

JANNAF

Liquid Propulsion Subcommittee and Advanced Materials Panel Additive Manufacturing for Propulsion Applications Technical Interchange Meeting

Evaluation of Additively Manufactured Demonstration Hardware for a Turbopump Application

September 4, 2014

Derek O'Neal

256-544-2543 derek.oneal@nasa.gov

MARSHALL SPACE FLIGHT CENTER

ENGINEERING DIRECTORATE
PROPULSION SYSTEMS DEPARTMENT
PROPULSION COMPONENT DESIGN & DEVELOPMENT DIVISION
TURBOMACHINERY DESIGN & DEVELOPMENT BRANCH

Agenda



- Introduction Turbomachinery and Additive Manufacturing (AM)
- NASA MSFC Turbomachinery Branch AM Goals
- Selective Laser Melting (SLM) Hardware Demonstrations
 - Images of Hardware
 - White Light Scan Results
 - Surface Evaluation
- SLM Material Test Specimens
 - Tensile Test Results
 - Fatigue Test Results
- Conclusion



Liquid Rocket Engine Turbopumps

Complex Geometries

Blades/Vanes
Complex Flow Passages & Ports









- High Shaft Speed
- Large Temperature Gradients
- High Pressure Loadings
- Dynamic Modes



Complex hardware, designed near the limits of the state-of-the-art, with predicted or demonstrated high reliability **leads to...**

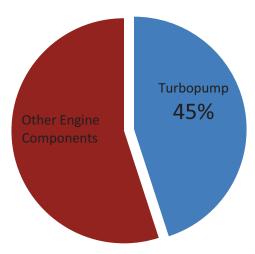
Introduction



Liquid Rocket Engine Turbopumps

- Long design and development lead time
 - Analyses for design and reliability
- Test data needed to verify models
- Long hardware fabrication lead times
 - Process development (castings, welds, etc.)
- Complex parts with many features
- Increased cost

Fastrac Development Engine Manufacturing Cost



Can we use **Additive Manufacturing** techniques to:

- Reduce manufacturing cost and lead time?
- Get hardware into test early enough to anchor models and provide a more robust design?

Turbomachinery Branch AM Goals



- Develop design experience and techniques to take full advantage of AM process benefits while understanding constraints
- Advance technology readiness level (TRL) of AM turbomachinery components and materials, allowing for easier insertion into mainline programs.
 - Demonstration of representative piece part designs
 - Material property verification

 Develop and test a turbopump assembly that uses AM techniques to the greatest extent possible.



Material Testing



Integrate AM into turbopump



Two SLM vendors were tasked with building selected turbopump components with lot test specimens from **IN718**.

Part	Model Image	Vendor	Surface Finishing	WLS	Surface Evaluation	Z Tensile	XY Tensile	Fatigue Surface Finish	Z Fatigue	XY Fatigue
Impeller		А	MMP	√	✓	4	2	MMP	6	0
		В	Ext: Bead Blast Int: Extrude Hone		✓	4	2	Bead Blast	6	0
Pump Volute		А	Ext: Bead Blast Int: Extrude Hone	✓	✓	4	2	Hand Polish	6	0
		В	Ext: Bead Blast Int: Extrude Hone	✓	✓	4	2	Bead Blast	6	0
Turbine Blisk		А	MMP	✓	✓	4	2	MMP	6	0
		В	Bead Blast	✓	✓	4	2	Bead Blast	6	0
Turbine Nozzle		А	MMP	✓	✓	4	2	MMP	6	0
		В	Bead Blast	✓	✓	6	0	Bead Blast	4	2
Turbine Stator		А	MMP	✓		4	2	MMP	6	0
		В	Bead Blast	✓		6	0	Bead Blast	4	2
Turbine Exit Guide Vanes		А	MMP	✓		4	2	MMP	6	0
		В	Bead Blast	✓		6	0	Bead Blast	4	2

MMP: Proprietary Micro Machining Process

WLS: White Light Scan



Impeller

Vendor A

Surface Finish: MMP





Vendor B

Surface Finish: Ext: Bead Blast Int: Extrude Hone

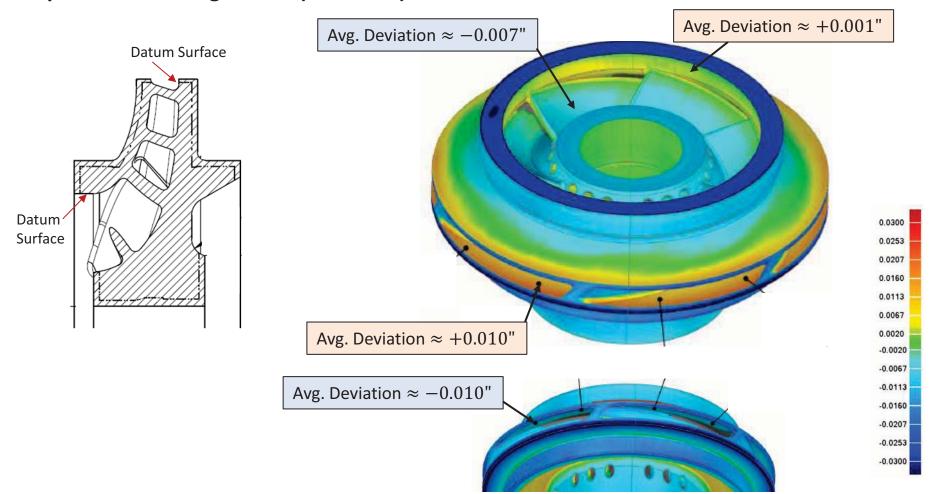




Build Direction

Build Direction

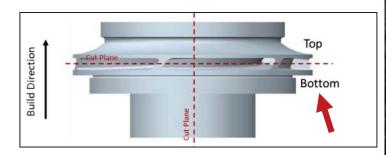
Impeller – White Light Scan (Vendor A)



Extra material is provided on external surfaces. Internal flow path surfaces are net shape.

