

Phenomena and Material Property Requirements for a Combined Structural and Thermal Ablation Model

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Logan, Utah



Applications

Phenomena

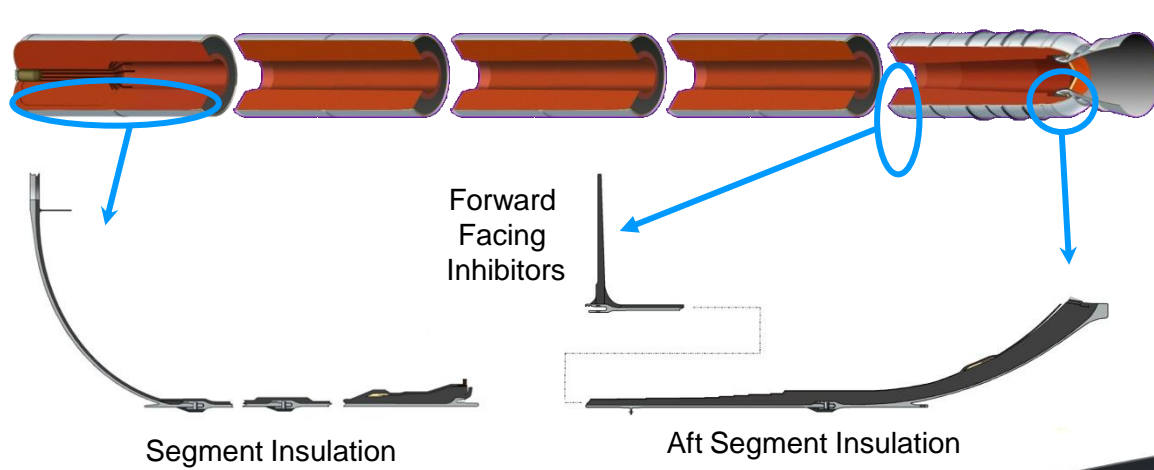
Mathematical models

ATK code development (ITRAC and Hero)

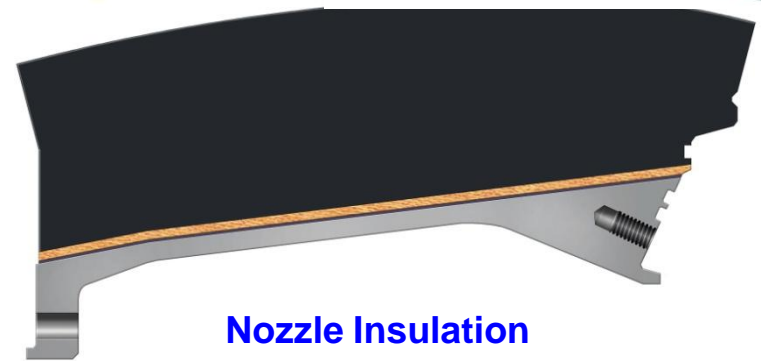
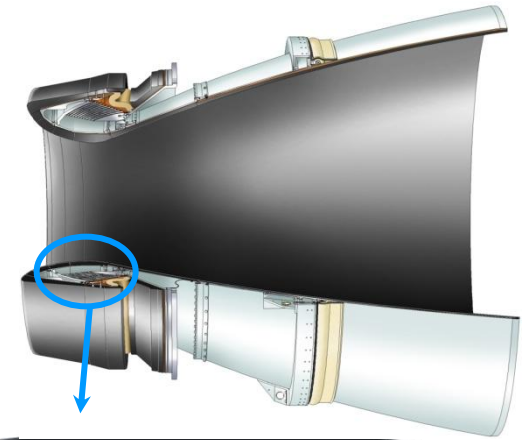
Thermal property requirements

