



Western Kentucky University



Lunabotics Mining:  
Evolution of A.R.T.E.M.I.S. PRIME

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# Western Kentucky University

A leading American university with international reach



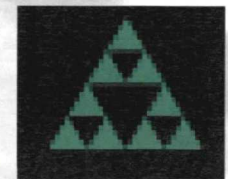
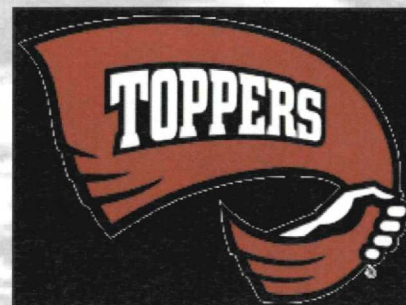
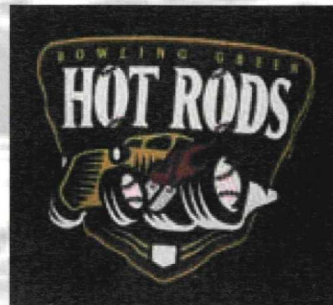
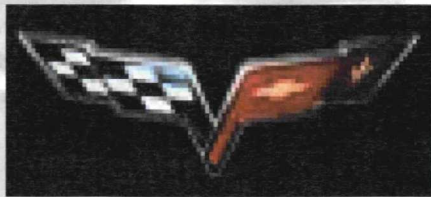
Located in Bowling Green, KY

- Between Louisville (110 miles) and Nashville (70 miles)
- Home of the Corvette Plant and National Corvette Museum
- Home of the minor league baseball Hot Rods

20,000+ WKU students

450+ Undergraduate Engineering students in CE, EE, and ME programs

Home of the ***Carol Martin Gatton Academy of Mathematics and Science in Kentucky***



A massing  
R egolith  
with  
T opper  
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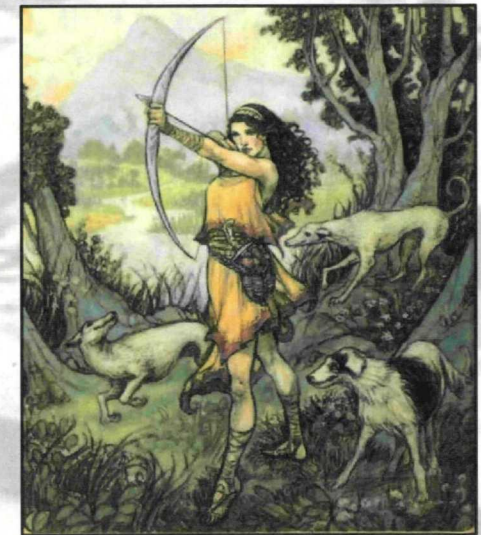
# ARTEMIS

## •A Multidisciplinary Team

- Electrical Engineering
- Mechanical Engineering
- Civil Engineering
- Physics
- Mathematics and Science (Gatton student)

## •All Female Members

- Unusual in Engineering
- Name inspired by the powerful Greek moon goddess, Artemis



# A.R.T.E.M.I.S. PRIME

Excavator Name: *A.R.T.E.M.I.S. PRIME*

Mass: **76.5 kg**

Max height: **1.5 meters**

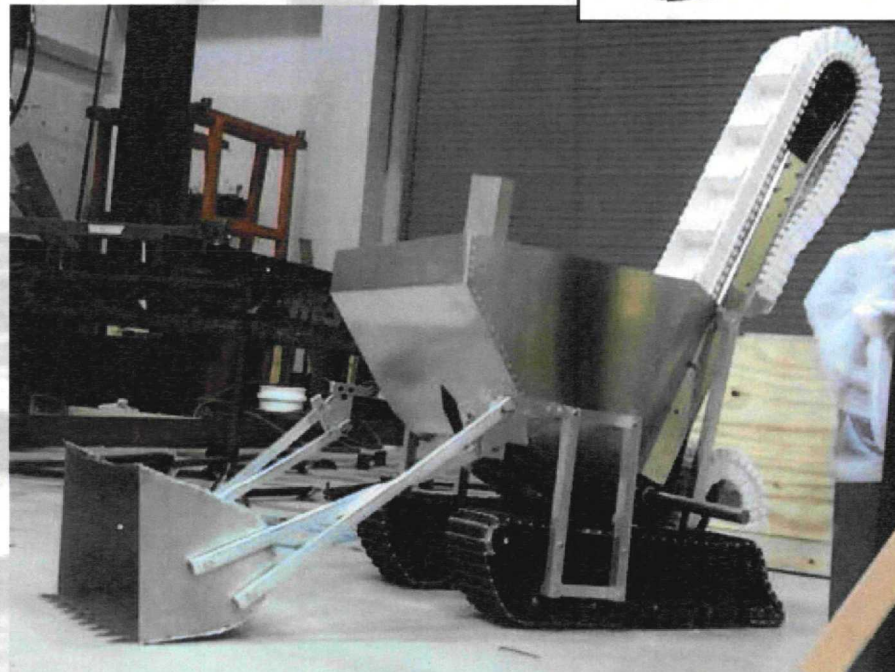
Max width: **0.75 meters**

Max length: **1.5 meters**

Regolith capacity: **113 kg**

Born: **Spring 2010**

Hobbies: **Lunar Mining**



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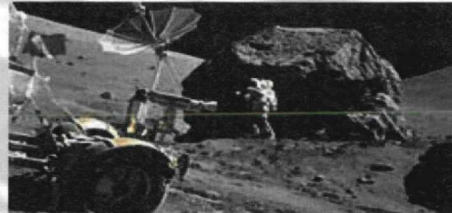
System Modeling

Preliminary Testing

System Integration

Risk Assessment

Performance Evaluation



### **Lunabotics Mining Competition**

“The purpose of the Lunabotics Mining Competition is to engage and retain students in science, technology, engineering, and mathematics, or STEM, in competitive environment that may result in innovative ideas and solutions, which could be applied to actual lunar excavation for NASA.”

-NASA Exploration Systems Mission Directorate Higher Education Project and the National Space Grant College and Fellowship Program

### **Student Design Team**

WKU’s A.R.T.E.M.I.S. team was formed in response to the announcement of the inaugural NASA Lunabotics Mining Competition. Competing will allow WKU engineering students to develop and be engaged in project-based learning activities by fulfilling the competition requirements, to proudly represent WKU at the Kennedy Space Center, contributing innovative ideas and solutions through a systems engineering approach, and to advance the fields of engineering through K-12 outreach programs.

