#### NASA/TM-2015-218781/Volume II/Part 2 NESC-RP-12-00783





### Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading

### Appendices

Steven J. Gentz/NESC Langley Research Center, Hampton, Virginia

David O. Ordway, David S. Parsons, Craig M. Garrison, C. Steven Rodgers, and Brian W. Collins Marshall Space Flight Center, Huntsville, Alabama

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### **Appendices**

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National Aeronautics and Space Administration

Langley Research Center Hampton, Virginia 23681-2199

#### Acknowledgments

The team would like to acknowledge the following for providing their expertise:

Dr. Curt Larsen/Johnson Space Center (JSC), of the NESC, for championing the Pyroshock Characterization of Composites task.

Mr. Ken Johnson/MSFC for his tireless support statistically analyzing the post-processed acceleration data.

Mr. Lee Allen/MSFC for supporting Ken Johnson in statistically analyzing the post-processed data

Ms. Barbara Breithaupt/MSFC for post-processing the pseudo-velocity data.

Mr. Justin Jackson/MSFC for providing his expertise in composite test panel fabrication.

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Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading			

# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

**Volume II: Appendices** 

Part 2

**April 16, 2015** 

NESC Request No.: TI-12-00783

NO PORTOR DE LA CONTRACTOR DE LA CONTRAC	NASA Engineering and Safety Center Technical Assessment Report	Document #:  NESC-RP- 12-00783	Version: 1.0
Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading			

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_	Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading			

### Appendix B. Pyroshock Test Reports (continued)

#### **B3.** Group II (Sandwich Panel Tests)

The test report documenting the test results for the Group II composite sandwich panel tests are documented in the attachment below.

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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

National Aeronautics and Space Administration

George C. Mershall Space Flight Center

Marshall Space Flight Center, AL 35812



Reply to Attn of: ET40-15-007

TO:

EV32/David O. Ordway

FROM:

ET40/Timothy C. Driskill

SUBJECT:

Composite Materials Pyroshock Development Test, Group II - Tests 11-18.

January 27, 2015

NESC-DEV-13-026

REF:

ED73.1

The composite sandwich panel test articles were tested in the ET40 Pyrotechnic Shock Facility, building 4619, room 170. Testing was completed on August 19, 2013. The test was run in accordance with Test and Checkout Procedure, (TCP) NESC-DEV-13-026. Eight tests were run on 8 different composite sandwich panels.

The accelerometer test setup is shown in the photographs section of this report. The aluminum LSC panel did not completely sever in Group II Test No. 11, panel 11. No visual damage to the test articles was noted.

Please direct any questions or comments regarding this test to Mr. Craig Garrison at (256) 544-7197 or craig.garrison@nasa.gov.

Timothy C. Driskill

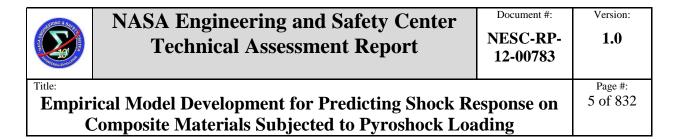
Branch Chief

Structural Dynamics Test Branch

Enclosure

cc:

ET01/File (w/o enclosure) C105/Steven J. Gentz ES22/David S. Parsons



### NESC-DEV-13-026 Composite Materials Shock Test

**General Test Setup** 

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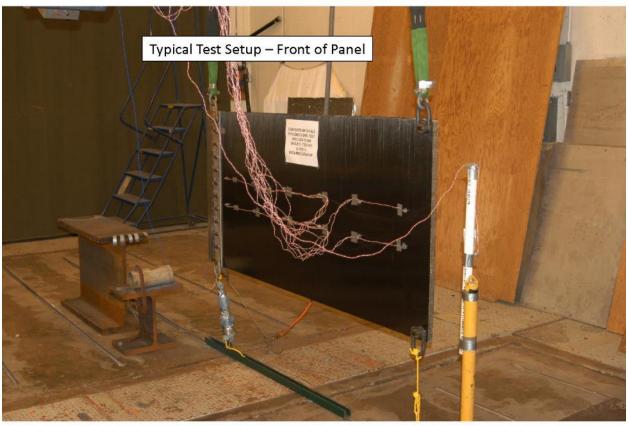
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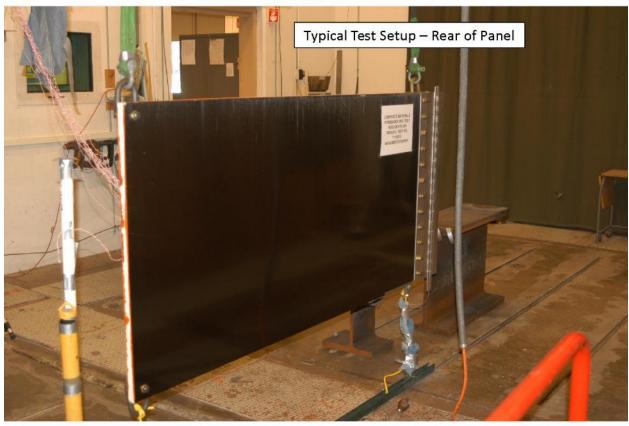
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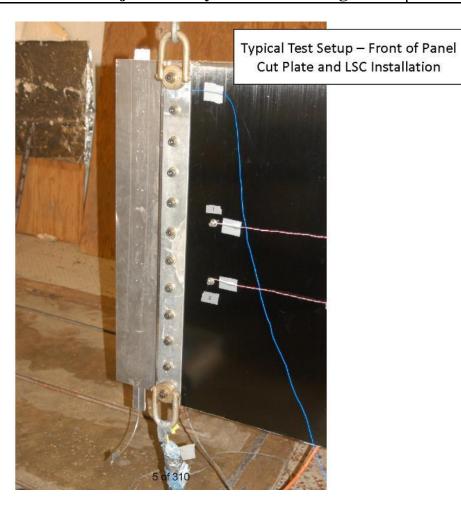
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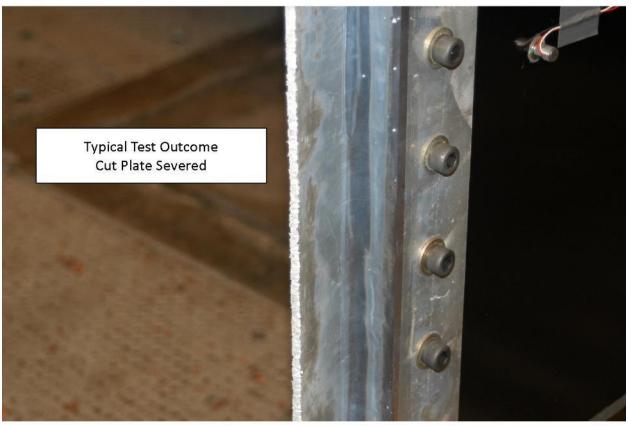


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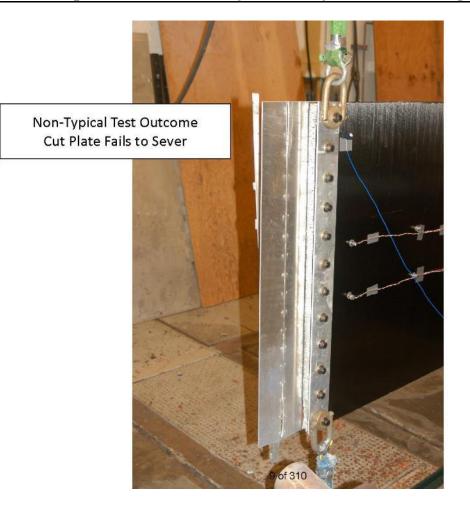
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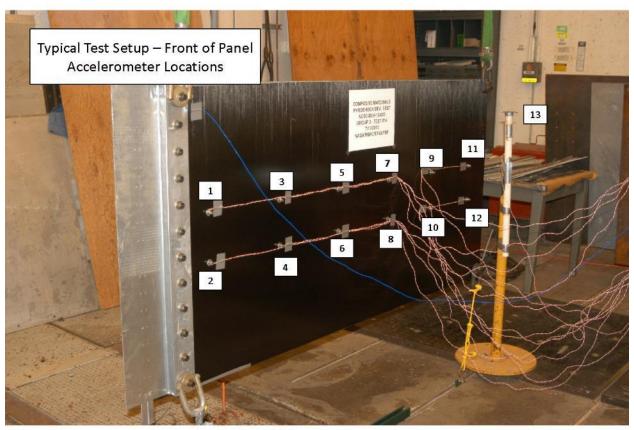
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

Test Matrix As Run			
Test	Date	Panel ID	Test ID
1	5-31-13	0332A011	Group 2 Test 11
2	6-28-13	0332A014	Group 2 Test 12
3	7-16-13	0332A013	Group 2 Test 13
4	7-17-13	0332A012	Group 2 Test 14
5	7-18-13	0332A016	Group 2 Test 15
6	7-23-13	0332A018	Group 2 Test 16
7	8-9-13	0332A015	Group 2 Test 17
8	8-19-13	0332A017	Group 2 Test 18

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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

Panel Location	Set 1 Accelerometers	Set 2 Accelerometers
1	350C02 31334	350D02 43026
2	350D02 43026	350C02 31340
3	350D02 43028	350C02 31338
4	350C02 31331	350D02 43028
5	350C02 31328	350D02 43029
6	350D02 43029	350C02 31333
7	350D02 43179	350C02 40292
8	350C02 31351	350D02 43179
9	350C02 31330	350D02 43180
10	350D02 43180	350C02 40295
11	350D02 43181	350C02 31336
12	350C02 40274	350D02 43181
13	350B02 11439	350B02 11439
Set 1 used in tests 1, 3, 7, and 8 Set 2 used in tests 2, 4, 5, and 6		

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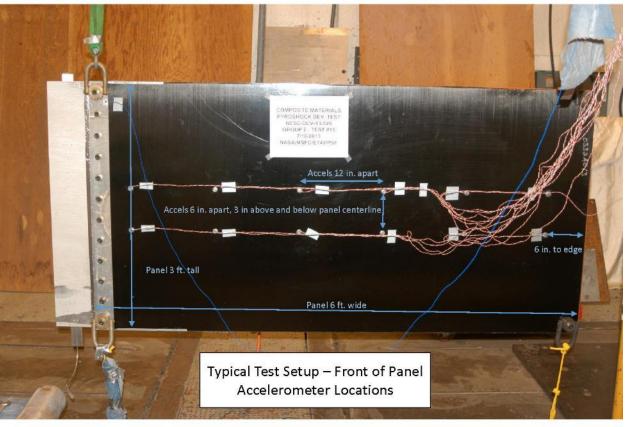
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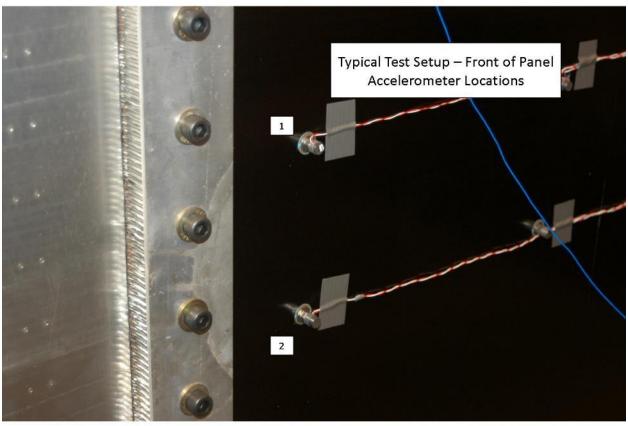
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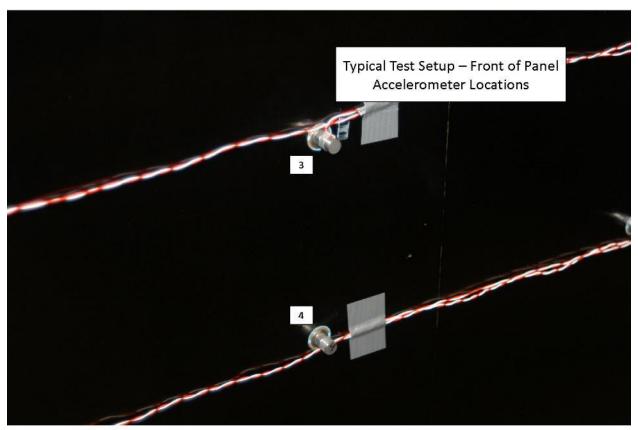
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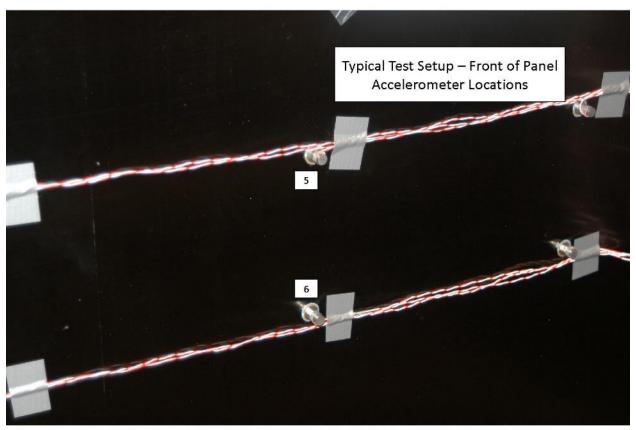
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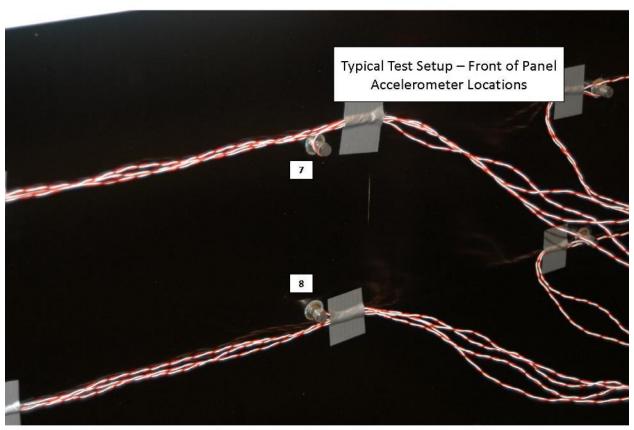
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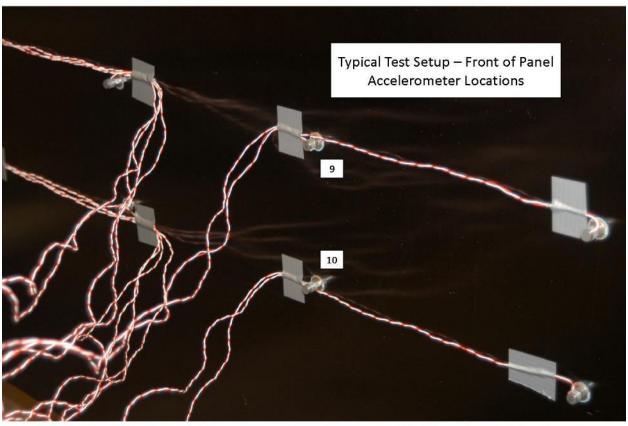
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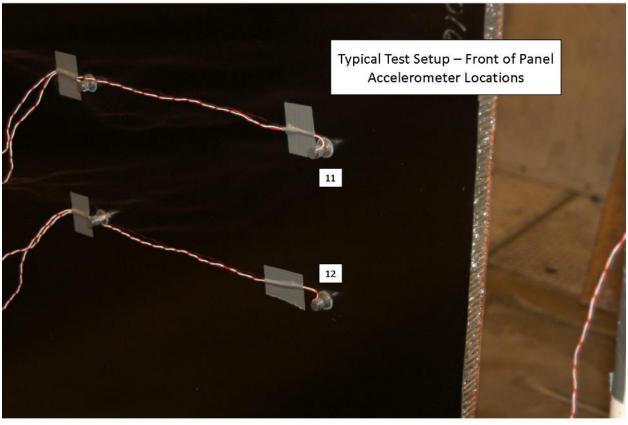
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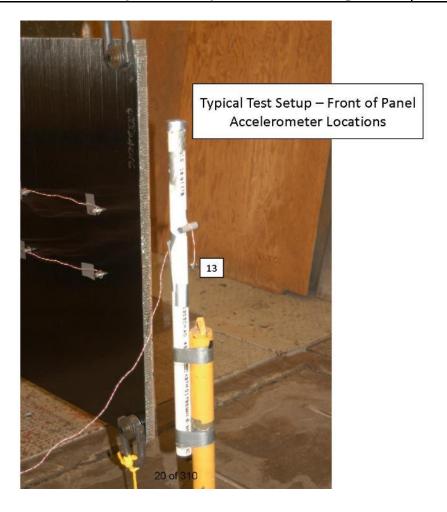
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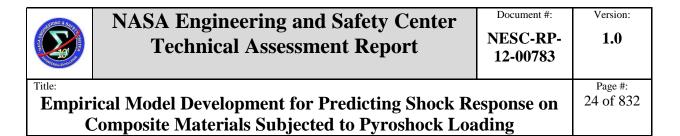
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Test #1 Setup

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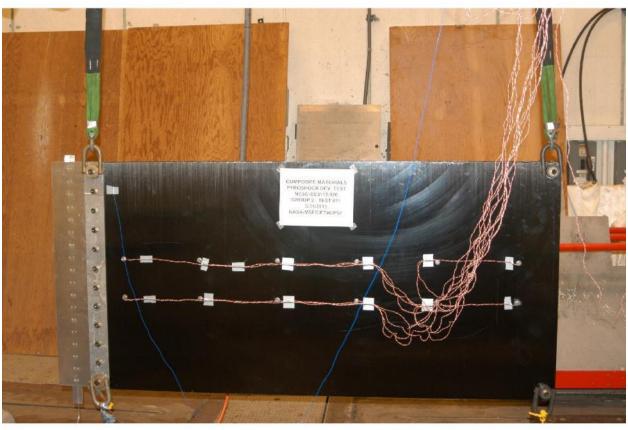
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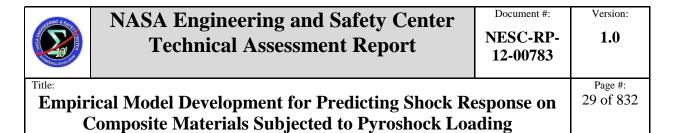
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### NESC-DEV-13-026 Composite Materials Shock Test

Test #2 Setup

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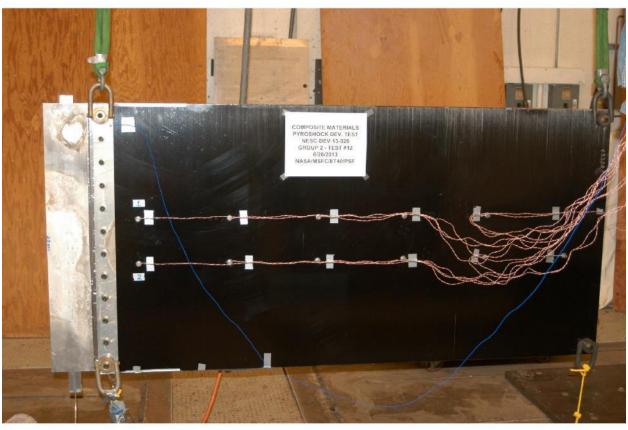
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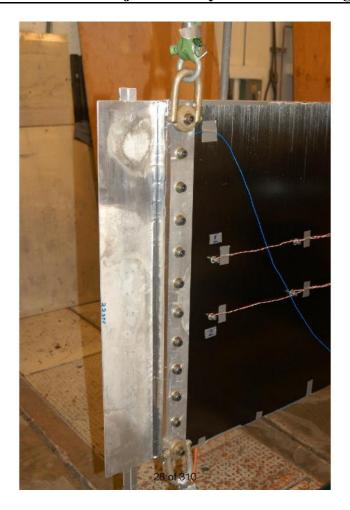


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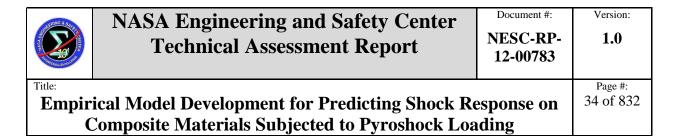
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## NESC-DEV-13-026 Composite Materials Shock Test

Test #3 Setup



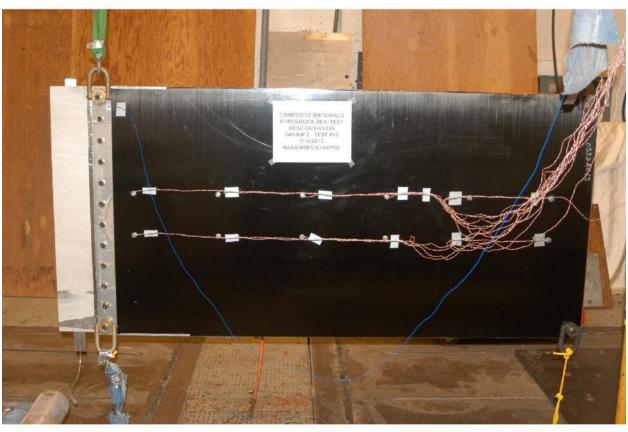
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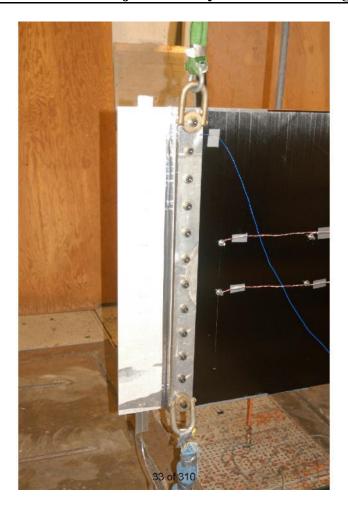
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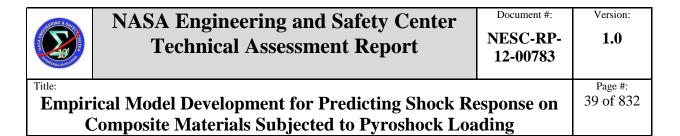
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## NESC-DEV-13-026 Composite Materials Shock Test

Test #4 Setup



Title:

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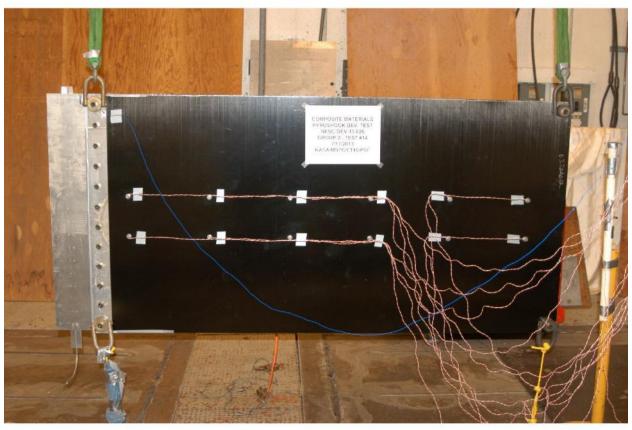
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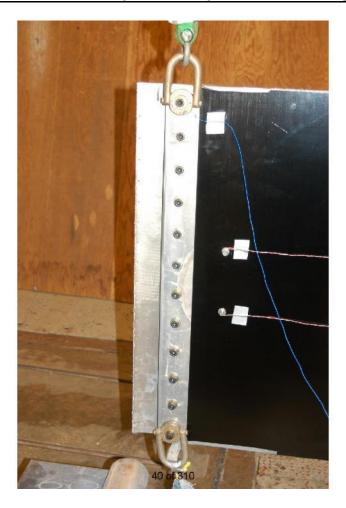


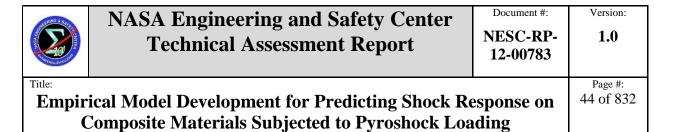
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## NESC-DEV-13-026 Composite Materials Shock Test

