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Effect of Background Pressure on the Plasma Oscillation Characteristics of the HiVHAc Hall Thruster



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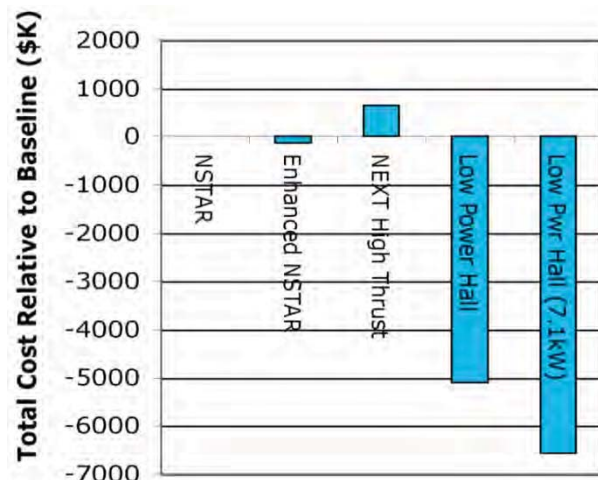
Present at 50th JPC, Cleveland OH, Jul 29, 2014



Introduction



- Two NASA studies^[1,2] show that Hall thruster can provide considerable cost benefits compared to Ion thruster for Discovery class missions and a high-voltage Hall thruster outperforms current SotA Hall thrusters
 - SMD initiated the High-Voltage Hall Accelerator (HiVHAc) project
- Latest version of HiVHAc is EDU2
- Latest test of the EDU2 is a component compatibility test (CCT)
 - Also reported in: IEPC-2013-445, IEPC-2013-058
- This paper describes characterization of the plasma oscillation characteristics as the background pressure is varied during the CCT



↑ [1]Oh, D., JPC, AIAA-2005-4270, 2005.

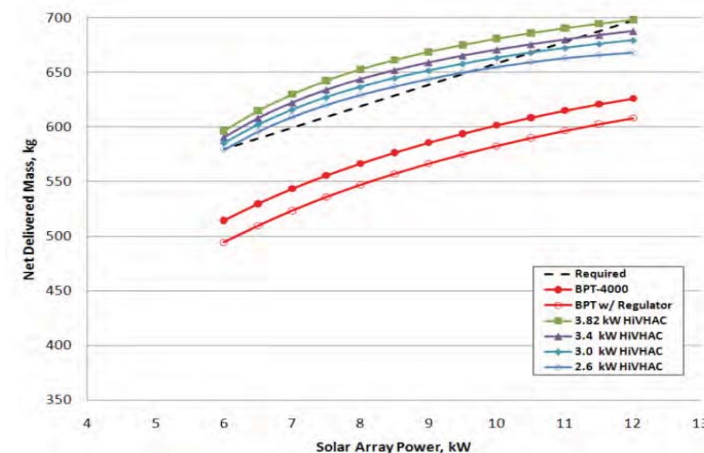


Figure 4. Performance comparison for the Dawn mission.

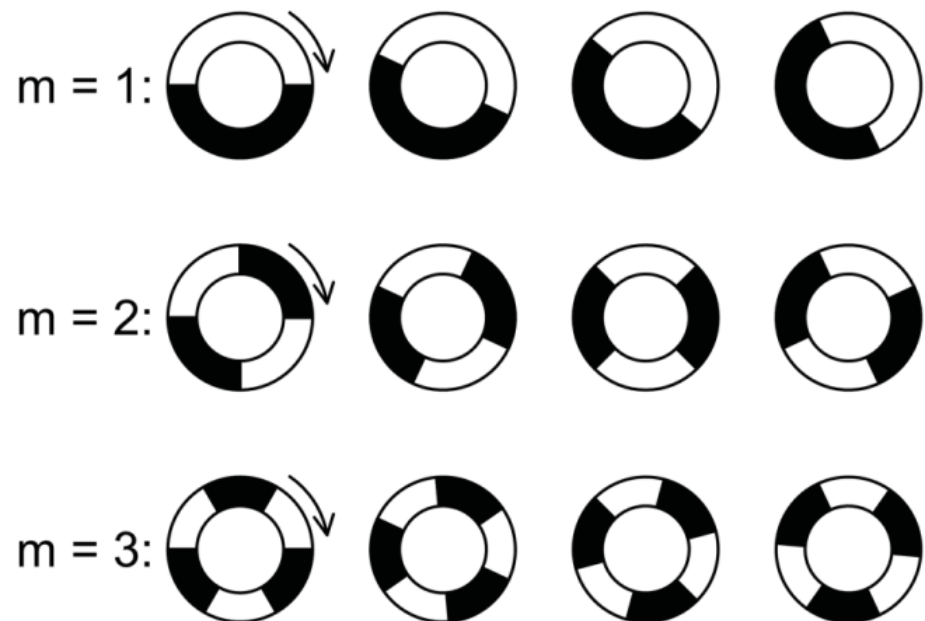
↑ [2]Dankanich, J. W., et al., IEPC, 2009-213, 2009.

Goal, Objective, and Oscillation Modes

- **Goal:** Determine the maximum acceptable background pressure for testing the HiVHAc EDU2 from the perspective of plasma oscillations
- **Objective:** Obtain data on how the characteristics (magnitude and frequency) of the plasma oscillations change with pressure
- Focused on breathing mode and spokes mode