Structural considerations

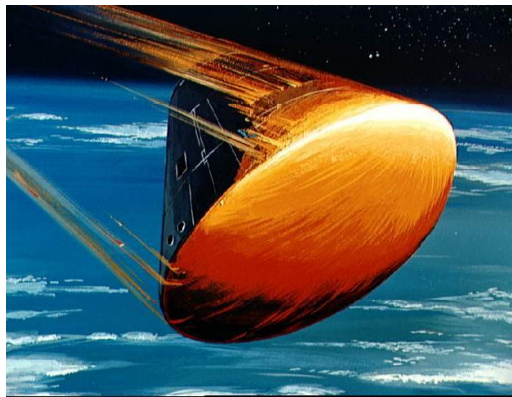
On-going work



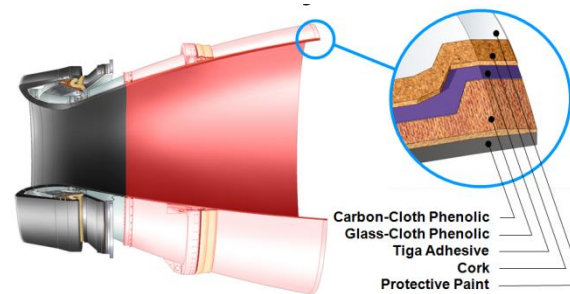
Internal Insulation



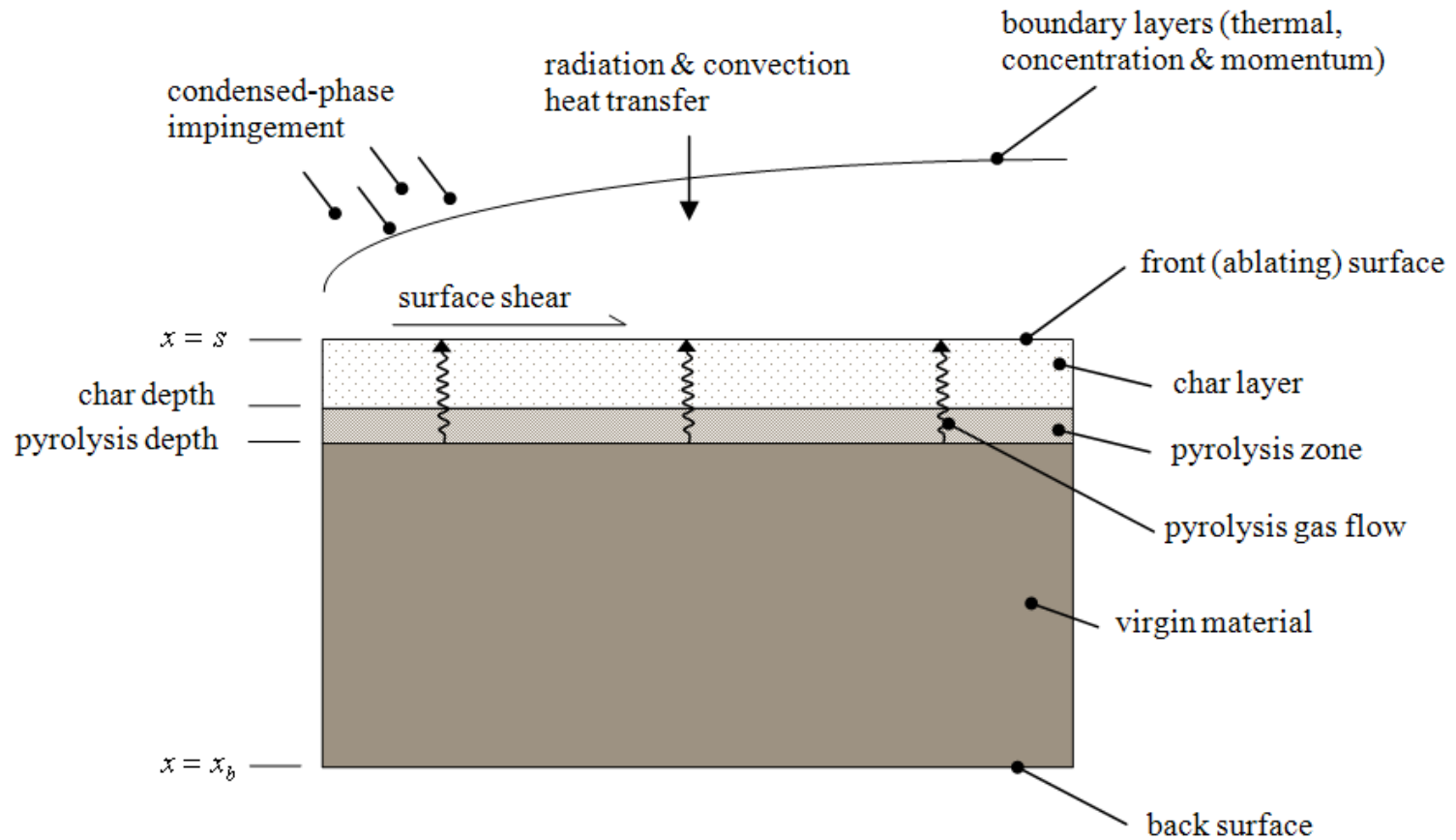
Nozzle Insulation



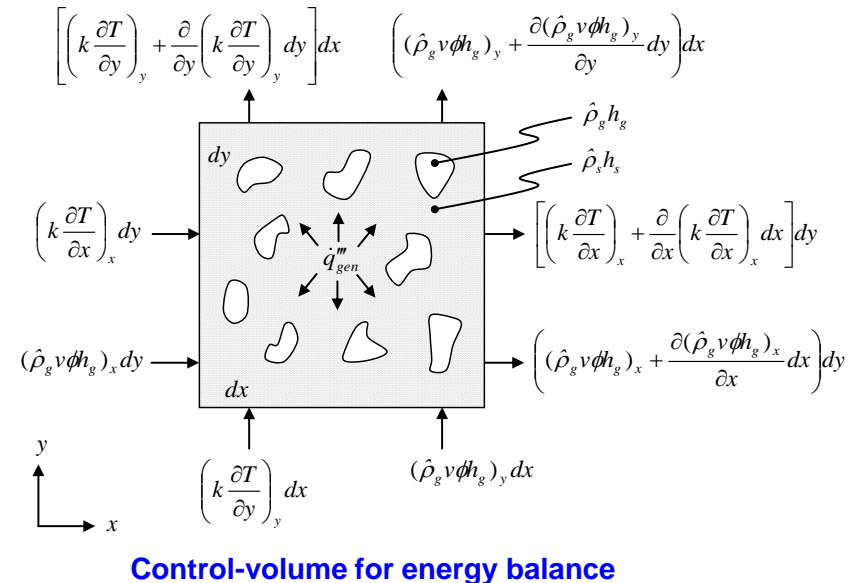
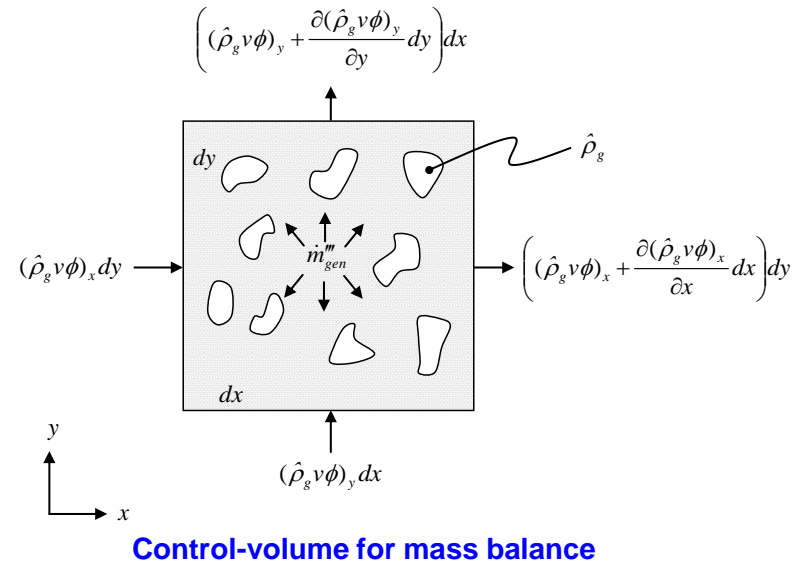
Re-entry Thermal Protection Systems



External Insulation



- **Mass balance (gas phase)**
 - Solid to gas conversion (pyrolysis)
 - Gas advection (permeation)
 - Storage (in the internal pores)
- **Momentum balance**
 - Balances friction with pressure gradient (Darcy's model)
- **Energy balance**
 - Conduction (in solid phase)
 - Storage (in solid and gas phases)
 - Pyrolysis energy
 - Advected energy (heat exchange with pyrolysis gases)



• Pyrolysis

$$\frac{\partial \rho_s}{\partial t} = -(\rho_v - \rho_c) \frac{\partial \alpha}{\partial t}$$

rate of solid density change

$$\alpha = \frac{\sum_i x_i \alpha_i}{\sum_i x_i}$$

overall versus component extent-of-reaction

$$\frac{\partial \alpha}{\partial t} = \frac{\sum_i \frac{\partial \alpha_i}{\partial t} x_i}{\sum_i x_i}$$

overall pyrolysis rate

$$\frac{d\alpha_i}{dt} = A_i e^{\left(\frac{-E_{a,i}}{RT}\right)} (1 - \alpha_i)^{m_i}$$