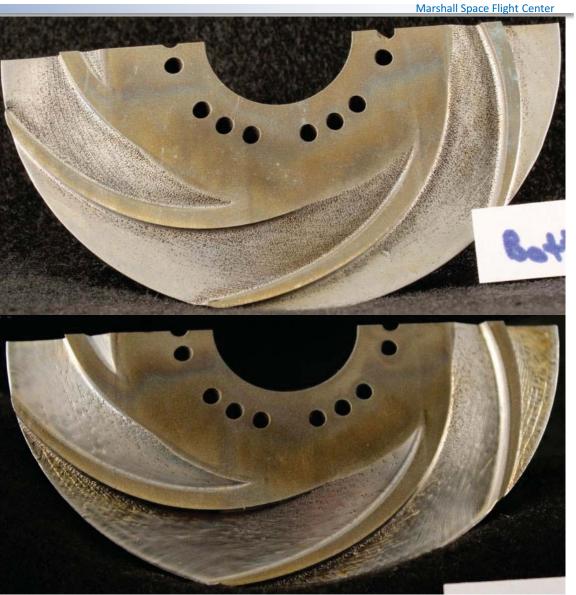


Impeller - Surface Evaluation



Vendor A – MMP (Bottom Surface)

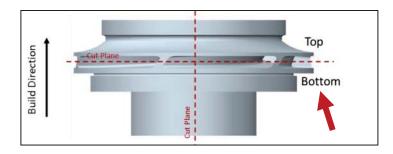
Vendor B – Extrude Hone (Bottom Surface)





Marshall Space Flight Center

Impeller - Surface Evaluation



Vendor A – MMP (Bottom Surface)

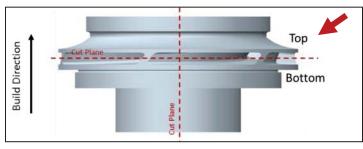
Vendor B – Extrude Hone (Bottom Surface)





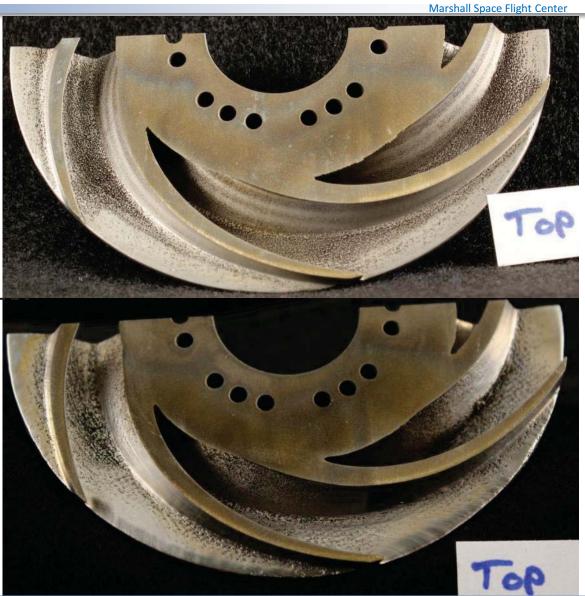


Impeller - Surface Evaluation



Vendor A – MMP (Top Surface)

Vendor B – Extrude Hone (Top Surface)





Impeller - Surface Evaluation







Vendor A – MMP (Top Surface)

Vendor B – Extrude Hone (Top Surface)





Approved for public release; distribution is unlimited.

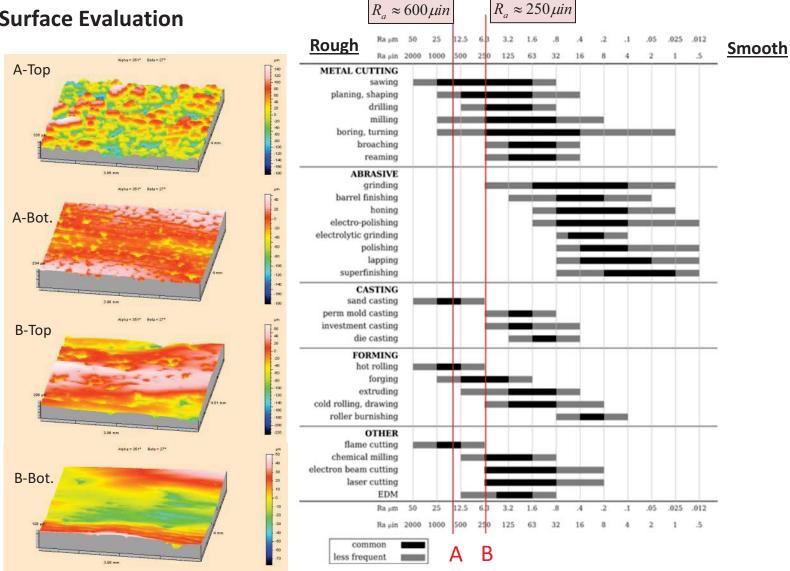




Impeller - Surface Evaluation

A - MMP

B – Extrude

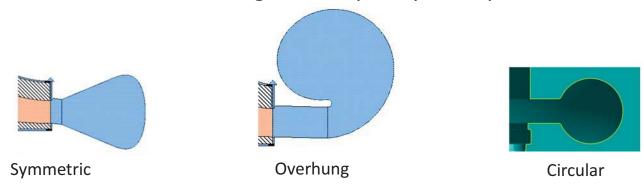




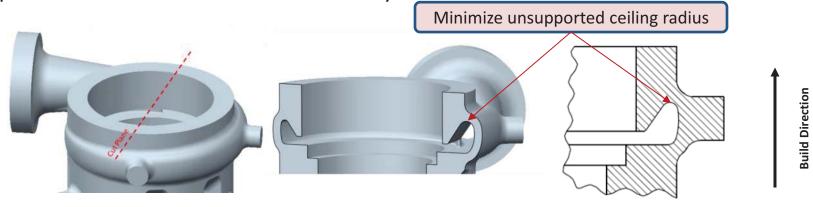
Pump Volute – Design Considerations

SLM Constraint – Unsupported ceiling radii should be minimized

Typical volute cross sections are designed for hydrodynamic performance.