Test #5 Setup



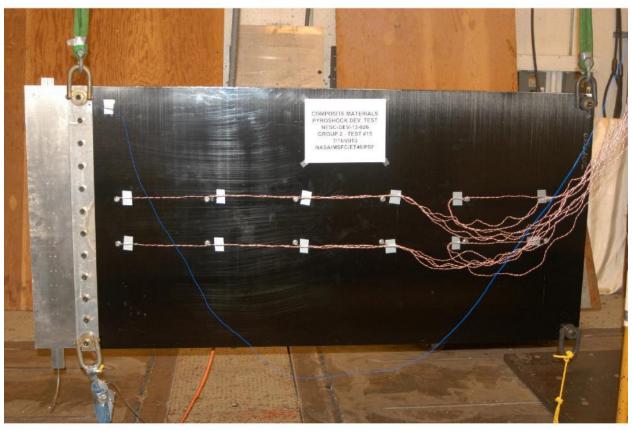
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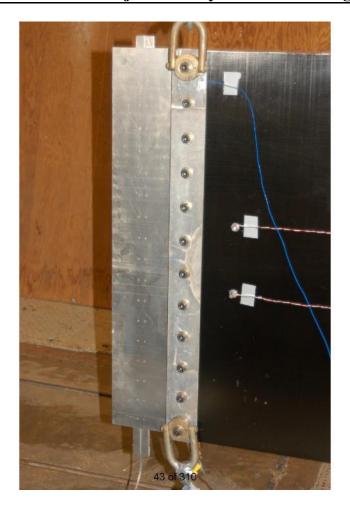
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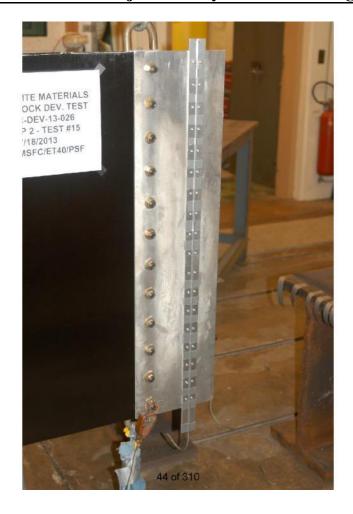
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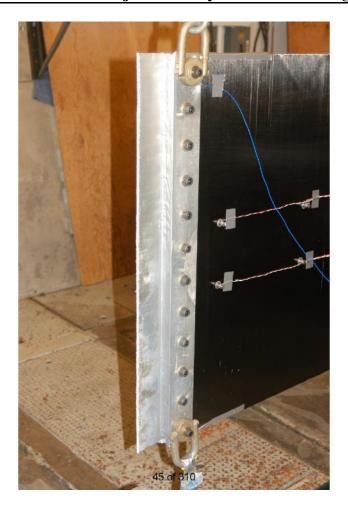
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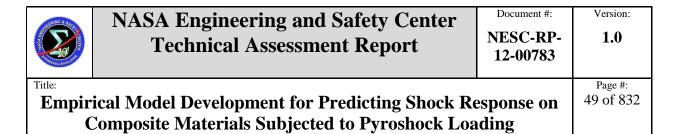
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Test #6 Setup



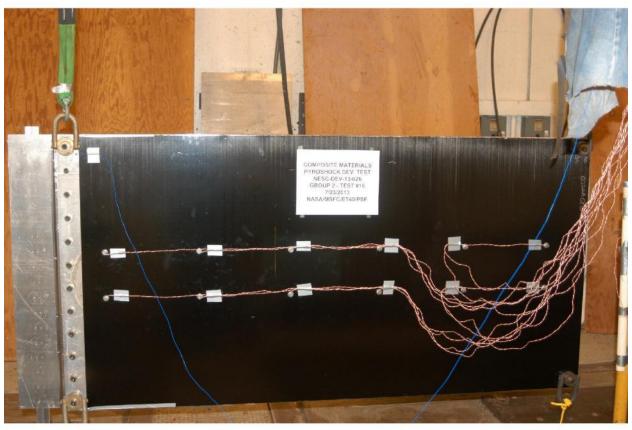
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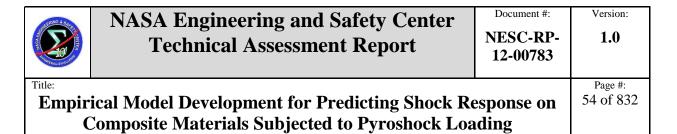
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Test #7 Setup

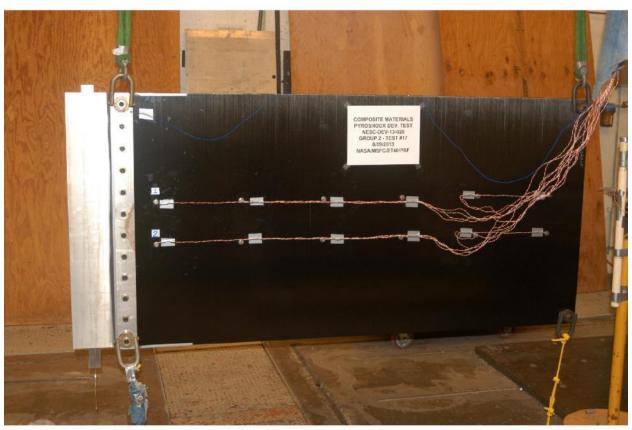


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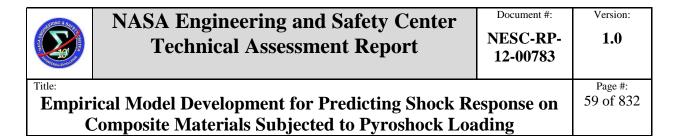
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Test #8 Setup

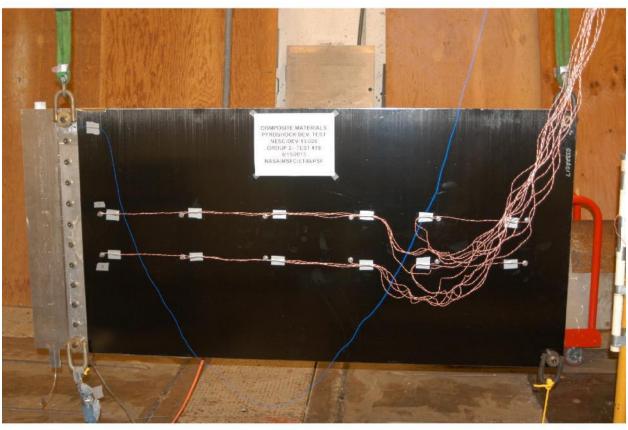


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## NESC-DEV-13-026 Composite Materials Shock Test

### Test #1 Accelerometer Data Panel 0332A011



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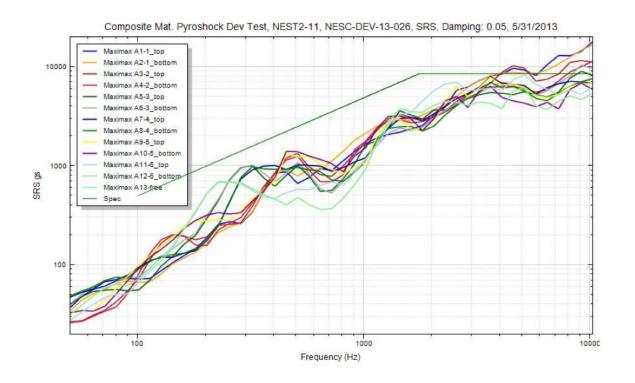
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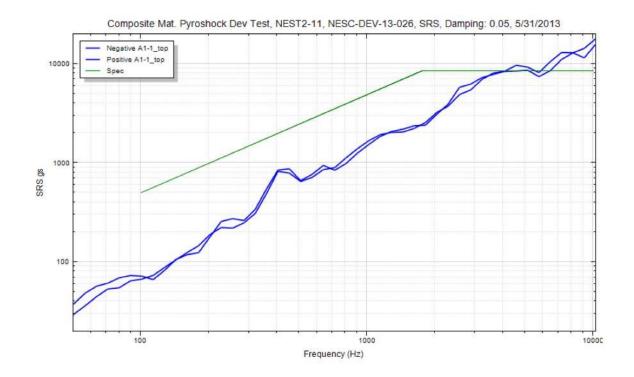
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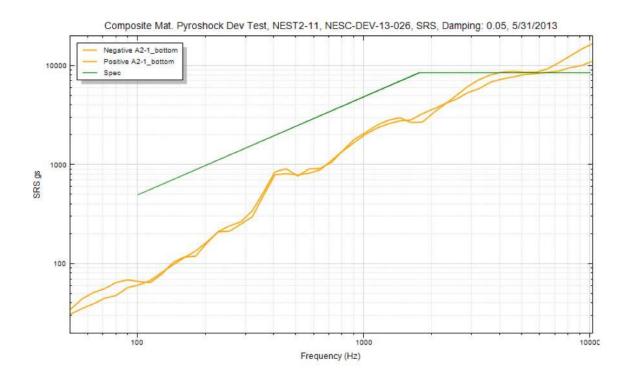
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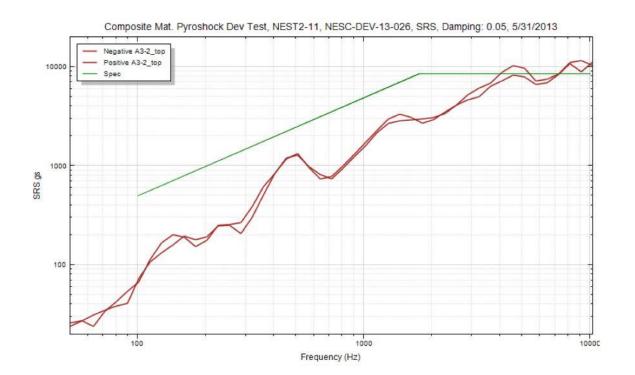
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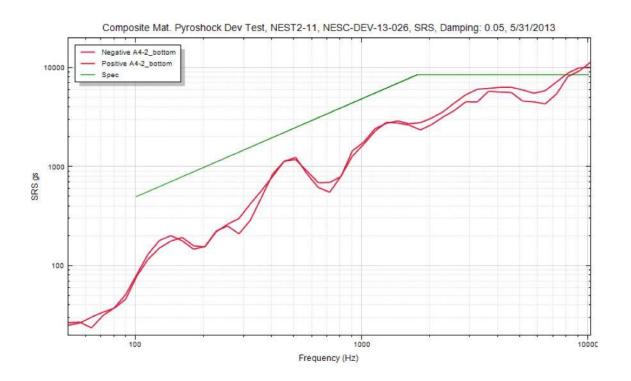
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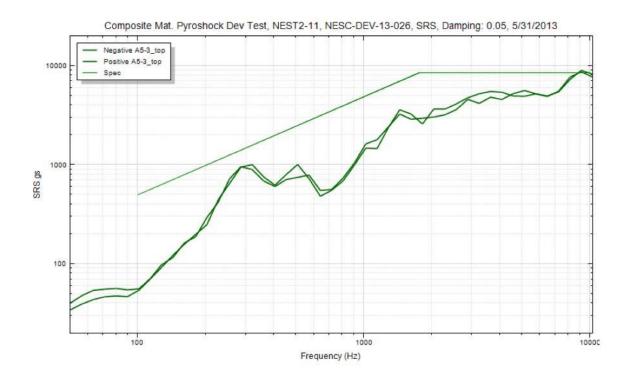
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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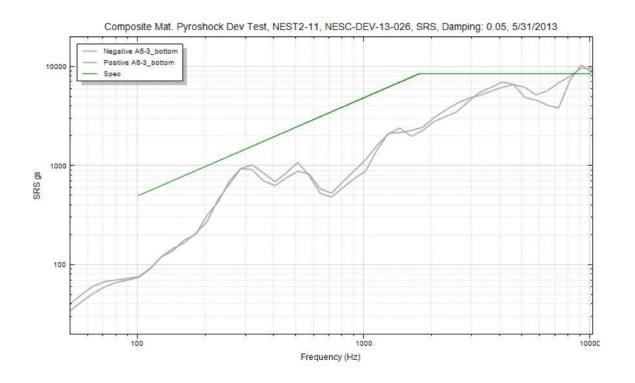
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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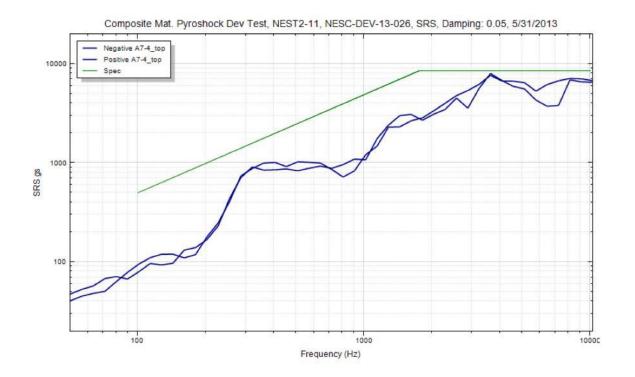
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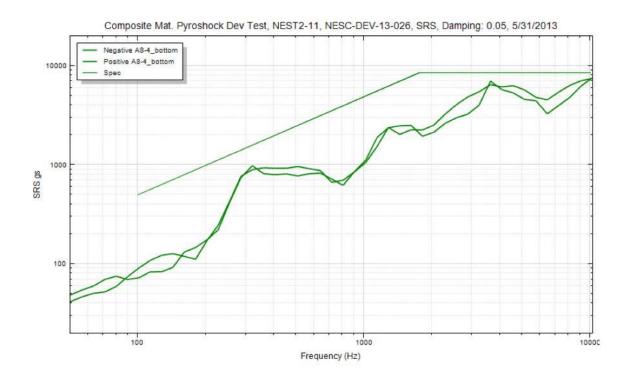
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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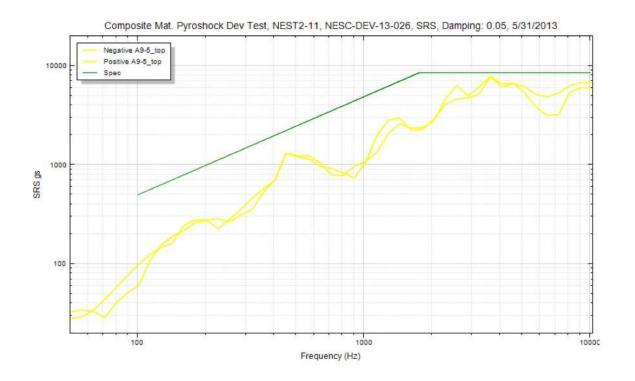
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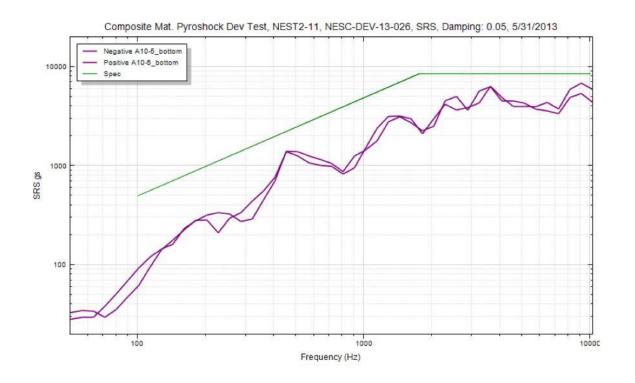
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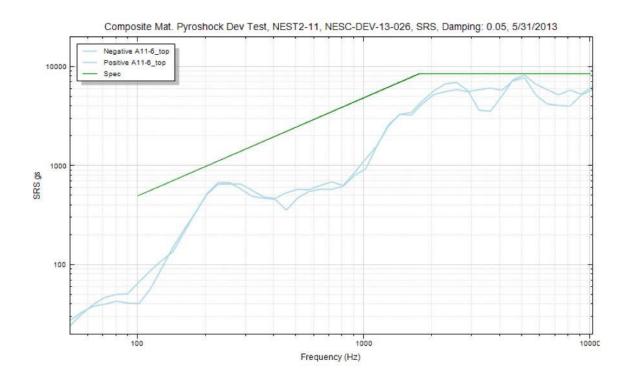
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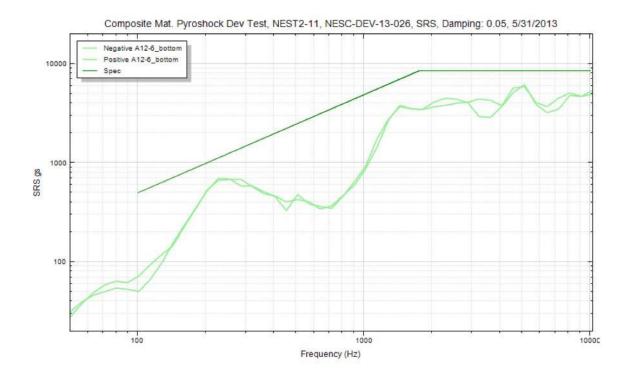
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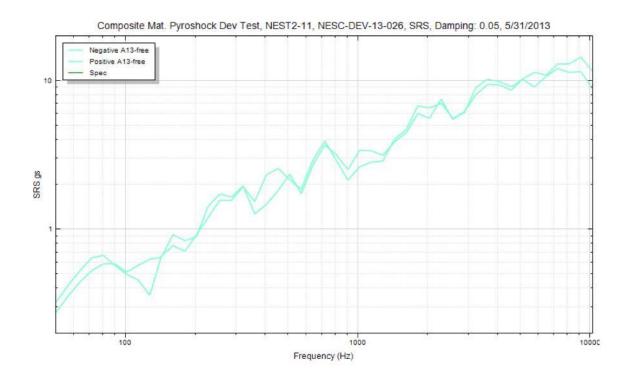
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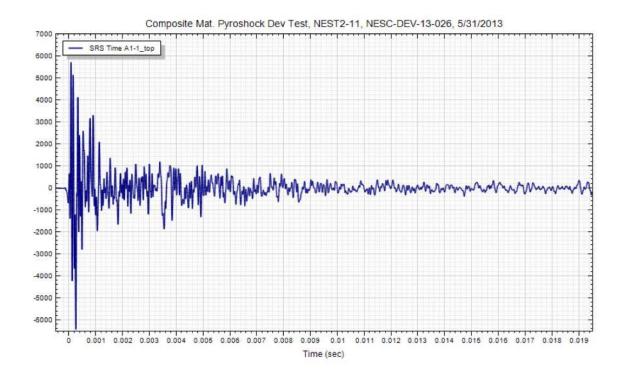
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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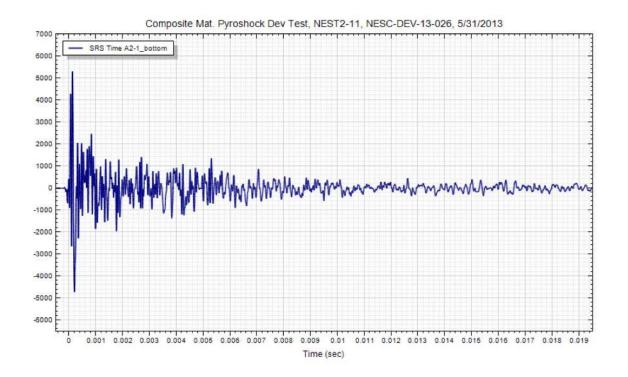
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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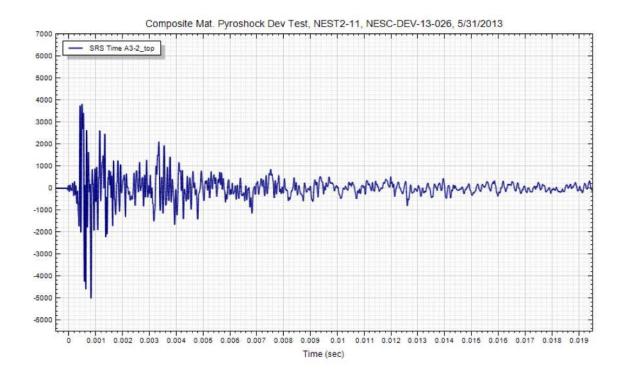
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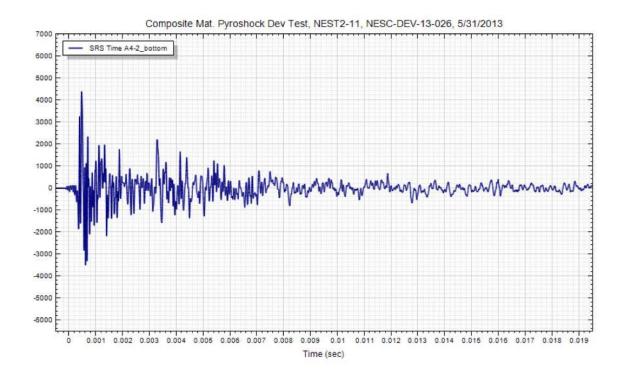
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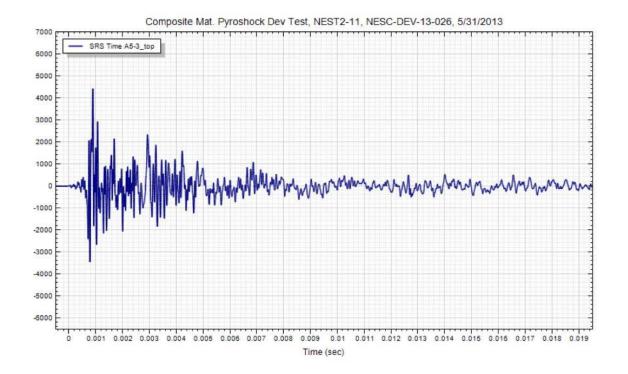
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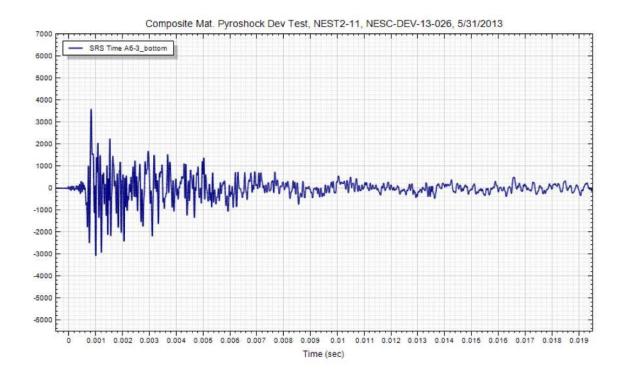
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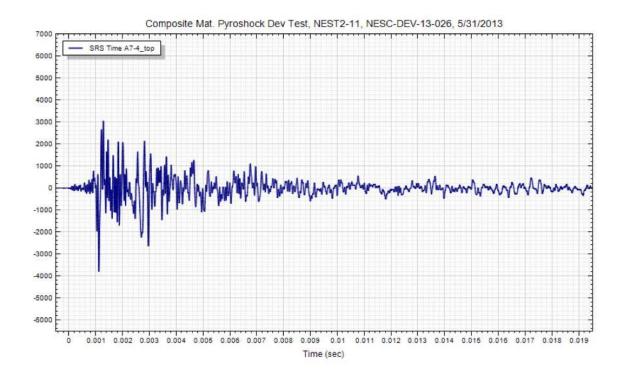
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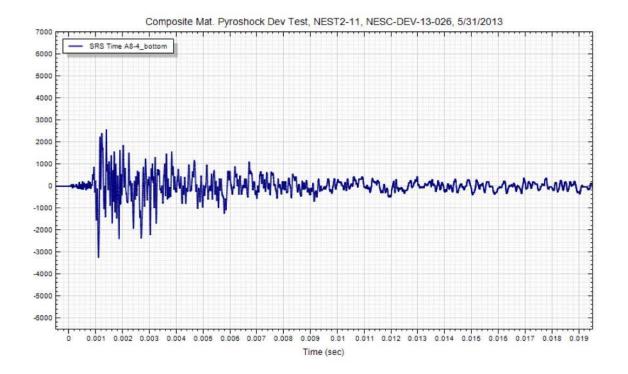
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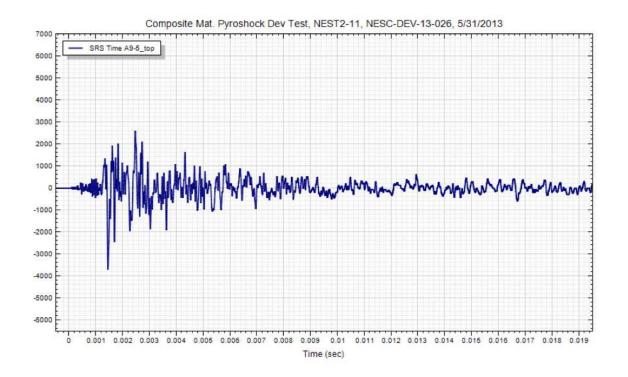
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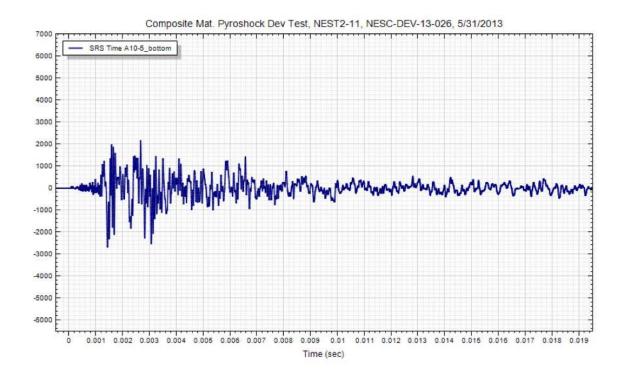
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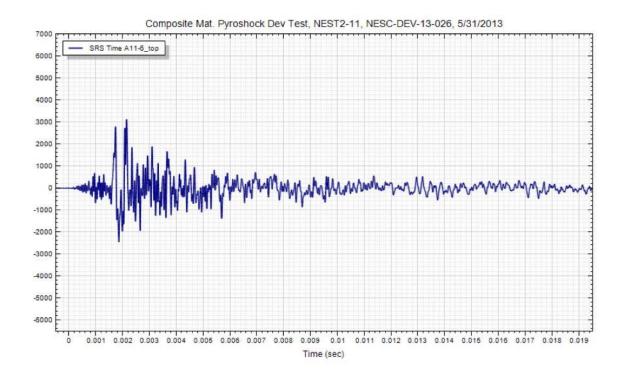
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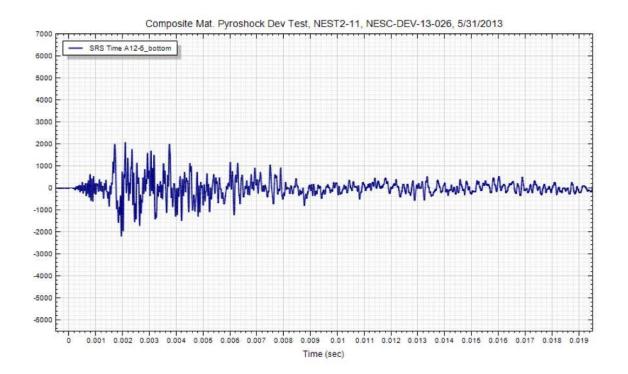
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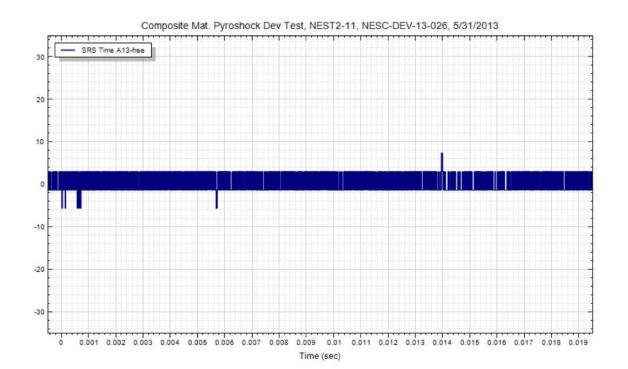
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-026 Composite Materials Shock Test