Arrhenius model for component extent-of-reaction

• Mass/Momentum Equation

$$(\rho_v - \rho_c) \frac{\partial \alpha}{\partial t} + \nabla \cdot \left(\frac{\hat{\rho}_g}{\mu_g} \Gamma \nabla P \right) - \hat{\rho}_g (\phi_c - \phi_v) \frac{\partial \alpha}{\partial t} - \frac{\phi \hat{\rho}_g}{k_g} \frac{\partial P}{\partial t} + \phi \beta_g \hat{\rho}_g \frac{\partial T}{\partial t} = 0$$

generation advection (permeation) storage

– Neglected storage (quasi-steady)

$$(\rho_v - \rho_c) \frac{\partial \alpha}{\partial t} + \nabla \cdot \left(\frac{\hat{\rho}_g}{\mu_g} \Gamma \nabla P \right) = 0$$

generation advection (permeation)

– 1-D simplification (with neglected storage)

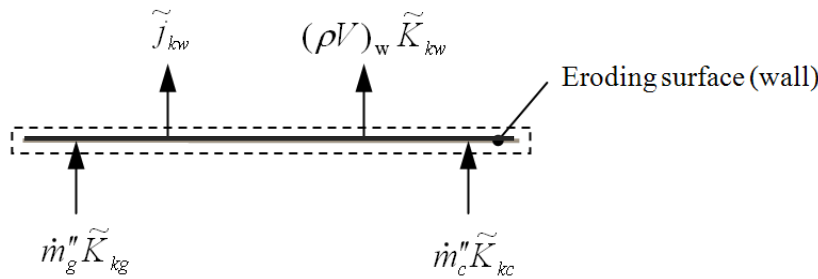
$$\dot{m}_g''(x_p) = -\frac{1}{A} \int_{x_p}^{x_b} A \frac{\partial \rho_s}{\partial t} dx$$

• Energy Equation

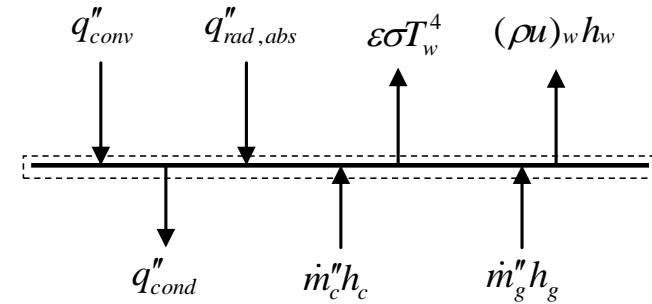
$$(Q_s - h_s + h_g)(\rho_v - \rho_c) \frac{\partial \alpha}{\partial t} + \hat{\rho}_g \phi \frac{\partial h_g}{\partial t} + \rho_s \frac{\partial h_s}{\partial t} + \hat{\rho}_g \mathbf{v}_D \cdot \nabla h_g - \nabla \cdot \mathbf{K} \nabla T = 0$$

pyrolysis energy storage advection conduction

Solution of M/M equation primarily needed for solution of energy equation. Analyst can choose level of fidelity.



Control-surface for elemental balance



Control-surface for energy balance

- **Unity Lewis number**

$$q_{cond}'' = \rho_e u_e C_H [H_r - (1+B')h_w + B'_c h_c + B'_g h_g] + \alpha_w q_{rad,inc}'' - \epsilon \sigma T_w^4$$

- **Unequal diffusion coefficients**

$$q_{cond}'' = \rho_e u_e C_H (H_r - h_w)_{f.e.g} + \rho_e u_e C_M \left[\left(\sum_i K_{ie} - \sum_i K_{iw} \right) h_i^{T_w} + B'_c h_c + B'_g h_g - B' h_w \right] + \alpha_w q_{rad,inc}'' - \epsilon \sigma T_w^4$$

- **Equal diffusion coefficients**

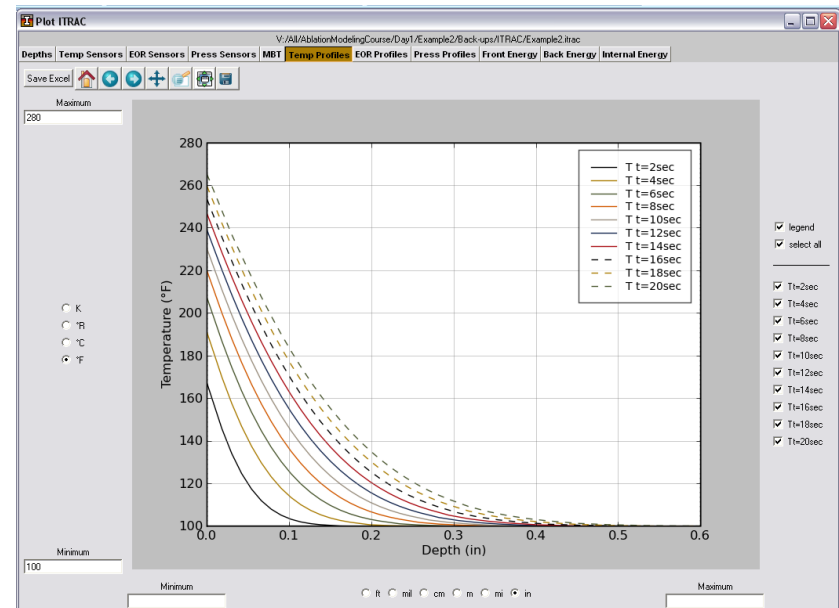
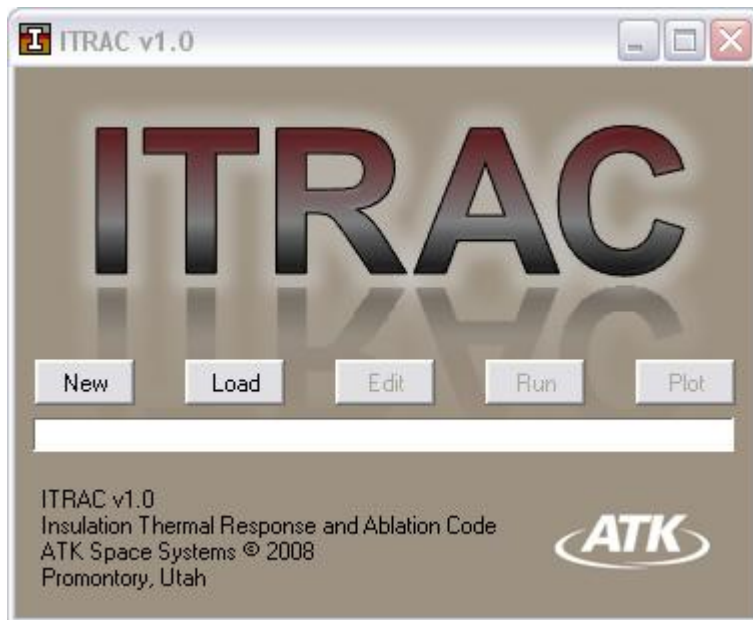
$$q_{cond}'' = \rho_e u_e C_H (H_r - h_w)_{f.e.g} + \rho_e u_e C_M \left[\left(\sum_i Z_{ie}^* - \sum_i Z_{iw}^* \right) h_i^{T_w} + B'_c h_c + B'_g h_g - B' h_w \right] + \alpha_w q_{rad,inc}'' - \epsilon \sigma T_w^4$$