Demonstration volute is designed as a compromise between hydrodynamic performance and SLM manufacturability.





Pump Volute

Vendor A

Surface Finish: Ext: Bead Blast

Int: Extrude Hone

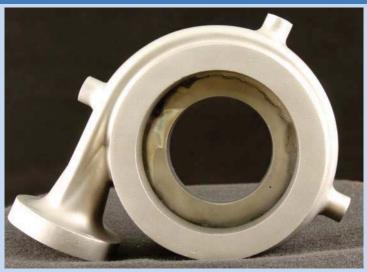




Vendor B

Surface Finish: Ext: Bead Blast

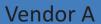
Int: Extrude Hone





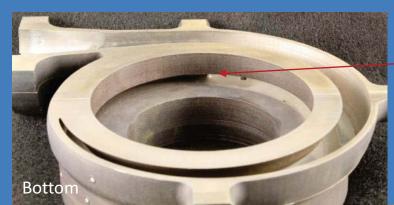


Pump Volute



Surface Finish: Ext: Bead Blast

Int: Extrude Hone



Vendor B

Surface Finish: Ext: Bead Blast

Int: Extrude Hone





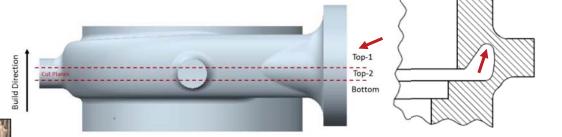
Top

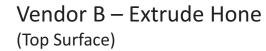


Marshall Space Flight Center

Pump Volute – Surface Evaluation

Vendor A – Extrude Hone (Top Surface)









