NASA/TM-2011-217072/Volume II NESC-RP-09-00506





Space Shuttle Program (SSP) Orbiter Main Propulsion System (MPS) Gaseous Hydrogen (GH₂) Flow Control Valve (FCV) Poppet Eddy Current (EC) Inspection Probability of Detection (POD) Study

Appendices

Robert S. Piascik/NESC and William H. Prosser/NESC Langley Research Center, Hampton, Virginia

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TOWN OF STREET	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Volume II: Appendices A and B

Space Shuttle Program (SSP) Orbiter
Main Propulsion System (MPS) Gaseous Hydrogen
(GH₂) Flow Control Valve Poppet Eddy Current (EC)
Inspection Probability of Detection (POD) Study

March 3, 2011

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Appendix A. Poppet Specimens Data

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Surface crack sizes and locations

Poppet#1				
Crack Number	Size (inch)	Angle (degrees)		
1	0.007	330		
2	0.001	330		
3	0.007	330		
4	0.002	330		
5	0.002	330		
6	0.008	150		
7	0.002	150		
8	0.005	150		
9	0.008	150		

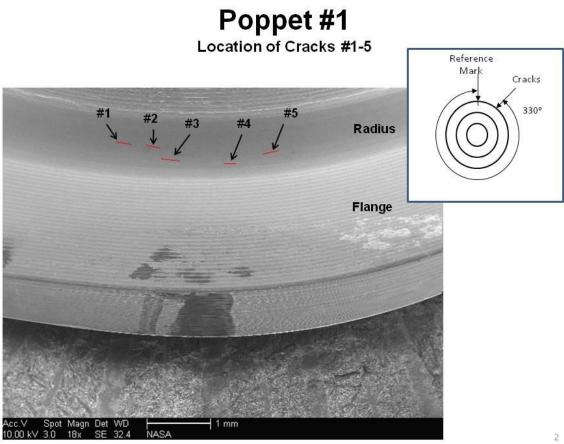
Boeing Eddy Current Findings

S	Poppet #1								
	Run Data (Vpp)								
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.157	0.146	0.147	0.178	0.152	0.163	0.157	Yes	140
J. Engel	-	49	3893	- 12	9	-8	8	No	330
B. Devries	2:	49	14	, n <u>a</u>	-0	150	- 1	No	330
B. Devries	0.151	0.158	0.126	0.147	0.154	0.153	0.148	Yes	140

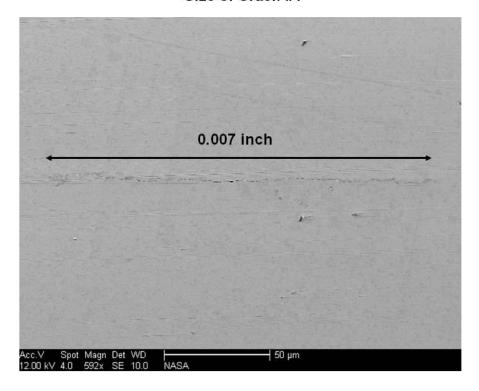
Note: Cracks at the 330° location were not detected by eddy current inspections

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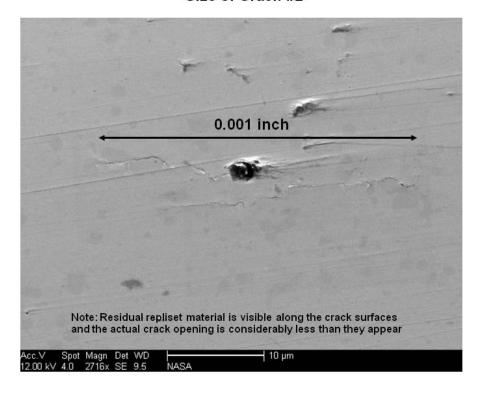




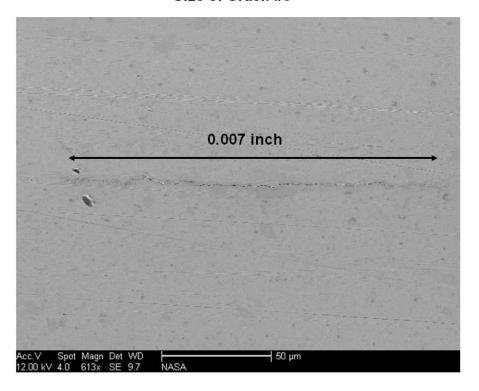
THE PARTY OF THE P	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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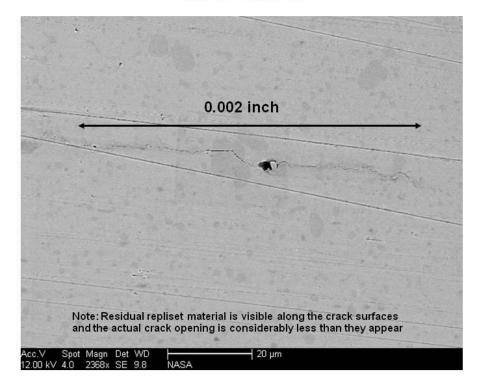
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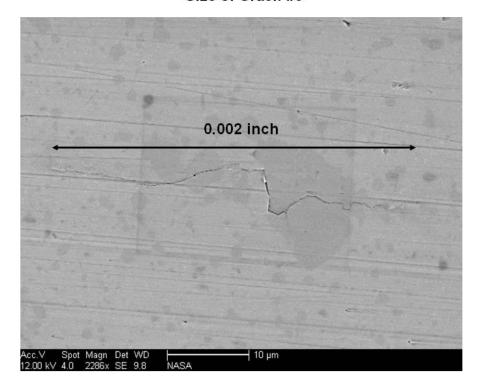
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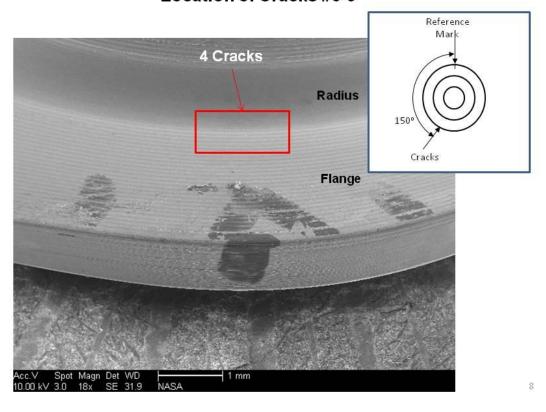


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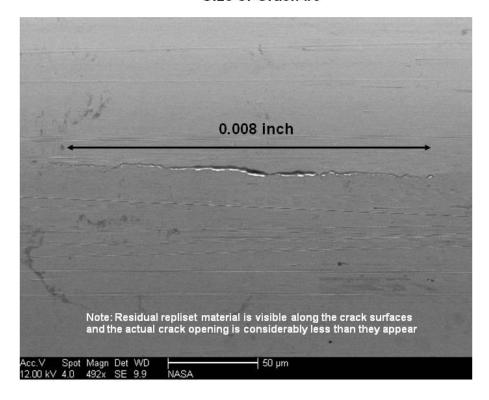


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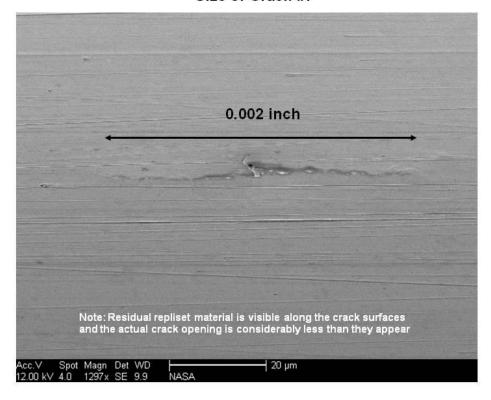
Poppet #1 Location of Cracks #6-9



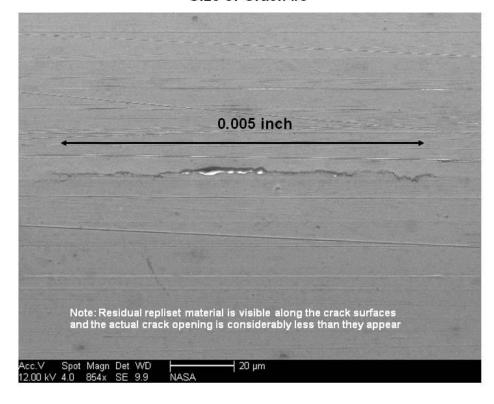
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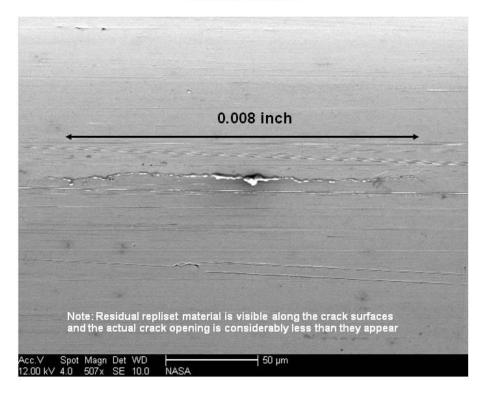
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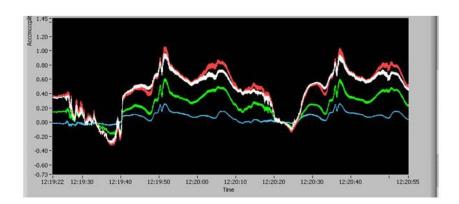


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Poppet #1
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

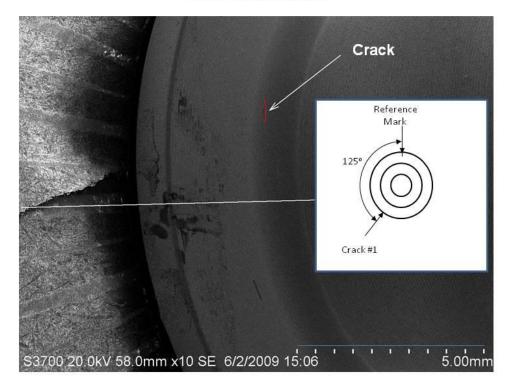
Poppet #7							
Crack Number	Size (inch)	Angle (degrees)					
1	0.027	125					
2	0.024	335					

Boeing Eddy Current Findings

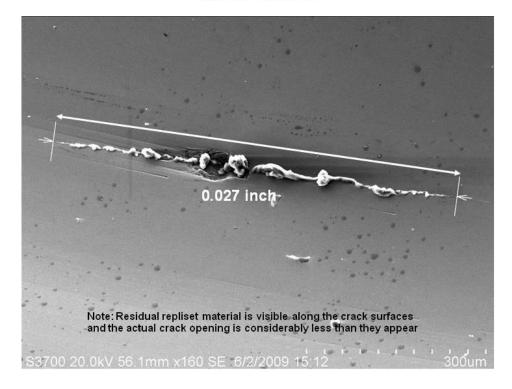
Poppet #7									
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.246	0.240	0.233	0.250	0.249	0.268	0.248	Yes	290
J. Engel	0.120	0.130	0.149	0.156	0.134	0.132	0.137	Yes	135
3. Devries	0.235	0.226	0.239	0.244	0.247	0.226	0.236	Yes	285
3. Devries	0.132	0.127	0.132	0.133	0.139	0.145	0.135	False	190
3. Devries	0.199	0.199	0.193	0.197	0.199	0.192	0.197	Yes	140

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Poppet #7
Location of Crack #1

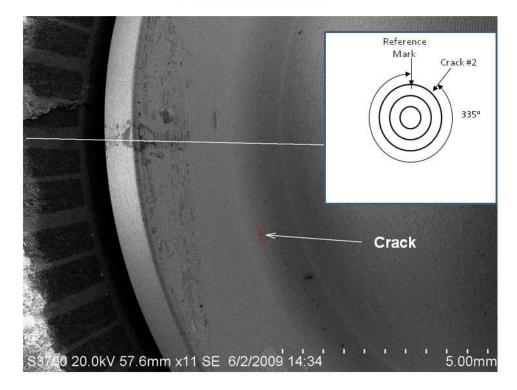


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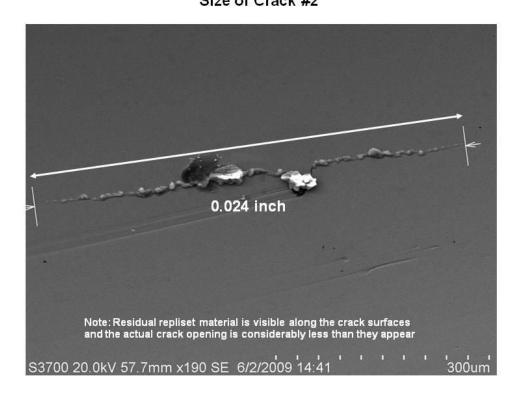


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Poppet #7
Location of Crack #2

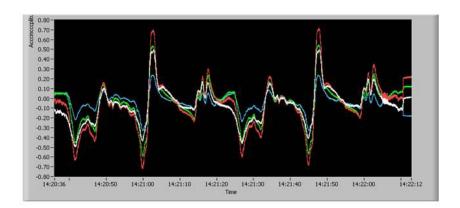


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Poppet #7
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

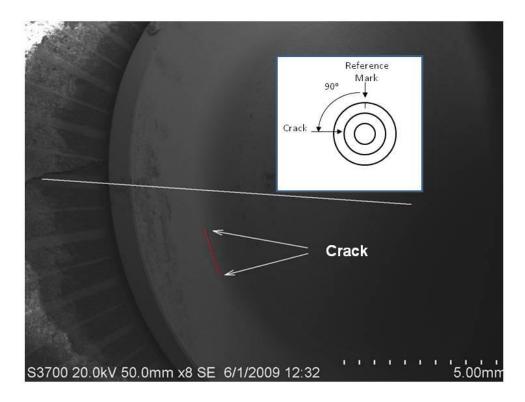
100	Poppet #11	
Crack Number	Size (inch)	Angle (degrees)
1	0.075	90

Boeing Eddy Current Findings

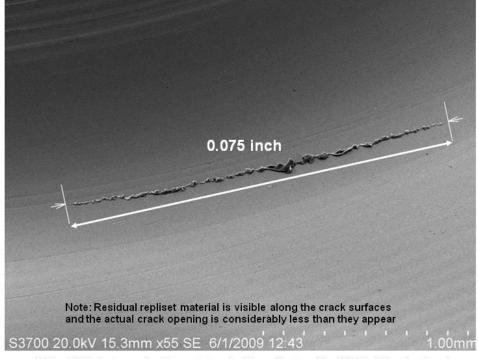
					F	oppet #	†11		V2
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	a	22	1871	1,5	-5	8	9	NA	- 1
3. Devries	1.041	1.064	1.063	1.054	1.113	1.122	1.076	Yes	70

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Poppet #11 Location of Crack #1



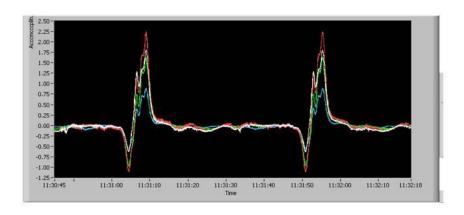
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Note: SEM photograph of Poppet Crack with repliset residue highlighting the Crack

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LaRC eddy current findings, the colors indicate ???



Note: The time trace accounts for 2 complete revolutions of the Poppet relative to the sensor

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Surface crack sizes and locations

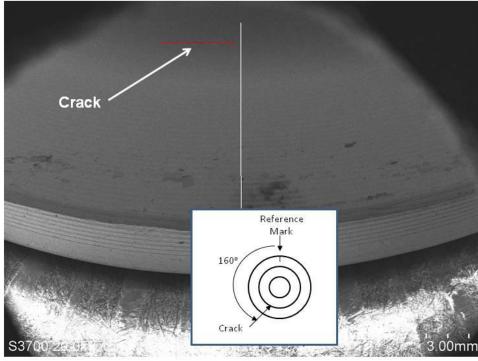
133	Poppet #12	
Crack Number	Size (inch)	Angle (degrees)
1	0.041	160

Boeing Eddy Current Findings

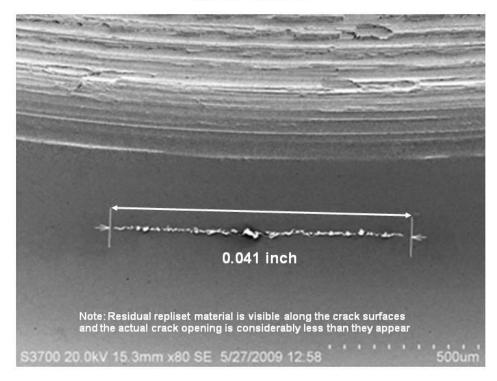
					F	oppet #	12			
	Run Data (Vpp)									
Inspector	1	2 3		4 5		6 Average		Crack Detected	Location (degrees)	
J. Engel	0.664	0.677	0.694	0.642	0.485	0.368	0.588	Yes	170	
B. Devries	0.658	0.655	0.654	0.666	0.658	0.675	0.661	Yes	170	

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Poppet #12 Location of Crack #1

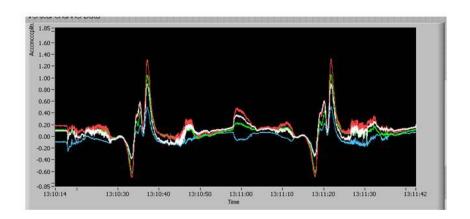


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LaRC eddy current findings, the colors indicate ???



Note: The time trace accounts for 2 complete revolutions of the Poppet relative to the sensor

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Surface crack sizes and locations

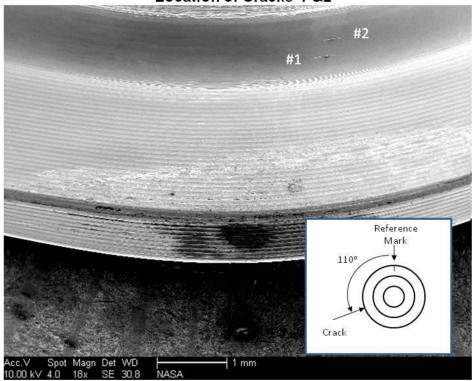
Poppet #13								
Crack Number	Size (inch)	Angle (degrees)						
1	0.014	110						
2	0.014	110						

Boeing Eddy Current Findings

					F	oppet #	13		
			Ru	ın Data (
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.185	0.192	0.196	0.202	0.202	0.213	0.198	Yes	120 (Small indication opposite)
3. Devries	0.190	0.187	0.185	0.186	0.192	0.192	0.189	Yes	110

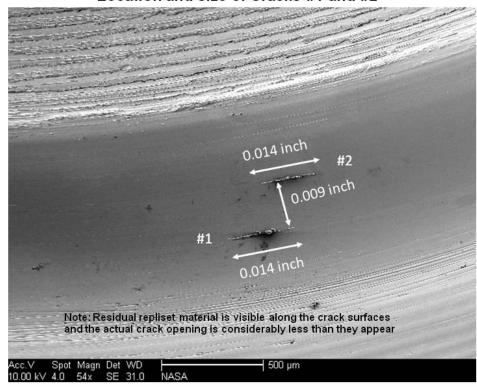
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Poppet #13 Location of Cracks 1 &2



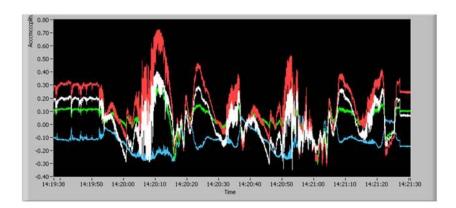
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Poppet #13 Location and size of Cracks #1 and #2



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Poppet #13
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

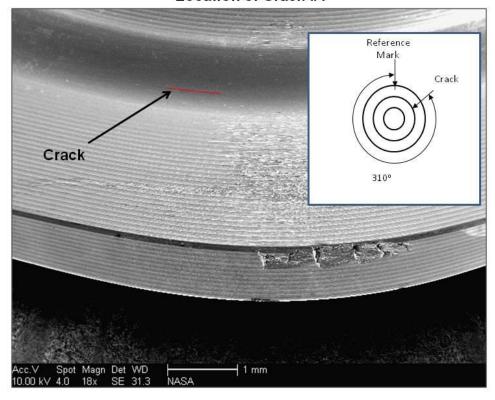
Poppet #14							
Crack Number	Size (inch)	Angle (degrees)					
1	0.033	310					

Boeing Eddy Current Findings

Poppet #14									
	ė –		Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.414	0.412	0.420	0.420	0.430	0.426	0.420	Yes	320
J. Engel	0.042	0.039	0.037	0.040	0.039	0.039	0.039	False	145 (Not 3:1 S/N ratio)
B. Devries	0.415	0.419	0.429	0.448	0.455	0.461	0.438	Yes	315

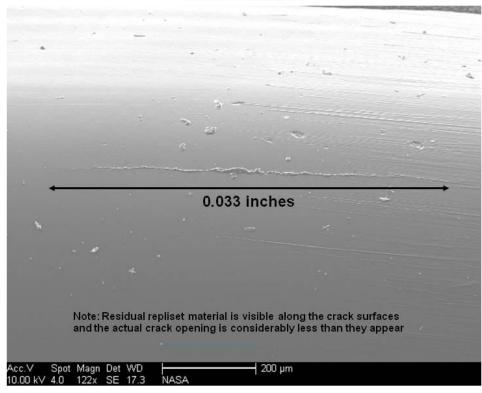
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Poppet #14 Location of Crack #1



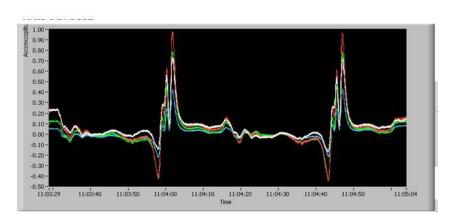
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Poppet #14 Size of Crack #1



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LaRC eddy current findings, the colors indicate ???



Note: The time trace accounts for 2 complete revolutions of the Poppet relative to the sensor

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Surface crack sizes and locations

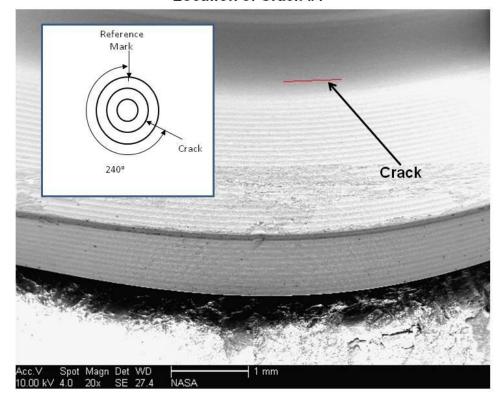
	Poppet #17	
Crack Number	Size (inch)	Angle (degrees)
1	0.026	240

Boeing Eddy Current Findings

Poppet #17									
		26 - 6	Ru	ın Data (√pp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.269	0.275	0.275	0.275	0.276	0.277	0.275	Yes	245
B. Devries	0.257	0.264	0.268	0.269	0.251	0.271	0.263	Yes	255

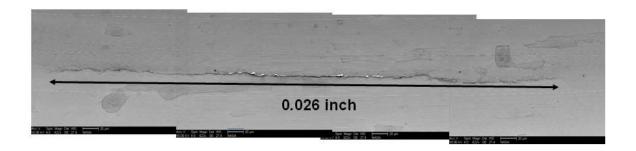
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Poppet #17 Location of Crack #1



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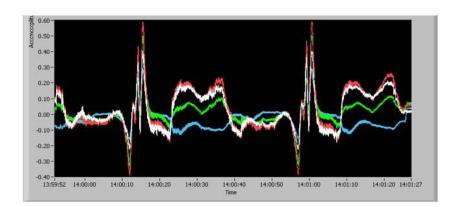
Poppet #17 Size of Crack #1



Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

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Poppet #17
LaRC eddy current findings, the colors indicate ???



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Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 43 of 538

Surface crack sizes and locations

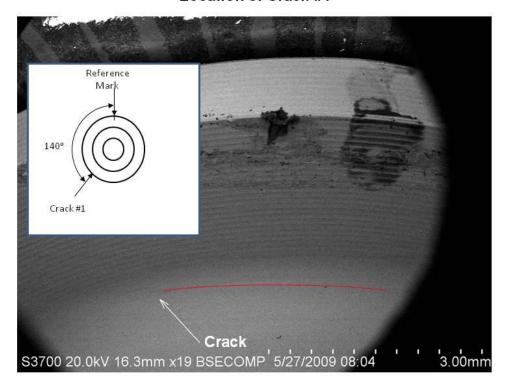
Poppet #18					
Crack Number	Size (inch)	Angle (degrees)			
1	0.078	140			
2	0.025	330			

Boeing Eddy Current Findings

	Poppet #18									
		w.	Rı	ın Data (√pp)	001		5		
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)	
J. Engel	1.026	1.028	1.066	1.077	1.088	1.091	1.063	Yes	120	
J. Engel	0.231	0.234	0.238	0.234	0.237	0.248	0.237	Yes	320	
B. Devries	1.028	1.061	1.041	1.051	1.051	1.096	1.055	Yes	120	
B. Devries	0.247	0.250	0.250	0.252	0.255	0.259	0.252	Yes	305	

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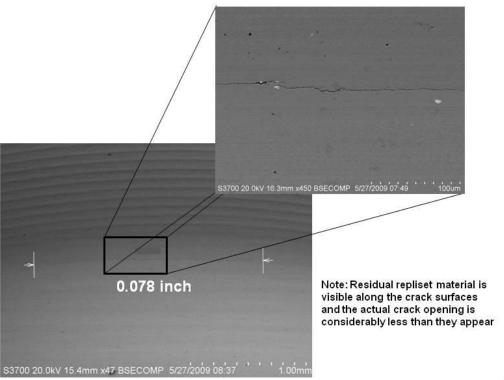
Poppet #18 Location of Crack #1



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THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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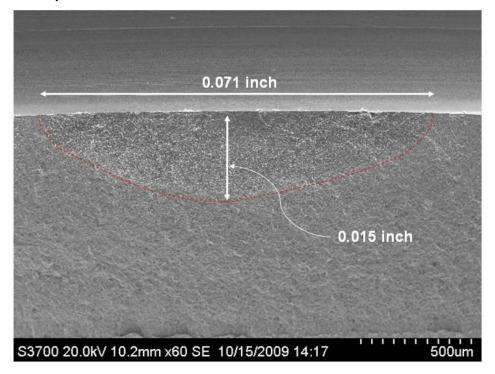
Poppet #18 Location and size of Crack #1



TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #18

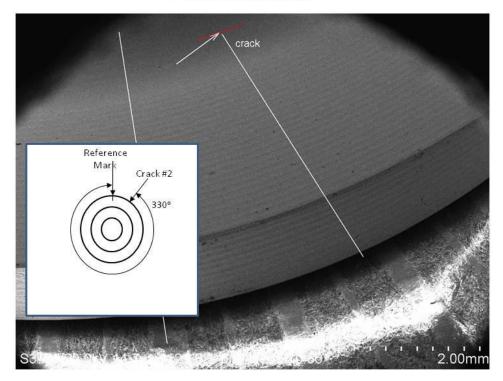
Crack depth and correlation with surface measurements for Crack #1



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THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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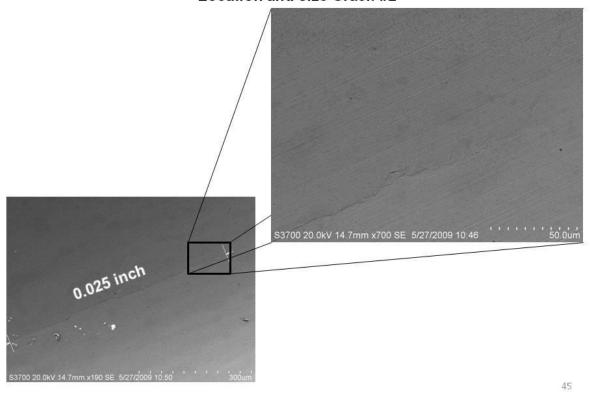
Poppet #18 Location of Crack #2



44

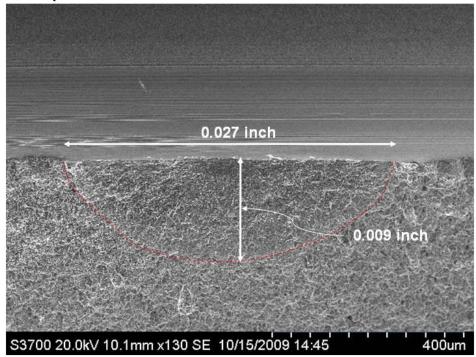
NG A CANAL PARTY OF THE PARTY O	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #18 Location and size Crack #2



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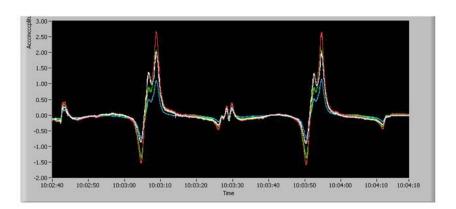
Poppet #18
Crack depth and correlation with surface measurements for Crack #2



TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #1

LaRC eddy current findings, the colors indicate ???



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Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet						

Surface crack sizes and locations

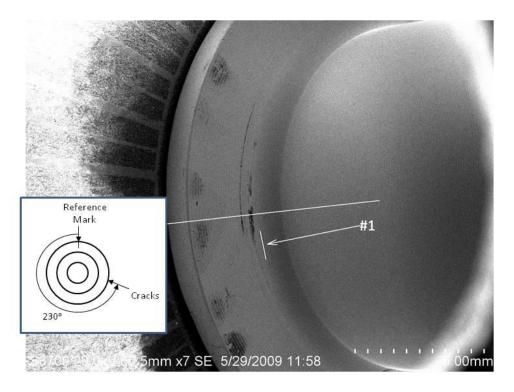
Poppet #19						
Crack Number	Size (inch)	Angle (degrees)				
1 1	0.037	230				
2	0.005	230				
3	0.004	230				
4	0.003	230				
5	0.001	230				

Boeing Eddy Current Findings

Poppet #19									
			Ru	n Data (Vpp)		100		
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.495	0.496	0.510	0.520	0.520	0.528	0.512	Yes	210
B. Devries	0.501	0.504	0.523	0.516	0.516	0.493	0.509	Yes	215

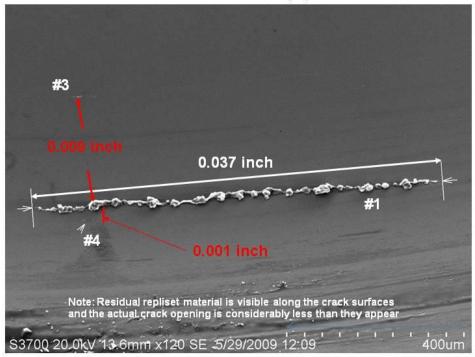
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet						

Poppet #19 Location of Crack #1



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A TO THE WAY TO SECURE THE WAY THE WAY TO SECURE THE WAY TO SECURE THE WAY THE WAY THE WAY TO SECURE THE WAY THE W	Technical Assessment Report	NESC-RP- 09-00506	1.0			
Title:						
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet						

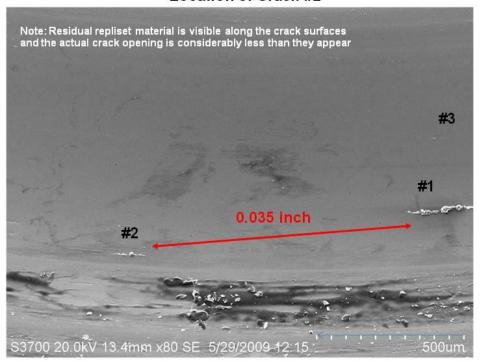
Location of Cracks #1, 3, and 4



50

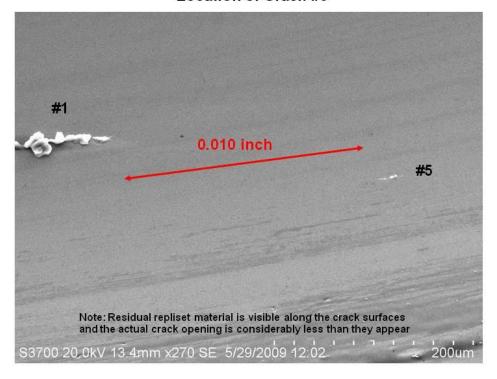
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet						

Poppet #19 Location of Crack #2



THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet						

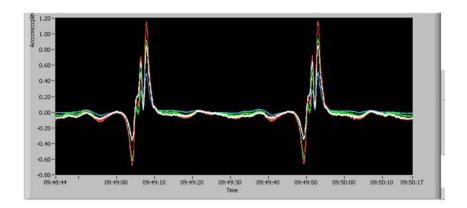
Poppet #19 Location of Crack #5



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WHICH STATES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
Title:						

Poppet #19
LaRC eddy current findings, the colors indicate ???



TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
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Surface crack sizes and locations

*	Poppet #20	80
Crack Number	Size (inch)	Angle (degrees)
1	0.017	110
2	0.003	110
3	0.003	110

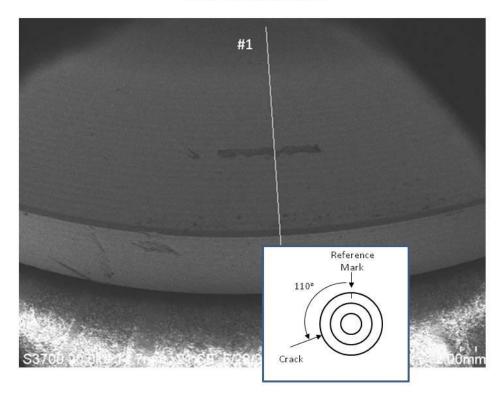
Boeing Eddy Current Findings

Poppet #20									
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.147	0.150	0.154	0.154	0.155	0.145	0.151	Yes	265
B. Devries	0.150	0.165	0.144	0.143	0.144	0.141	0.148	Yes	255

Note: It appears that both Engel and DeVries used a machining mark on the flange as the reference rather than the mark that we put on the specimen. The difference between the two marks was about 160°

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Title:	1 07-00300					

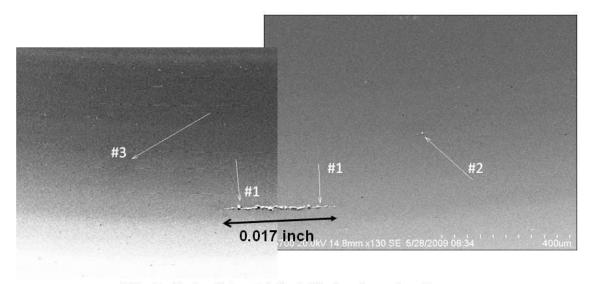
Poppet #20 Location of Crack #1



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THING A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

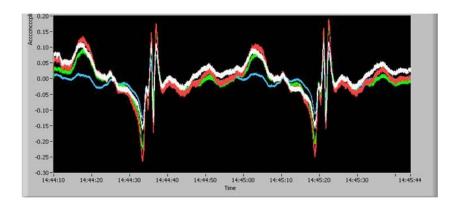
Poppet #20 Location of Cracks #1, 2, and 3



Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

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Poppet #20
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

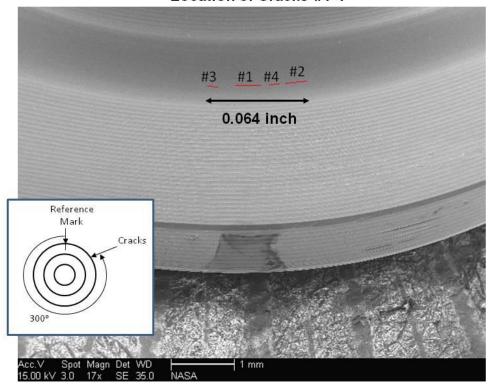
7	Poppet #21	8
Crack Number	Size (inch)	Angle (degrees)
1	0.017	300
2	0.015	300
3	0.009	300
4	0.008	300

Boeing Eddy Current Findings

00					F	oppet #	[‡] 21		
			Ru	n Data (√pp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.143	0.137	0.130	0.143	0.153	0.145	0.142	Yes	300
B. Devries	0.133	0.139	0.137	0.126	0.149	0.128	0.135	Yes	295

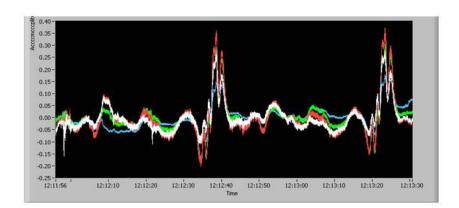
THING A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Poppet #21 Location of Cracks #1-4



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Poppet #21
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

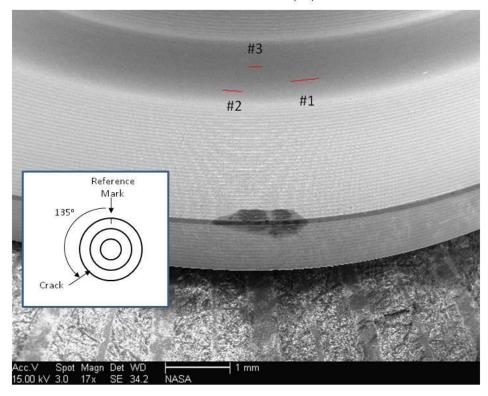
Poppet #22					
Crack Number	Size (inch)	Angle (degrees)			
1	0.022	135			
2	0.009	135			
3	0.004	135			

Boeing Eddy Current Findings

Poppet #22									
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.223	0.215	0.234	0.224	0.233	0.230	0.227	Yes	140
3. Devries	0.220	0.225	0.223	0.226	0.225	0.229	0.225	Yes	120

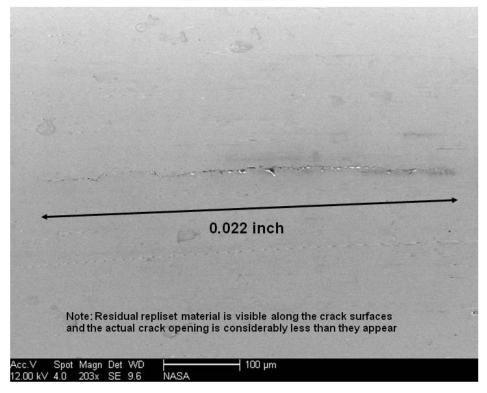
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Poppet #22 Location of Cracks #1, 2, and 3



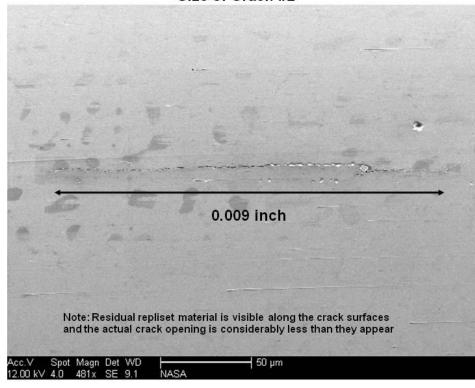
TO THE PROPERTY OF THE PROPERT	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:				

Poppet #22 Size of Crack #1



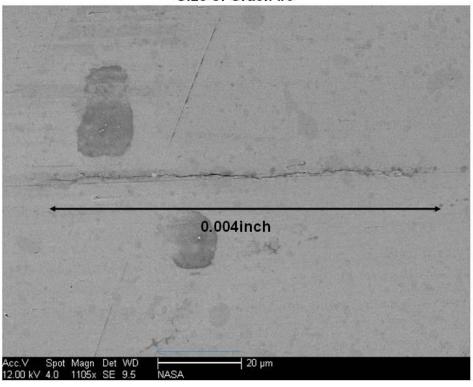
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #22 Size of Crack #2



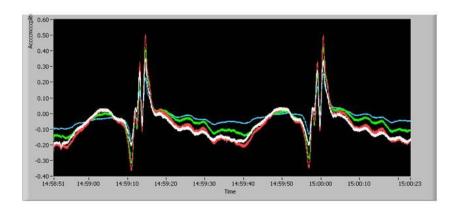
THING A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #22 Size of Crack #3



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Poppet #22
LaRC eddy current findings, the colors indicate ???



NO A STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:	1 07-00300			

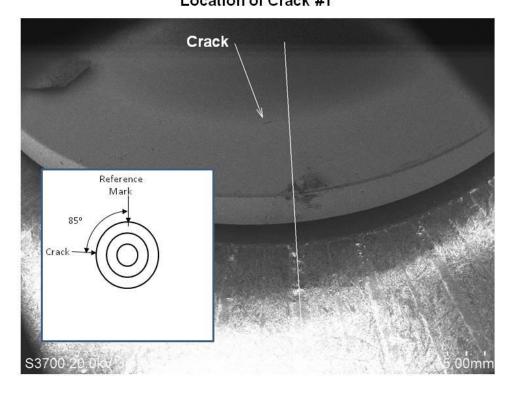
Surface crack sizes and locations

Poppet #23					
Crack Number	Size (inch)	Angle (degrees)			
1	0.026	85			

	Poppet #23								40
	0	77 - 15	Ru	ın Data (Vpp)		g		
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	12	1920	12	9:	20	144	2 1	NA	1/2
B. Devries	0.180	0.193	0.180	0.180	0.178	0.184	0.183	Yes	75

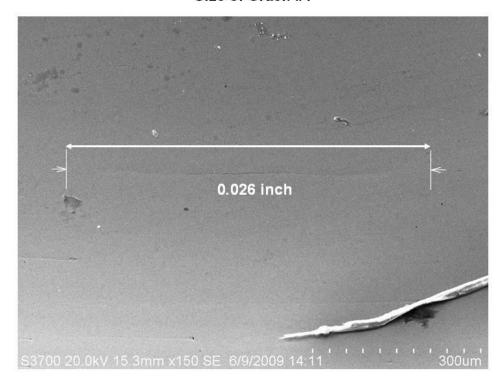
THIS A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Poppet #23
Location of Crack #1



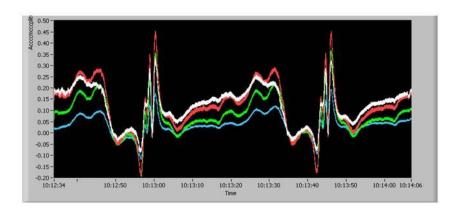
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
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Poppet #23 Size of Crack #1



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Poppet #23
LaRC eddy current findings, the colors indicate ???



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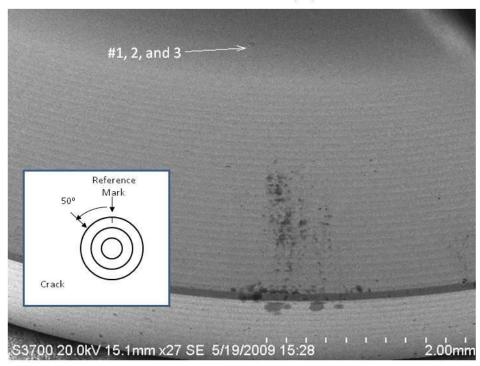
Surface crack sizes and locations

E 68	Poppet #24	8
Crack Number	Size (inch)	Angle (degrees)
1	0.012	50

*	Poppet #24								
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.078	0.096	0.089	0.077	0.090	0.089	0.087	Yes	45 (Not 3:1 S/N ratio)
B Devries	0.084	0.101	0.084	0.086	0.092	0.096	0.091	Yes	30

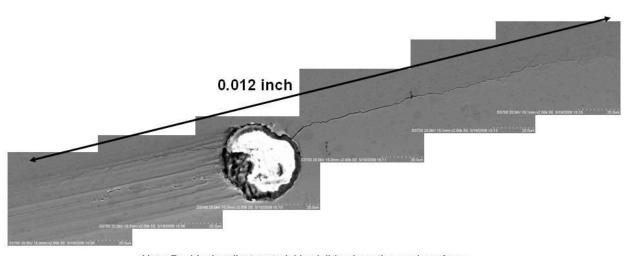
The state of the s	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
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Poppet #24 Location of Cracks #1, 2, and 3



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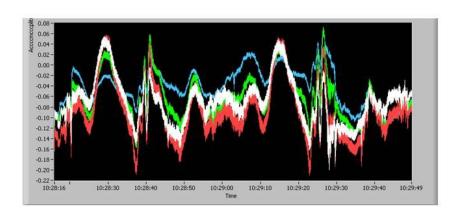
Poppet #24 Size of Crack #1



Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

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Poppet #24
LaRC eddy current findings, the colors indicate ???