### Test #2 Accelerometer Data Panel 0332A014



NESC-RP-12-00783

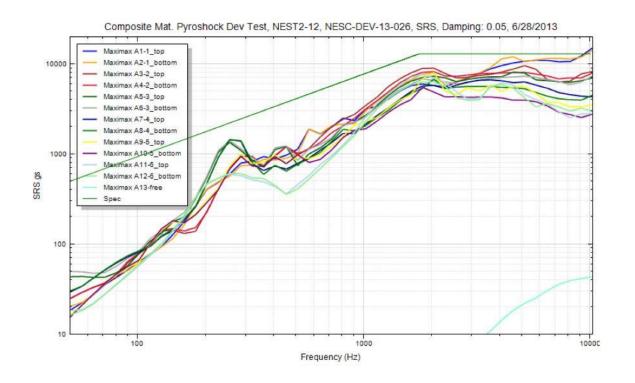
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

Composite Mat. Pyroshock Dev Test, NEST2-12, NESC-DEV-13-026, SRS, Damping: 0.05, 6/28/2013

Negative A1-1\_top
Positive A1-1\_top
Spec

1000
Frequency (Hz)



NESC-RP-12-00783

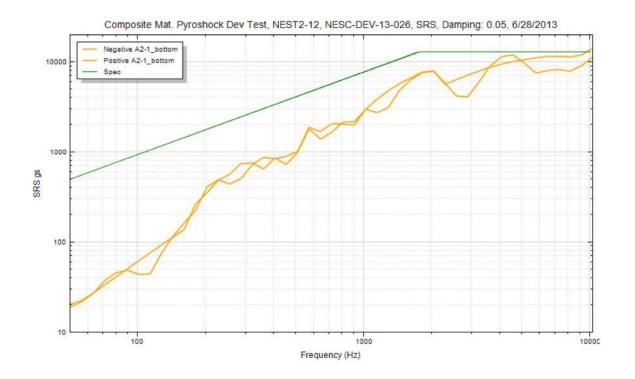
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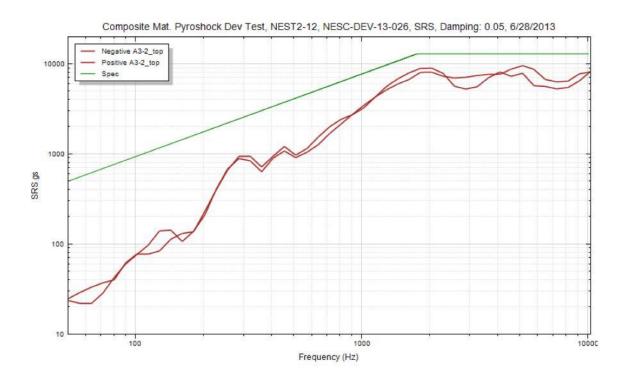
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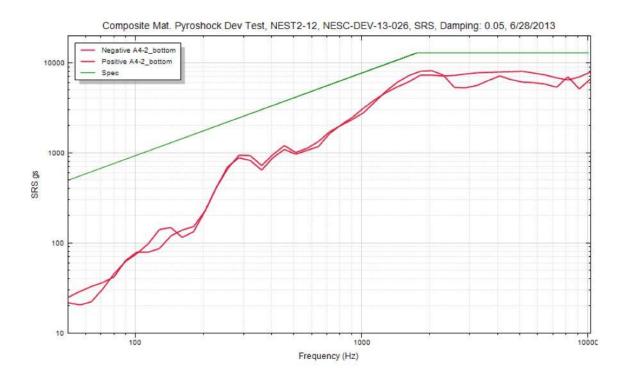
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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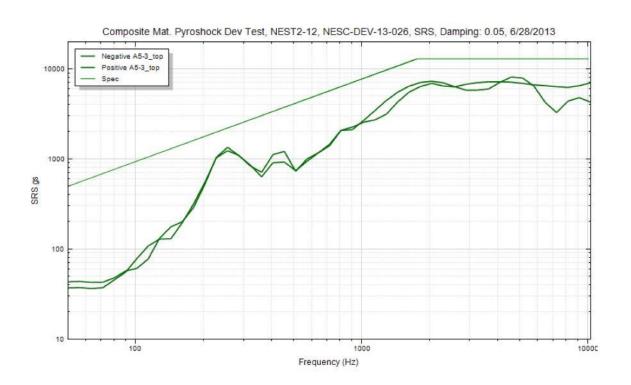
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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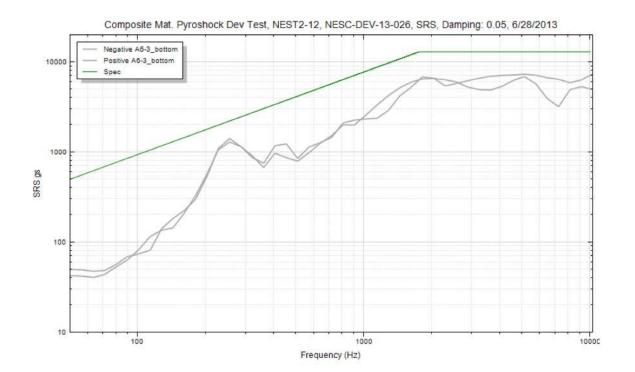
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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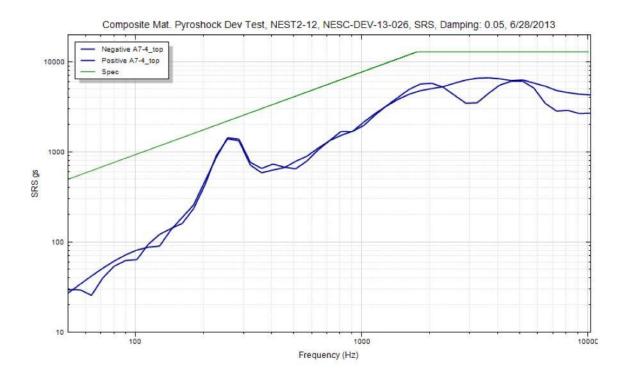
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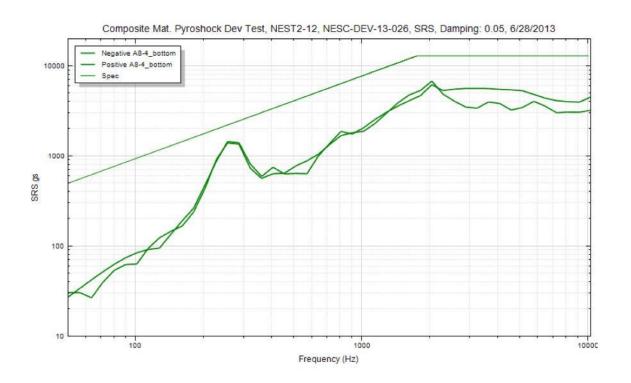
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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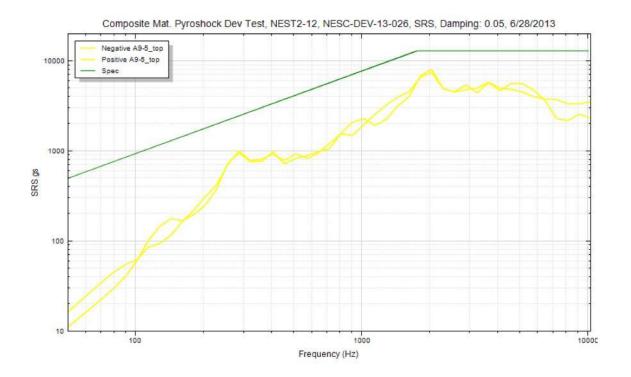
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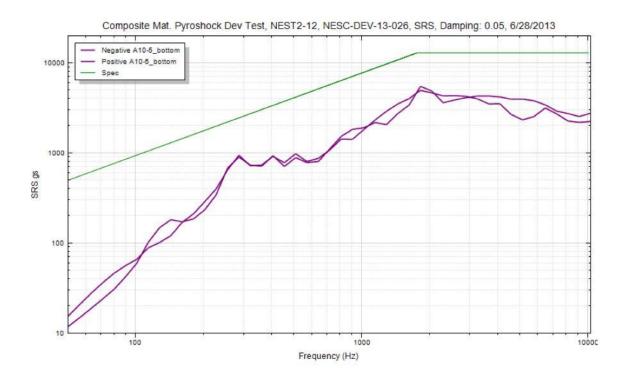
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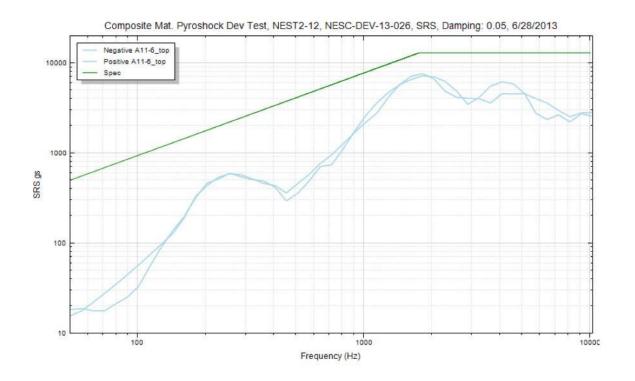
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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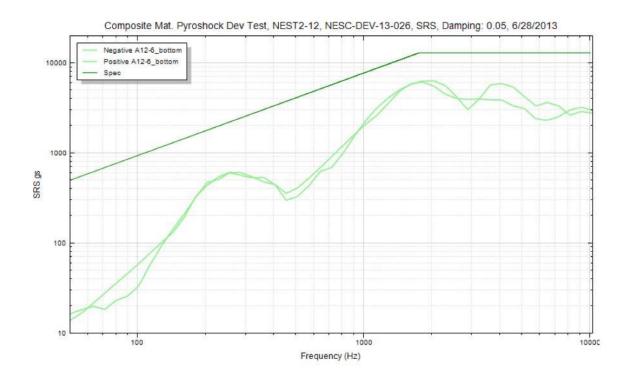
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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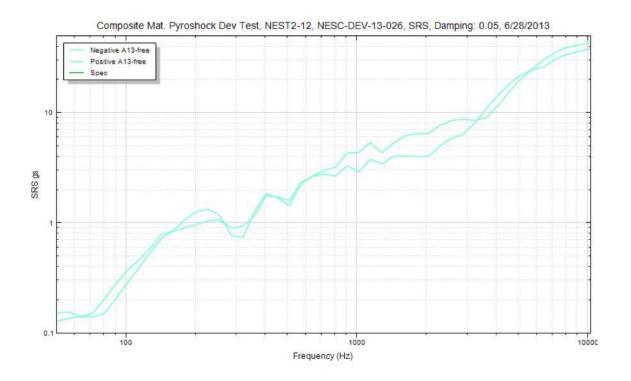
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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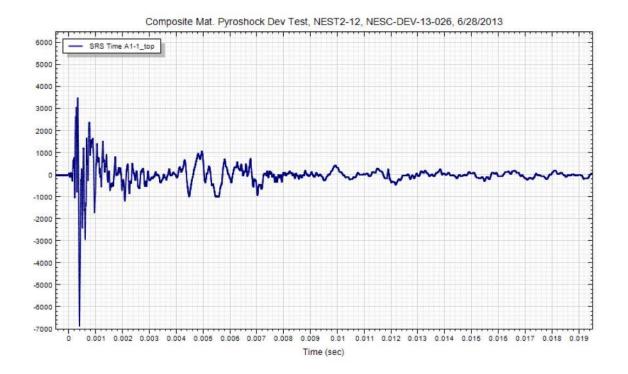
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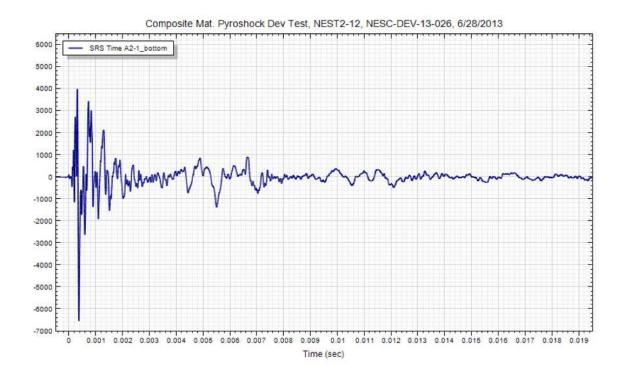
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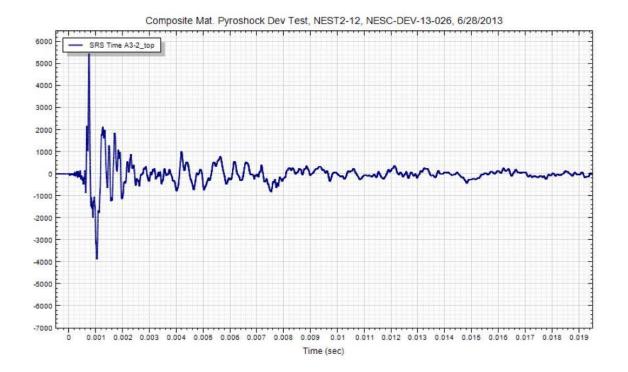
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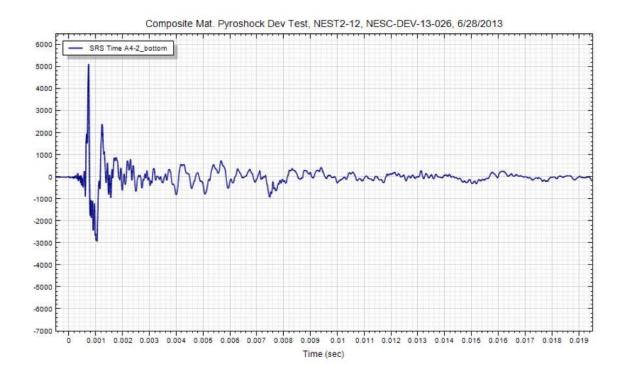
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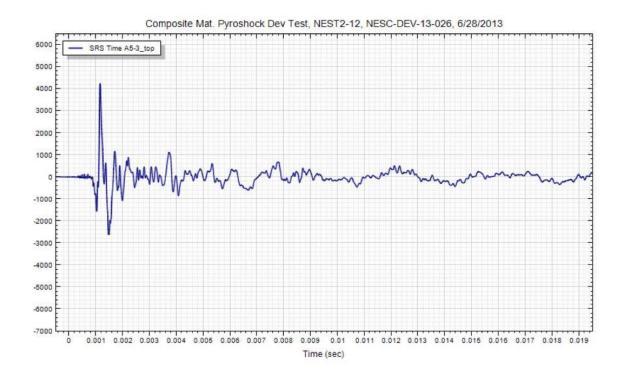
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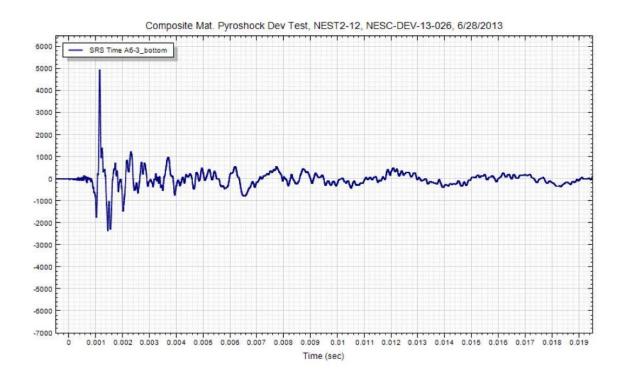
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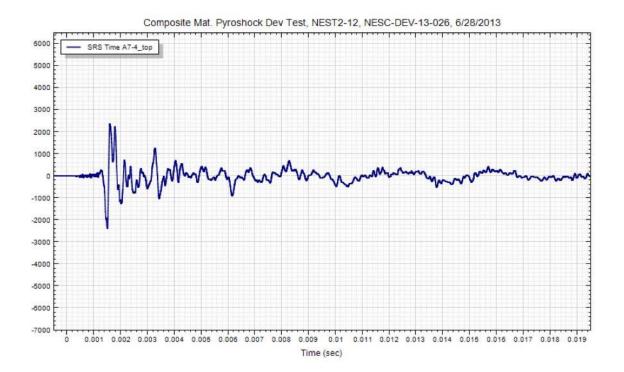
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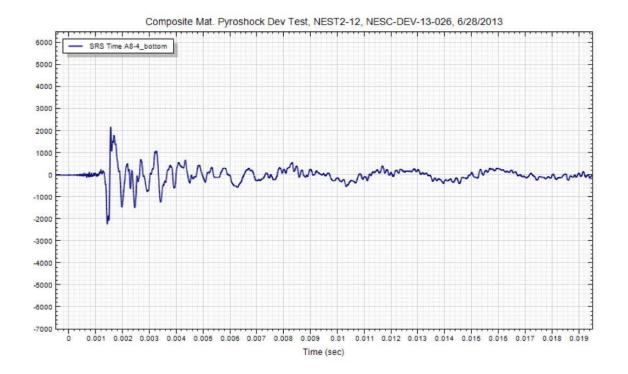
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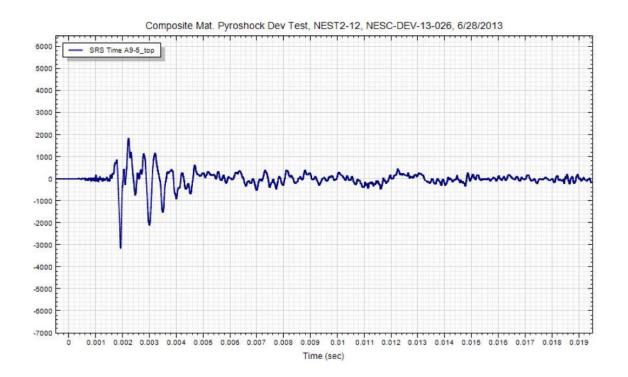
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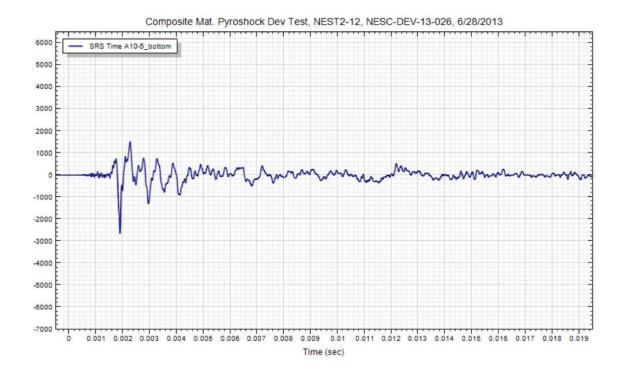
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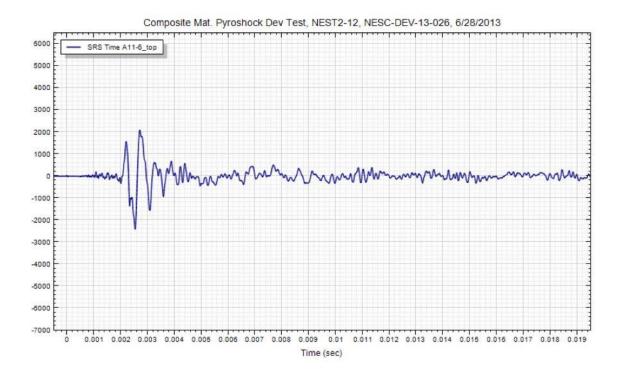
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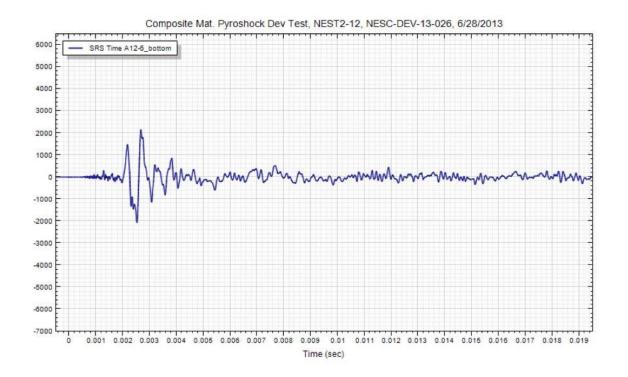
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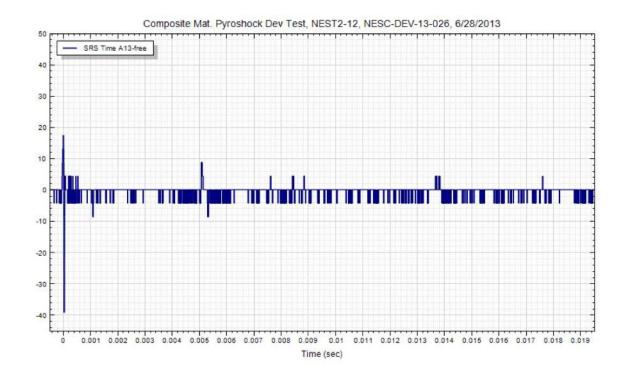
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-026 Composite Materials Shock Test