- **Surface ablation rate**

$$\dot{s}_{chem} = \frac{\dot{m}_c''}{\rho_c} = \frac{B'_c \rho_e u_e C_M}{\rho_c}$$

Thermochemistry
("b-prime") tables
from ACE code

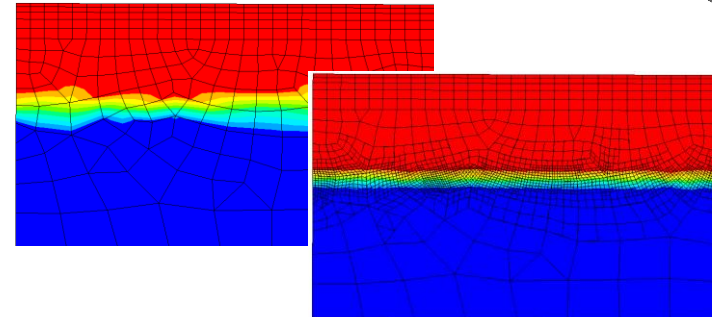
- 1-D (planar, cylindrical, and spherical)
- Variable-grid finite-volume method
- Heat transfer, material pyrolysis, pore pressure, thermochemical ablation
- Various mechanical erosion models
- User-defined dynamic link libraries (DLLs)
- Ignition model



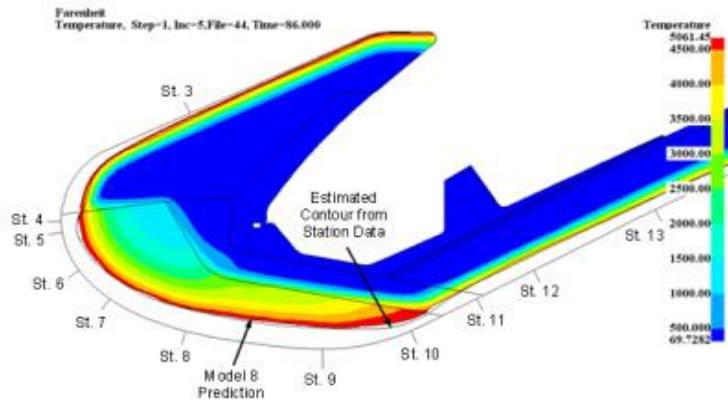
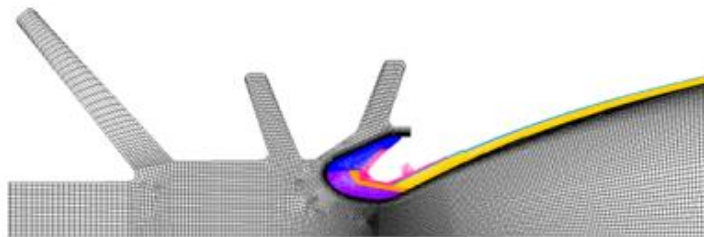
Hero (Heat Transfer and Erosion Analysis Program)

A premier aerospace and defense company

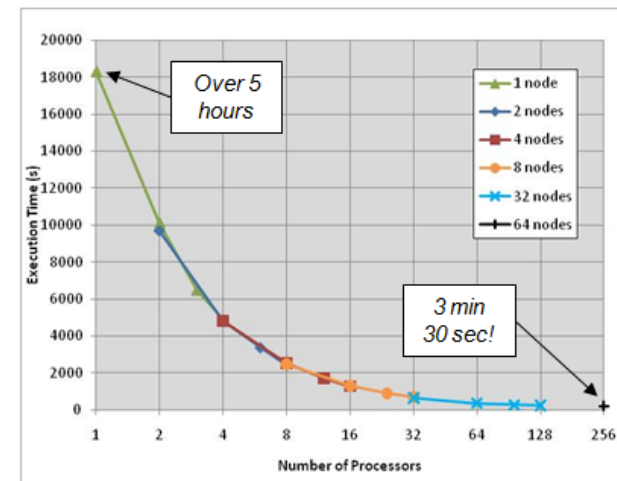
- Multi-D
- Variable-grid finite-element (hierarchical) method
- Heat transfer, material pyrolysis, pore pressure, thermochemical ablation, radiosity, structural
- Fluid-thermal-structural interaction (FTSI) capabilities (FEM Builder)
- Adaptive refinement
- Parallel processing