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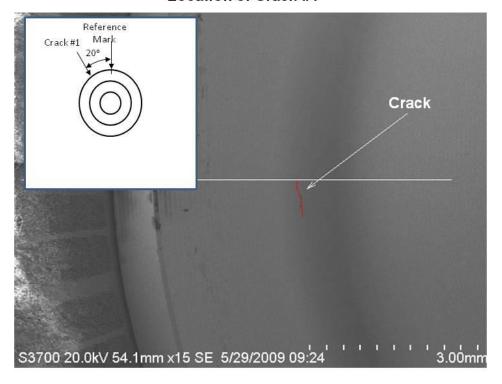
Surface crack sizes and locations

	Poppet #26					
Crack Number	Size (inch)	Angle (degrees)				
1	0.034	20				
2	0.005	210				

	Poppet #26									
84 175 X		22 7	Ru	ın Data (Vpp)	9900				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)	
J. Engel	-83	120	18	9	-8	3243	-	No	210	
J. Engel	0.363	0.348	0.361	0.356	0.353	0.363	0.357	Yes	65	
B. Devries	. 28	323	. 72	- 5	- 92	7528	2	No	210	
B. Devries	0.348	0.349	0.362	0.365	0.363	0.367	0.359	Yes	75	

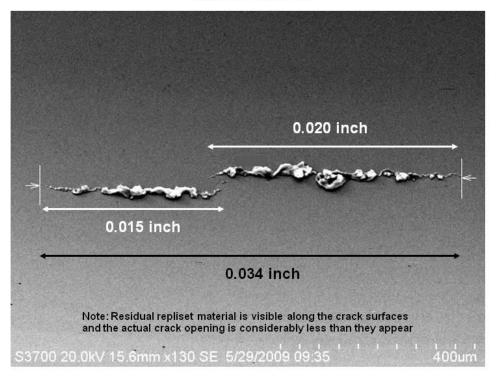
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #26 Location of Crack #1



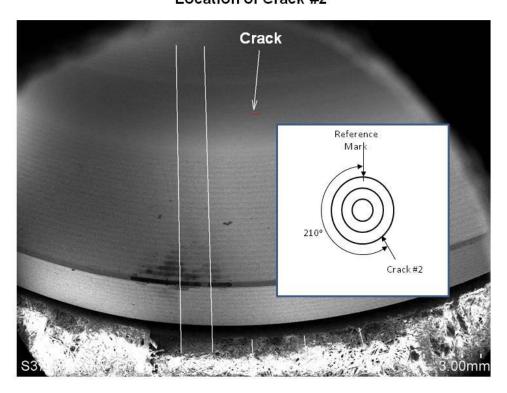
SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:			
TO THE STATE OF TH	Technical Assessment Report	NESC-RP- 09-00506	1.0			
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STS	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Poppet #26 Size of Crack #1



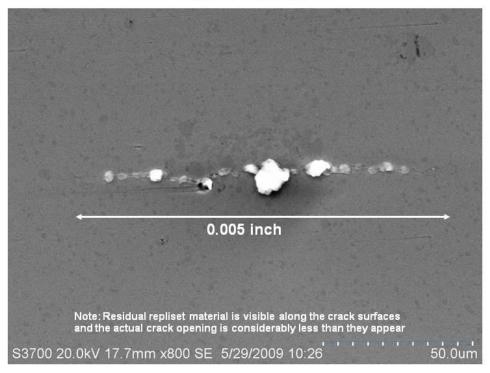
SHERWING & SAFE	NASA Engineering and Safety Center	Document #:	Version:			
Marketon executed	Technical Assessment Report	NESC-RP- 09-00506	1.0			
Title:			Page #:			
ST	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Poppet #26 Location of Crack #2



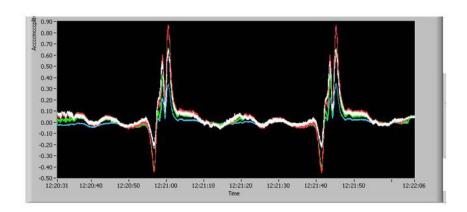
SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:			
TO LEASE STATE OF THE STATE OF	Technical Assessment Report	NESC-RP- 09-00506	1.0			
Title:			Page #:			
STS	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Poppet #26 Size of Crack #2



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LaRC eddy current findings, the colors indicate???



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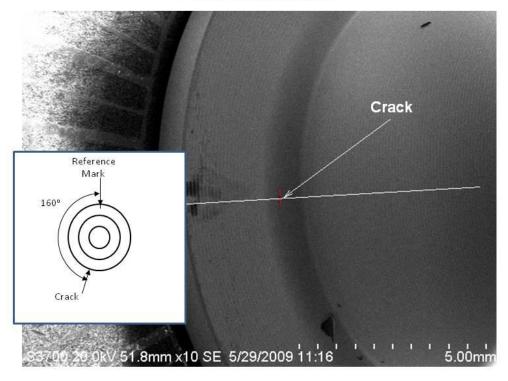
Surface crack sizes and locations

W 8	Poppet #27	3/
Crack Number	Size (inch)	Angle (degrees)
1	0.032	160

60					F	oppet #	27		
	Run Data (Vpp)								
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.423	0.441	0.437	0.449	0.456	0.456	0.444	Yes	155
B. Devries	0.449	0.460	0.463	0.464	0.471	0.467	0.462	Yes	145

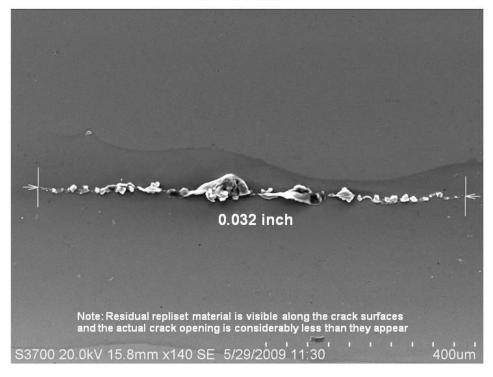
TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #27 Location of Crack #1



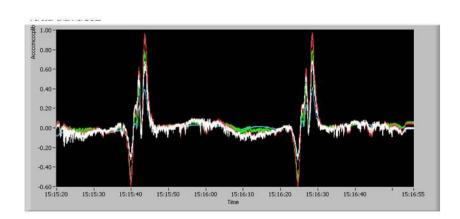
SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:
TO THE STATE OF TH	Technical Assessment Report	NESC-RP- 09-00506	1.0
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Poppet #27 Size of Crack #1



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Poppet #27
LaRC eddy current findings, the colors indicate ???



SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:			
A SANTER	Technical Assessment Report	NESC-RP- 09-00506	1.0			
Title:			Page #:			
STS	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

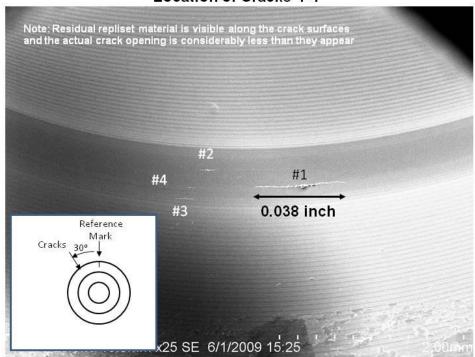
Surface crack sizes and locations

70	Poppet #29					
Crack Number	Size (inch)	Angle (degrees)				
1	0.038	30				
2	0.010	30				
3	0.008	30				
4	0.006	30				

10 %					F	oppet #	29		YK
5			Ru	n Data (Vpp)				3
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.531	0.538	0.535	0.538	0.539	0.558	0.540	Yes	30
B. Devries	0.549	0.559	0.558	0.570	0.567	0.568	0.562	Yes	35

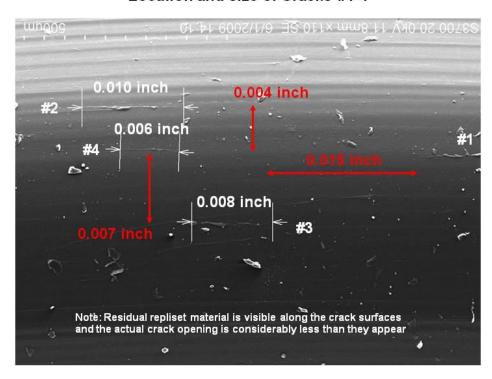
THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location of Cracks 1-4



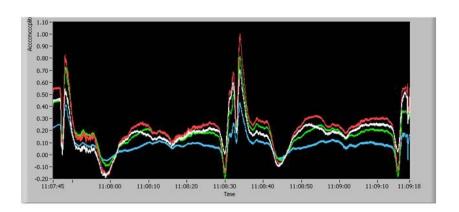
TO THE PERSON NAMED IN COLUMN TO THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #1-4



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Poppet #29
LaRC eddy current findings, the colors indicate ???



SHEERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:		
A SAINES	Technical Assessment Report	NESC-RP- 09-00506	1.0		
Title:			Page #:		
STS	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

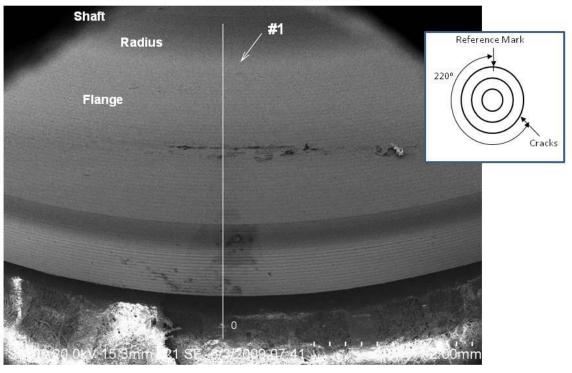
Surface crack sizes and locations

Poppet #31					
Crack Number	Angle (degrees)				
1	0.003	220			
2	0.014	40			
3	0.007	40			
4	0.004	40			
5	0.015	40			
6	0.004	40			
7	0.006	40			
8	0.012	40			
9	0.010	40			
10	0.006	40			
11	0.004	40			
12	0.002	40			

-0 70	Poppet #31										
5			Ru	n Data (\	√pp)				3		
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)		
J. Engel	0.233	0.233	0.230	0.230	0.231	0.230	0.231	Yes	40		
J. Engel	8	- 38	3373	1 55	8	86	15	No	220		
3. Devries		 58	875	l æ				No	220		
3. Devries	0.221	0.226	0.222	0.215	0.221	0.226	0.222	Yes	45		

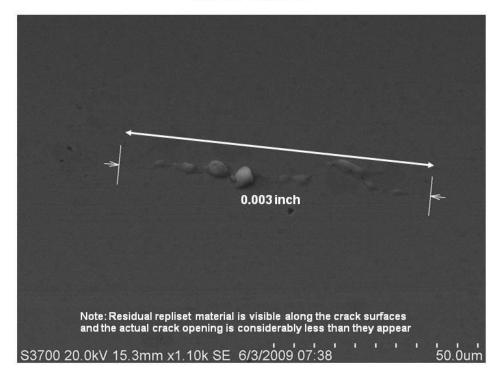
THE STATE OF THE S	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #31 Location of Crack #1



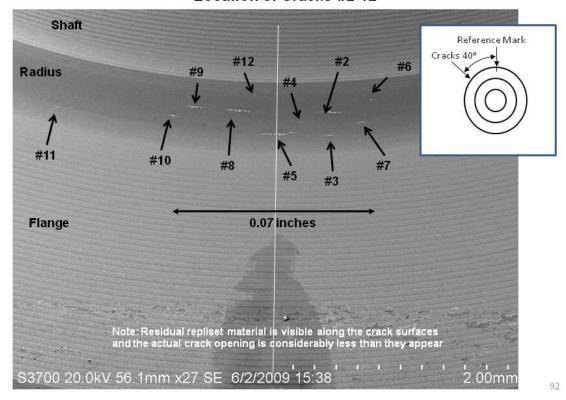
NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:	
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Poppet #31 Size of Crack #1



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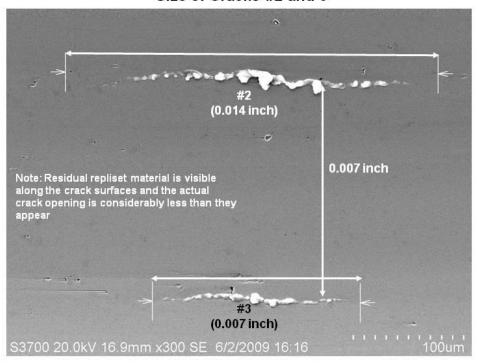
Location of Cracks #2-12



NESC Request No.: 09-00506

TO THE RESIDENCE OF THE PARTY O	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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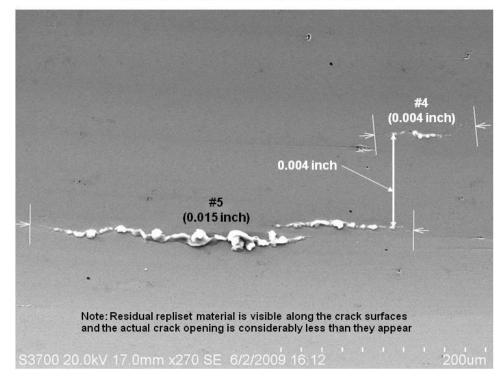
Size of Cracks #2 and 3



THE STATE OF THE S	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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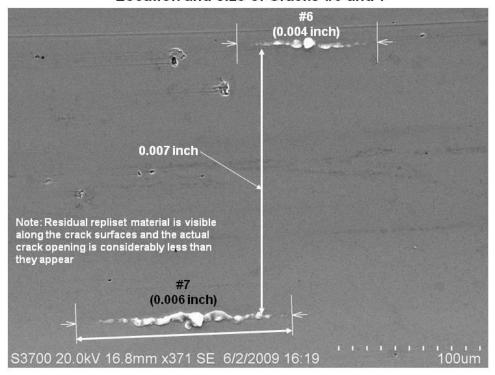
Poppet #31

Location and size of Cracks #4 and 5



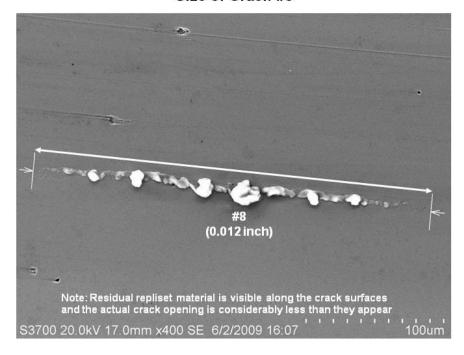
STATE STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #6 and 7



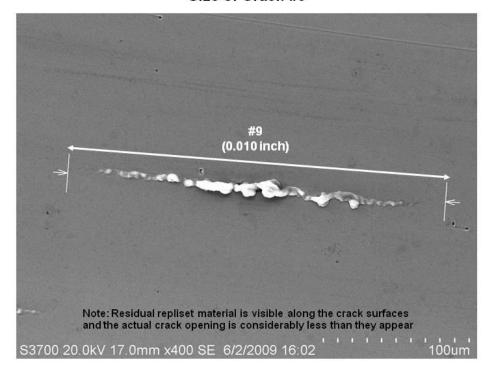
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #31 Size of Crack #8



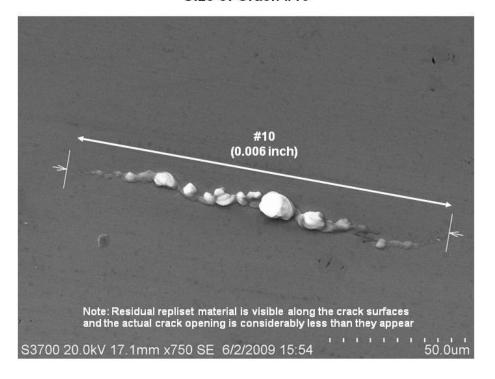
SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:
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Title:			Page #:
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Poppet #31 Size of Crack #9



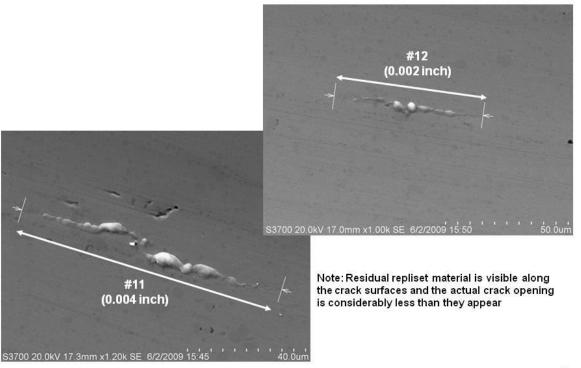
NG & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #10



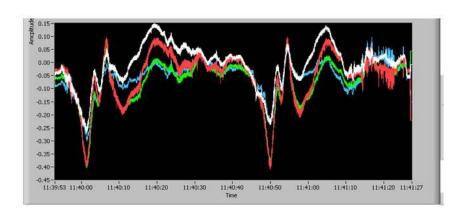
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #31 Size of Cracks #11 and 12



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Poppet #31
LaRC eddy current findings, the colors indicate ???

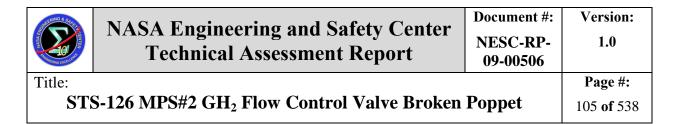


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TO THE REPORT OF THE PARTY OF T		NESC-RP- 09-00506	1.0
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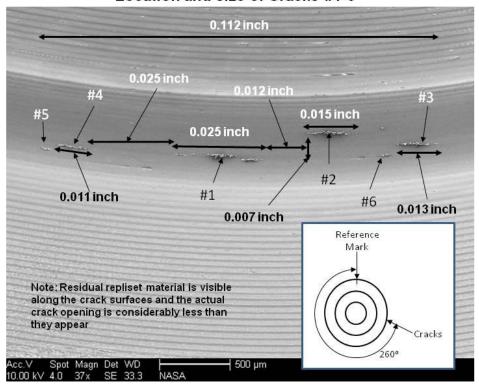
Surface crack sizes and locations

Poppet #32						
Crack Number	Size (inch)	Angle (degrees) 260				
1	0.025					
2	0.015	260				
3	0.013	260				
4	0.011	260				
5	0.006	260				
6	0.005	260				
7	0.012	80				
8	0.010	80				
9	0.005	80				

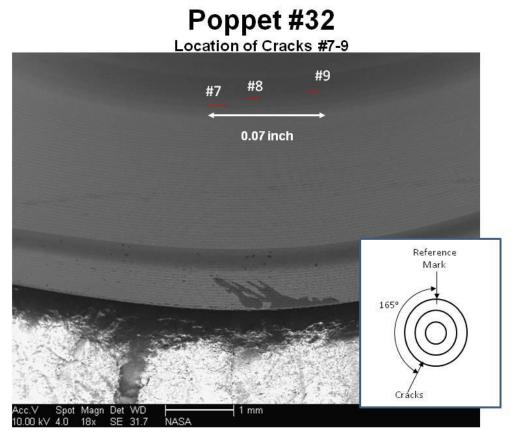
	Poppet #32										
Inspector	Run Data (Vpp)										
	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)		
J. Engel	0.248	0.241	0.241	0.249	0.250	0.250	0.247	Yes	75		
J. Engel	0.077	0.075	0.076	0.081	0.084	0.086	0.080	Yes	255 (Marginal 3:1 S/N)		
B. Devries	0.074	0.073	0.067	0.072	0.067	0.070	0.071	Yes	255		
B. Devries	0.244	0.246	0.243	0.244	0.251	0.244	0.245	Yes	75		



Location and size of Cracks #1-6

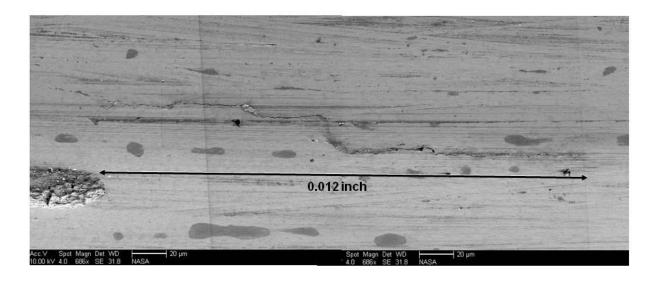


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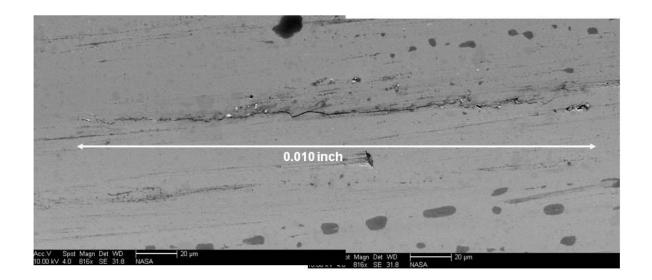
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Poppet #32 Size of Crack #7



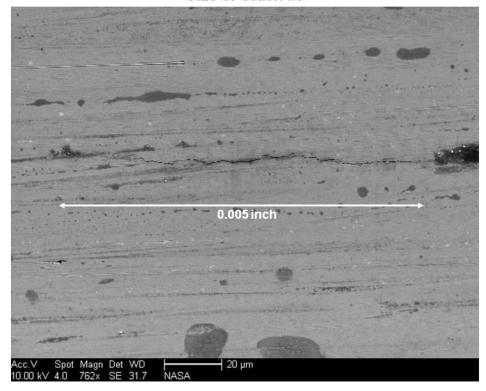
THE A STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #32 Size of Crack #8



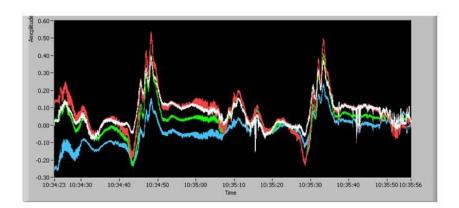
THE A STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #32 Size of Crack #9



THE THING A SALES	NASA Engineering and Safety Center Tochnical Assessment Report	Document #: NESC-RP-	Version: 1.0	
Tech	Technical Assessment Report	09-00506		
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ST	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

LaRC eddy current findings, the colors indicate???



THIS A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 111 of 538

Surface crack sizes and locations

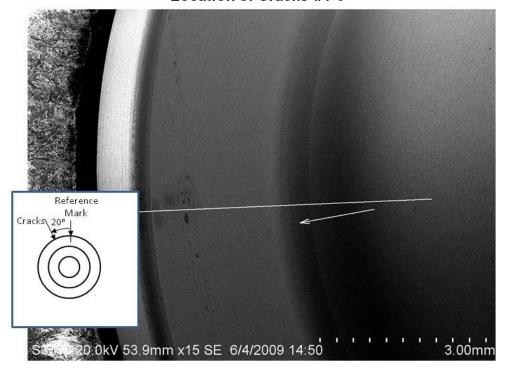
Poppet #33					
Crack Number	Size (inch)	Angle (degrees)			
1	0.026	20			
2	0.014	20			
3	0.002	20			
4	0.003	200			

Boeing Eddy Current Findings

Poppet #33									
9			Ru	ın Data (\	√pp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	2:	- 41	1823	72	0	- 15	1 12	No	200
J. Engel	0.220	0.221	0.223	0.223	0.227	0.218	0.222	Yes	10
B. Devries		-	8758	ge.			15	No	200
B. Devries	0.211	0.216	0.216	0.215	0.209	0.218	0.214	Yes	15

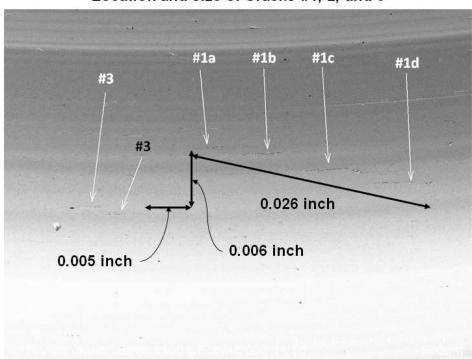
SHERING & SARE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
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Poppet #33 Location of Cracks #1-3



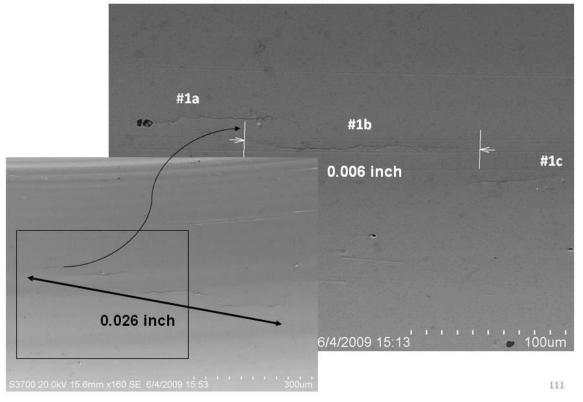
SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:
TENTER OF THE PROPERTY OF THE	Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
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Poppet #33
Location and size of Cracks #1, 2, and 3



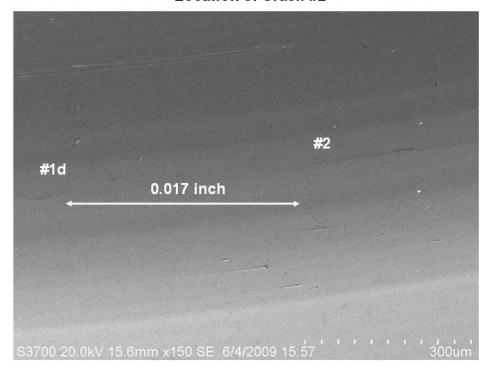
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #33 Location of Crack #1



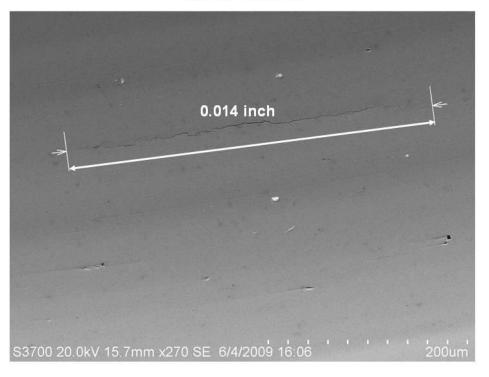
SHERING & SAFE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
TENTER OF THE PROPERTY OF THE		NESC-RP- 09-00506	1.0
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Poppet #33
Location of Crack #2



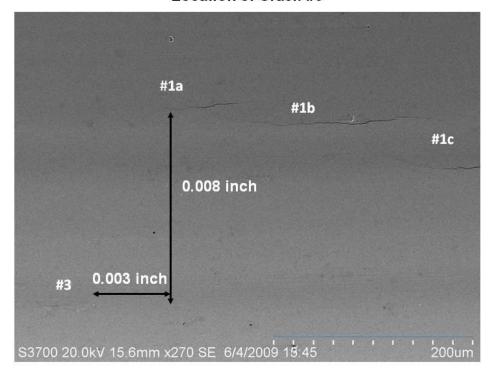
	neering and Safety Center cal Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #33 Size of Crack #2



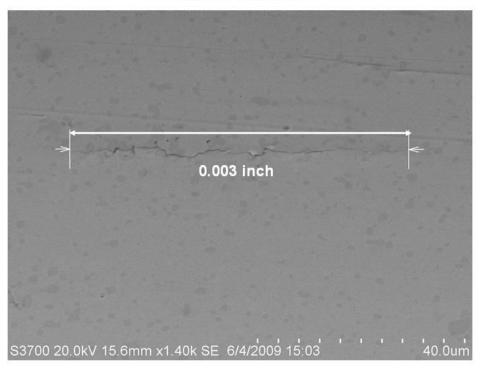
SHERING & SAFE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:	
SENTER STANDARD		NESC-RP- 09-00506	1.0	
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Poppet #33
Location of Crack #3



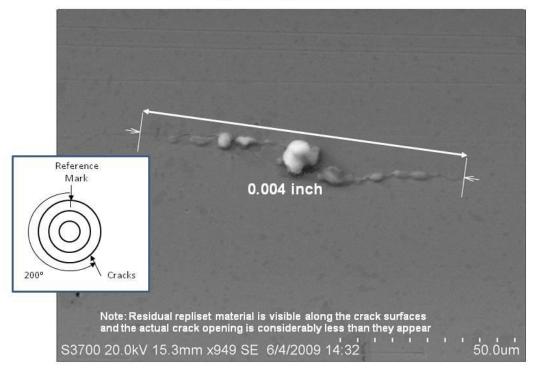
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #33 Size of Crack #3



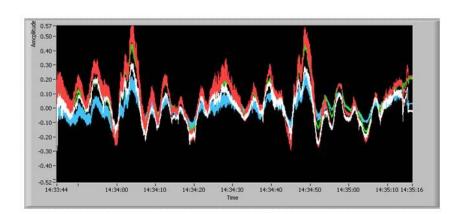
NAME OF STREET	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #33 Size of Crack #4



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Poppet #33
LaRC eddy current findings, the colors indicate ???



SHERING & SACE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
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Surface crack sizes and locations

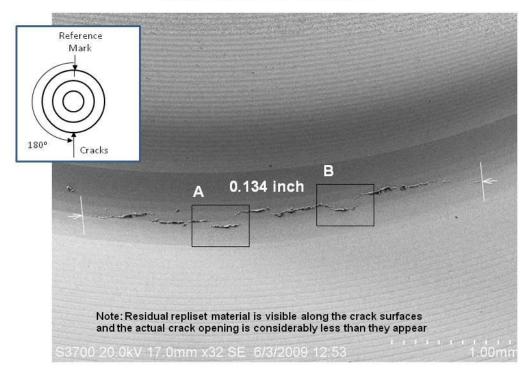
Poppet #34				
Crack Number	Size (inch)	Angle (degrees)		
1	0.134	180		
2	0.027	0		
3	0.010	0		
4	0.021	0		
5	0.009	0		
6	0.004	0		
7	0.005	0		
8	0.005	0		

Boeing Eddy Current Findings

Poppet #34									
26191603396000	0		Ru	n Data (√pp)				0
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.681	0.688	0.683	0.690	0.693	0.671	0.684	Yes	180
J. Engel	0.233	0.230	0.235	0.231	0.233	0.233	0.233	Yes	355
9. Devries	0.215	0.218	0.213	0.218	0.217	0.219	0.217	Yes	0
A Devries	0.648	0.659	0.649	0.654	0.653	0.670	0.656	Yes	180

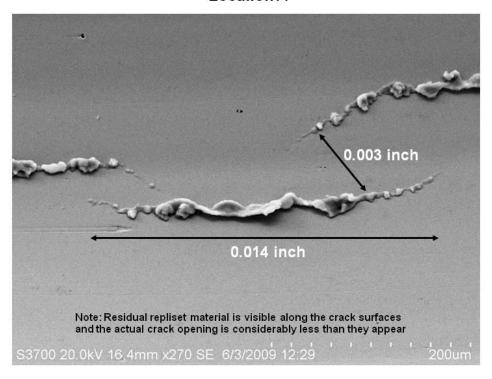
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #34 Location and size of Crack #1



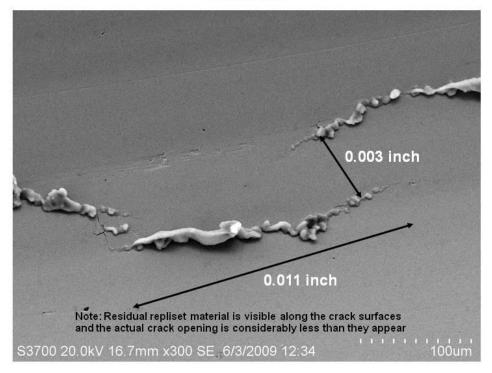
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

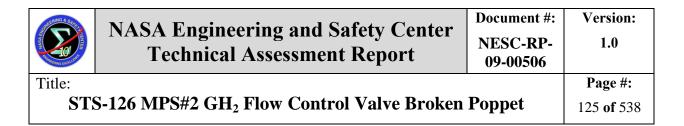
Location A



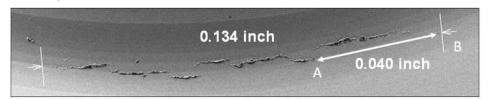
SHERING & SAFE	NASA Engineering and Safety Contor	Document #:	Version:
TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report		1.0
Title:			Page #:
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Poppet #34 Location B

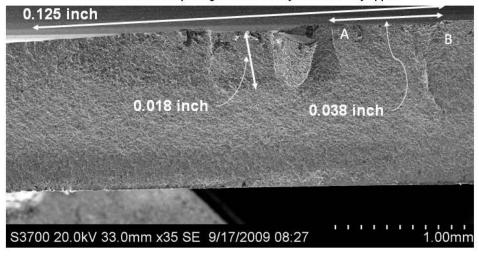




Crack depth and correlation with surface measurements for Crack #1

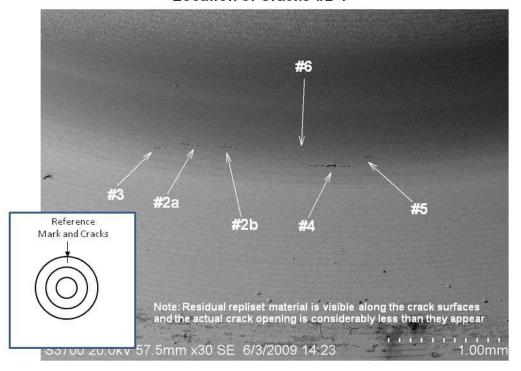


Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear



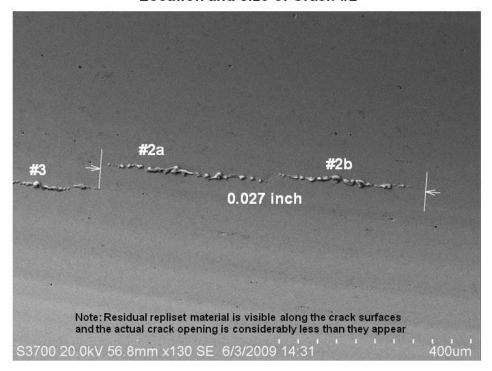
TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #34 Location of Cracks #2-4



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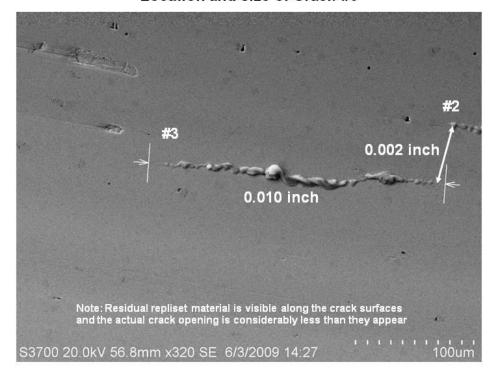
Location and size of Crack #2



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STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

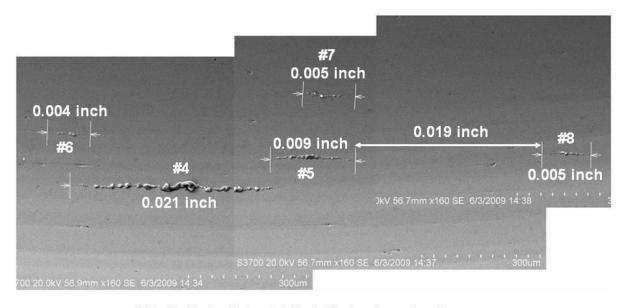
Poppet #34

Location and size of Crack #3



TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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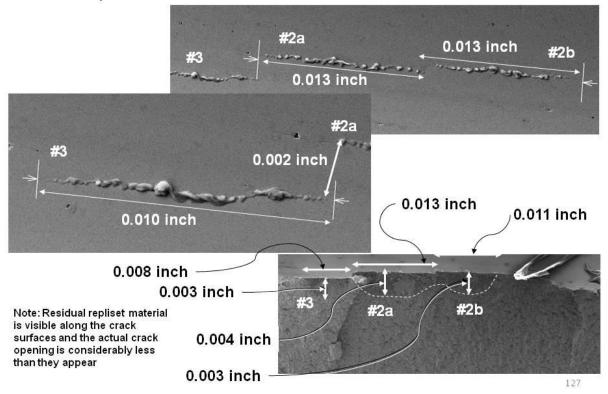
Location and size of Cracks #4 - 8



Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

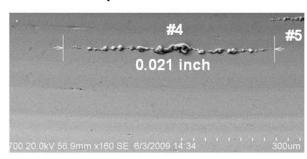
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			Page #: 130 of 538

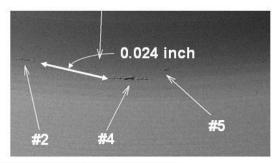
Crack depth and correlation with surface measurements for Cracks #2 & 3

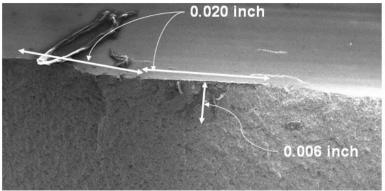


THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Crack depth and correlation with surface measurements for Crack #4



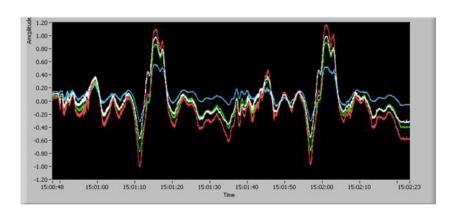




Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

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Poppet #34
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

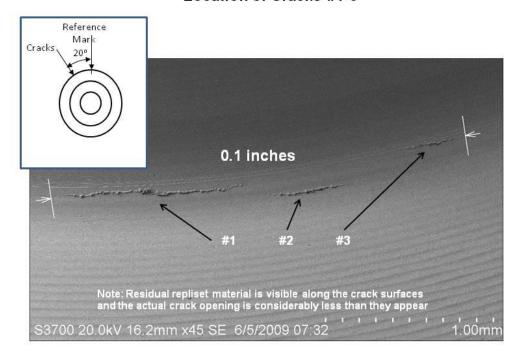
Poppet #35				
Crack Number	Size (inch)	Angle (degrees)		
1	0.044	20		
2	0.021	20		
3	0.018	20		
4	0.006	200		
5	0.005	200		
6	0.004	200		

Boeing Eddy Current Findings

Poppet #35									
			Ru	n Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.564	0.585	0.597	0.596	0.604	0.595	0.590	Yes	10
J. Engel	0.065	0.074	0.055	0.074	0.082	0.068	0.070	Yes	195 (Not315/Nratio)
B. Devries	0.573	0.580	0.584	0.577	0.574	0.564	0.575	Yes	15
B. Devries	9	<u> </u>	194	, Nº	· · ·	9	. 12	No	200

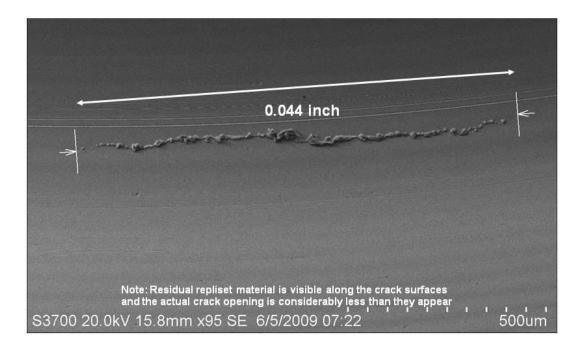
TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location of Cracks #1-3



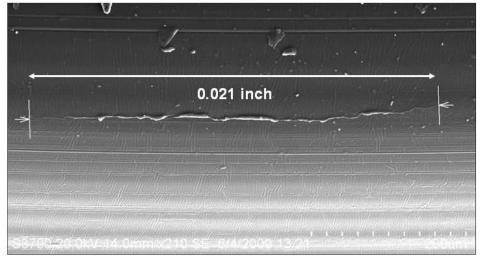
THING & STATE OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #1



THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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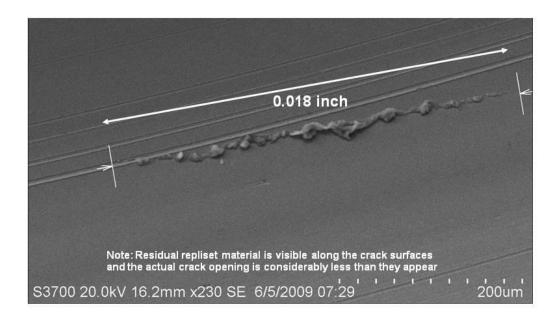
Poppet #35 Size of Crack #2



Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

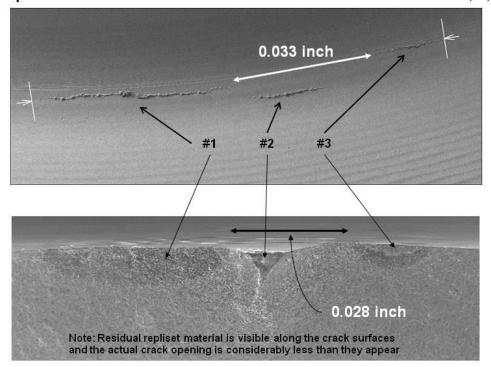
THE REPORT OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
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Size of Crack #3



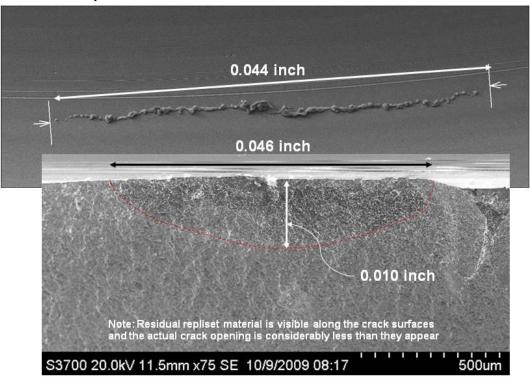
NASA Engineering and Safety Cer Technical Assessment Report	er Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Crack depth and correlation with surface measurements for Cracks #1, 2, and 3



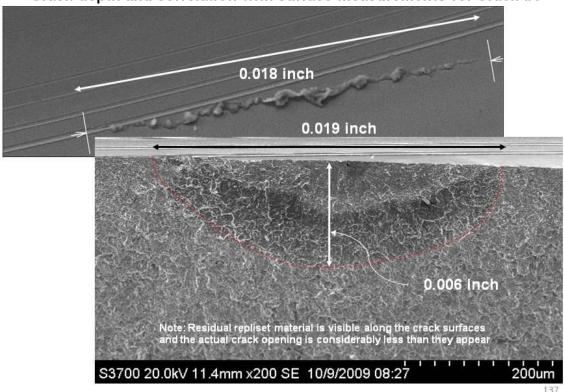
STATE OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
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Crack depth and correlation with surface measurements for Crack #1

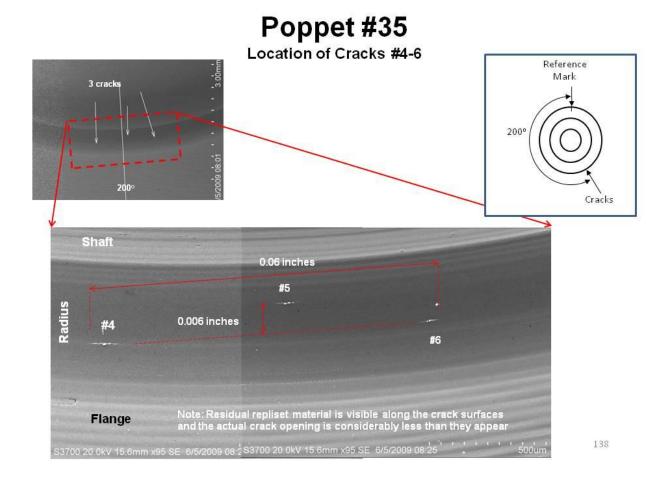


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Crack depth and correlation with surface measurements for Crack #3

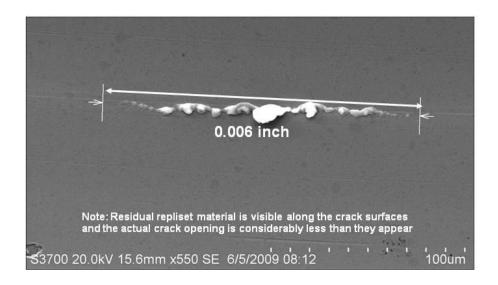


THE PARTY OF THE P	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				



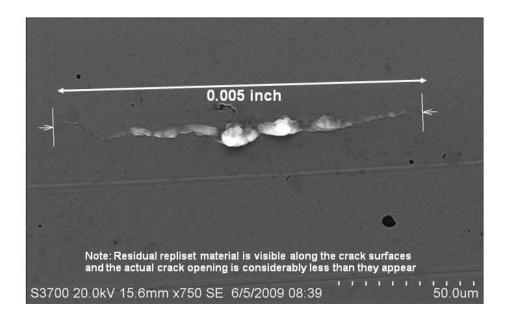
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

Size of Crack #4



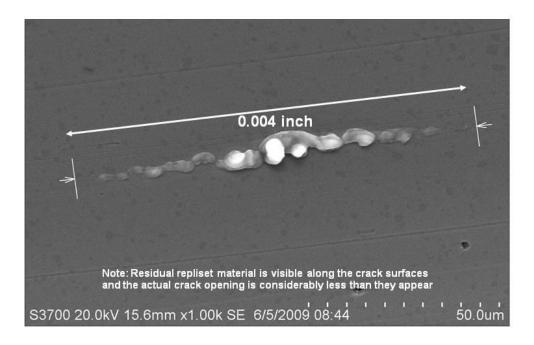
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SENTER SECURITION OF THE PERSON OF THE PERSO	Technical Assessment Report		1.0
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Size of Crack #5



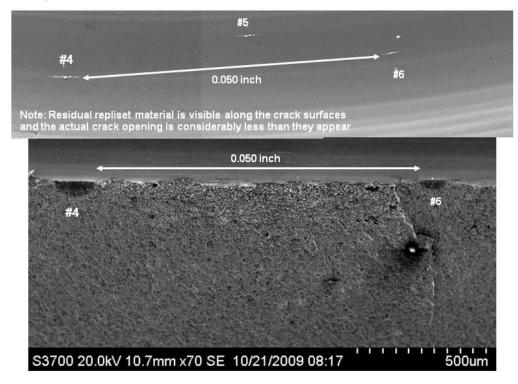
NG & COLUMN AND A	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #6



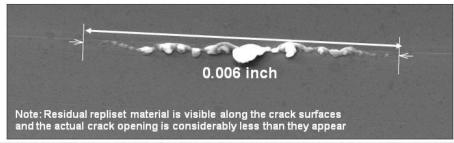
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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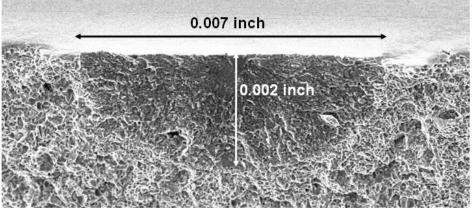
Crack depth and correlation with surface measurements for Cracks #4 and 6



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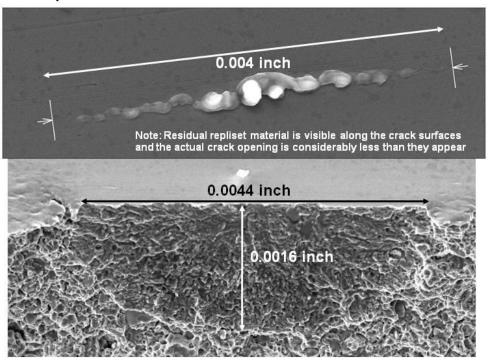
Crack depth and correlation with surface measurements for Crack #4





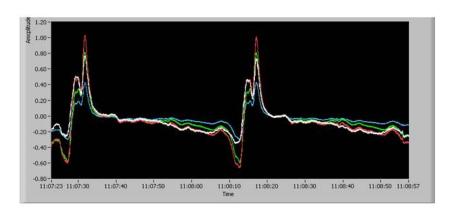
NO A PARTIE OF THE PARTIE OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Crack depth and correlation with surface measurements for Crack #6



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Poppet #35
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