### Test #3 Accelerometer Data Panel 0332A013



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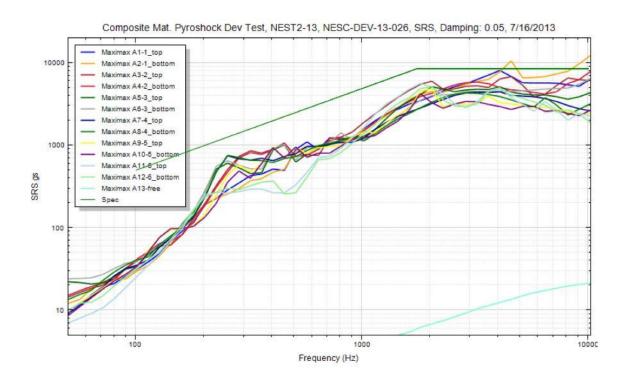
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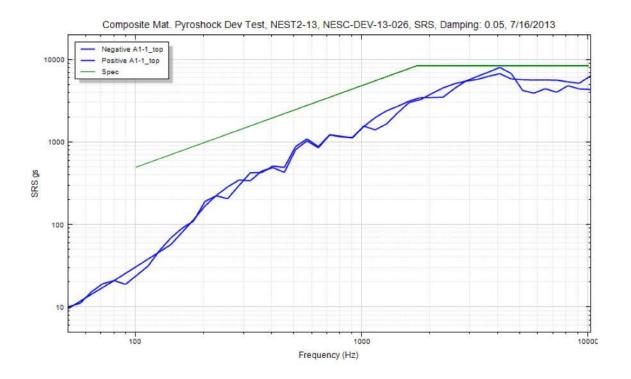
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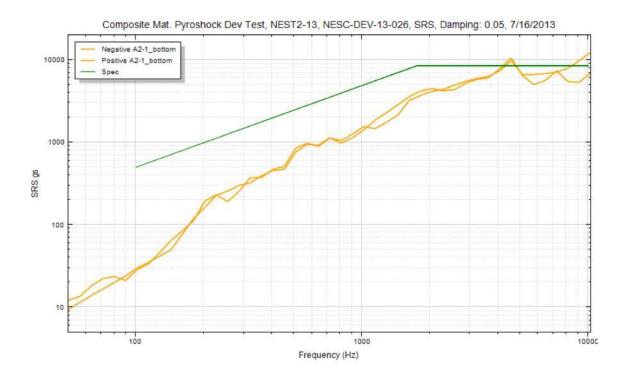
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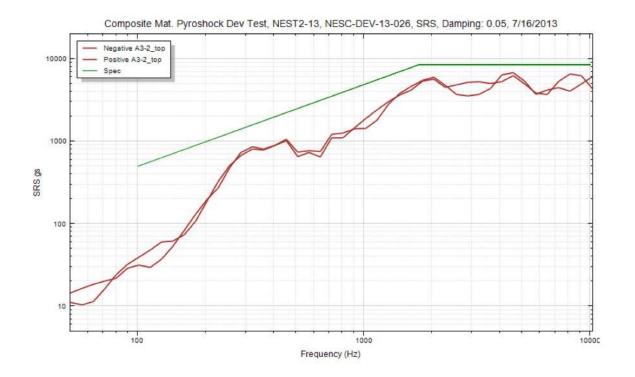
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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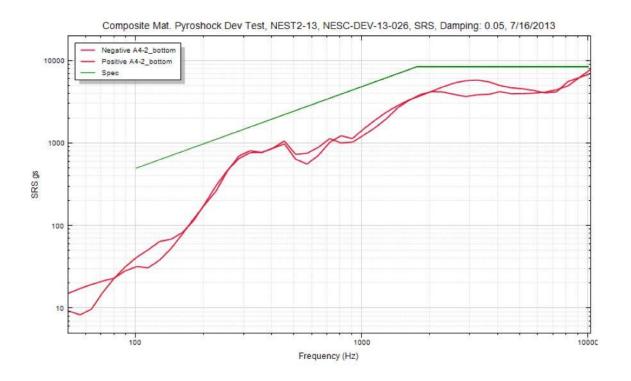
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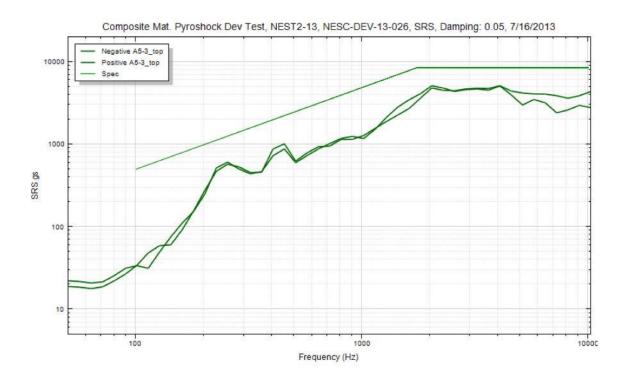
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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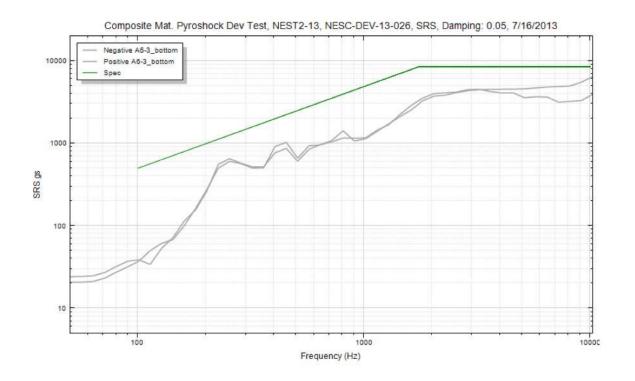
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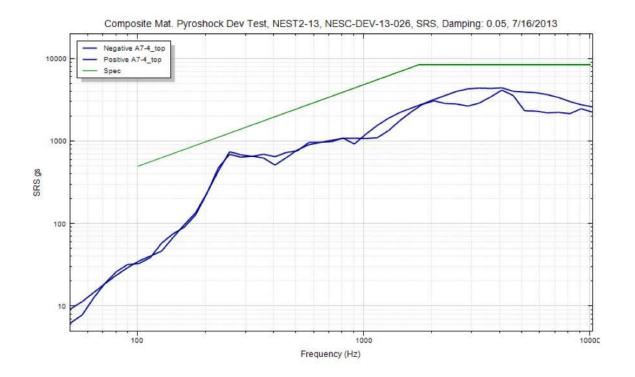
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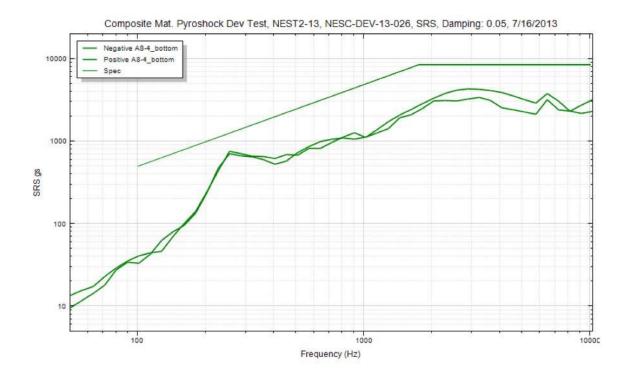
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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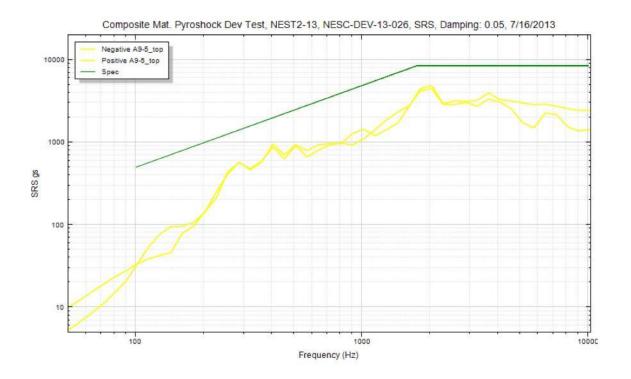
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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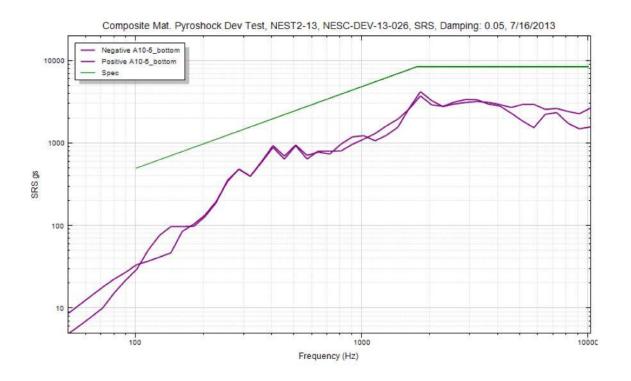
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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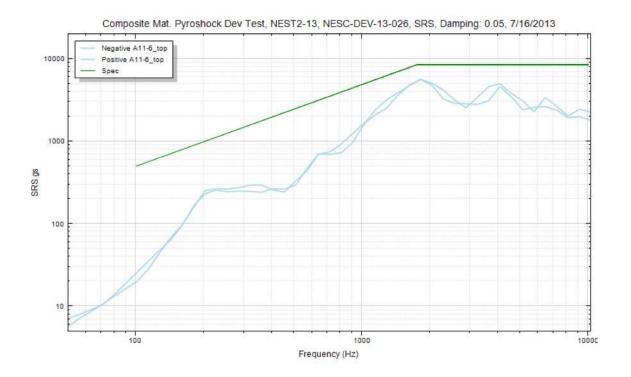
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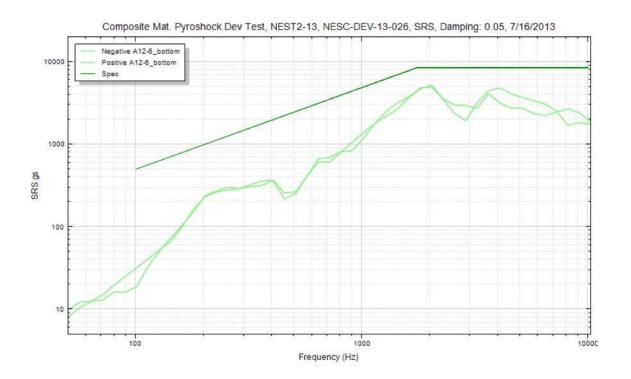
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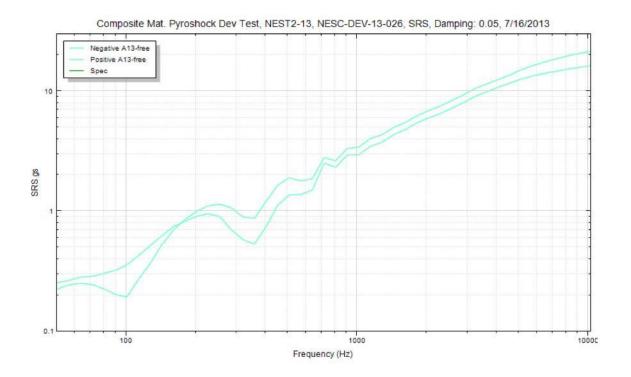
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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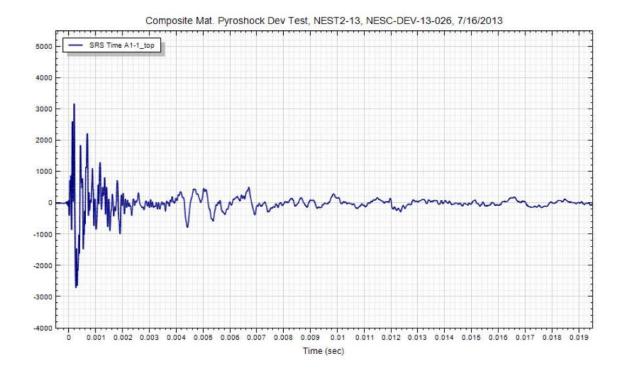
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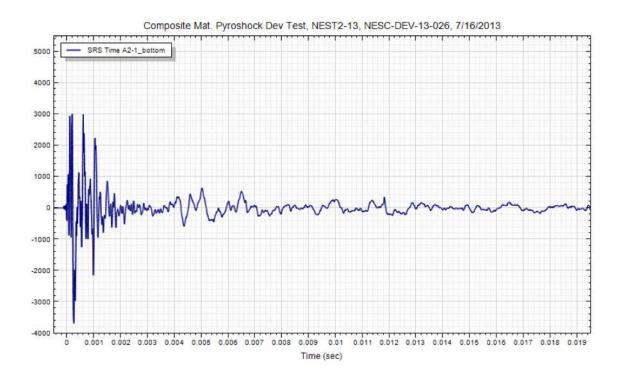
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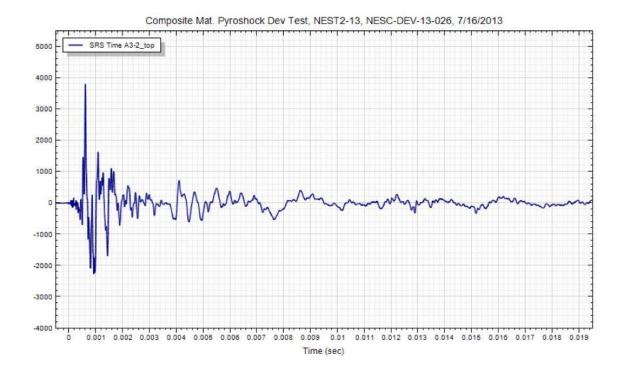
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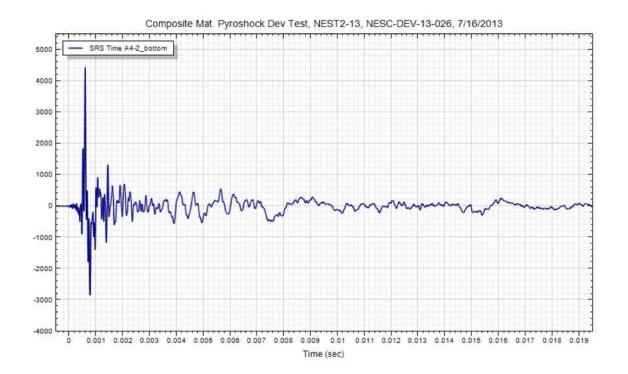
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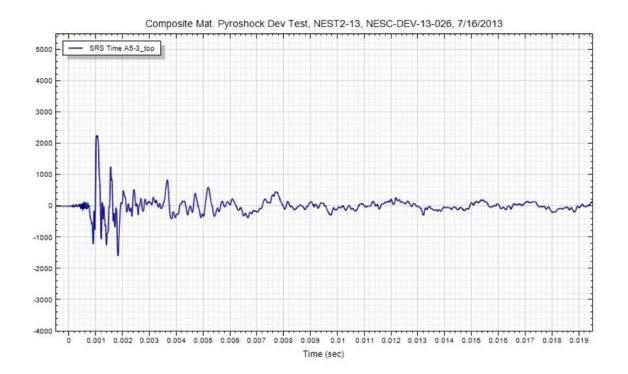
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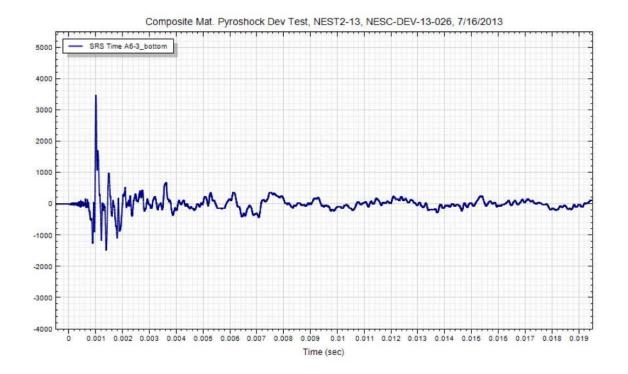
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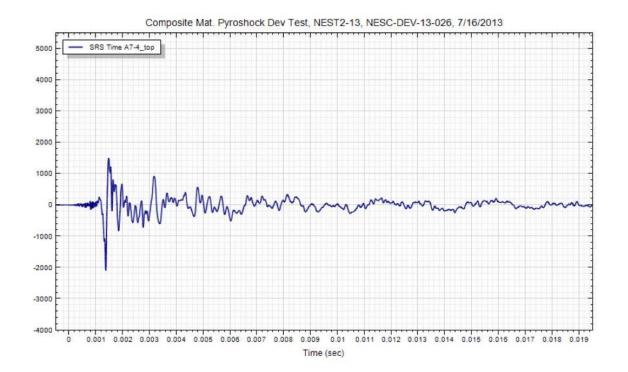
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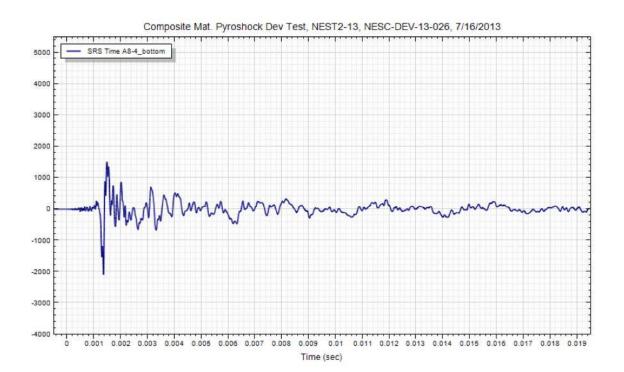
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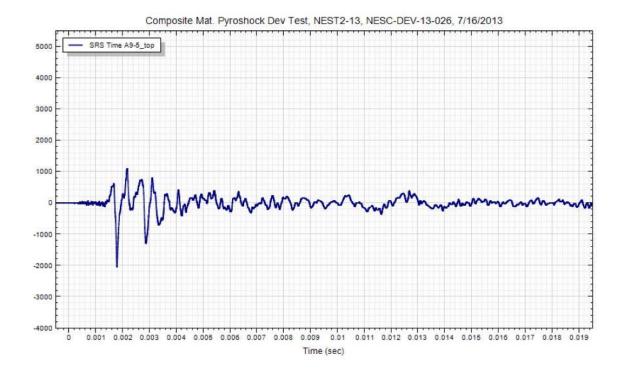
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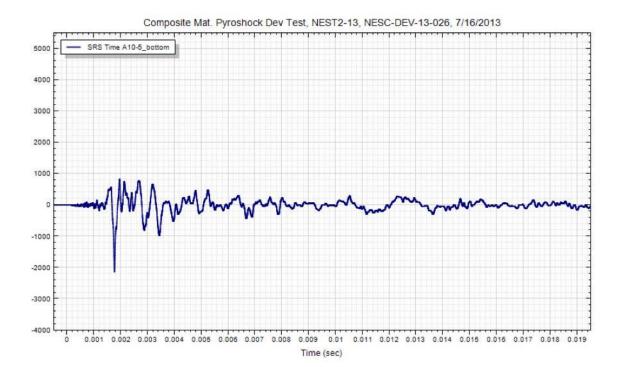
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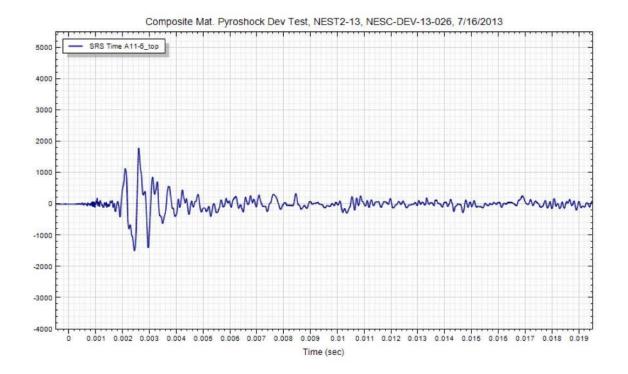
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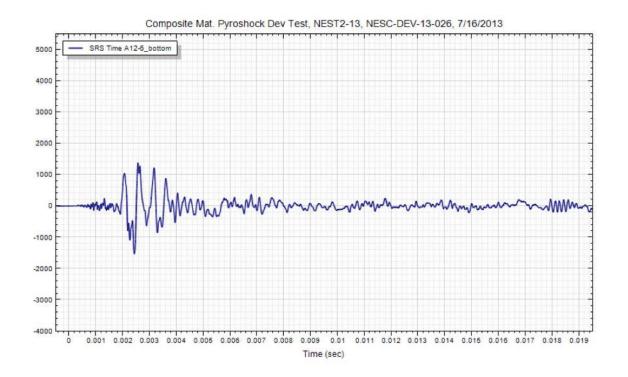
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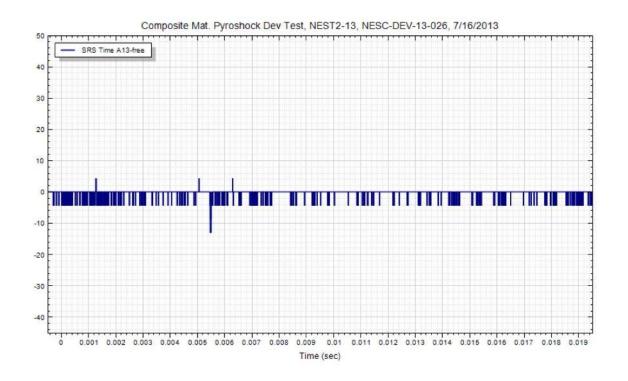
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### NESC-DEV-13-026 Composite Materials Shock Test