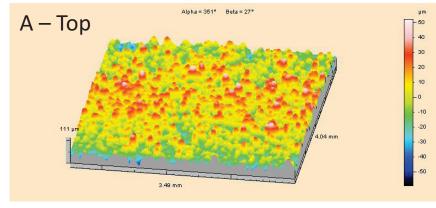


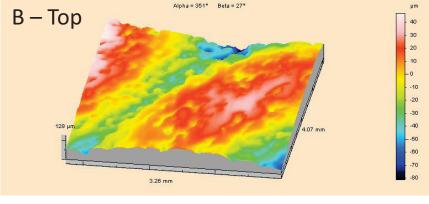
Marshall Space Flight Center

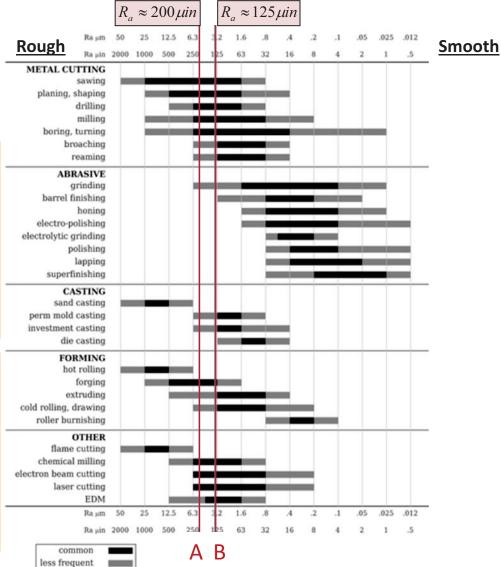


Vendor A – Extrude Hone

Vendor B – Extrude Hone









Turbine Blisk

Vendor A

Surface Finish: MMP

Vendor B

Surface Finish: **Bead Blast**





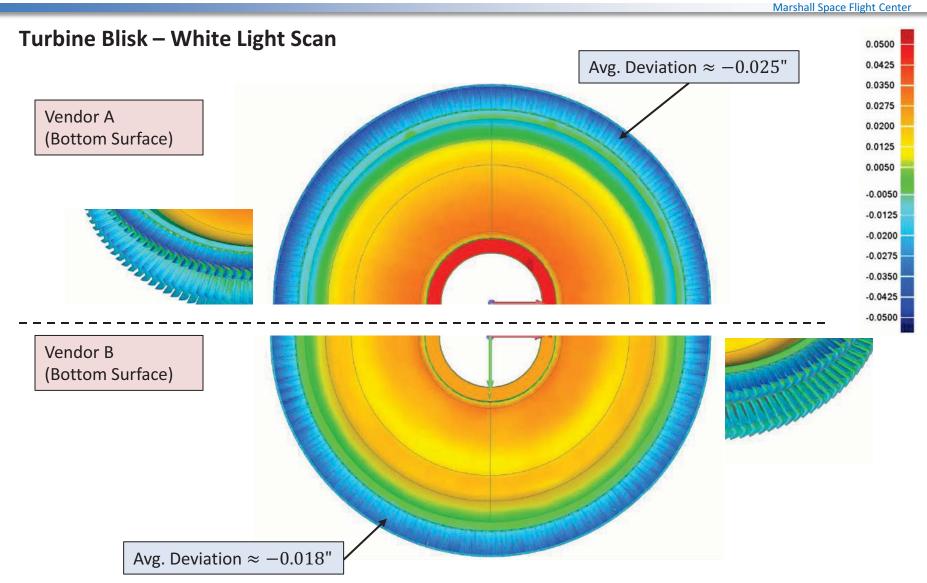




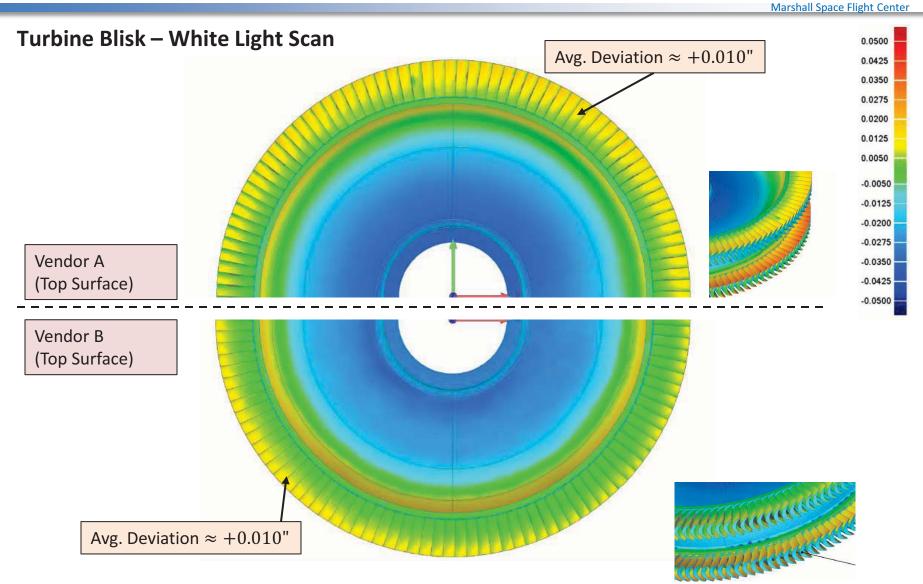


Approved for public release; distribution is unlimited.











Turbine Blisk – Surface Evaluation





Vendor A – MMP



Turbine Blisk – Surface Evaluation





Vendor B – Bead Blast



Marshall Space Flight Center

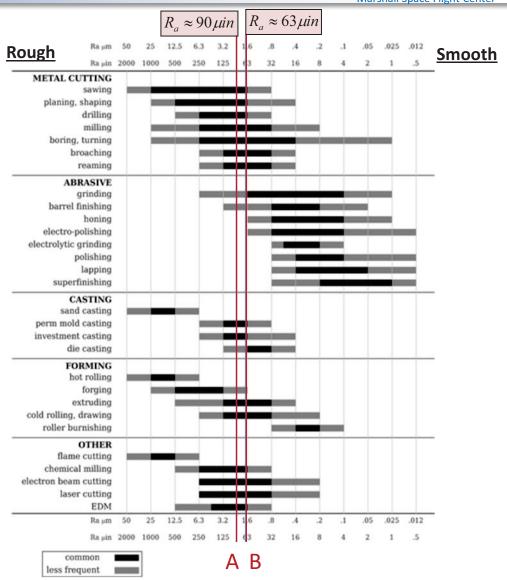
Turbine Blisk – Surface Evaluation

Vendor A - MMP

Vendor B – Bead Blast



Stylus profiling of Blisk Blade (EM10-Tribology)



SLM Material Test Specimens



Tensile Test Results

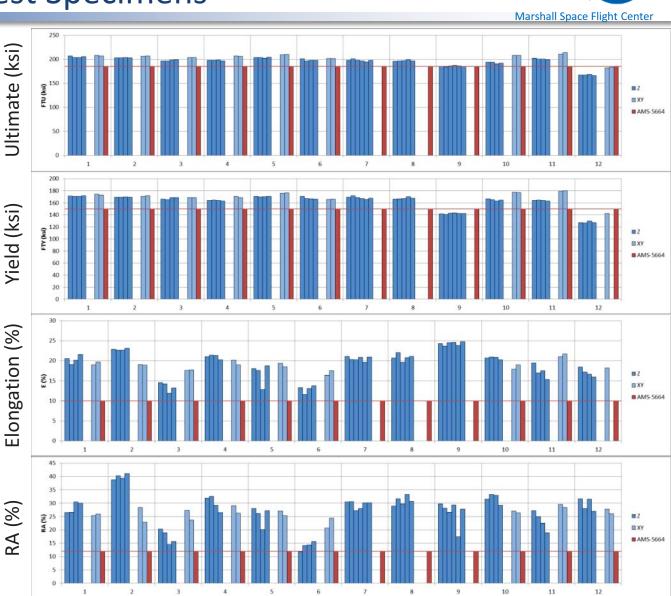
X-Axis

Vendor A Builds

- 1. Turbine Nozzle
- 2. Turbine Exit Guide Vanes
- 3. Turbine Stator
- 4. Turbine Blisk
- 5. Impeller