Breathing mode video



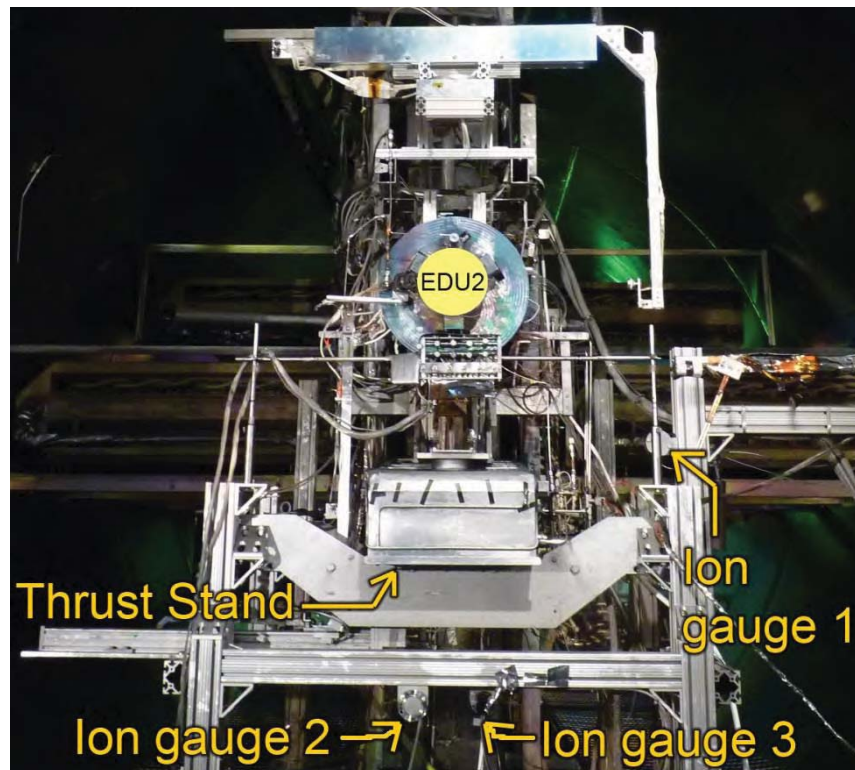
Spokes mode illustration



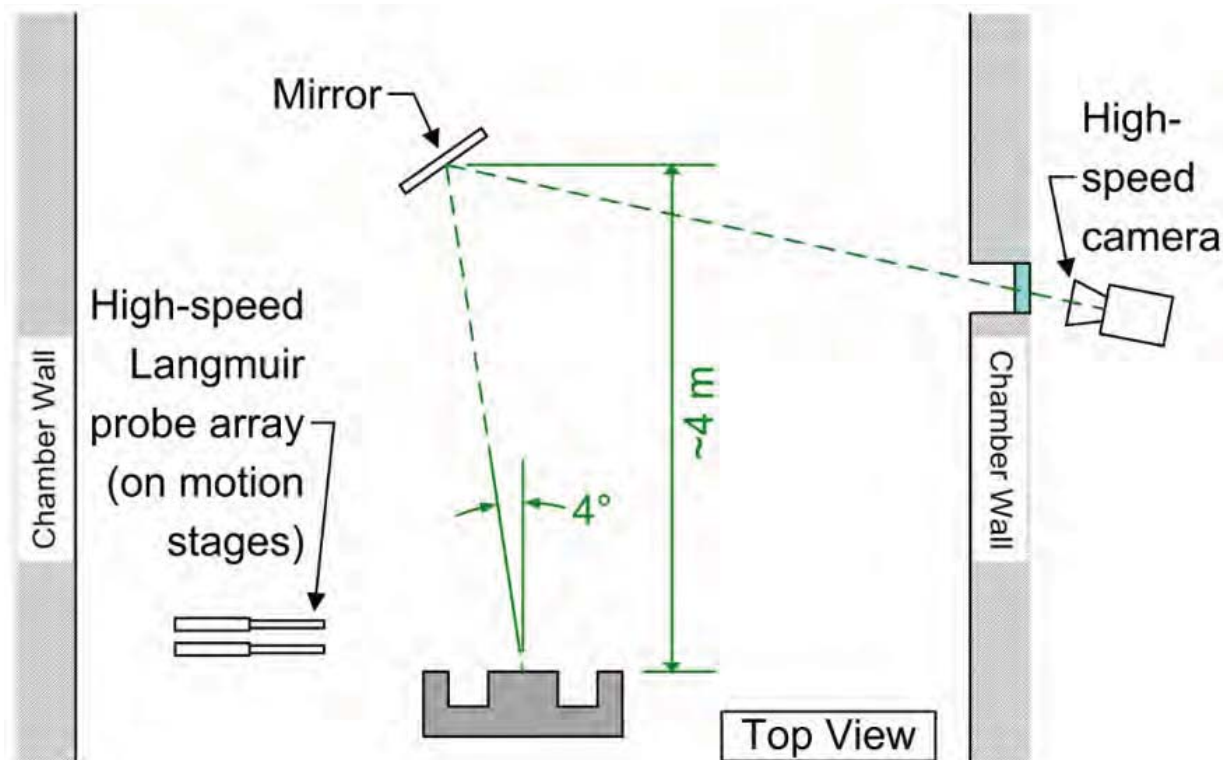
NASA HiVHAc EDU2 Hall thruster



- NASA HiVHAc EDU2 is a 3.9-kW Hall thruster
 - Can throttle from 200 to 650 V (1200 to 2700 sec); 0.3 to 3.9 kW
 - Magnetic lens topology symmetric about the channel centerline
 - Seven throttle points selected: 300 to 500 V, 1.5 to 3.9 kW
- Ion gauge 2 used to determine background pressure
- Label: vvv-k.k-Pnx (voltage-power-normalized pressure)
 - Pressure normalized by lowest achievable pressure
- Mass flow rates set while at lowest background pressure (P1x) and fixed for a given throttle point; CFF = 6.7%
- Magnetic field strength set to optimize efficiency at P1x



