Paper for AIAA Joint Propulsion Conference, July 27-29, 2015, Orlando, Florida Use of Generalized Fluid System Simulation Program (GFSSP) for Teaching and Performing Senior Design Projects at the Educational Institutions

A. K. Majumdar and A. Hedayat Propulsion Systems Department Marshall Space Flight Center, MSFC, AL 35812

Abstract

This paper describes the experience of the authors in using the Generalized Fluid System Simulation Program (GFSSP) in teaching Design of Thermal Systems class at University of Alabama in Huntsville. GFSSP is a finite volume based thermo-fluid system network analysis code, developed at NASA/Marshall Space Flight Center, and is extensively used in NASA, Department of Defense, and aerospace industries for propulsion system design, analysis, and performance evaluation. The educational version of GFSSP is freely available to all US higher education institutions. The main purpose of the paper is to illustrate the utilization of this user-friendly code for the thermal systems design and fluid engineering courses and to encourage the instructors to utilize the code for the class assignments as well as senior design projects.

1. Introduction

There is increasing use of computational tools for design and analysis in thermal and fluid engineering industries. The engineers fresh from school are generally proficient with the use of software for writing reports or preparing presentations. However, they receive very little training on thermal and fluid analysis software in undergraduate program. One of the reason for such deficiency is the lack of availability of the software that can be integrated with the engineering course. Many of these software are commercial and universities may not have the resources for licensing these software. To fill this gap, NASA/Marshall Space Flight Center have developed an educational version of Generalized Fluid System Simulation Program (GFSSP) which is available free of cost for classroom use in engineering universities at United States.

GFSSP is a finite volume based flow network analysis software. Finite Volume Method (FVM) is extensively used in Computational Fluid Dynamics (CFD) to solve Navier-Stokes equation. Many commercial CFD codes (FLUENT, CFX, and FLOW3D) are based on FVM. FVM is an extension of the control volume analysis technique of classical thermodynamics. In Figure 1a, a typical control volume is shown for mass and energy conservation based on the first law of thermodynamics. Finite volume analysis of fluid network is shown in Figure 1b. A fluid network consists of boundary nodes, internal nodes and branches. Boundary and internal nodes are connected through branches in series or parallel arrangements. Mass and energy conservation equations are solved in internal nodes similar to control volumes of classical thermodynamics. Flowrates are calculated in branches. In control volume analysis, mass flow rates are known. In finite volume

analysis, the mass flow rates are not known and therefore momentum equations are solved in branches to calculate flowrates which are necessary to solve mass and energy conservation equations in the node. In addition to flow energy, heat and work can be added to internal nodes similar to control volumes. Thus the finite volume procedure can model thermodynamics, fluid flow and heat transfer within a unified mathematical framework.



(a)



(b)

Figure 1. (a) Control Volume Analysis in Classical Thermodynamics; (b) Finite Volume Analysis in Fluid Network

This paper presents a brief overview of GFSSP and describes how it is used in a senior undergraduate Thermal Design class at University of Alabama in Huntsville.

2. GFSSP Overview

In GFSSP, a fluid circuit is constructed with boundary nodes, internal nodes and branches (Figure 1b) while the solid circuit is constructed with solid nodes, ambient nodes and conductors. The solid and fluid nodes are connected with solid-fluid conductors. Users must specify conditions, such as pressure, temperature and concentration of species at the boundary nodes. These variables are calculated at the internal nodes by solving conservation equations of mass, energy and species in conjunction with the thermodynamic equation of state. Each internal node is a control volume where there are inflow and outflow of mass, energy and species at the boundaries of the control volume. The internal node also has resident mass, energy and concentration. The momentum conservation equation is expressed in flowrates and is solved in branches. At the solid node, the energy conservation equation for solid is solved to compute temperature of the solid node. Figure 2 shows a schematic and GFSSP flow circuit of a counter-flow heat exchanger. Hot nitrogen gas is flowing through a pipe, colder nitrogen is flowing counter to the hot stream in the annulus pipe and heat transfer occurs through metal tubes. The problem considered is to calculate flowrates and temperature distributions in both streams.



Figure 2. A Typical Flow Network consists of Fluid Node, Solid Node, Flow Branches and Conductors

GFSSP has a unique data structure shown in Figure 3; this allows constructing all possible arrangements of a flow network with no limit on the number of elements. The elements of a flow network are boundary nodes where pressure and temperature are specified, internal nodes where pressure and temperature are calculated, and branches where flowrates are calculated. For conjugate heat transfer problems, there are three additional elements: solid node, ambient node, and conductor. The solid and fluid nodes are connected with solid-fluid conductors.