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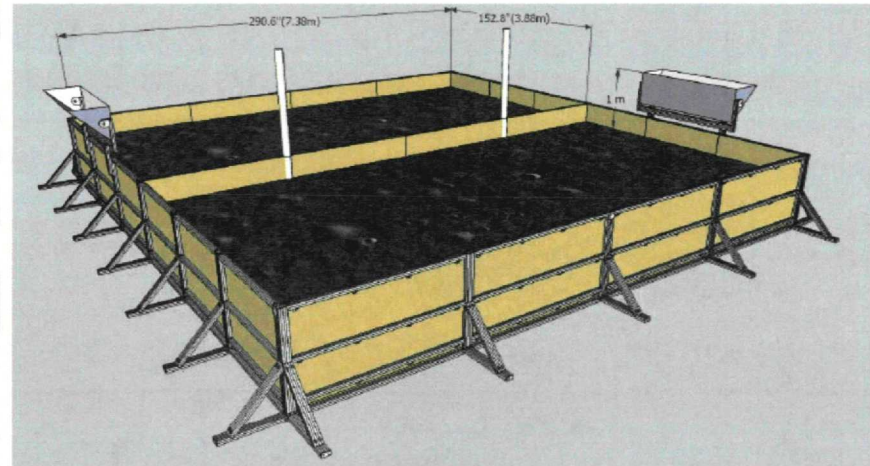
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# Statement of Lunabotics Problem

•Design, build, and operate a remotely controlled device that is capable of excavating, transporting, and discharging lunar regolith simulant in a lunar environment over a 15-minute period.

## Key NASA Specifications:

- ✓ Maximum dimensions: 2m x 0.75m x 1.5m
- ✓ Maximum hardware mass: 80.0 kg
- ✓ Communication bandwidth: <5.0 Mbps
- ✓ All power must be provided onboard
- ✓ Hardware may not change the physical or chemical properties of the simulant
- ✓ Processes or materials used must be appropriate for the lunar surface



Sandbox Diagram (side view)

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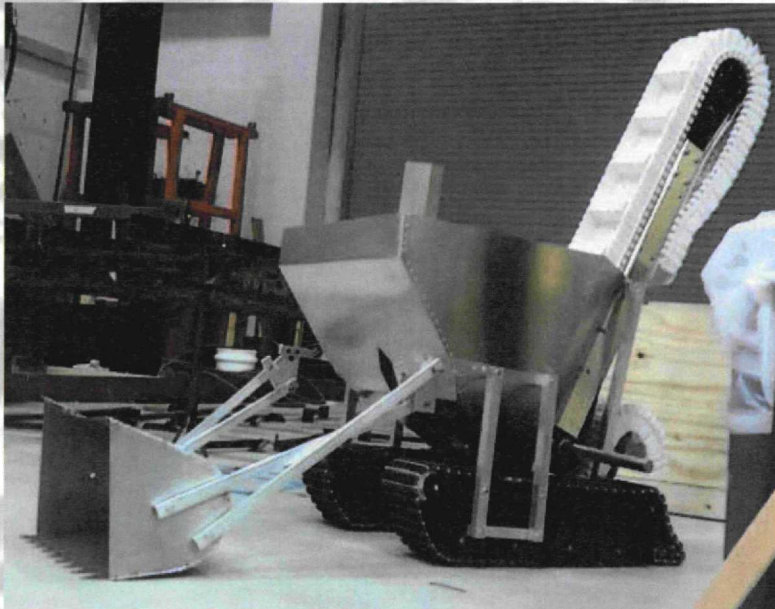
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# Statement of Lunabotics Problem



## A.R.T.E.M.I.S. Criteria:

- ✓ Maximize regolith carrying capacity
- ✓ Minimize complexity of system
- ✓ Minimize regolith contamination of moving and electrical components
- ✓ Minimize regolith loss
- ✓ Optimize battery capacity/weight
- ✓ Reduce costs if quality is not sacrificed

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# The Major System Functions

**DRIVE**

Transportation across simulated lunar surface

Avoidance of obstacles

**DIG**

Removal of regolith from competition surface

Storage of collected regolith on/in device

**DEPOSIT**

Removal of regolith from device storage

Placement of regolith in provided collection bin



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## Drive Function Decisions

- Wheels vs. Track Systems
- Required power
- Maneuverability; ability to handle obstacles
- Fabricate at WKU vs. professional manufacturing with enhanced reliability



### Evaluation Matrix

<b>Wheels vs. Tracks</b>	<b>Cost</b>	<b>Weight (3x)</b>	<b>Time</b>	<b>Resources</b>	
<b>Wheels</b>	<b>+</b>	<b>-</b>	<b>-</b>	<b>-</b>	<b>-4</b>
	<b>\$1,300</b>	<b>~65 lbs</b>	<b>&gt;1 week</b>		
<b>Tracks</b>	<b>0</b>	<b>0</b>	<b>0</b>	<b>0</b>	<b>0</b>
	<b>\$2,000</b>	<b>~52 lbs</b>	<b>1 week</b>	<b>none needed</b>	

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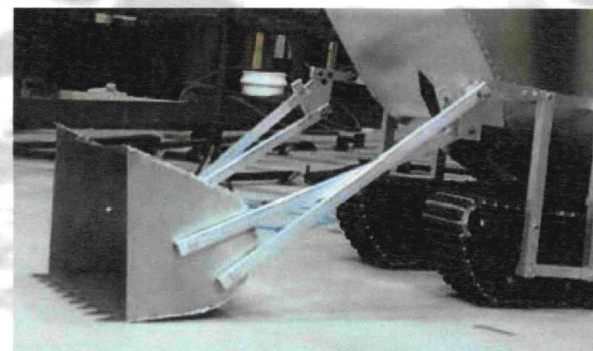
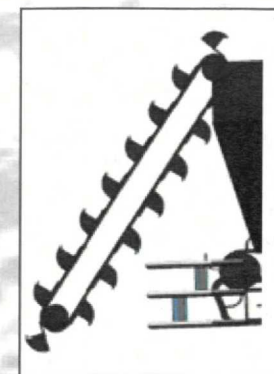
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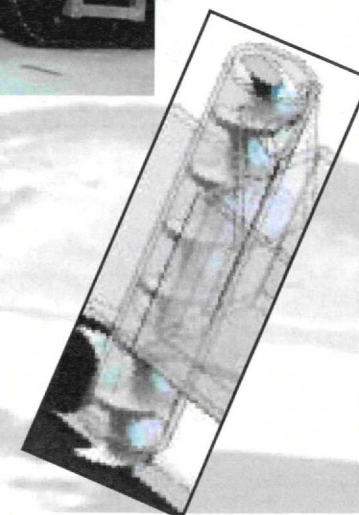
# Dig Function Decisions

- Auger vs. conveyor vs. scoop
- Regolith excavation and storage in hopper
- Required power
- Weight and integration challenges



## Evaluation Matrix

DIG	Cost	Efficiency (2x)	Weight (2x)	Design Time	Construct Time	Power Needs	TOTAL
Auger	0	0	0	0	0	0	0
Conveyor	+	+	-	+	0	0	2
Scoop	0	+	+	-	-	+	3



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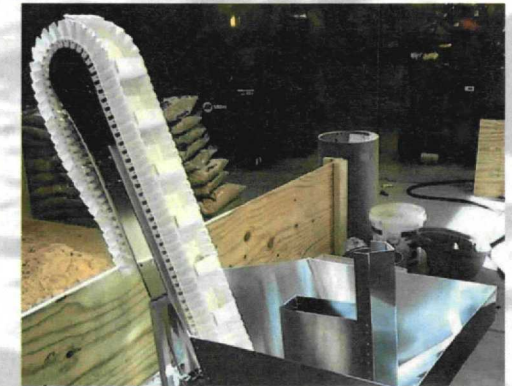
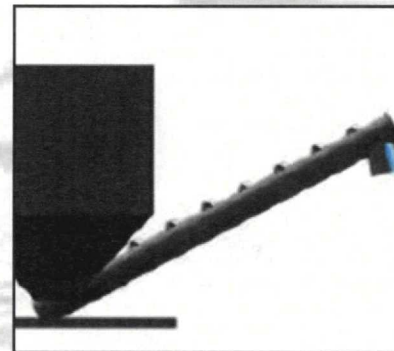
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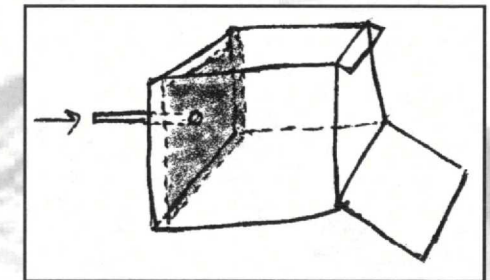
# Deposit Function Decisions