- 6. Pump Volute

Vendor B Builds

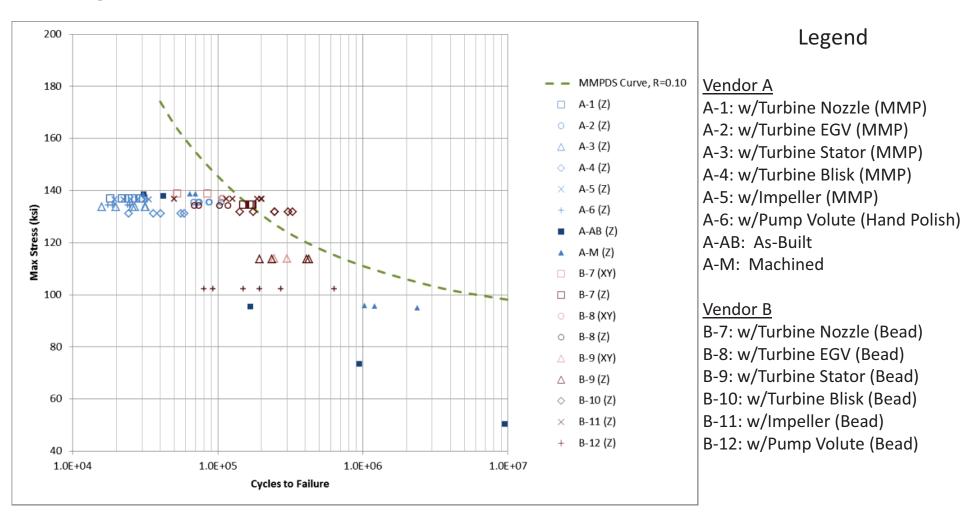
- 7. Turbine Nozzle
- 8. Turbine Exit Guide Vanes
- 9. Turbine Stator
- 10. Turbine Blisk
- 11. Impeller
- 12. Pump Volute



SLM Material Test Specimens



Fatigue Test Results



Conclusion



The SLM hardware demonstrations help fulfill Turbomachinery Branch, AM Goals:

- Develop AM design experience ✓
- Advance TRL of AM turbomachinery components and materials
 - Demonstration of representative piece part designs ✓
 - Continue to improve process (surface finishing, removing supports and powder, dimensional tolerance).
 - Material property verification ✓
 - Continue to grow material property database. Build lot test specimens with all parts.
 - Develop and test a turbopump assembly that uses AM techniques to the greatest extent possible. (The next step)

The SLM demonstration hardware met most of the design intentions. With a few process improvements, these geometries can be integrated into a turbopump assembly.

