

Amine Swingbed Payload Technology Demonstration

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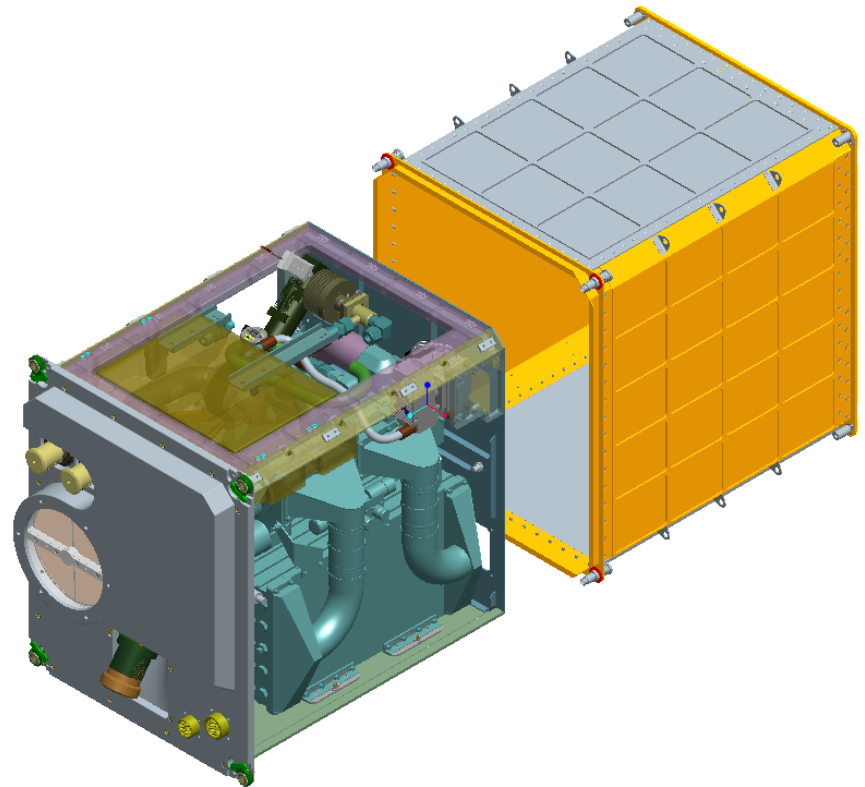
NASA/Johnson Space Center