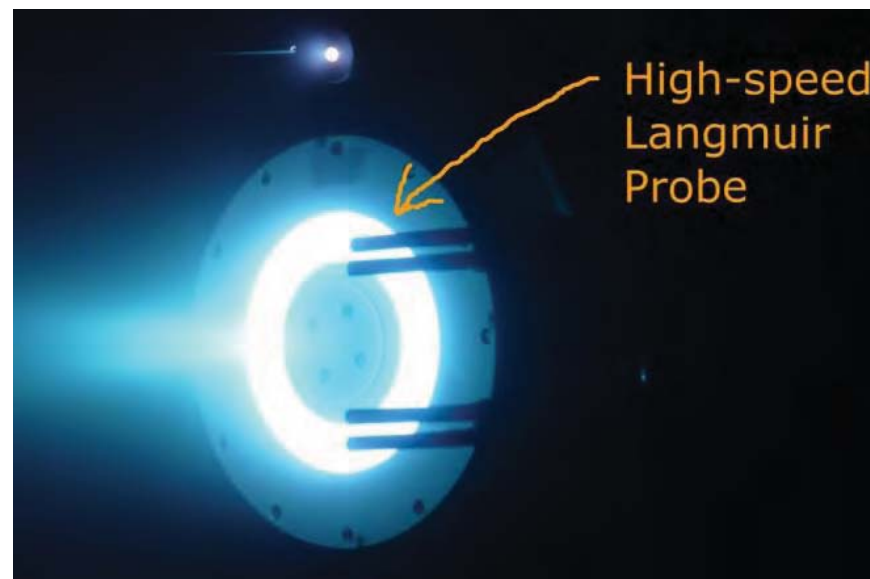
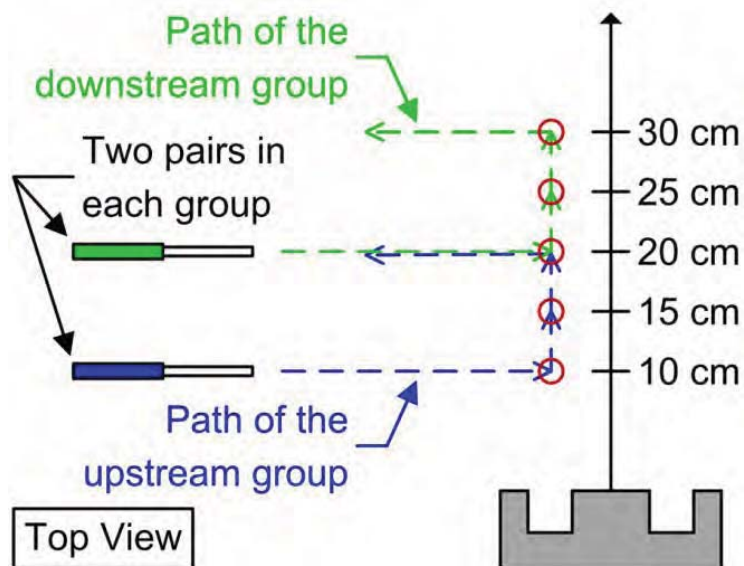
Experimental Setup: High-speed Camera



- High-speed camera filmed at $\sim 80,000$ fps
- Thruster occupy 100×100 pixels
- 4 degrees off-axis (to make room for an IR camera)
- High-speed video at various pressure collected only for 500 V conditions; video at lowest pressure available at other conditions



Experimental Setup: High-speed Langmuir Probe



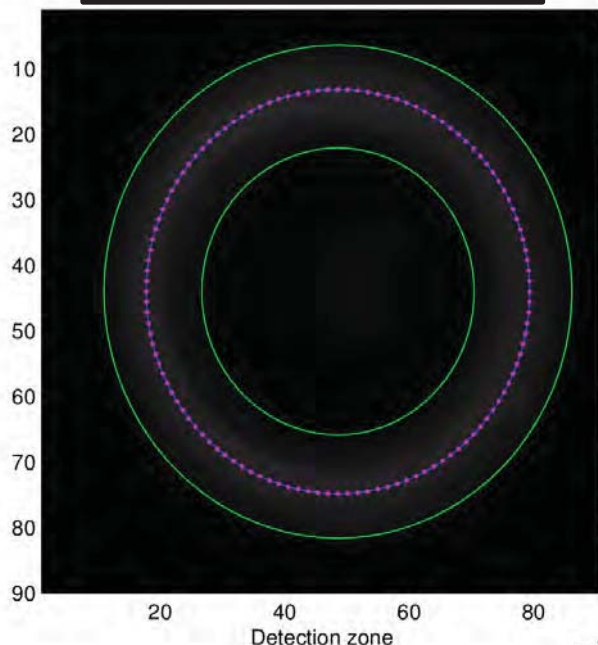
- Four pairs of Langmuir probes in a movable array
 - Each pair consist of an active and a null probe for high-speed sweeping
- Two pairs are axially upstream while the other two pairs are axially downstream, each pair is set azimuthally apart
- Sweep rate is 200 kHz
- Array moves to three axial locations along the channel centerline