- Piston vs. Auger vs. conveyor
- Integration into hopper
- Weight
- Enhanced reliability, professional manufacturing
- Decreased construction time
- Required power



## Evaluation Matrix

DEPOSIT	Cost	Efficiency (2x)	Weight (2x)	Design Time	Construct Time	Power Needs	TOTAL
Auger	0	0	0	0	0	0	0
Conveyor	+	0	-	+	+	0	1
Push Out	0	-	+	-	-	+	-1



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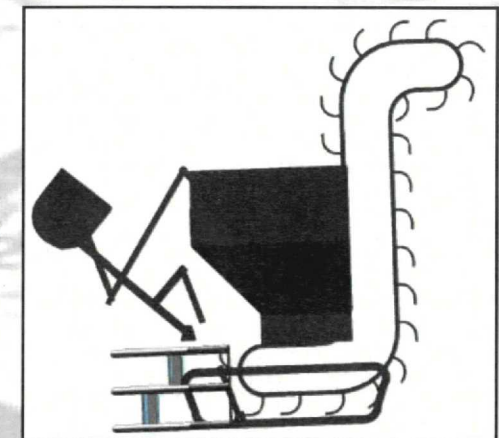
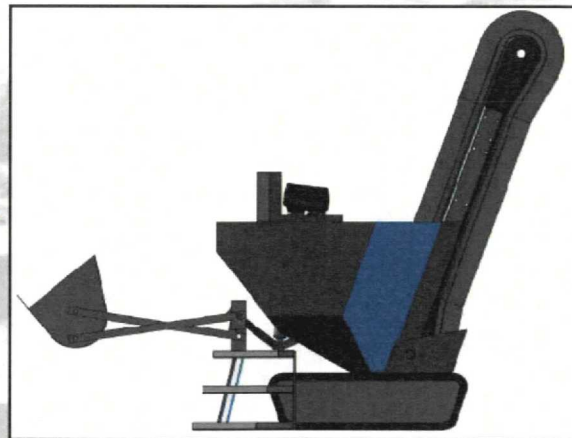
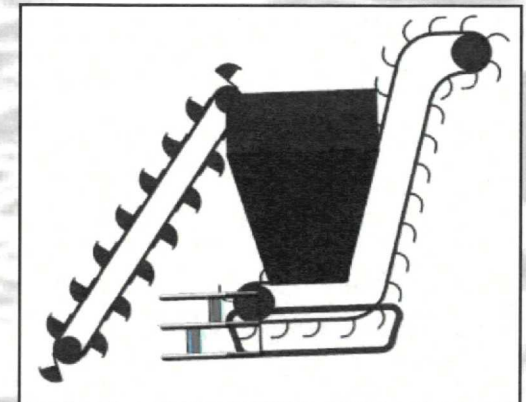
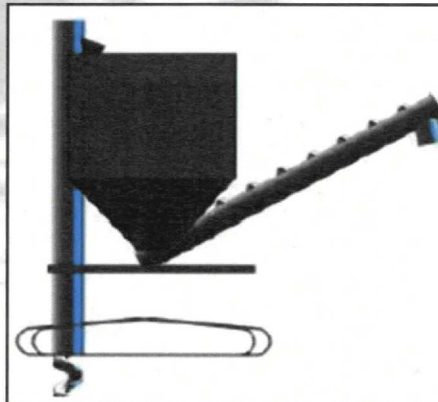
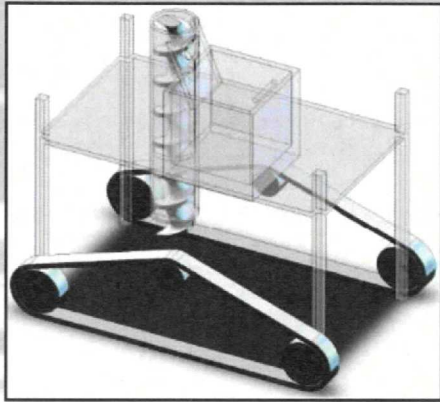
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# Mechanical Design Iterations



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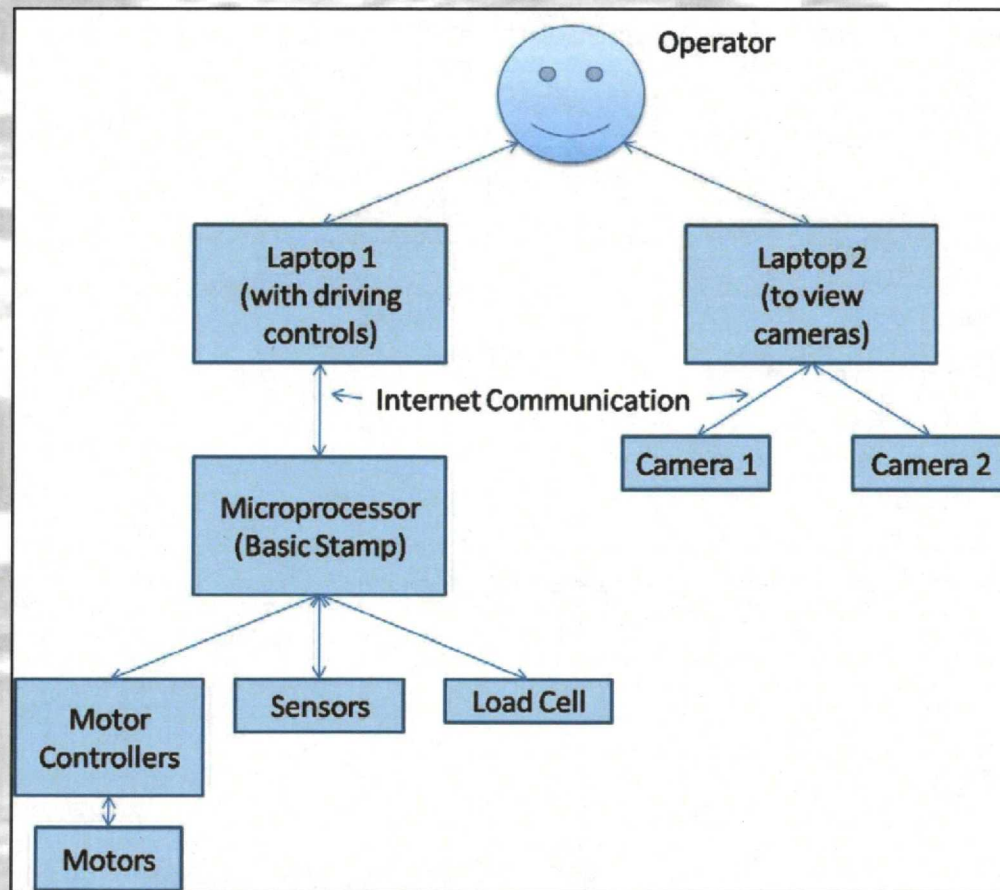
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# Iterative Design Process: Electrical System Design



## Design Decisions:

Considering the distance between the operator and the device at the competition, an IP control system was chosen over a remote control system for reliability.