### Test #4 Accelerometer Data Panel 0332A012



NESC-RP-12-00783

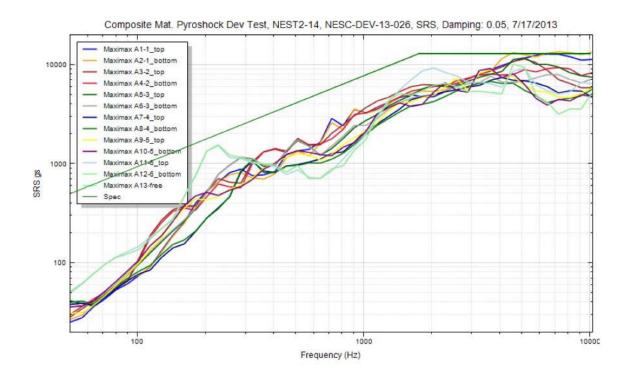
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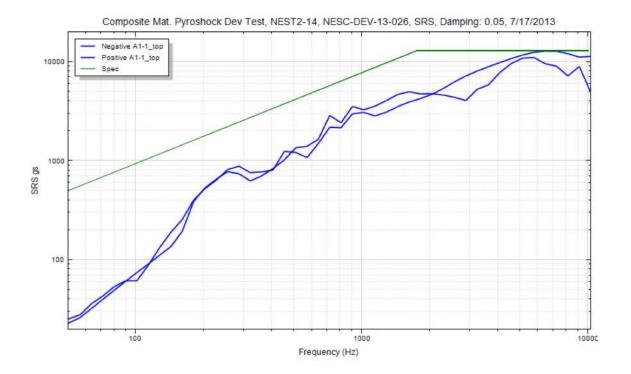
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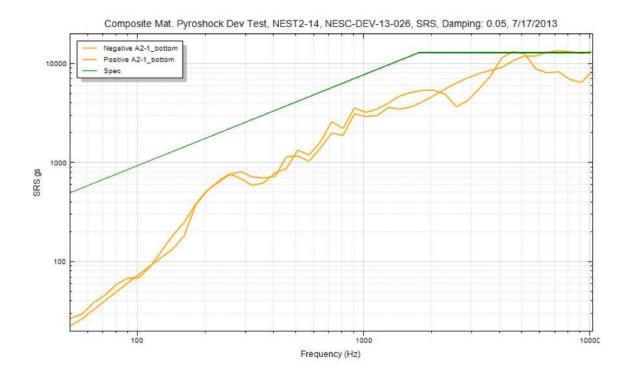
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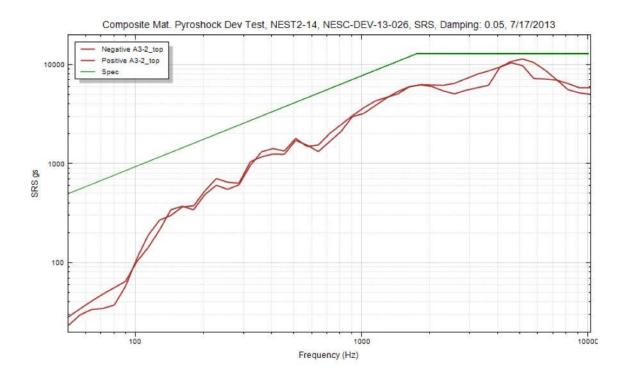
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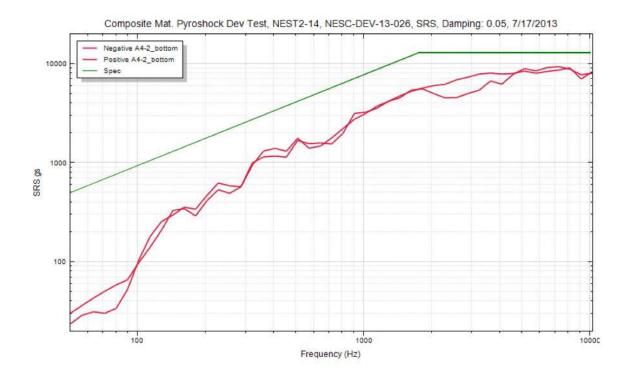
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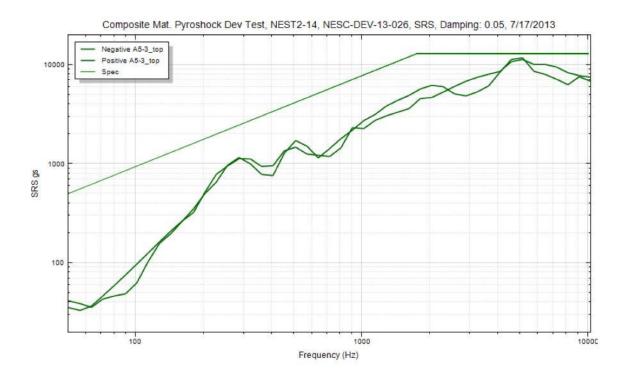
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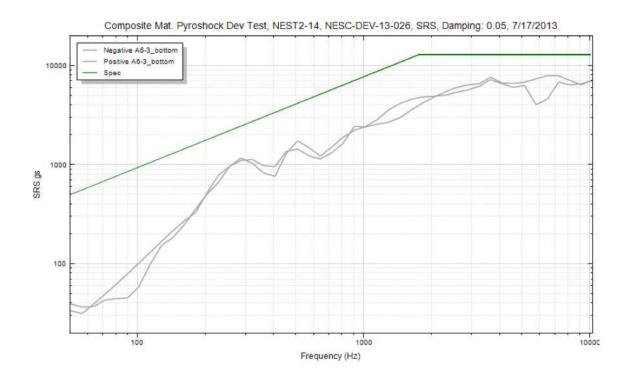
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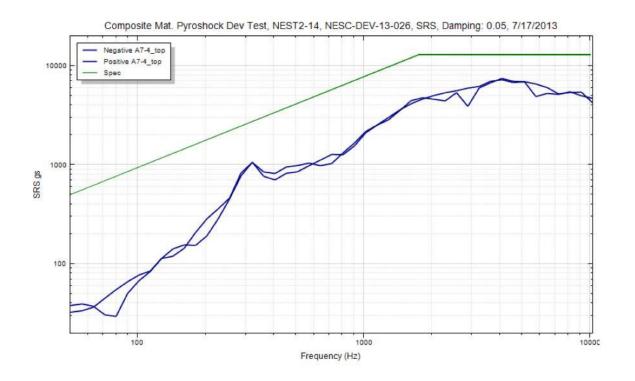
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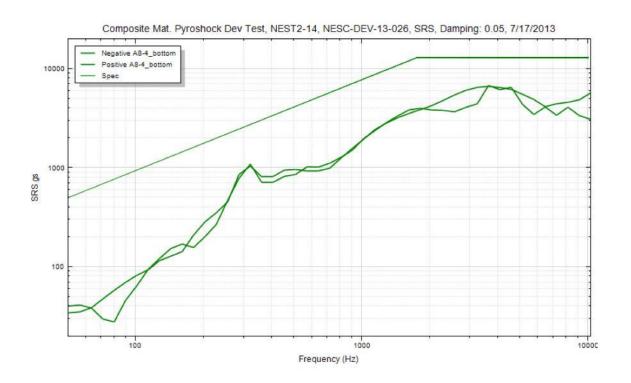
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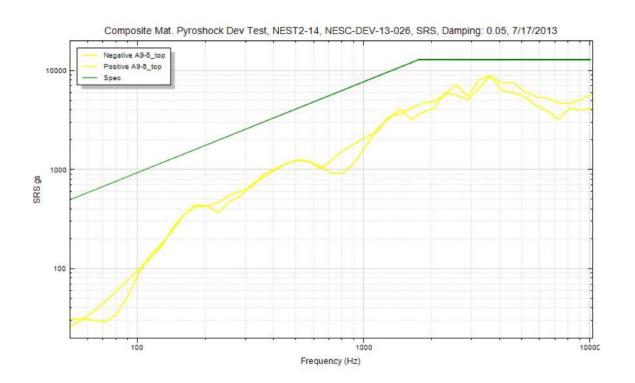
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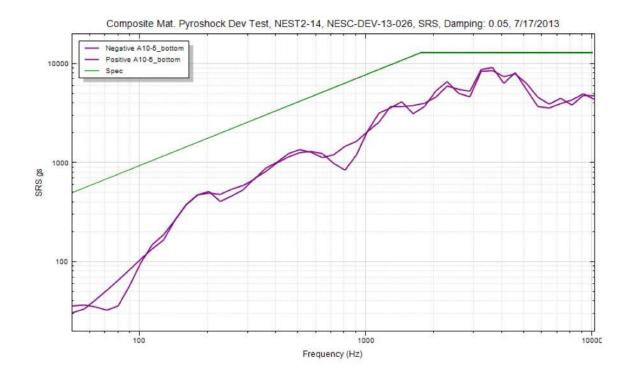
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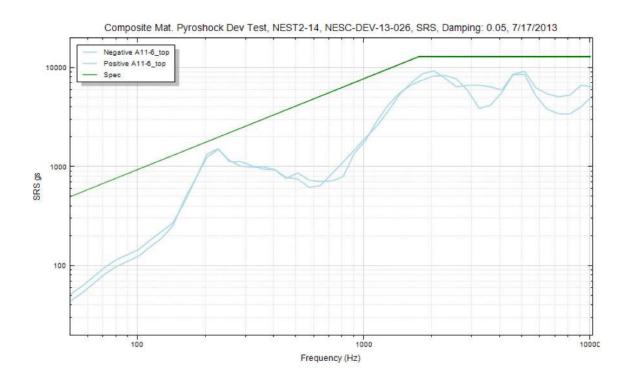
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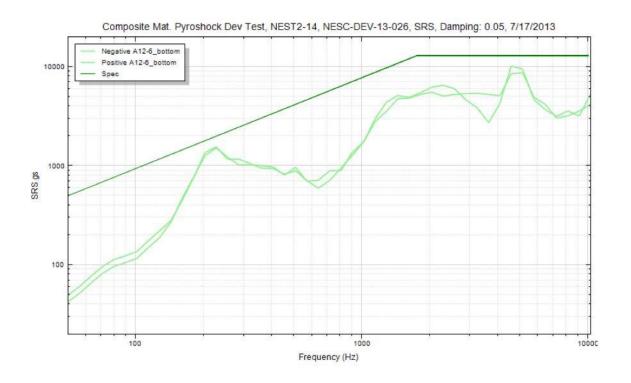
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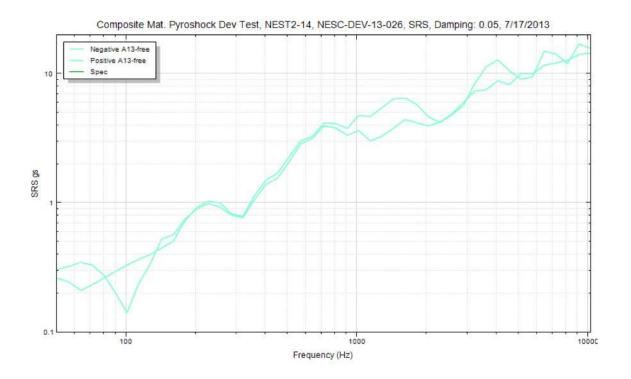
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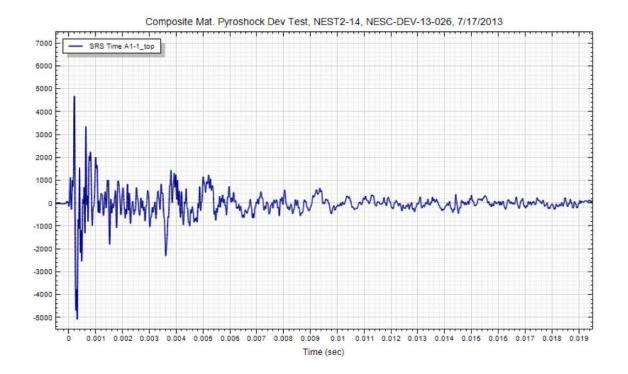
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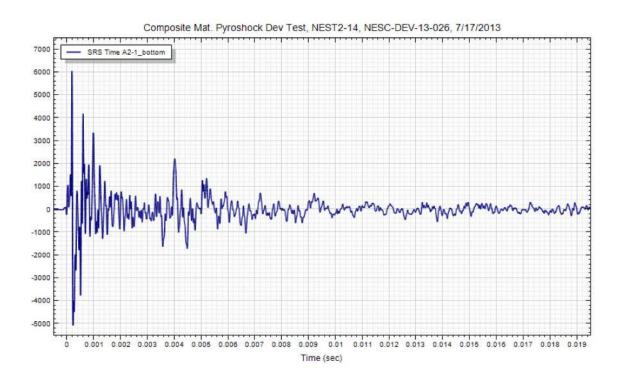
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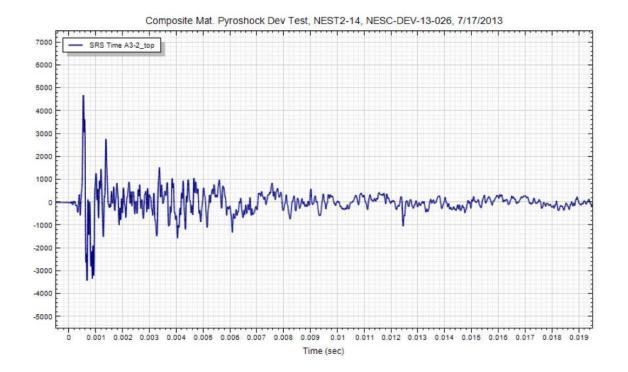
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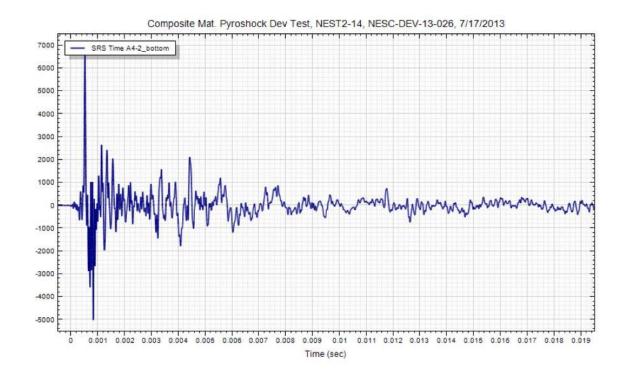
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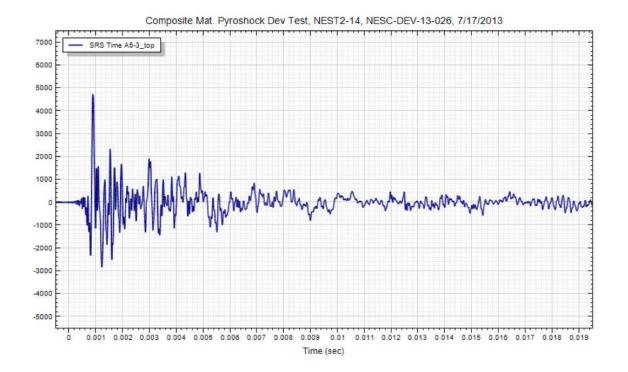
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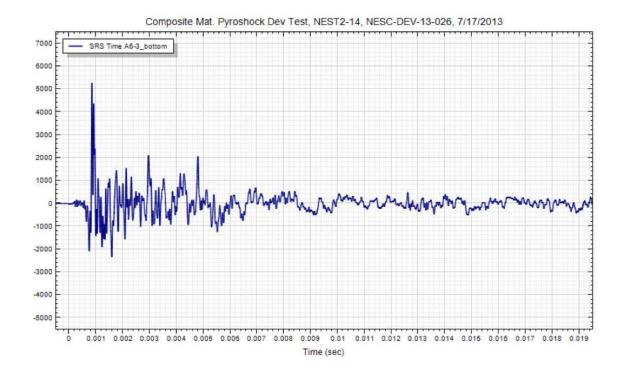
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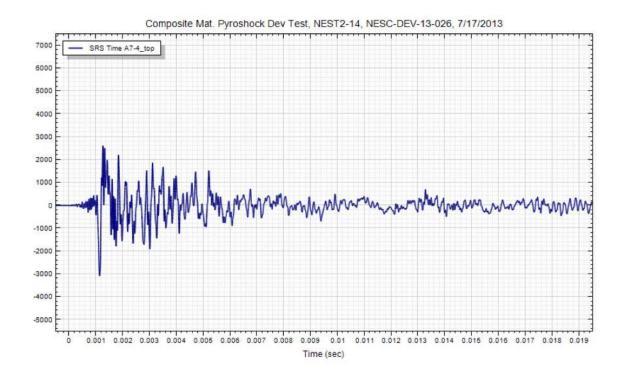
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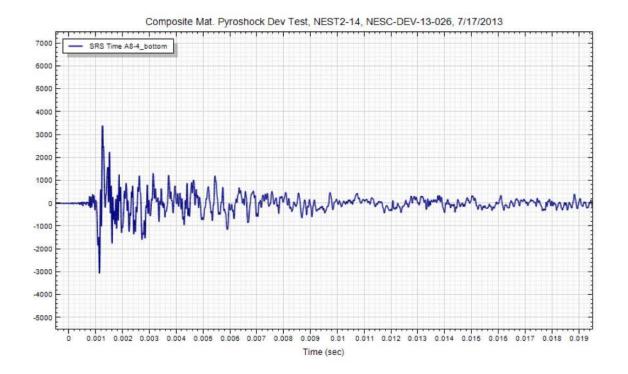
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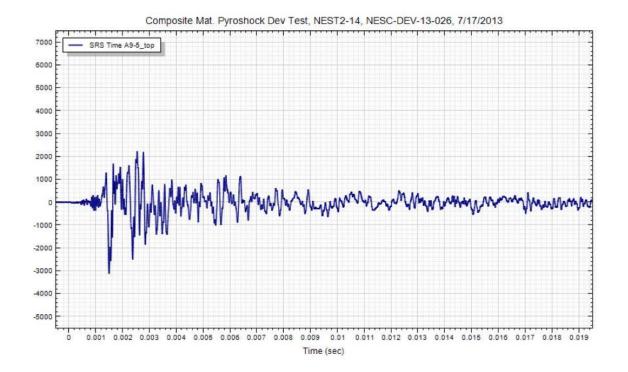
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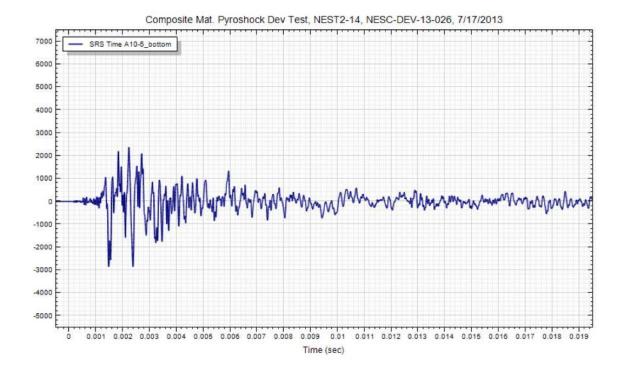
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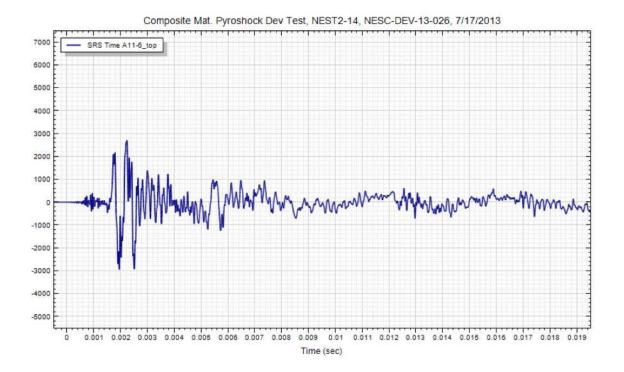
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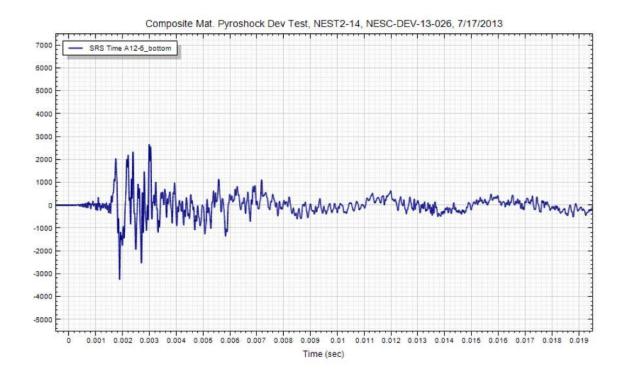
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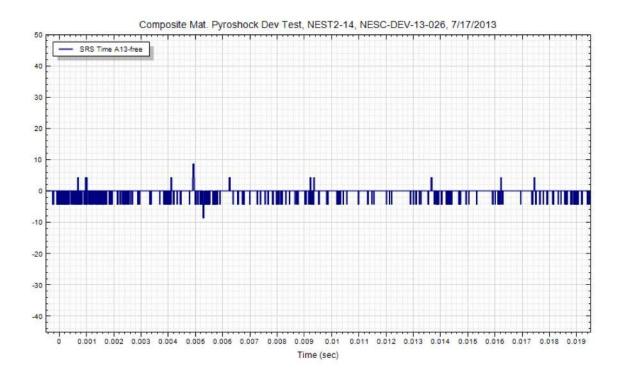
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### NESC-DEV-13-026 Composite Materials Shock Test

### Test #5 Accelerometer Data Panel 0332A016



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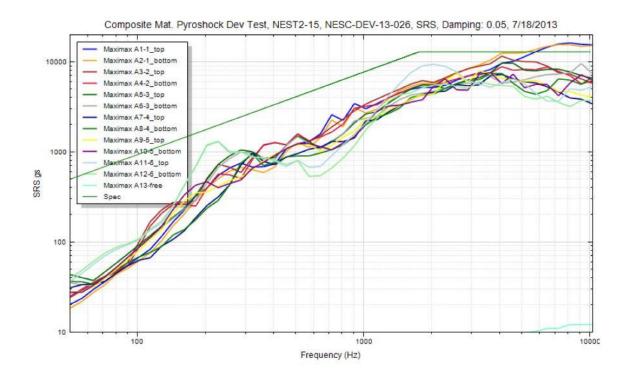
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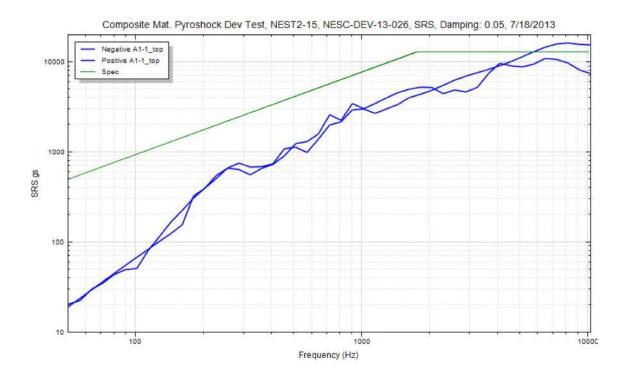
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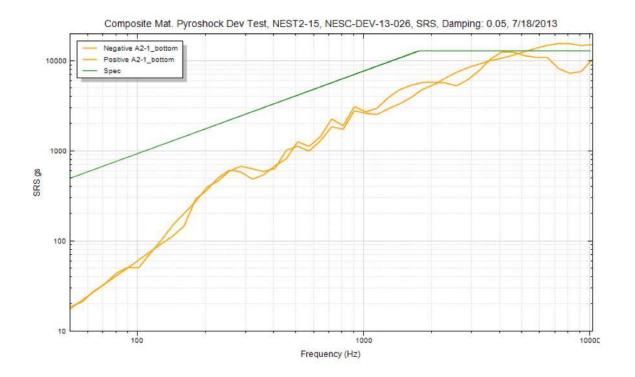
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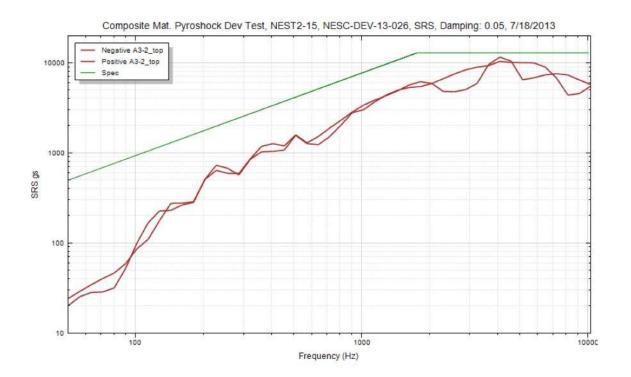
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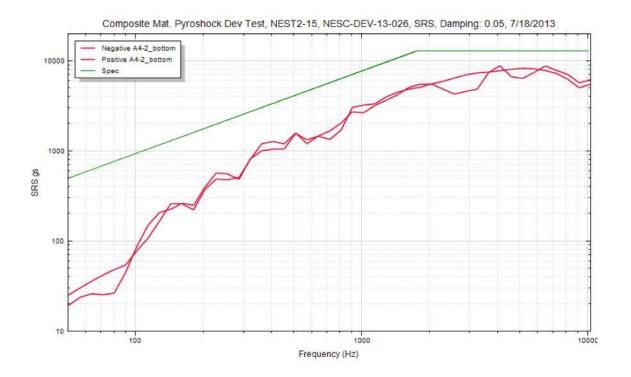
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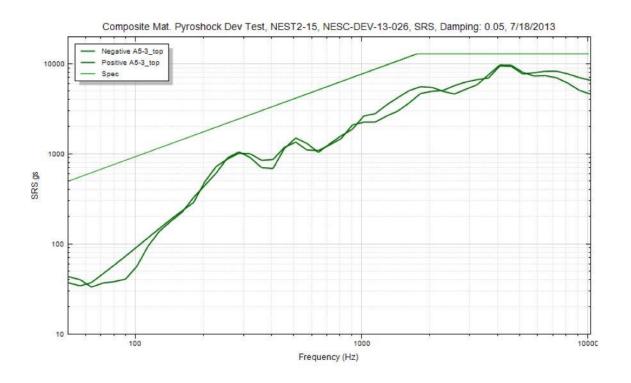
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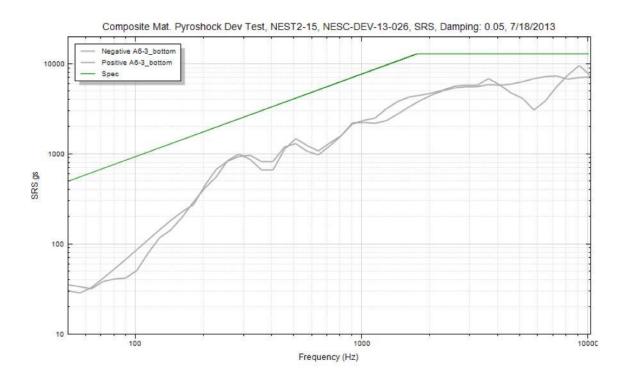
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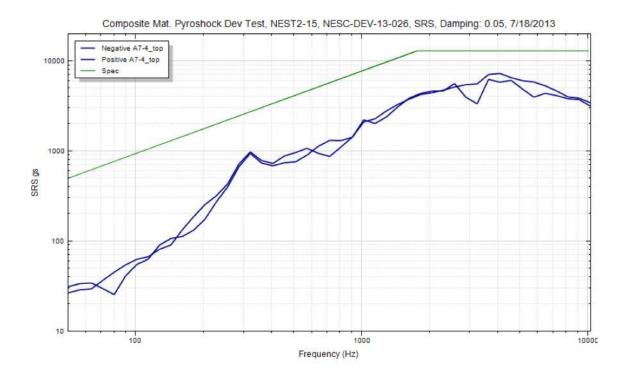
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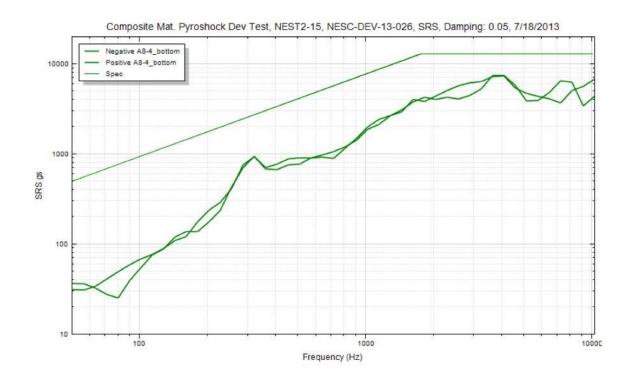
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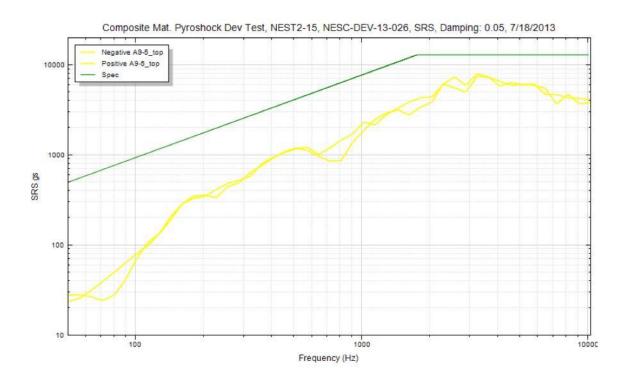
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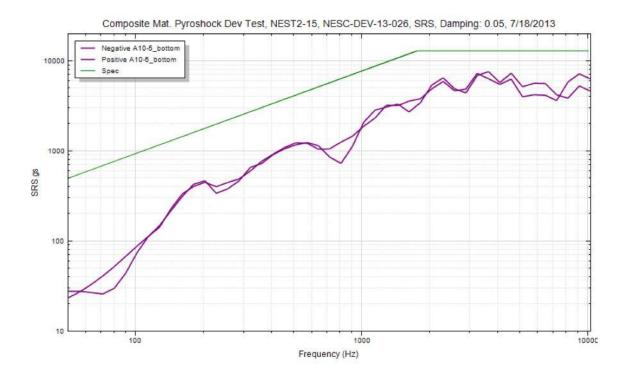
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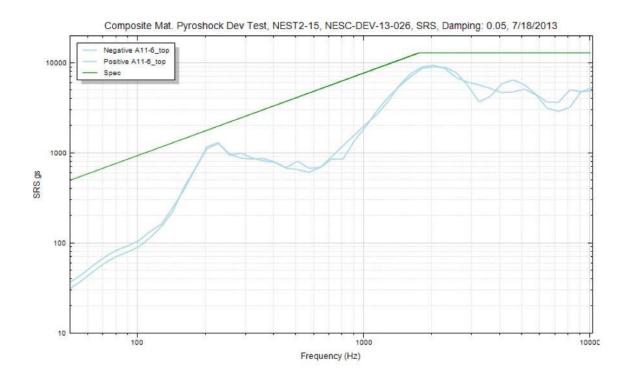
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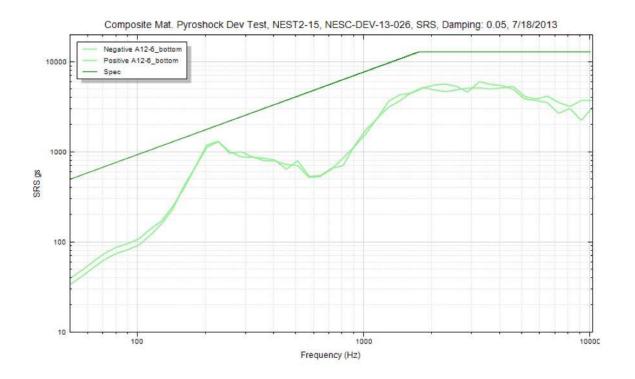
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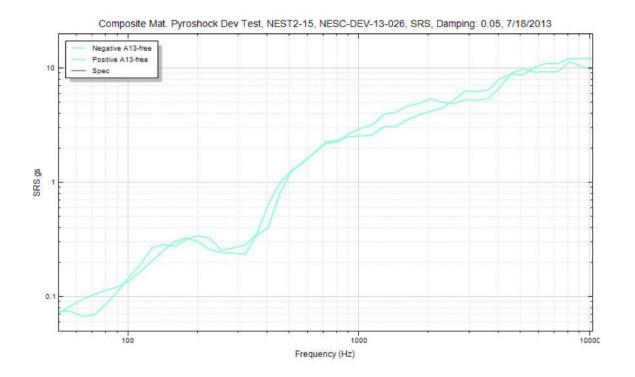
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