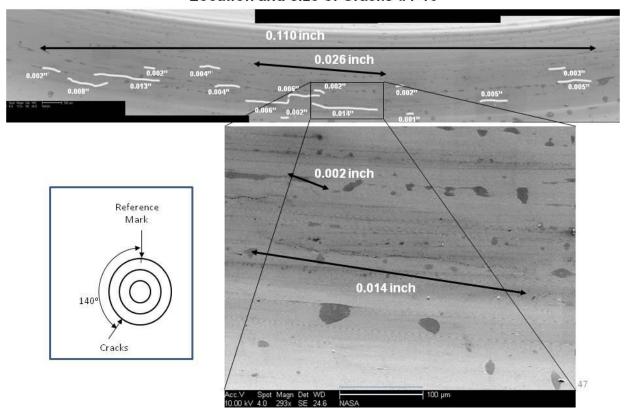
Poppet #36					
Crack Number	Size (inch)	Angle (degrees)			
1	0.005	140			
2	0.001	140			
3	0.001	140			
4	0.006	140			
5	0.002	140			
6	0.014	140			
7	0.006	140			
8	0.004	140			
9	0.002	140			
10	0.005	140			
11	0.003	140			
12	0.013	140			
13	0.008	140			
14	0.002	140			
15	0.002	140			
16	0.004	140			
17	0.009	310			
18	0.004	310			
19	0.002	310			

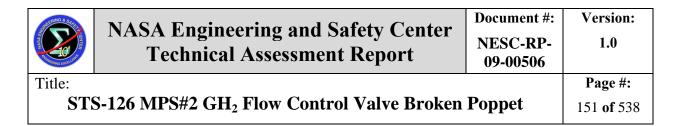
Boeing Eddy Current Findings

Poppet #36									
			Ru	n Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.157	0.160	0.163	0.168	0.150	0.160	0.160	Yes	135 (Wobble in poppet)
J. Engel	0.099	0.096	0.096	0.096	0.092	0.093	0.095	Yes	310 (Wobble in poppet)
3. Devries	0.175	0.169	0.181	0.171	0.161	0.158	0.169	Yes	130
3. Devries	0.086	0.084	0.087	0.084	0.090	0.076	0.085	Yes	310

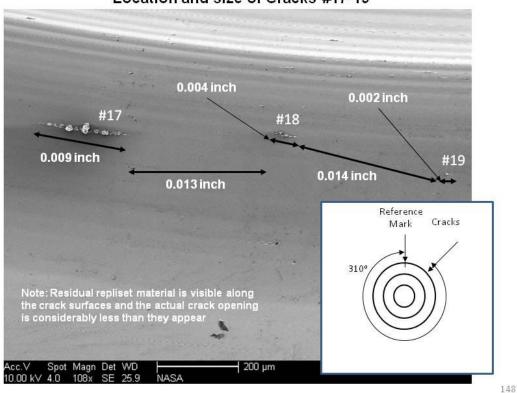
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

Poppet #36 Location and size of Cracks #1-16



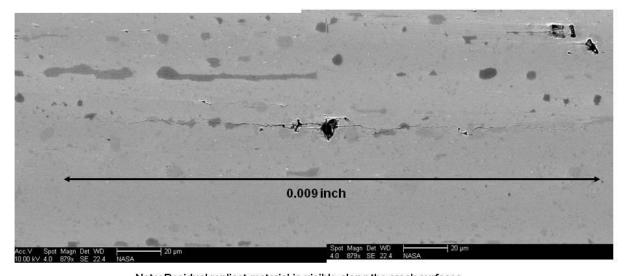


Location and size of Cracks #17-19



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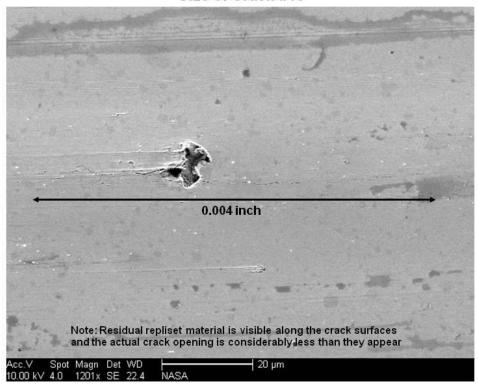
Poppet #36 Size of Crack #17



Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

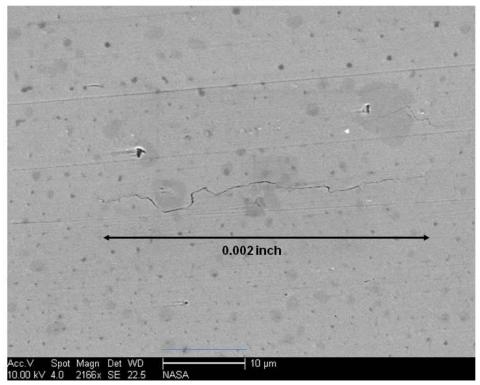
THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #36 Size of Crack #18



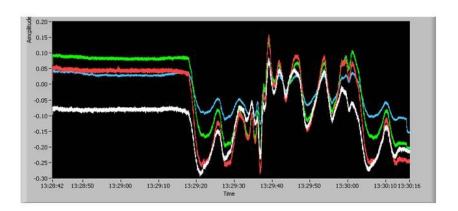
THE STREET	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #36 Size of Crack #19



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Poppet #36
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

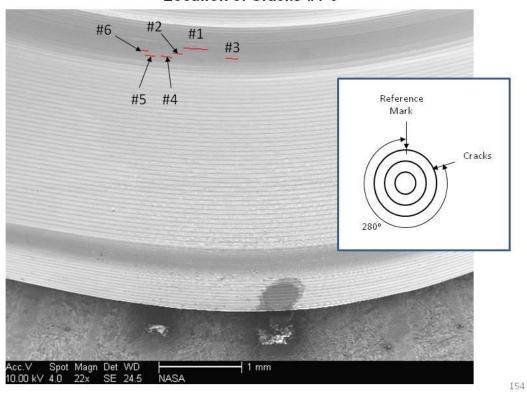
Poppet #37			
Crack Number	Size (inch)	Angle (degrees)	
1	0.022	280	
2	0.002	280	
3	0.004	280	
4	0.002	280	
5	0.001	280	
6	0.004	280	
7	0.002	110	
8	0.002	110	
9	0.002	110	
10	0.002	110	

Boeing Eddy Current Findings

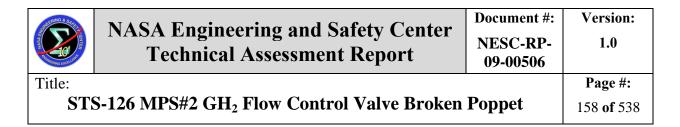
Poppet #37									
			Ru	ın Data (√pp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.243	0.251	0.248	0.254	0.247	0.251	0.249	Crack	290
J. Engel	- 10	20	8988	(e-	8	8	100	No	110
J. Engel	0.053	0.050	0.050	0.052	0.054	0.071	0.055	False	45 (Not 3:1 S/N ratio)
B. Devries	2	25	12	72	-0	9:	12	No	110
B. Devries	0.259	0.258	0.258	0.260	0.262	0.257	0.259	Crack	280

THIS A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

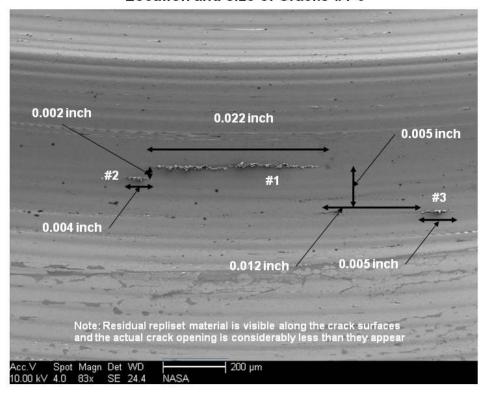
Poppet #37 Location of Cracks #1-6



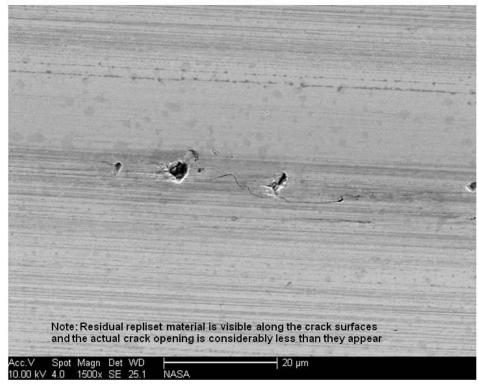
NESC Request No.: 09-00506



Location and size of Cracks #1-3



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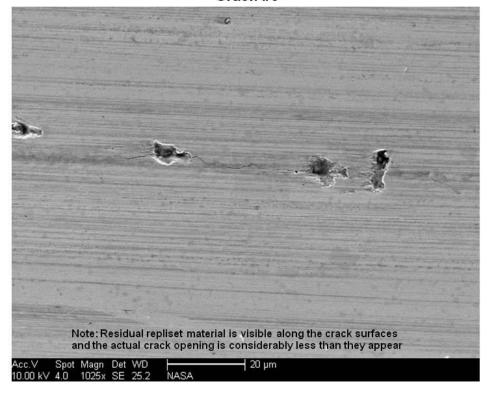


THE REPORT OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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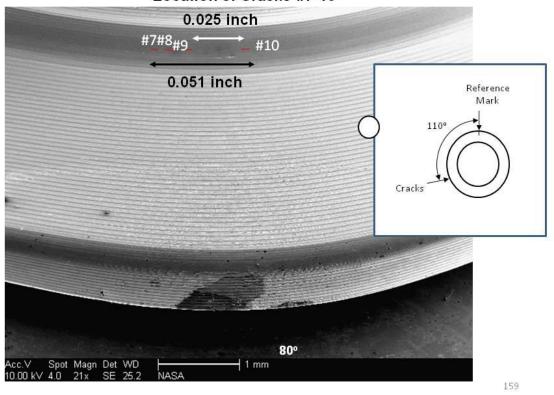
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Crack #6

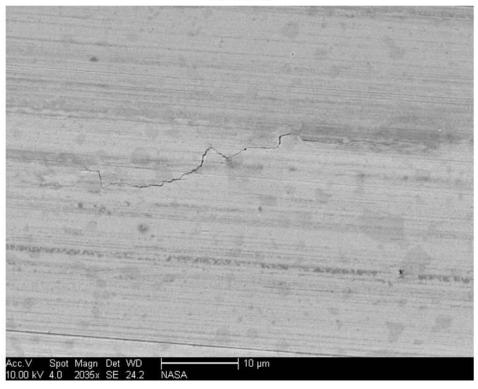


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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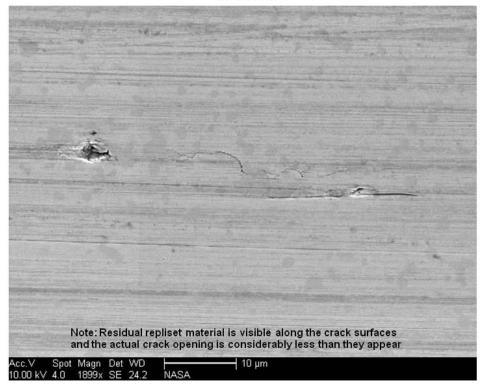
Poppet #37 Location of Cracks #7-10



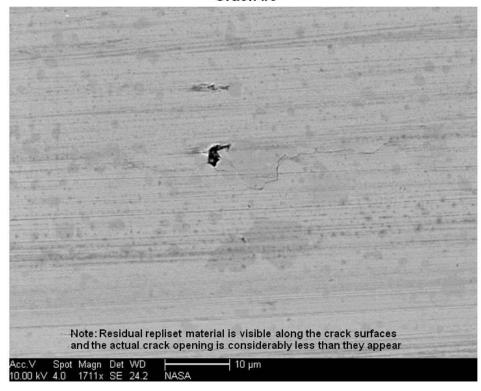
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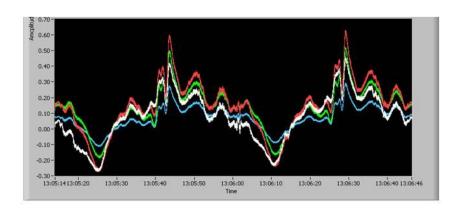


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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A SAINES			1.0
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Poppet #37
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

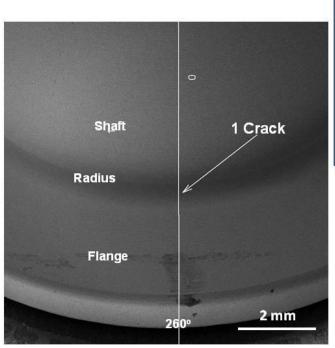
Poppet #38					
Crack Number	Size (inch)	Angle (degrees)			
1	0.074	260			
2	0.025	80			
3	0.011	80			
4	0.007	80			

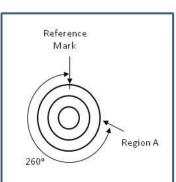
Boeing Eddy Current Findings

ř.	Poppet #38								
6) V			Ru	n Data (√pp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.901	0.885	0.891	0.902	0.911	0.912	0.900	Yes	95
J. Engel	0.078	0.075	0.075	0.078	0.076	0.080	0.077	Yes	265 (Not315/Nratio)
B. Devries	0.080	0.075	0.093	0.082	0.079	0.079	0.081	Yes	275
B. Devries	0.875	0.866	0.898	0.908	0.913	0.911	0.895	Yes	95

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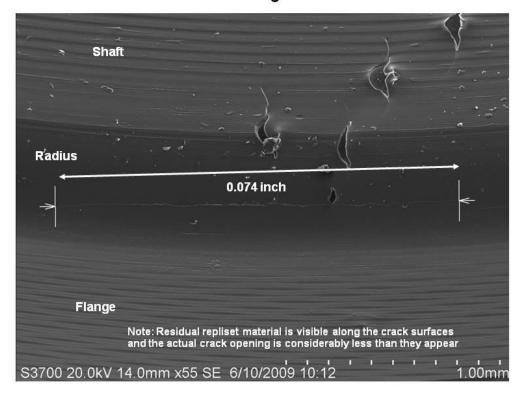
Poppet #38 Location of Region A





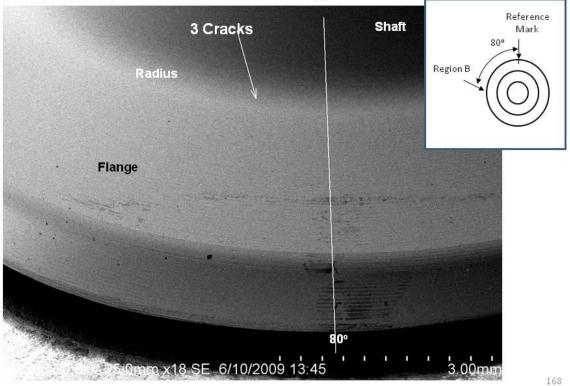
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Region A Cracks #1-3



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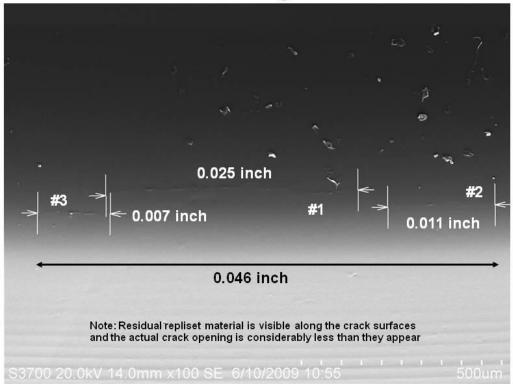
Poppet #38 Location of Region B



NESC Request No.: 09-00506

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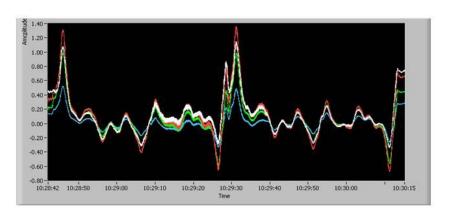
Poppet #38
Location and size of Region B Cracks #1-3



NESC Request No.: 09-00506

A STATE OF THE STA	NASA Engineering and Safety Center	Document #:	Version:		
	Technical Assessment Report	NESC-RP- 09-00506	1.0		
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LaRC eddy current findings, the colors indicate???



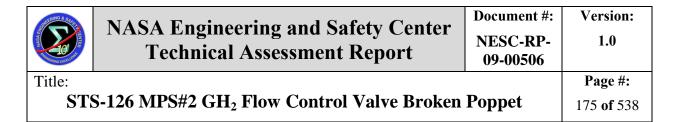
THIRD & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

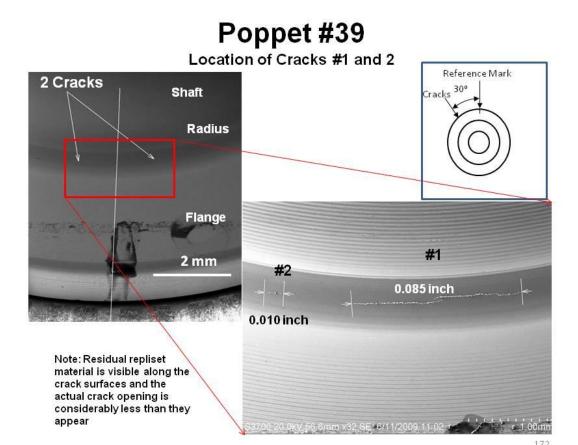
Surface crack sizes and locations

Poppet #39						
Crack Number	Size (inch)	Angle (degrees)				
1	0.085	30				
2	0.010	30				
3	0.056	210				

Boeing Eddy Current Findings

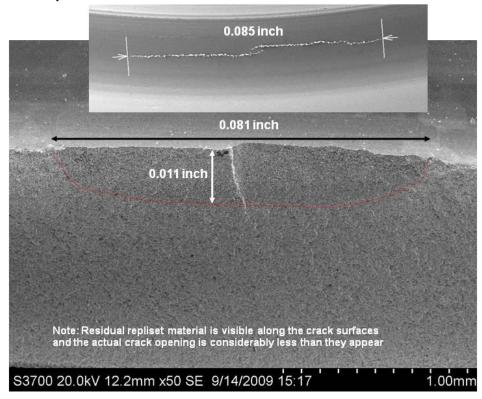
0 19	Poppet #39								
Inspector	Run Data (Vpp)								
	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.983	0.993	1.010	1.015	1.014	1.012	1.005	Yes	25
J. Engel	0.652	0.649	0.679	0.691	0.662	0.668	0.667	Yes	200
B. Devries	0.963	0.968	0.963	0.968	0.970	0.974	0.968	Yes	30
B. Devries	0.646	0.633	0.647	0.631	0.631	0.659	0.641	Yes	210

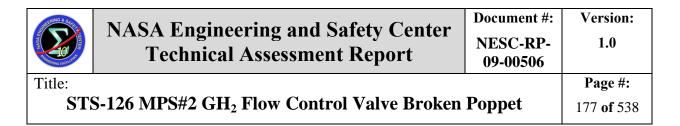




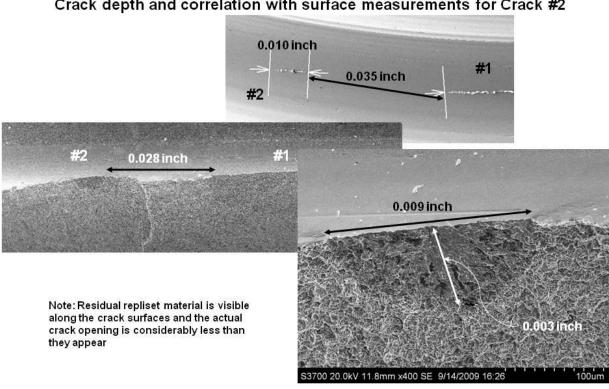
TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Crack depth and correlation with surface measurements for Crack #1

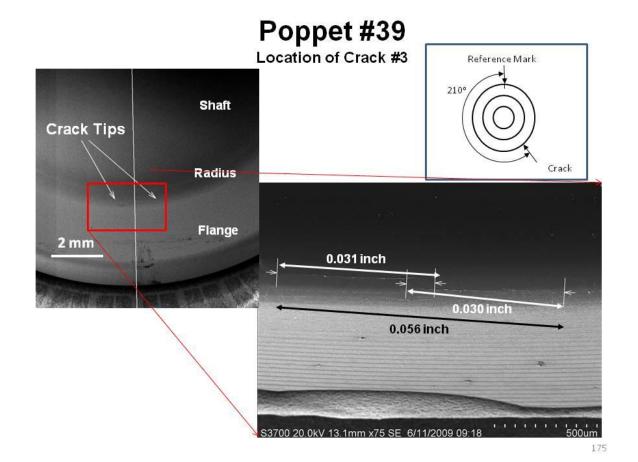




Crack depth and correlation with surface measurements for Crack #2

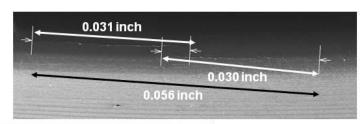


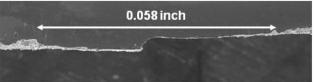
TO STORY OF THE PARTY OF THE PA	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
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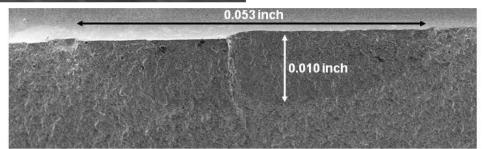


NO A CANAL DE LA C	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Crack depth and correlation with surface measurements for Crack #3

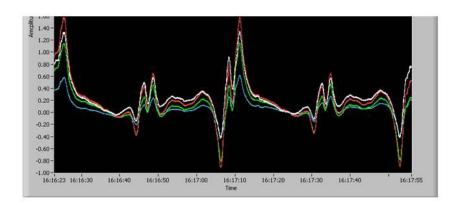






THE PARTY OF THE P	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #39
LaRC eddy current findings, the colors indicate ???



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A SAMPLE SERVICES	Technical Assessment Report	NESC-RP- 09-00506	1.0		
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Surface crack sizes and locations

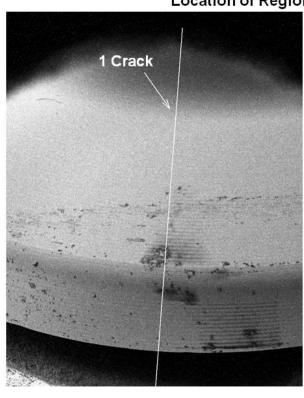
ž ė	Poppet #40	8
Crack Number	Size (inch)	Angle (degrees)
1	0.010	240

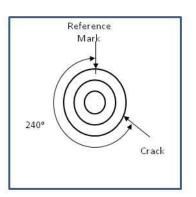
Boeing Eddy Current Findings

	Poppet #40									
ii.			Ru	n Data (√pp)				7	
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)	
J. Engel	0.082	0.077	0.079	0.081	0.078	0.081	0.080	Yes	230 (Small indication opposite)	
B. Devries	0.772	0.772	0.769	0.791	0.820	0.818	0.790	False	135	
B. Devries	- 2	E)	9826	144	9	9	14	No	240	

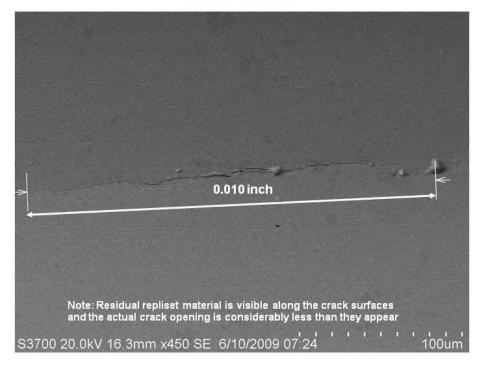
THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Poppet #40 Location of Region A



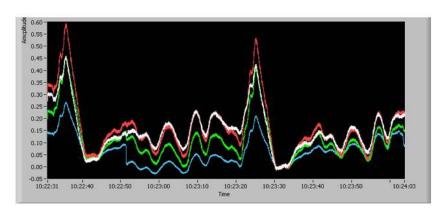


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Poppet #40
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

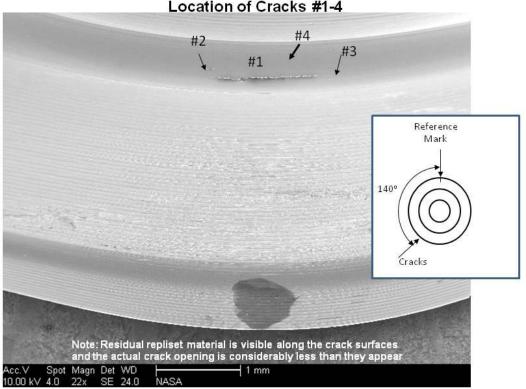
Poppet #41						
Crack Number	Size (inch)	Angle (degrees)				
1	0.050	140				
2	0.010	140				
3	0.001	140				
4	0.002	140				
5	0.002	320				

Boeing Eddy Current Findings

Poppet #41										
			Rı	ın Data (Vpp)					
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)	
J. Engel	0.767	0.791	0.808	0.812	0.805	0.828	0.802	Yes	130	
J. Engel	1.5	6	52	7.0	1553	1 25	125	No	320	
B. Devries	S#	- 8	- 8	#3	698	8		No	320	
B. Devries	0.820	0.824	0.819	0.824	0.819	0.817	0.821	Yes	130	

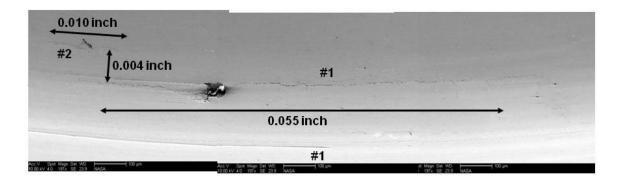
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #41 Location of Cracks #1-4



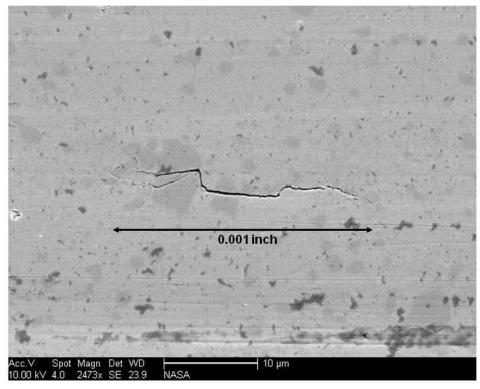
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #41 Location and size of Cracks #1 and 2

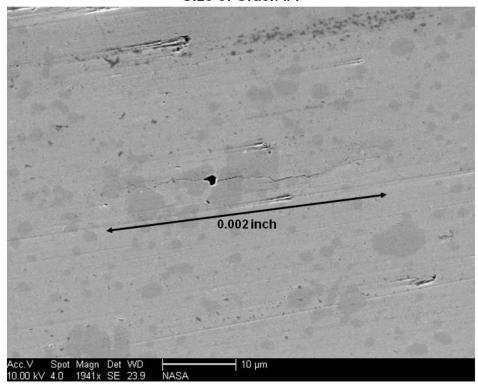


Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

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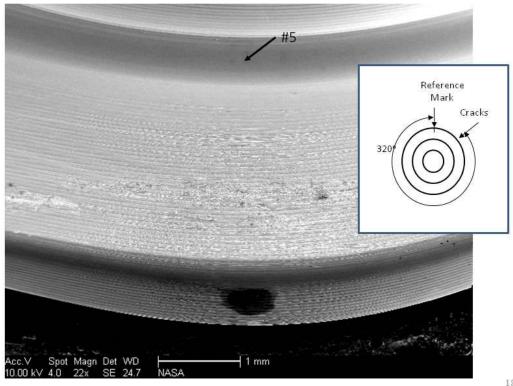


THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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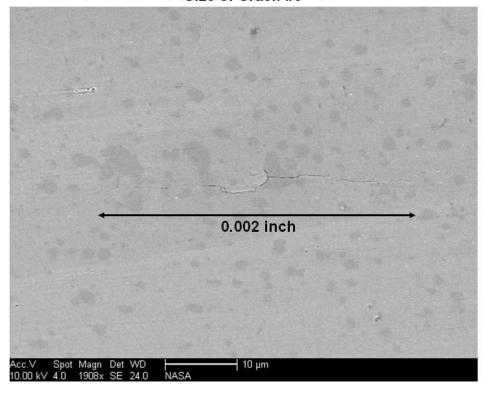


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Poppet #41 Location of Crack #5

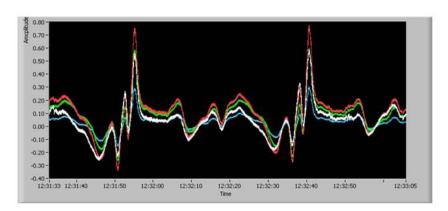


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Poppet #41
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

Poppet #42					
Crack Number	Size (inch)	Angle (degrees)			
1	0.025	260			
2	0.002	260			
3	0.004	260			
4	0.002	260			
5	0.004	260			
6	0.004	260			

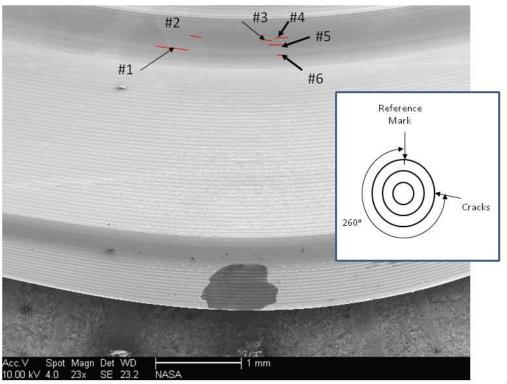
Boeing Eddy Current Findings

Poppet #42									
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.242	0.243	0.243	0.240	0.243	0.246	0.243	Crack	60
J. Engel	0.043	0.040	0.042	0.041	0.040	0.040	0.041	False	280 (Not315/N ratio)
B. Devries	0.253	0.268	0.263	0.259	0.264	0.281	0.265	Crack	75

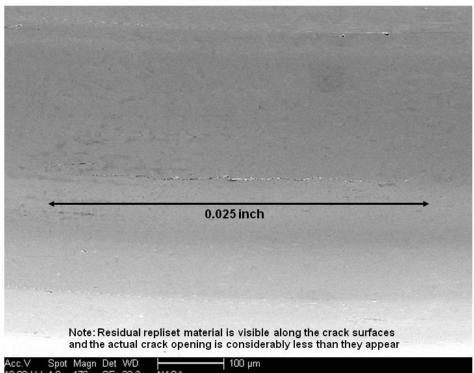
Note: The reference mark was very faint and a machining mark may have been mistaken for the reference mark

TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #42 Location of Cracks #1-6



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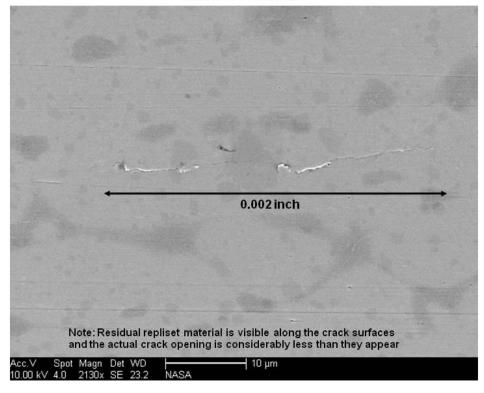


Spot Magn Det WD 4.0 176x SE 23.3

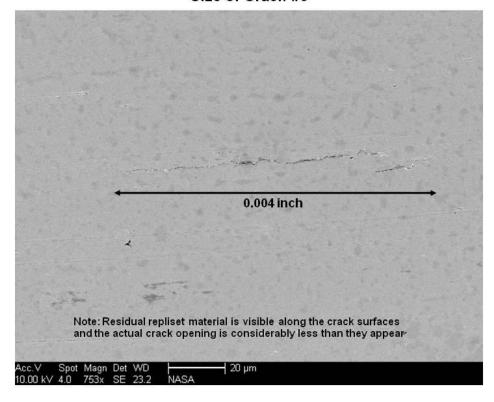
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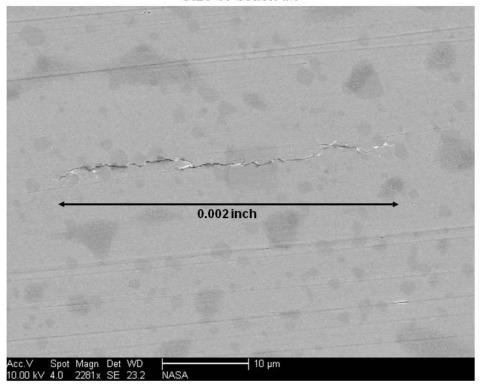
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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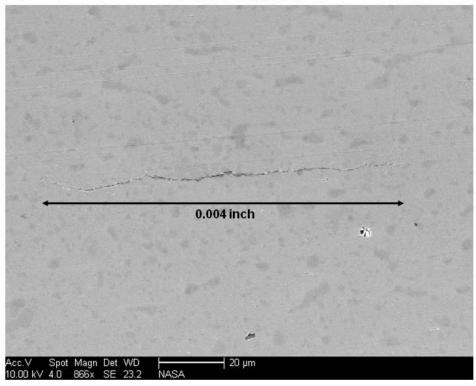
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Title:				



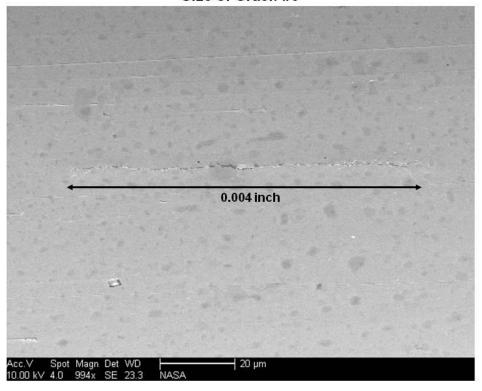
TOTAL STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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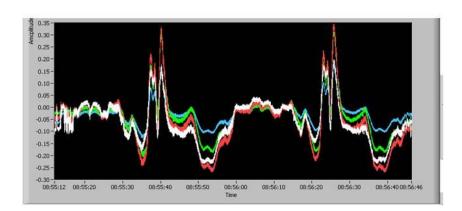


TOTAL STREET	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Poppet #42
LaRC eddy current findings, the colors indicate ???



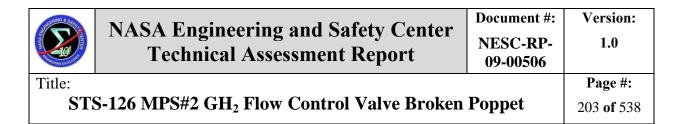
SHERING & SAFE	NASA Engineering and Safety Center		Version:	
TO LEASE STATE OF THE STATE OF	Technical Assessment Report	NESC-RP- 09-00506	1.0	
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Surface crack sizes and locations

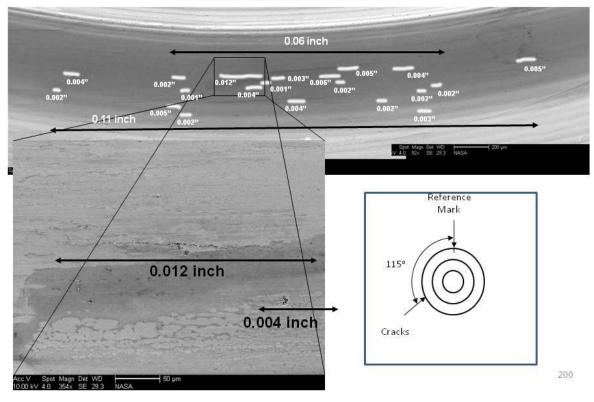
	Poppet #43						
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)		
1	0.001	115	16	0.002	115		
2	0.005	115	17	0.006	115		
3	0.002	115	18	0.005	115		
4	0.004	115	19	0.004	115		
5	0.003	115	20	0.005	115		
6	0.002	115	21	0.002	295		
7	0.002	115	22	0.001	295		
8	0.002	115	23	0.002	295		
9	0.002	115	24	0.001	295		
10	0.003	115	25	0.002	295		
11	0.001	115	26	0.001	295		
12	0.004	115	27	0.002	295		
13	0.012	115			983031		
14	0.004	115					
15	0.001	115					

Boeing Eddy Current Findings

	Poppet #43								
25.50-000.03000.000	7		Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.152	0.147	0.144	0.143	0.148	0.145	0.147	Yes	355?
J. Engel	0175	578	. 35	. 5	- 51	8975	1	No	295
J. Engel	0.047	0.045	0.040	0.041	0.045	0.043	0.044	Yes	115
B. Devries	0.146	0.150	0.148	0.144	0.151	0.151	0.148	Yes	115
B. Devries	2928	363	99	- 2	. 25	7528	S 50	No	295

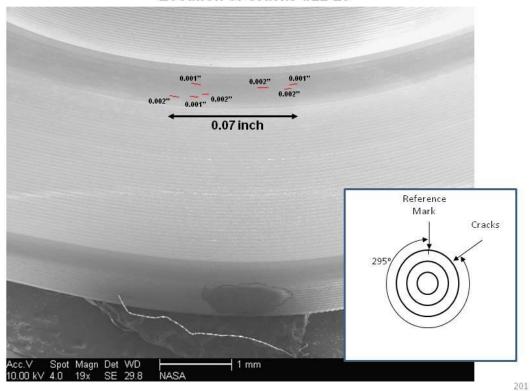


Location and size of Cracks #1-20



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ST	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

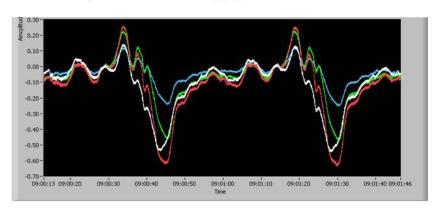
Poppet #43 Location of Cracks #22-29



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Poppet #43
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

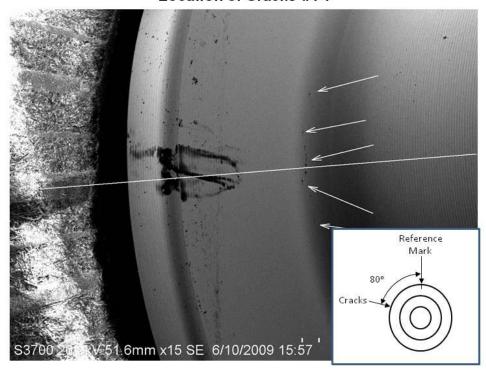
Poppet #44			
Crack Number	Size (inch)	Angle (degrees)	
1	0.029	80	
2	0.008	80	
3	0.002	80	
4	0.002	80	
5	0.009	80	
6	0.002	80	
7	0.007	80	
8	0.004	80	
9	0.004	80	
10	0.002	80	
11	0.008	80	
12	0.004	80	
13	0.006	260	
14	0.004	260	
15	0.003	260	

Boeing Eddy Current Findings

Poppet #44									
SHEETSEN SOLIS	Run Data (Vpp)								
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.246	0.253	0.279	0.279	0.281	0.282	0.270	Yes	80
J. Engel	0.040	0.043	0.041	0.045	0.041	0.045	0.043	Yes	260 (Not3:15/Nratio)
B. Devries	S23	, E	-0	. 2	12	N2	\$1 J	No	260
B. Devries	0.284	0.277	0.282	0.284	0.251	0.271	0.275	Yes	75

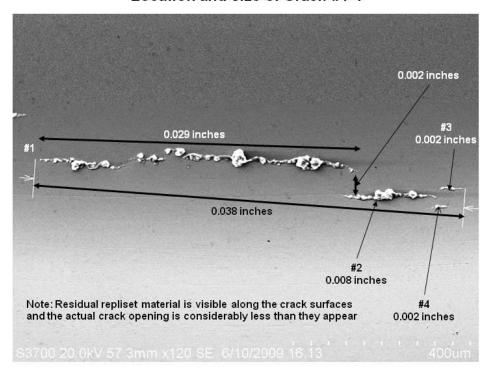
TO STORY OF THE PARTY OF THE PA	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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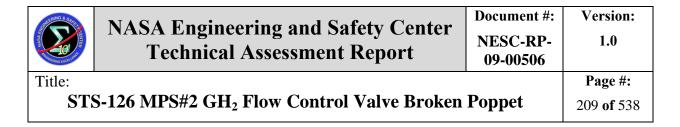
Poppet #44 Location of Cracks #1-7



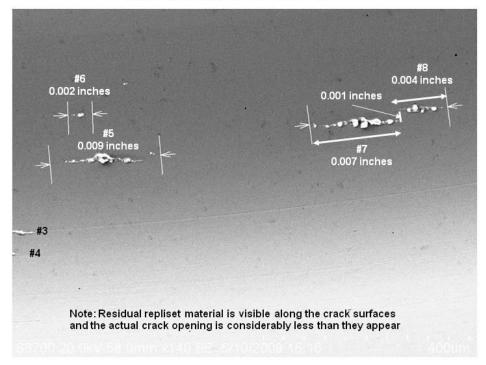
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Crack #1-4



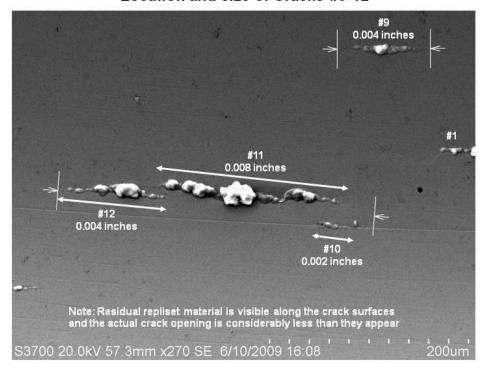


Location and size of Cracks #5-8



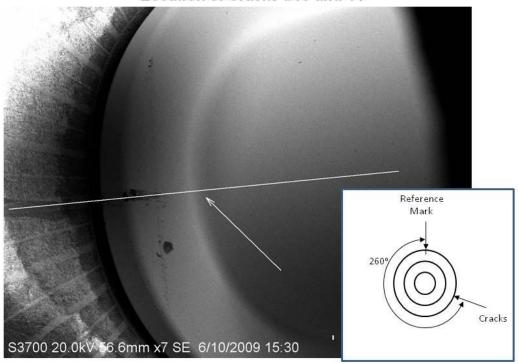
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #9-12



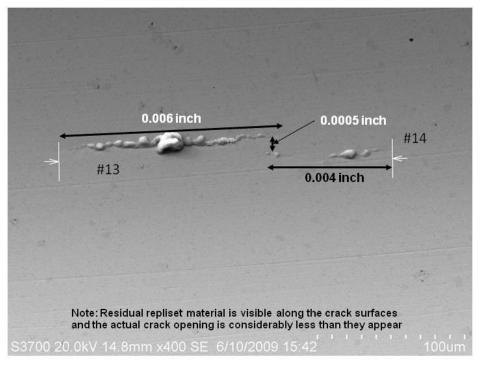
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #44 Location of Cracks #13 and 14



THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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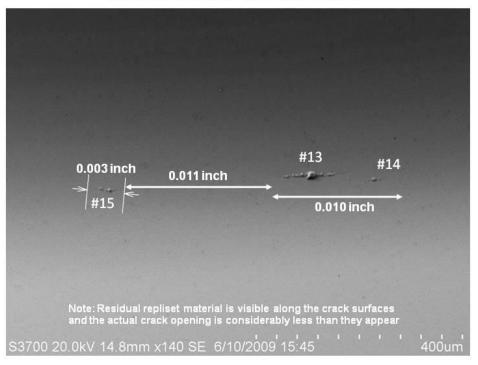
Poppet #44 Location and size of Cracks #8 & 9



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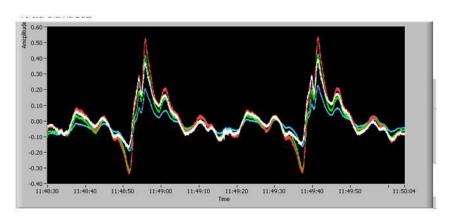
Poppet #44

Location and size of Cracks #13-14



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Poppet #44
LaRC eddy current findings, the colors indicate ???