Acknowledgements



Mechanical Test Branch – EM10

Doug Wells (EM20) – Test Planning

Vann Bradford (EM10) – Material Test

Chip Moore (EM10) – Surface Evaluation

Brian West (EM42) – White Light Scanning



BACK UP

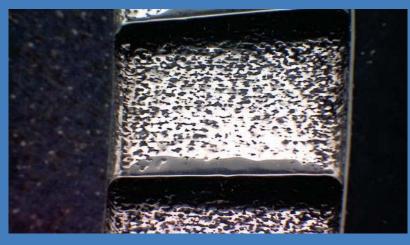


Turbine Nozzle

Vendor A

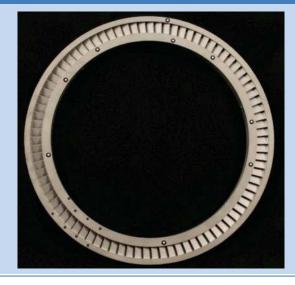
Surface Finish: MMP

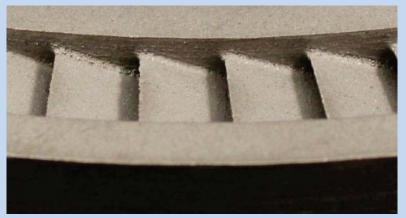




Vendor B

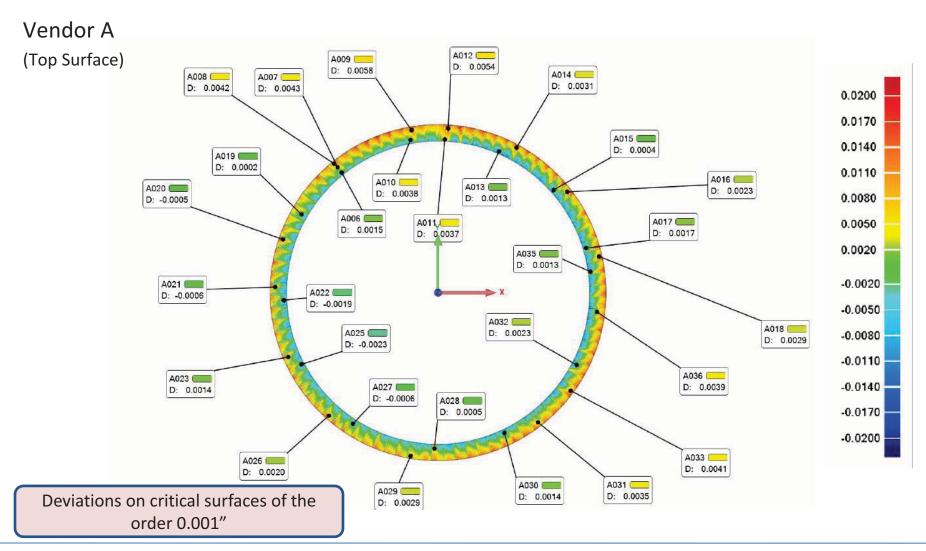
Surface Finish: Bead Blast



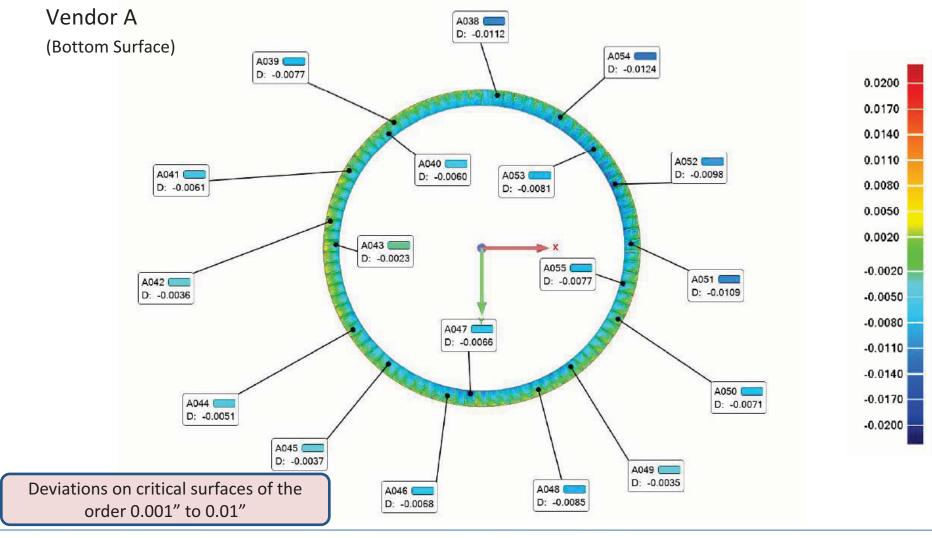




Turbine Nozzle – White Light Scan



Turbine Nozzle – White Light Scan





0.0100

0.0085 0.0070 0.0055

0.0040

0.0025 0.0010

-0.0010

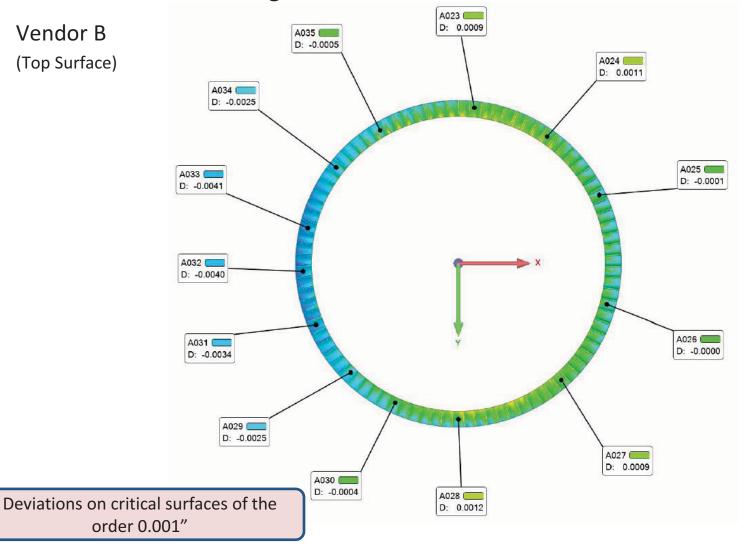
-0.0025 -0.0040

-0.0055

-0.0070 -0.0085 -0.0100

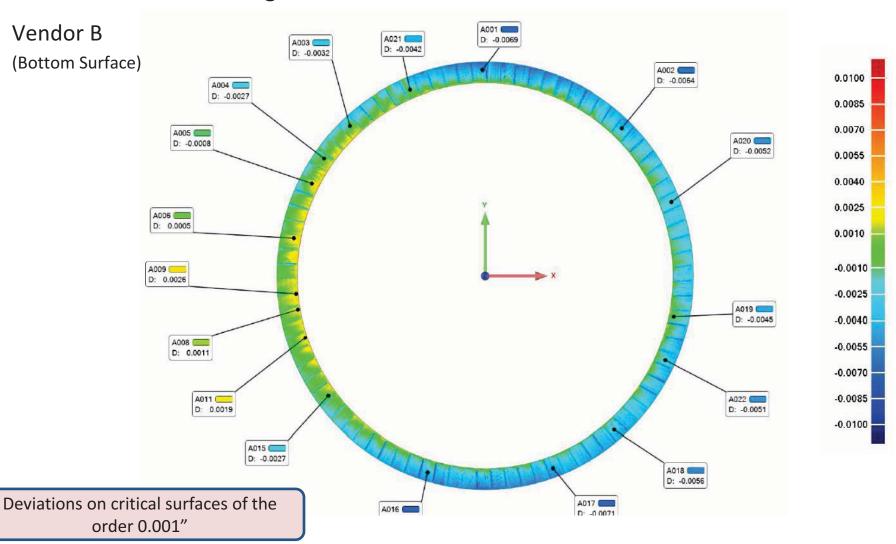
Turbine Nozzle – White Light Scan

Vendor B (Top Surface)