Data analysis: High-speed Camera

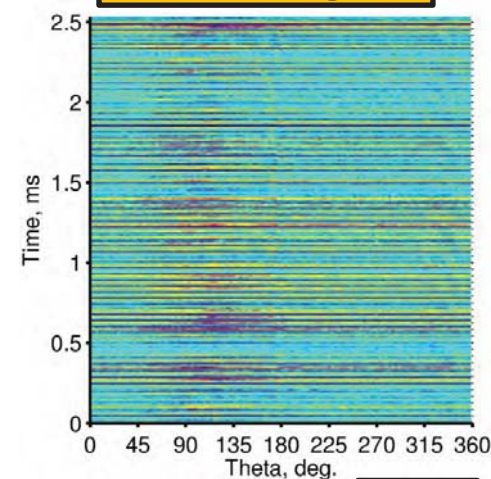


Plot of crop & circle-fit

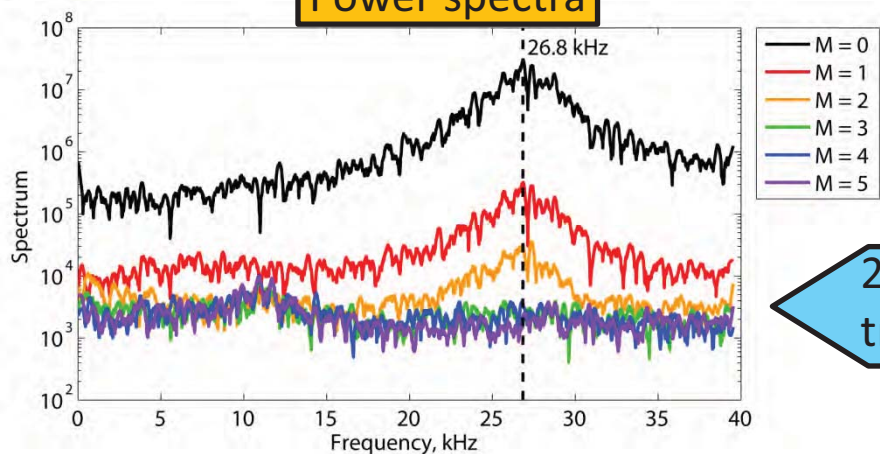


Bin data by azimuthal angle

Theta-t diagram



Power spectra



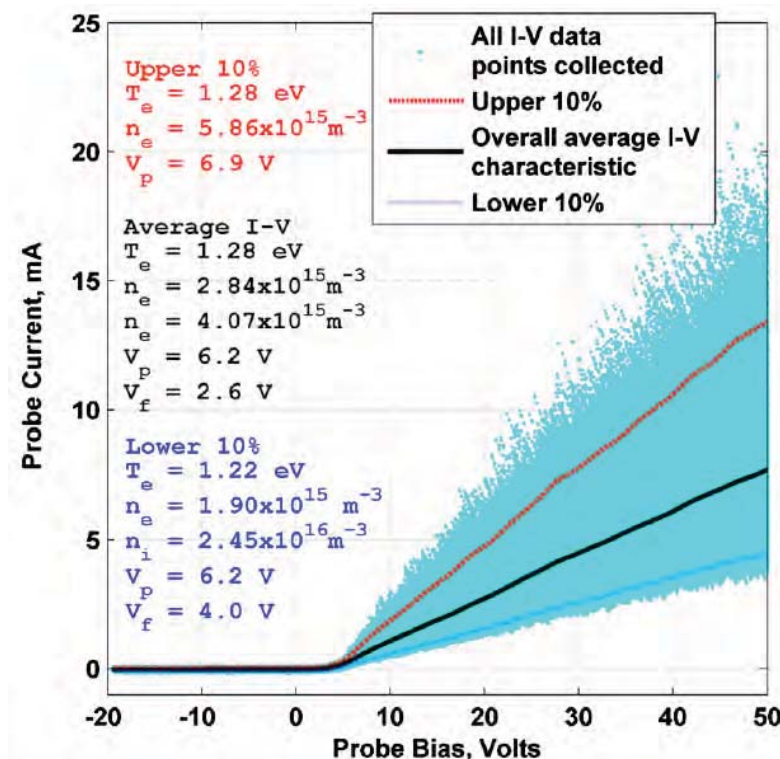
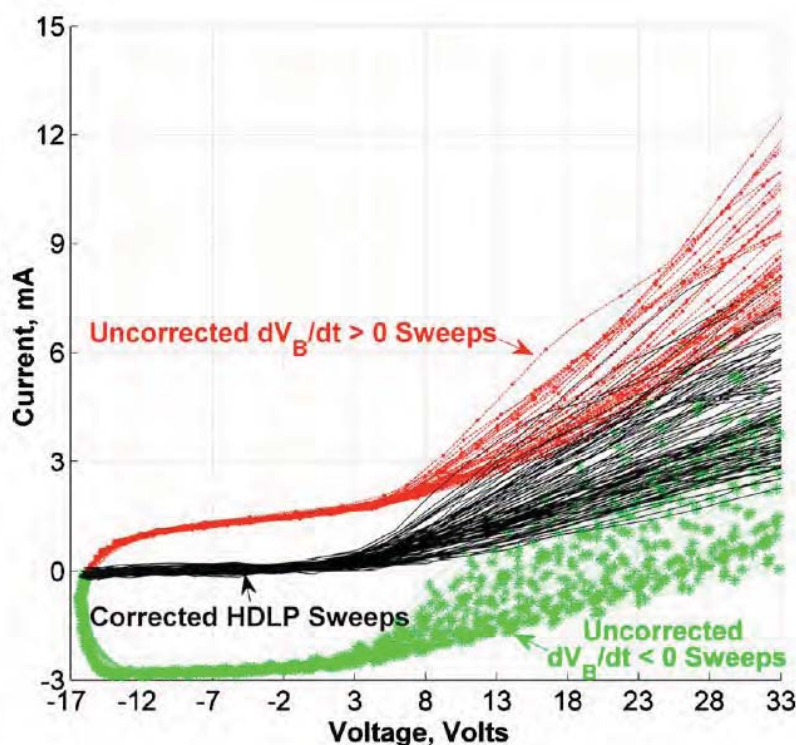
2D Fourier transform



Data analysis: High-speed Langmuir Probe



- Null probe data subtracted from active probe data to correct for capacitive effects associated with high-speed sweeping
- Analyzed using standard collisionless non-magnetized thin-sheath Langmuir probe theory