Figure 3. Data structure of the fluid-solid network has six major elements.

The mathematical closure is described in Table 1. GFSSP uses a pressure based scheme as pressure is computed from mass conservation equation. The mass and momentum conservation equations and thermodynamic equation of state are solved simultaneously by the Newton-Raphson method while energy conservation equations of fluid and solid are solved separately but implicitly coupled with the other equations stated above. Further details of the mathematical formulation and solution procedure are described in reference 1.

GFSSP is linked with two thermodynamic property programs, GASP² and WASP³ and GASPAK⁴, that provide thermodynamic and thermophysical properties of selected fluids. Both programs cover a range of pressure and temperature that allows fluid properties to be evaluated for liquid, liquid-vapor (saturation), and vapor region. GASP and WASP provide properties of 12 fluids. GASPAK includes a library of 36 fluids.

Table 1. Mathematical Closure

Unknown Variables	Available Equations to Solve
1. Pressure	1. Mass Conservation Equation
2. Flowrate	2. Momentum Conservation Equation
3. Fluid Temperature	3. Energy Conservation Equation of Fluid
4. Solid Temperature	4. Energy Conservation Equation of Solid
5. Fluid Mass (Unsteady Flow)	5. Thermodynamic Equation of State

GFSSP has three major parts. The first part is the graphical user interface (GUI), visual thermofluid analyzer of systems and components (VTASC). VTASC allows users to create a flow circuit by a 'point and click' paradigm. It creates the GFSSP input file after the completion of the model building process. GFSSP's GUI provides the users a platform to build and run their models. It also allows post-processing of results. The network flow circuit is first built using three basic elements: boundary node, internal node, and branch.



Figure 4 - GFSSP's Program Structure showing the interaction of three major modules

3. Utilization of GFSSP in Senior Design Project

Utilization of GFSSP has been incorporated in the syllabus of Design of Thermal Systems course, a senior level class, at the University of Alabama in Huntsville. The utilization of GFSSP includes tutorials, homework assignments, and a design project. For the design project, the class divided into multiple teams, each having 3 or 4 team members. The students were encouraged to form and select a team lead who oversaw overall progress of

the project. Each team is required to submit a project report and make a presentation. A tutorial example and brief review of the design project assignment submitted by one of the team are presented in the following sections.

3.1 Tutorial Example: Pressurization of a Propellant Tank

This example demonstrates the use of GFSSP's unsteady formulation by predicting the pressure and temperature history during the blowdown of a pressurized tank. A schematic of a propellant pressurization system is shown in Figure 5. It is assumed that initially the ullage, space is filled with pressurant, helium (He), at propellant temperature. As the warm pressurant enters the ullage space, it mixes with the cold ullage gas and the temperature of the ullage gas starts to increase due to mixing and compression. Initially, the walls of the tank are also at propellant temperature. Heat transfer from the ullage gas to the propellant, liquid oxygen (LOX), and the tank wall and mass transfer from the propellant to the ullage start immediately after the pressurant begins flowing into the tank. Propellant flows from the tank to the engine under the influence of ullage pressure and gravitational head in the tank. In this model, condensation of propellant vapor has been neglected.



Figure 5. Schematic of propellant tank pressurization system.

GFSSP Model

A five-node pressurization system, as shown in Figure 6(a), was developed. Helium at 95 psia and 120 °F enters the ullage space, which is initially filled with helium at 67 psia and -264 °F. Node 2 represents the ullage space, which has an initial volume of 25 ft³. A pseudo boundary node (node 3) has been introduced to exert ullage pressure on the initial propellant volume of 475 ft³, which is represented by node 4. The pressure at the pseudo boundary node is calculated from the ullage pressure and gravitational head and is the driving force to supply the propellant to the engine. This pressure is calculated at the beginning of each time step. Branch 12 models the tank inlet, branch 34 represents the

propellant surface, and branch 45 represents the line to the engine. All three branches were modeled using a Flow Through a Restriction. The flow coefficient of branch 45 is adjusted to restrict the propellant flow such that all propellant is expelled from the tank over the course of the run. In this test model, the engine inlet pressure was set at 50 psia. Figure 6(b) shows how the model looks in VTASC. Figure 7 shows the VTASC tank pressurization dialog and inputs for example 10.



Figure 6. Simple pressurization system test model: (a) Detailed model schematic and (b) VTASC model.