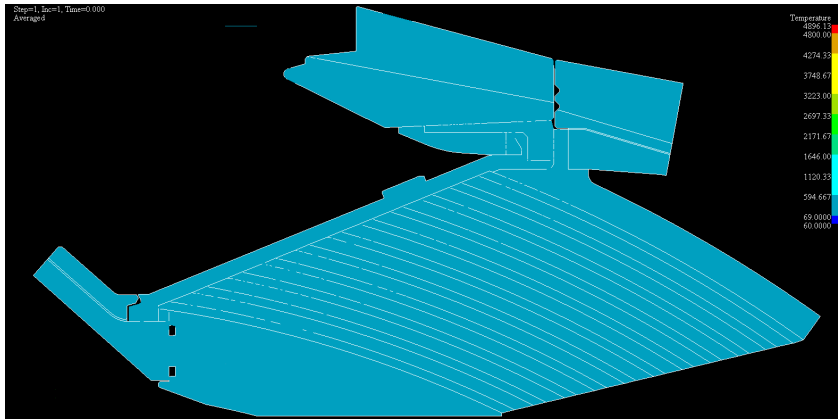
Hero Adaptive Refinement



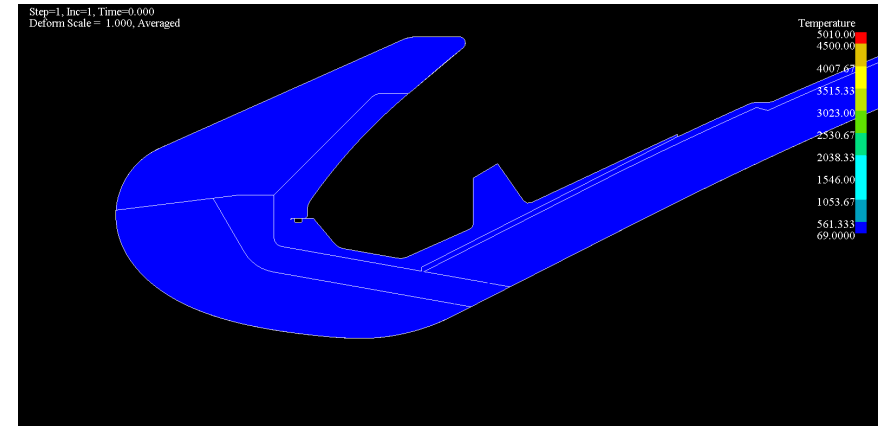
FTI Nozzle Analysis



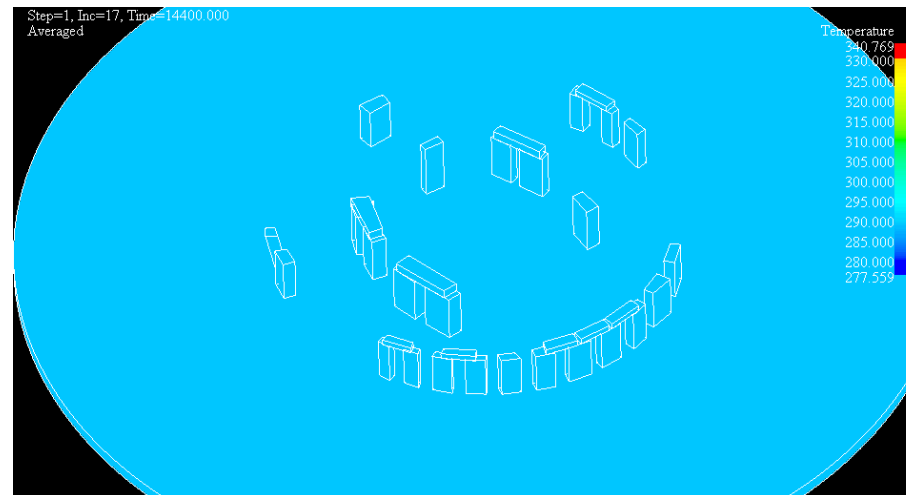
Run-Time Improvement with Parallel Processing



Modeling of Complex Ablation Scenario



Nozzle Thermal Analysis Results



Radiosity with Solar Effects

Pyrolysis Properties

$$\frac{\partial \rho_s}{\partial t} = -(\rho_v - \rho_c) \frac{\partial \alpha}{\partial t}$$

rate of solid density change

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overall versus component extent-of-reaction

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– 1-D simplification (with neglected storage)

$$\dot{m}_g''(x_p) = -\frac{1}{A} \int_{x_p}^{x_b} A \frac{\partial \rho_s}{\partial t} dx$$

Energy Equation

$$(Q_s - h_s + h_g)(\rho_v - \rho_c) \frac{\partial \alpha}{\partial t} + \hat{\rho}_g \phi \frac{\partial h_g}{\partial t} + \rho_s \frac{\partial h_s}{\partial t} + \hat{\rho}_g \mathbf{v}_D \cdot \nabla h_g - \nabla \cdot \mathbf{K} \nabla T = 0$$

pyrolysis energy storage advection conduction

Insulator Properties

Pyrolysis kinetics parameters

$$x_i, E_i, A_i, m_i$$

Virgin and char densities

$$\rho_v, \rho_c$$

$$\rho_s = \rho_v(1 - \alpha) + \rho_c \alpha$$

Virgin and char specific heat

$$c_{p,v}, c_{p,c}$$

$$h_v = \int_{T_{ref}}^T c_{p,v} dt \quad h_c = \int_{T_{ref}}^T c_{p,c} dt$$

$$h_s = h_v(1 - \alpha) + h_c \alpha$$

Virgin and char conductivity

$$k_v, k_c$$

$$k_s = k_v(1 - \alpha) + k_c \alpha$$

Elemental compositions (v & c)

$$\overset{\text{equilibrium analysis}}{\bar{K}_{k,v}, \bar{K}_{k,c}} \rightarrow h_g, MW_g$$

Heats of formation

$$h_{f,v}^o, h_{f,c}^o \rightarrow Q_s$$

$$q''_{cond} = \rho_e u_e C_H \left[(H_r - h_w)_{f.e.g.} + \frac{C_M}{C_H} \sum_i (K_{i,e} - K_{i,w}) h_i^{T_w} + B'_g h_g + B'_c h_c - B' h_w \right] + \alpha q''_{rad,inc} - \epsilon \sigma T_w^4$$

Boundary Conditions

- Transport coefficient
- Enthalpy (recovery and frozen)
- Incident radiation heat flux

Propellant Properties

- Stanton number ratio
- Elemental composition

Surface Product Properties

- Elemental composition
- Enthalpy of products

Insulator Properties

- Char enthalpy (specific heat) and pyrolysis gas enthalpy
- Radiation properties (emissivity and absorptivity)

Pyrolysis kinetic parameters

Density

- Virgin and char

Specific heat

- Virgin and char
- Versus temperature

Thermal conductivity

- Virgin and char
- Versus temperature

Elemental compositions

- Virgin and char
- Pyrolysis gas calculated

Heats-of-formation

- Virgin and char
- Pyrolysis gas value calculated
- Used to calculate heat-of-pyrolysis
- Used in surface thermochemistry

