



NASA Plum Brook Station Spacecraft Propulsion Research Facility (B-2)

B-2 Facility Capability Summary

Hydrogen/Oxygen Rocket

Test Capability With Vacuum Start

- 30K lbf thrust engine for 400 sec test duration
- 100K lbf thrust engine for 270 sec test duration
- 300K lbf thrust engine for 15 sec test duration

Space Simulation

B–2's test chamber is capable of reproducing space environment temperatures and vacuum.

Vacuum pressure LN₂ chamber temperature Infrared radiation intensity 5x10⁻⁷ torr -320 °F 130 W/ft²

Control and Data Systems

- Test operations are controlled by a fully redundant computerized control system via a remote control room located 2400 ft from the test site.
- A state-of-the-art data acquisition system records all parameters associated with the test. Features include 600 channels at direct current to 1 kHz per channel and a high-speed system capability up to 115 kHz per channel.

Spray Chamber

- 67 ft diameter by 120 ft deep underground chamber holds 1.7 million gallons of chilled water.
- Four 2000-hp water pumps recirculate 224,000 gal of chilled water per minute to condense engine exhaust.
- Three propellant dump tanks are available for enhanced safety.



* All values are approximate and for reference only.

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