Before the mechanical system was complete, electrical design began with a schematic of the basic communication process.

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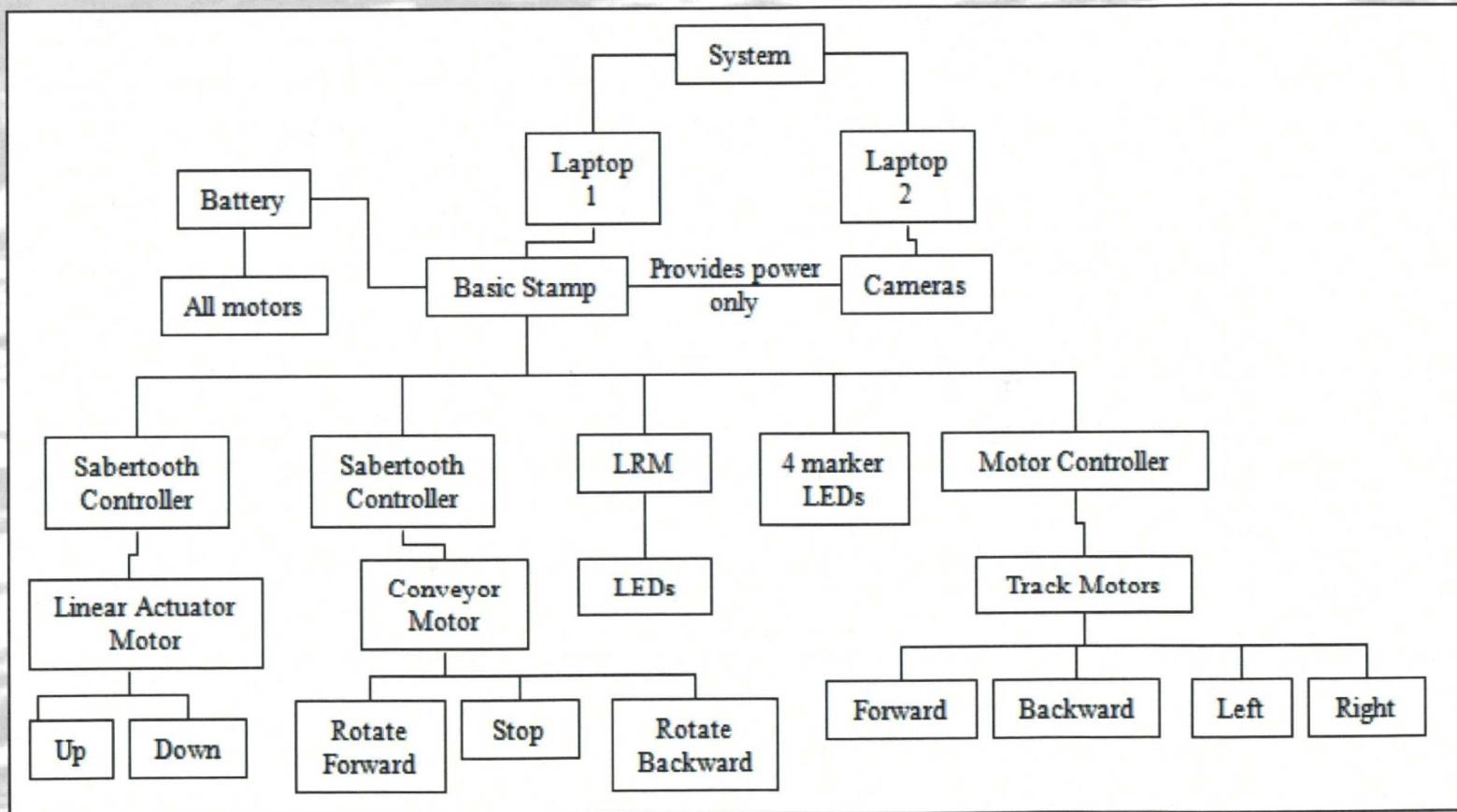
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# Electrical System Design



After mechanical components were chosen, the final schematic of control system was created.

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# Hardware Requirements

**Objective:** Simplicity of system

**Solution:** Uniform voltage in system

**Implementation:** Track motors (24V) dictated the voltage of linear actuator and conveyor motors.

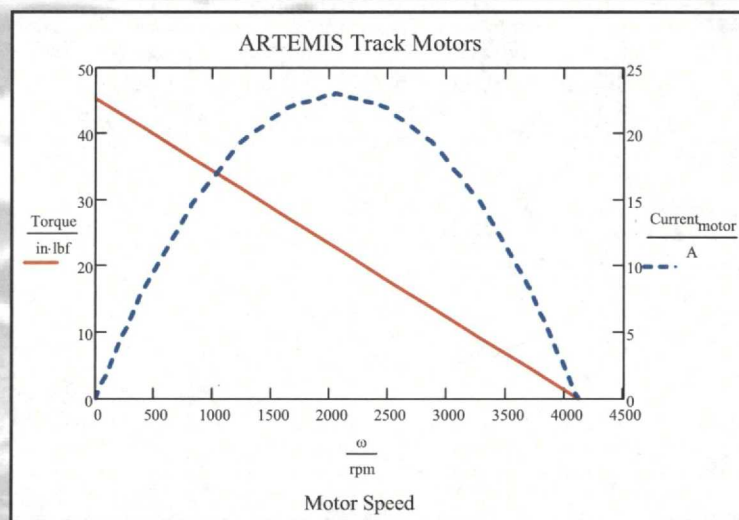
**Evaluation:** Voltage was easily accommodated for the linear actuator and conveyor motors.

**Objective:** On-board power system

**Solution:** Rechargeable battery

**Implementation:** Design parameters included max drawn current, operation time of each component in a cycle, and ultimate amp-hour requirements.

**Evaluation:** Initial battery choice proved adequate but exceeded the weight budget. Lithium ion battery was chosen for weight, amp-hour reliability, and peak current accommodation, as these design factors were more significant than cost.