NESC-RP-12-00783

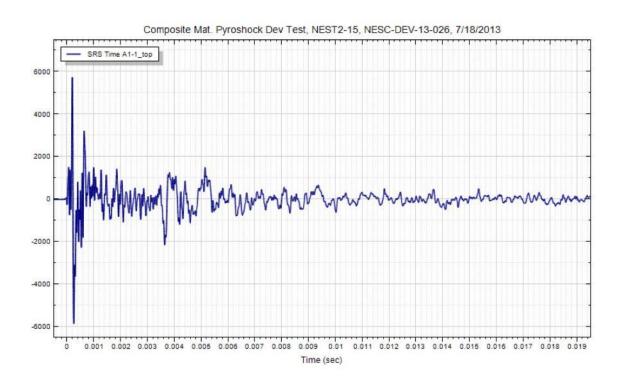
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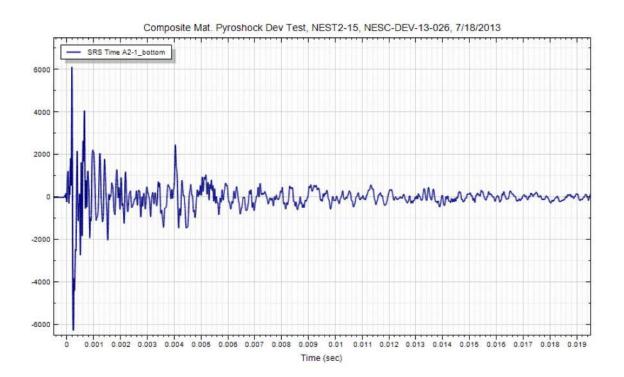
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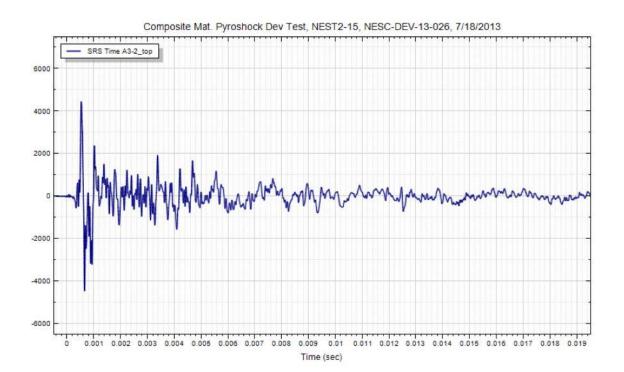
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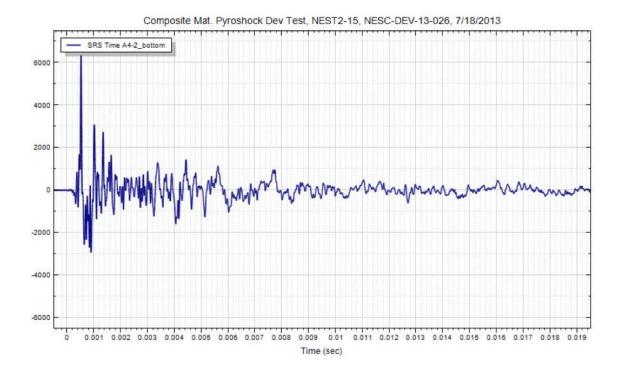
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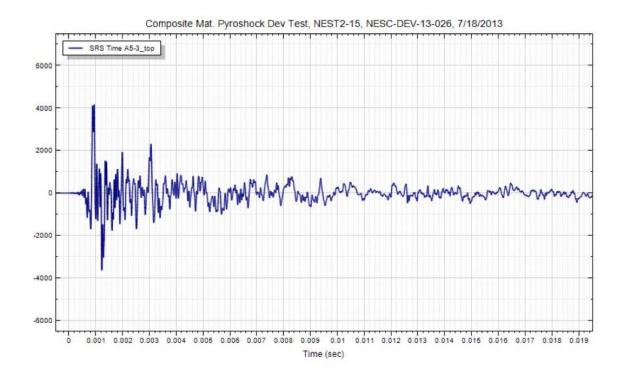
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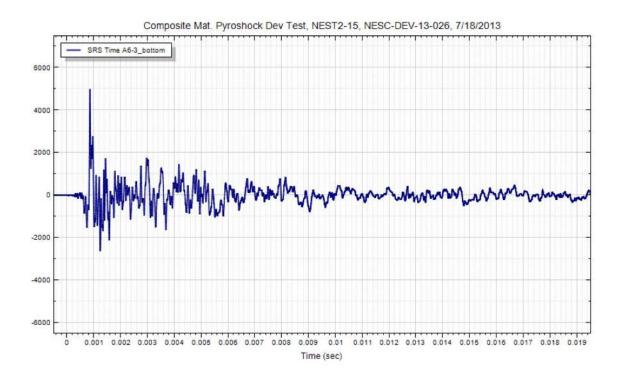
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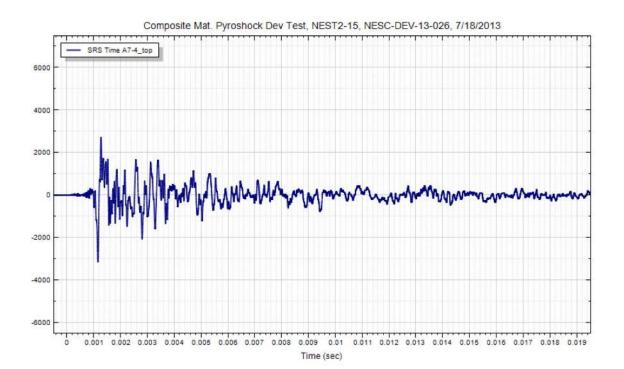
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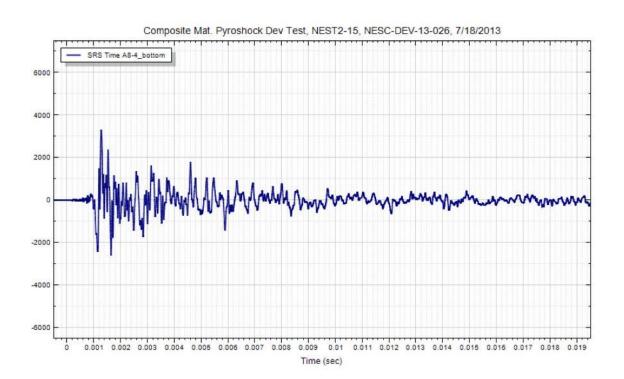
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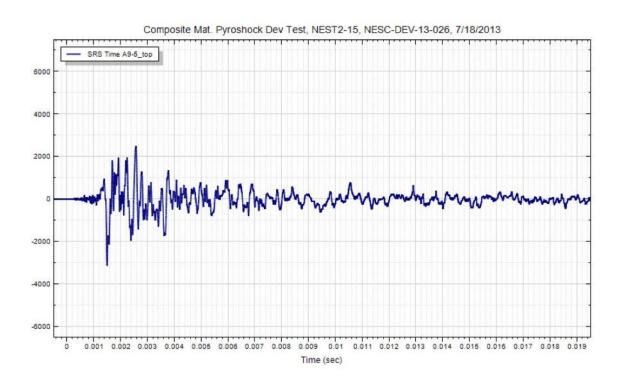
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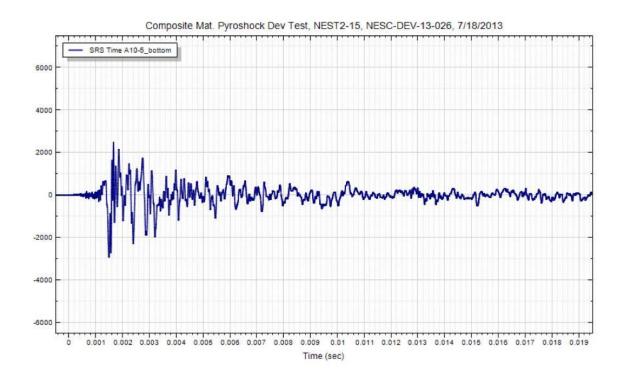
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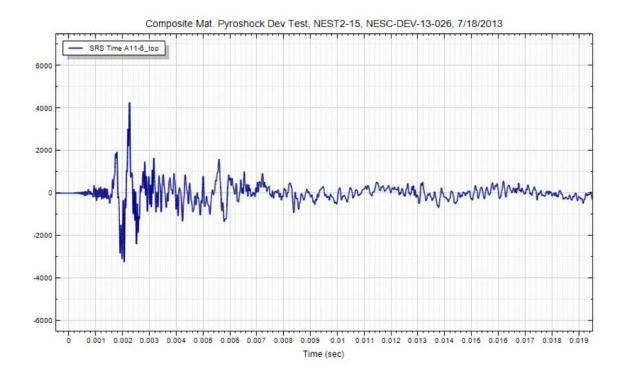
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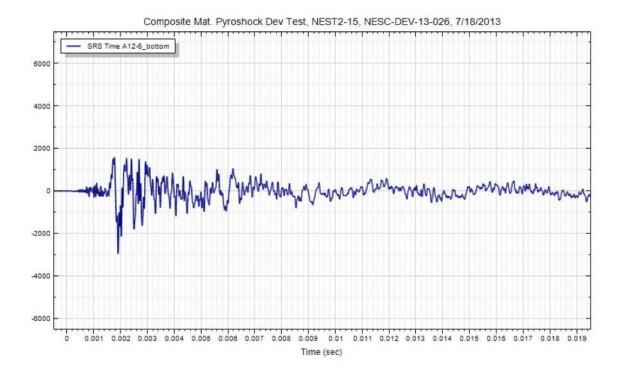
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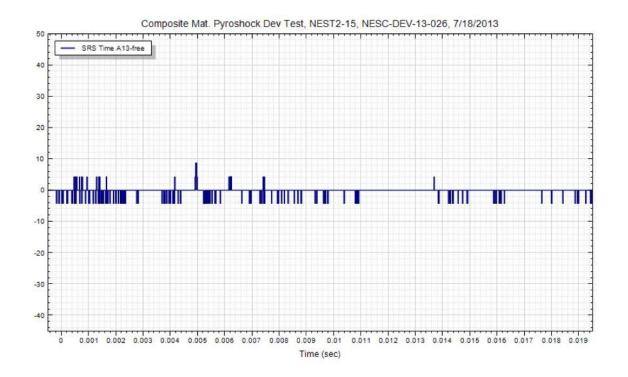
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-026 Composite Materials Shock Test

### Test #6 Accelerometer Data Panel 0332A018



NESC-RP-12-00783

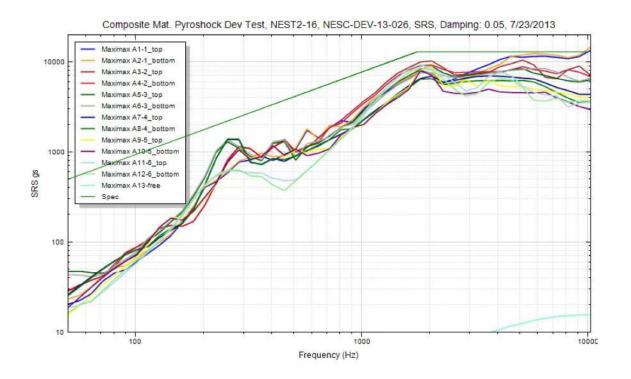
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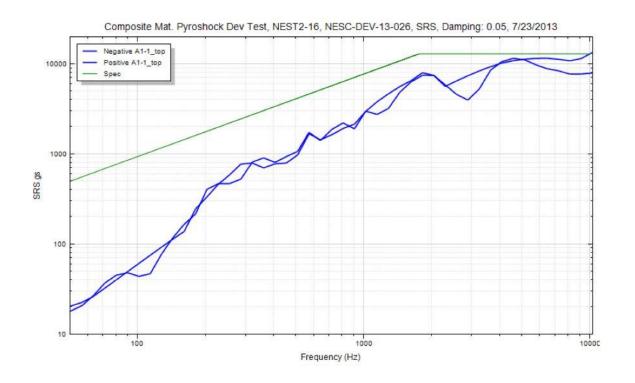
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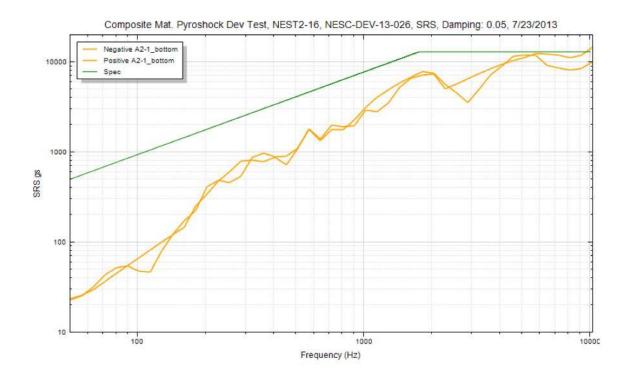
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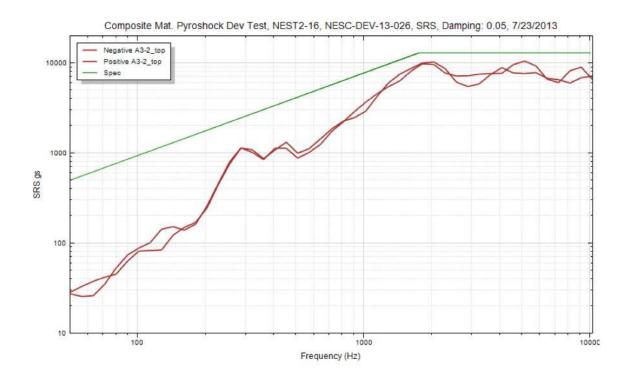
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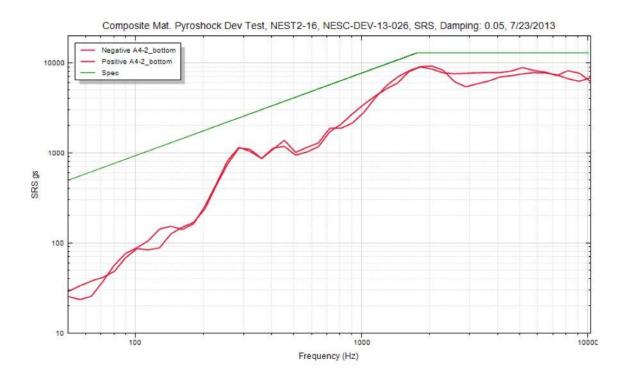
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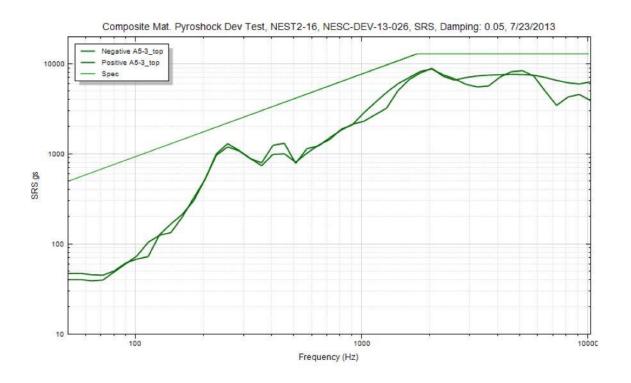
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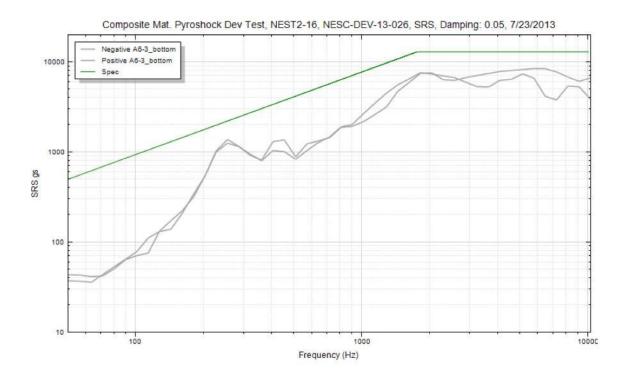
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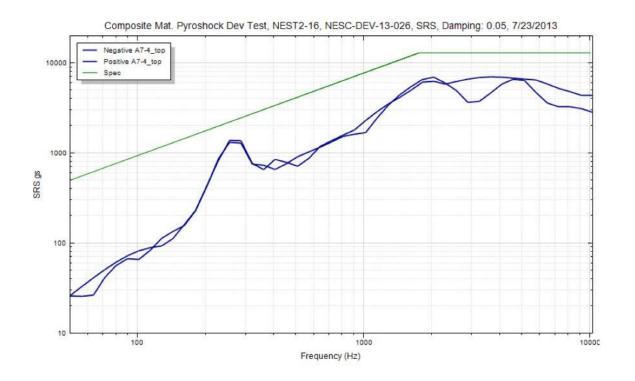
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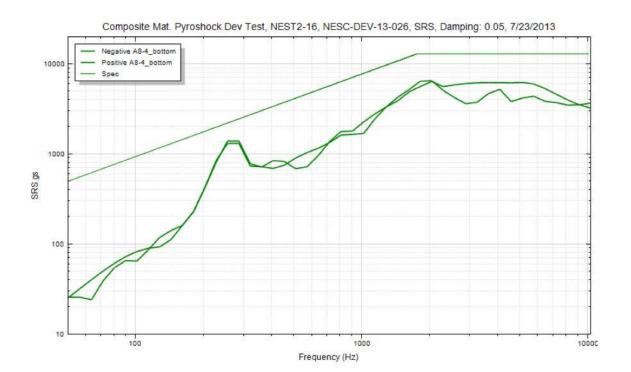
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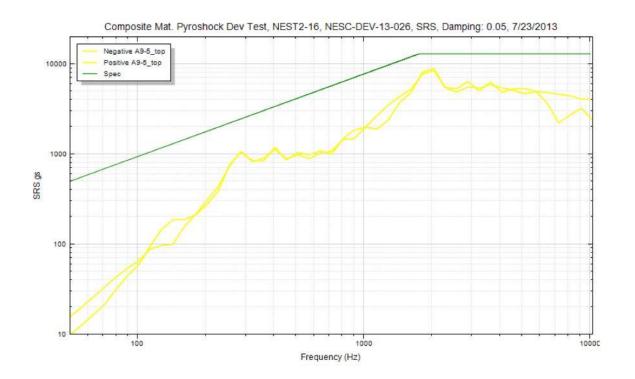
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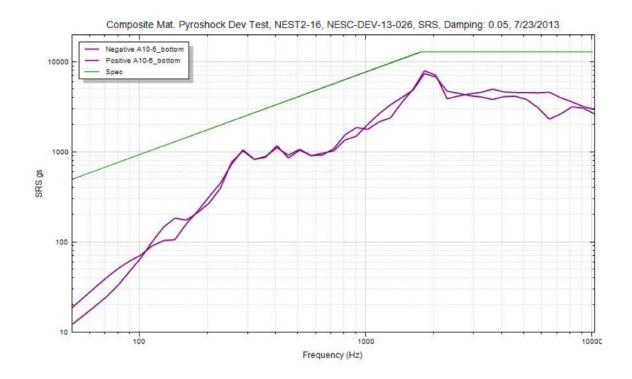
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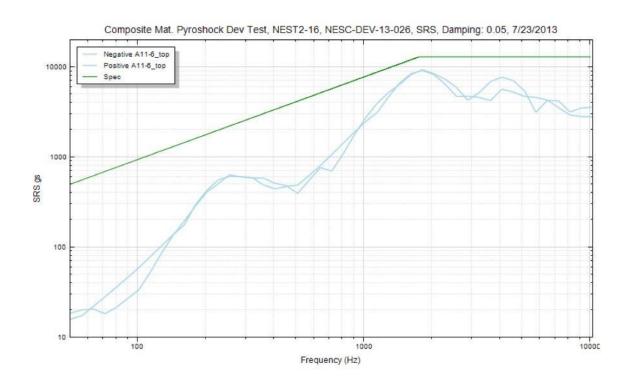
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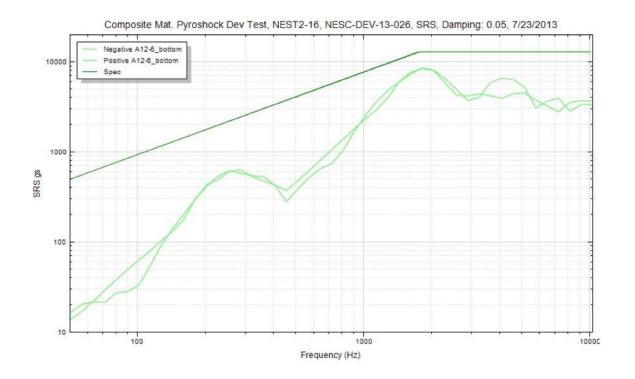
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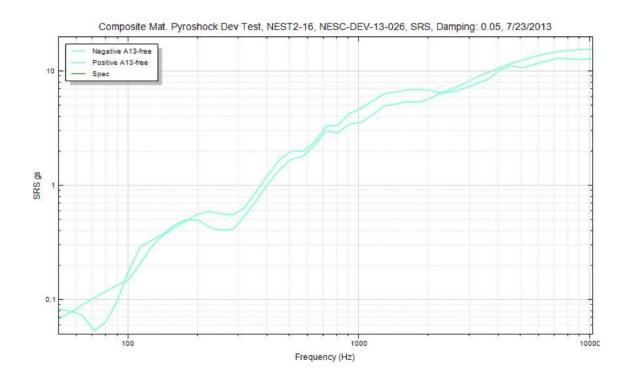
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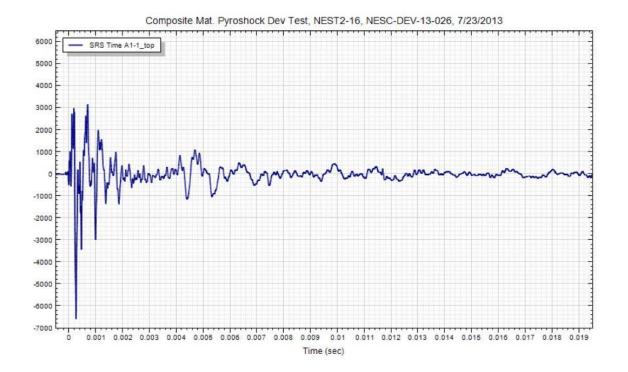
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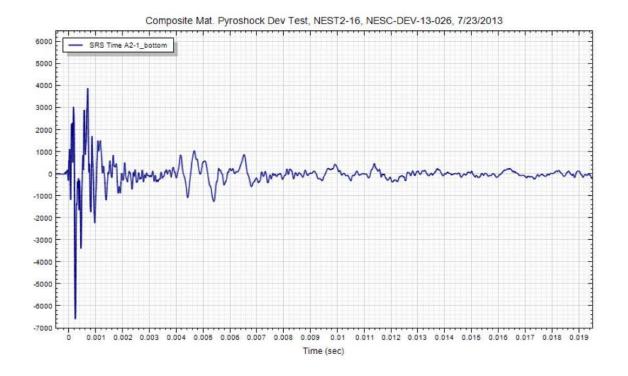
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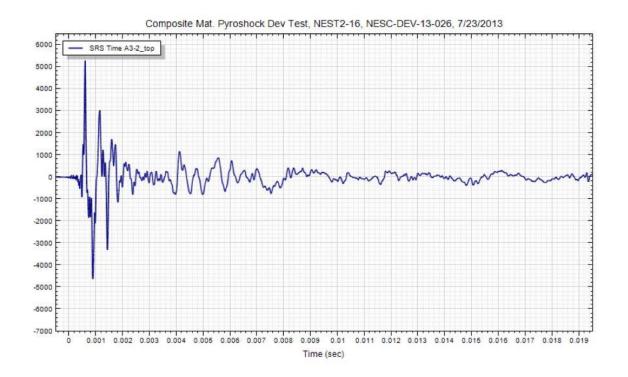
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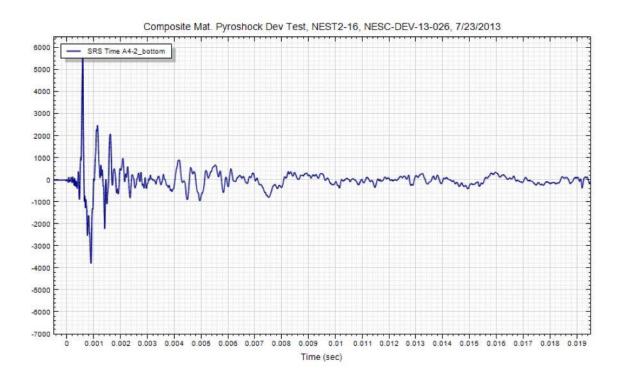
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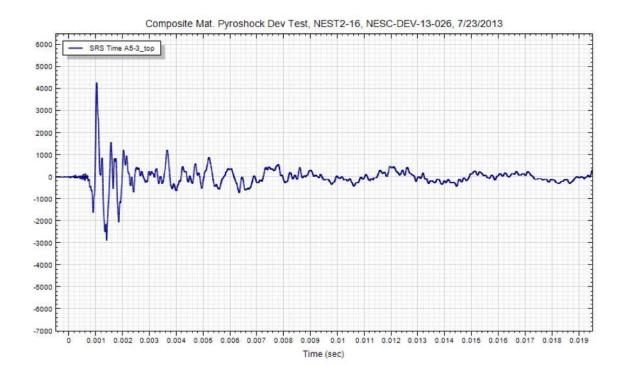
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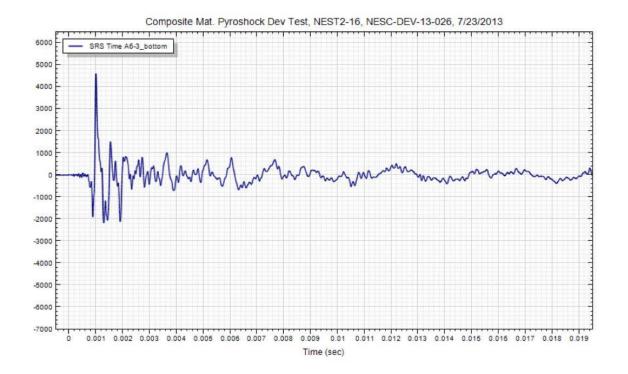
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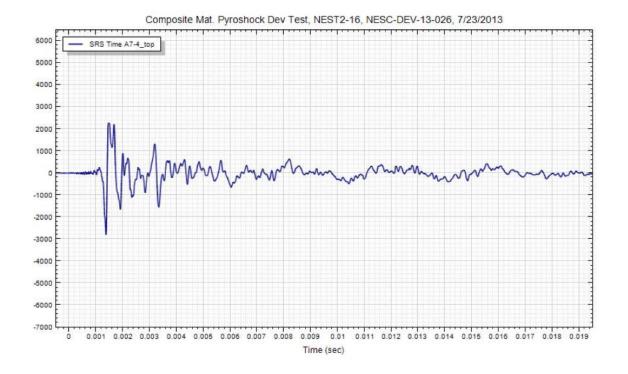
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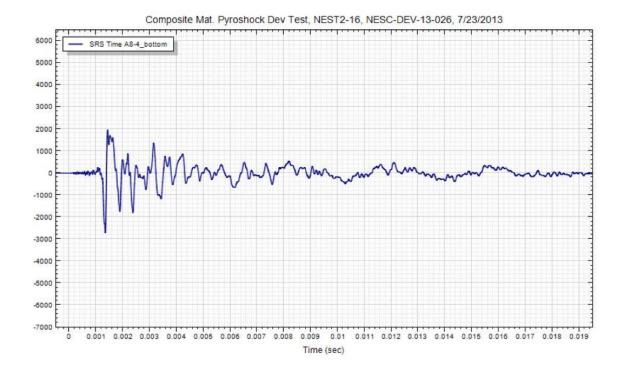
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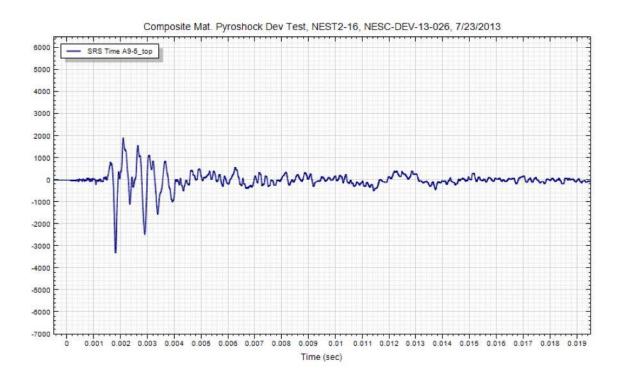
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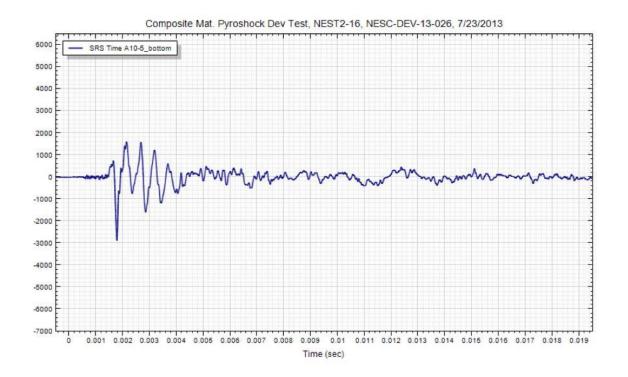
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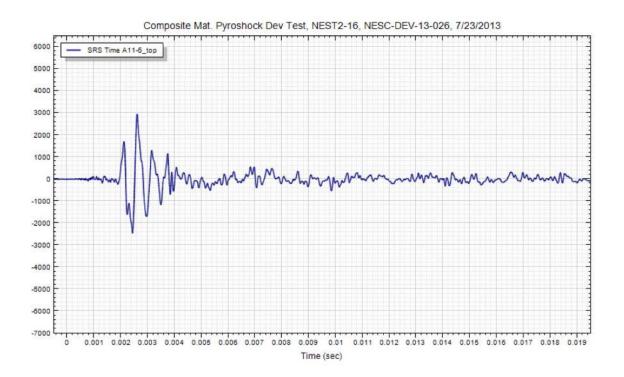
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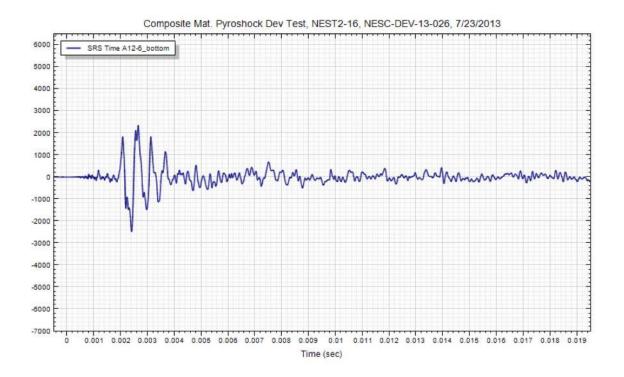
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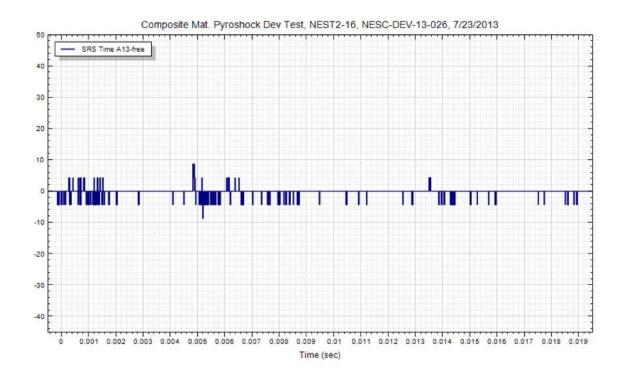
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### NESC-DEV-13-026 Composite Materials Shock Test