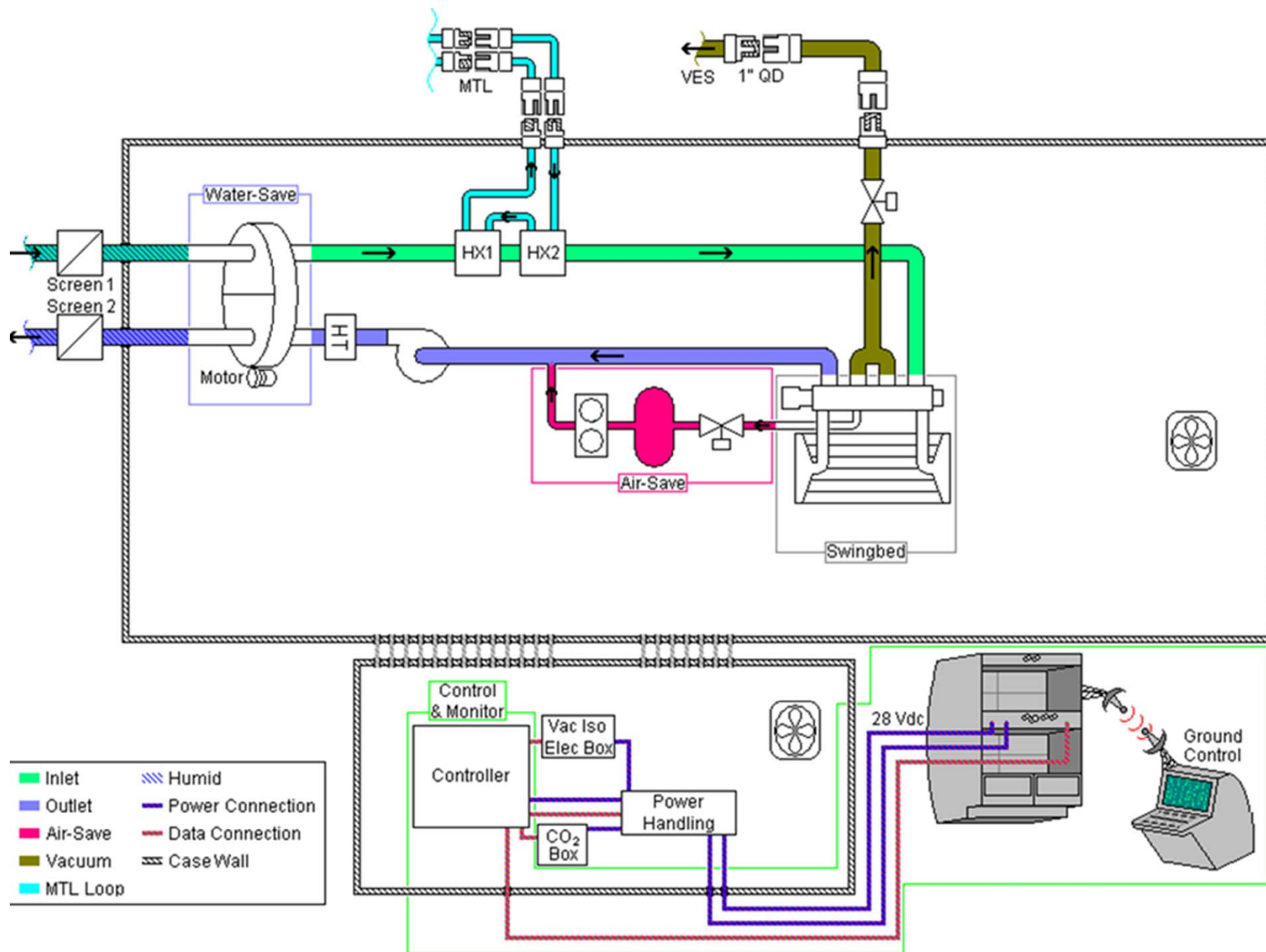
Amine Swingbed Technology

- The Amine Swingbed is an amine-based, vacuum-regenerated adsorption technology for removing carbon dioxide and humidity from a habitable spacecraft environment, and is the baseline technology for the Orion Program's Multi-Purpose Crew Vehicle (MPCV).
 - Uses a pair of interleaved-layer beds filled with SA9T, the amine sorbent.
 - A linear multiball valve rotates 270° back and forth to control the flow of air and vacuum to adsorbing and desorbing beds: one bed adsorbs CO₂ and H₂O from cabin air while the other bed is exposed to vacuum for regeneration by venting the CO₂ and H₂O.
 - The two beds are thermally linked, so no additional heating or cooling is required.
- The technology can be applied to habitable environments where recycling CO₂ and H₂O is not required such as short duration missions.

What is the Amine Swingbed Payload?

- The Amine Swingbed Payload consists of the swingbed itself, a desiccant wheel (to reduce water losses), an air-save tank and compressor (to reduce air losses), a blower, an air/liquid heat exchanger, a heater, various sensors, and a controller/power handler.
- Located in ExPRESS Rack 8 in the US Lab
- Controlled from the ground at JSC's Payload Operations Control Center (POCC).





Technology Maturation for Orion

- Ground tests of the Amine Swingbed technology provide for:
 - detailed performance data
 - reduces technology risk
 - integrated technology evaluations (multiple components together in one test)
- Test conducted at JSC is more complicated and more integrated with other ARS systems than previous tests.
 - 2006/07/08: unmanned tests
 - 2008: human test subjects
 - 2009: reduced pressure, elevated oxygen
 - 2011: suited test subjects

