



Conflict-Aware Scheduling Algorithm

An algorithm is being developed to automate NASA's Deep Space Network antenna allocation.

NASA's Jet Propulsion Laboratory, Pasadena, California

A conflict-aware scheduling algorithm is being developed to help automate the allocation of NASA's Deep Space Network (DSN) antennas and equipment that are used to communicate with interplanetary scientific spacecraft. The current approach for scheduling DSN ground resources seeks to provide an equitable distribution of tracking services among the multiple scientific missions and is very labor intensive. Due to the large (and increasing) number of mission requests for DSN services, combined with technical and geometric constraints, the DSN is highly oversubscribed. To help automate the process, and reduce the DSN and spaceflight project labor effort required for initiating, maintaining, and negotiating schedules, a new scheduling algorithm is being developed.

The scheduling algorithm generates a "conflict-aware" schedule, where all requests are scheduled based on a dynamic priority scheme. The conflict-aware scheduling algorithm allocates all requests for DSN tracking services while identifying and maintaining the conflicts to facilitate collaboration and negotiation between spaceflight missions. These contrast with traditional "conflict-free"

scheduling algorithms that assign tracks that are not in conflict and mark the remainder as unscheduled. In the case where full schedule automation is desired (based on mission/event priorities, fairness, allocation rules, geometric constraints, and ground system capabilities/constraints), a conflict-free schedule can easily be created from the conflict-aware schedule by removing lower priority items that are in conflict.

Unlike most existing scheduling engines that require fixed length schedule items in the request, the conflict-aware schedule provides a dynamic scheduling engine to determine allocation length during the scheduling process. This is made necessary by the variety of mission-tracking request types faced by the DSN. In addition to fixed track requests, missions may also need continuous coverage or may need to segment a track related to multiple ground assets to support a given request for service. In these cases, the schedule allocation length (time) will depend on the availability of each resource.

The conflict-aware scheduling algorithm combines scheduling heuristics, optimization, a search algorithm, and computational intelligence. At the be-

ginning of the procedure, all requests pass through a scoring system (chosen from a simple mathematical equation or fuzzy logic) that determines the priority of each request on the basis of measures of fairness of the allocation, importance of the request, the type of request, and the allocation length. Starting with the highest priority request, all technical and geometric constraints are combined to determine the available "timeline/antenna groups" for scheduling. A scoring system that considers items already in the schedule and the request characteristics then identifies the best timeline/antenna group and start times for each request. This then continues for each successive priority request (priority is recomputed dynamically) until all requests are scheduled.

The conflict-aware algorithm is not limited to DSN application. It can also be applicable to solution of scheduling problems in areas such as manufacturing and traffic control.

This work was done by Yeou-Fang Wang and Chester Borden for NASA's Jet Propulsion Laboratory. Further information is contained in a TSP (see page 1). NPO-41320

Real-Time Diagnosis of Faults Using a Bank of Kalman Filters

Gradual changes associated with aging are taken into account in the diagnostic process.

John H. Glenn Research Center, Cleveland, Ohio

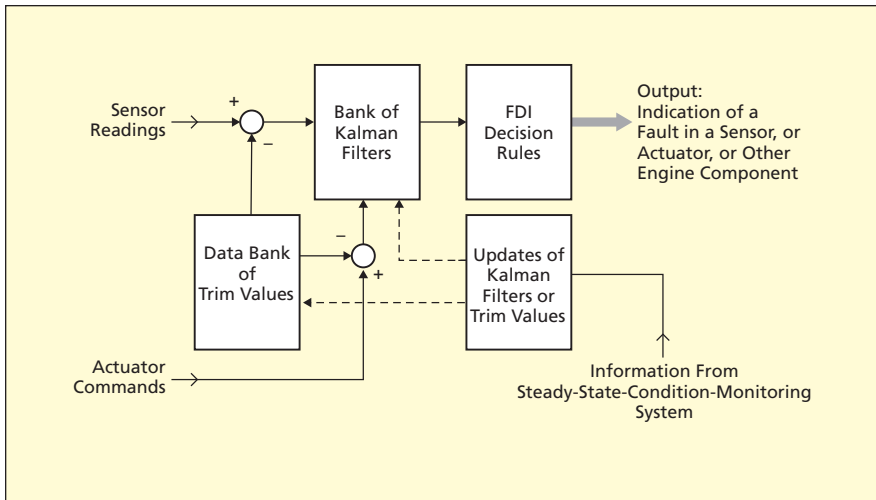
A new robust method of automated real-time diagnosis of faults in an aircraft engine or a similar complex system involves the use of a bank of Kalman filters. In order to be highly reliable, a diagnostic system must be designed to account for the numerous failure conditions that an aircraft engine may encounter in operation. The method achieves this objective through the utilization of multiple Kalman filters, each of which is uniquely designed based on a specific failure hypothesis. A fault-detection-and-isolation (FDI) system, developed based on this method, is able to isolate faults in sensors and actuators while

detecting component faults (abrupt degradation in engine component performance). By affording a capability for real-time identification of minor faults before they grow into major ones, the method promises to enhance safety and reduce operating costs.

The robustness of this method is further enhanced by incorporating information regarding the aging condition of an engine. In general, real-time fault diagnostic methods use the nominal performance of a "healthy" new engine as a reference condition in the diagnostic process. Such an approach does not ac-

count for gradual changes in performance associated with aging of an otherwise healthy engine. By incorporating information on gradual, aging-related changes, the new method makes it possible to retain at least some of the sensitivity and accuracy needed to detect incipient faults while preventing false alarms that could result from erroneous interpretation of symptoms of aging as symptoms of failures.

The figure schematically depicts an FDI system according to the new method. The FDI system is integrated with an engine, from which it accepts two



This **FDI System** accepts information on gradual aging of an engine to retain its effectiveness in fault diagnosis.

sets of input signals: sensor readings and actuator commands. Two main parts of the FDI system are a bank of Kalman filters and a subsystem that implements FDI decision rules. Each Kalman filter is

designed to detect a specific sensor or actuator fault. When a sensor or actuator fault occurs, large estimation errors are generated by all filters except the one using the correct hypothesis. By monitor-

ing the residual output of each filter, the specific fault that has occurred can be detected and isolated on the basis of the decision rules.

A set of parameters that indicate the performance of the engine components is estimated by the “correct” Kalman filter for use in detecting component faults. To reduce the loss of diagnostic accuracy and sensitivity in the face of aging, the FDI system accepts information from a steady-state-condition-monitoring system. This information is used to update the Kalman filters and a data bank of trim values representative of the current aging condition.

This work was done by Takahisa Kobayashi of QSS Group, Inc. and Donald L. Simon of Army Research Laboratory for Glenn Research Center. Further information is contained in a TSP (see page 1).

Inquiries concerning rights for the commercial use of this invention should be addressed to NASA Glenn Research Center, Commercial Technology Office, Attn: Steve Fedor, Mail Stop 4-8, 21000 Brookpark Road, Cleveland, Ohio 44135. Refer to LEW-17457.