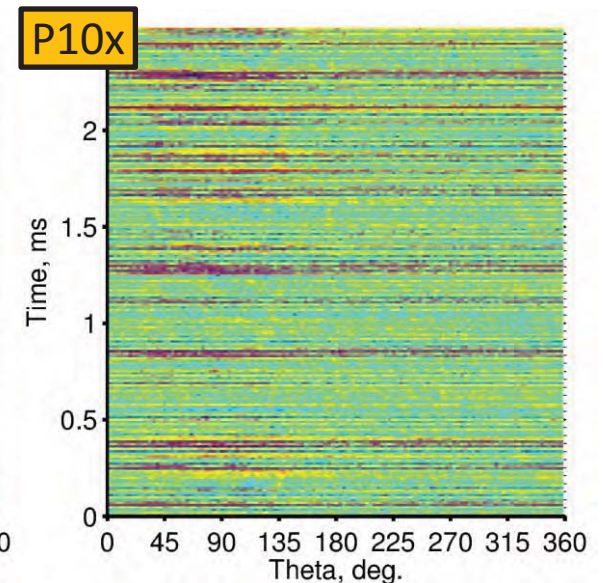
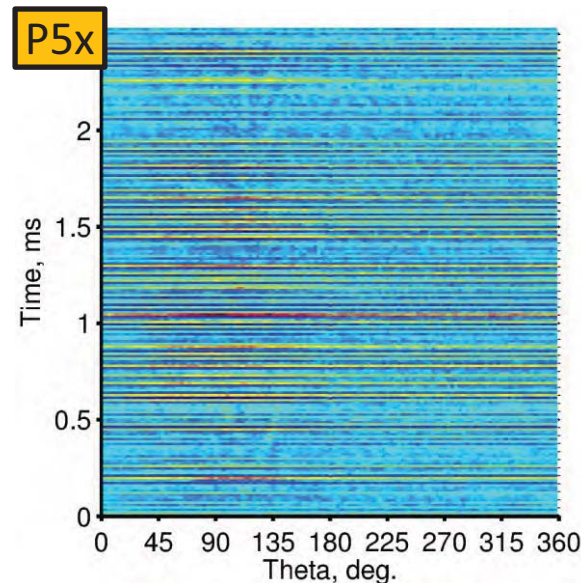
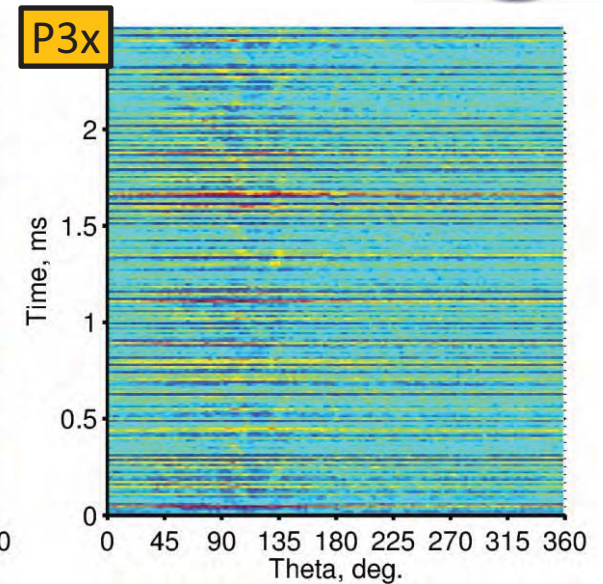
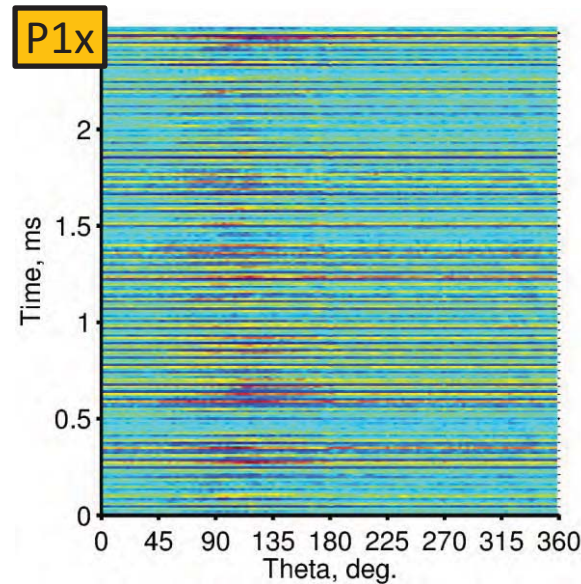




Results: High-speed Camera



- High-speed videos show mostly breathing mode oscillations
- Spokes occasionally appear but is not sustained
- Theta-t diagrams for the 500 V, 3.9 kW throttle point

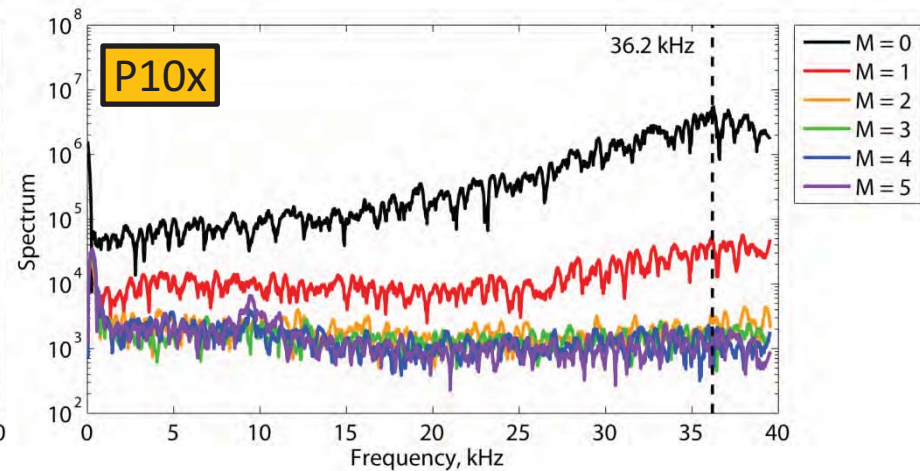
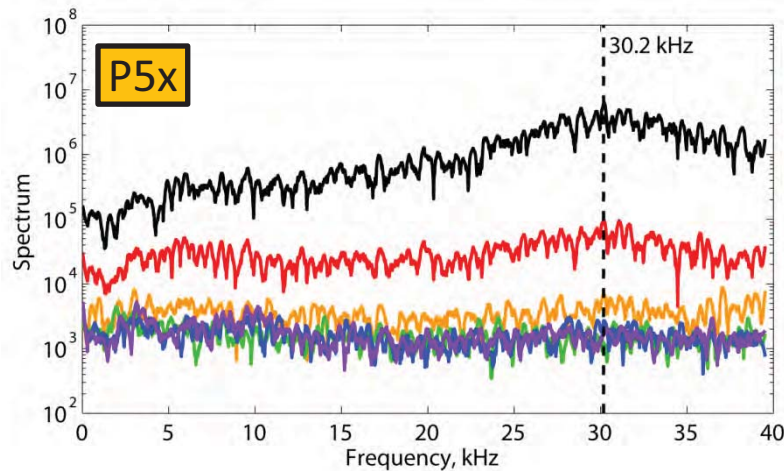
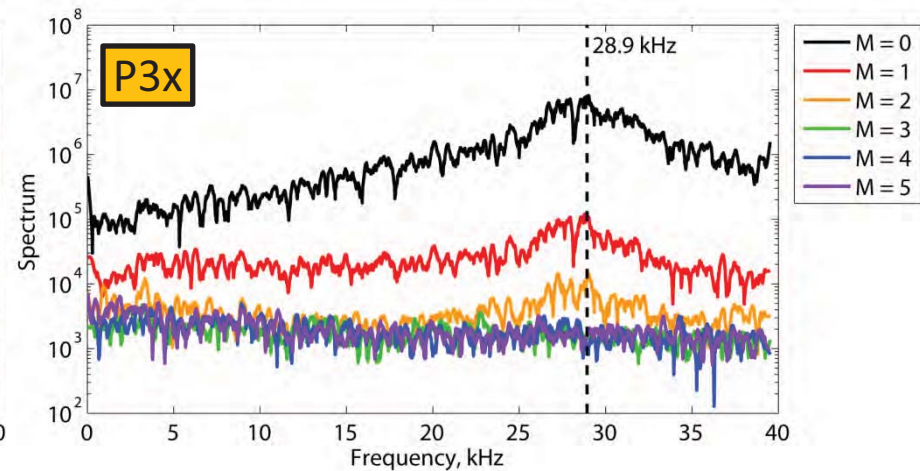
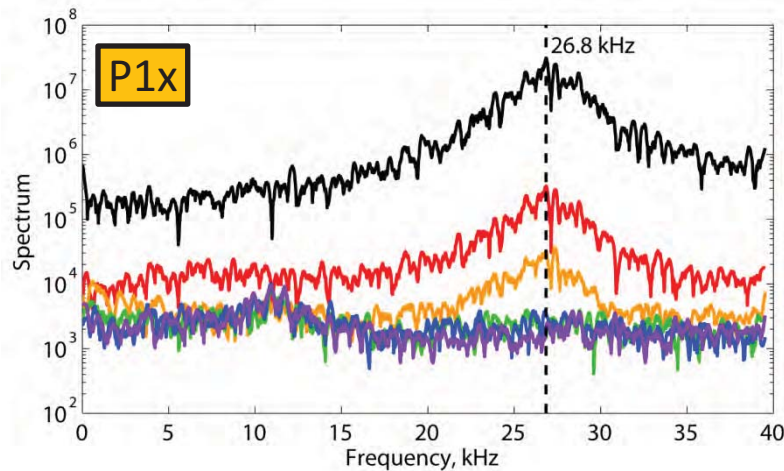




