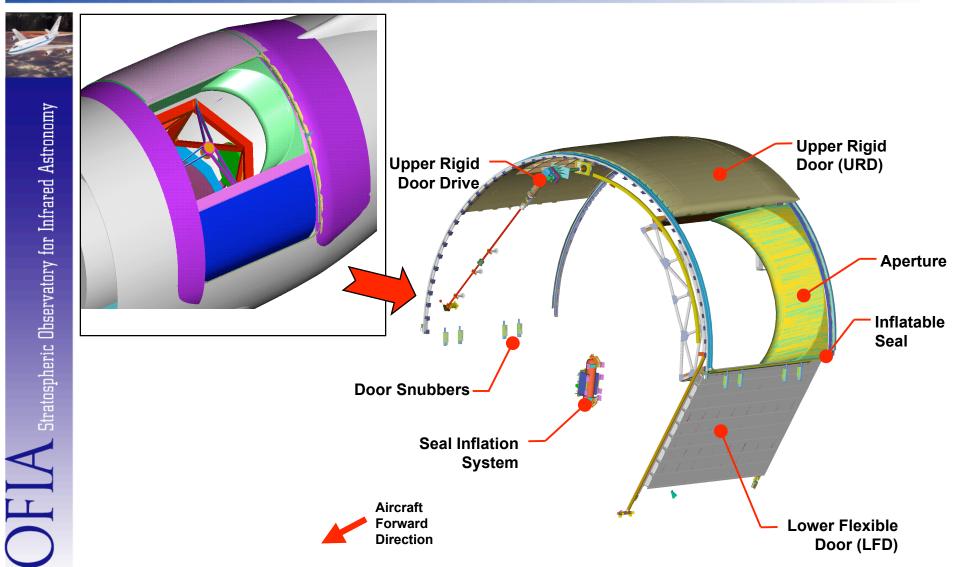




Large Structural Opening



Cavity Door System





Technical Challenges



σ Open Port cavity

- σ = Final Verification pending completion of Flight Tests
 - σ Influence on aircraft Stability & Control
 - - **TRESONANCE**
 - **™** Structural Fatigue
 - **™** Environment for Telescope Performance
 - **™** Drag (aircraft performance)
- Structural Modification
 - Strength
 - Stiffness
 - Transition to unmodified areas



Technical Challenges



- **Thermal Environment**
 - **™** Systems exposure
 - **™** Science performance
- **™** Cavity Door

 - **™** Track Telescope motion
 - ^π Drive system safety
- Lightweight Primary Mirror
- Rotational Isolation System
 - KAO used air bearing but this technology does not scale well...



SOFIA Wind Tunnel Testing Overview



7% Scale Tests

- SOFIA I March 1990 to July 1990 Forward Cavity configuration
- SOFIA II June 1994 to August 1994 Aft Cavity configuration
- SOFIA III February 1995 -SP only -Aperture Geometry -TA loads
- SOFIA IV Sept 1995 to Dec1995 Door design space evaluation
- SOFIA V November 1997
 - Adjustment of Boundary Layer profile to match Baseline Flight tests
 - Verification of Final Partial External Door (PED) Design
 - Measurement of loads on Final Telescope design (pointing performance)
 - Measurement of loads for use in PED design

3% Scale Tests

Stability & Control - measure aero-coefficients between baseline 747-SP and SOFIA and provide substantiation for reduced flight test program

- Low Speed Tests University of Washington Kirsten Wind Tunnel
 - Part 1 Sept 1998 to Oct 1998 & Part 2 Jan 1999 to Feb 1999
- High Speed Tests Boeing Transonic Wind Tunnel
 - November 1998

Administration Ames Research Center SOFIA 7% model in Ames 14ft Transonic Wind Tunnel