Left: Lead Acid  
Below: Lithium Ion



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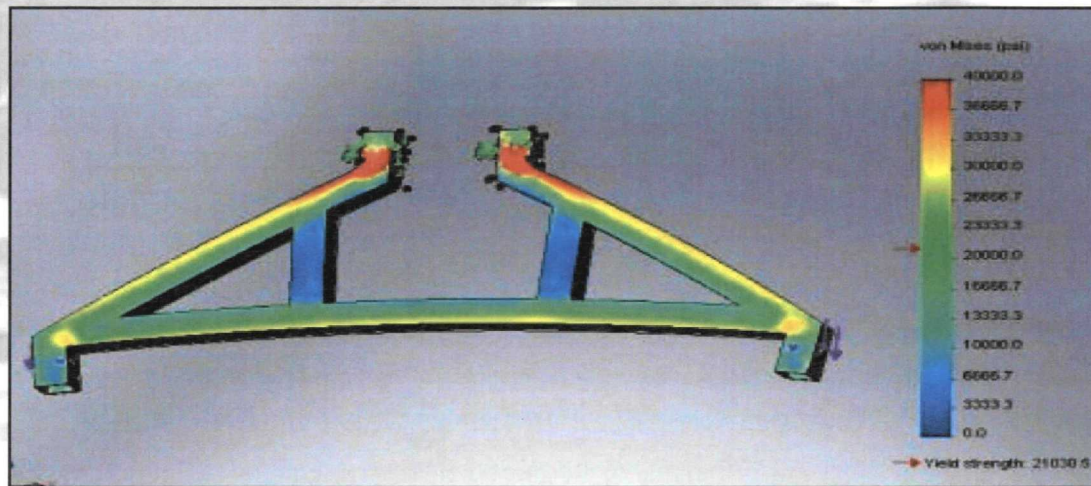
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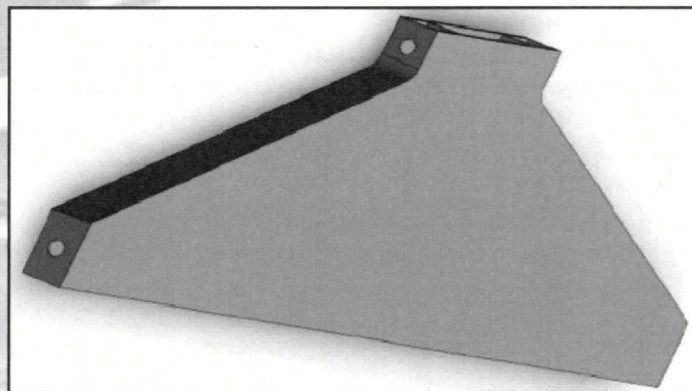
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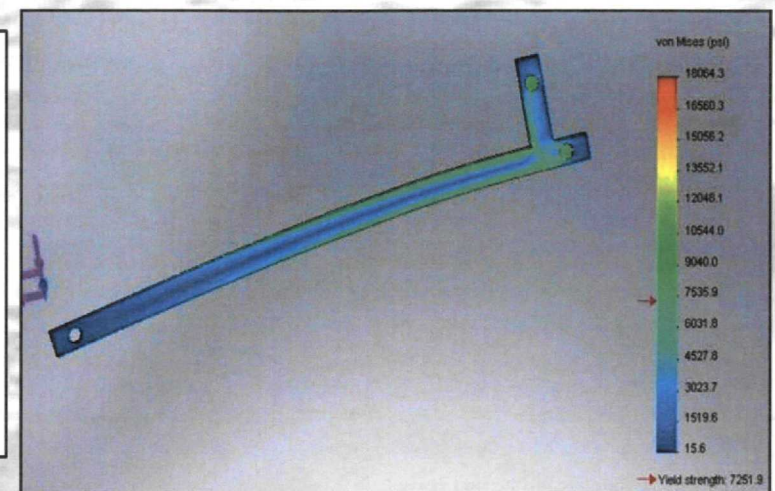
# System Modeling



A finite element analysis (FEA) was performed using *SolidWorks* to determine maximum stresses the supports must withstand. Analysis of the initial design led to the manufacture of the final support brace.



Left: Final support brace  
Right: FEA of scoop arm confirmed the design was adequate.





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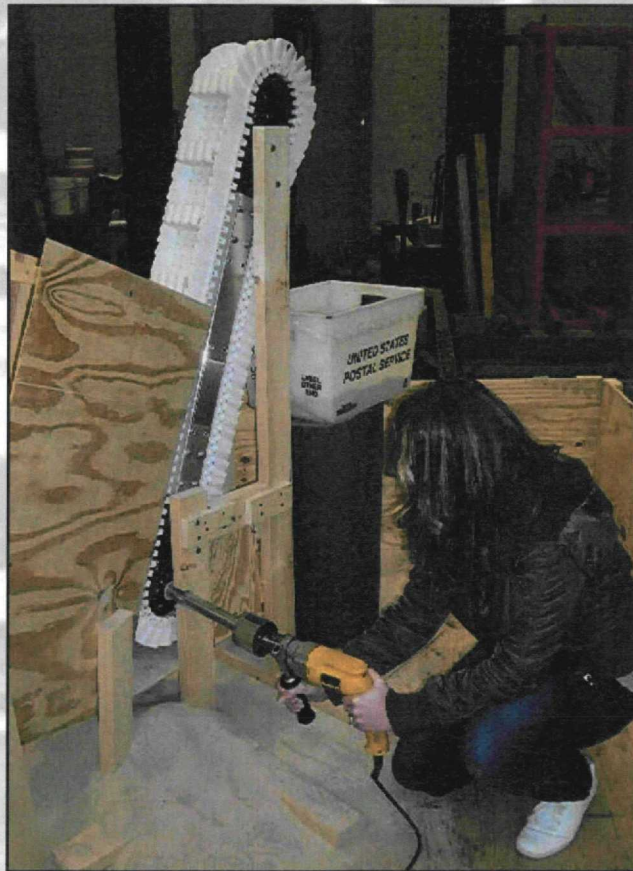
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## Preliminary Testing



A mock-up of the conveyor/ hopper assembly was created for regolith retention tests. The team also experimented with various delivery angles for the conveyor deposit system.



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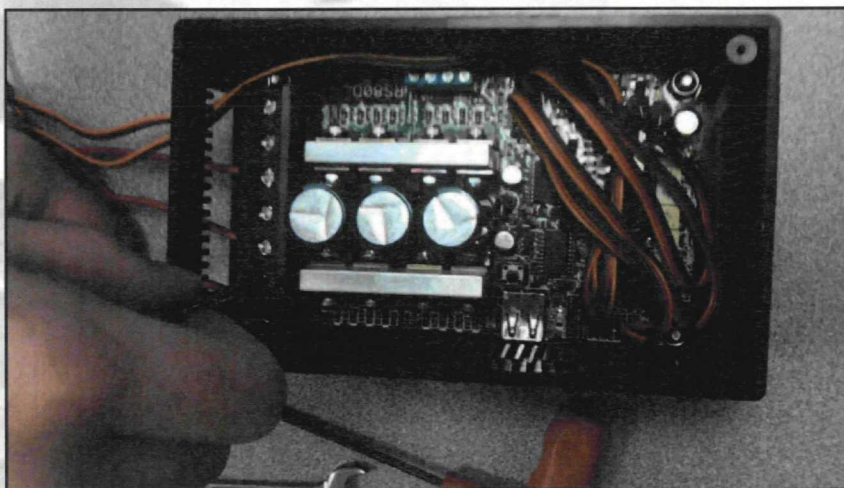
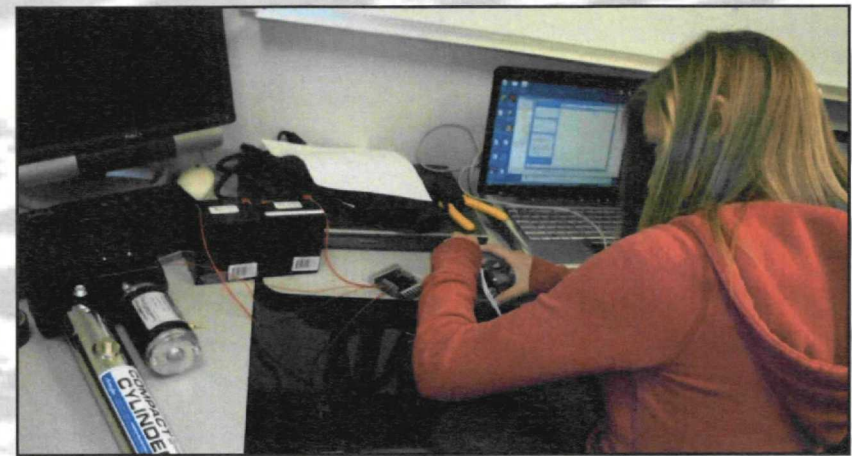
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## Preliminary Testing



Top left: A platform was assembled to test the tracks for uniform speed and power.