Turbine Nozzle – White Light Scan





Turbine Nozzle – Surface Evaluation

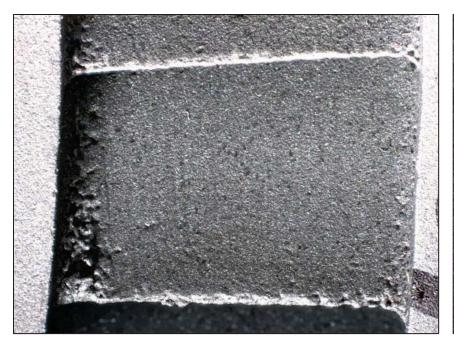




Vendor A – MMP



Turbine Nozzle – Surface Evaluation





Vendor B – Bead Blast



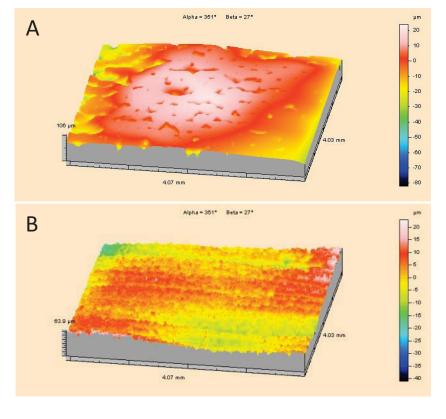


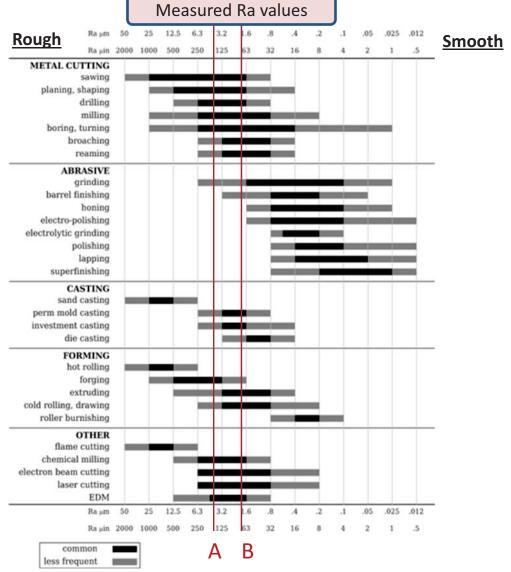
Marshall Space Flight Center

Turbine Nozzle – Surface Evaluation

Vendor A - MMP

Vendor B – Bead Blast







Turbine Stator

Vendor A

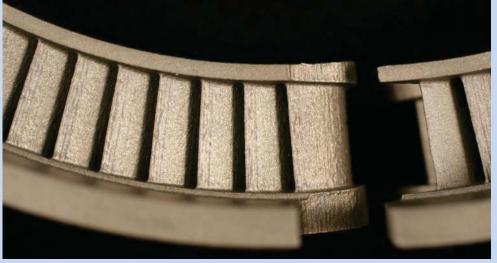
Surface Finish: MMP

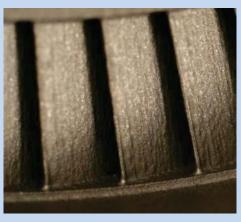




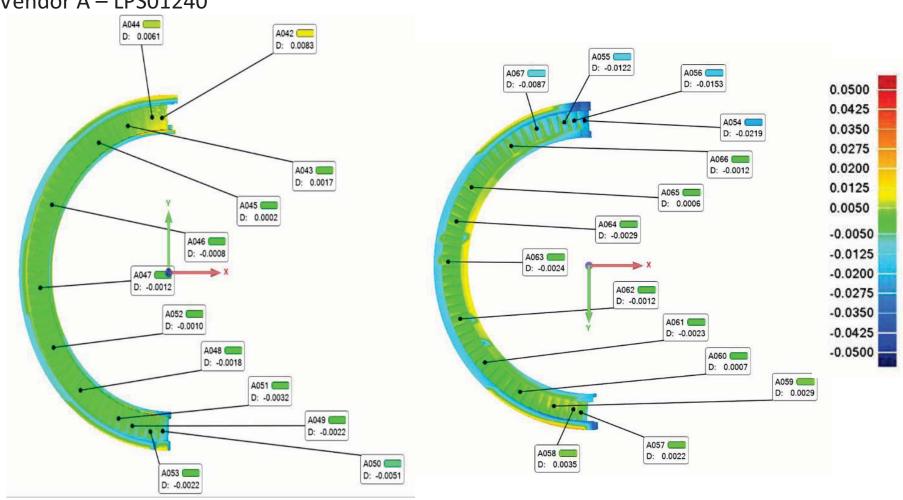
Vendor B

Surface Finish: Bead Blast



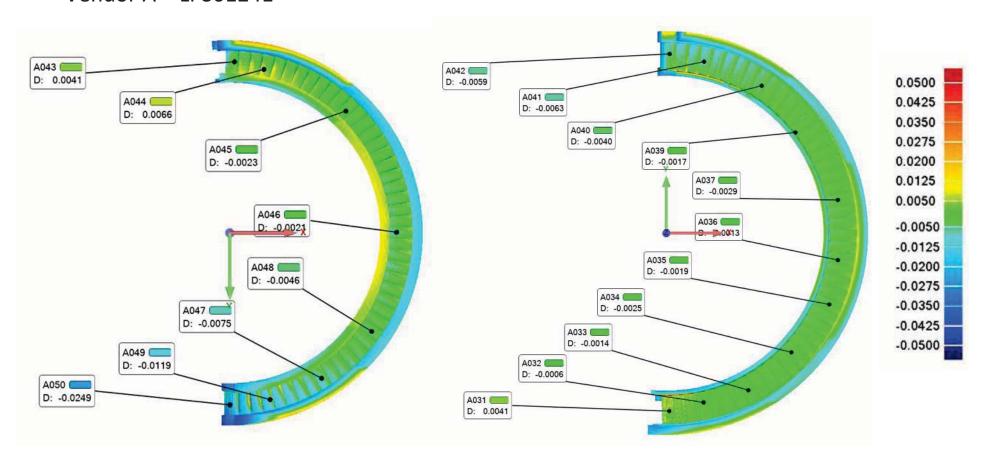


Vendor A - LPS01240

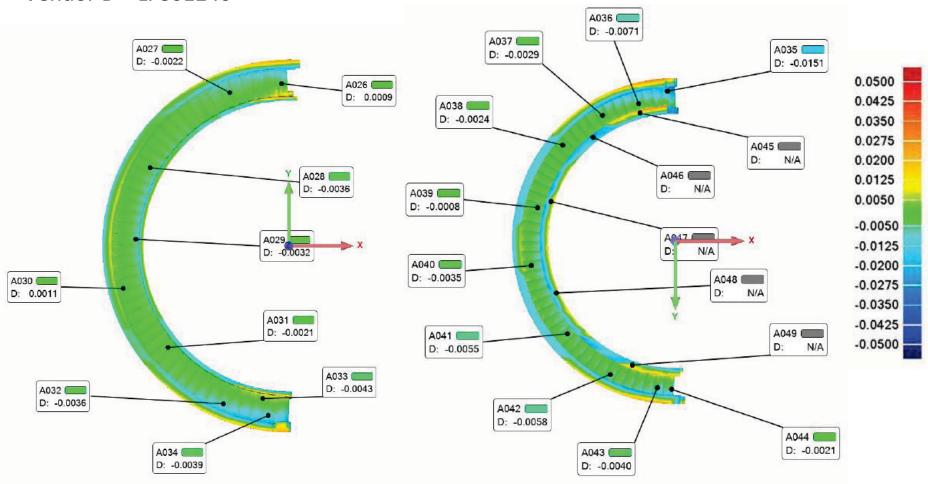