Results: High-speed Camera



- Power spectra for the 500 V, 3.9 kW throttle point





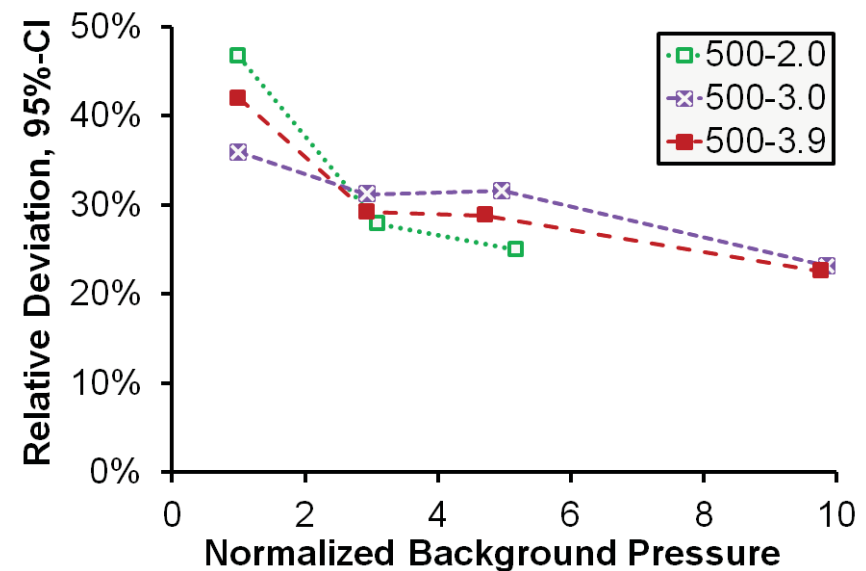
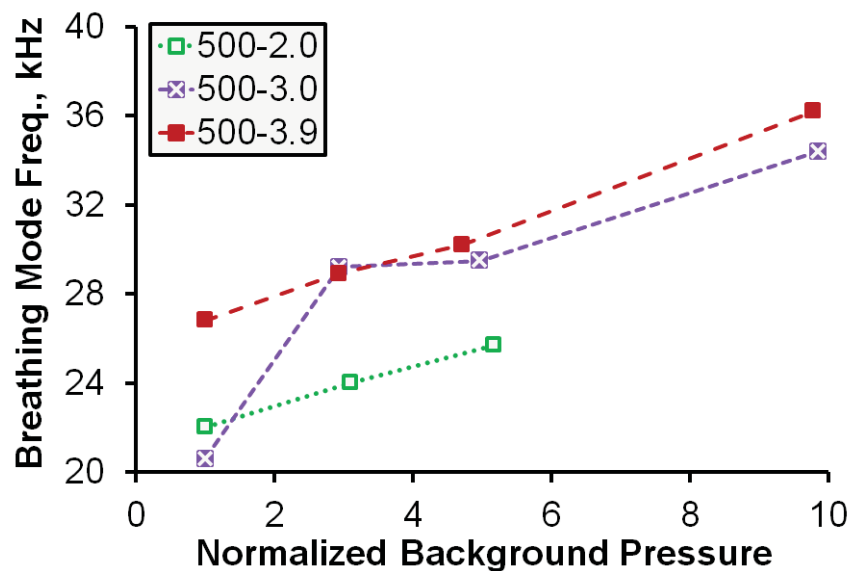
Results: High-speed Camera



- High-speed camera shows breathing mode frequency increases and magnitude decreases with rising background pressure
- Scaling relationship from Fife's work (AIAA-1997-3052) suggests that the characteristic axial length of the ionization/acceleration zone (I/AZ) decreases with rising background pressure