Top right: Testing Sabertooth motor controller for compatibility with the linear actuator motor.

Left: Wiring RS80D motor controller for tracks compatibility testing

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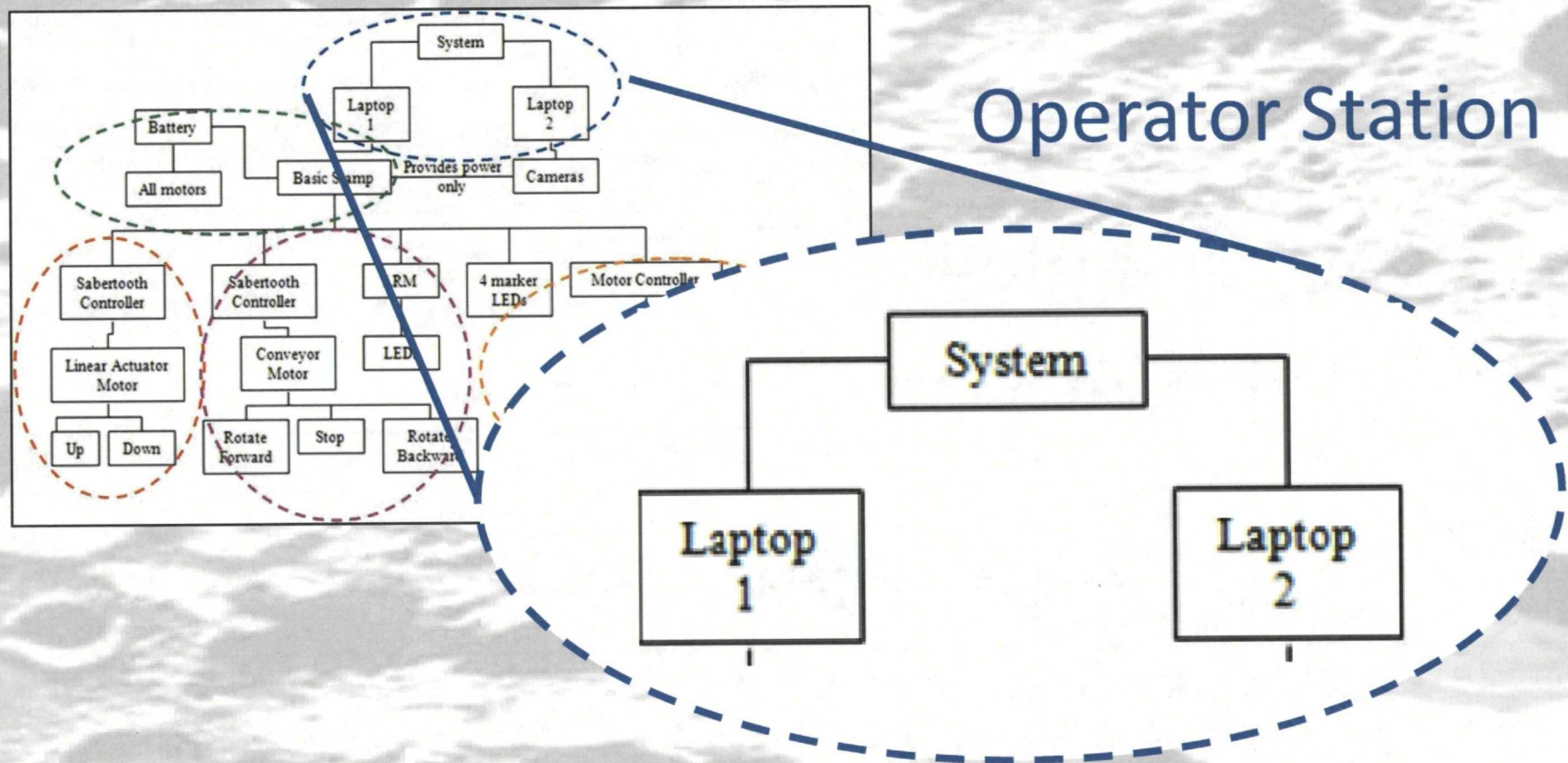
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# System Integration



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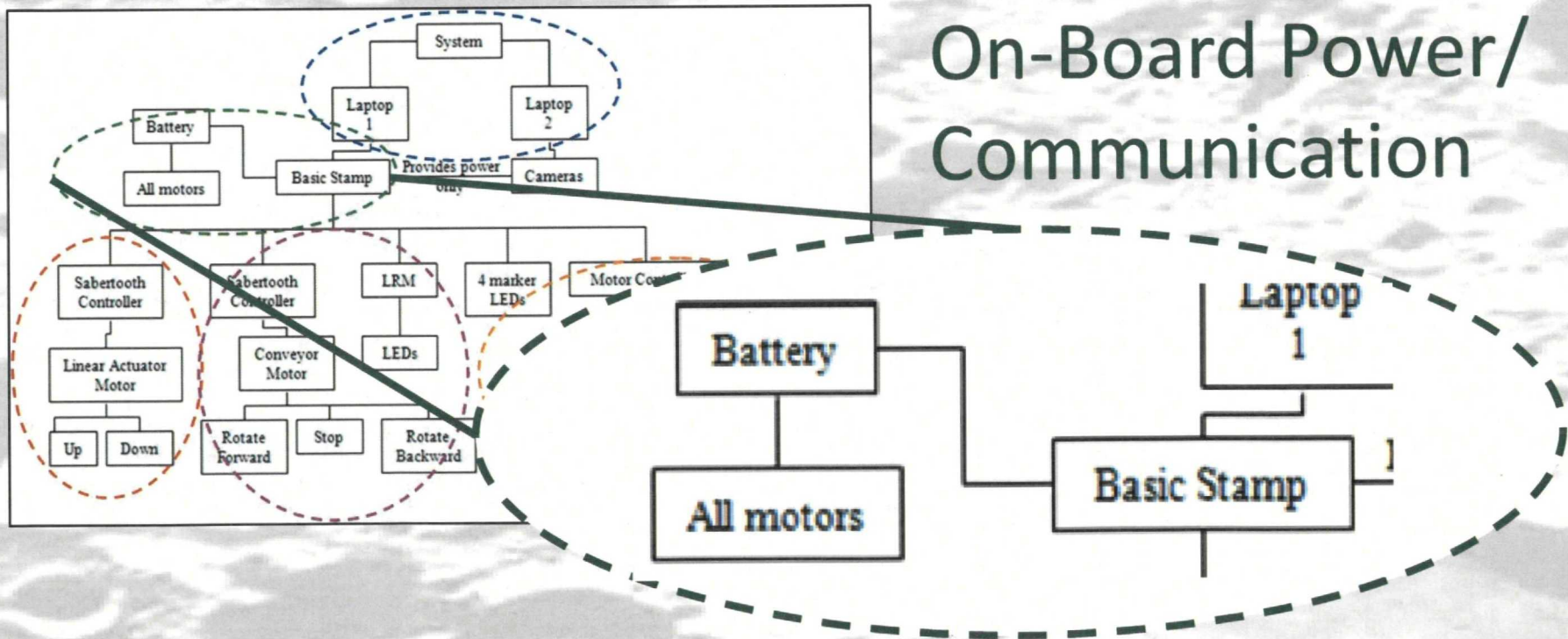
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# System Integration