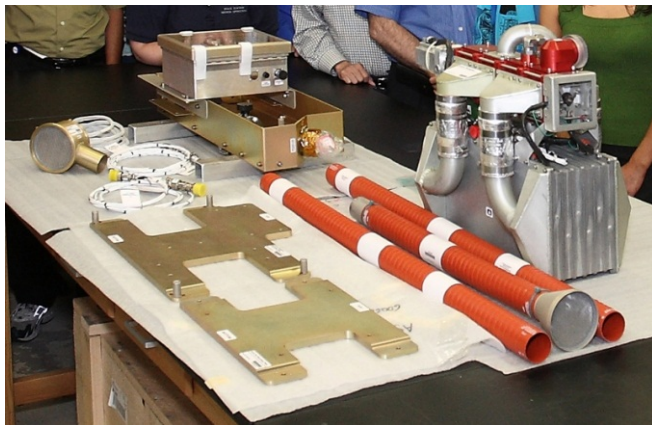


Orion technology demonstration on ISS

- Orion's Multi-Purpose Crew Vehicle (MPCV) is designed for short-duration missions where conserving resources (especially H₂O) is not critical, but maintaining low levels of CO₂ and comfortable humidity levels is.
- However, it is important for ISS to limit the amount of H₂O and air that are vented overboard because it is a long duration mission where resources are recycled as much as practical.
- To reduce H₂O and air losses while testing the Amine Swingbed on the ISS, new subsystems were developed, based on proven technologies, and were integrated with the Amine Swingbed.
 - Water Save subsystem
 - Air Save subsystem
- Using the ISS environment to test these subsystems integrated with the Amine Swingbed helps reduce risk for these technologies for potential future exploration missions.

On-Orbit Payload Integration

- Due to logistical constraints (mass, volume), the Payload was delivered to ISS in two Phases
 - Phase A delivered Feb 2011
 - Phase B delivered July 2011
- On orbit was the first time that the Phase A equipment physically mated with the Phase B equipment.
 - Crew feedback was there was less than 1 mm of clearance. (Whew!)
- But no payload experience is complete without having to troubleshoot something, so



Anomaly and troubleshooting history

- ☹ GMT 2011/342: ISS Fire Alarm went off
- ☹ GMT 2012/013: Amine Swingbed valve stopped moving. Incorrect wiring = blown fuse.
- ☹ GMT 2012/041: Initial on-orbit electrical measurements indicated motor was wired wrong.
- ☹ GMT 2012/102: Motor rewire attempted, but visual indication and crewmember confirmed wiring was correct
- 😊 GMT 2012/159, 160: Installed new fuse on control board, replaced motor/gearbox assembly