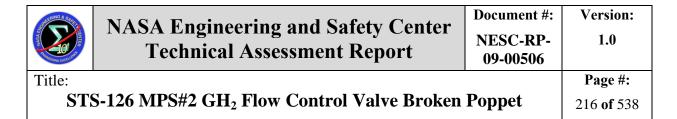
TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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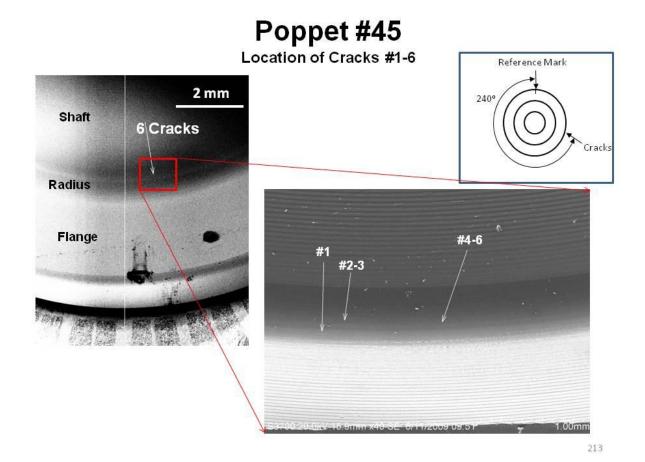
Surface crack sizes and locations

Poppet #45				
Crack Number	Size (inch)	Angle (degrees)		
1	0.014	240		
2	0.009	240		
3	0.006	240		
4	0.005	240		
5	0.004	240		
6	0.002	240		
7	0.005	55		
8	0.004	55		
9	0.009	55		
10	0.006	55		

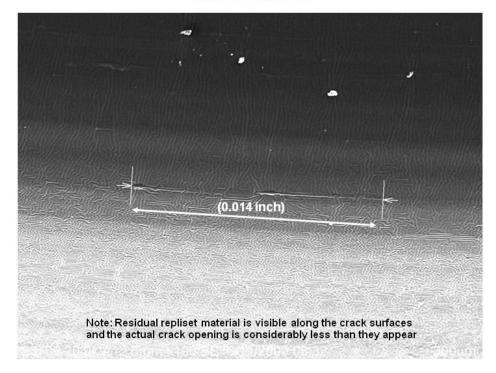
Boeing Eddy Current Findings

Poppet #45									
8 69 6			Ru	ın Data (Vpp)		10.		
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.091	0.075	0.075	0.082	0.078	0.078	0.080	Yes	225 (Short indication)
J. Engel	0.078	0.072	0.079	0.078	0.081	0.088	0.079	Yes	60 (Longer indication)
3. Devries	0.068	0.060	0.065	0.069	0.063	0.066	0.065	Yes	240
3. Devries	0.084	0.079	0.086	0.091	0.090	0.095	0.088	Yes	60



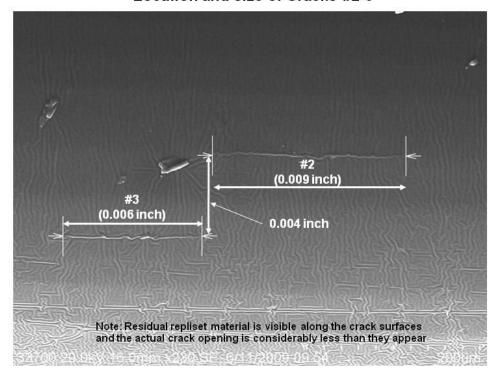


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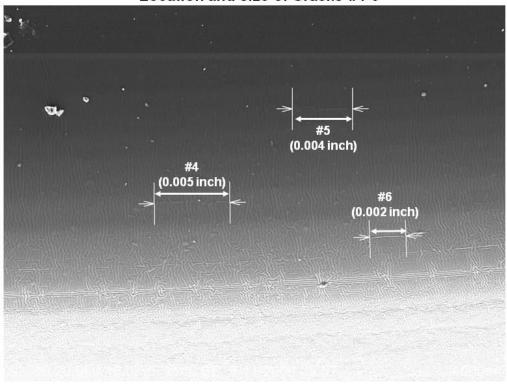
SHERING & SAFE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
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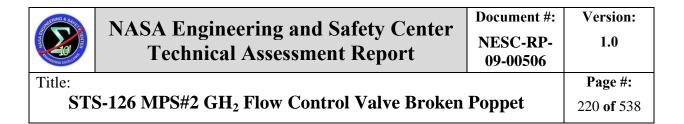
Poppet #45 Location and size of Cracks #2-3

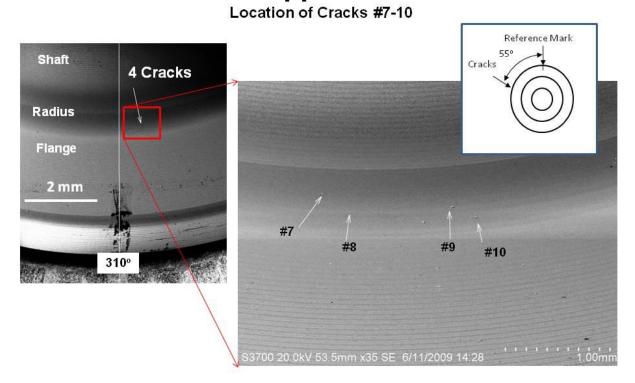


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #45 Location and size of Cracks #4-6

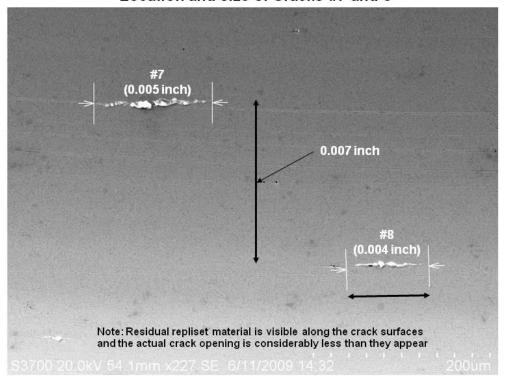






	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		Page #: 221 of 538	

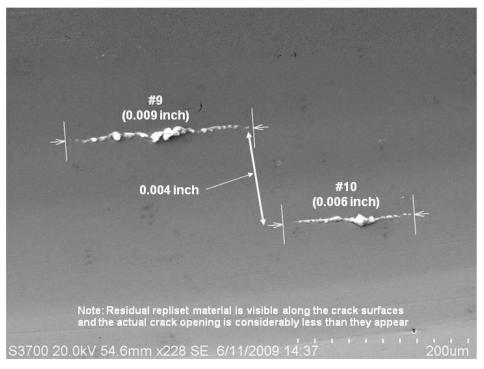
Poppet #45
Location and size of Cracks #7 and 8



	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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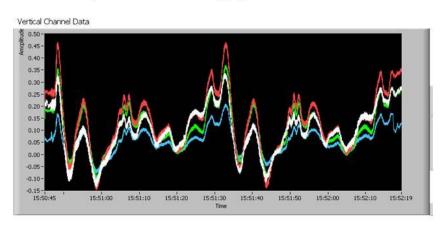
Poppet #45

Location and size of Cracks #9 and 10



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Poppet #45 LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

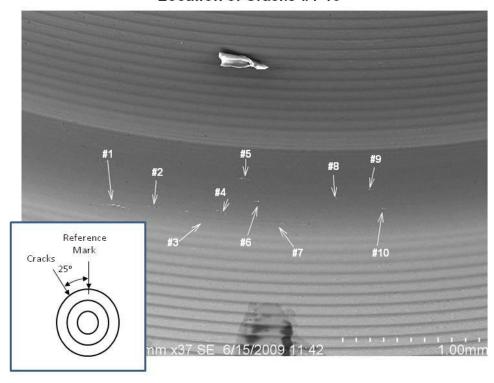
Poppet #46			
Crack Number	Size (inch)	Angle (degrees)	
1	0.004	25	
2	0.005	25	
3	0.003	25	
4	0.004	25	
5	0.006	25	
6	0.005	25	
7	0.003	25	
8	0.003	25	
9	0.004	25	
10	0.004	25	
11	0.012	230	
12	0.007	230	
13	0.003	230	
14	0.005	230	
15	0.004	230	
16	0.003	230	
17	0.002	220	
18	0.002	220	

Boeing Eddy Current Findings

	Poppet #46								
SOUR DANIES AND COM-			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.098	0.099	0.100	0.093	0.092	0.103	0.098	Yes	195
J. Engel	0.067	0.066	0.073	0.066	0.068	0.073	0.069	Yes	10 (Not315/Nratio)
B. Devries	899	8	9	- 8	35913	89	49	No	25
B. Devries	0.103	0.104	0.106	0.102	0.095	0.098	0.101	Yes	200

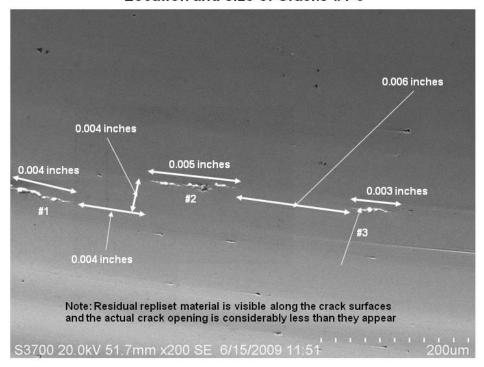
TO THE PROPERTY OF THE PROPERT	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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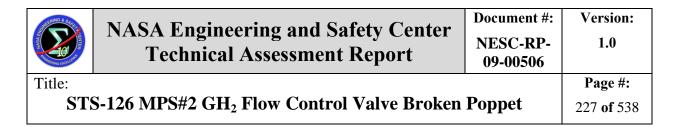
Poppet #46 Location of Cracks #1-10



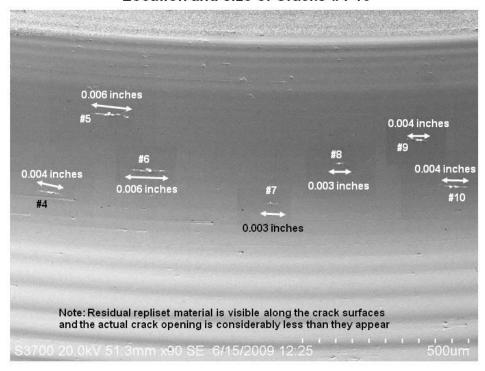
TO SECOND	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #46 Location and size of Cracks #1-3



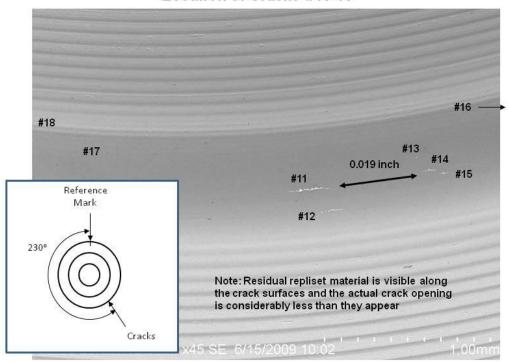


Location and size of Cracks #4-10



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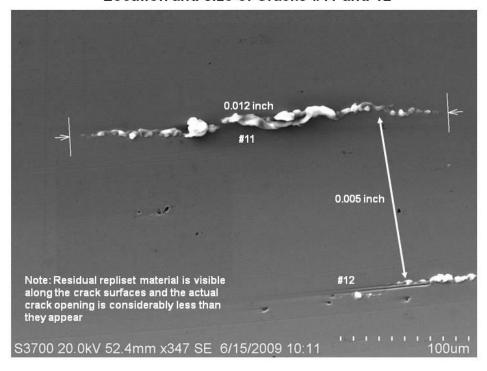
Poppet #46 Location of Cracks #11-18



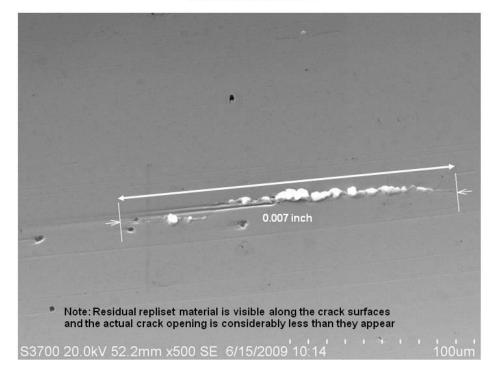
TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #46

Location and size of Cracks #11 and 12

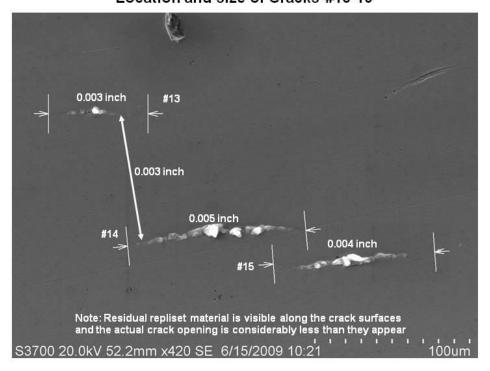


THE PART OF THE PA	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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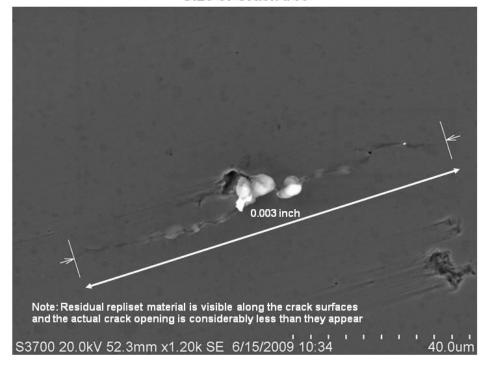
THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #46
Location and size of Cracks #13-15

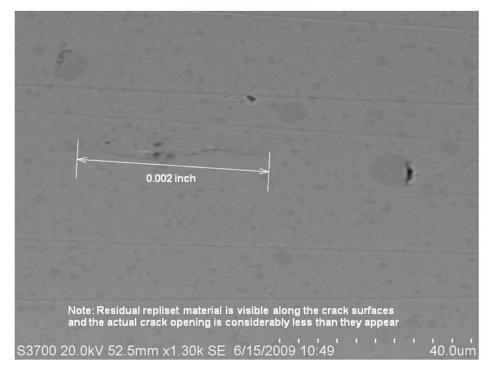


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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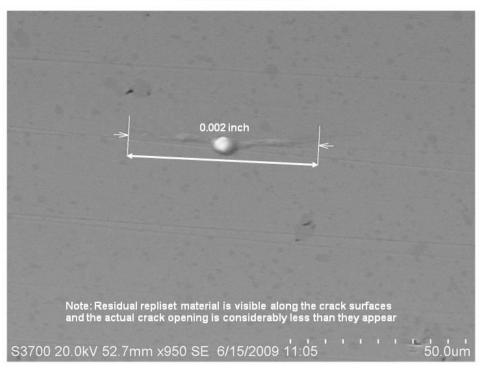
Size of Crack #16



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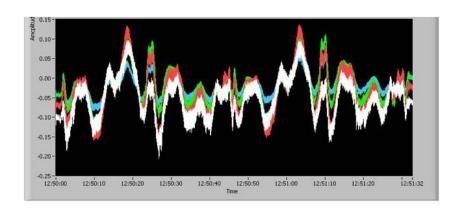


TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #46
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

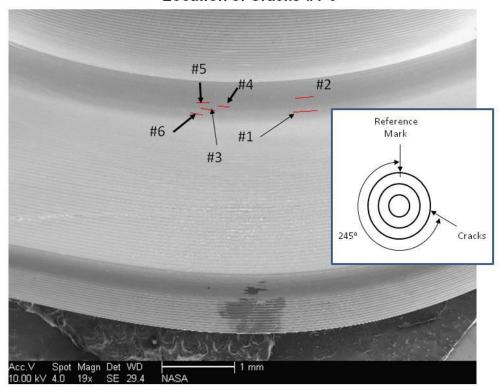
Poppet #47			
Crack Number	Size (inch)	Angle (degrees)	
1	0.014	245	
2	0.011	245	
3	0.005	245	
4	0.005	245	
5	0.004	245	
6	0.004	245	
7	0.004	65	

Boeing Eddy Current Findings

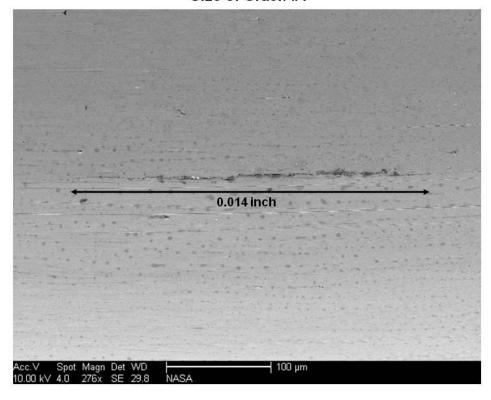
Poppet #47									
			Ru	ın Data (√pp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.139	0.140	0.140	0.141	0.142	0.142	0.141	Yes	245
J. Engel	0.056	0.059	0.062	0.060	0.058	0.059	0.059	Yes	65 (Not3:1S/N ratio)
3. Devries	0.144	0.141	0.146	0.150	0.144	0.141	0.144	Yes	255
B. Devries	9-	25	848	12	-2	9:	1 12	No	65

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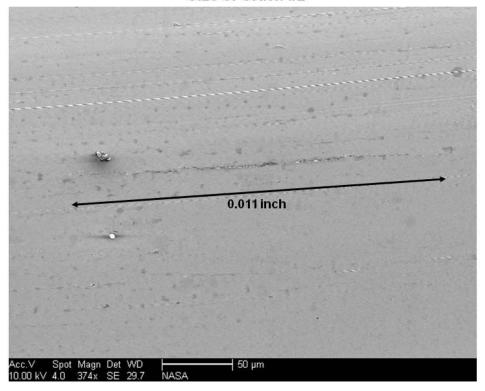
Poppet #47 Location of Cracks #1-6



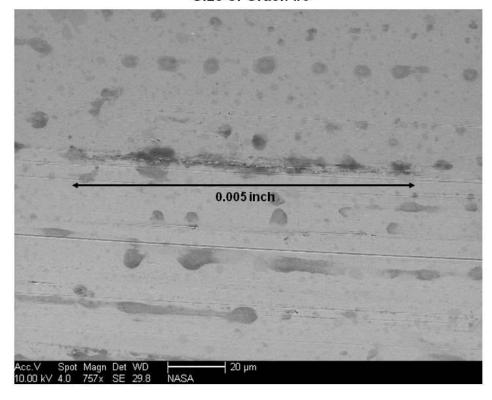
TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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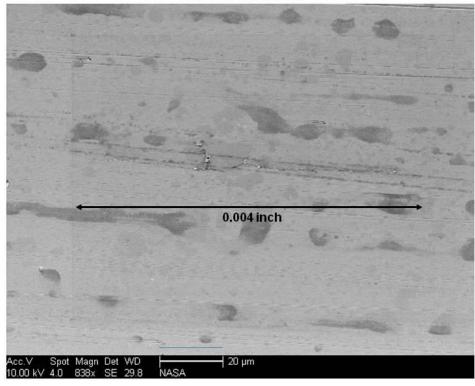
THE STATE OF THE S	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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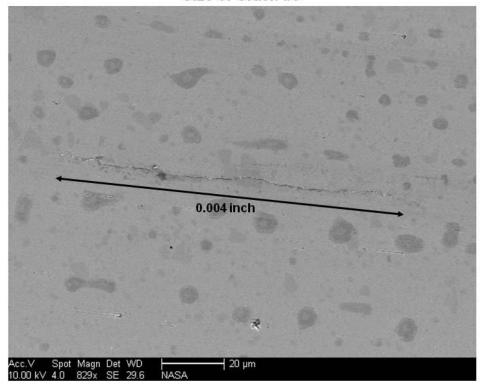
SHEERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:	
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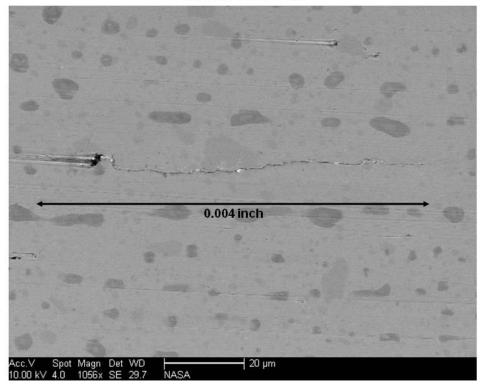
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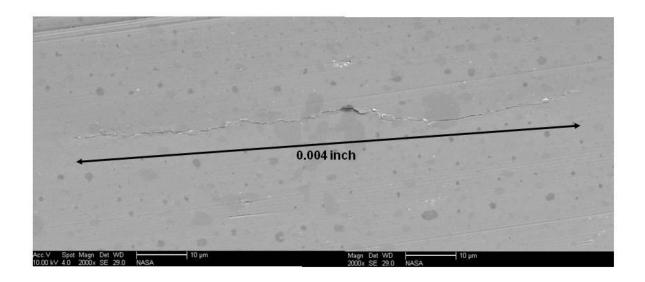


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Poppet #47 Location of Crack #7

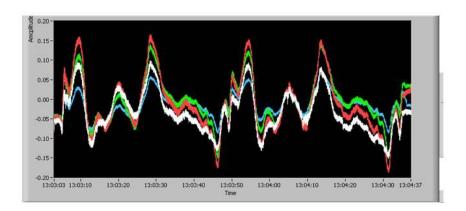
Reference Mark 65° Cracks

TOWN A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #47
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

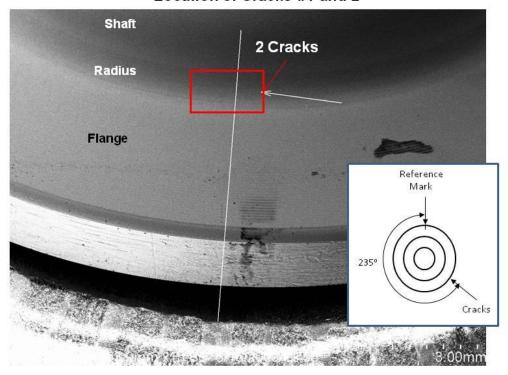
Poppet #48					
Crack Number	Size (inch)	Angle (degrees)			
1	0.005	235			
2	0.004	235			
3	0.002	55			
4	0.034	55			
5	0.007	55			
6	0.012	55			
7	0.006	55			
8	0.002	55			
9	0.001	55			
10	0.003	55			
11	0.002	55			
12	0.002	55			
13	0.004	55			
14	0.002	55			

Boeing Eddy Current Findings

Poppet #48									
Inspector	Run Data (Vpp)								
	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.344	0.350	0.351	0.358	0.361	0.355	0.353	Yes	45
J. Engel		8	. 20	898	5 7 86	95	8 7 /8	Yes	235 (Samll indication slightly above noise)
B. Devries	95	8	86	1250	953	85	950	No	235
B. Devries	0.342	0.340	0.342	0.342	0.341	0.344	0.342	Yes	45

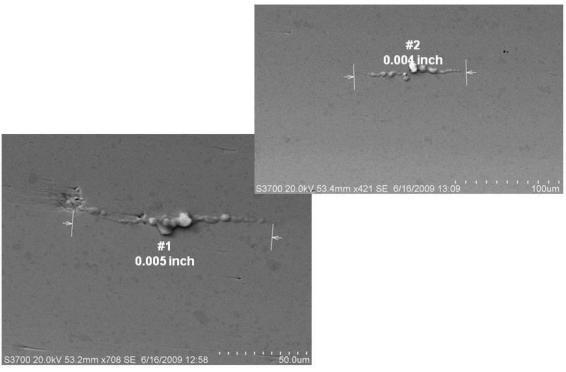
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0			
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet						

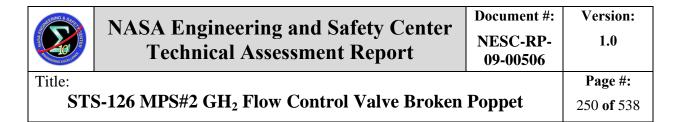
Poppet #48 Location of Cracks #1 and 2

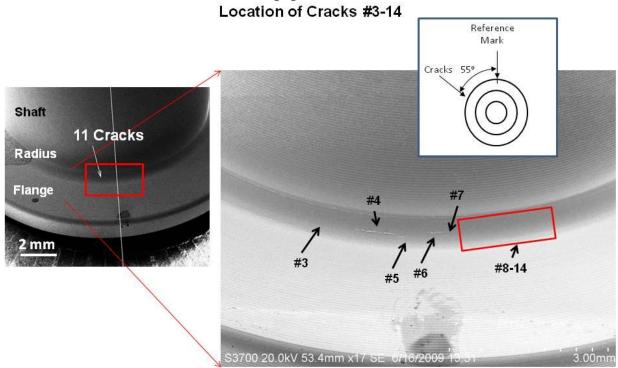


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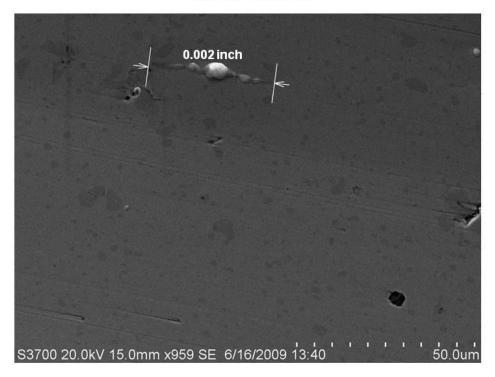
Poppet #48 Location and size of Cracks #1 and 2





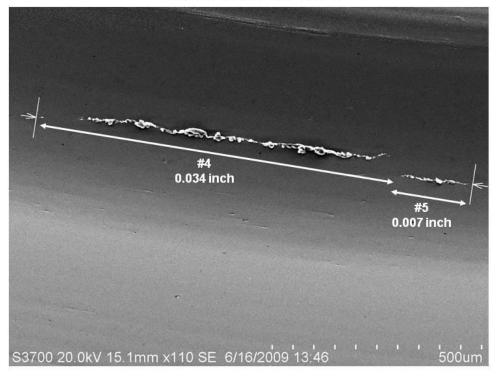


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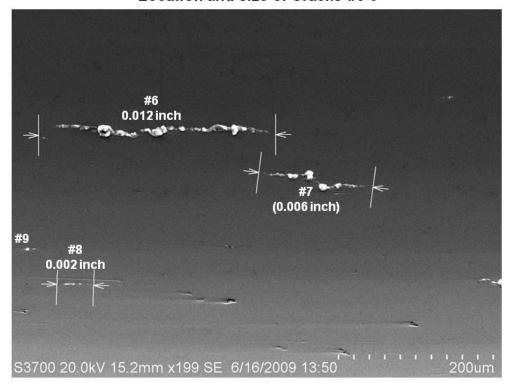
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #48 Size of Cracks #4 and 5



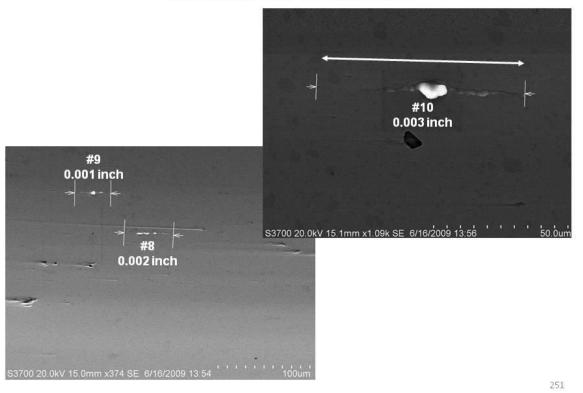
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #6-9



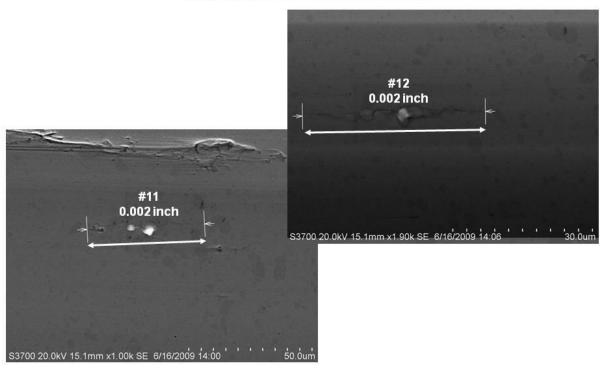
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #48 Location and size of Cracks #8-10



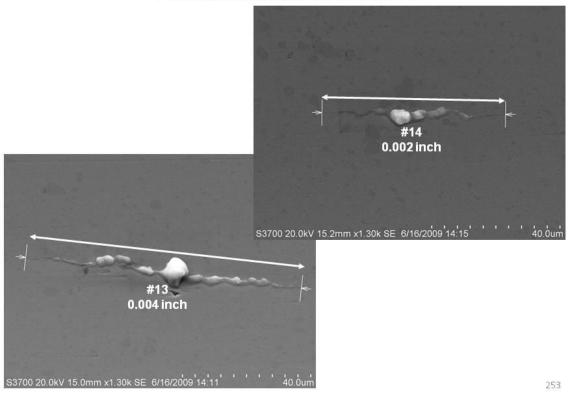
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #48 Location and size of Cracks #11-12



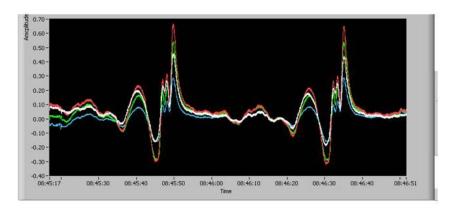
THING & CALLED	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Poppet #48 Location and size of Cracks #13-14



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Poppet #48
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

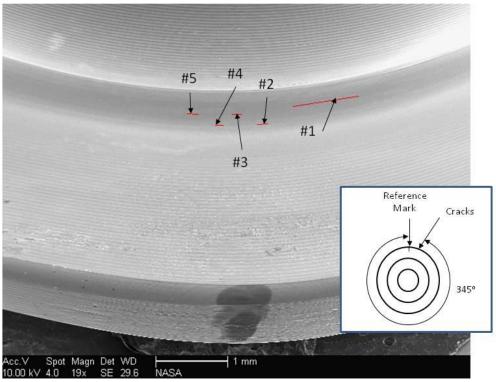
Poppet #49					
Crack Number	Size (inch)	Angle (degrees)			
1	0.045	345			
2	0.003	345			
3	0.003	345			
4	0.007	345			
5	0.004	345			
6	0.002	165			

Boeing Eddy Current Findings

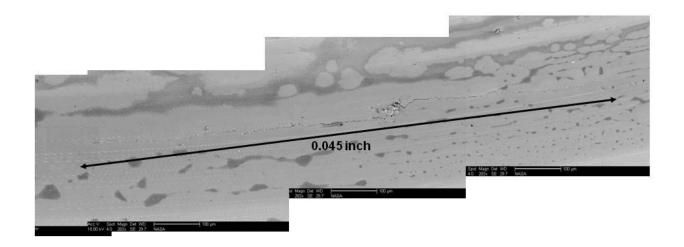
	Poppet #49									
SE 307 SE			Ru	n Data (\	√pp)	- 112/11 				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)	
J. Engel	-	40	3593	-	9	-8	-	No	165	
J. Engel	0.827	0.795	0.801	0.805	0.810	0.802	0.807	Yes	350	
B. Devries	2	28	7028	- 84	- 2	84	99	No	165	
B. Devries	0.777	0.789	0.789	0.818	0.814	0.822	0.802	Yes	350	

THING A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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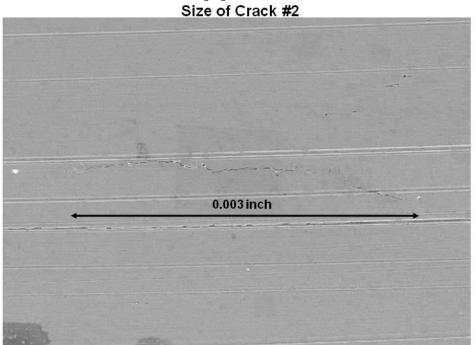
Poppet #49 Location of Cracks #1-6



TRING & CALLEGE AND A CALLEGE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:				

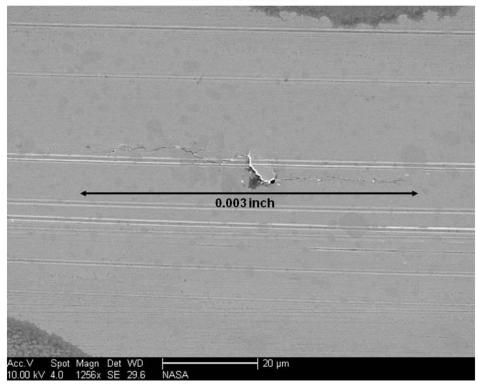


TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

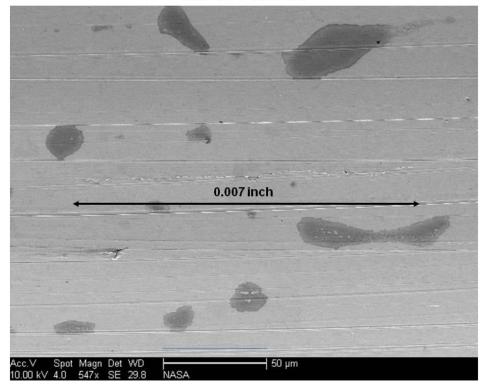


20 µm

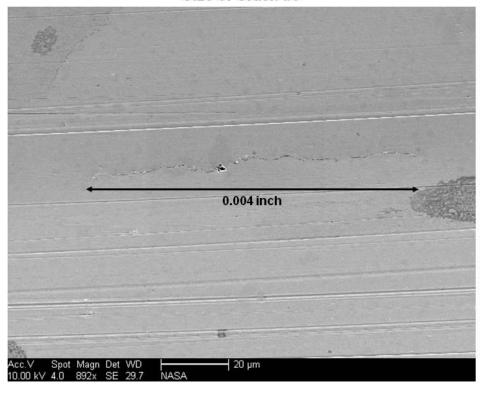
SHEERING & SAFE	NASA Engineering and Safety Center		Version:	
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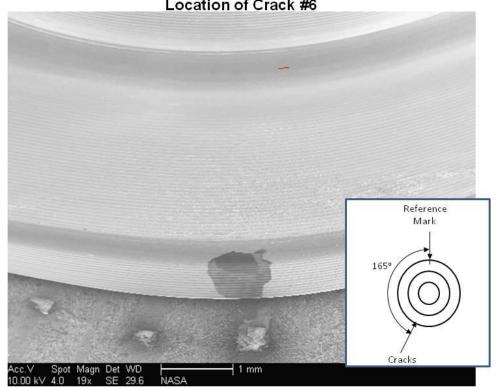


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

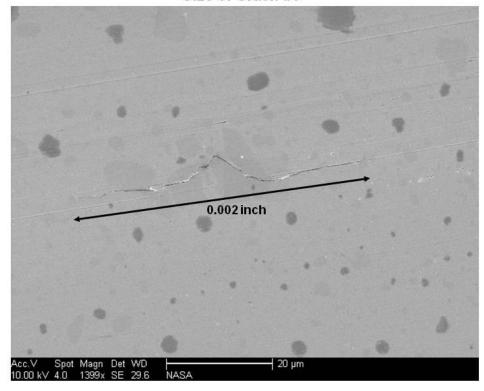


muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #49 Location of Crack #6

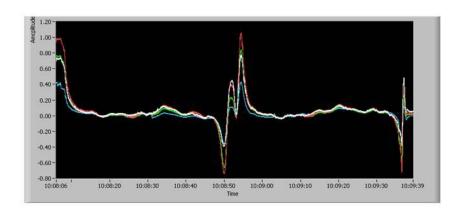


NG & CALLED TO THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #49
LaRC eddy current findings, the colors indicate ???



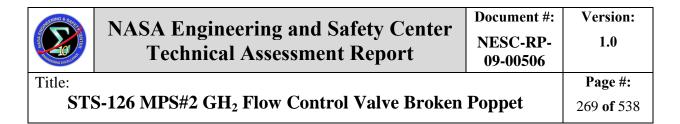
TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Surface crack sizes and locations

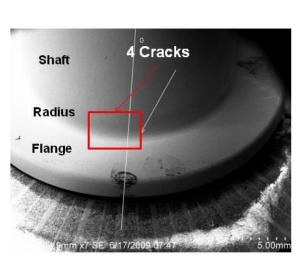
Poppet #50					
Crack Number	Size (inch)	Angle (degrees)			
1	0.004	155			
2	0.003	155			
3	0.005	155			
4	0.003	155			
5	0.006	335			
6	0.009	335			
7	0.010	335			
8	0.005	335			
9	0.010	335			
10	0.003	335			
11	0.002	335			

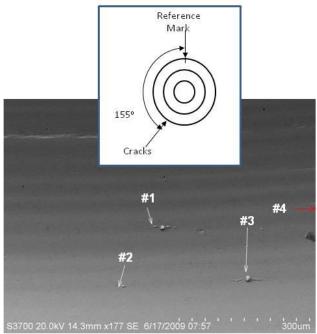
Boeing Eddy Current Findings

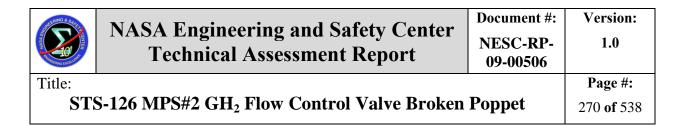
5	Poppet #50								
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.213	0.204	0.211	0.216	0.215	0.212	0.212	Yes	340
J. Engel	0.055	0.052	0.059	0.061	0.062	0.053	0.057	Yes	170 (Not31S/N ratio)
B. Devries		. 59	8758	1 55			15	No	155
B. Devries	0.205	0.210	0.210	0.211	0.208	0.207	0.209	Yes	330



Location of Cracks #1-4

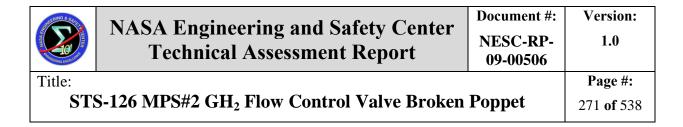




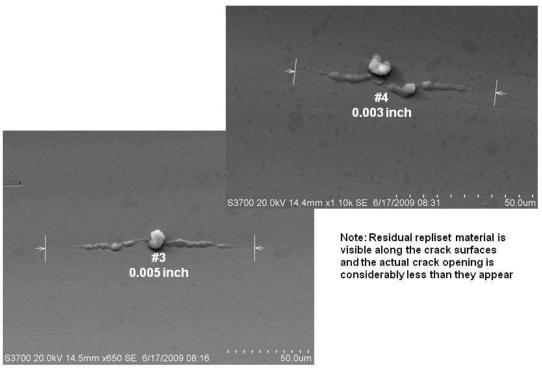


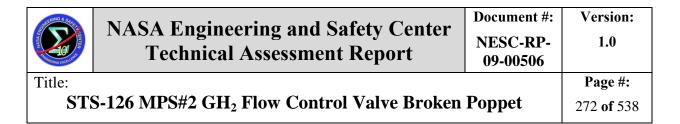
Size of Cracks #1 and 2



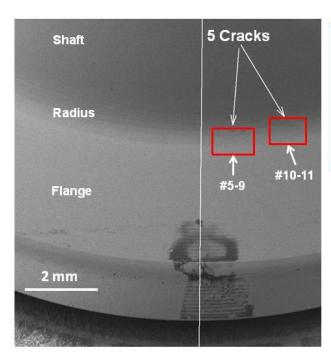


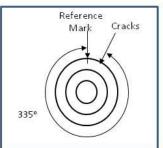
Location and size of Cracks #3 and 4





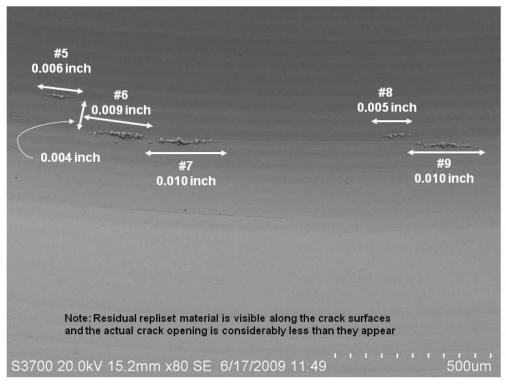
Location of Cracks #5-9

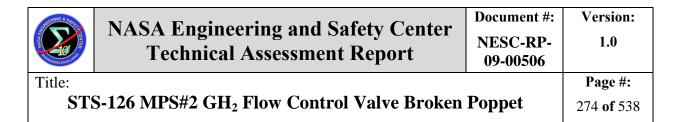




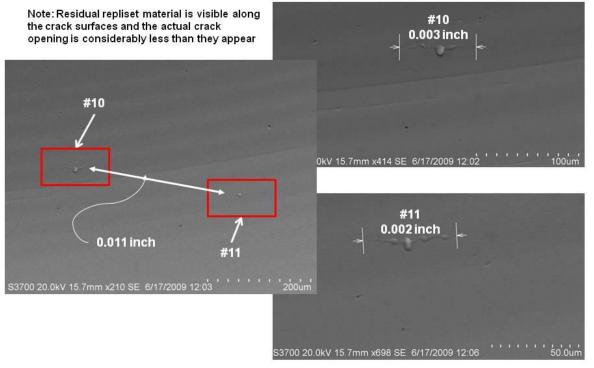
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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Location and size of Cracks #5-9



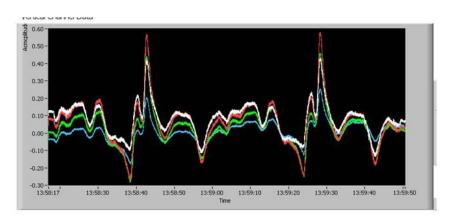


Size of Cracks #10-11



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LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

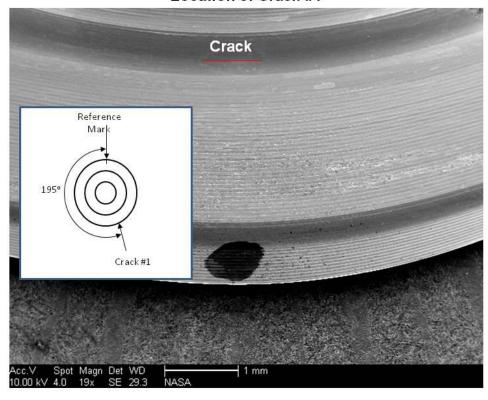
Poppet #51					
Crack Number	Size (inch)	Angle (degrees)			
1	0.032	195			
2	0.108	15			

Boeing Eddy Current Findings

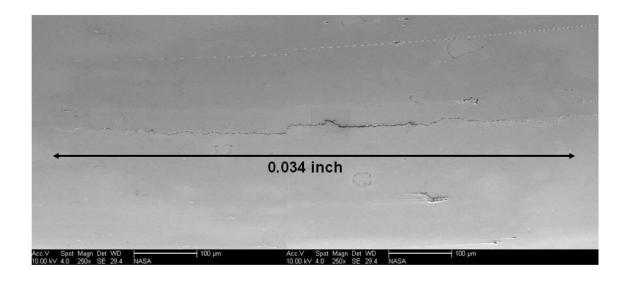
Poppet #51									
8 10 10		ns.	Ru	ın Data (Vpp)	3	ag.		53
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	1.006	1.022	1.048	1.054	1.071	1.068	1.045	Yes	35
J. Engel	0.410	0.415	0.412	0.422	0.422	0.430	0.419	Yes	210
B. Devries	1.047	1.059	1.067	1.063	1.082	1.064	1.064	Yes	20
B. Devries	0.420	0.420	0.420	0.416	0.420	0.420	0.419	Yes	200

TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #51 Location of Crack #1

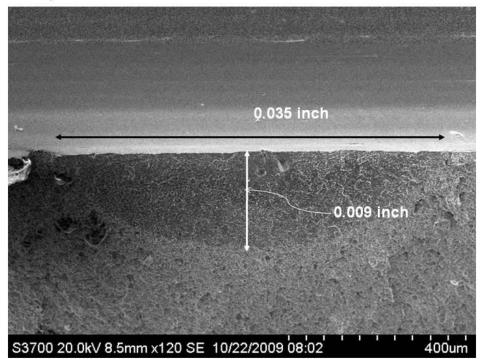


TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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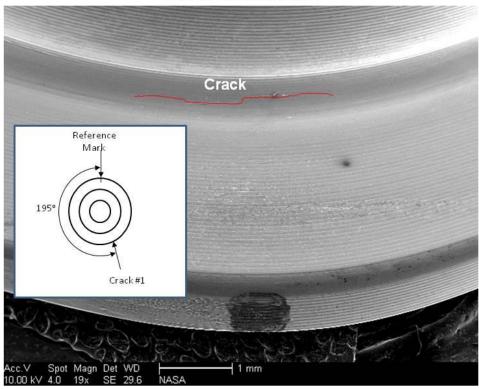
THING & CALLED THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #51
Crack depth and correlation with surface measurements for Crack #1

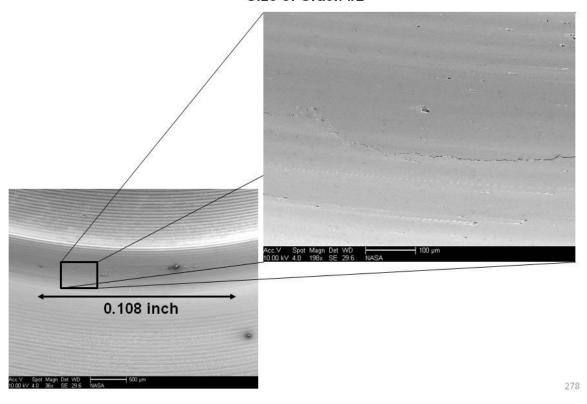


muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #51 Location of Crack #2

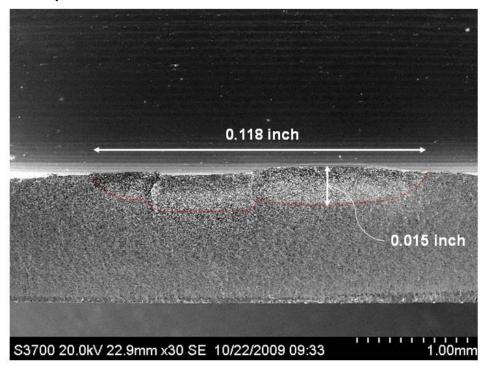


THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		



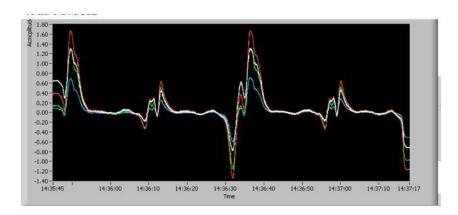
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Crack depth and correlation with surface measurements for Crack #2



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Poppet #51
LaRC eddy current findings, the colors indicate ???