## On-Board Power/ Communication



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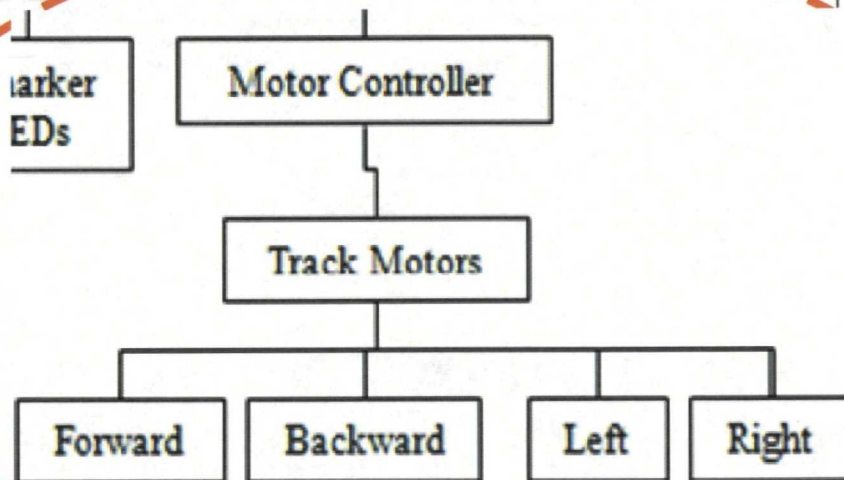
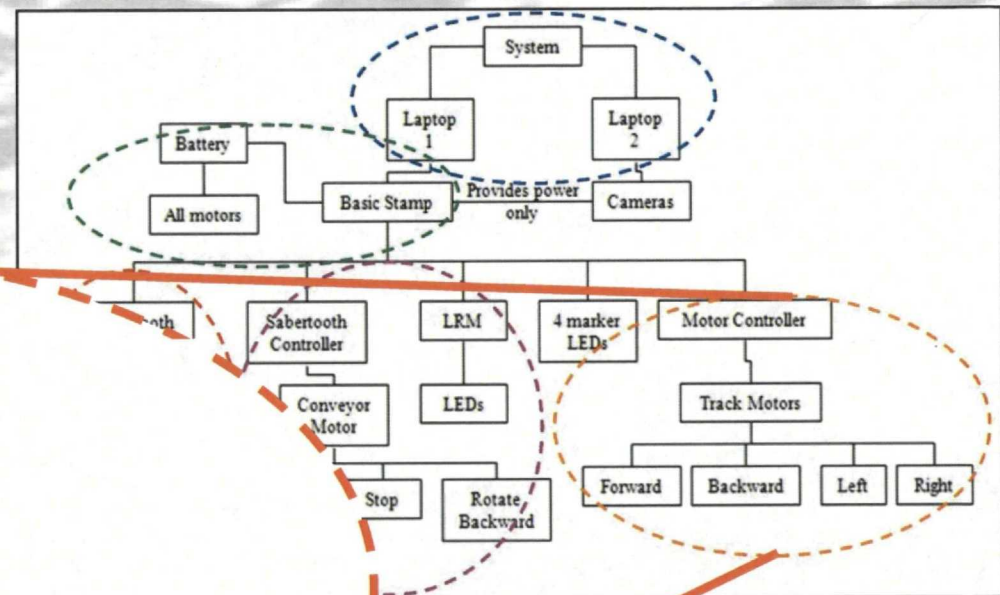
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# System Integration

Drive  
Function  
(Tracks)



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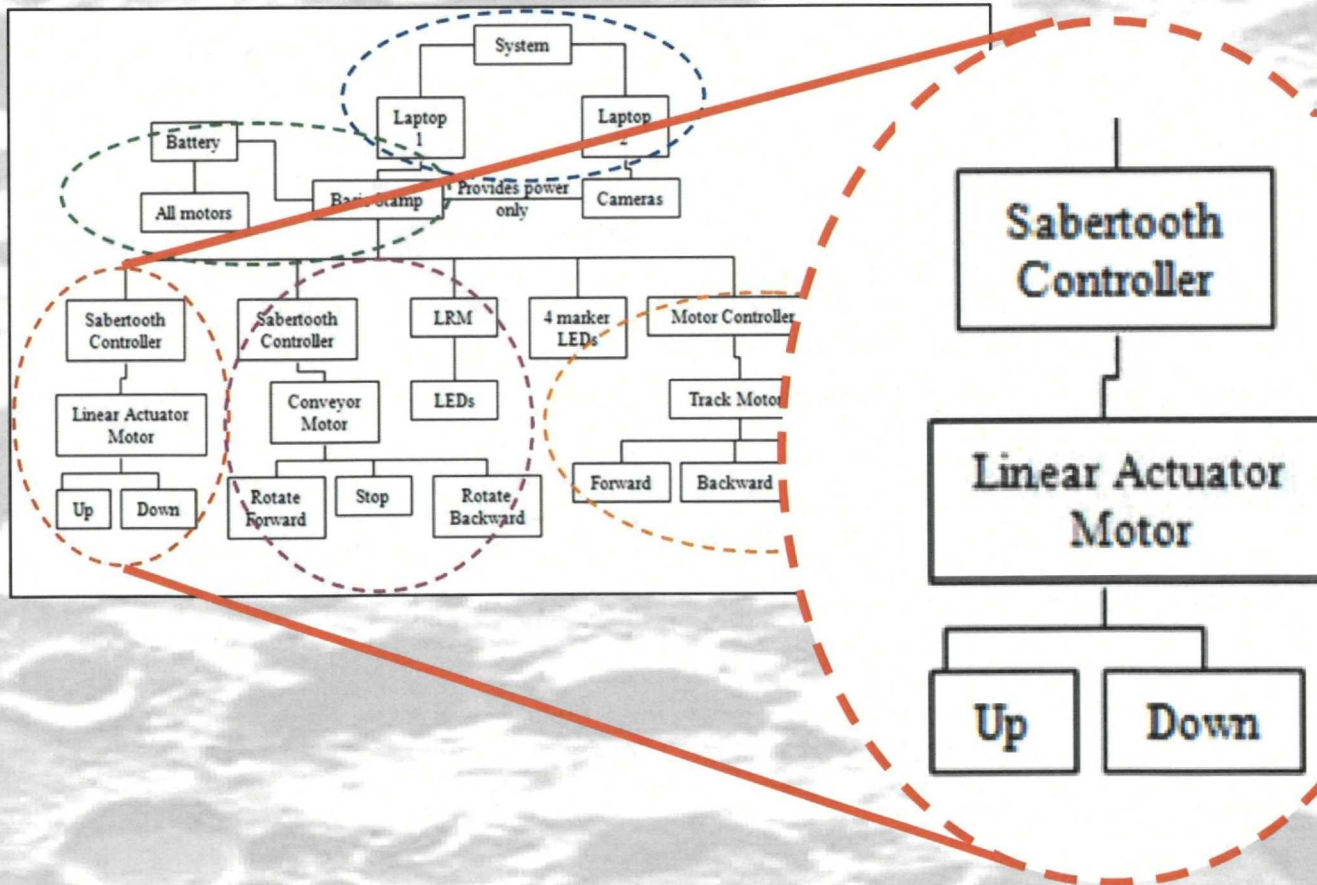
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# System Integration