Anomaly and troubleshooting history

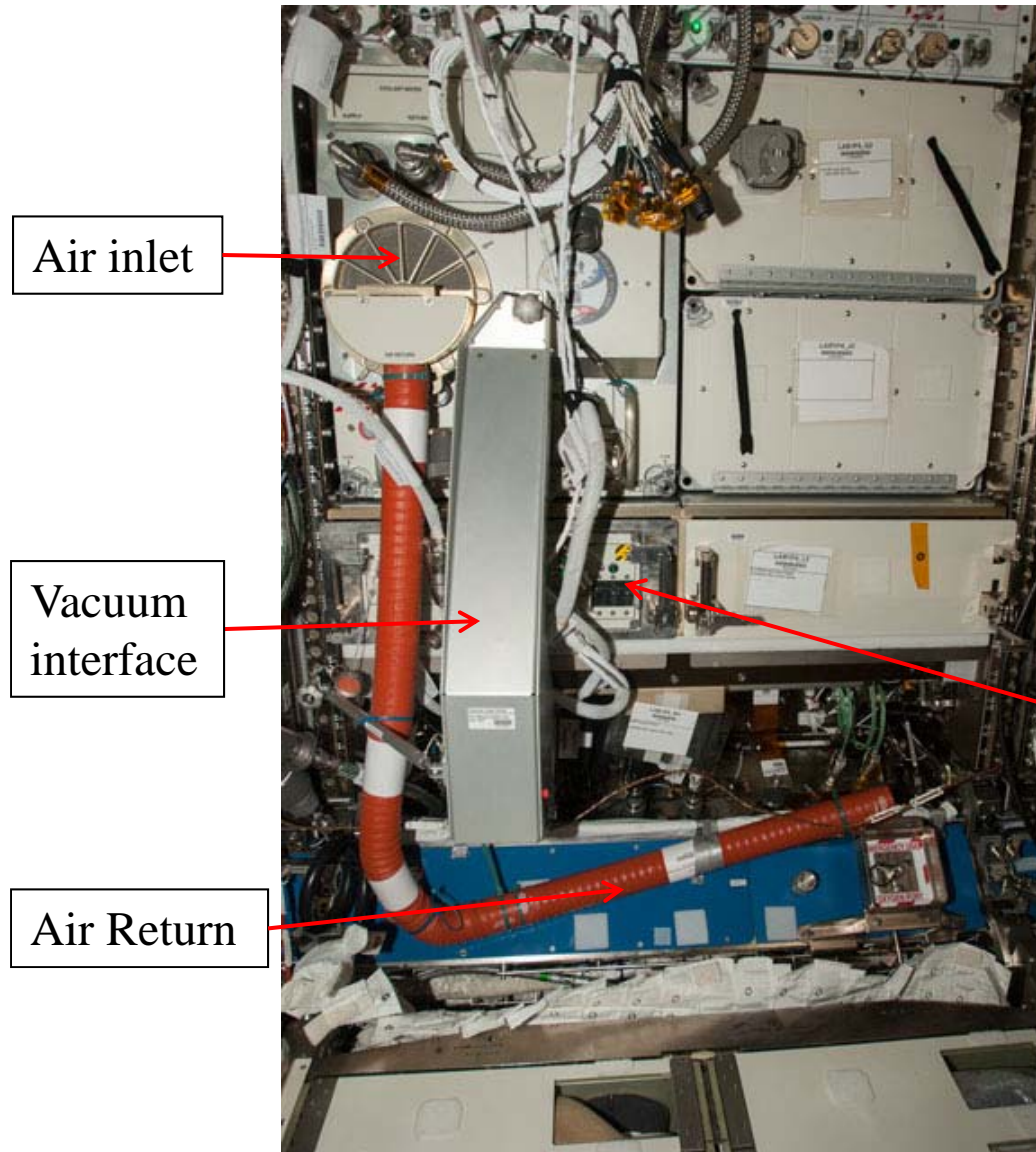
- 😊 GMT 2012/172: Final installation of payload into Express Rack 8 – successful.
- ☹ GMT 2012/175: Attempted Test Point 1. Proximity sensor failed to provide confirmation of complete rotation of valve. Mechanical cause: Set screws used to secure drive key inside gearbox were insufficiently torqued.
- ☹ GMT 2013/030: Torqued the set screws on the drive key, but valve failed to rotate properly, then stopped altogether. Possible cause: rotary valve failure, gearbox failure, motor failure.
- 😊 GMT 2013/059, 060: Evaluated valve torque resistance and replaced suspect gearbox with original gearbox – successful.
- 😊 GMT 2013/079: Payload installed into Express Rack 8.



Troubleshooting Ops

- Microgravity issues – assembly procedures.
- Clear instructions – “Just-in-time” videos.
- Direct communication with the crew.
 - More efficient, and when talking directly with crewmembers the actions are more coherent.
- More inflight anomaly resolutions.
 - Do whatever can be done on orbit before bringing it home.
- Engage the crew early.