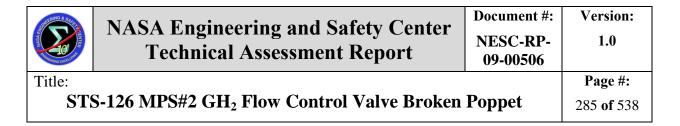
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Surface crack sizes and locations

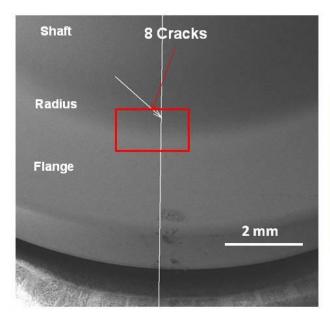
	Poppet #52										
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees						
1	0.003	115	11	0.014	295						
2	0.004	115	115 12 0.006		295						
3	0.004	115	13	0.014 0.008 0.006	295 295 295						
4	0.004	115	14								
5	0.006	115	15								
6	0.005	115	16	0.014	295						
7	0.010	115	17	0.008	295						
8	8 0.005		18	0.004	295						
9	0.012	295	19	0.005	295						
10	0.013	295	20	0.003	295						

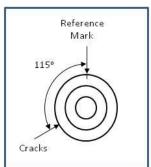
Boeing Eddy Current Findings

Poppet #52											
Inspector	Run Data (Vpp)										
	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)		
J. Engel	0.214	0.219	0.223	0.220	0.227	0.222	0.221	Yes	290		
l. Engel	0.096	0.096	0.096	0.101	0.099	0.106	0.099	Yes	115		
3. Devries	0.093	0.098	0.092	0.099	0.092	0.097	0.095	Yes	110		
B. Devries	0.218	0.217	0.215	0.218	0.219	0.218	0.218	Yes	290		



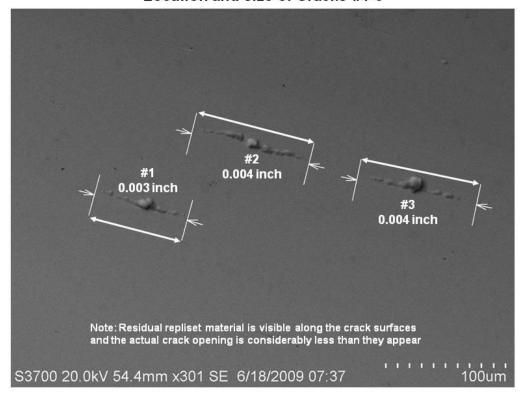
Location of Cracks #1-8





TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:				

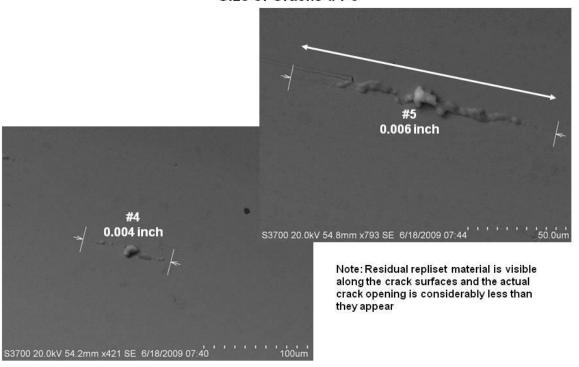
Location and size of Cracks #1-3



NESC Request No.: 09-00506

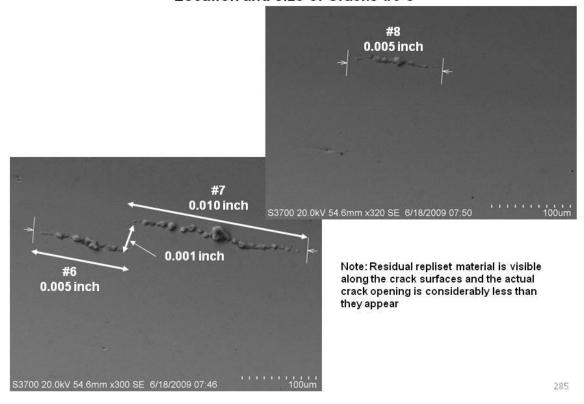
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #52 Size of Cracks #4-5

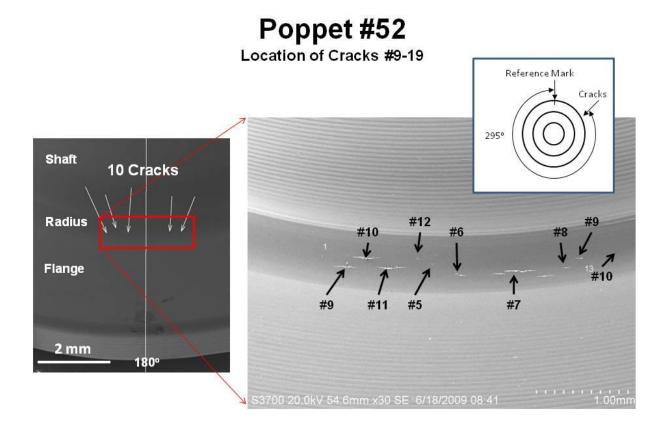


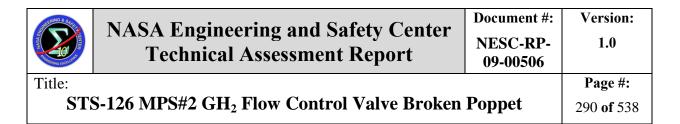
STORY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #6-8

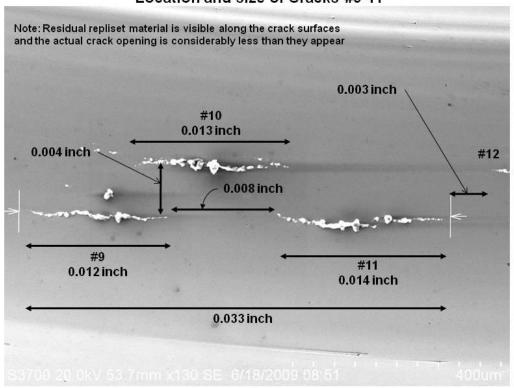


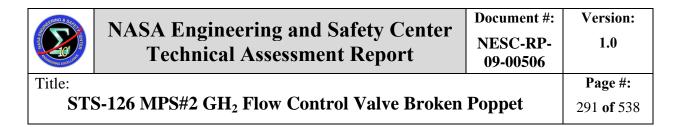
STORY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
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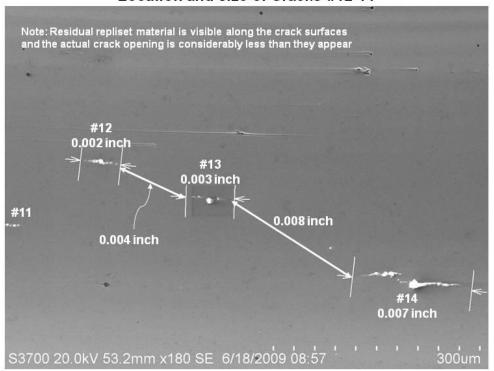


Location and size of Cracks #9-11



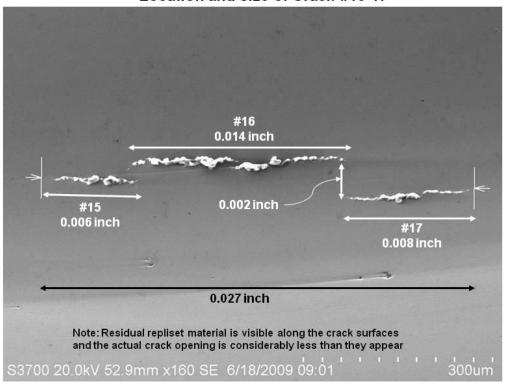


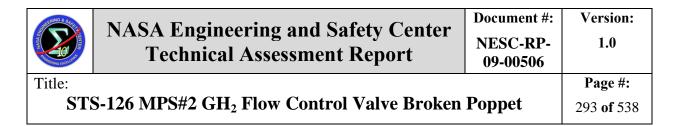
Location and size of Cracks #12-14

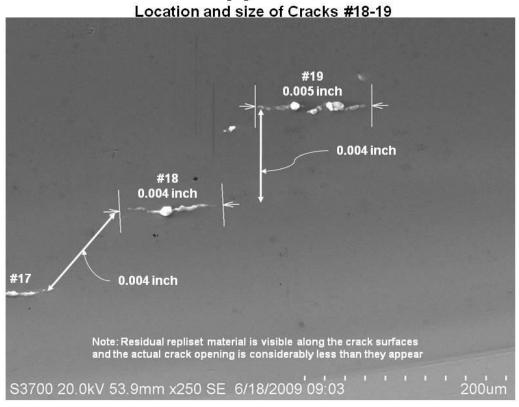


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Location and size of Crack #15-17







-

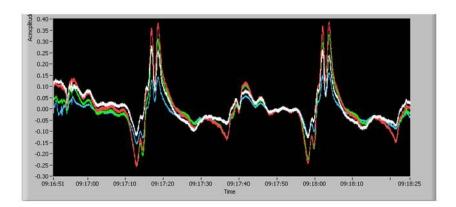
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #20



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Poppet #52
LaRC eddy current findings, the colors indicate ???



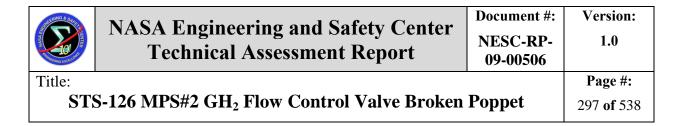
NASA Engineering and Safety Cente		Document #:	Version:
TO SERVICE STATE OF THE SERVIC	Technical Assessment Report		1.0
Title:		Page #:	
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Surface crack sizes and locations

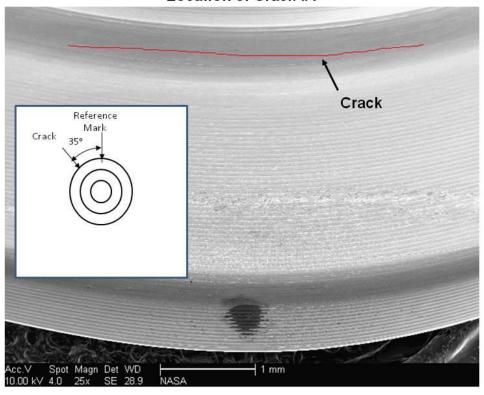
£ 43	Poppet #53	8
Crack Number	Size (inch)	Angle (degrees)
1	0.136	35

Boeing Eddy Current Findings

Poppet #53									
8: 30: 5			Ru	n Data (√pp)	0			
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	1.080	1.093	1.115	1.135	1.139	1.129	1.115	Yes	20
B. Devries	1.080	1.103	1.114	1.113	1.099	1.129	1.106	Yes	25

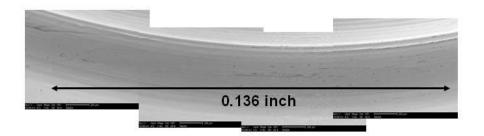


Poppet #53 Location of Crack #1



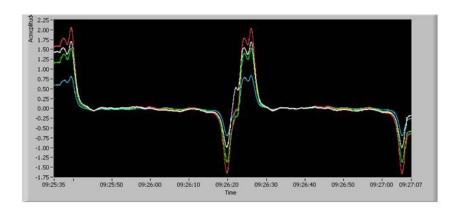
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #53 Size of Crack #1



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Poppet #53
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

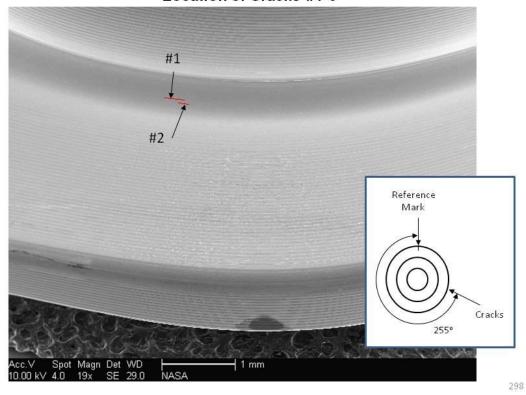
	Poppet #54	
Crack Number	Size (inch)	Angle (degrees)
1	0.017	255
2	0.007	255

Boeing Eddy Current Findings

Poppet #54									
		200	Ru	ın Data (Vpp)	10	.,,		
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.212	0.211	0.209	0.212	0.212	0.216	0.212	Yes	250
B. Devries	0.217	0.216	0.226	0.216	0.221	0.219	0.219	Yes	255

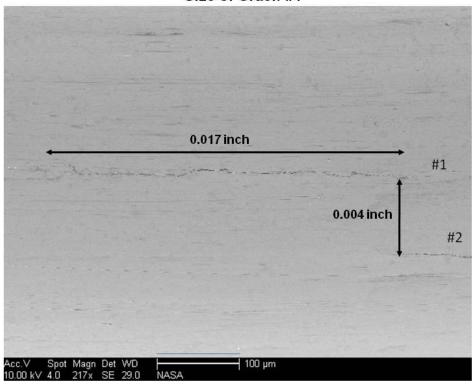
THING A SALES			Version: 1.0		
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

Poppet #54 Location of Cracks #1-6



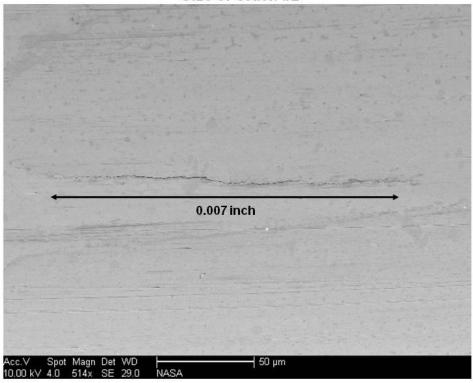
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

Poppet #54 Size of Crack #1



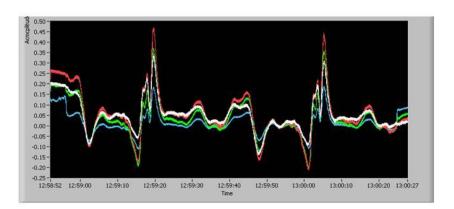
SHEERING & SAFEL	NASA Engineering and Safety Center	Document #:	Version:		
TO THE STATE OF TH	Technical Assessment Report	NESC-RP- 09-00506	1.0		
Title:			Page #:		
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Poppet #54 Size of Crack #2



THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title:					

LaRC eddy current findings, the colors indicate ???



THE RESIDENCE OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Surface crack sizes and locations

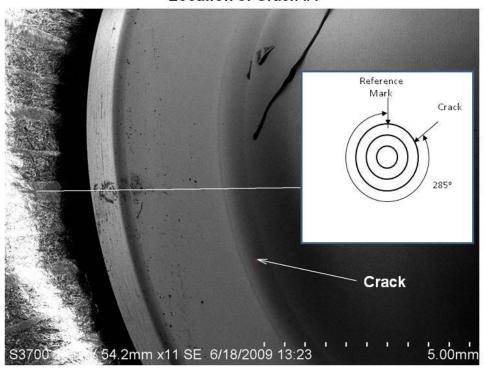
* **	Poppet #55	8
Crack Number	Size (inch)	Angle (degrees)
1	0.004	285

Boeing Eddy Current Findings

3					F	oppet #	55		
			Ru	ın Data (Vpp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.044	0.039	0.035	0.041	0.040	0.044	0.041	No	285
B. Devries	0.036	0.035	0.033	0.038	0.035	0.031	0.035	No	285

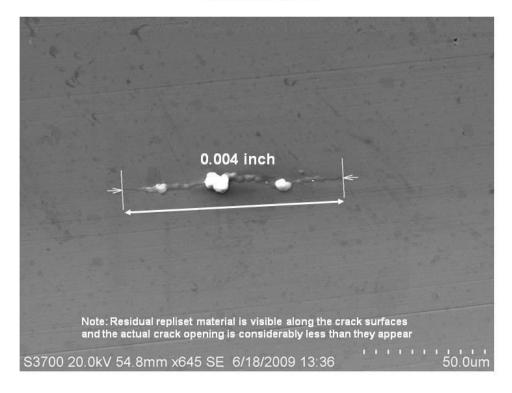
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Poppet #55 Location of Crack #1



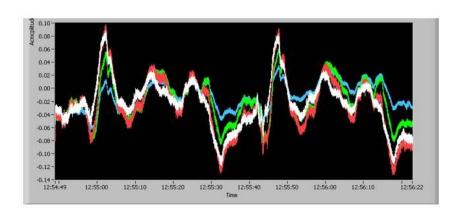
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title:			Page #:		
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					

Poppet #55 Size of Crack #1



SHERING & SALE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
THE STATE OF THE S		NESC-RP- 09-00506	1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #55
LaRC eddy current findings, the colors indicate ???



SHERING & SAFE	NASA Engineering and Safety Center Technical Assessment Report	Document #:	Version:
TO TOWN		NESC-RP- 09-00506	1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Surface crack sizes and locations

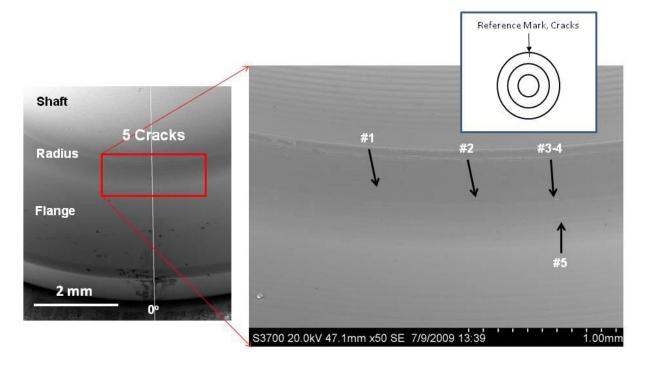
Poppet #56					
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)
1	0.002	0	11	0.003	180
2	0.003	0	12	0.002	180
3	0.004	0	13	0.002	180
4	0.005	0	14	0.002	180
5	0.006	0	15	0.002	180
6	0.003	180	16	0.004	180
7	0.002	180	17	0.004	180
8	0.002	180	18	0.002	180
9	0.004	180	19	0.003	180
10	0.045	180			4-360

Boeing Eddy Current Findings

None Provided

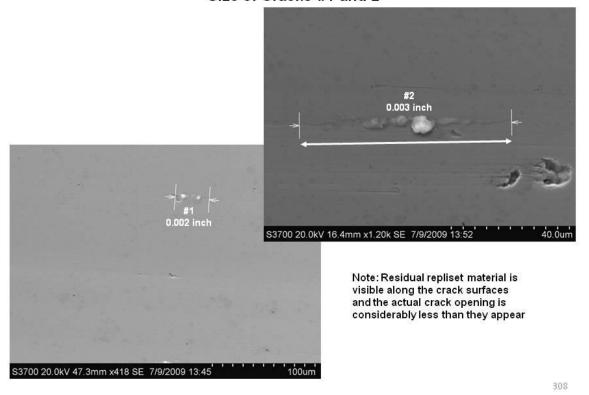
STORY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			310 of 538

Poppet #56 Location of Cracks #1-5



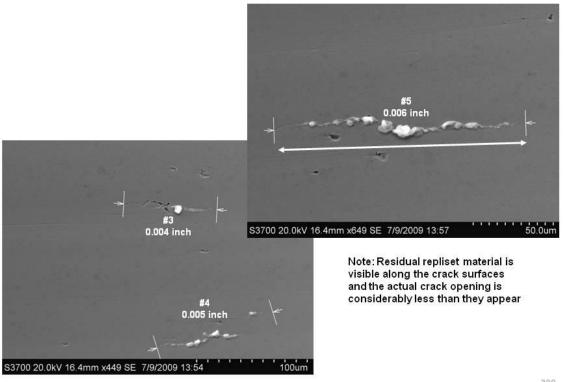
TO THE TENT OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken		Page #: 311 of 538

Poppet #56 Size of Cracks #1 and 2

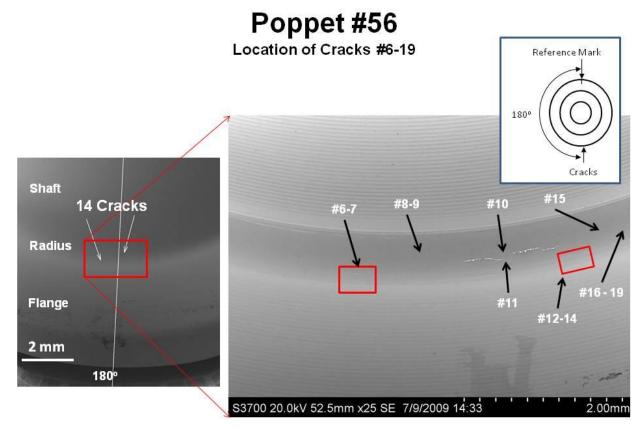


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			Page #: 312 of 538

Poppet #56 Size of Cracks #3-5



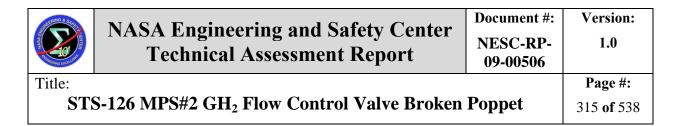
S STATE OF THE STA	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			



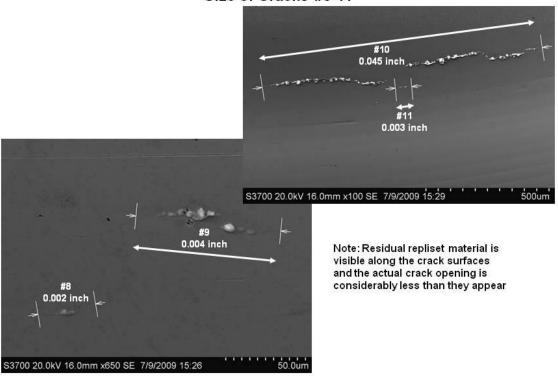
A STATE OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			Page #: 314 of 538
515 120 MISH2 GILLION CONTROL VALVE BIONEH I OPPER			314 01 330

Size of Cracks #6 and 7



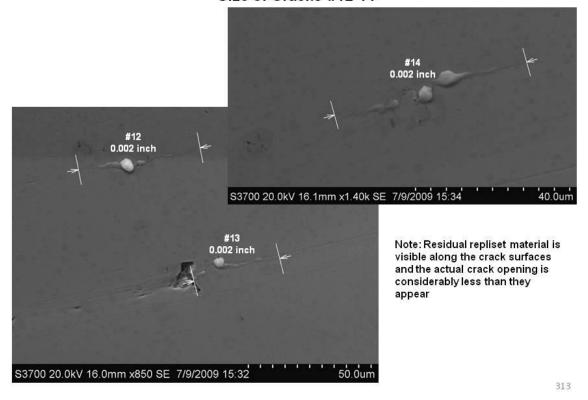


Size of Cracks #8-11



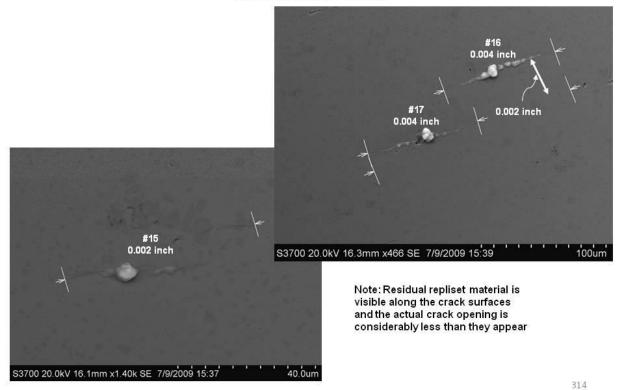
TO THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			Page #: 316 of 538

Poppet #56 Size of Cracks #12-14



THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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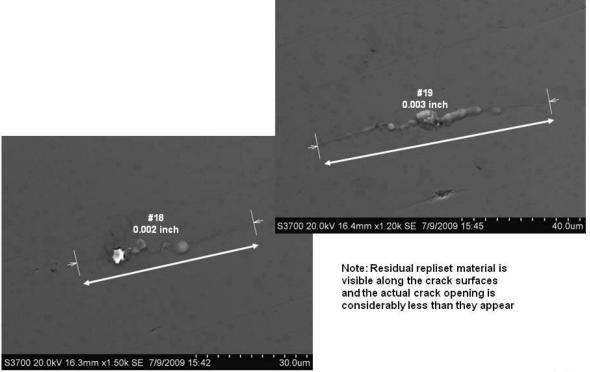
Poppet #56 Size of Cracks #15-17



NESC Request No.: 09-00506

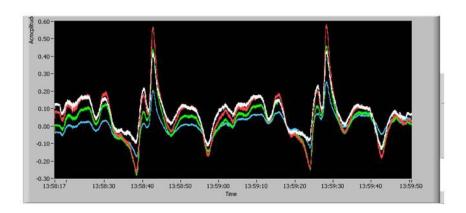
STATE OF THE PERSON OF THE PER	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP-	Version: 1.0
Title:	recimient rissessment report	09-00506	Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #56 Size of Cracks #18 and 19



THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #56
LaRC eddy current findings, the colors indicate ???



TO SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP-	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	09-00506 Poppet	Page #: 320 of 538

Surface crack sizes and locations

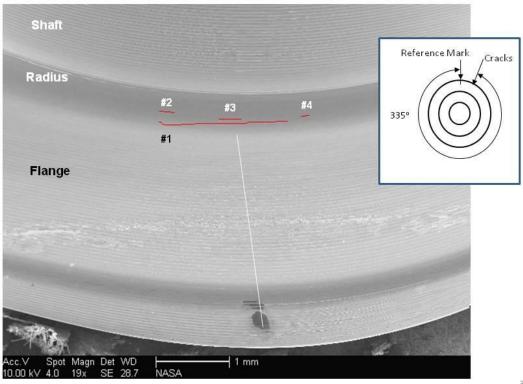
Poppet#57						
Crack Number	Size (inch)	Angle (degrees) 335				
1	0.079					
2	0.005	335				
3	0.014	335				
4	0.005	335				
5	0.007	155				
6	0.002	155				
7	0.005	155				

Boeing Eddy Current Findings

Poppet #57									
Inspector	Run Data (Vpp)								
	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.498	0.488	0.480	0.478	0.486	0.480	0.485	Yes	335
J. Engel	0.170	0.160	0.169	0.165	0.174	0.171	0.168	Yes	160
B. Devries	0.487	0.491	0.488	0.494	0.447	0.494	0.484	Yes	330
B. Devries	0.170	0.167	0.168	0.165	0.149	0.157	0.163	No	160

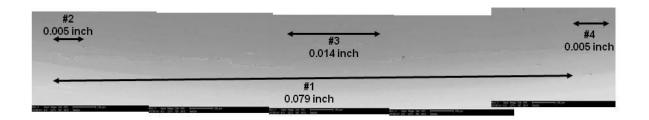
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #57 Location of Cracks #1-4



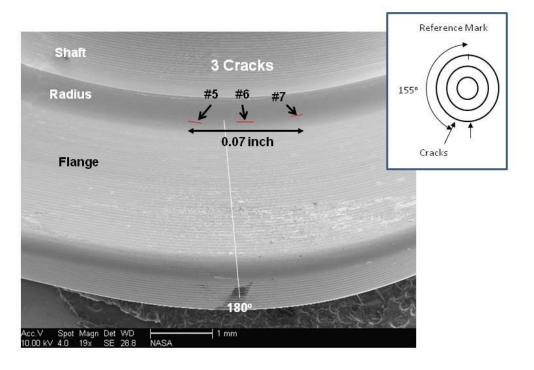
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #57 Size of Cracks #1-4



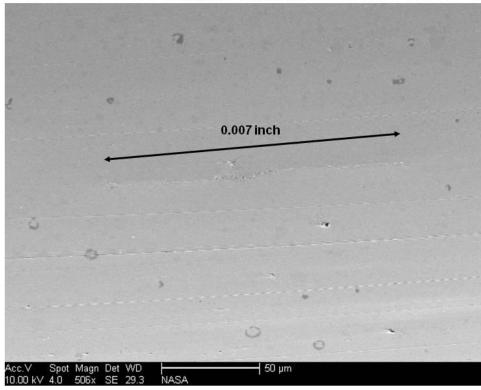
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		Page #: 323 of 538	

Poppet #57 Location of Cracks #5-7



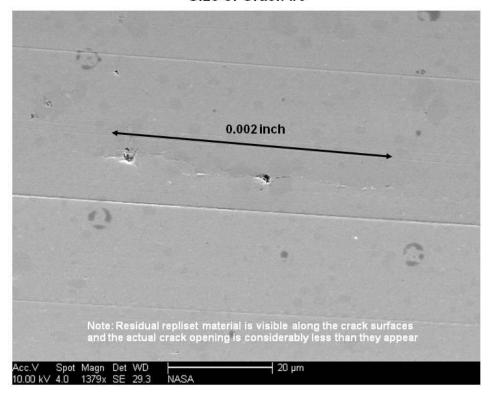
THE RESERVE TO SERVE THE PARTY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #57 Size of Crack #5



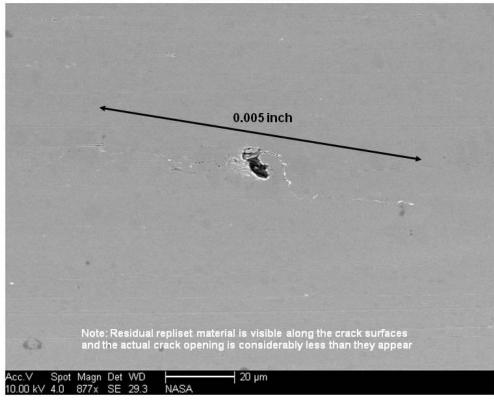
NASA Engineering and Safety	NASA Engineering and Safety Center	Document #:	Version:
TO THE STATE OF TH	NASA Engineering and Safety Center Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #57 Size of Crack #6



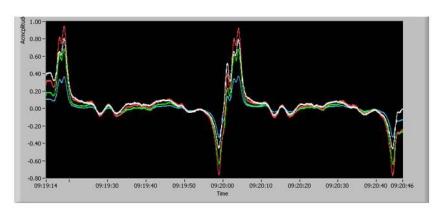
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #57 Size of Crack #7



THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		Page #: 327 of 538	

Poppet #57
LaRC eddy current findings, the colors indicate ???



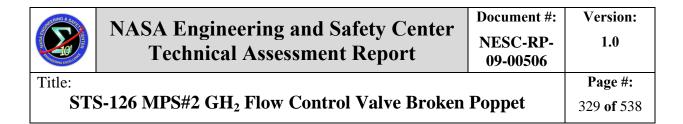
THING & CALLED	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 328 of 538

Surface crack sizes and locations

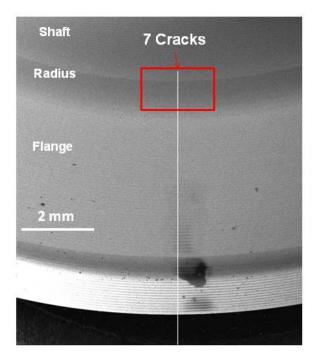
	Poppet #58					
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)	
1	0.005	0	12	0.002	180	
2	0.003	0	13	0.003	180	
3	0.004	0	14	0.002	180	
4	0.005	0	15	0.009	180	
5	0.008	0	16	0.015	180	
6	0.002	0	17	0.003	180	
7	0.002	0	18	0.014	180	
8	0.005	180	19	0.003	180	
9	0.005	180	20	0.002	180	
10	0.004	180	21	0.002	180	
11	0.002	180				

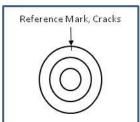
Boeing Eddy Current Findings

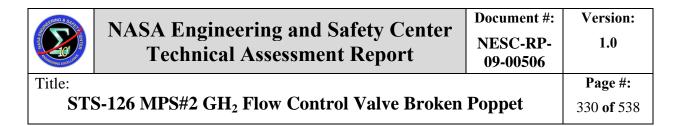
None Provided



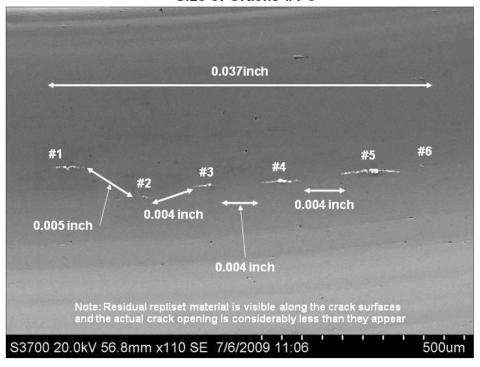
Location of Cracks #1-7





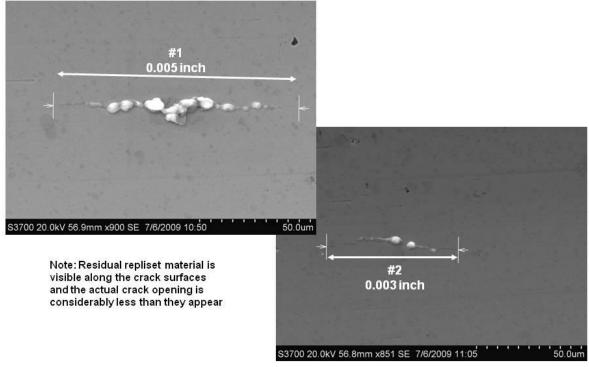


Size of Cracks #1-5



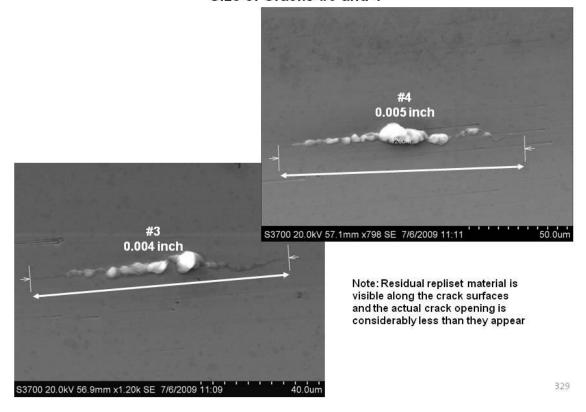
NASA Engineering and Safety Center	Document #:	Version:	
TO LEASE STATE OF THE STATE OF	Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #58 Size of Cracks #1 and 2



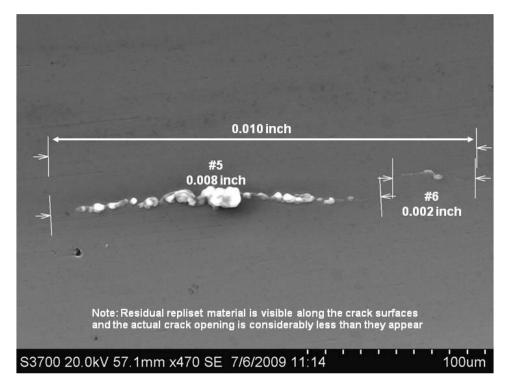
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			Page #:
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Poppet #58 Size of Cracks #3 and 4



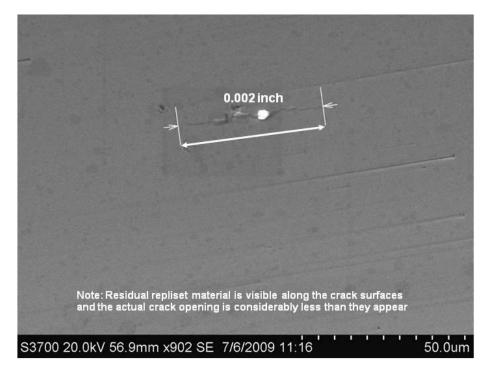
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			333 of 538

Size of Cracks #5 and 6



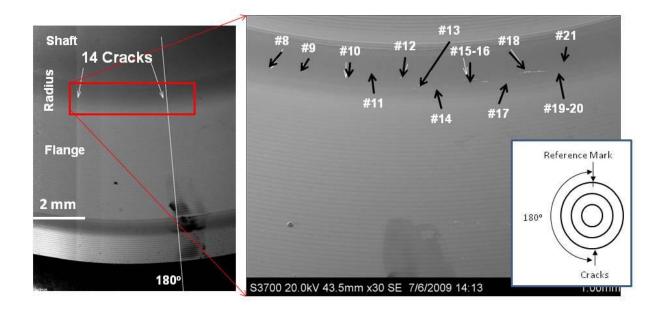
THIRD & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 334 of 538

Poppet #58 Size of Crack #7



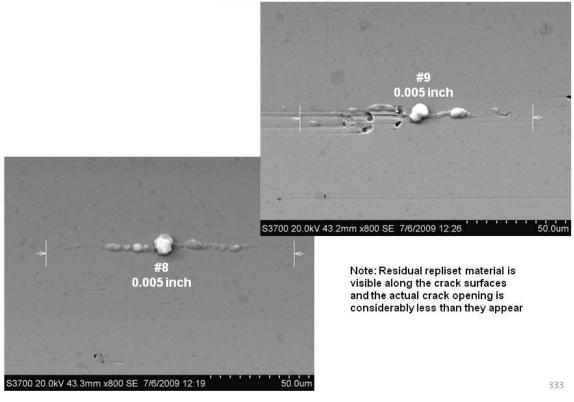
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:		_	Page #:
STS	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	335 of 538

Location of Cracks #8-21



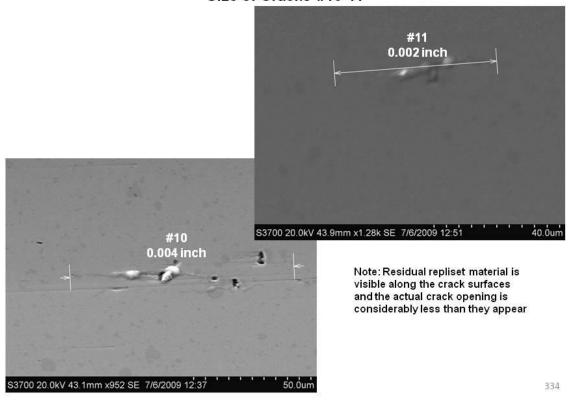
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:			Page #:
ST	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	336 of 538

Poppet #58 Size of Cracks #8-9



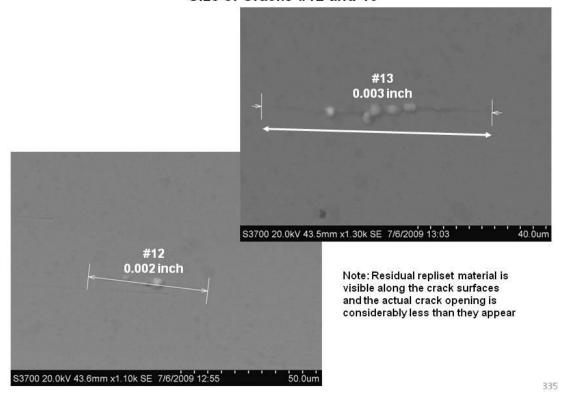
STORY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:			Page #:
STS	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	337 of 538

Poppet #58 Size of Cracks #10-11



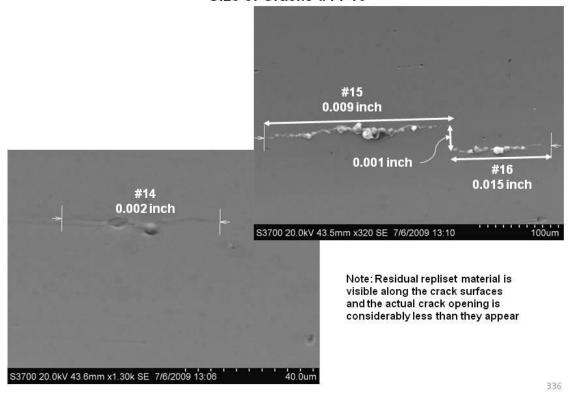
TO THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 338 of 538

Size of Cracks #12 and 13



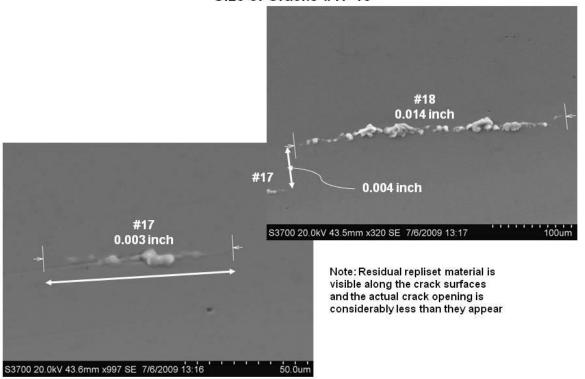
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Cracks #14-16



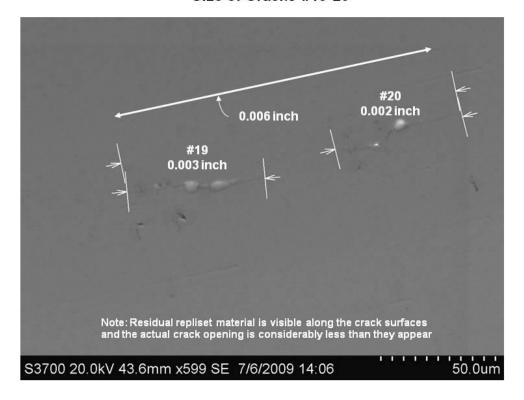
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Cracks #17-18



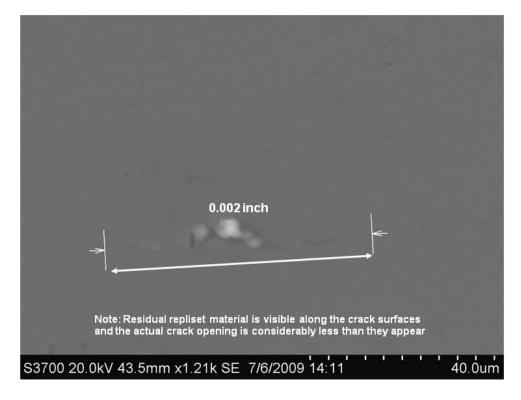
SHERING & SAFE	NASA Engineering and Safety Center	and Safaty Contar Document #: Vers	Version:
TO THE REPORT OF THE PARTY OF T	Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
STS	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	341 of 538

Size of Cracks #19-20



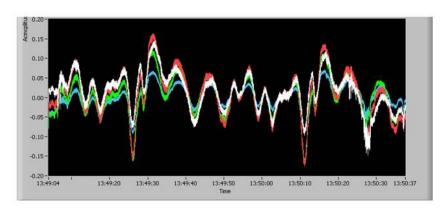
SHEERING & SAFE	NASA Engineering and Safety Center	Document #: Version:	#: Version:
STORY TO SECULAR	Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
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Size of Crack #21



TO THE PART OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 343 of 538

Poppet #58
LaRC eddy current findings, the colors indicate ???



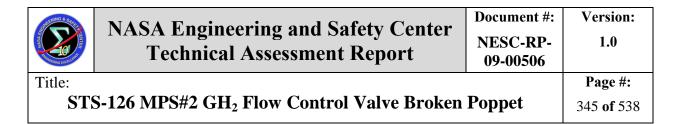
NG A STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	S-126 MPS#2 GH ₂ Flow Control Valve Broken	Poppet	Page #: 344 of 538

Surface crack sizes and locations

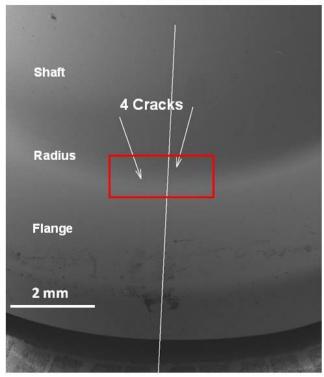
Poppet #59		
Crack Number	Size (inch)	Angle (degrees)
1	0.008	0
2	0.002	0
3	0.002	0
4	0.010	0
5	0.002	180
6	0.002	180
7	0.003	180
8	0.004	180
9	0.001	180

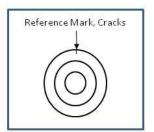
Boeing Eddy Current Findings

None Provided



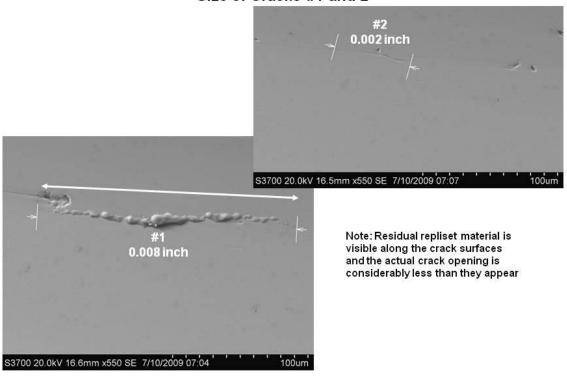
Location of Cracks #1-4





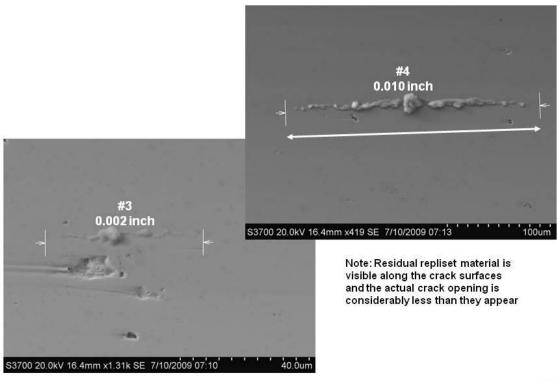
THING A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			Page #: 346 of 538

Size of Cracks #1 and 2



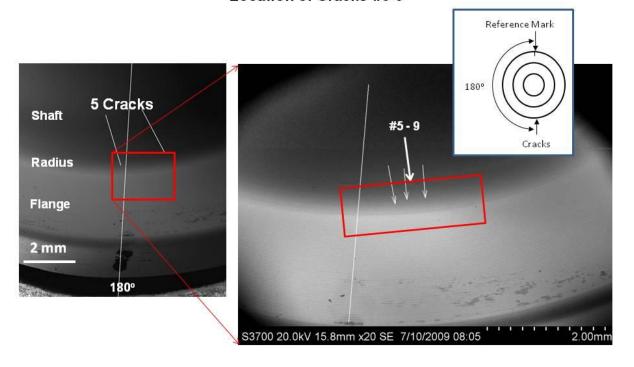
TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #59 Size of Cracks #3 and 4



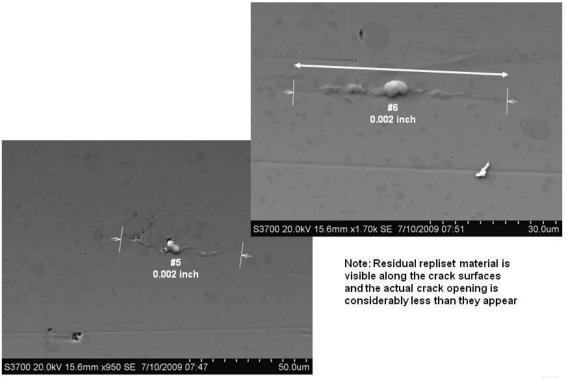
TO STORY OF THE PARTY OF THE PA	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #59 Location of Cracks #5-9



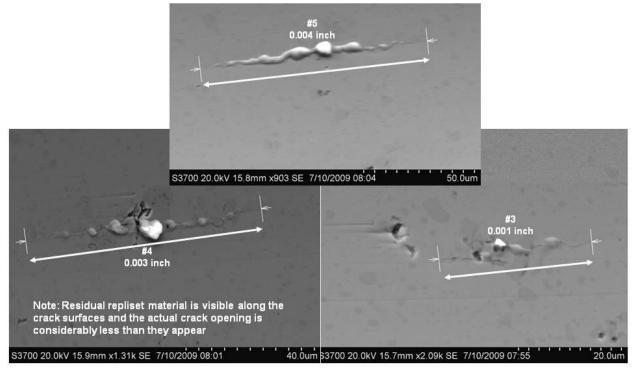
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		Page #: 349 of 538

Poppet #59 Size of Cracks #5-6



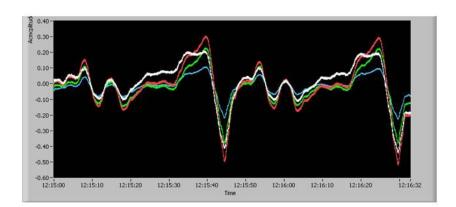
	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #59 Size of Cracks #7-9



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Poppet #59
LaRC eddy current findings, the colors indicate ???