$$2\pi f_i = \frac{\sqrt{u_i u_n}}{L_i}$$

- Note the high-speed camera captures mostly light from I/AZ

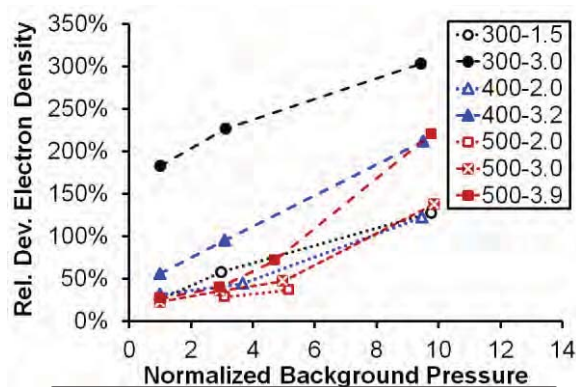
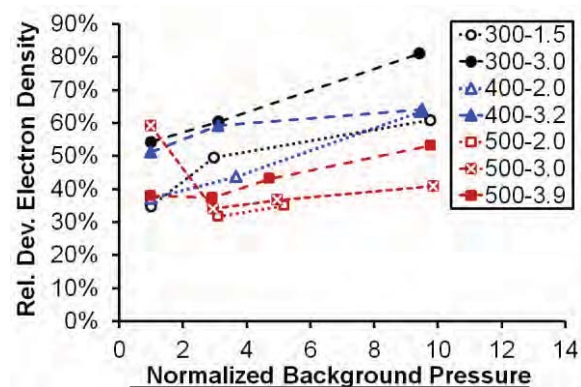
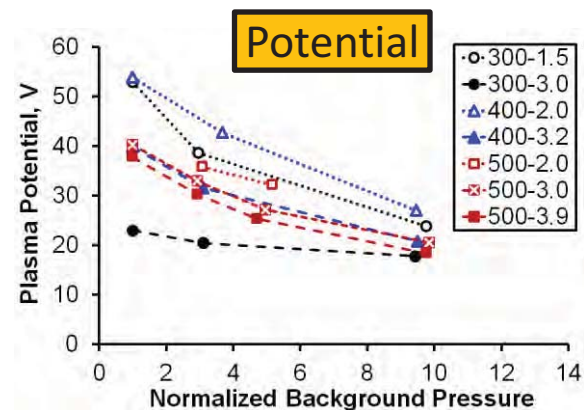
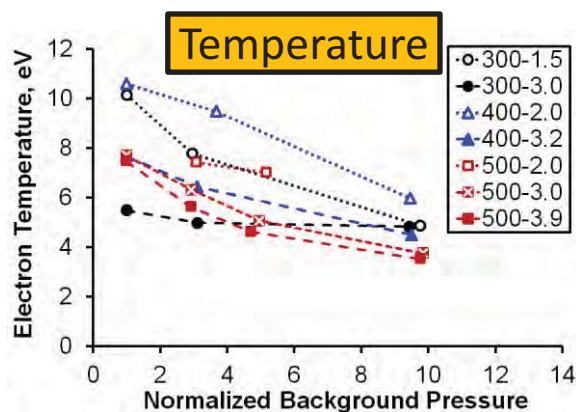
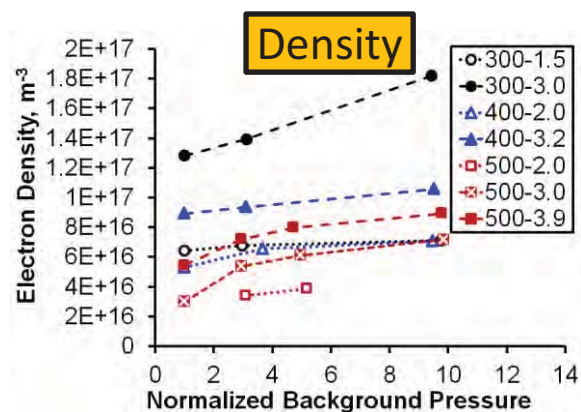




Results: High-speed Langmuir Probe



- Data shown is averaged of the two upstream probes at 10 cm axially downstream of the exit plane

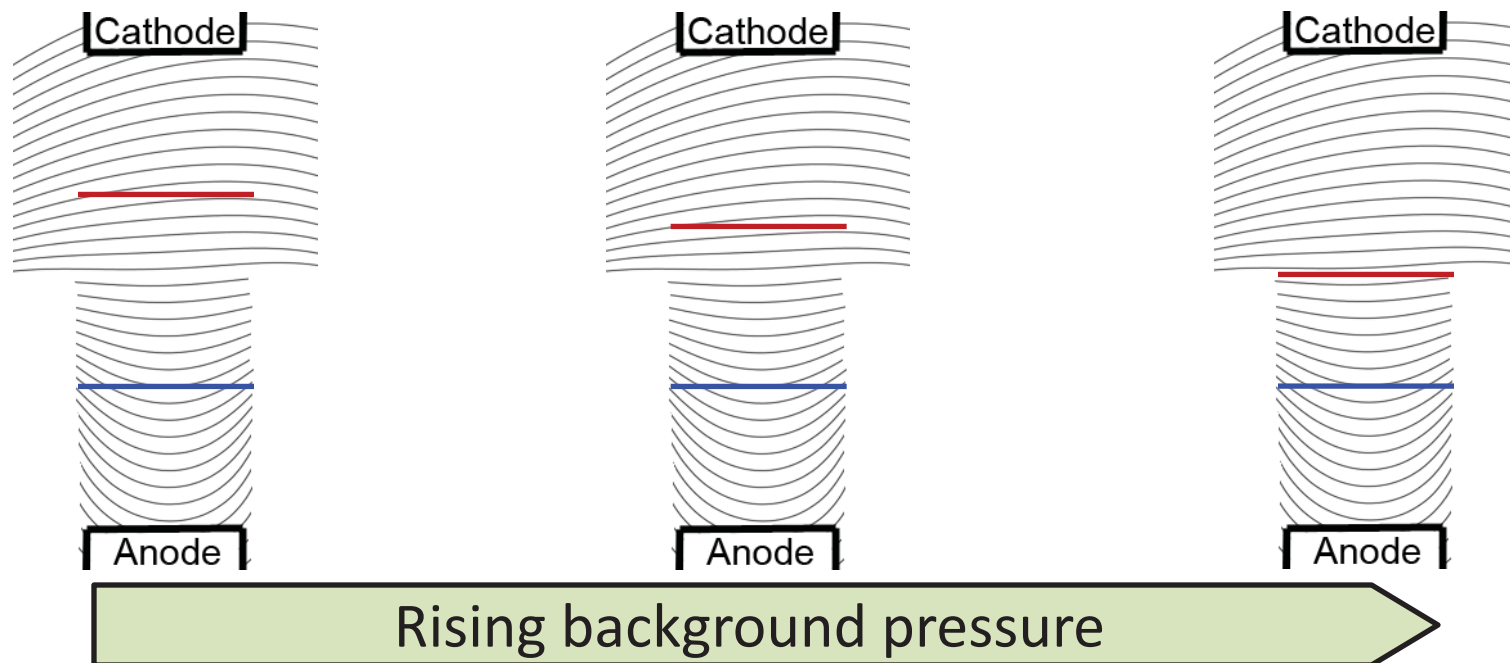




The ionization/acceleration zone is shortening with rising background pressure.