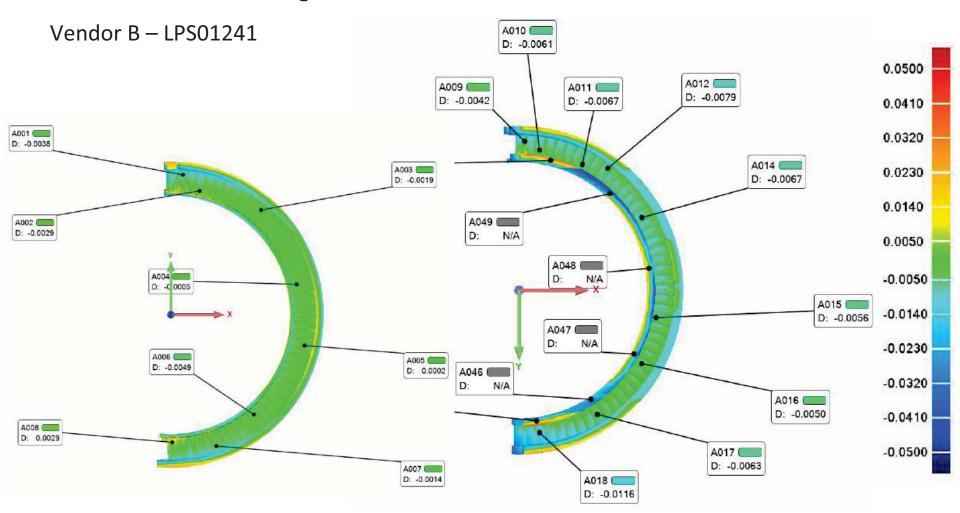
Vendor A - LPS01241



Vendor B - LPS01240





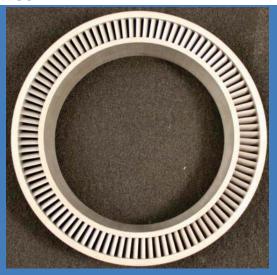




Turbine Exit Guide Vanes

Vendor A

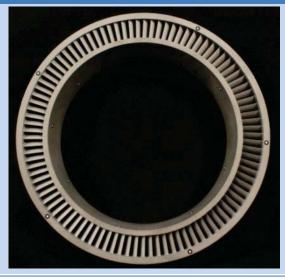
Surface Finish: MMP

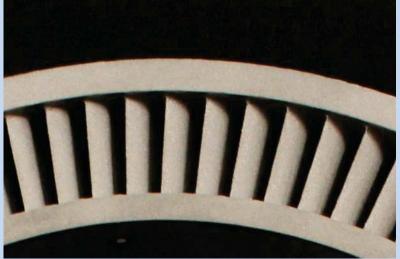




Vendor B

Surface Finish: Bead Blast

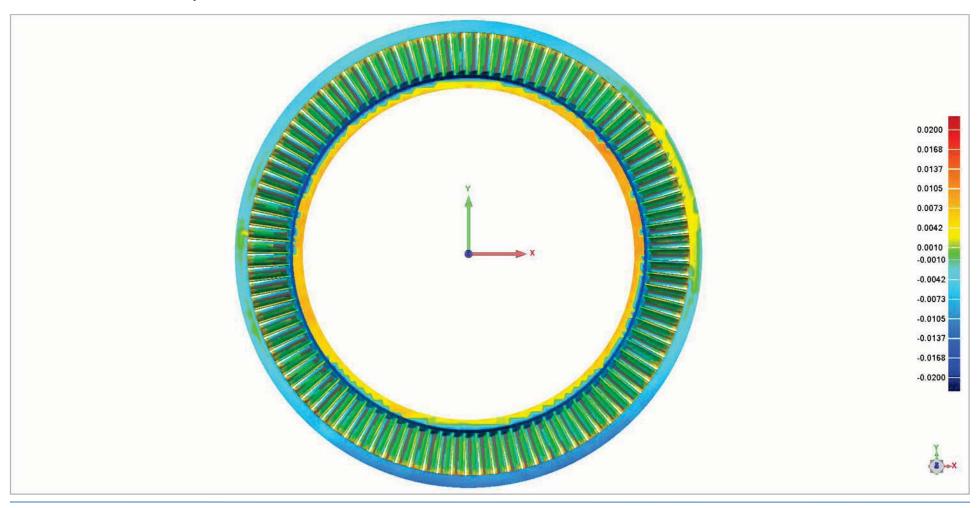






Turbine Exit Guide Vanes – White Light Scan

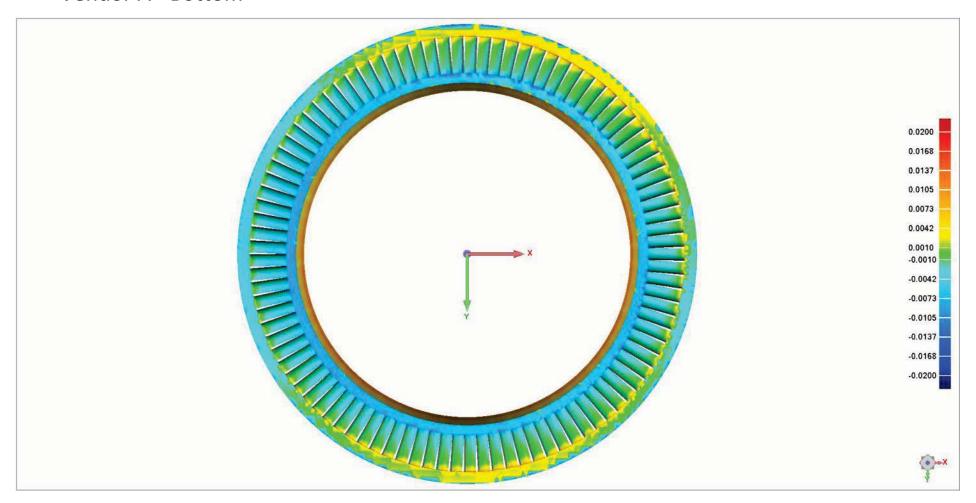
Vendor A - Top





Turbine Exit Guide Vanes – White Light Scan

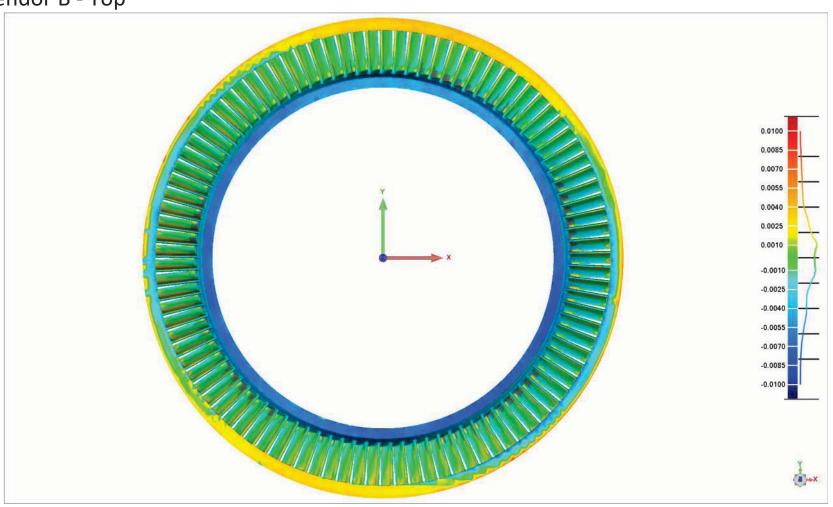
Vendor A - Bottom





Turbine Exit Guide Vanes – White Light Scan

Vendor B - Top





Turbine Exit Guide Vanes – White Light Scan

Vendor B - Bottom