Results and Discussions

The pressurization system transient test model was run for 200 seconds (s) with 0.1s time step. Figure 7 shows both the ullage pressure and tank bottom pressure histories for the test model. After an initial pressure rise due to a 'ramping up' transient effect, both pressures begin a slow but steady decline for the remainder of the run. It should be noted that tank bottom pressure was calculated by adding ullage pressure with pressure due to the gravitational head. Figure 7 shows that as the gravitational head decreases, the ullage and tank bottom pressures slowly converge until all propellant is drained from the tank. The slow decline in ullage pressure is mainly due to the expanding ullage volume.

Figure 8 shows the histories for the ullage temperature and the tank wall temperature. This figure shows that the tank wall temperature rises 32 °F over the course of the model run. It

reveals that the 120 °F helium gas entering the tank has an increasing effect on the tank wall as propellant is drained from the tank and the wall surface area exposed to the warmer ullage gas grows. This effect is somewhat dampened, however, because the heat gained by the wall is conducted to the portion of the tank that is submerged in LOX, which acts as a heat sink. The ullage temperature rises 192 °F during the first 60 seconds of tank pressurization before beginning a slow decline for the remainder of the simulation. This large initial temperature rise is primarily due to the mixing of hot helium gas with the relatively cold gas present in the ullage. The decline in temperature is a result of expansion due to a continuous increase of the ullage volume.



Figure 7. Ullage and tank bottom pressure history.

Figure 8. Ullage and tank wall temperature history.

Helium flow rate into the tank is shown in Figure 9. The helium flow rate was found to drop initially as the start transient takes place, which is consistent with the 'ramp up' effect noted in Figure 7. Then the flow rate begins to gradually increase as ullage pressure drops due to the expanding ullage volume. LOX flow rate into the engine is shown in Figure 10. The LOX flow rate curve mirrors the ullage and tank bottom pressure curves, rising through an initial start transient to a peak value and then declining for the remainder of the run as tank pressure drops.



Figure 9. Helium mass flow rate history.



Figure 10. LOX mass flow rate history.

3.2 Design Project Assignment

The objective of the design project was to develop a design for a chilled-water airconditioning system. This design supplies water from a chiller to air handling units (AHUs) in a four-story office complex, and meets design requirements set forth in the design project description. These requirements govern the flow rates in different branches of the system, the maximum allowable velocity through the pipes, the pressure drops across components, and the pipe length. In chilled-water air-conditioning systems water is pumped through a chiller, into a network of pipes that supply water to the AHUs. Air is passed through the AHUs and the chilled water extracts heat from the air, cooling the airflow, which is then distributed to the different rooms on the floor. These air conditioning systems are commonly used in large buildings, such as hotels or office complexes, where each floor has its own AHU, allowing different floors to be cooled individually. The basic flow circuit is shown in Figure 11. The "P" component represents the pump, and the "C" component represents the chiller.



Figure 11. Basic flow circuit shown in project requirements.

Design Requirements

Several design requirements had to be met for this chilled-water flow network, which are outlined in Table 2. These requirements specify flow rates, flow velocities, pressure drops, and pipe lengths. For the prescribed requirements, several quantities had to be determined. These quantities were:

- 1. Floor pipe size
- 2. Supply and return pipe size
- 3. Pressure rise across the pump
- 4. Pump horsepower for an efficiency of 55%
- 5. Cost to operate the system for a day, assuming a price of \$0.10 per KW-hr

Requirement Number	Title	Description	
1	Flow Rate	Each floor requires a chilled-water flow rate of 18,000 lbm/hr	
2	Flow Velocities	All flow velocities are not to exceed 5 ft/s	
3	Floor Spacing	Each floor is 12 feet higher than the one below	
4	Supply and Return Pipe Sizing	The supply and return pipes must all be the same diameter (Schedule 80)	
5	Supply Pipe Length	The total length of supply pipe running from the exit of the bottom floor, through the pump and chiller, and up to the entrance of the top floor is 300 feet. (See Figure #1).	
6	Floor Pipe Sizing	The floor pipes must all be the same diameter (Schedule 80)	
7	Floor Pipe Length	Each floor must have 100 feet of pipe	
8	Floor Valves	Each floor must have two gate valves	
9	Air Handling Unit Pressure Drop	The pressure drop across each AHU must be 50 psi	
10	Chiller Pressure Drop	The pressure drop across the chiller must be 100 psi	

Table 2: Air-Conditioning System Requirements.

Assumptions

- 1. Neglected losses due to couplings/welds required to join sections of steel pipe, which typically come in 10-12 foot lengths, to obtain the lengths specified in the problem (100 feet of 2" and 300 feet of 4", etc).
- 2. Assumed a <0.5% deviation from values specified in project requirements was acceptable.

GFSSP Model

The flow circuit, illustrated in Figure 11, was created in GFSSP and is shown in Figure 12.