- Assumption: breathing mode enhance electron transport, **upstream** edge fixed by magnetic field and anode
- Downstream edge of I/AZ recedes into channel -> less field lines to cross between **downstream** and **upstream** edges -> lower breathing mode magnitude needed
- At the same time, more field lines to cross between **downstream** edge and cathode



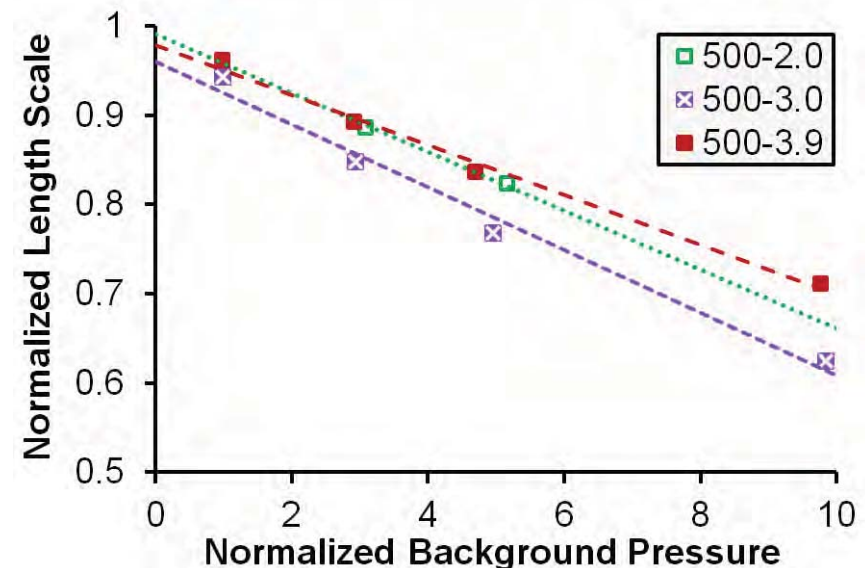
*Red line denote **downstream** edge of I/AZ, blue line denote **upstream** edge



Engineering Analysis: Test Guidelines



- If preceding physical picture is correct, can formulate test guidelines by levying a requirement on the erosion band location
- Assume: I/AZ axial length of few tens mm (from probe and LIF data), impose a requirement that the erosion band shift by no more than a few mm (assumed design margin) -> I/AZ length cannot change by more than 10%
- Axial length scale change of 0.9 -> normalized pressure of 2-3
- Conservative estimate translates this normalized pressure to $1.2e-6$ Torr per mg/s





Dimensional Analysis: Shortening of Ionization/Acceleration Zone

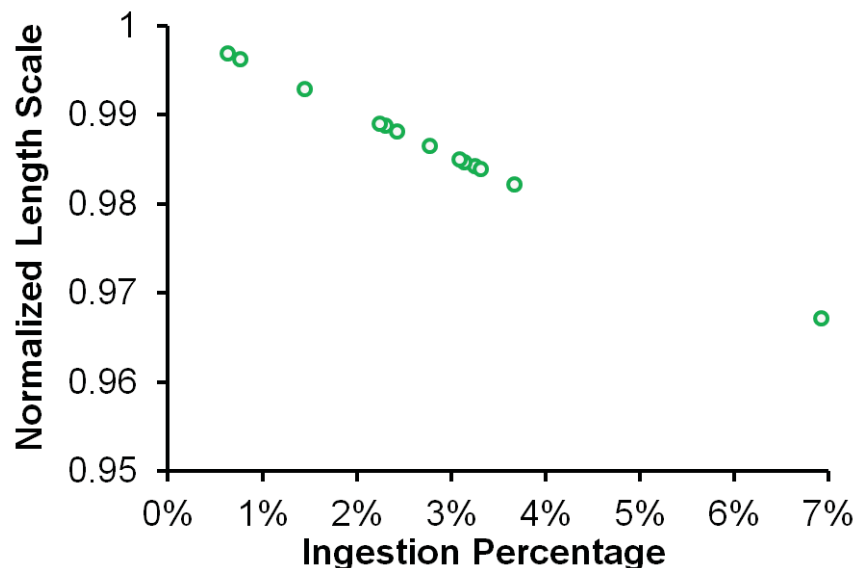


- Try dimensional analysis to discover cause of I/AZ shortening
- Start with relationship between plasma potential and charge density
- Note Laplacian of Φ scales as V/L^2

$$\nabla^2 \Phi = -\rho/\epsilon_0$$

$$L_i \sim \frac{1}{\sqrt{n_e}} \sim \frac{1}{\sqrt{MR_a + MR_i}}$$

- Assume ingested stream has same species composition, propellant utilization and current utilization as main propellant stream
- Use ratio of discharge current to mass flow rate and excess discharge current to calculated ingested mass flow rate
- Predicted change is about one order of magnitude too low





Conclusion



- Combined high-speed camera and Langmuir probe data suggests upstream movement of the ionization/acceleration zone with rising background pressure
- A physical model is proposed where the upstream edge is fixed while the downstream edge recedes into the channel with rising background pressure
 - Data is consistent with assuming that breathing mode oscillation enhances electron transport across magnetic field lines
- Assuming ionization/acceleration zone length inversely proportional to breathing mode frequency (from Fife's work), can convert requirement on erosion band location into requirement on maximum background pressure -> recommend at most **1.2e-6 Torr per mg/s** for HiVHAc EDU2 testing
- Previous work (IEPC-2013-058) suggested **5e-7 Torr per mg/s** for at most 2% increase in measured thrust, which is more stringent



Acknowledgment



- We thank,
 - NASA Science Mission Directorate In-Space Propulsion project for funding this work,
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 - [Kevin L. Blake](#) and [George P. Jacynycz](#) for thruster fabrication, assembly of the test setup, and operation of the vacuum facility.
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Questions