### Test #7 Accelerometer Data Panel 0332A015



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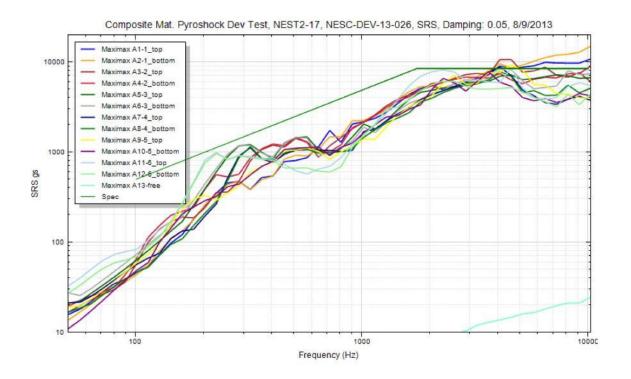
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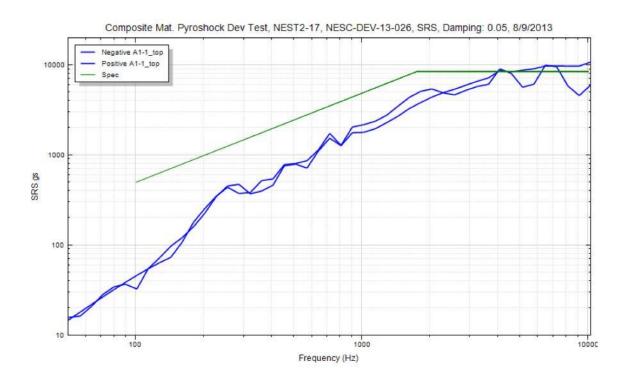
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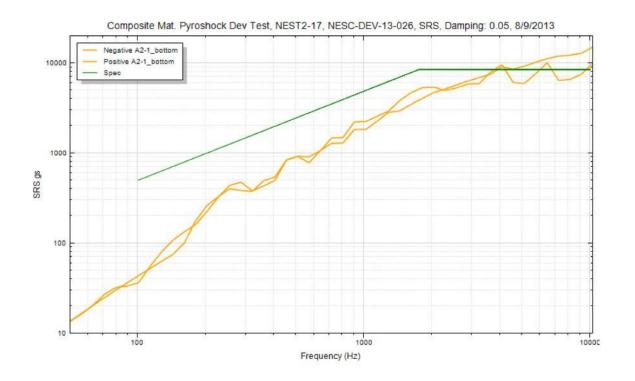
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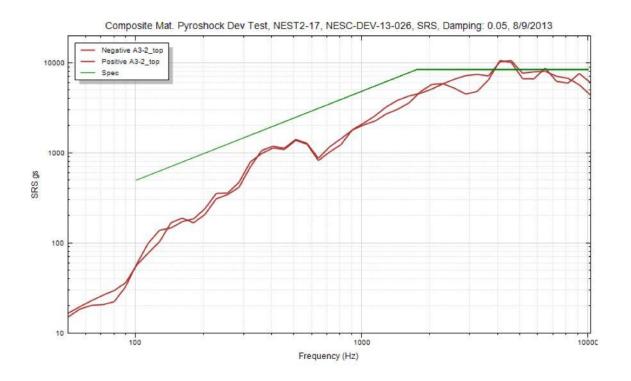
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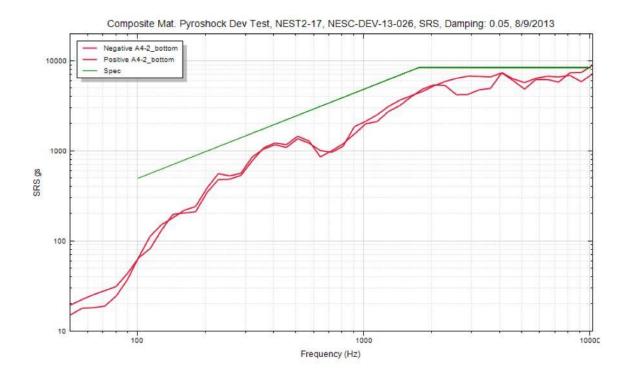
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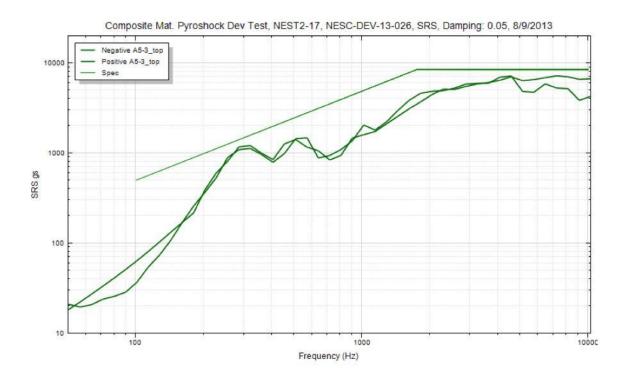
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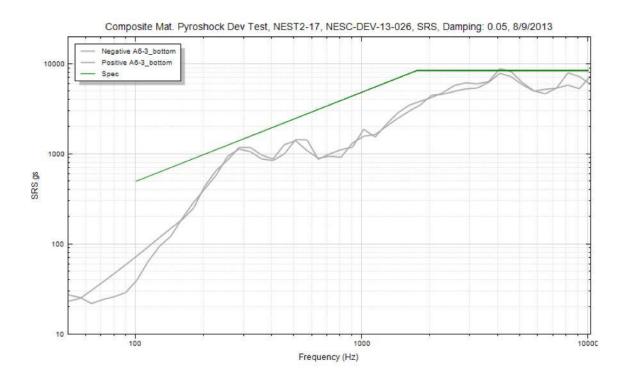
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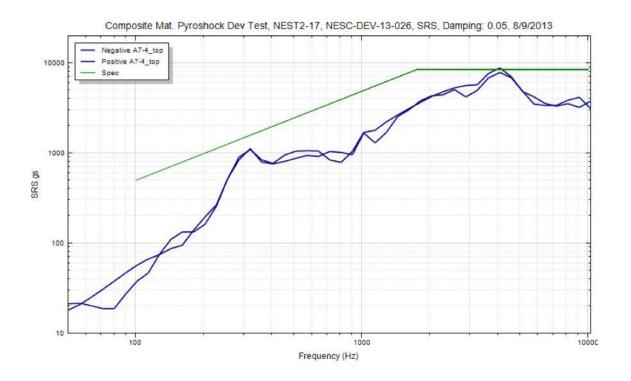
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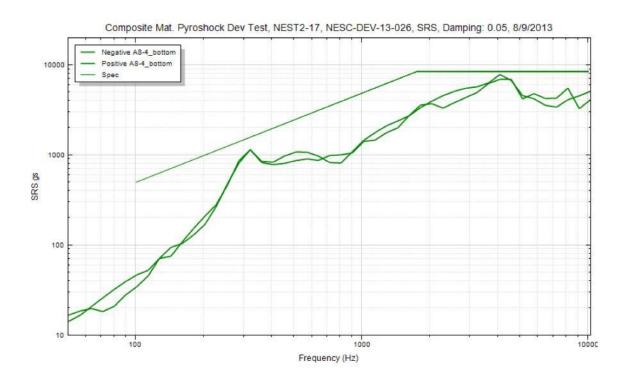
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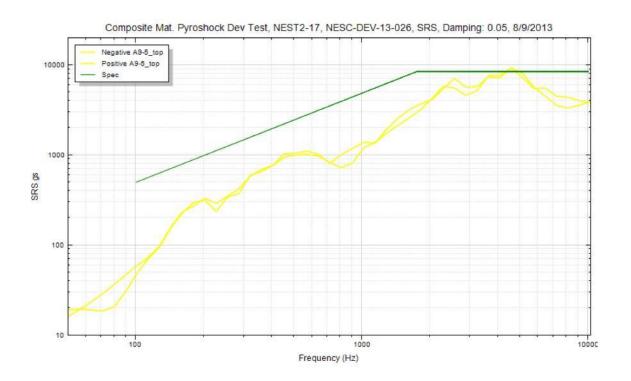
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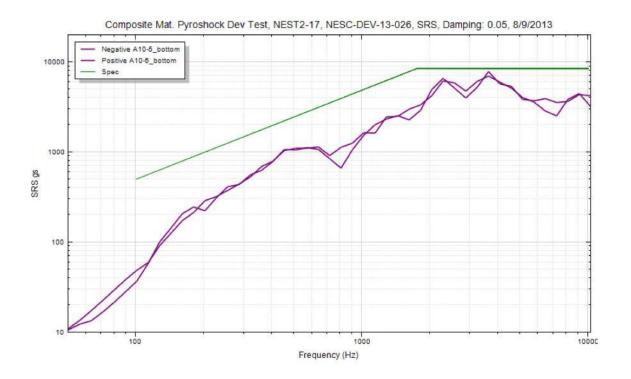
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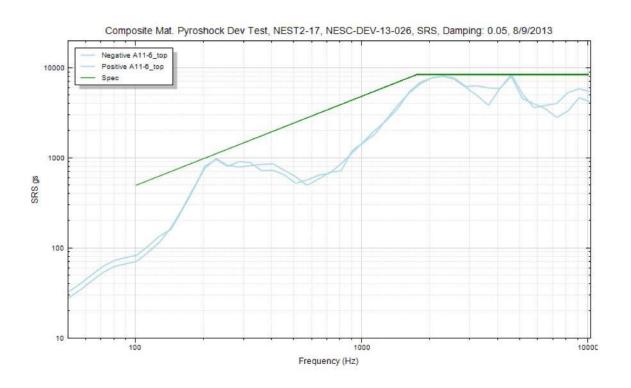
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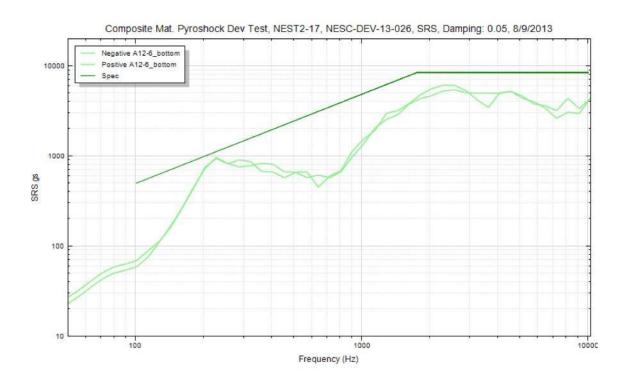
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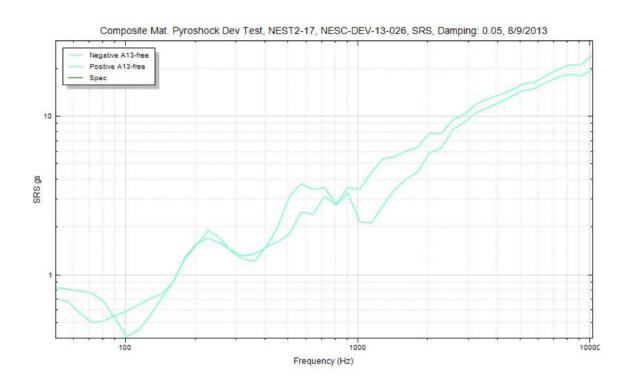
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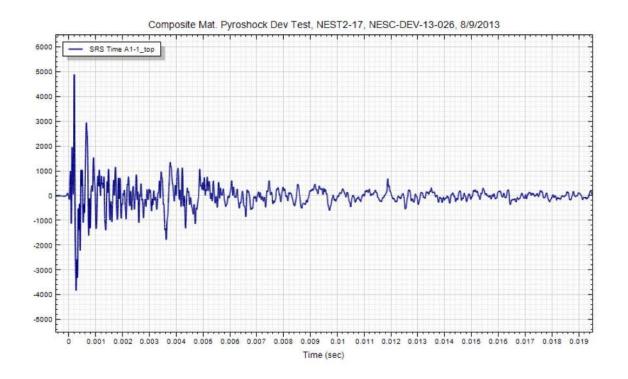
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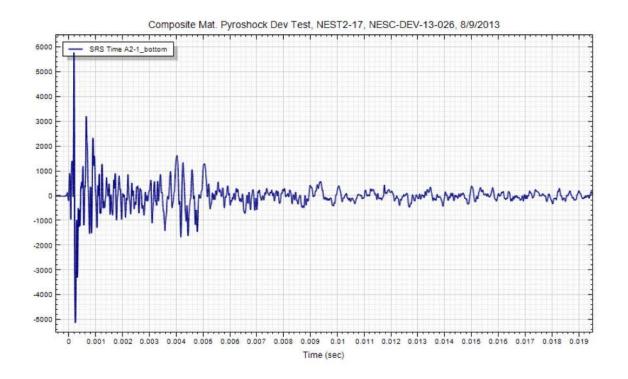
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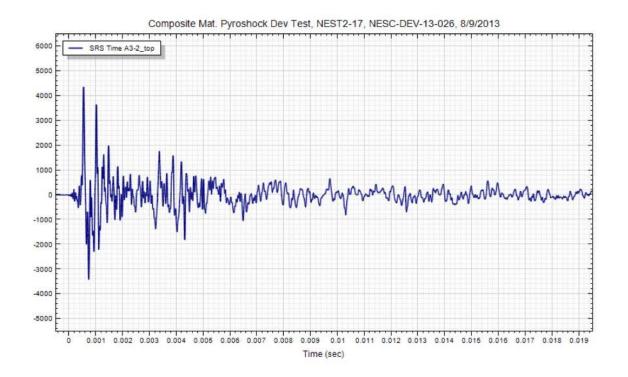
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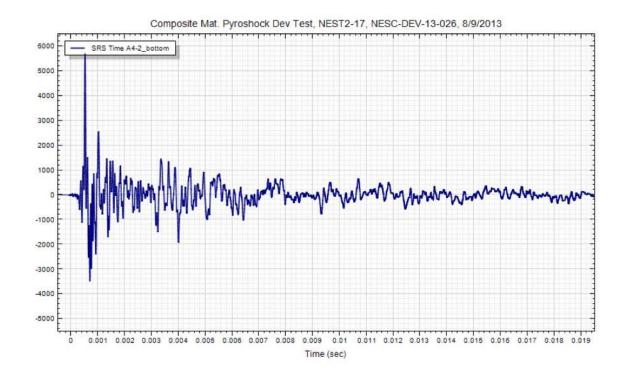
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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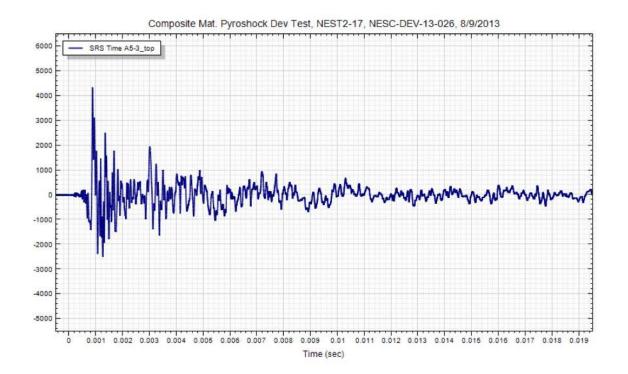
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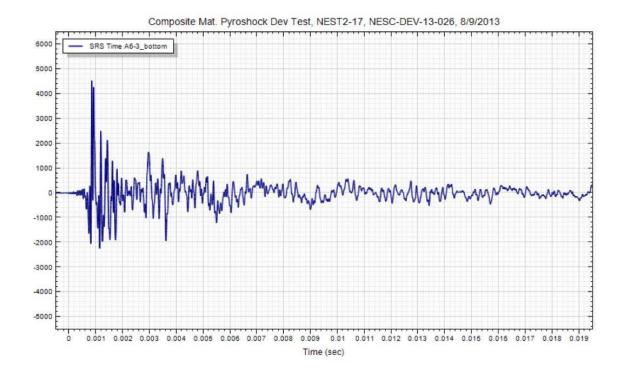
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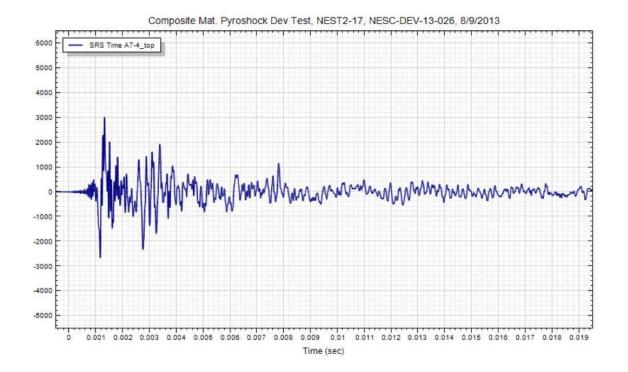
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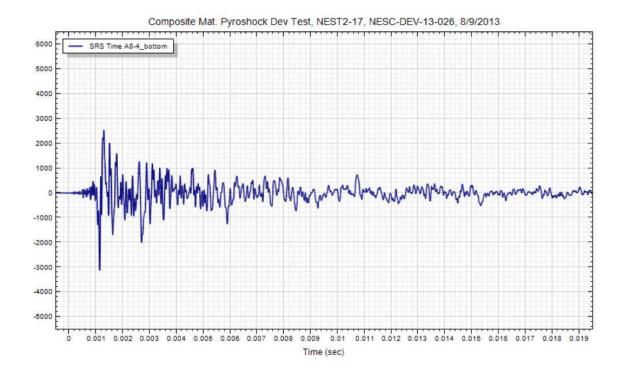
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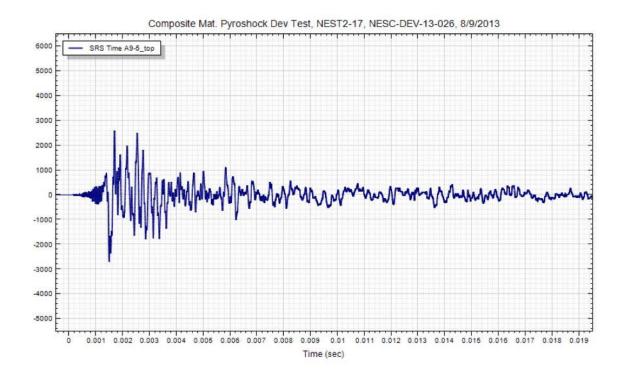
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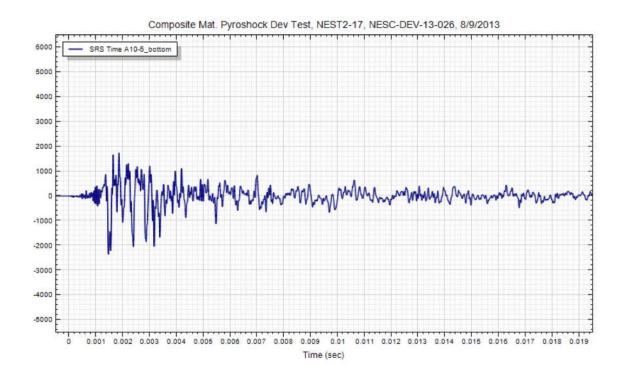
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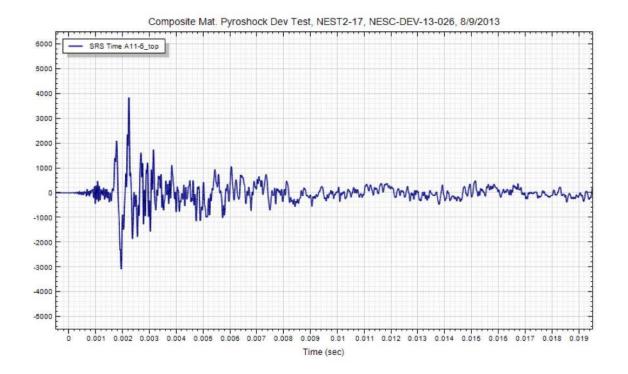
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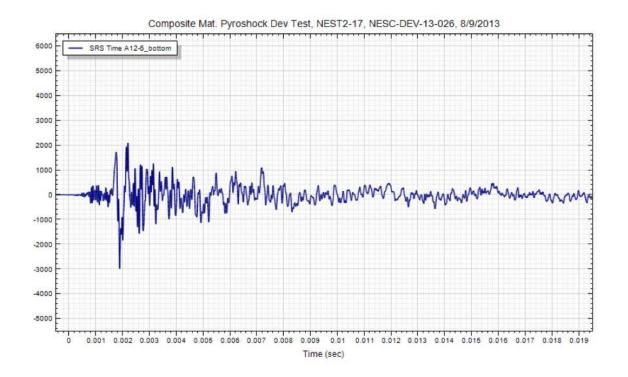
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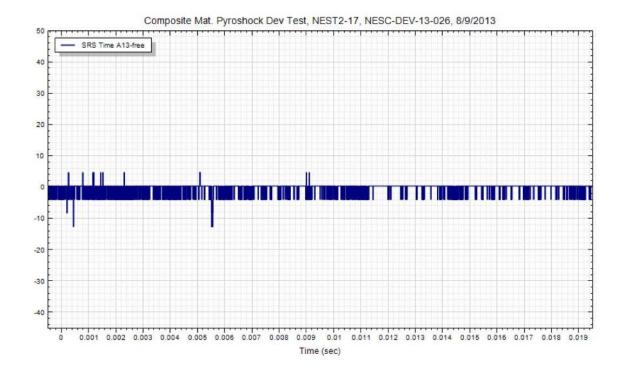
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-026 Composite Materials Shock Test