Figure 12. GFSSP flow circuit with numbered nodes and branches.

GFSSP model simulation of the flow circuit yielded a set of design results for pipe sizes, pressure drops, flow rates, flow velocities, required horsepower, and cost of operation. These results are summarized in the Table 3, and demonstrate that the design fully meets all project requirements. Overall, as seen from Table 3, after defining and/or solving for all unknown quantities, the results of the design closely meet the design requirements. The results of the GFSSP model are within 0.3% percent difference of the ideal design requirements.

	GFSSP Value	Requirements Volue	Notes
Courte of Determ Direction		value	
(Schedule 80)	3.826 in		
Floor Pipe Size (Schedule 80)	1.939 in		
Pressure Rise Across Pump	154.0 psi		
Pump Power Required	23.5 hp (17.5 kW)		At 55% efficiency
Cost of Operation (per 24 hrs)	\$42		\$0.10 per kW-hr for 24 hrs
Mass Flow Rate in Pipe 1011 (4 th Floor)	18036 lbm/hr	18000 lbm/hr	% Difference: 0.2 %
Mass Flow Rate in Pipe 3031 (3 rd Floor)	18018 lbm/hr	18000 lbm/hr	% Difference: 0.1 %
Mass Flow Rate in Pipe 3738 (2 nd Floor)	18018 lbm/hr	18000 lbm/hr	% Difference: 0.1 %
Mass Flow Rate in Pipe 4344 (1 st Floor)	18040 lbm/hr	18000 lbm/hr	% Difference: 0.22 %
Mass Flow Rate in 300' Supply Pipe	72108 lbm/hr	72000 lbm/hr	% Difference: 0.15 %
AHU 4 Pressure Drop	49.96 psi	50 psi	% Difference: 0.08%
AHU 3 Pressure Drop	49.86 psi	50 psi	% Difference: 0.28%
AHU 2 Pressure Drop	49.86 psi	50 psi	% Difference: 0.28%
AHU 1 Pressure Drop	49.96 psi	50 psi	% Difference: 0.08%
Chiller Pressure Drop	100.1 psi	100 psi	% Difference: 0.1%
Fluid Velocity in Pipe 1011 (4 th Floor)	3.917 ft/sec	<5 ft/s	Meets Requirement
Fluid Velocity in Pipe 3031 (3 rd Floor)	3.913 ft/sec	<5 ft/s	Meets Requirement
Fluid Velocity in Pipe 3738 (2 nd Floor)	3.913 ft/sec	<5 ft/s	Meets Requirement
Fluid Velocity in Pipe 4344 (1 st Floor)	3.917 ft/sec	<5 ft/s	Meets Requirement
Velocity in 300' Supply Pipe	4.022 ft/sec	<5 ft/s	Meets Requirement

Table 3.	Project	Design	Results.
1 4010 01	1101000	Dongin	1 CO GILO

4. Conclusions

This paper demonstrates how a system level, user friendly network flow analysis code can be integrated in a senior level thermal design class. The intent was to introduce a state of the art computational tool to perform a real world technical task. The introduction of GFSSP was done through lectures, tutorials and senior design project. The authors' experience of using GFSSP in the class was very positive mainly due to very positive feedback from students about their learning experience while performing the project. GFSSP is available free of cost to all universities in the United States from NASA/MSFC's Technology Transfer Office.

Acknowledgement

This work was supported by National Institute of Rocket Propulsion (NIRPS) at NASA/ MSFC. The authors wish to thank the STI Publication Office of NASA/MSFC for the preparation of this manuscript.

References

- A.K. Majumdar, A.C. LeClair, R. Moore, P.A. Schallhorn, Generalized Fluid System Simulation Program, Version 6.0, NASA/TM—2013–217492 (October 2013) <<u>https://gfssp.msfc.nasa.gov>.</u>
- Hendricks, R.C.; Baron, A.K.; and Peller, I.C.: "GASP A Computer Code for Calculating the Thermodynamic and Transport Properties for Ten Fluids: Parahydrogen, Helium, Neon, Methane, Nitrogen, Carbon Monoxide, Oxygen, Fluorine, Argon, and Carbon Dioxide," NASA TN D-7808, NASA Lewis Research Center, Cleveland, OH, February 1975.
- 3. Hendricks, R.C.; Peller, I.C.; and Baron, A.K., "WASP A Flexible Fortran IV Computer Code for Calculating Water and Steam Properties," NASA TN D-7391, NASA Lewis Research Center Cleveland, OH, November 1973.
- 4. User's Guide to GASPAK, Version 3.20, Cryodata, Inc., November 1994.