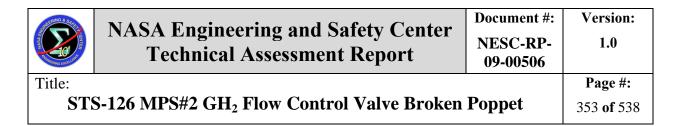
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SV SWILL TO SECULAR SECONDARY	Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

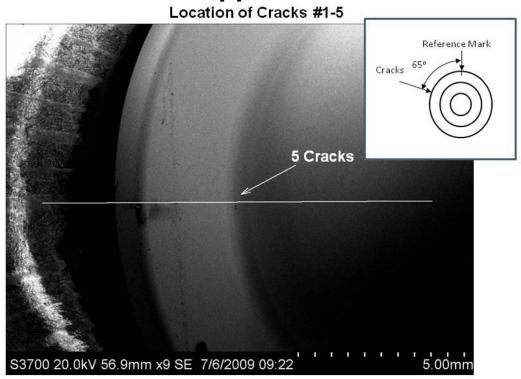
Surface crack sizes and locations

Poppet #60			
Crack Number	Size (inch)	Angle (degrees)	
1	0.038	65	
2	0.003	65	
3	0.005	65	
4	0.003	65	
5	0.004	65	

Boeing Eddy Current Findings

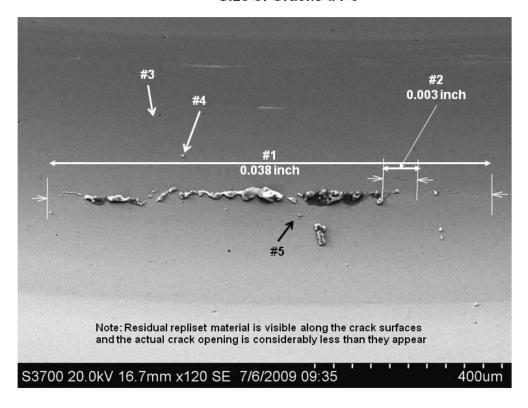
Poppet #60									
93 100 51			Ru	n Data (√pp)	110000			
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.589	0.591	0.613	0.620	0.622	0.622	0.610	Yes	80
3. Devries	0.614	0.611	0.609	0.607	0.607	0.615	0.611	Yes	75





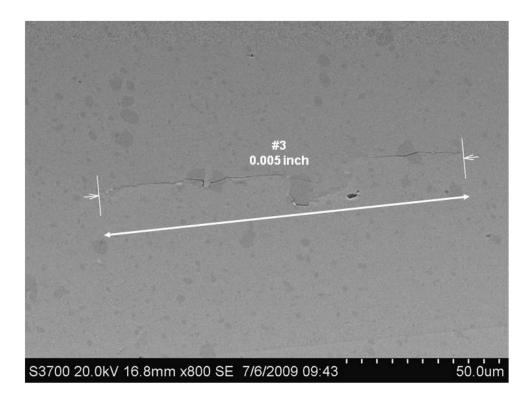
THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #60 Size of Cracks #1-5



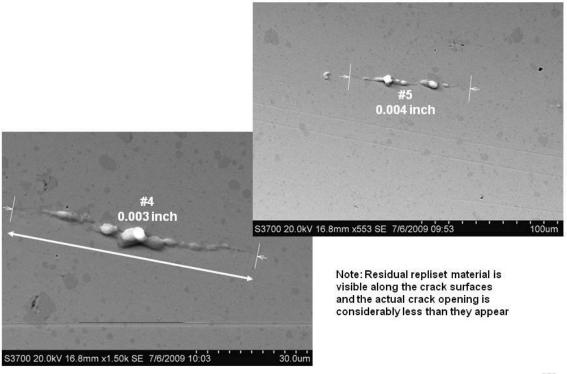
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #60 Size of Crack #3



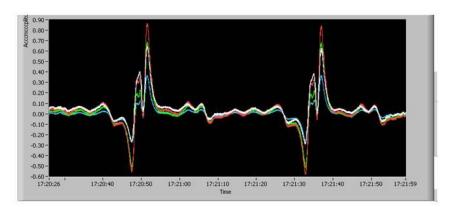
TO THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP-	Version: 1.0
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Poppet #60 Size of Cracks #4 and 5



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Poppet #60
LaRC eddy current findings, the colors indicate ???



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TO LEASE STATE OF THE STATE OF		NESC-RP- 09-00506	1.0	
Title:			Page #:	
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

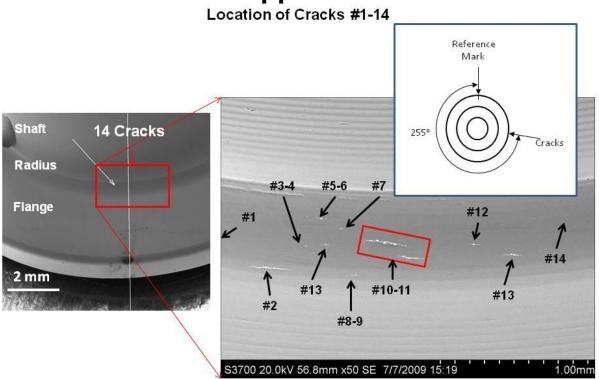
Surface crack sizes and locations

	Poppet #61						
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)		
1	0.003	255	15	0.003	75		
2	0.011	255	16	0.006	75		
3	0.002	255	17	0.002	75		
4	0.003	255	18	0.003	75		
5	0.002	255	19	0.003	75		
6	0.004	255	20	0.006	75		
7	0.002	255	21	0.003	75		
8	0.001	255	22	0.006	75		
9	0.003	255	23	0.004	75		
10	0.012	255	24	0.004	75		
11	0.007	255	25	0.013	75		
12	0.004	255	26	0.005	75		
13	0.005	255	27	0.006	75		
14	0.002	255	28	0.004	75		

Boeing Eddy Current Findings

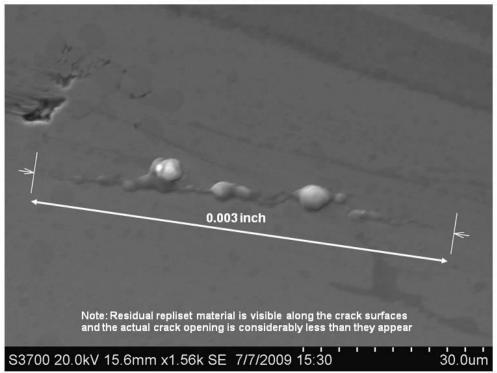
					F	oppet #	61	,	10
		ço.	Ru	ın Data (Vpp)		gc -		33
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.188	0.198	0.190	0.188	0.194	0.200	0.193	Yes	75
J. Engel	0.168	0.176	0.176	0.170	0.185	0.183	0.176	Yes	255
B. Devries	0.196	0.199	0.203	0.207	0.204	0.206	0.203	Yes	80
B. Devries	0.174	0.175	0.177	0.175	0.181	0.178	0.177	Yes	260

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Title:	S 126 MDS#2 CH Flow Control Volvo Prokon	Donnet	Page #: 359 of 538
517	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		



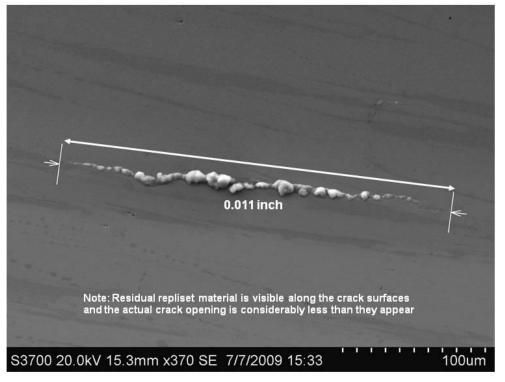
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		Page #: 360 of 538	

Poppet #61 Size of Crack #1



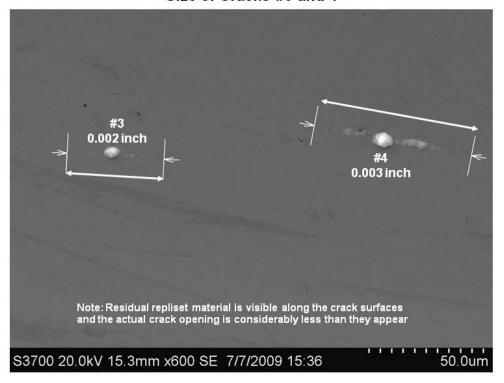
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Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #61 Size of Crack #2



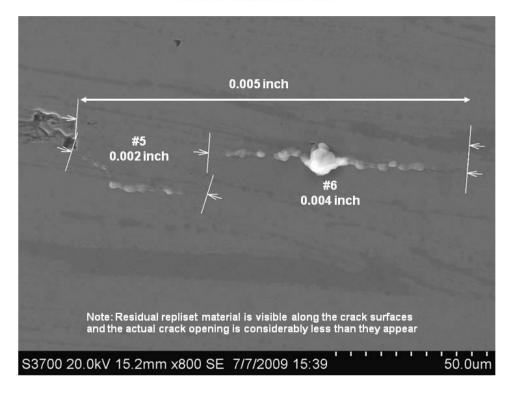
NG & COLUMN AND ADDRESS OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #61 Size of Cracks #3 and 4



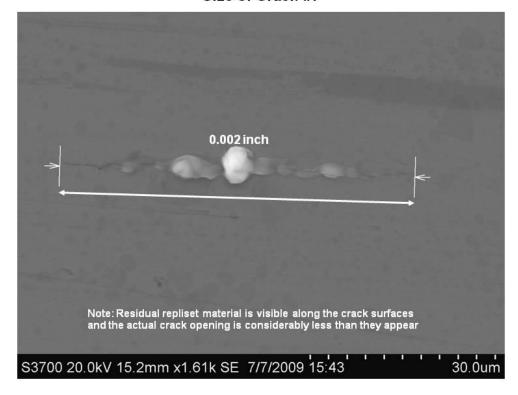
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Title:			Page #:		
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Poppet #61 Size of Cracks #5 and 6



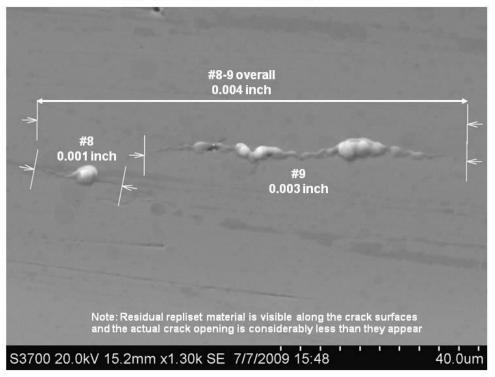
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Title:	E 126 MDC#2 CIL Flory Control Volvo Prokon	Donnat	Page #: 364 of 538
513	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #61 Size of Crack #7



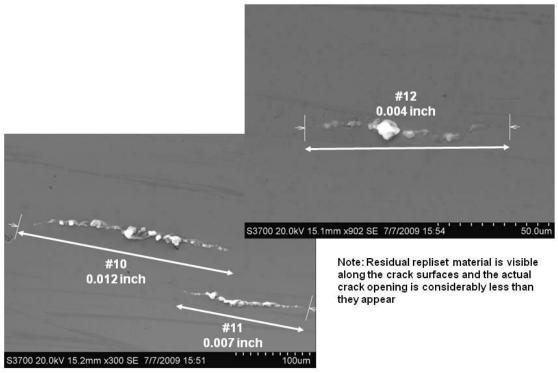
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Location and size of Cracks #8 and 9



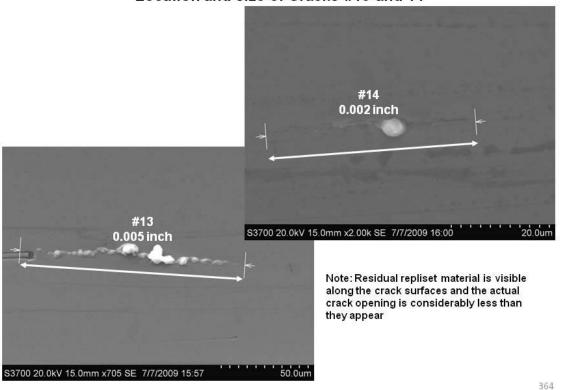
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Location and size of Cracks #10-12



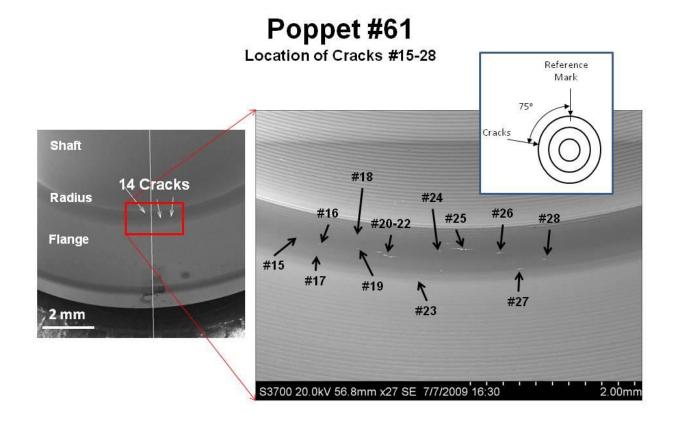
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Poppet #61 Location and size of Cracks #13 and 14



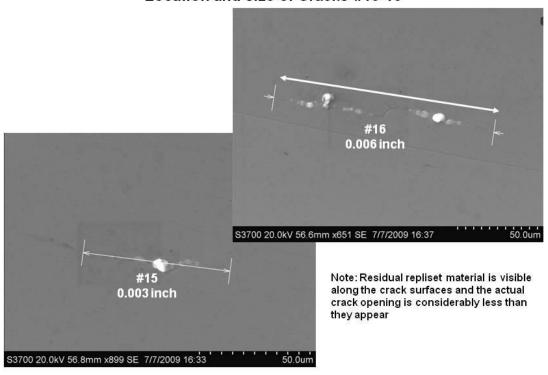
NESC Request No.: 09-00506

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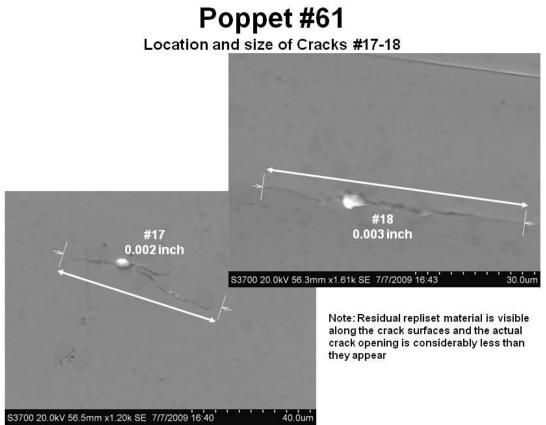


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #15-16

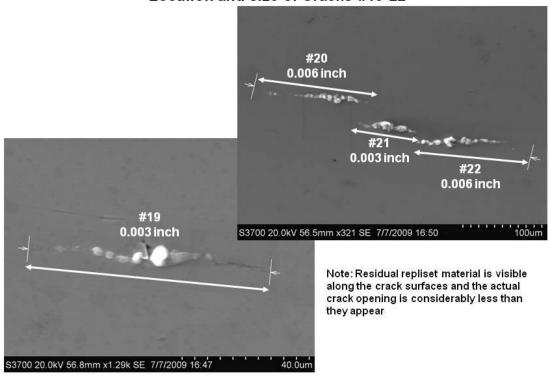


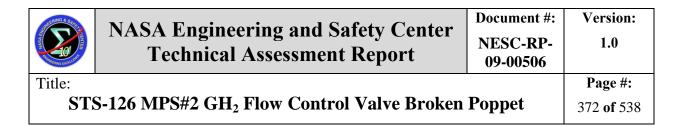
THE RESERVE TO SERVE THE PARTY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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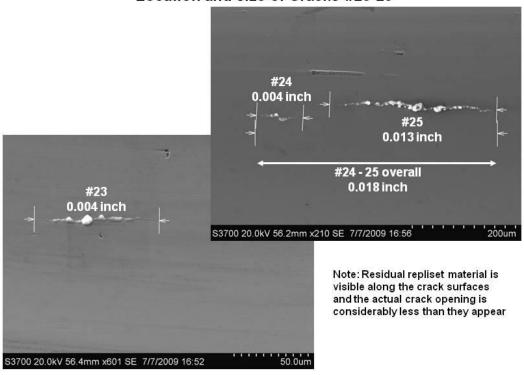
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Location and size of Cracks #19-22



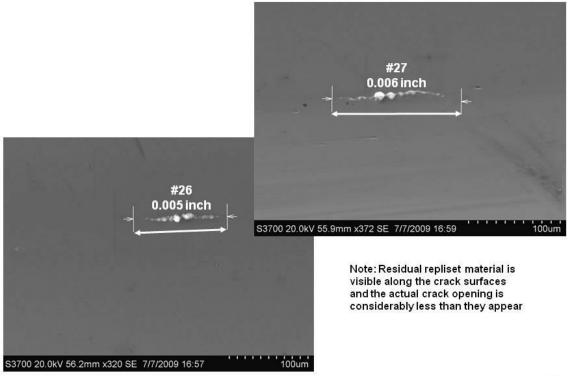


Location and size of Cracks #23-25



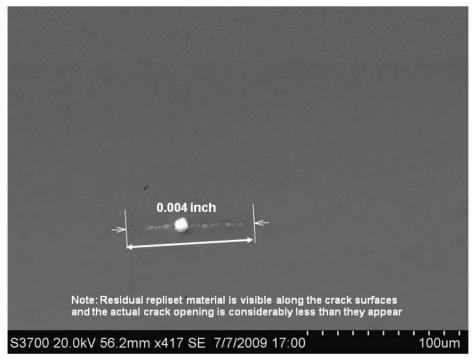
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Poppet #61 Location and size of Cracks #26-27



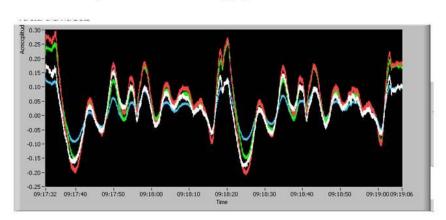
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #61 Size of Crack #28



THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #61
LaRC eddy current findings, the colors indicate ???



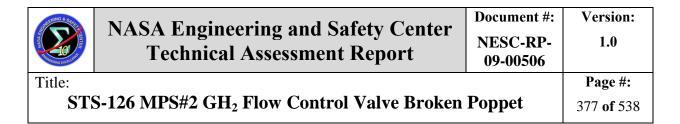
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Surface crack sizes and locations

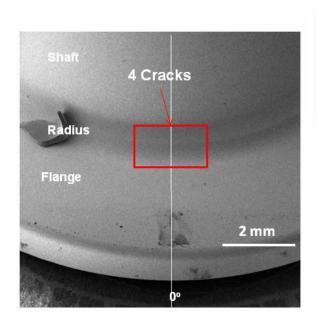
Poppet #62					
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)
1	0.009	0	11	0.008	180
2	0.006	0	12	0.002	180
3	0.008	0	13	0.003	180
4	0.006	0	14	0.005	180
5	0.004	180	15	0.004	180
6	0.004	180	16	0.002	180
7	0.002	180	17	0.002	180
8	0.003	180	18	0.004	180
9	0.008	180	19	0.002	180
10	0.004	180	20	0.004	180

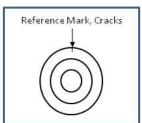
Boeing Eddy Current Findings

None Provided



Location of Cracks #1-4

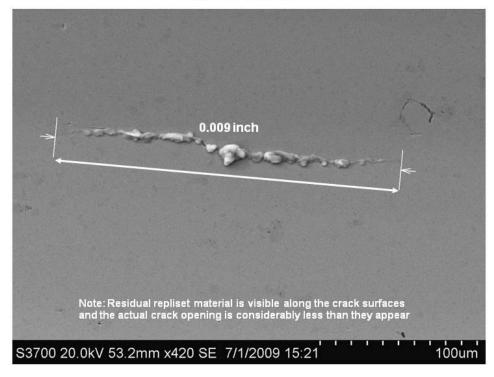




37/

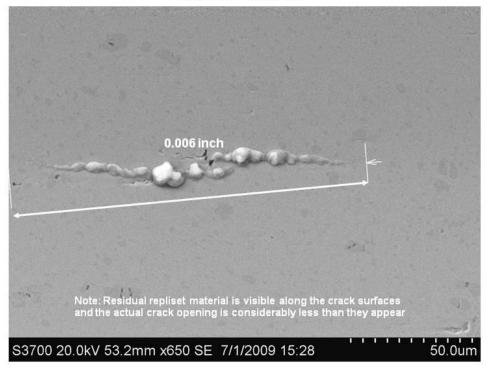
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Strangering executed	Technical Assessment Report	NESC-RP- 09-00506	1.0
Title:			Page #:
STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Poppet #62 Size of Crack #1



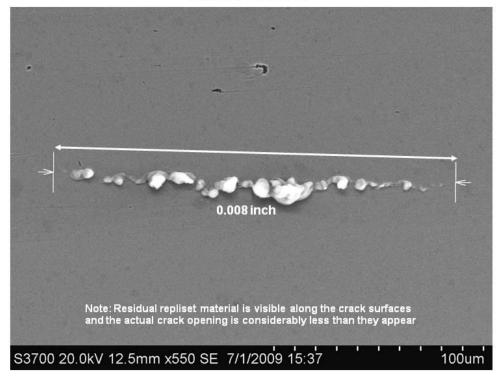
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #62 Size of Crack #2



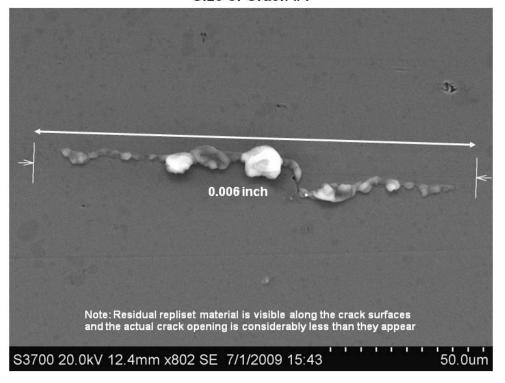
THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #3

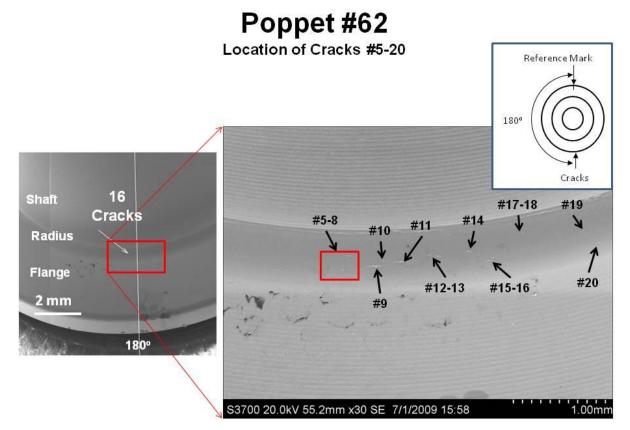


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #62 Size of Crack #4

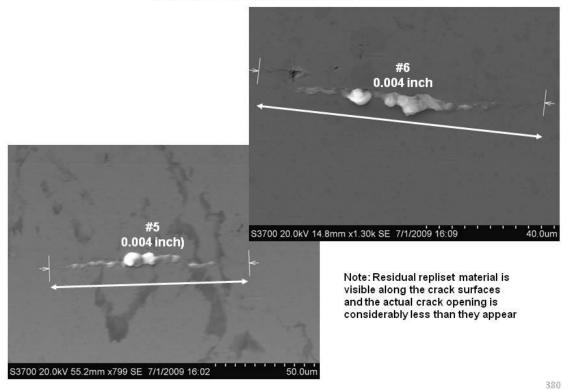


STORY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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THE A STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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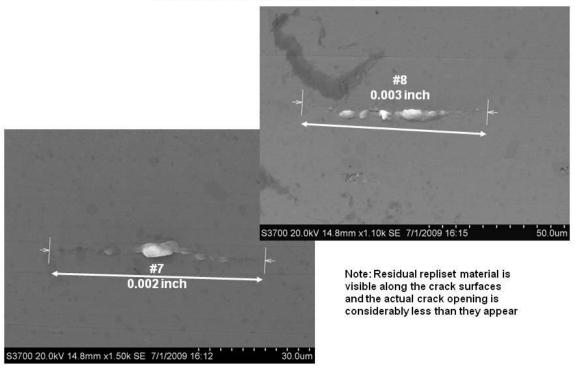
Poppet #62 Location and size of Cracks #5 and 6



NESC Request No.: 09-00506

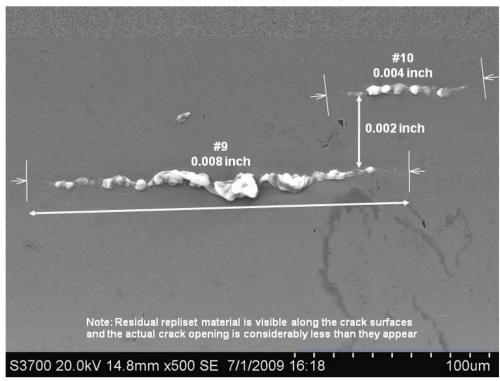
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #62 Location and size of Cracks #7 and 8



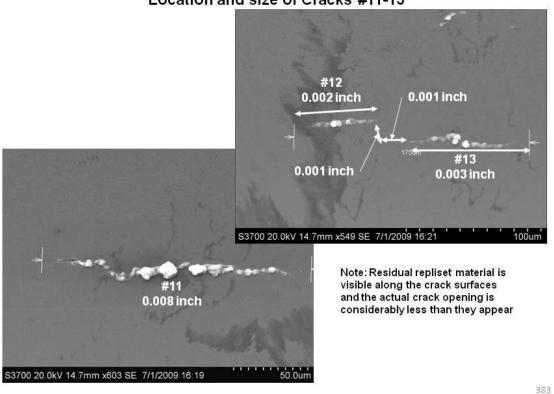
	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #9 and 10



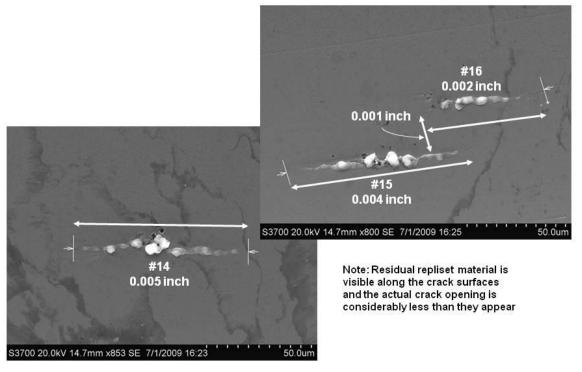
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #11-13



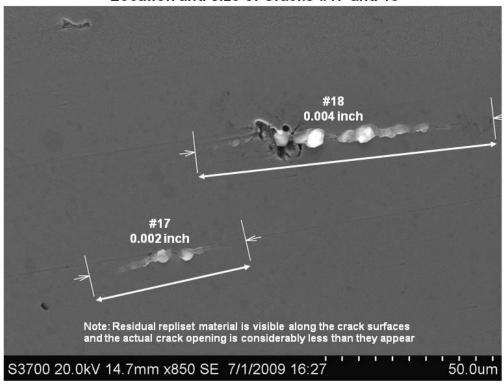
THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #14-16



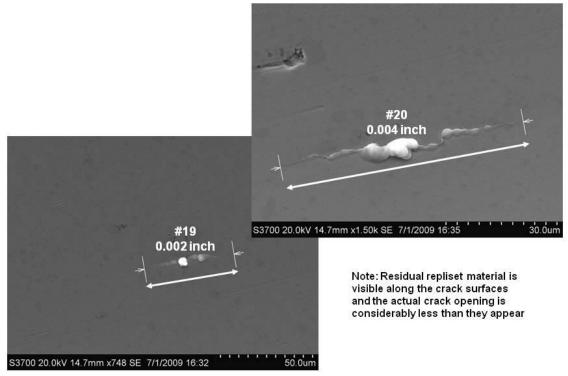
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		Page #: 388 of 538	

Poppet #62 Location and size of Cracks #17 and 18



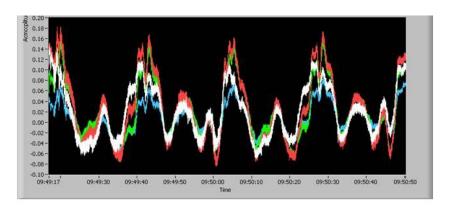
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Cracks #19 and 20



THE A COUNTY OF THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #62
LaRC eddy current findings, the colors indicate ???



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Surface crack sizes and locations

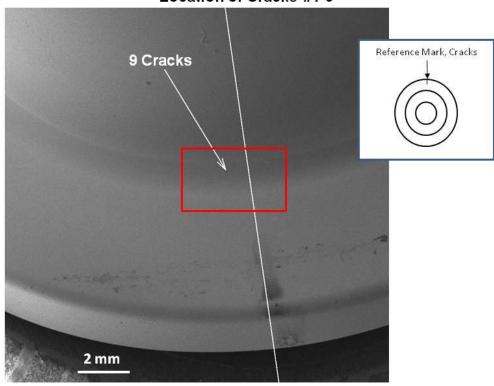
Poppet #63							
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)		
1	0.005	0	11	0.006	180		
2	0.002	0	12	0.012	180		
3	0.015	0	13	0.002	180		
4	0.002	0	14	0.002	180		
5	0.004	0	15	0.007	180		
6	0.004	0	16	0.010	180		
7	0.006	0	17	0.006	180		
8	0.004	0	18	0.015	180		
9	0.002	0	19	0.002	180		
10	0.002	180	20	0.002	180		

Boeing Eddy Current Findings

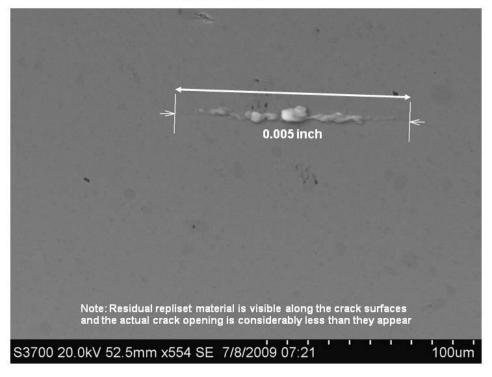
None Provided

TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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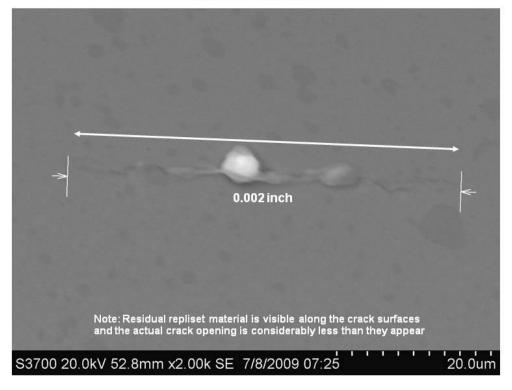
Poppet #63 Location of Cracks #1-9



NASA Engineering and Safety Center Technical Assessment Report	NASA Engineering and Safety Center	Document #:	Version:
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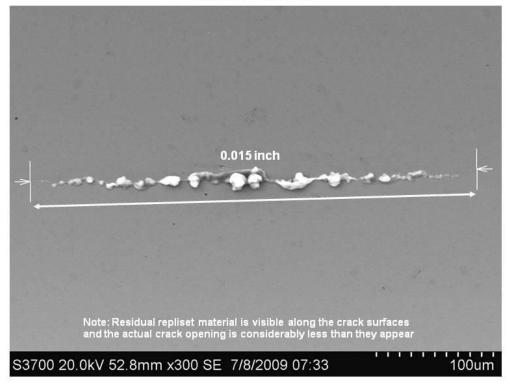


TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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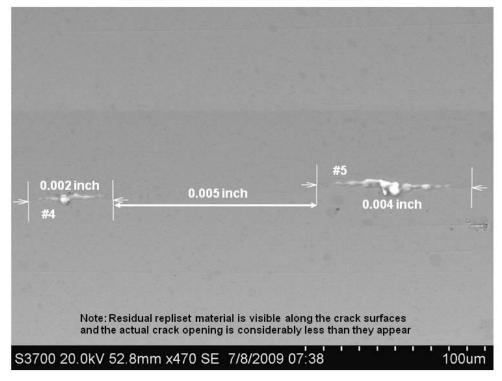
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #3

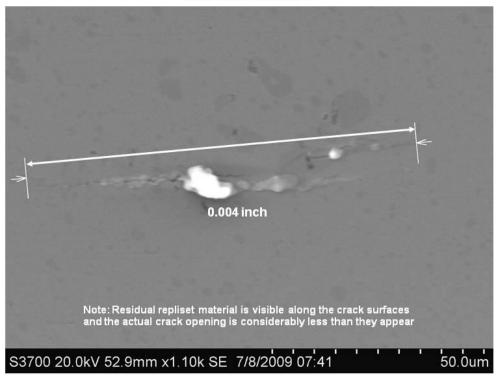


SHERING & SAFE	NASA Engineering and Safety Center	Document #:	Version:
Technical Assessment Report	NESC-RP- 09-00506	1.0	
Title:			Page #:
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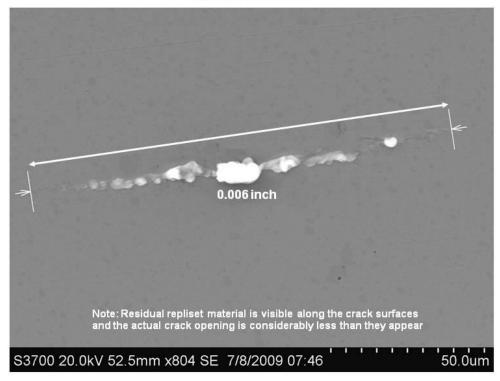
Location and size of Cracks #4 and 5



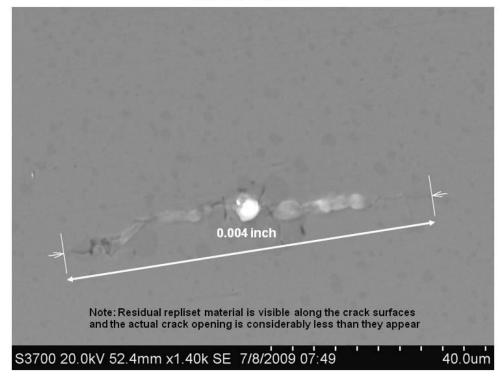
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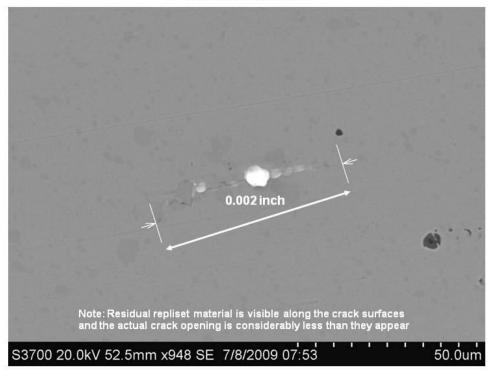
THE PART OF THE PA	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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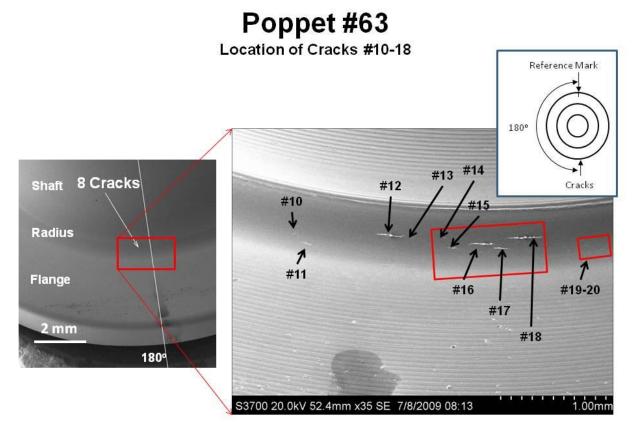
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TO LEASE STATE OF THE STATE OF	NASA Engineering and Safety Center Technical Assessment Report	NESC-RP- 09-00506	1.0
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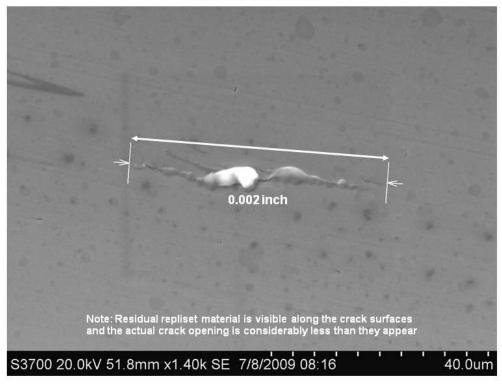
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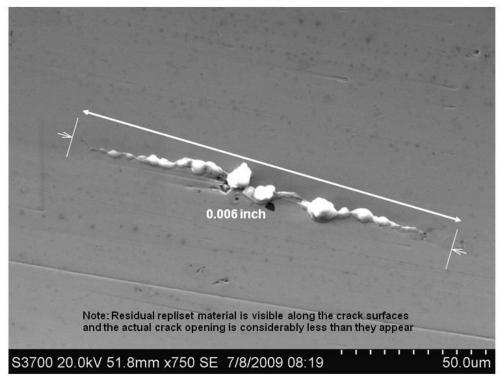
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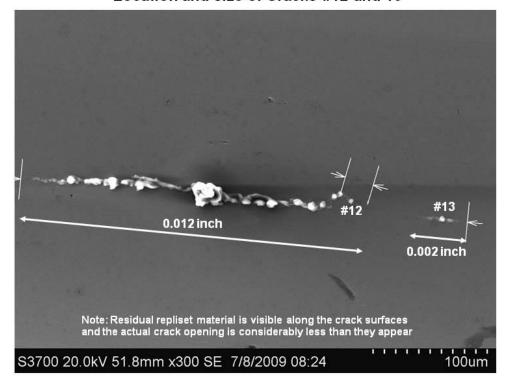
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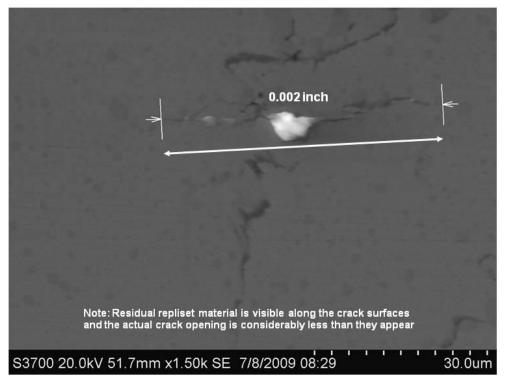
Poppet #63

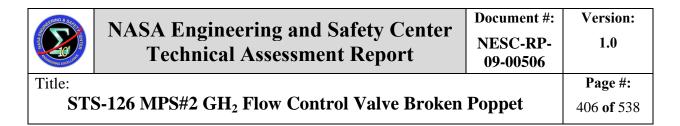
Location and size of Cracks #12 and 13



THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

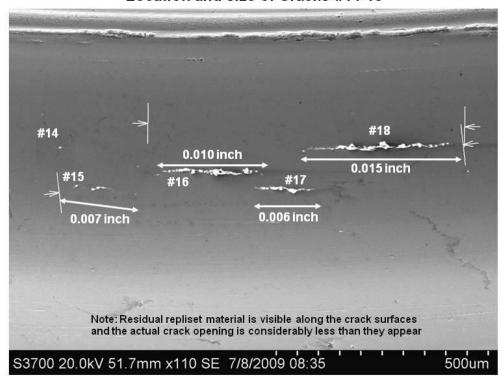
Size of Crack #14





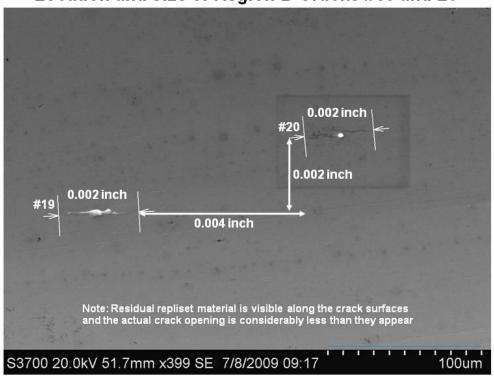
Poppet #63

Location and size of Cracks #14-18



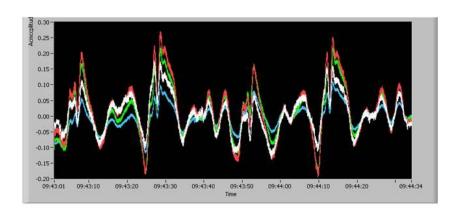
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #63
Location and size of Region B Cracks #19 and 20



THE RESERVE TO SERVE THE PARTY OF THE PARTY	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #63
LaRC eddy current findings, the colors indicate ???