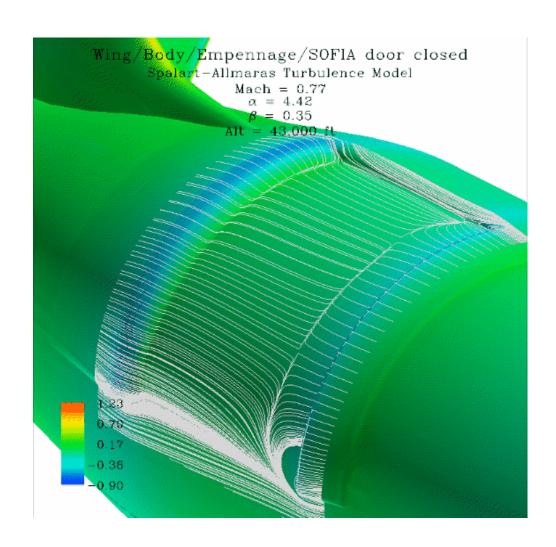


Primarily used to development shear layer control design technology and to determine cavity acoustic environment and resultant loads on Telescope



SOFIA CFD Predictions

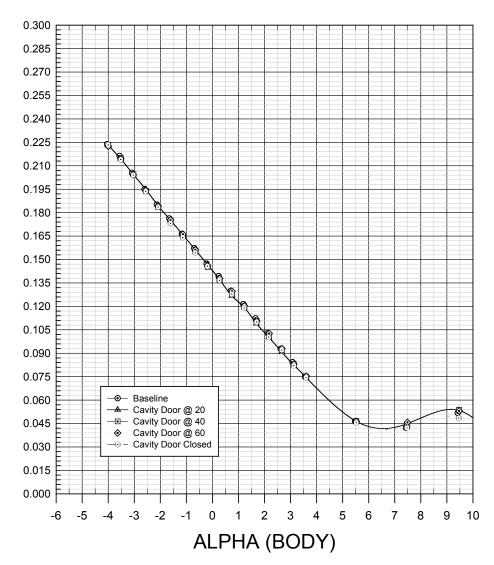
 Example of CFD flow over the mod





SOFIA Wind Tunnel Data

- Stab & Control
- Negligible change in drag and pitching moment
- No other F&Ms affected





Objectives



- Per all Test and Analyses completed, data indicates Objectives will be met
 - Minimal impact on Stability & Control of Aircraft
 - Robust Non resonating cavity (structural/safety)
 - "Quiet" cavity for optimum TA pointing performance
 - Minimize drag to maintain Aircraft performance
 - Optimize Aero-Optic performance "seeing" for short wave length image quality performance
- Flight Testing is remaining step to Verify



Summary



- SOFIA SLC development began with KAO heritage
- Open port cavity/SLC issues identified early (1980's) as risk areas
 - Risk reduction activities were planned & completed accordingly
- Eight Separate Wind Tunnel Test Series Completed
 - Results Indicate:
 - Shear layer control implementation will provide quiet well behaved cavity acoustic environment
 - Stability & Control of aircraft will be essentially unaffected
- Multiple CFD and other analyses completed
 - Results concur with wind tunnel tests and provide additional data
- Multiple Independent Reviews Concur with approach
 - Latest NESC review extensively examined test and analysis data and planned program approach and recommend proceeding to flight test
- All data indicates that SOFIA will fly like an unmodified 747-SP







