In another prior approach — the one leading to the present improved CSPEs — one utilizes a highly-Li<sup>+</sup>-conductive inorganic filler material to increase the effective Li<sup>+</sup>-conductivity of the solid electrolyte. In prior CSPE formulations following this approach, the highly-Li<sup>+</sup>-conductive fillers have been in the form of particles. It has been found that Li<sup>+</sup>-ion conductivity can be increased to about 10<sup>4</sup> S/cm by use of particles, but that the potential for any further increase is limited by the inherently restrictive nature of contacts between particles.

In contrast, in a CSPE of the present type, interparticle contact or the lack thereof is no longer an issue. In a typical application, a CSPE is formed as a film. The highly-Li<sup>+</sup>-conductive fibers can penetrate the entire thickness of the film and can thereby effectively constitute a relatively-long-distance Li<sup>+</sup>-transfer tunnel. The Li<sup>+</sup>-ion conductivity of the film as a whole is thus increased substantially beyond that achievable by use of particles. For example, the figure presents results of conductivity measurements on two CSPEs made from a PEO-LiN( $SO_2CF_2CF_3$ )<sub>2</sub> polymer electrolyte filled with 20 weight percent of the highly-Li<sup>+</sup>-conductive compound La<sub>0.55</sub>Li<sub>0.35</sub>TiO<sub>3</sub>.

An additional advantage of using filler fibers is that mechanical properties of the resulting CSPEs are superior to those attainable by use of particle fillers. One disadvantage — at the present state of development — is that relative to particles, fibers are less effective for interface stabilization and trapping of impurities. In contemplated further development, it may be pos-

sible to overcome this disadvantage by reducing fiber diameters to the order of nanometers. Other avenues of development could include selection of fiber materials having greater Li<sup>+</sup>-ion conductivities and finding ways to arrange fibers in velvetlike mats to maximize through-the-thickness conductivities.

This work was done by A. John Appleby, Chunsheng Wang, and Xiangwu Zhang of Texas A&M University for Glenn Research Center. Further information is contained in a TSP (see page 1).

Inquiries concerning rights for the commercial use of this invention should be addressed to NASA Glenn Research Center, Commercial Technology Office, Attn: Steve Fedor, Mail Stop 4–8, 21000 Brookpark Road, Cleveland, Ohio 44135. Refer to LEW-17470-1.

## **Electrically Conductive Anodized Aluminum Surfaces**

These coatings are highly adherent, transparent, and relatively inexpensive.

Marshall Space Flight Center, Alabama

Anodized aluminum components can be treated to make them sufficiently electrically conductive to suppress discharges of static electricity. The treatment was conceived as a means of preventing static electric discharges on exterior satin-anodized aluminum (SAA) surfaces of spacecraft without adversely affecting the thermal-control/optical properties of the SAA and without need to apply electrically conductive paints, which eventually peel off in the harsh environment of outer space. The treatment can also be used to impart electrical conductivity to anodized housings of computers, medical electronic instruments, telephoneexchange equipment, and other terrestrial electronic equipment vulnerable to electrostatic discharge.

The electrical resistivity of a typical anodized aluminum surface layer lies between  $10^{11}$  and  $10^{13}$   $\Omega$ ·cm. To suppress electrostatic discharge, it is necessary to reduce the electrical resistivity significantly — preferably to  $\leq 10^9 \ \Omega \cdot \text{cm}$ . The present treatment does this. The treatment is a direct electrodeposition process in which the outer anodized surface becomes covered and the pores in the surface filled with a transparent, electrically conductive metal oxide nanocomposite. Filling the pores with the nanocomposite reduces the transverse electrical resistivity and, in the original intended outer-space application, the exterior covering portion of the nanocomposite would afford the requisite electrical contact with the outer-space plasma.

The electrical resistivity of the nanocomposite can be tailored to a value between  $10^7$  and  $10^{12} \,\Omega$ -cm. Unlike electrically conductive paint, the nanocomposite becomes an integral

part of the anodized aluminum substrate, without need for adhesive bonding material and without risk of subsequent peeling. The electrodeposition process is compatible with commercial anodizing production lines.

At present, the electronics industry uses expensive, exotic, electrostatic-discharge-suppressing finishes: examples include silver impregnated anodized, black electroless nickel, black chrome, and black copper. In comparison with these competing finishes, the present nanocomposite finishes are expected to cost 50 to 20 percent less and to last longer.

This work was done by Trung Hung Nguyen of EIC Laboratories for Marshall Space Flight Center. Further information is contained in a TSP (see page 1). MFS-32092-1