Radiation properties

- Emissivity and absorptivity

For pore pressure

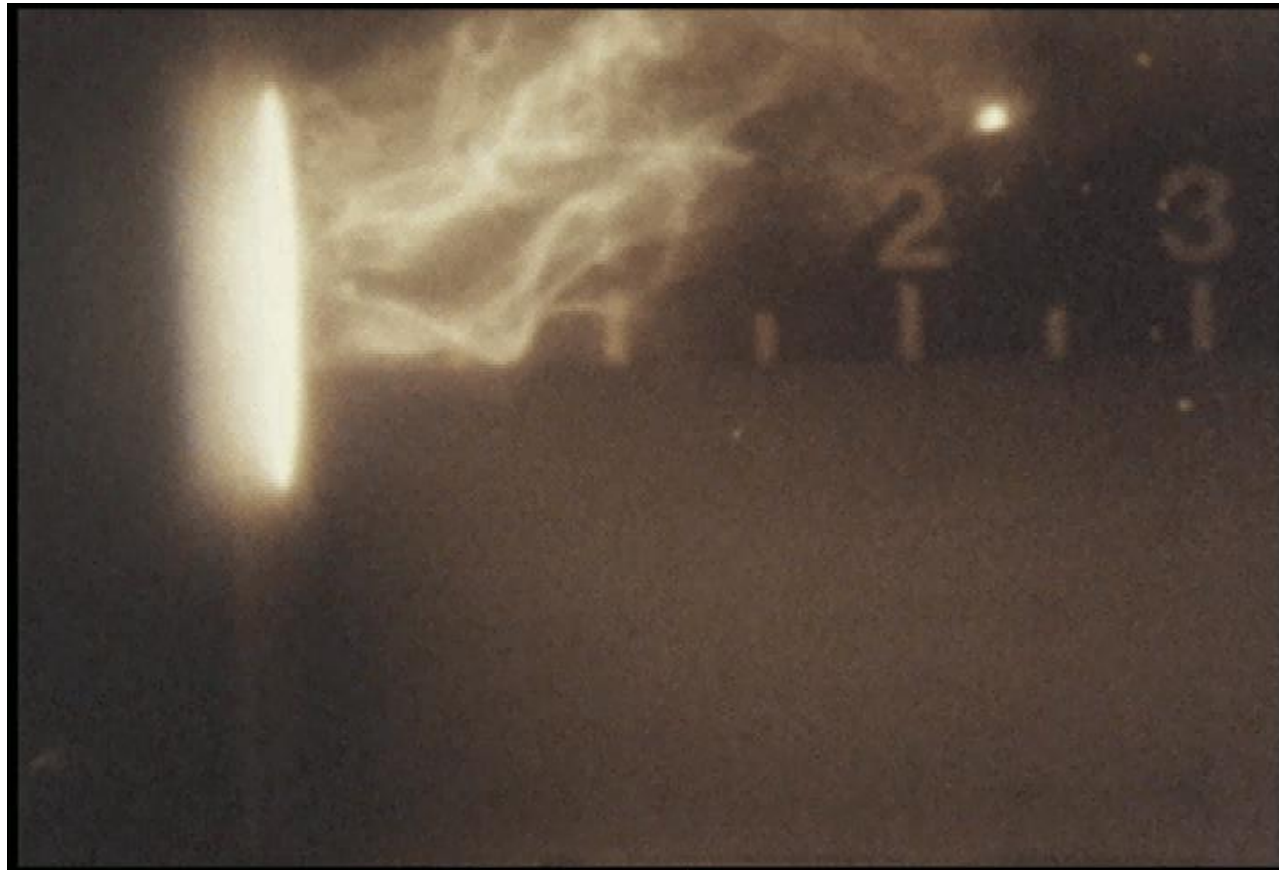
- Porosity
- Permeability
- Pyrolysis gas molecular weight
- Pyrolysis gas viscosity

Structural modeling is often required for accurate assessment of structural integrity (including phenomena such as pocketing, ply-lifting, wedge-outs, and delamination)

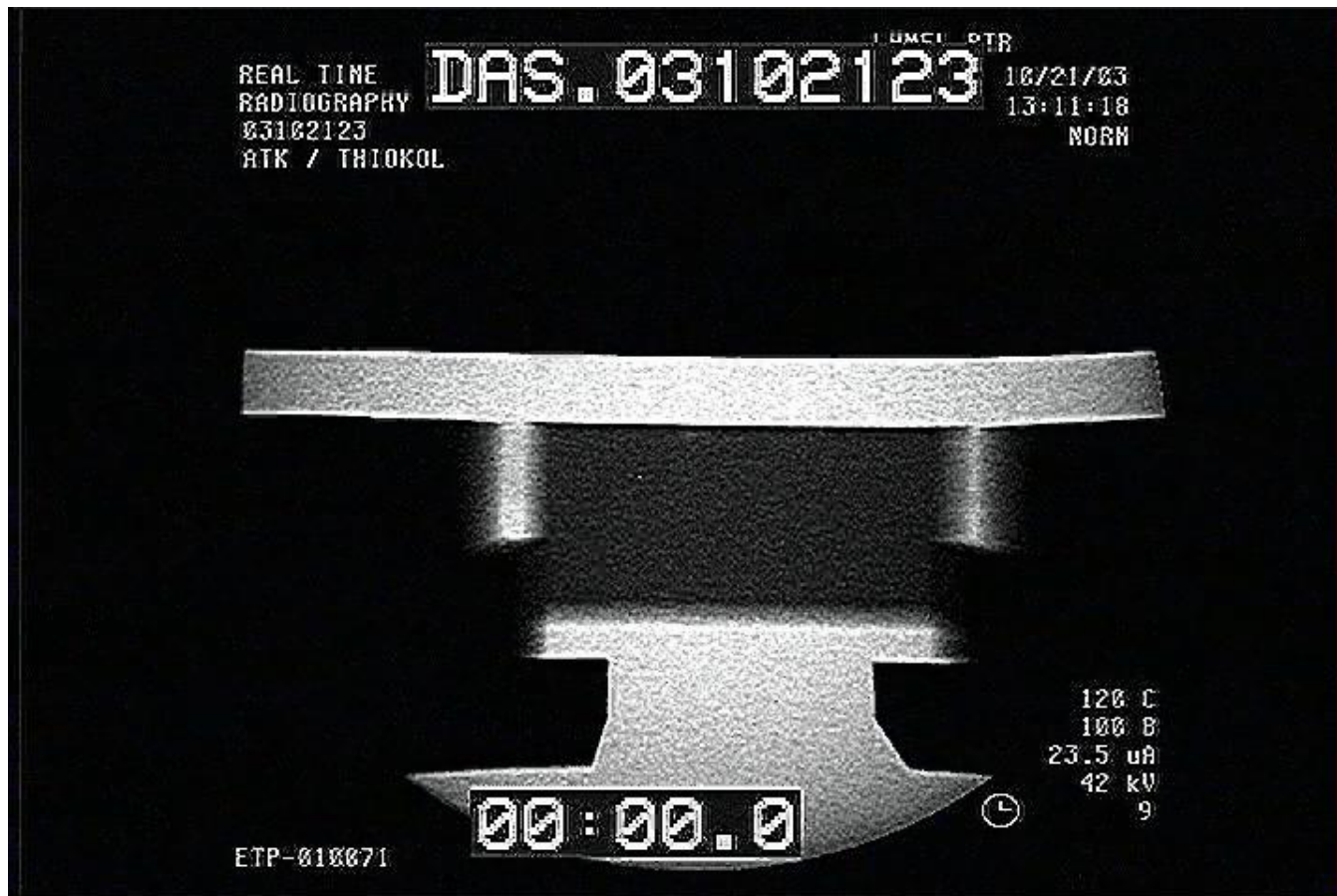
- Phenomena can have significant impact on thermal protection
- Integrally tied to thermal responses
- Accurate modeling of heat transfer, material pyrolysis, pore pressure, thermochemical ablation, etc. is critical

Pocketing involves the expulsion of carbon-cloth phenolic (CCP) material due to a combination of stresses from thermal expansion and pore pressure driven stresses

- Can significantly affect thermal protection



Ply lifting involves the lifting of a char cap of CCP material. Initial failure is caused by thermal expansion/contraction, lifting is caused by pore pressures



Structural – High Temperature Issues



A premier aerospace and defense company

Ply lifting char caps can slough, causing loss of thermal protection

-3.103 sec
Pre-fire

SRTM NC-6 24" Motor Nozzle Real-Time Radiography, 9/14/2006
Marshall Space Flight Center NDE Team/EM20

Nozzle o.d.