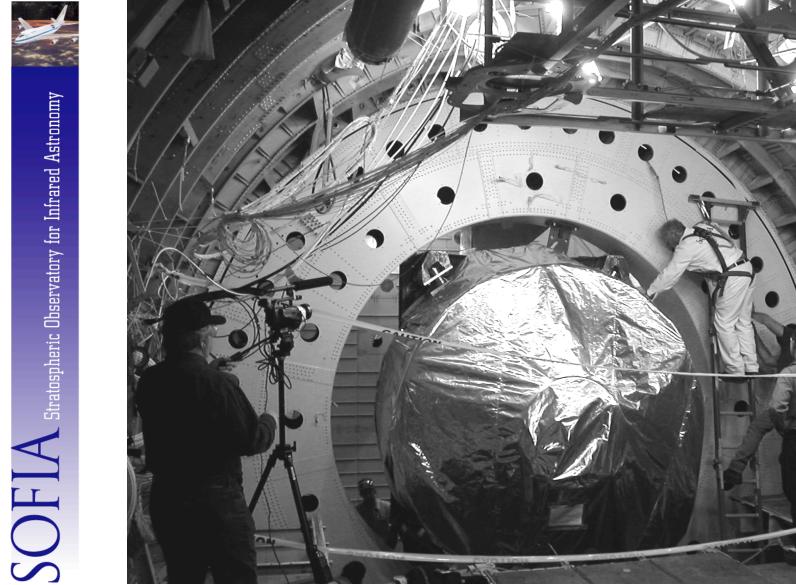


Inside aircraft just before SUA installation

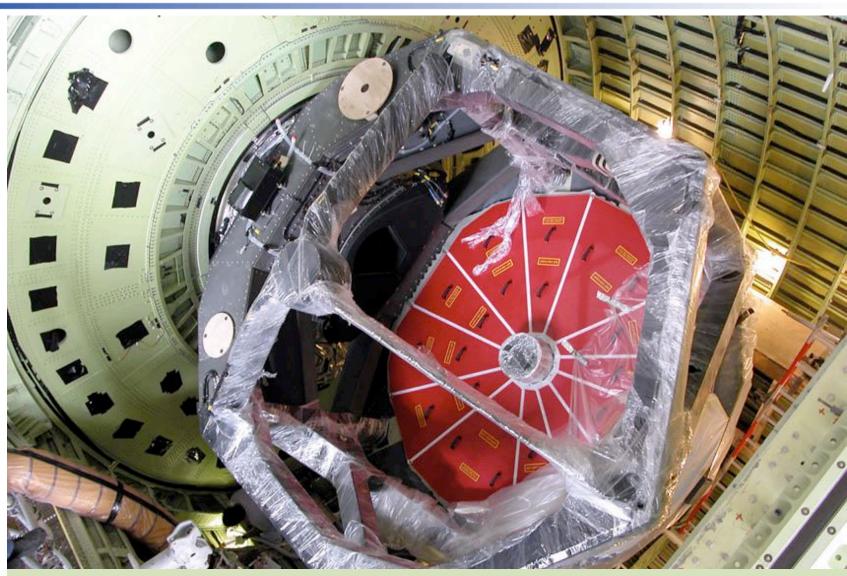




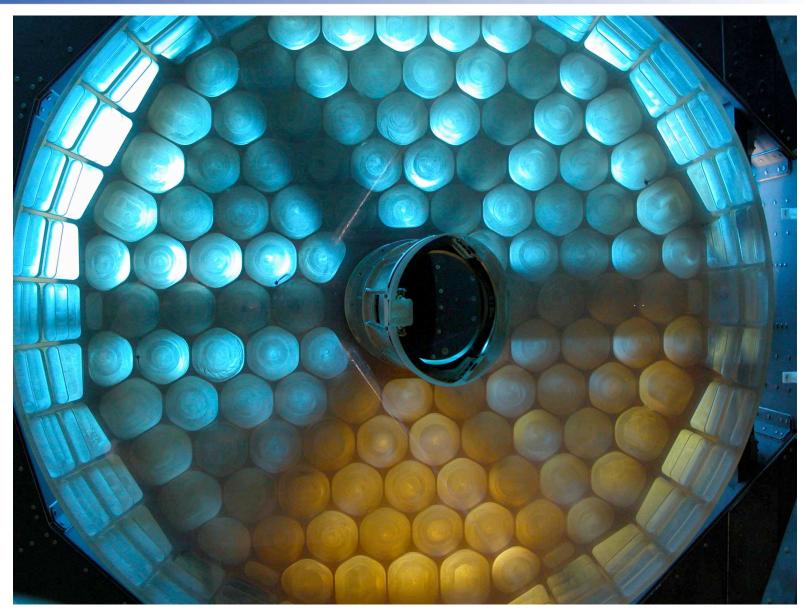








Telescope inside Aircraft Cavity











Roll out from paint hangar September 2006







First Flight









Ferry to Dryden







Questions?



National Aeronautics and Space

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