On-orbit installation in the US Lab



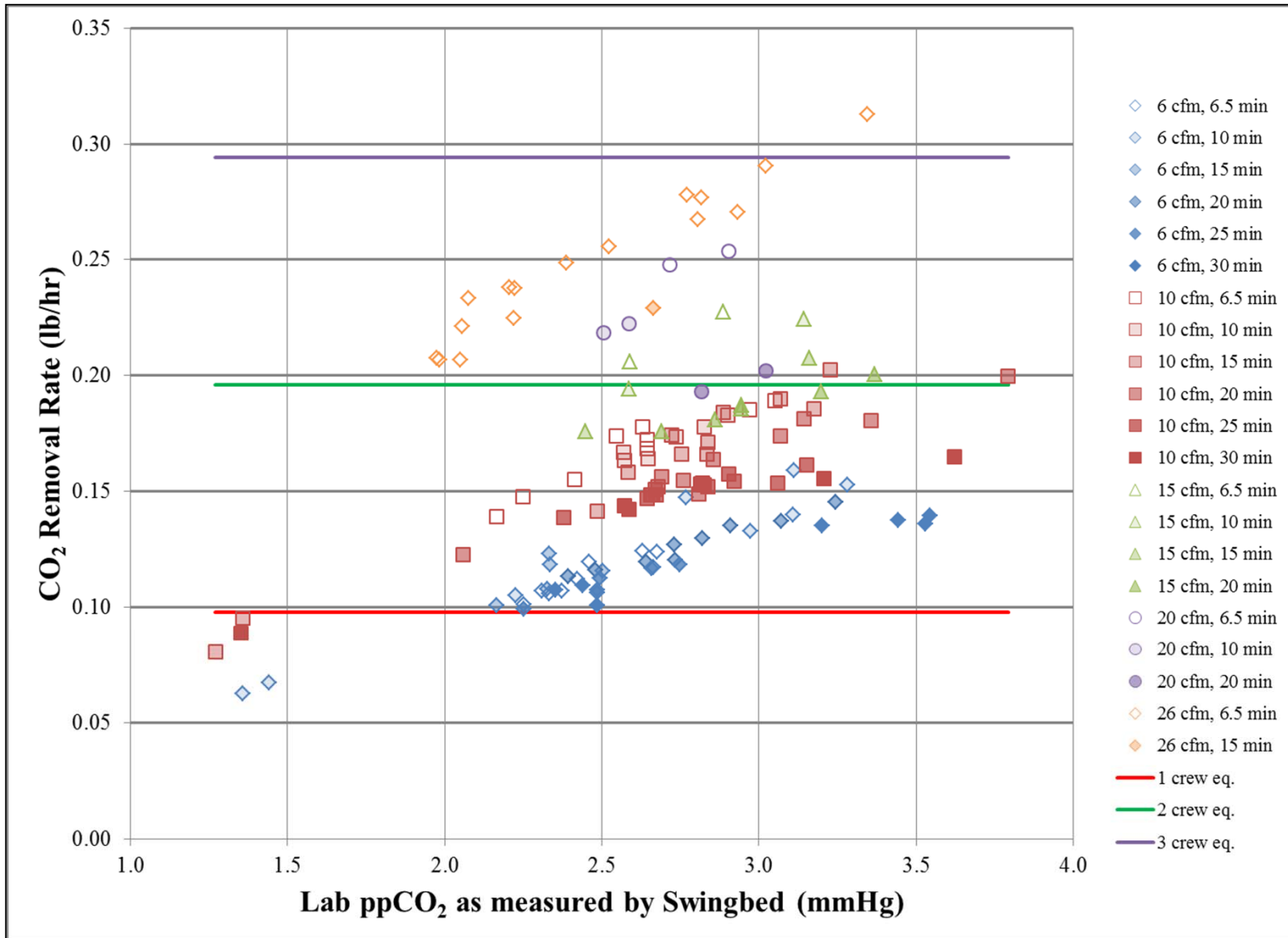
Swingbed inside locker chassis

Payload controller containing power and control functions and data collection

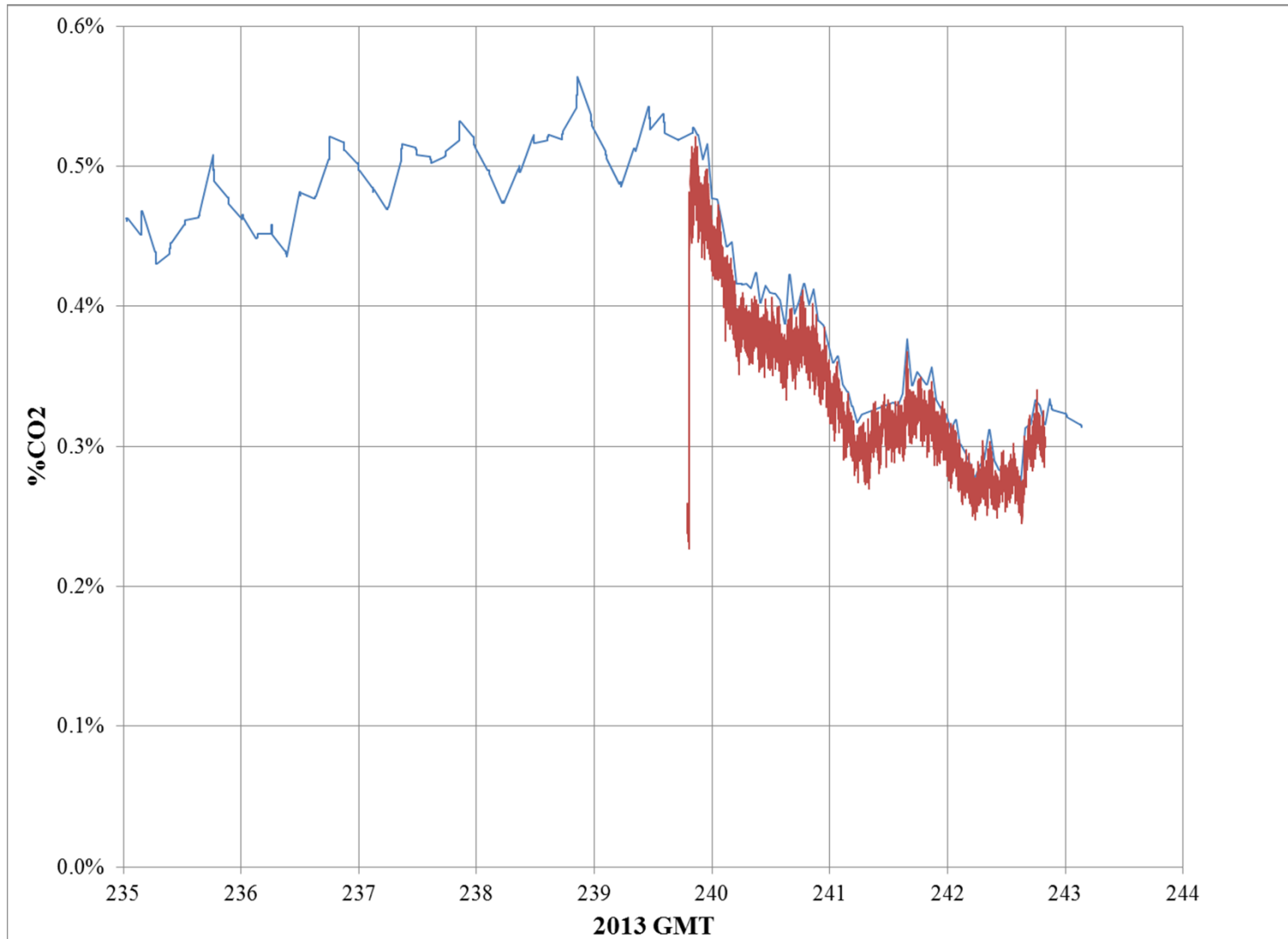
On-orbit operations

- The Amine Swingbed Payload is controlled by the Amine Swingbed team from the Payload Operations Control Center (POCC) at the Johnson Space Center in Houston by the Amine Swingbed team.
- Amine Swingbed Payload can run autonomously.
- Conducted 1000 hours of tests between May 2013 – February 2014, typically 8 – 16 hours at a time.
- Varied the air flow rate and rotary valve cycle period of the system.

CO₂ scrubbing performance



Case study: How much can Amine Swingbed impact ISS CO₂ levels?



Benefits of the Payload Experience

- Buys down risk for Orion
 - Important lessons related to motor/gearbox
 - If the rotary valve was the problem, could have major impacts to Orion
- Going through the Safety Review Process helps expand your on knowledge of your payload
- Team is better because of the payload experience
 - Technology development experts caught of glimpse of the Ops world

Amine Swingbed as an ISS Resource

- ISS Program will maintain the capability to operate the Amine Swingbed Payload through at least 2015.
 - Provide backup CO₂ removal for contingency ops.
 - Provide supplementary CO₂ removal.
- Recently supported Node 3 CDRA maintenance activity (April, 2014).
 - The Node 3 Carbon Dioxide Removal Assembly (CDRA) was taken offline for maintenance in April 2014. ISS program requested the Amine Swingbed Payload to provide supplemental and contingency CO₂ removal during this period.

Acknowledgements

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