DIG Function  
(Scoop)



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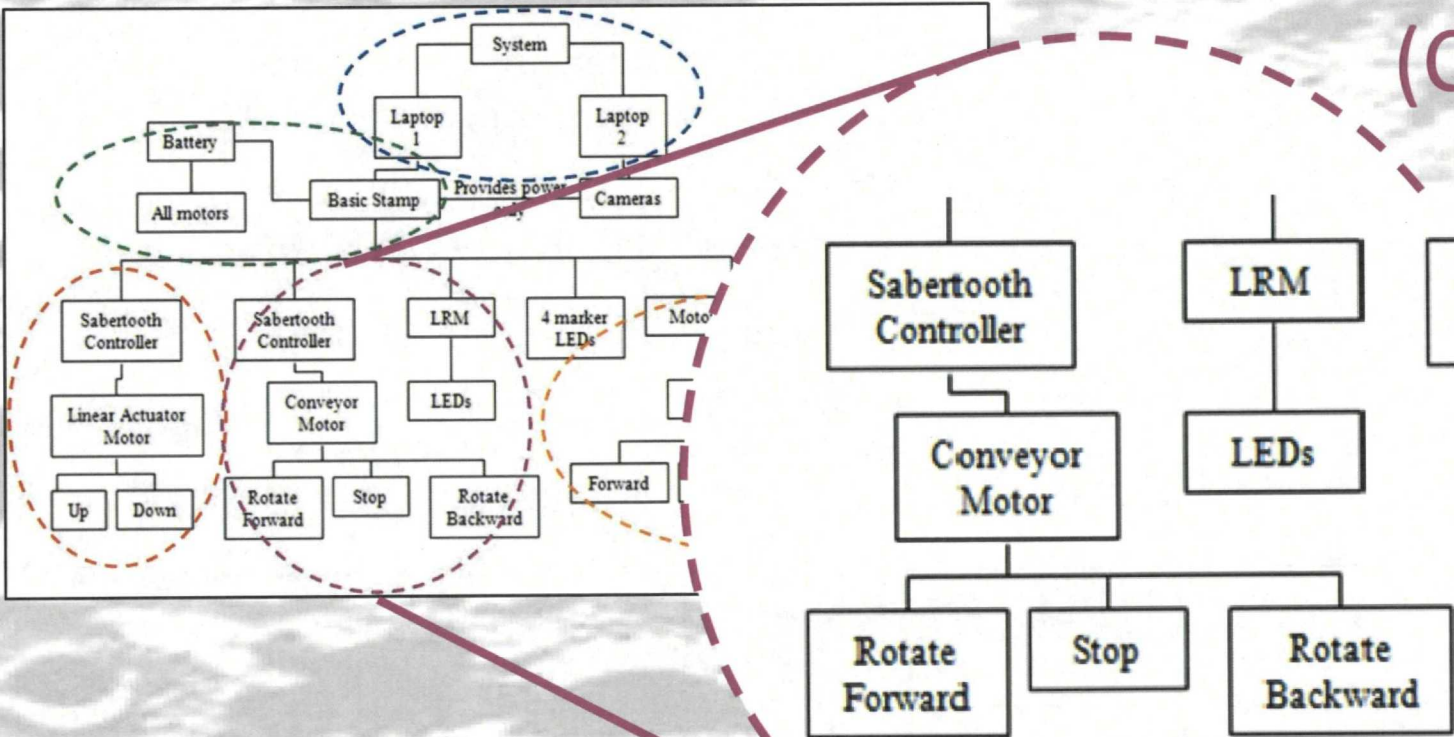
**System Integration**

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# System Integration

Deposit  
Function  
(Conveyor)



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# Risk Assessment

## **System Model:**

Excavator is primarily a series system  
Limited redundancy due to weight constraints  
Virtually every component is a single point failure

## **Potential Failure Modes:**

Yielding  
Current overload  
Motor torque overload

## **Risk Minimization:**

Professional vs. onsite fabrication  
Safety factor of at least 1.5 for each component  
Conservative load estimates  
Familiarity with motor controllers  
Safety factor for maximum current  
Derating of purchased components



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## Reliability Ratings for Key Components

$$R(t) = e^{-\left(\frac{t}{\eta}\right)^{\beta}}$$

*R = Reliability*

*$\beta$  = Shape Parameter (0.8;  
indicates wear-in period)*

*$\eta$  = Scale Parameter (Expected  
Life) (700 hours)*

*t = operation time (10 hours)*

Two-Parameter Weibull Distribution equation yields a reliability rating of **96.7%** for the linear actuator based on device information.

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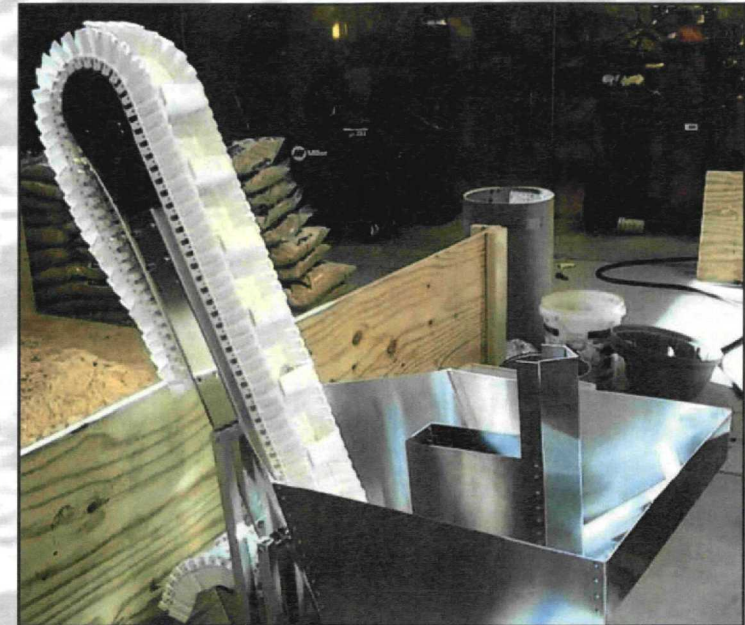
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## Performance Evaluation Plan



Subsystems integration is completed as expected, with weight and size dimensions within specification. CE sub-team has constructed a 1:4 scale practice box to test excavator operation. Sufficient re-design time has been allotted in the project schedule for components that must be re-engineered for improved performance.

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