THIRD & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

Surface crack sizes and locations

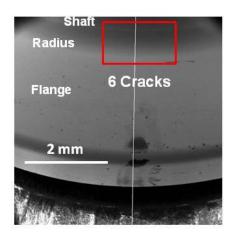
Poppet #64						
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)	
1	0.002	225	11	0.009	45	
2	0.002	225	12	0.009	45	
3	0.004	225	13	0.007	45	
4	0.002	225	14	0.006	45	
5	0.001	225	15	0.009	45	
6	0.002	225	16	0.002	45	
7	0.004	45	17	0.003	45	
8	0.004	45	18	0.003	45	
9	0.002	45	19	0.004	45	
10	0.004	45	20	0.025	45	

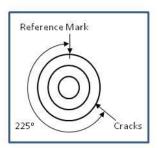
Boeing Eddy Current Findings

Poppet #64									79
			Ru	ın Data (Vpp)				3
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.233	0.237	0.239	0.244	0.240	0.244	0.240	Yes	195
J. Engel	0.066	0.068	na	0.070	0.071	0.072	0.069	Yes	5 (Not3:15/N ratio)
3. Devries	8	R2	8958	î se î	- 5	8	100	No	45
B. Devries	0.242	0.241	0.247	0.246	0.240	0.235	0.242	Yes	210

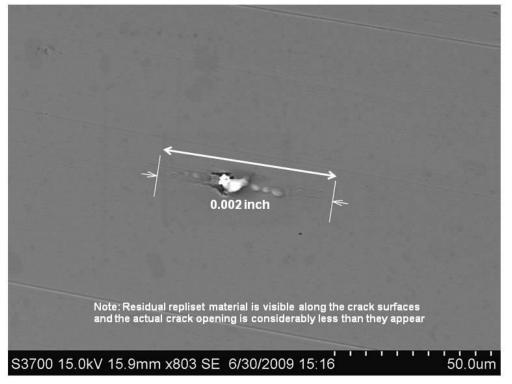
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:			Page #:
STS	STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

Poppet #64 Location of Cracks #1-6

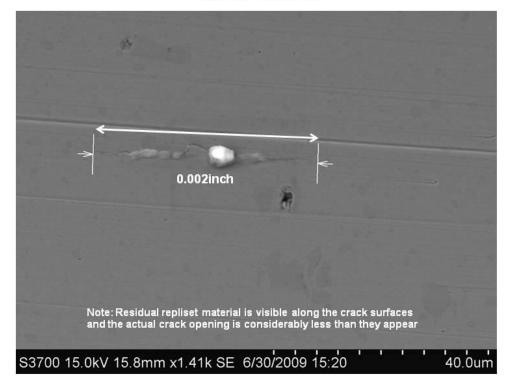




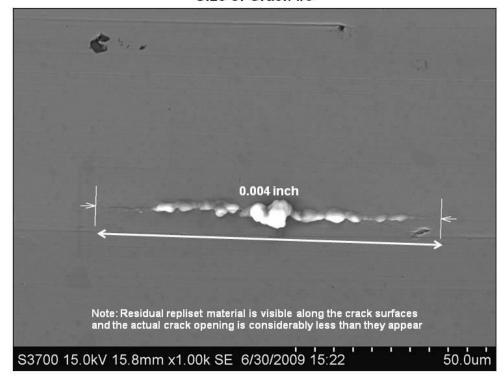
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THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

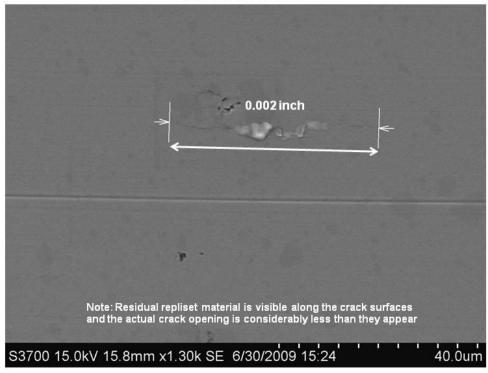


SHERING & SALE	NASA Engineering and Safety Center		Version:
THE STATE OF THE S	Technical Assessment Report	NESC-RP- 09-00506	1.0
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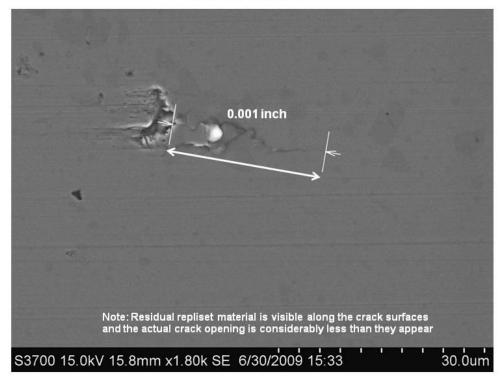


SHEERING & SAFEL	NASA Engineering and Safety Center		Version:
TO THE STATE OF TH	Technical Assessment Report	NESC-RP- 09-00506	1.0
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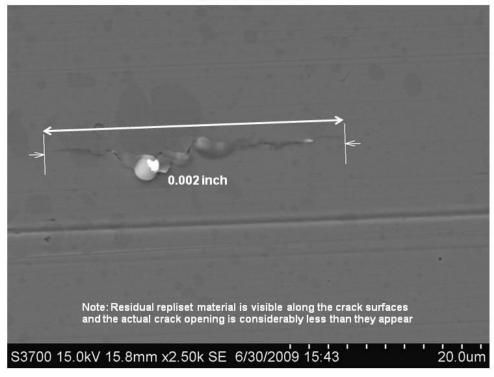
Size of Crack #4

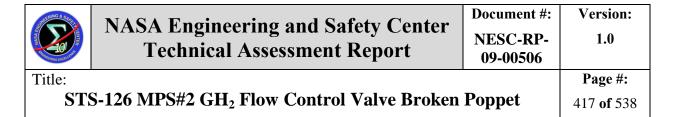


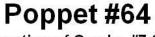
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Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet			

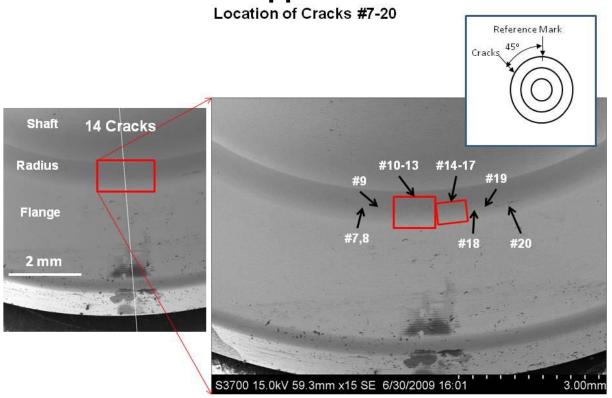


NASA Engineering and Safety Center	Document #:	Version:	
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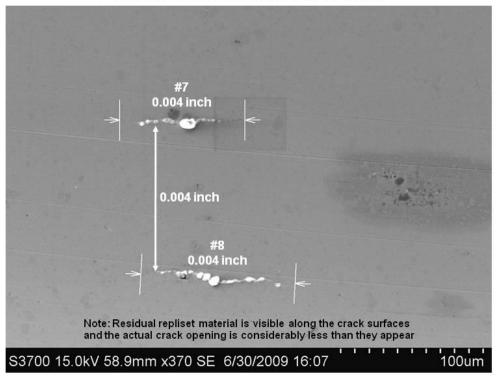




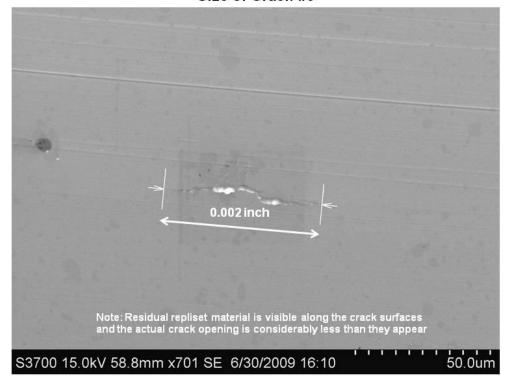


E TO LEAD TO L	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #64 Location and size of Cracks #7 and 8

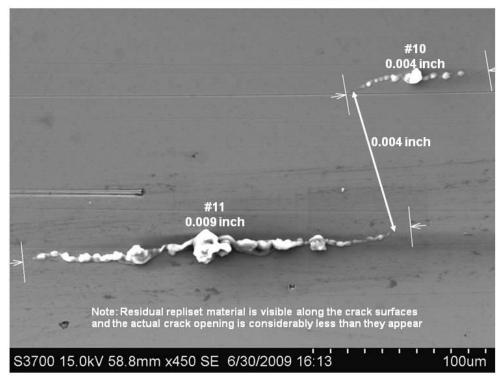


muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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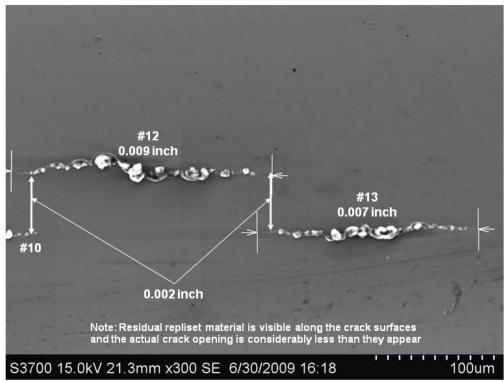
THING & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #10-11



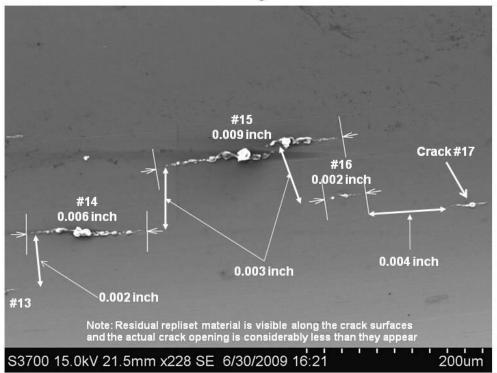
TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #12-13

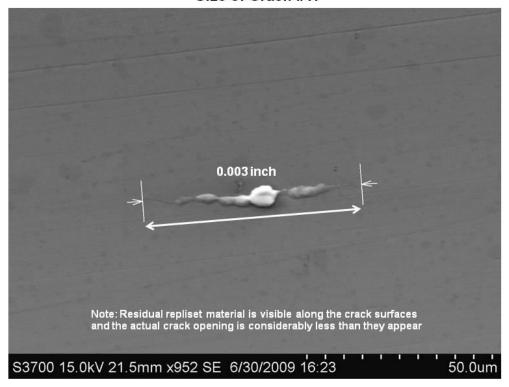


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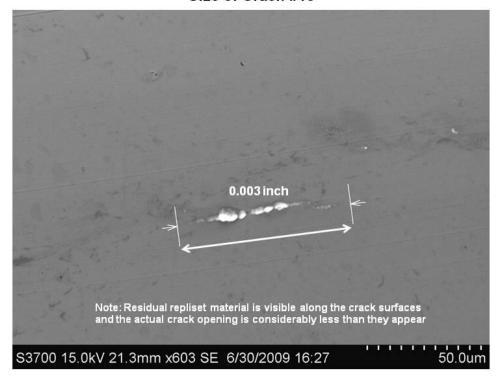
Poppet #64
Location and size of Region B Cracks #14-16



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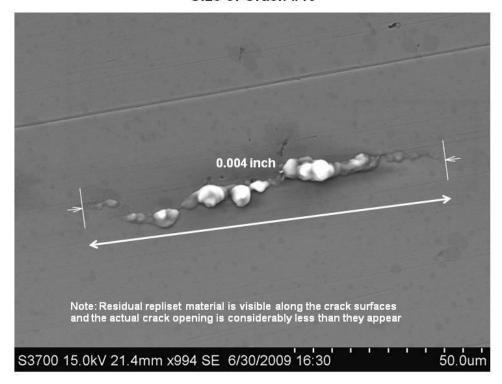


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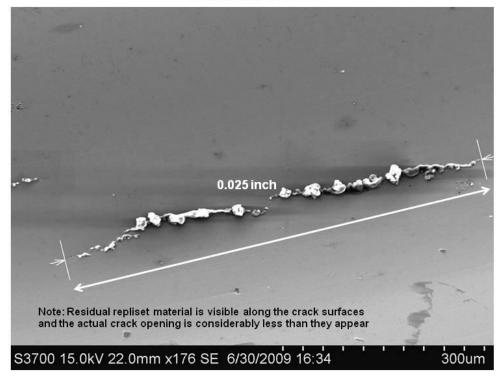


NASA Engineering and Safety C	NASA Engineering and Safety Contor	Document #:	Version:
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Size of Crack #19

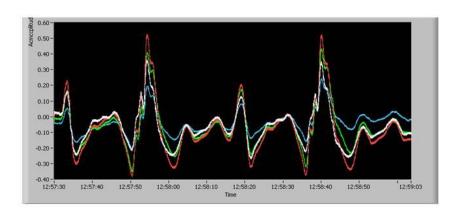


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Poppet #64
LaRC eddy current findings, the colors indicate ???



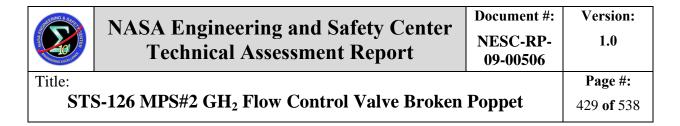
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Surface crack sizes and locations

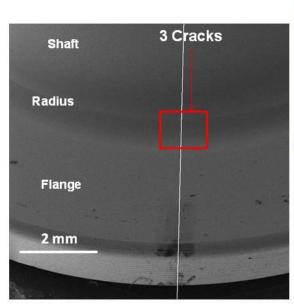
Poppet #65							
Crack Number	Size (inch)	Angle (degrees) 115					
1	0.002						
2	0.002	115					
3	0.001	115					
4	0.003	295					
5	0.004	295					

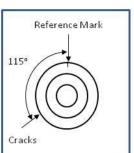
Boeing Eddy Current Findings

	Poppet #65										
Inspector	Run Data (Vpp)										
	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)		
J. Engel	0.074	0.075	0.078	0.078	0.080	0.079	0.077	No	0 ?		
J. Engel	- 1	- 40	37413	[pa]	9	- 8	9	No	295		
B. Devries	-	49	3943	194	9	- 8	8	No	295		
B. Devries	0.085	0.078	0.077	0.078	0.081	0.088	0.081	No	15?		

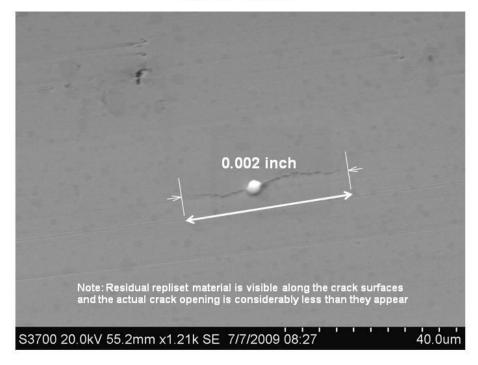


Location of Cracks #1-3

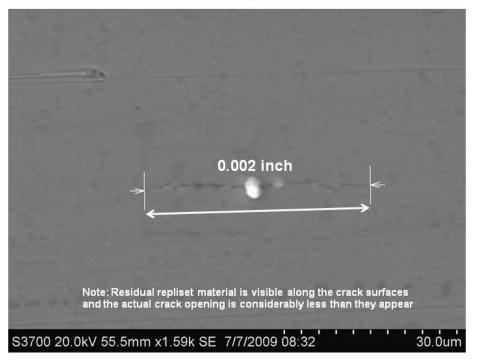




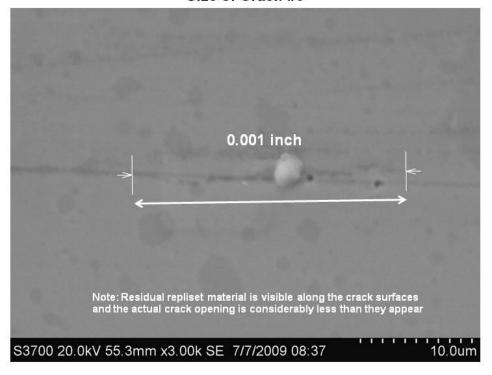
SHERWING & SAFE	NASA Engineering and Safety Center	Document #:	Version:	
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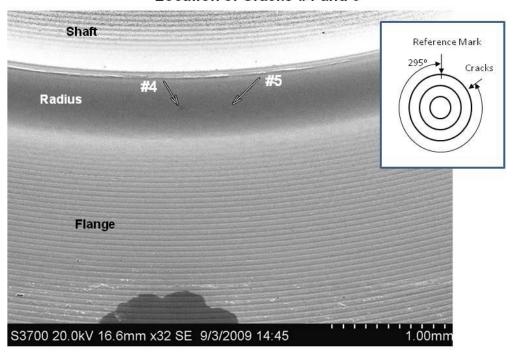


THING & STATE OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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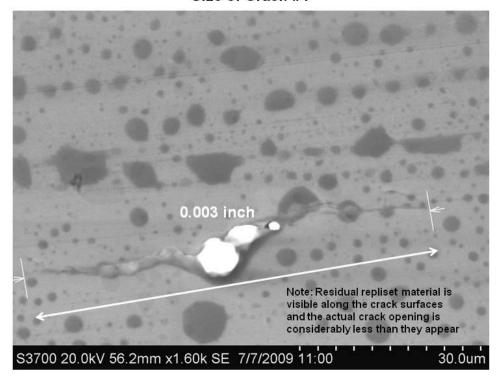


TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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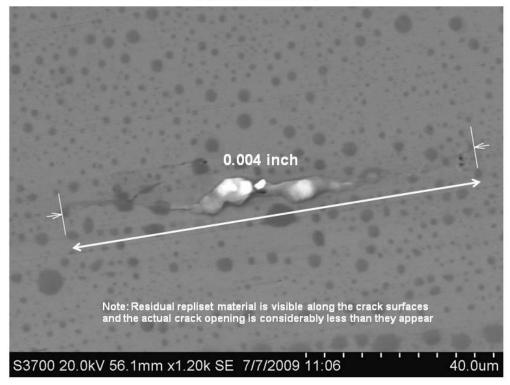
Location of Cracks #4 and 5



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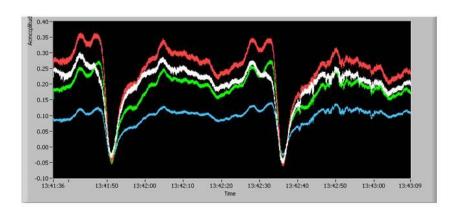


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Poppet #65
LaRC eddy current findings, the colors indicate ???



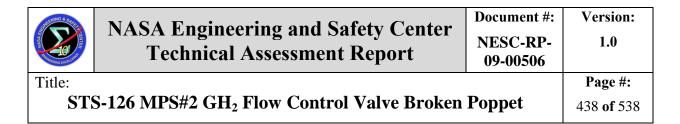
THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Surface crack sizes and locations

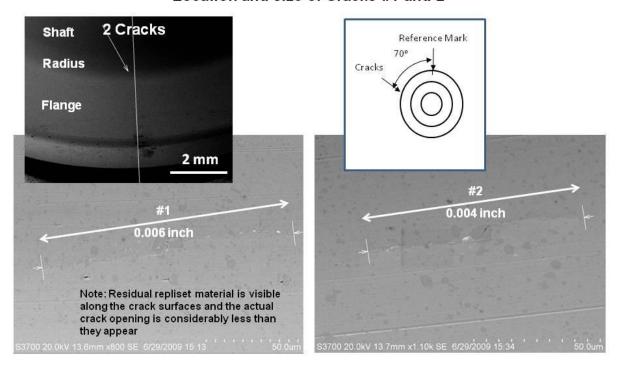
Poppet #66				
Crack Number	Size (inch)	Angle (degrees)		
1	0.006	70		
2	0.004	70		
3	0.025	250		
4	0.018	250		
5	0.024	250		
6	0.006	250		
7	0.005	250		

Boeing Eddy Current Findings

Poppet #66									
18 60 0			Ru	n Data (/pp)				
Inspector	1	2	3	4	5	6	Average	Crack Detected	Location (degrees)
J. Engel	0.190	0.194	0.183	0.189	0.186	0.192	0.189	Yes	235
J. Engel	8	-83	0.23] %	9]	- 8	. 8	No	70
3. Devries	2:	20	848	72	-2	27	74	No	70
A Devries	0.186	0.185	0.188	0.201	0.193	0.190	0.191	Yes	240

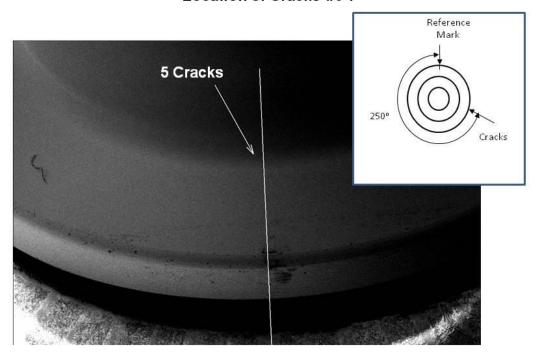


Location and size of Cracks #1 and 2

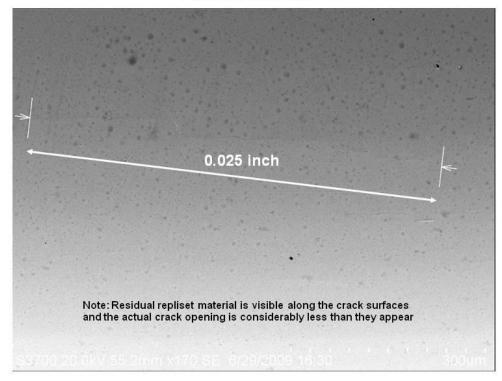


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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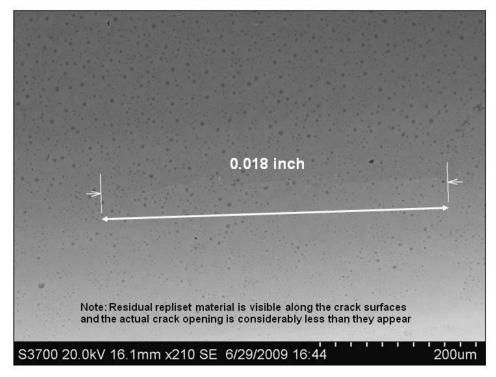
Poppet #66 Location of Cracks #3-7



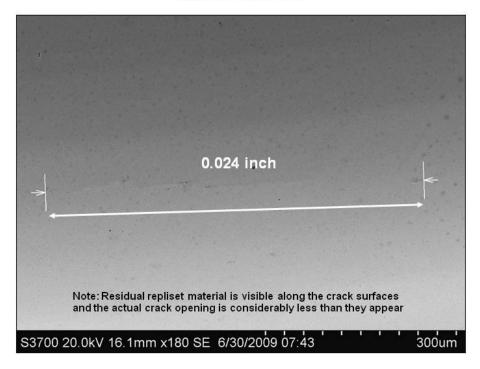
mita & garage	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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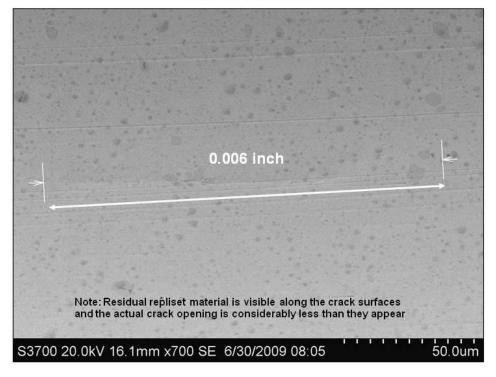
SHEERING & SAFE	NASA Engineering and Safety Center		Version:
TO THE STATE OF TH	Technical Assessment Report	NESC-RP- 09-00506	1.0
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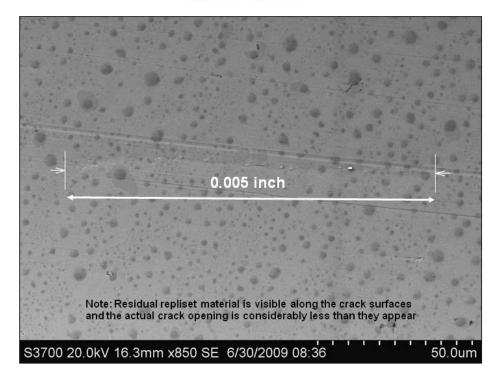
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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TENTER OF THE PROPERTY OF THE	Technical Assessment Report	NESC-RP- 09-00506	1.0
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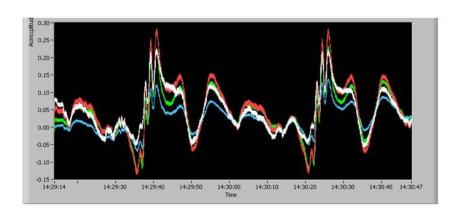


THING & COLUMN TO SERVICE AND ADDRESS OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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SECTING & SARE	NASA Engineering and Safety Center Technical Assessment Report		Version:
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Poppet #66
LaRC eddy current findings, the colors indicate ???



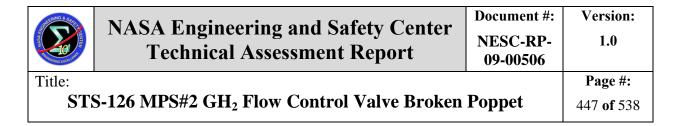
SHERING & SARE	NASA Engineering and Safety Center Technical Assessment Report		Version:
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Title:			Page #:
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Surface crack sizes and locations

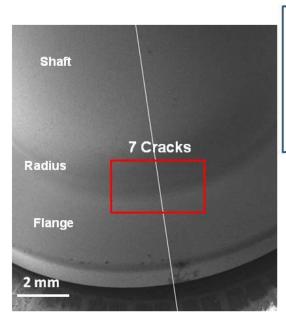
	Poppet #67			
Crack Number	Size (inch)	Angle (degrees)		
1	0.002	285		
2	0.006	285		
3	0.003	285		
4	0.004	285		
5	0.002	285		
6	0.008	285		
7	0.002	285		
8	0.009	105		
9	0.006	105		
10	0.002	105		
11	0.003	105		
12	0.003	105		
13	0.003	105		

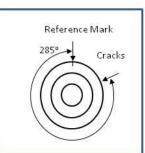
Boeing Eddy Current Findings

	Poppet #67								
7) S			Ru	n Data (\	/pp)				8
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.089	0.086	0.089	0.092	0.095	0.095	0.091	Yes	275 (Not 3:1 S/N ratio)
J. Engel	9	20	1923	312		. 20	12	No	105
3. Devries	8	78	2250	95	8	56	35	No	105
3. Devries	0.091	0.089	0.091	0.095	0.117	0.100	0.097	No	285



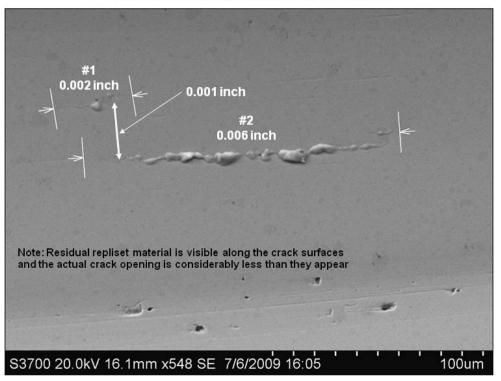
Location of Cracks #1-7



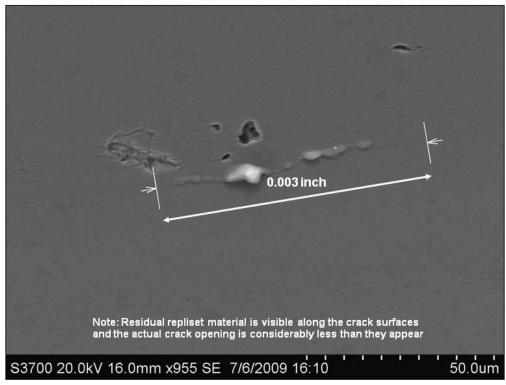


THE REPORT OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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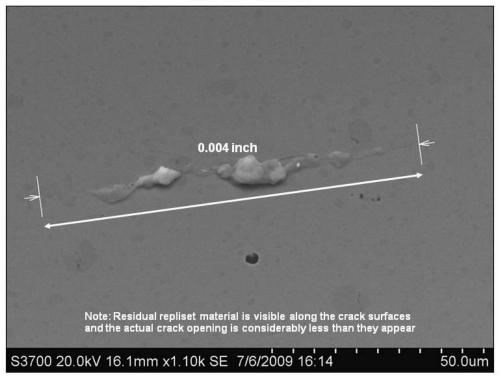
Location and size of Cracks #1 and 2



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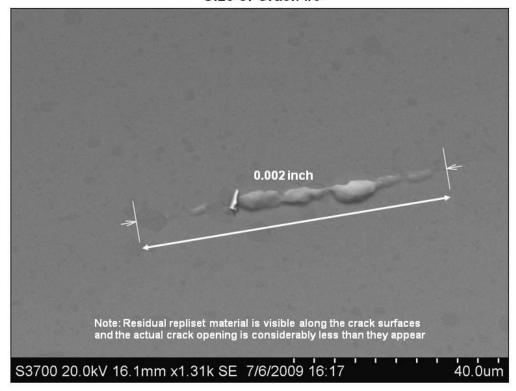


TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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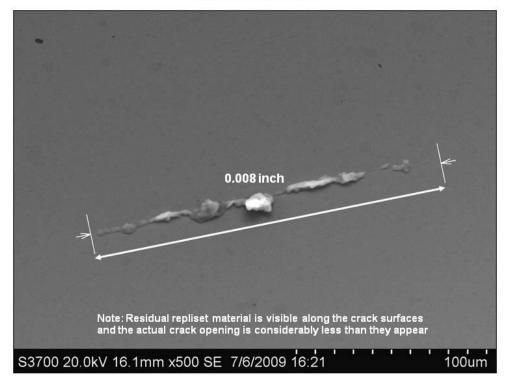


THE STATE OF THE S	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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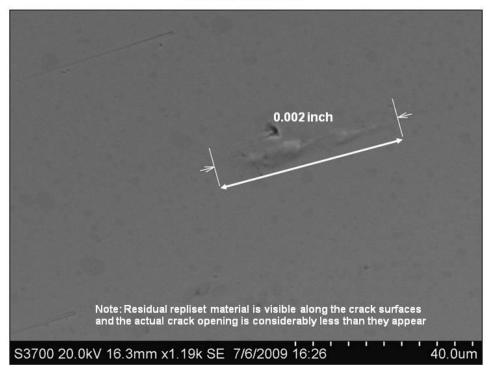
Size of Crack #5



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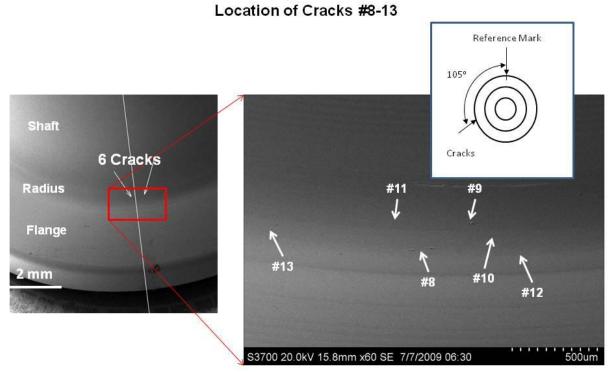


TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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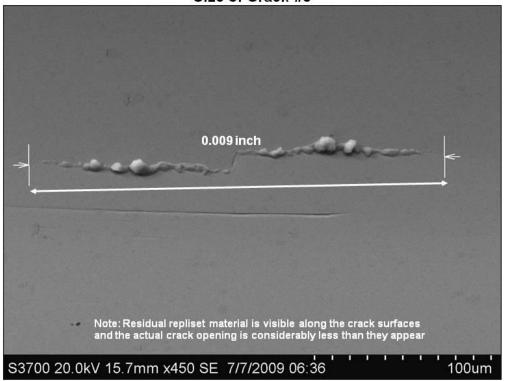
TO THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #67 Location of Cracks #8-13

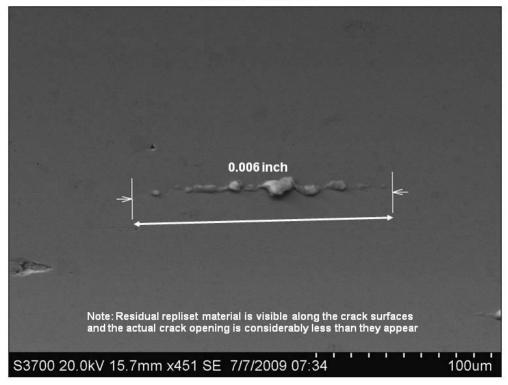


WIND A STATE OF THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #8

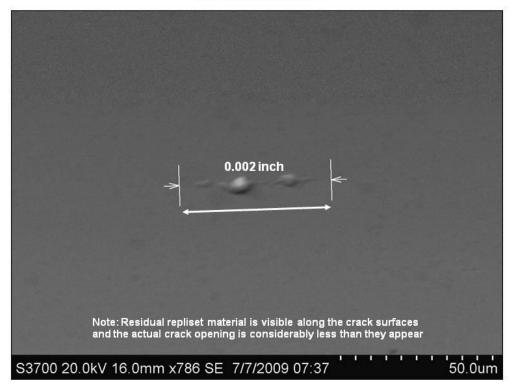


TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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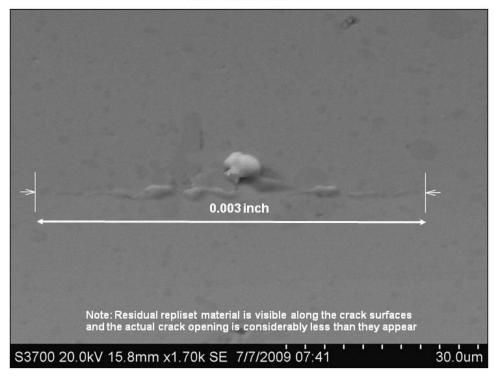


TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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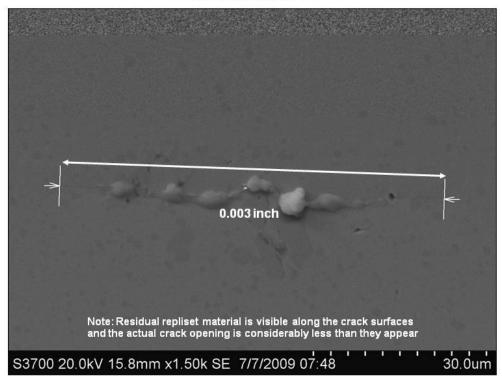
Size of Crack #10



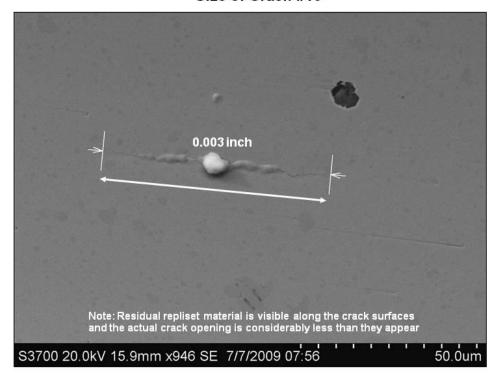
TO STATE OF THE ST	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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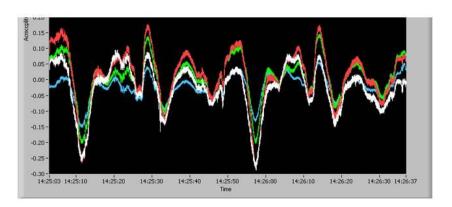


NG & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #67
LaRC eddy current findings, the colors indicate ???



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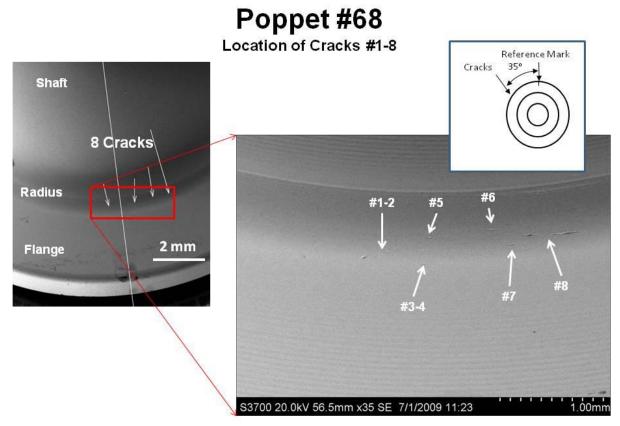
Surface crack sizes and locations

Poppet #68		
Crack Number	Size (inch)	Angle (degrees)
1	0.003	35
2	0.005	35
3	0.003	35
4	0.002	35
5	0.008	35
6	0.009	35
7	0.006	35
8	0.026	35
9	0.010	215
10	0.003	215
11	0.003	215
12	0.005	215
13	0.012	215
14	0.006	215
15	0.004	215

Boeing Eddy Current Findings

Poppet #68											
			Ru	n Data (
Inspector	1	2	3	4	5 6 Average CrackDetecte		Crack Detected	Location (degrees)			
J. Engel	0.197	0.202	0.216	0.220	0.218	0.208	0.210	Yes	40		
J. Engel	0.117	0.120	0.124	0.130	0.125	0.127	0.124	Yes	215		
3. Devries	0.213	0.219	0.222	0.220	0.224	0.220	0.220	Yes	45		
B. Devries	0.118	0.120	0.119	0.121	0.122	0.123	0.121	Yes	225		

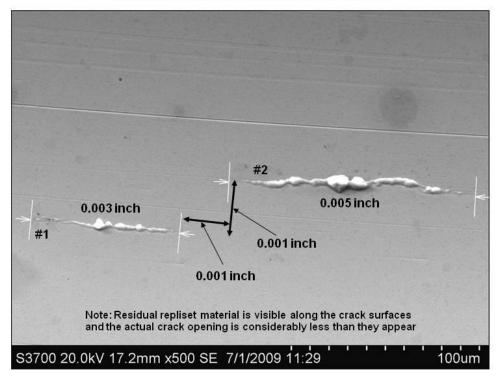
TO THE REPORT OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0		
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet					
515-120 WII 5#2 GII2 Flow Collifor Valve Drokell I opper					



THE STATE OF THE S	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0	
Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet				

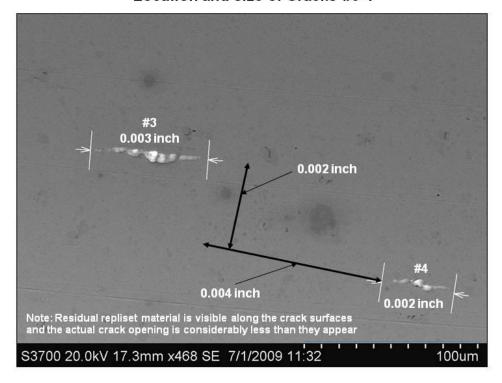
Poppet #68

Location and size of Cracks #1 and 2

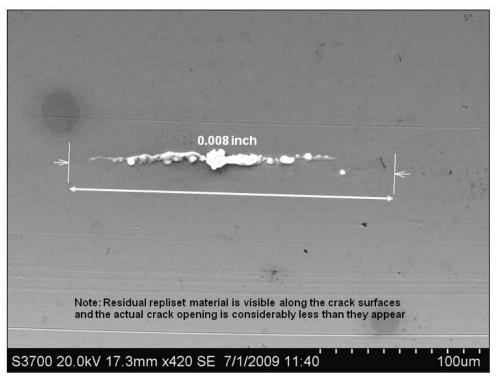


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Poppet #68
Location and size of Cracks #3-4

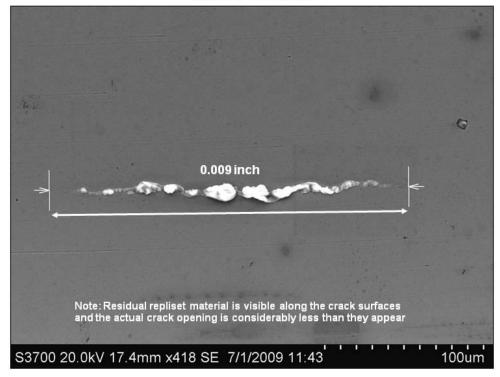


TOTAL SECTION OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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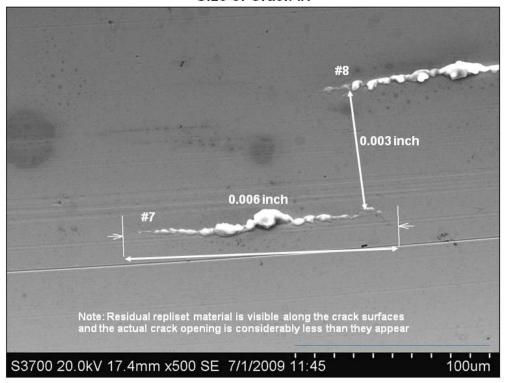


muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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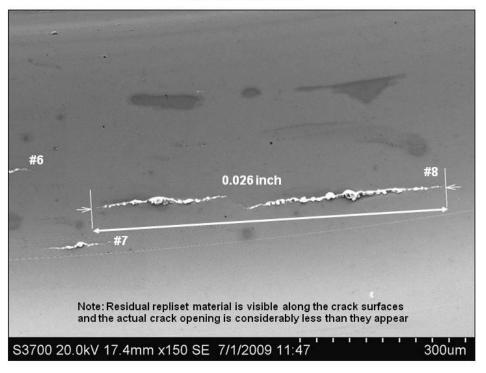
Size of Crack #6



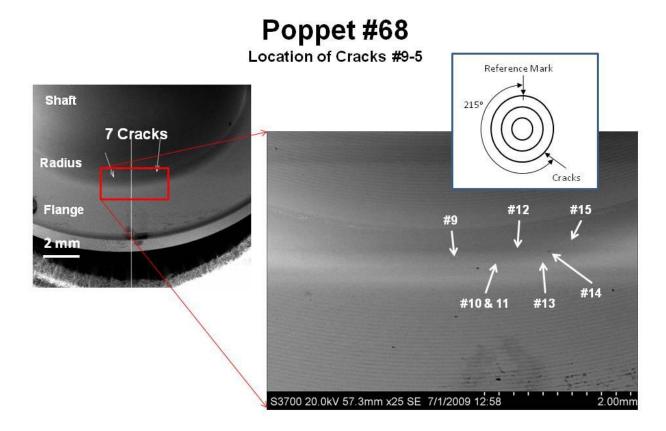
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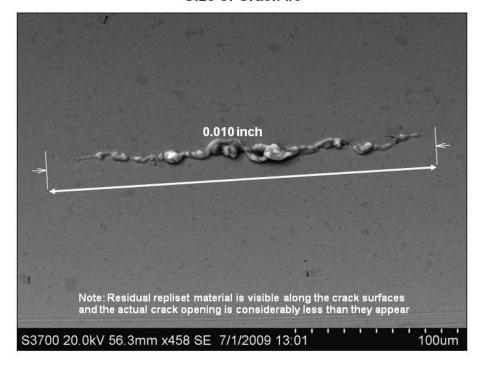
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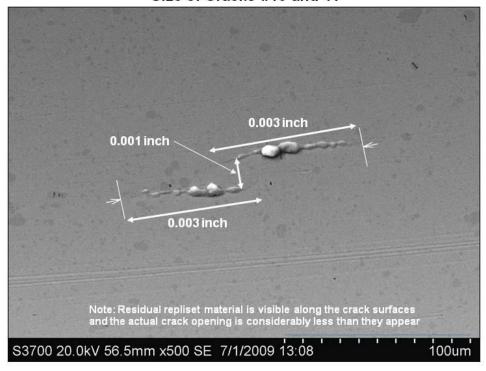


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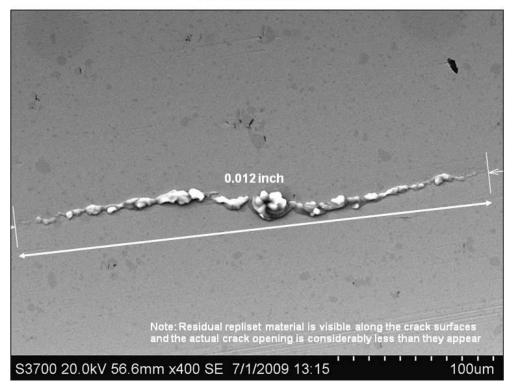
Size of Cracks #10 and 11



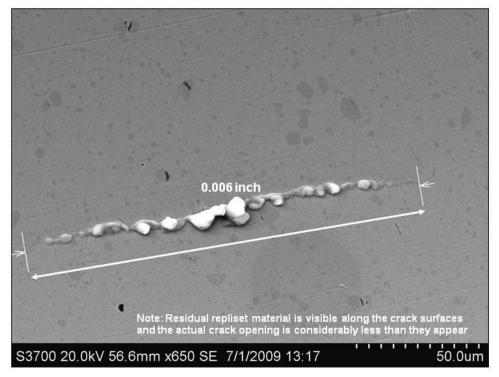
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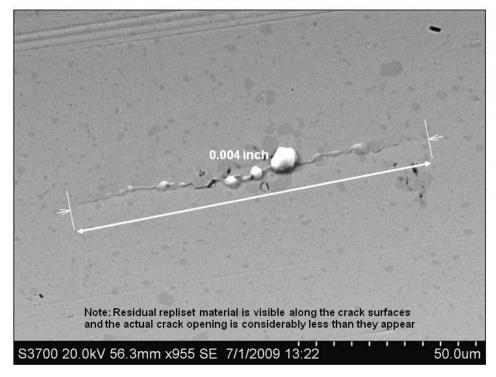
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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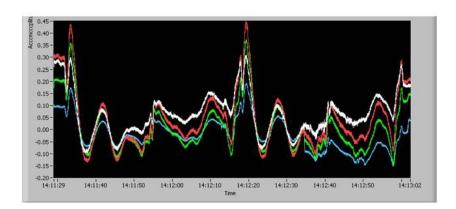


THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #68
LaRC eddy current findings, the colors indicate ???