Nozzle i.d.

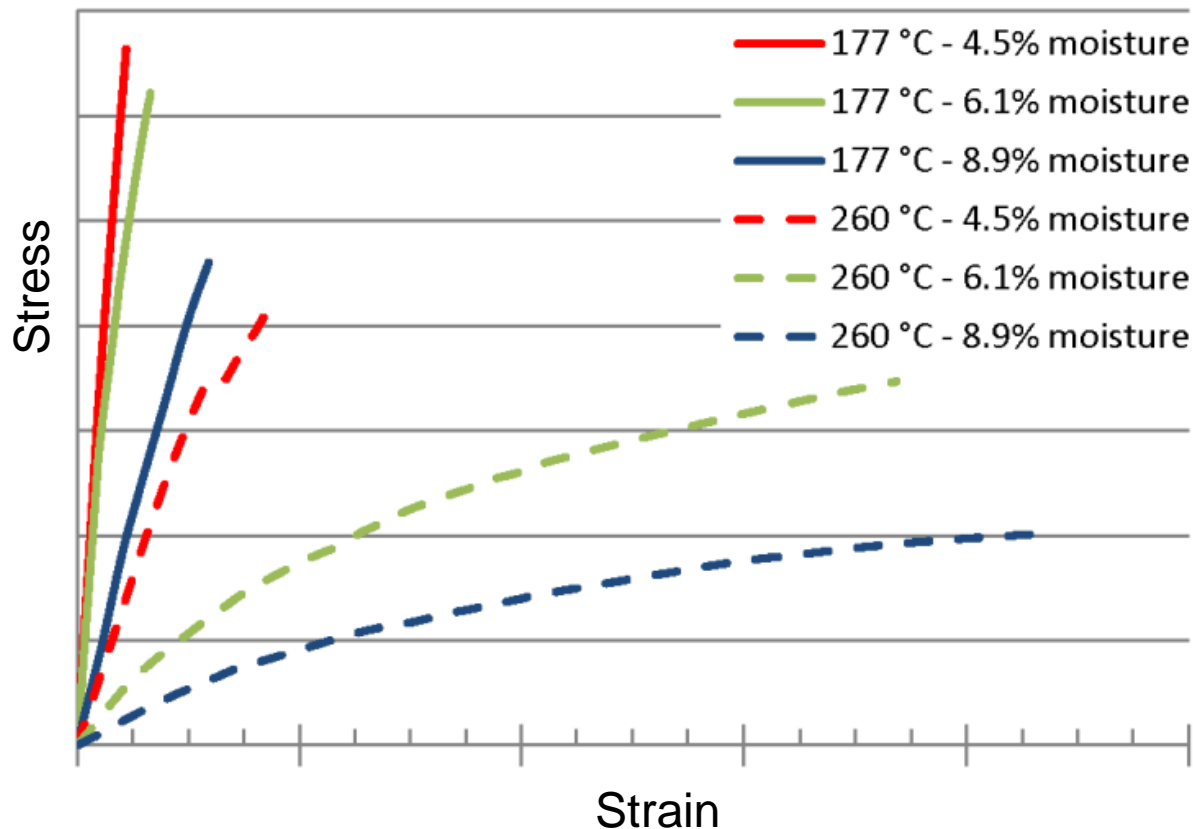
Aft

Forward



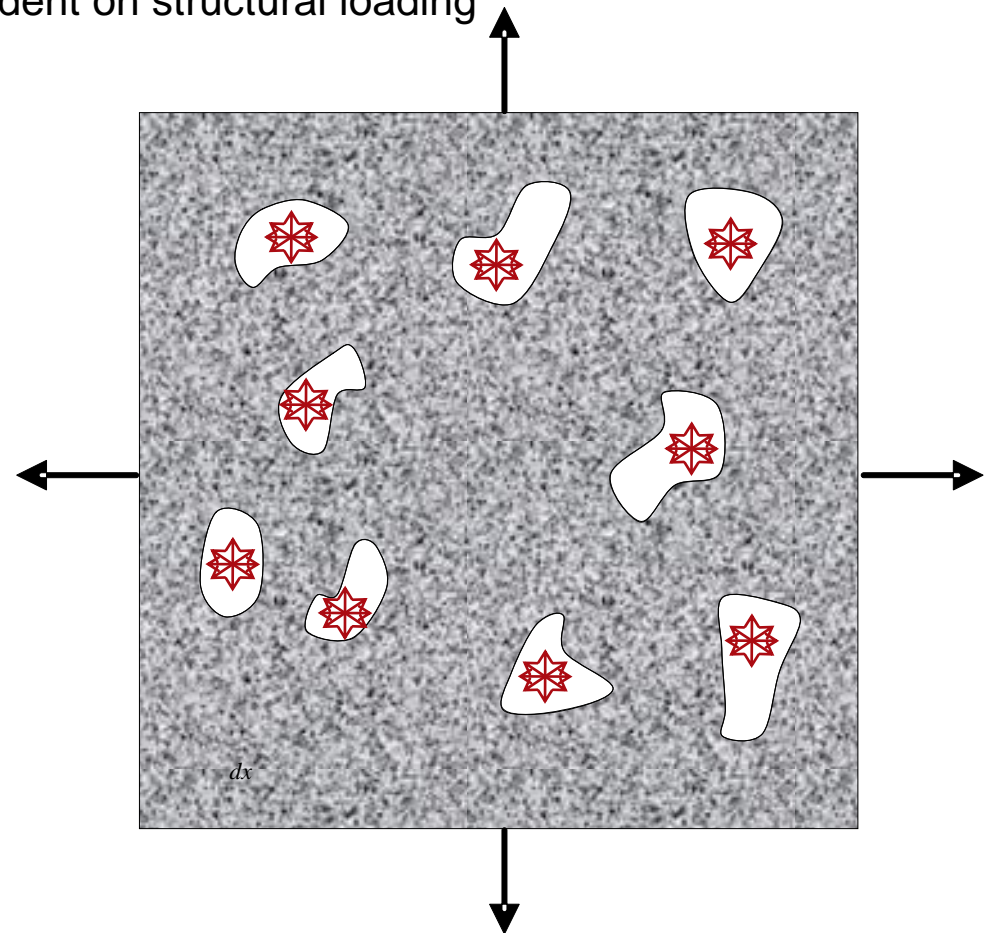
Structural modeling is dependent on thermal modeling, for example:

- Stiffness can be dependent on degree-of-char and presence of moisture
 - Charring causes material changes that affect the magnitude of the modulus
 - Moisture causes a plasticization of the matrix material



Structural modeling is dependent on thermal modeling, for example:

- Structural behavior is also influenced by pressures
- Pressure magnitudes are highly dependent on structural loading
 - Permeability is affected by load
 - Pressure is affected by permeability



The following is a simple first principles equation that may be used for the structural model (complexity is usually added to address nonlinearities)

$$\varepsilon_{ij} = \underbrace{S_{ijkl} \sigma_{kl}^m}_{\text{stress loading}} + \underbrace{\eta_{ij} \Delta P}_{\text{pore pressures}} + \underbrace{\alpha_{ij} \Delta T}_{\text{thermal expansion}} + \underbrace{\beta_{ij} \Delta M}_{\text{moisture swelling}} + \underbrace{\chi_{ij} \Delta v_c}_{\text{decomposition}}$$

Material properties summary (dependent on thermal state)

Stiffness matrix (includes moduli and Poisson's ratios) S_{ijkl}

Pressure stress coupling, η_{ij}

Coefficient of thermal expansion, α_{ij}

Moisture expansion coefficient, β_{ij}

Decomposition coefficient, χ_{ij}

Failure criteria

Conjugate models are required for accurate simulation of thermal and structural behavior

Thermal

- 3-D surface ablation in Hero
- General thermal contact in Hero
- Advanced surface thermochemistry code (ACE replacement)
 - Alumina impingement
 - Chemical kinetics
- Material properties
- Comprehensive validations against historical nozzle data

Flow modeling

Advanced carbon-cloth phenolic CCP modeling

- Moisture
- Structural coupling

- Kendall, R. M., and Bartlett, E. P., Rindal, R. A., and Moyer, C. B. "An Analysis of the Coupled Chemically Reacting Boundary Layer and Charring Ablator: Part I, Summary Report," NASA CR-1060, Washington, D. C., June 1968.
- Moyer, C. B., and Rindal, R. A., "An Analysis of the Coupled Chemically Reacting Boundary Layer and Charring Ablator: Part II, Finite Difference Solution for the In-depth Response of Charring Materials Considering Surface Chemical and Energy Balances," NASA CR-1061, Washington, D. C., June 1968.
- Kendall, R. M., Rindal, R. A., and Bartlett, E. P., "A Multicomponent Boundary Layer Chemically Coupled to an Ablating Surface," *AIAA Journal*, Vol. 5, No. 6, June 1967.
- Blackwell, B. F., "Numerical Prediction of One-Dimensional Ablation Using a Finite Control-Volume Procedure with Exponential Differencing," *Numerical Heat Transfer*, Vol. 14, 1988.
- Blackwell, B. F., and Hogan, R. E., "One-Dimensional Ablation Using Landau Transformation and Finite Control Volume Procedure," *Journal of Thermophysics and Heat Transfer*, Vol. 8, No. 2, 1994.
- Hogan, R. E., Blackwell, B. F., and Cochran, R. J., "Application of Moving Grid Control Volume Finite Element Method to Ablation Problems," *Journal of Thermophysics and Heat Transfer*, Vol. 10, No. 2, 1996.
- Amar, A. J., Blackwell, B. F., and Edwards, J. R., "One-Dimensional Ablation Using a Full Newton's Method and Finite Control Volume Procedure," *Journal of Thermophysics and Heat Transfer*, Vol. 22, No. 1, January-March 2008.
- Milos, F. S., and Rasky, D. J., "Review of Numerical Procedures for Computational Surface Thermochemistry," *Journal of Thermophysics and Heat Transfer*, Vol. 8, No. 1, January-March 1994.
- Chen, Y.-K., and Milos, F. S., "Ablation and Thermal Response Program for Spacecraft Heatshield Analysis," *Journal of Spacecraft and Rockets*, Vol. 36, No. 3, 1999.
- Chen, Y.-K., and Milos, F. S., "Two-Dimensional Implicit Thermal Response and Ablation Program for Charring Materials on Hypersonic Space Vehicles," AIAA Paper 2000-0206, January 2000.
- Chen, Y.-K., and Milos, F. S., "Three-Dimensional Ablation and Thermal Response Simulation System," AIAA Paper 2005-5064, June 2005.
- Wu, Y., and Katsube, N., "A Thermomechanical Model for Chemically Decomposing Composites – I. Theory," *International Journal of Engineering Science*, Vol. 35, No. 2, 1997.
- Sullivan, R. M. and Stokes, E. H., "A model for the Effusion of Water in Carbon Cloth Phenolic Composites," *Mechanics of Materials*, Vol. 26, 1977.
- Sullivan, R. M., Stokes, E. H., and Baker, E. H., "Effect of Time at Temperature on the X-ply Tensile Modulus of Carbon Cloth Phenolic Composites," NASA/TM-2010-216921.
- Sullivan, R. M. and Salamon, N. J., "A Finite Element Method for the Thermochemical Decomposition of Polymeric Materials - I. Theory," *International Journal of Engineering Science*, Vol. 30, No. 4. 1992.
- Sullivan, R.M., "The Effect of Water on Thermal Stresses in Polymer Composites," *Journal of Applied Mechanics*, Vol. 63, 1996.

Ewing, M. E., and Laker, T. S., “Insulation Thermal Response and Ablation Code, ITRAC Version 1.0, Theory Manual,” ATK Launch Systems Manual, MAN000004, Promontory, UT, March 2009.

Ewing, M. E., Laker, T. S., and P. H. Bauer, “Insulation Thermal Response and Ablation Code, ITRAC Version 1.0, User Manual,” ATK Launch Systems Manual, MAN000005, Promontory, UT, March 2009.

Laker, T. S., and Ewing, M. E., “Insulation Thermal Response and Ablation Code, ITRAC Version 1.0, Verification Manual,” ATK Launch Systems Manual, MAN000006, Promontory, UT, March 2009.

Ewing, M. E., Walker, D. T., Isaac, D. A., Phipps, B. E., and Dewey, H. H., “Heat Transfer and Erosion Analysis Program, Hero 2010, Theory Manual,” ATK Launch Systems, Promontory, UT, 2010.

Isaac, D. A., Dewey, H. H., “Heat Transfer and Erosion Analysis Program, Hero 2010, User Manual,” ATK Launch Systems, Promontory, UT, 2010.

Isaac, D. A., Walker, D. T., Ewing, M. E., and Dewey, H. H., “Heat Transfer and Erosion Analysis Program 2010, Test Manual,” ATK Launch Systems, Promontory, UT, 2010.