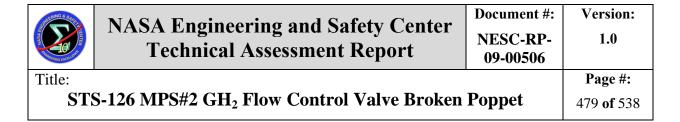
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AND THE PROPERTY OF THE PROPER	Technical Assessment Report	NESC-RP- 09-00506	1.0
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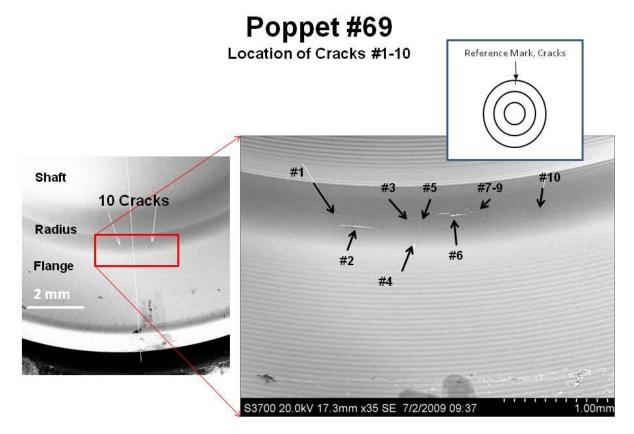
Surface crack sizes and locations

Poppet #69					
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)
1	0.003	0	11	0.004	180
2	0.015	0	12	0.004	180
3	0.002	0	13	0.011	180
4	0.002	0	14	0.007	180
5	0.001	0	15	0.006	180
6	0.014	0	16	0.006	180
7	0.003	0	17	0.008	180
8	0.003	0	18	0.003	180
9	0.003	0	19	0.008	180
10	0.002	0	20	0.003	180
			21	0.005	180

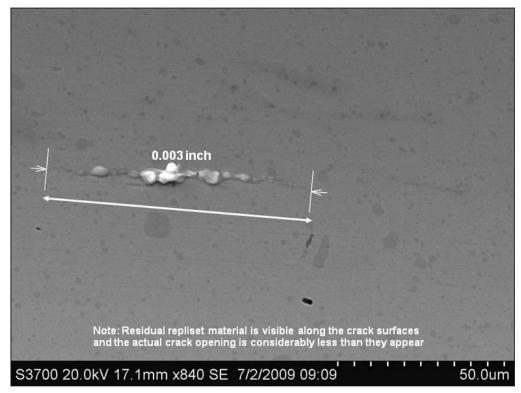
Boeing Eddy Current Findings

None Provided



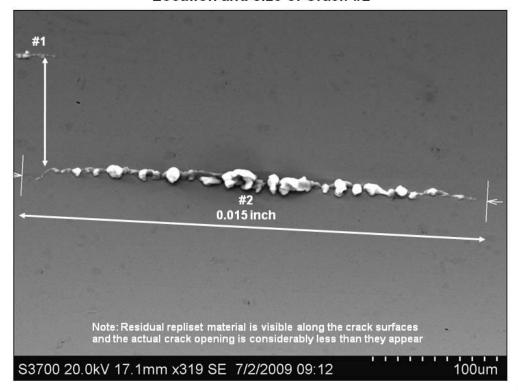


THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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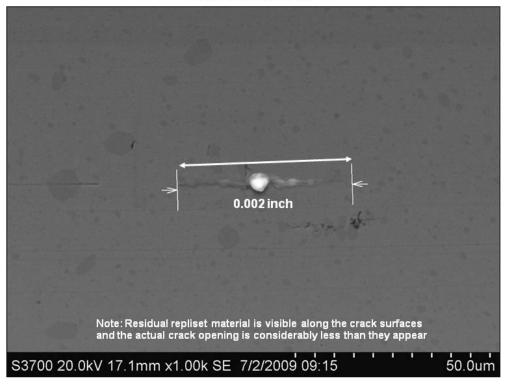


TO THE TENT OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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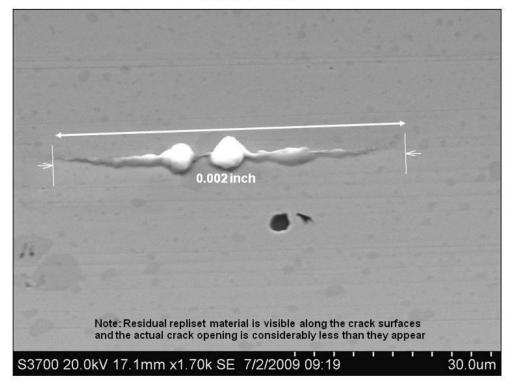
Location and size of Crack #2



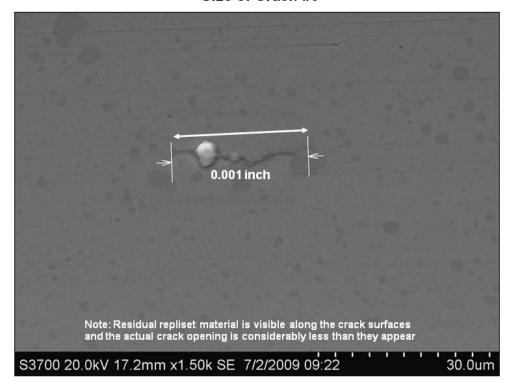
TO THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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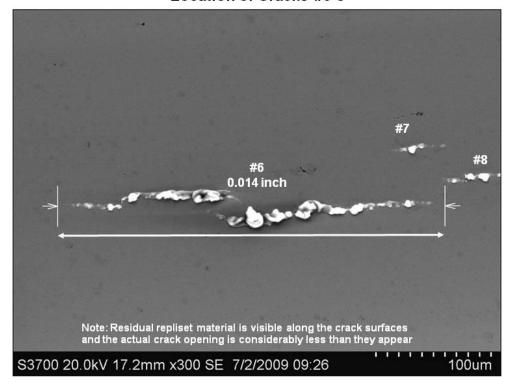


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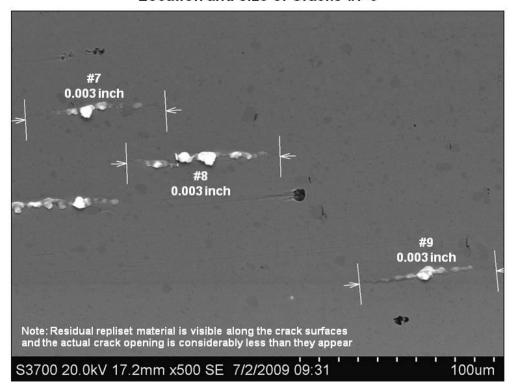
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location of Cracks #6-8



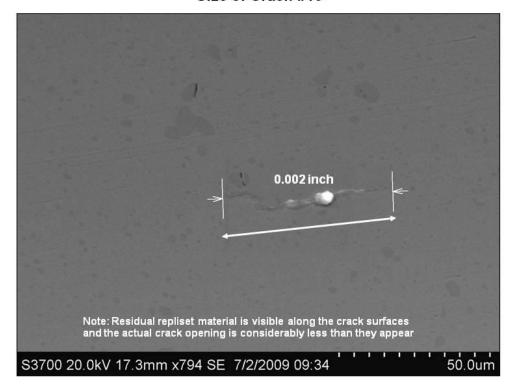
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #7-9

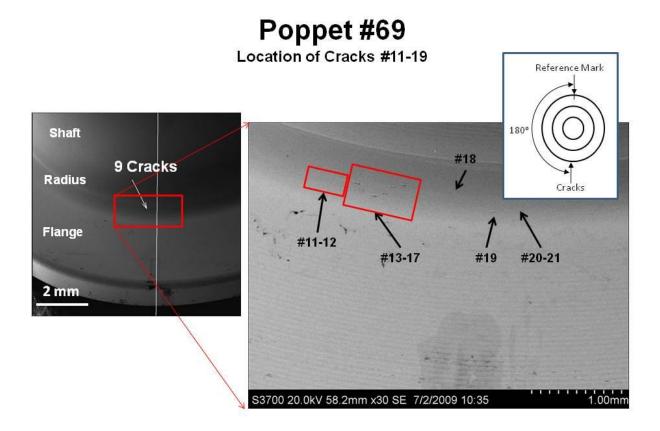


THE RESERVE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
Title:	Title: STS-126 MPS#2 GH ₂ Flow Control Valve Broken Poppet		

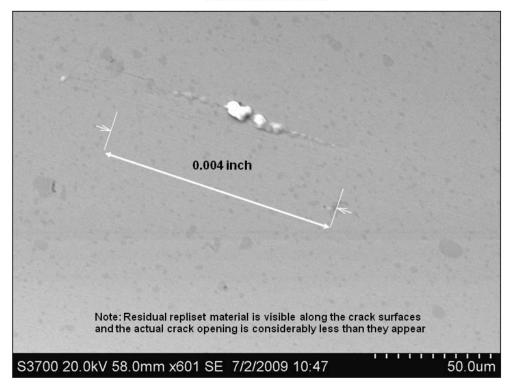
Size of Crack #10



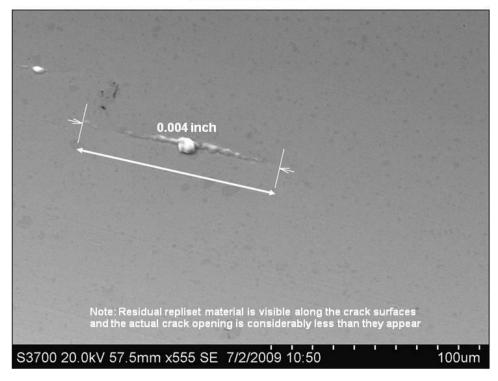
THURS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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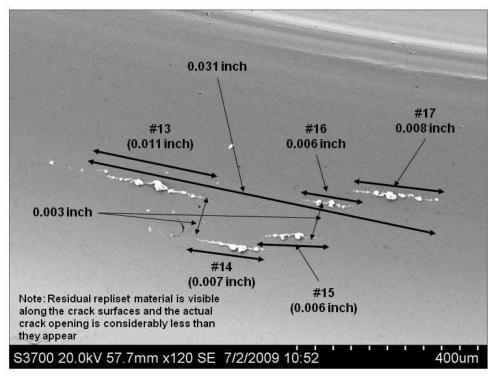


THE A STATE OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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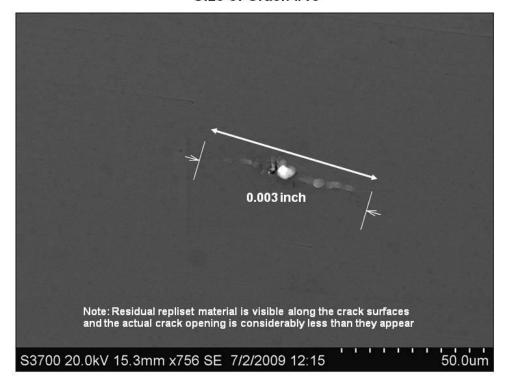
THE REPORT OF THE PARTY OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #13-17



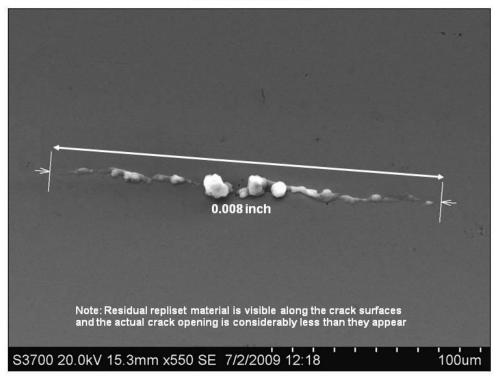
TRING & COLOR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #18



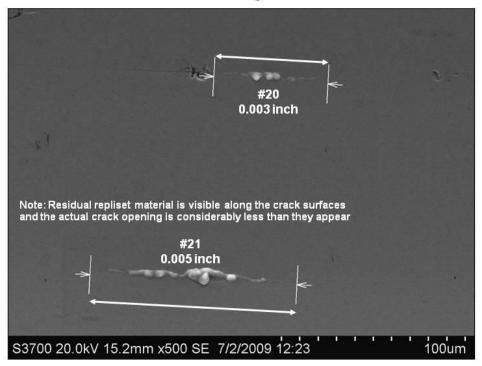
THIS & SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #19



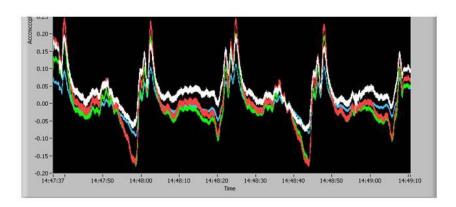
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Title:			Page #:
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Poppet #69
Location and size of Region B Cracks #20-21



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A SALVANO EXCENSES	Technical Assessment Report		1.0
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Poppet #69
LaRC eddy current findings, the colors indicate ???



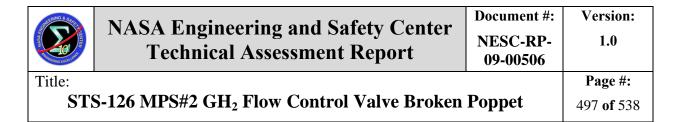
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Surface crack sizes and locations

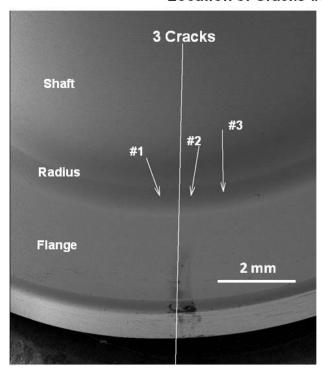
	Poppet #70	
Crack Number	Size (inch)	Angle (degrees)
1	0.005	145
2	0.016	145
3	0.003	145
4	0.008	325
5	0.004	325

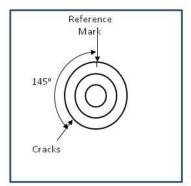
Boeing Eddy Current Findings

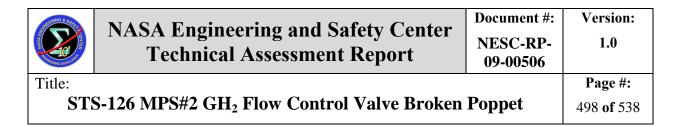
Poppet #70									
			Ru	ın Data (Vpp)		ψ.		
Inspector	1	2	3	4	5	6	Average	CrackDetected	Location (degrees)
J. Engel	0.184	0.187	0.174	0.172	0.176	0.178	0.179	Yes	145
J. Engel	0.093	0.097	0.100	0.100	0.102	0.100	0.099	Yes	345 (Not31S/Nratio)
B. Devries	0.171	0.181	0.177	0.190	0.189	0.201	0.185	Yes	150
B. Devries	0.095	0.089	0.096	0.108	0.101	0.102	0.099	No	355



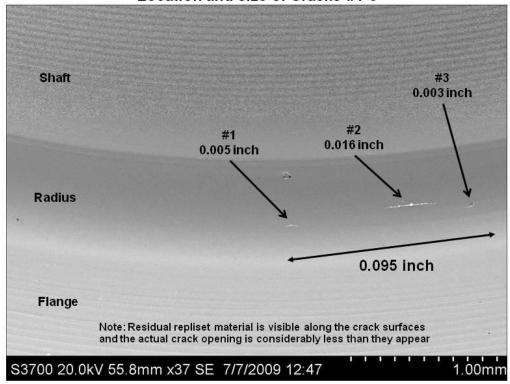
Location of Cracks #1-3





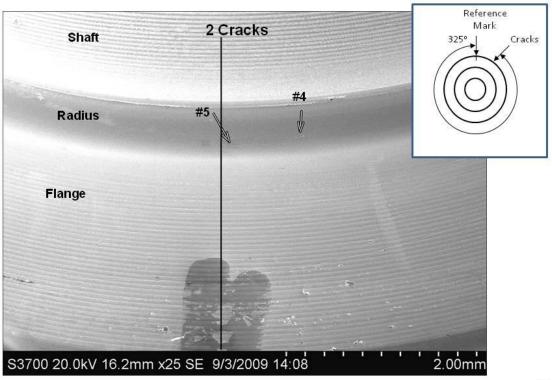


Location and size of Cracks #1-3

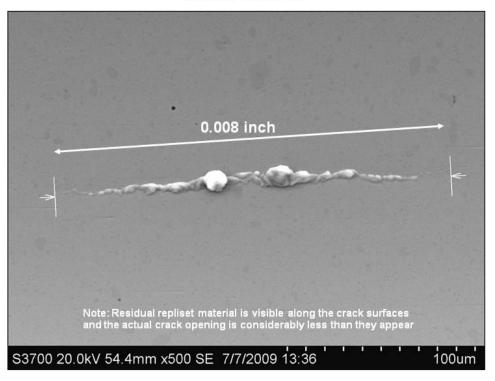


muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location of Cracks #4 and 5

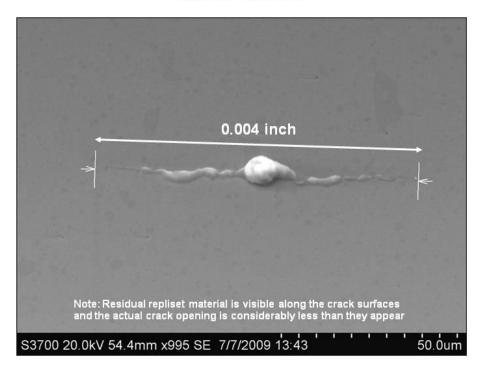


muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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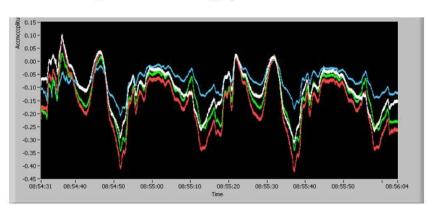
TOTAL STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Size of Crack #5



STATE LANGE AS A STATE OF THE S	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #70
LaRC eddy current findings, the colors indicate ???

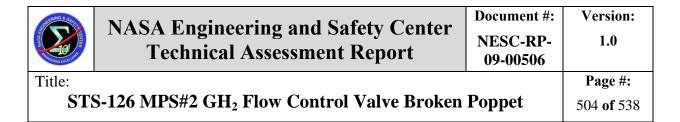


NG 2 STATE OF THE	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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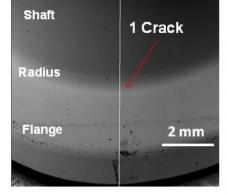
Surface crack sizes and locations

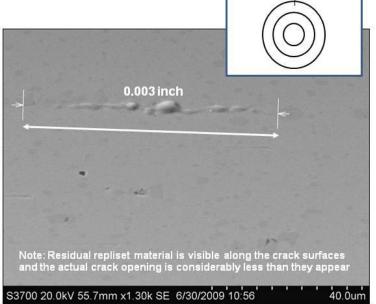
Boeing Eddy Current Findings

None Provided





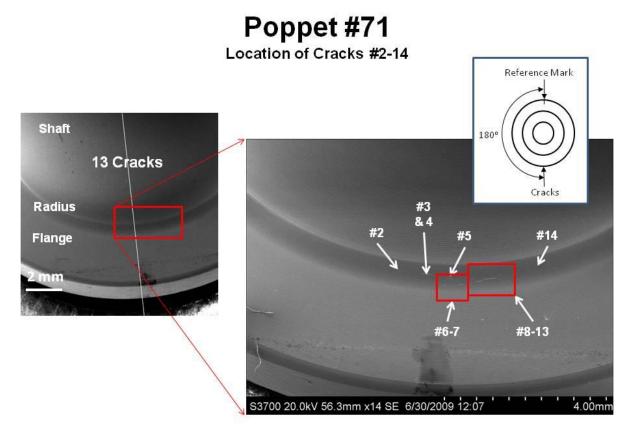




501

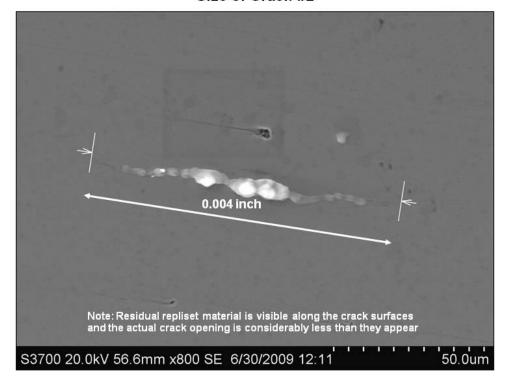
Reference Mark, Crack #1

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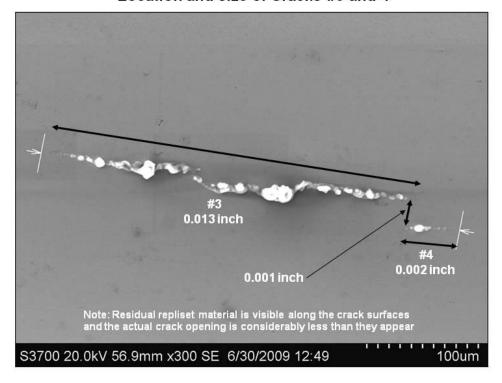
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TO SWEET	Technical Assessment Report	NESC-RP- 09-00506	1.0
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Poppet #71 Size of Crack #2



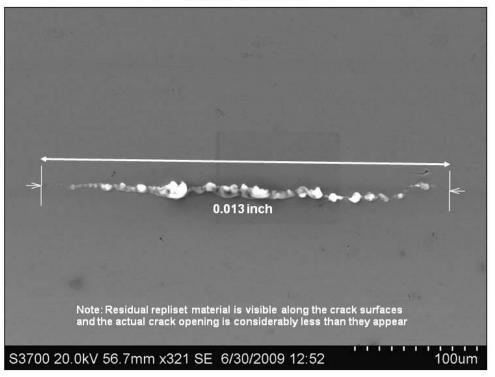
THING & CALLED	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #3 and 4



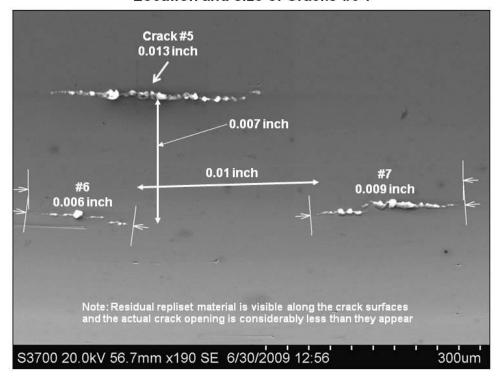
muld & Sol	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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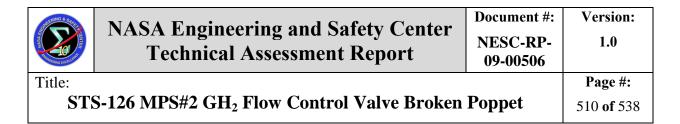
Poppet #71 Size of Crack #5



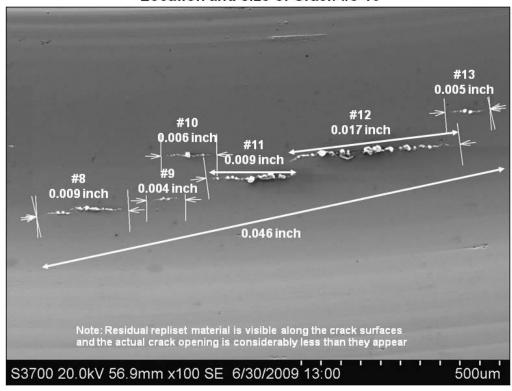
TO THE REAL PROPERTY OF THE PARTY OF THE PAR	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #5-7



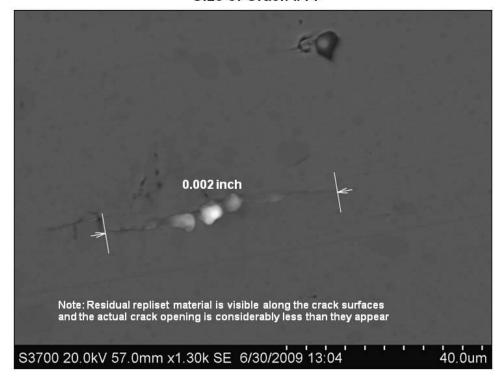


Location and size of Crack #8-13



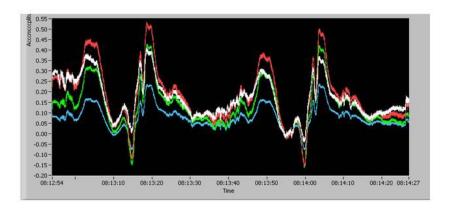
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THE STATE OF THE S	Technical Assessment Report	NESC-RP- 09-00506	1.0
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Size of Crack #14



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Poppet #71
LaRC eddy current findings, the colors indicate ???

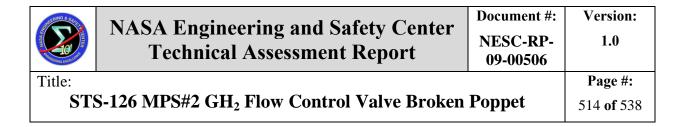


THING & CALLED	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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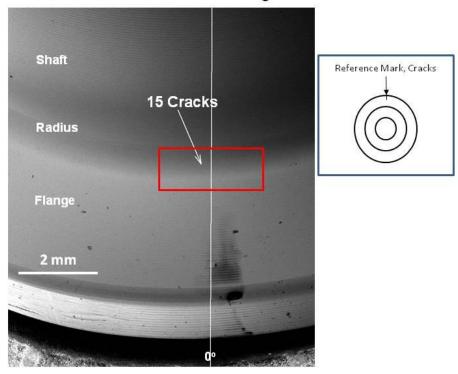
Surface crack sizes and locations

Poppet #72					
Crack Number	Size (inch)	Angle (degrees)	Crack Number	Size (inch)	Angle (degrees)
1	0.006	0	15	0.003	0
2	0.002	0	16	0.003	180
3	0.004	0	17	0.005	180
4	0.002	0	18	0.010	180
5	0.004	0	19	0.006	180
6	0.006	0	20	0.002	180
7	0.002	0	21	0.027	180
8	0.001	0	22	0.006	180
9	0.001	0	23	0.007	180
10	0.002	0	24	0.009	180
11	0.003	0	25	0.004	180
12	0.002	0	26	0.004	180
13	0.002	0	27	0.005	180
14	0.002	0	28	0.004	180

Boeing Eddy Current Findings None Provided



Location of Region A



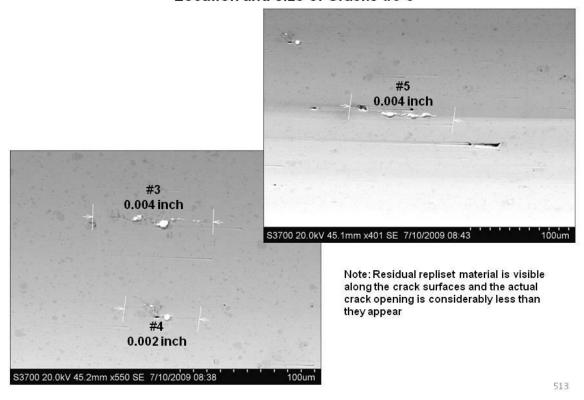
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Location and size of Cracks #1 and 2



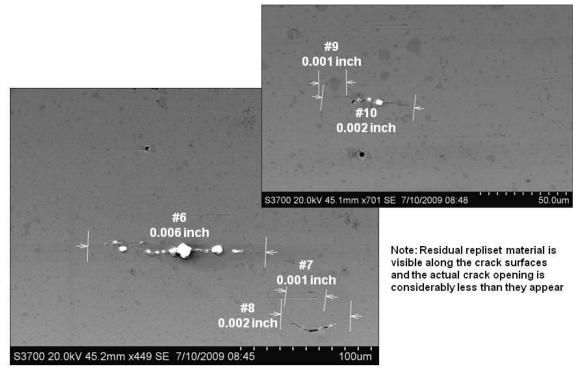
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Poppet #72 Location and size of Cracks #3-5



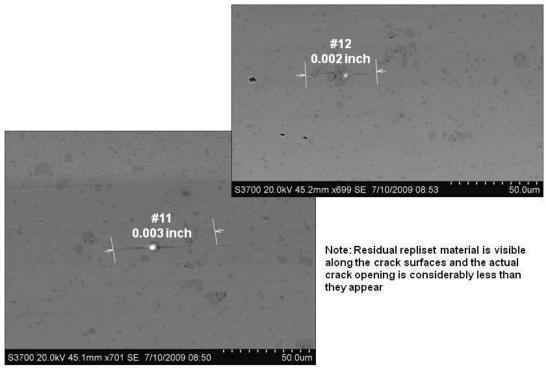
THING A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Location and size of Cracks #6-10



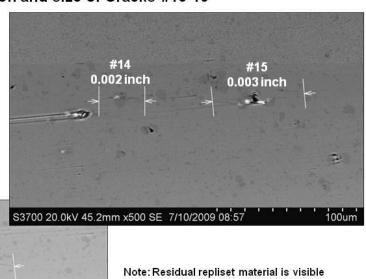
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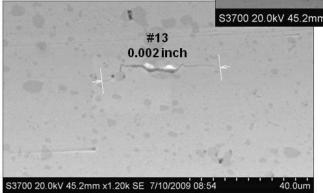
Location and size of Cracks #11-12



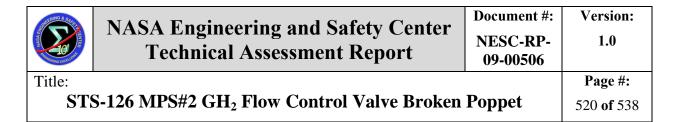
THING & STATE OF THE PARTY OF T	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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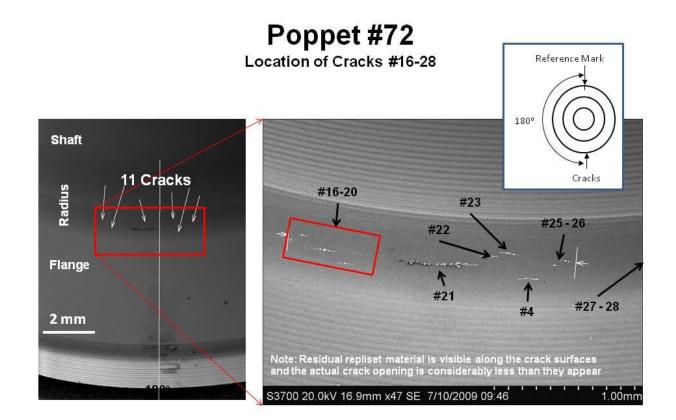
Location and size of Cracks #13-15





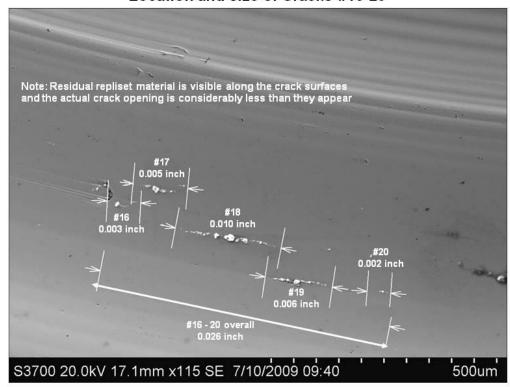
Note: Residual repliset material is visible along the crack surfaces and the actual crack opening is considerably less than they appear

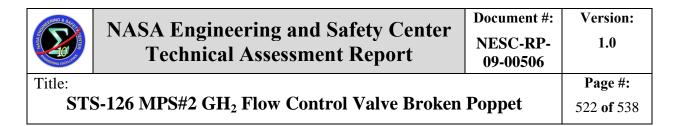




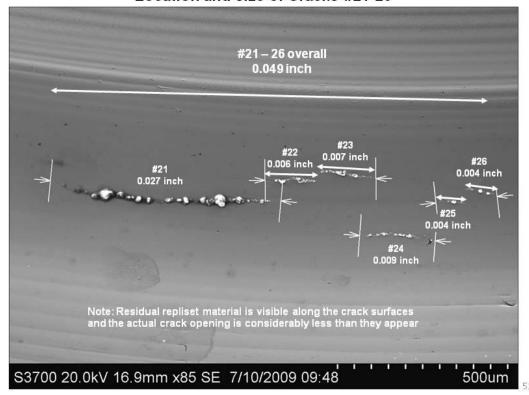
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Location and size of Cracks #16-20





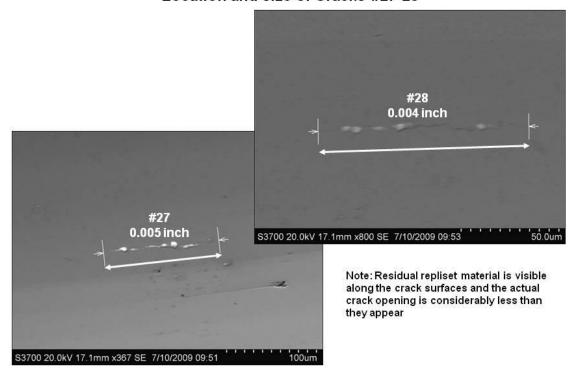
Location and size of Cracks #21-26



NESC Request No.: 09-00506

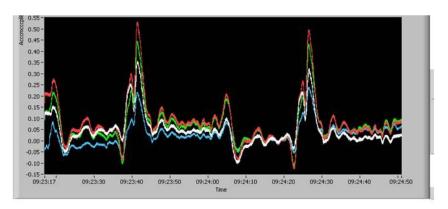
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Poppet #72 Location and size of Cracks #27-28



THE A SALES	NASA Engineering and Safety Center Technical Assessment Report	Document #: NESC-RP- 09-00506	Version: 1.0
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Poppet #72
LaRC eddy current findings, the colors indicate ???



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Appendix B. Boeing Eddy Current Procedure/Technique Sheet Flow Control Valve Poppet - #SSO-01, Revision C

Authors of the contents of this appendix were Team members for the NESC and were under NASA funding at the time the work was conducted.



Eddy Current Procedure/Technique Sheet Flow Control Valve Poppet - #SSO-01 Revision: C

Title: Flow Control Valve Poppet

Program: Space Shuttle Orbiter

Part Name and Number: Flow Control Valve Poppet, 80913 (Figure 1)

Applicable Documents: MT0501-513, "Inspection, Eddy Current, Requirements for"

Material & Condition: Cres 440A or 440C

Material Thickness: Approximately 0.100" in shank and 0.045" at flange

Instrumentation: UniWest US-454 Eddy Current Flaw Detector (Figure 2), or equivalent.

Recording Devices: US-454 electronic strip chart

Alarm: Not required

Probes: US-1839 differential coil, 0.0312" radius, S/N 37072 (Figure 3), 37073 or equivalent.

Fixtures and Scanning Aids: US-1779 Bolt Inspection Scanner (Figure 4). The scanner

includes a probe fixture to hold probe at a 45-degree angle into the flange radius. In addition to the scanner, a threaded bolt with handle and a rubber band allows a slow and steady

ndex.

Reference Standard: NASA supplied Titanium fastener with EDM notches (Figure 5-7)

"VM Ti/TB-STD-JSC, VM PRODUCTS, INC., JS-061208-01"

Standardization Checks: Before inspecting the first part and after inspecting all poppet

hardware, or at 4-hour intervals if less time, check the system sensitivity by scanning the reference standard EDM notch A in the

shank and notch B in the radius.

Scan Rate and Index: Scan speed shall be the same for both reference standard and

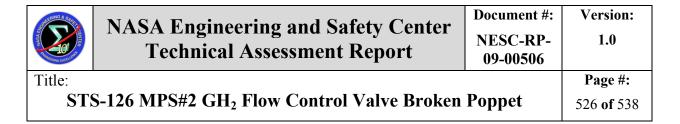
hardware. Index turn speed used to inspect the part shall not exceed

that used to scan the reference standard.

Acceptance Criteria: Reject all relevant indications that exceed 0.2 Vpp.

Test Facility: Boeing, 5301 Bolsa Ave., Huntington Beach, California. 92647

Prepared By: Bob DeVries Validated By: Jim Engel



EDEINO	Flow Control Valve Poppet - #SSO-01 Revision: C		
I evel III Annroval		Date: 11/6/09	



Document #: Version:

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Eddy Current Procedure/Technique Sheet Flow Control Valve Poppet - #SSO-01 Revision: C

A. Setup/Preparation

- Verify that the serial number on the part is identical to the serial number on the Fabrication Order (FO) or Work Order (WO). Notify inspection supervisor if the above criterion fail.
- 2. Verify calibration of the eddy current instrument prior to inspecting hardware.
- Prior to performing eddy current inspection, remove dust, machining chips, oil, or other foreign matter that could interfere with the inspection from the inspection surface.
- 4. Prior to performing eddy current inspection, test each poppet for a residual magnetic field using a field indicator. If the field is greater than 0.5 Gauss, demagnetize the poppet using a Parker Probe on the AC setting.
- 5. Prior to performing eddy current inspection, replace the Kapton tape on the probe tip. After inspecting six poppets, replace Kapton tape. Install probe in probe holder at bottom of slot in probe with probe S/N annotation located on the right side.
- 6. Connect probe, power and printer cables.

B. Standardization

- 1. Turn on the eddy current test instrument.
- Choose large strip chart display on the left side and small impedance plane display on the right side by pressing DISPLAY ON/OFF.
- 3. Establish the following parameters on the eddy current instrument. Recall file # "poppet", by pressing the ENTER button, turning the scroll wheel until RECALL SET is highlighted, press ENTER, scroll to the file "poppet" and press ENTER. Verify the following inspection parameters by comparing them to the displayed values on screen (press ENTER and scroll the wheel to see these parameters).
 - Frequency (FREQ1) = 2 MHz
 - Channel Gain = 20.0 dB
 - Phase Angle (ROT) = 294 Deg.
 - Probe Drive = HI
 - Low Pass Filter = 100 Hz
 - High Pass Filter = OFF
 - X sensitivity = 0.5 V/div
 - Y sensitivity = X
- 4. Place the reference standard on the rollers of the scanner (Figure 8). Adjust the head side thrust bearing so that it lightly contacts the standard head at centerline of head. Spin test by turning the roller RPM control to lowest RPM (first setting). Confirm that the standard spins freely but does not rock or climb onto rollers and stop the scanner.



Title:

STS-126 MPS#2 GH₂ Flow Control Valve Broken Poppet

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Eddy Current Procedure/Technique Sheet Flow Control Valve Poppet - #SSO-01 Revision: C

- 5. Lower the probe, rotate to a vertical position, make an adjustment to probe height in probe holder, if necessary, and contact the spinning standard such that the probe coil area contacts the bolt shank. Locate probe on the shaft near the threads, turn index screw to meet probe holder and apply rubber bands as shown in Figure 8.
- Shank Standardization. Scroll to SAVE DATA, leaving the cursor in that position. Start scanner and press NULL and ERASE. Scan the reference standard shank by turning the index handle and slowly index probe across shank to image the small notch with the probe held at zero degrees (normal). Press ENTER. Stop scanner. Remove probe from shank.
- 7. Continue to save data by scrolling to the file name "tishank1" and press ENTER four times. Scroll to RECALL DATA, press ENTER and scroll to saved file name "tishank1" and press ENTER. Press F1 (FRAME AT) and scroll to view the largest signal on the strip chart and press F2 (FRAME LENGTH), ENTER and scroll to enlarge the chart, keeping the largest signal in the right side chart (showing twice). Press CANCEL twice.
- 8. Evaluate the signal from Notch A, 0.030-inch deep notch (Figure 6) and verify that the signal height from this notch is at least 0.2 Vpp or re-scan shank.
- 9. <u>Radius Standardization.</u> Press ENTER and scroll to SAVE DATA. Move the probe onto the standard shank; translate the probe to contact the 0.030" radius with the rubber band removed. Move probe in probe holder to contact part radius, if needed. Angle the probe 45-degrees and turn on bolt scanner. Press NULL and erase. Maximize signal from 0.050" long EDM notch by angulating probe. Press ENTER.
- 10. Re-save scan data under the file name "tirad1" and move probe from reference standard. Save, recall file and press CANCEL, as in step 7.
- 11. Evaluate the signal from Notch B in the radius (only one in radius), and verify that the signal height from this notch is at least 0.7 Vpp or re-scan radius.
- 12. The eddy current signal from the reference notch in the radius shall be greater than 0.7 Vpp or re-run the standardization. Retract probe and remove reference standard from bolt scanner.

C. Inspection

- Personnel performing this inspection require a minimum Level II certification in eddy current per BSS 7698. The inspector must wear clean gloves while handling flight poppets. Perform a magnified visual inspection of the 0.045" flange and radius at 25 to 30X magnification and report all nicks, burrs, dents, pitting, or scratches on the appropriate documentation.
- 2. Inspection area is limited to the 0.480" diameter shank (as much as possible with the probe normal to the shank) and the 0.030" radius located between the 0.480" shank, and 0.045" flange of the part unless otherwise specified.



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Assure that the scanner is not moving. Install probe in probe holder at top of slot in probe with probe S/N annotation located on the right side. Verify that there is Teflon tape attached to the entire end of the head side thrust bearing or apply new tape.

CAUTION: Use extreme care to prevent damage to the end flange of the poppet as this is a critical dimension for the functionality of the poppet.

- 4. Wearing clean gloves place the poppet on the rollers of the scanner and attach a tiestrap to secure poppet during flange scanning (Figure 9). Move probe in probe holder to top of slot in probe. Lower the probe to verify that the probe coil is in contact with the shank. If needed, adjust the probe in the probe holder to assure intimate contact with the radius. Retract the probe. Rotate the indexing screw clockwise to allow full range of probe.
- 5. Shank Inspection. Spin the poppet by turning the scanner using the first speed setting and assure that the poppet spins in a stable manner. Lower the probe onto the poppet shank normal to the shank so that the coil area contacts the shank between the flanges of the poppet and away from either flange. Attach the rubber band and press NULL. Turn the index screw clockwise until reaching the mid-flange and press ERASE. Scan the shank by turning the indexing handle counterclockwise and slowly indexing probe across shank to cover entire length between the end flange and the mid-flange. Press ENTER and retract the probe and stop the scanner. Save file with the part S/N appended with an "s" for shank scan as follows. Press ENTER, scroll to locate each alphanumeric character, press F2 after displaying each character and press ENTER (e.g. 00023s1 for first inspection, 00023s2 for second, etc.). Recall file and print data to maintain hard copy records (described in step 7).
- 6. Radius Inspection. Place the poppet on the rollers of the scanner and attach a tiestrap to secure poppet during flange scanning (Figure 10). Adjust the head thrust bearing to contact the poppet flange. Move probe in probe holder to contact part radius. Turn on scanner, move the probe tip into the flange radius, and turn the probe to 45-degrees. Ensure a light but steady contact and press NULL and ERASE. After scanning the radius area for a few seconds, slowly lift the probe vertically by up to 1/16" to scan the radius flange area. Slowly lower the probe; continue scanning across the radius into the shank while gradually changing the probe angle lower than 45 degrees until the probe reaches between 0 and 30 degrees to the shank. Stop scanning after approximately 1/8 inch of shank has been scanned. Relocate the probe in the radius at 45-degrees. Move probe slightly by angling while maintaining contact with the radius to maximize any signals shown on the strip chart. Press ENTER on SAVE DATA. Retract the probe and stop the scanner.
- 7. <u>Save and Print Data.</u> Save file with the part S/N appended with an "r" for radius scan. Press ENTER, scroll to PRINT SCREEN and press ENTER. The screen will freeze as the printing process takes about one minute. Press CANCEL twice after printing to return to the scrolling strip chart screen.



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8. Perform standardization after inspecting the last poppet in the lot, if four hours have passed or before turning off the instrument. Scan and save files with names of "tishank2" and "tirad2" as detailed in step B for this post-inspection reference standard scan. If standardization results do not repeat per step B-8 and B-11, reinspect all hardware up to the point of the last successful standardization.

D. Evaluation:

- Analyze the data by observing phase angle and vertical amplitude of the indication (Figure 11). Interpret signals by comparison of strip chart signature from poppets with strip chart signatures of reference standard EDM notches.
- Eddy current signal amplitude in volts, peak-to-peak (Vpp) displays a relative severity of cracks. Crack length, depth and separation will increase the amplitude. Length effect may saturate the signal on longer cracks. Depth effect saturates the signal on cracks deeper than four mils.
- 3. Crack signals have a phase angle of 100 to 130 degrees and typically have higher amplitudes than machine grove signals. Small, tight fatigue cracks produce a signal with a single hump similar to the calibration notch. Longer cracks (>.2") show two humps due to the differential coil design. Machine grove (tool mark) signals are less than 0.15 0.2 Vpp and may have a flatter shape.
- 4. A small circumferential crack signature appears as a single hump. A small surface anomaly (missing carbide particle, pore) would also appear as small hump. Use high magnification visual inspection to rule out these non-relevant indications (if applicable). A long circumferential crack (>.15") may appear as a double-humped signal.
- 5. Ensure that minimal lift-off signal is included in the Vpp calculation (signal that projects to the left of the baseline or negative reading). Also, ensure that the etched in serial number on the shank does not create non-relevant signals with a flat top shape that may exceed 0.2 Vpp.
- Reject all relevant indications that exceed 0.2 Vpp. Relevant indications are suspect
 if they increase by ≥0.020 Vpp from a previous scan of ≥0.050 Vpp (send reports to
 engineering for further evaluation).
- Report all rejections on nonconformance paperwork, noting the Vpp level and circumferential location of indications.

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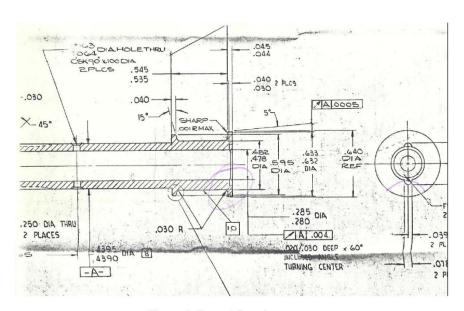


Figure 1: Poppet Drawing

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Figure 2: UniWest US-454 Eddy Current Flaw Detector

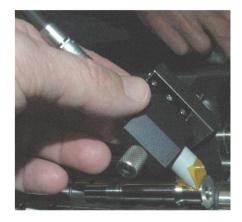


Figure 3: US-1839 Probe with "V" Shape 0.0312" Radius

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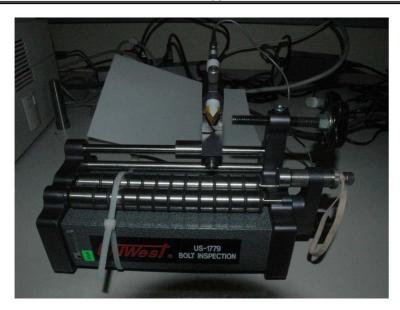
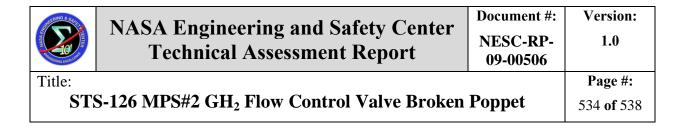


Figure 4: US-1779 Bolt Inspection Scanner



Figure 5: Reference Standard





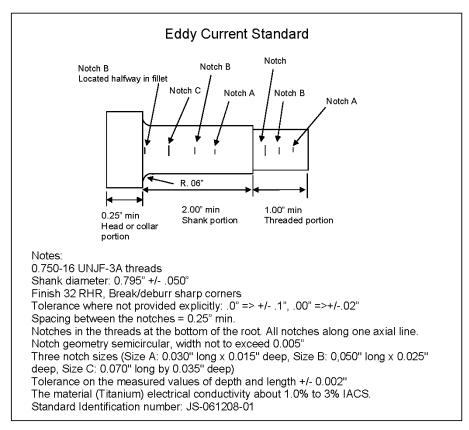
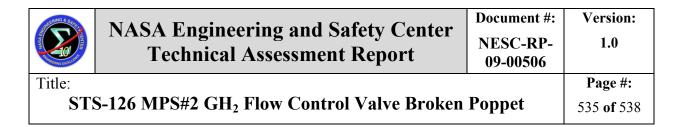


Figure 6: Reference Standard Design Drawing



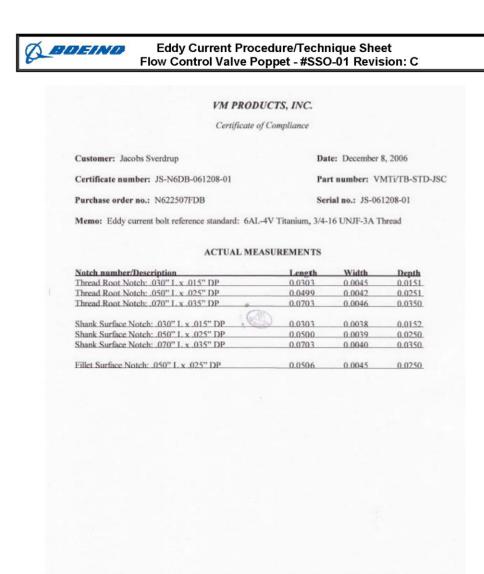


Figure 7: Reference Standard Certification

By Mike Porter Signed:

The above measurements were taken with the use of: optical Comparator ID No. 146011789, and/or micrometer SN 2236-4211, traceable to NIST.

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Figure 8: Reference Standard Positioned on Scanner

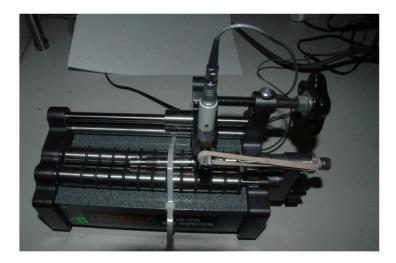


Figure 9: Poppet Positioned for Shank Scanning

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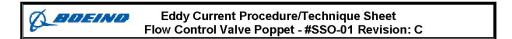
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Figure 10: Poppet Positioned for Radius Scanning

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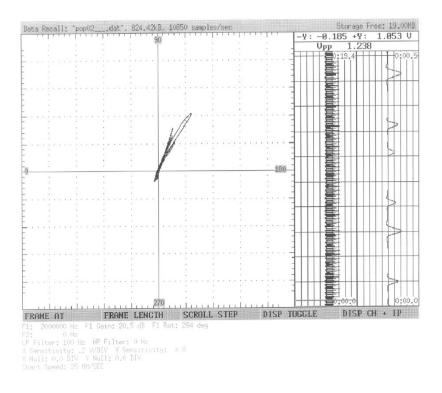


Figure 11: Eddy Current Crack Signal on Poppet

REPORT DOCUMENTATION PAGE

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13. SUPPLEMENTARY NOTES

14. ABSTRACT

The Director of the NASA Engineering and Safety Center (NESC), requested an independent assessment of the anomalous gaseous hydrogen (GH2) flow incident on the Space Shuttle Program (SSP) Orbiter Vehicle (OV)-105 during the Space Transportation System (STS)-126 mission. The main propulsion system (MPS) engine #2 GH2 flow control valve (FCV) LV-57 transition from low towards high flow position without being commanded. Post-flight examination revealed that the FCV LV-57 poppet had experienced a fatigue failure that liberated a section of the poppet flange. The NESC assessment provided a peer review of the computational fluid dynamics (CFD), stress analysis, and impact testing. A probability of detection (POD) study was requested by the SSP Orbiter Project for the eddy current (EC) nondestructive evaluation (NDE) techniques that were developed to inspect the flight FCV poppets. This report contains the Appendices to the main report.

15. SUBJECT TERMS

Computational fluid dynamics; Nondestructive evaluation; NASA Engineering and Safety Center; Flow control valve; Anomalous gaseous hydrogen flow; Poppet

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