### Test #8 Accelerometer Data Panel 0332A017



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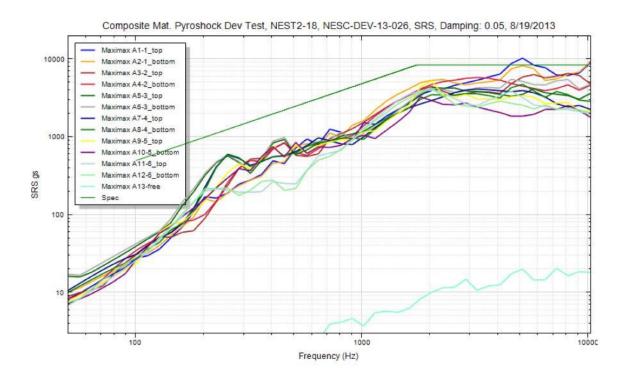
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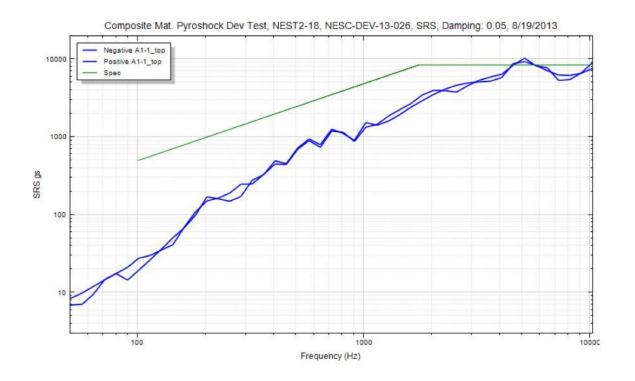
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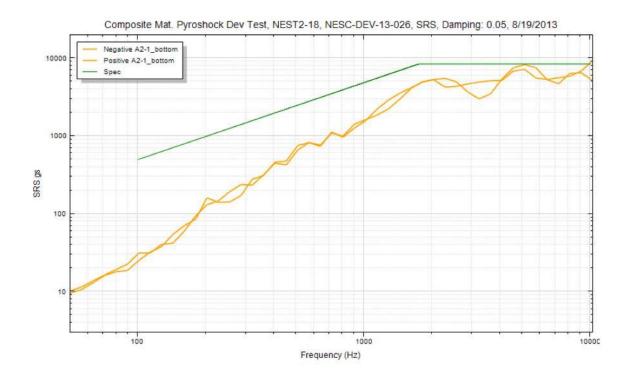
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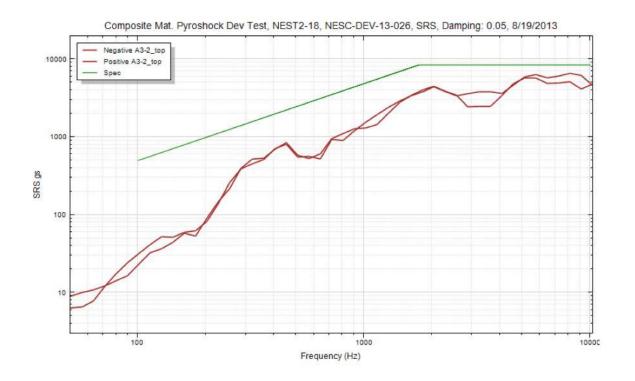
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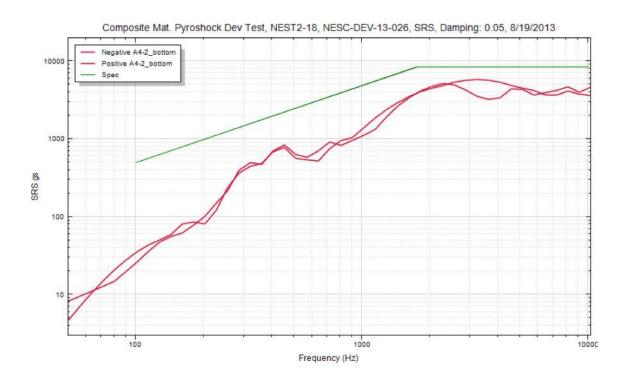
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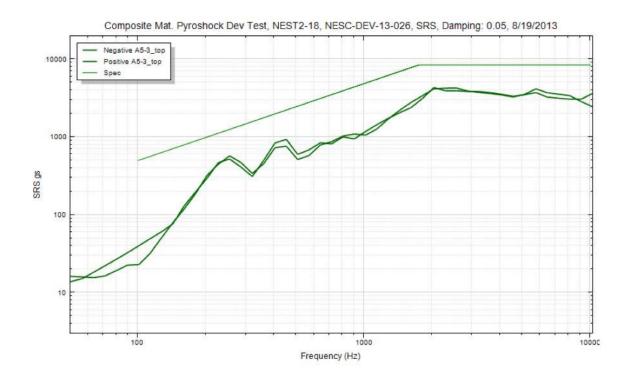
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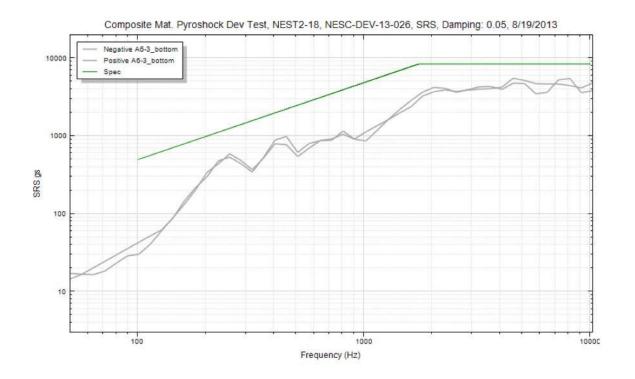
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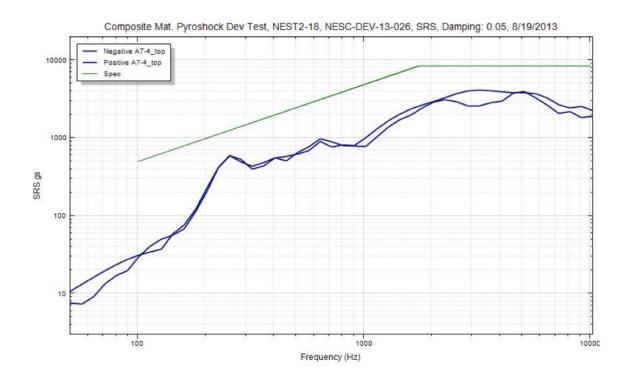
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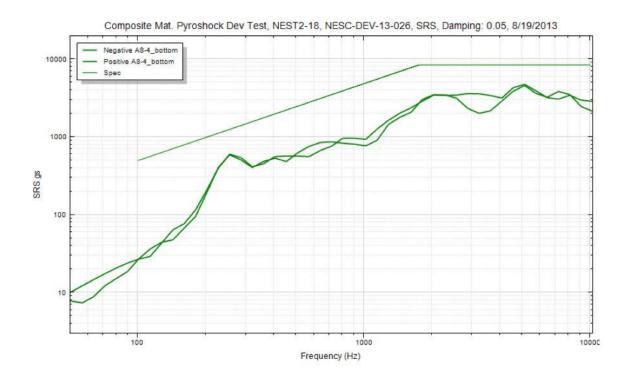
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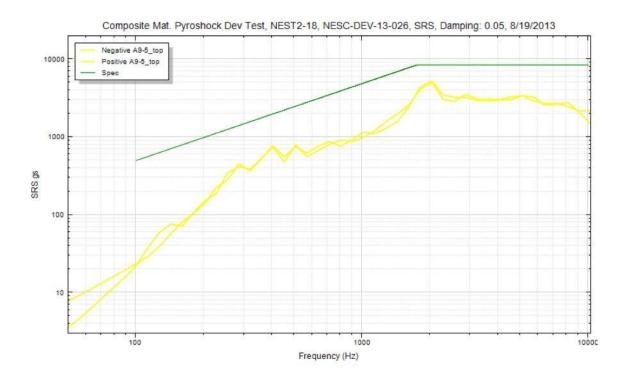
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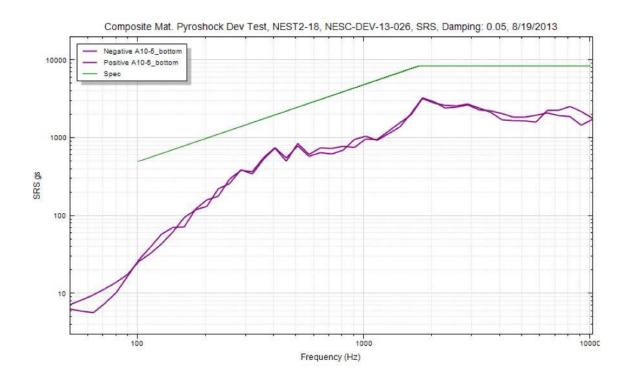
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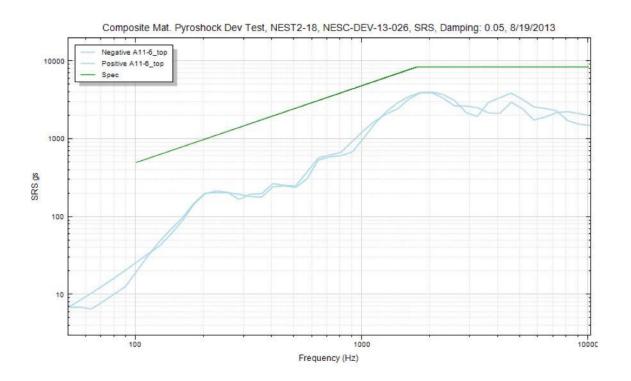
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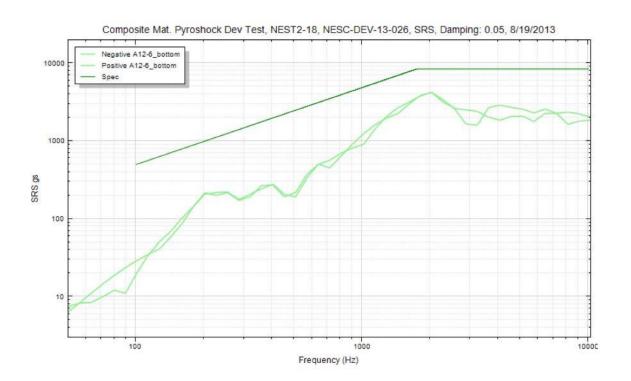
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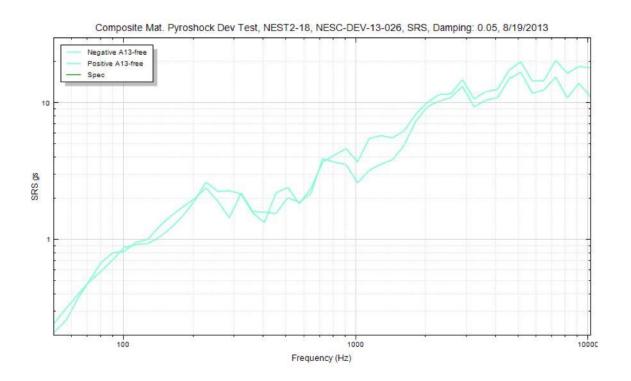
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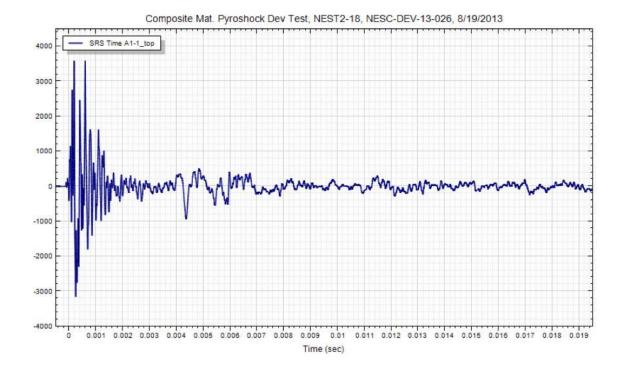
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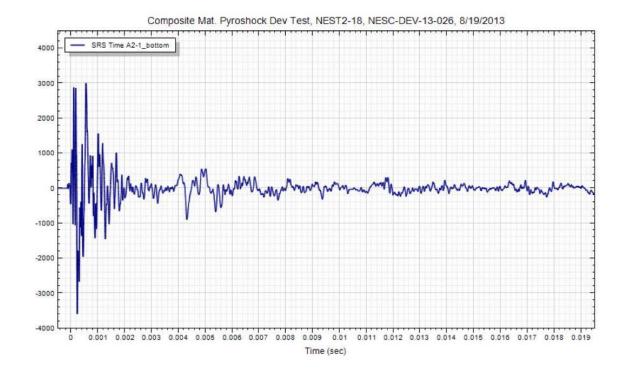
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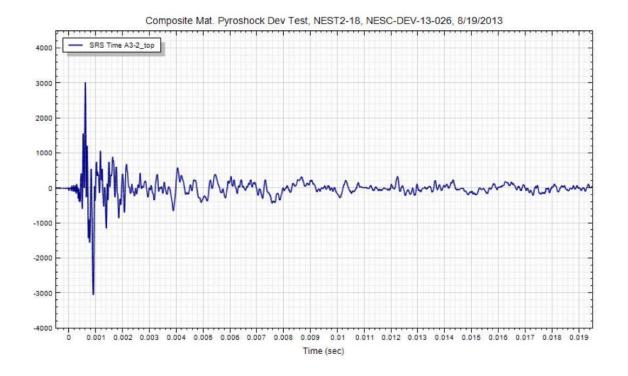
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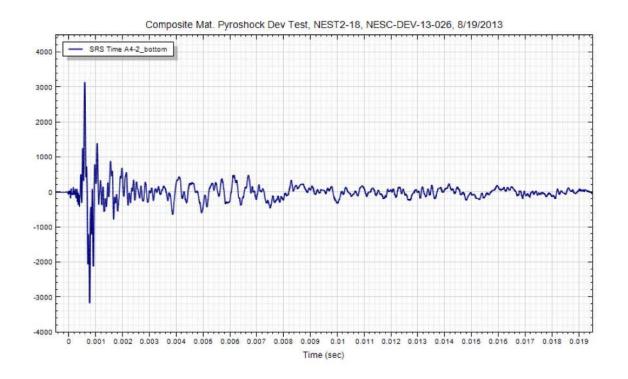
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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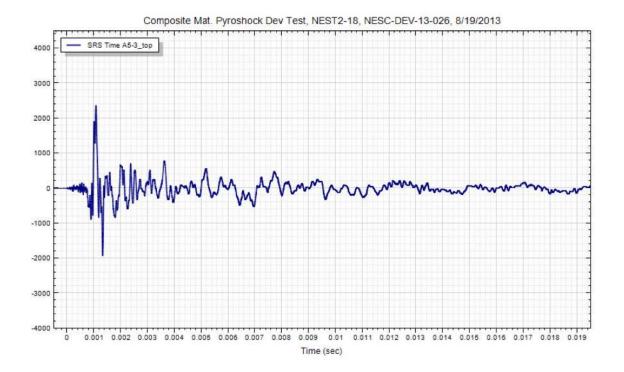
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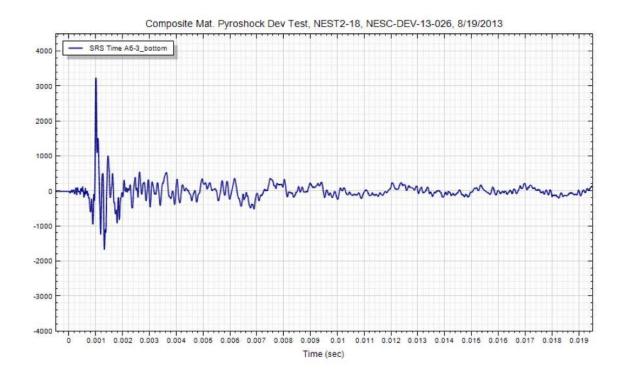
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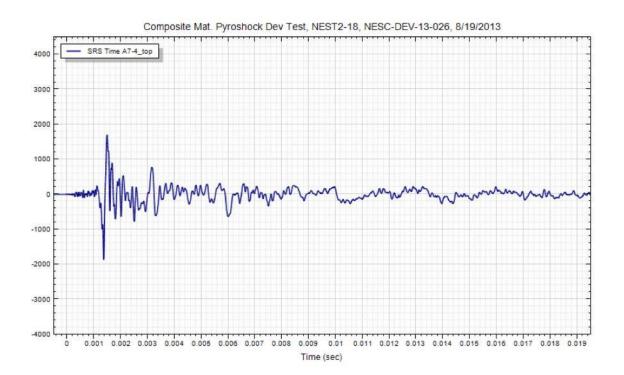
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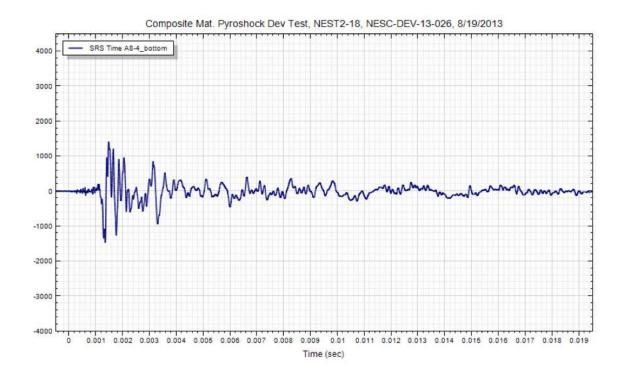
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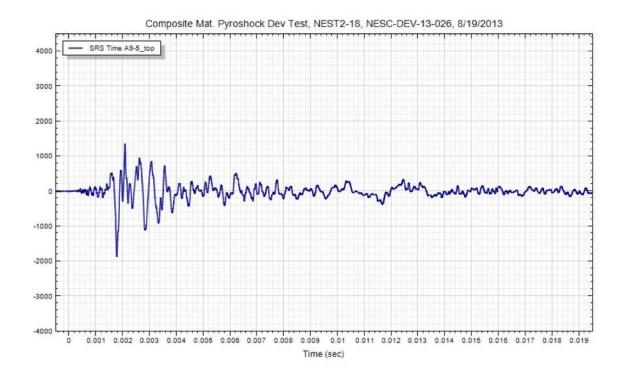
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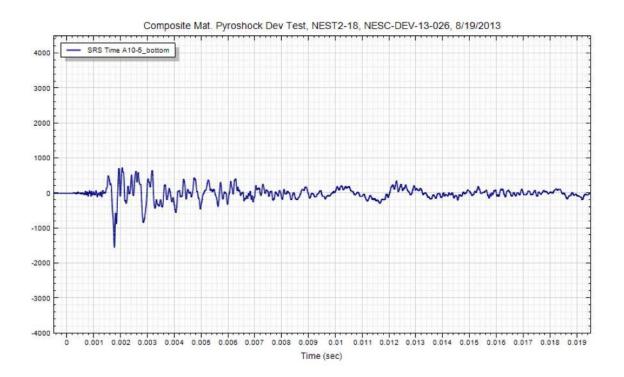
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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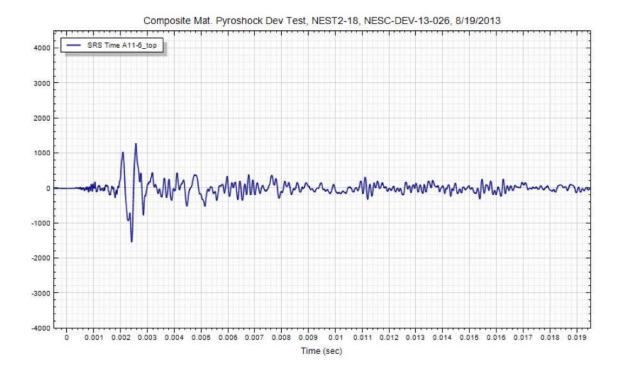
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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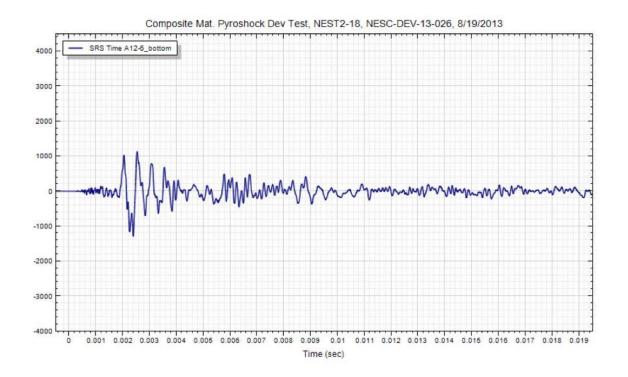
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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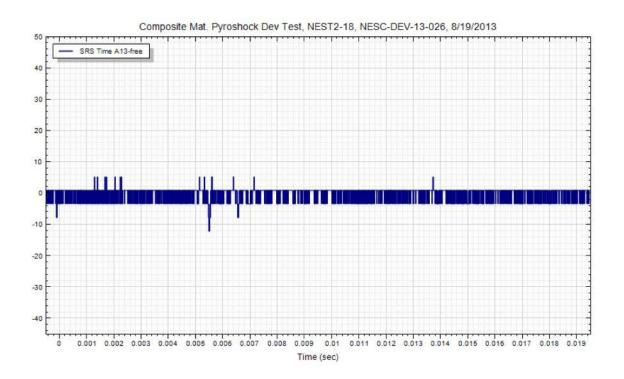
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-026 Composite Materials Shock Test

### **Test and Checkout Procedure**



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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

As Run Record Copy

> NESC-DEV-13-026 5/30/2013

George C. Marshall Space Flight Center Marshall Space Flight Center, Alabama 35812

### TEST AND CHECK-OUT PROCEDURE

ET40 / VIBRATION, ACOUSTICS, AND SHOCK TEAM

# COMPOSITE MATERIALS PYROSHOCK DEVELOPMENT TEST

Group II - Tests 11 to 18

This Procedure Describes Safety Critical Operations



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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Composite Materials	Revision: Baseline		
Group II - Tests 11 to	18	Date: 5/30/2013	Page 1 of 18
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

#### 1.0 <u>INTRODUCTION</u>

#### 1.1 PURPOSE

The purpose of this procedure is to define the steps necessary to perform a pyrotechnic shock test in the Pyrotechnic Shock Facility in Building 4619 using pyrotechnic devices.

Test Matrix Test Articles: 3'x6' Sandwich composite panels with LSC plate and LSC backing plate. The 8 test articles of the test matrix are described in test plan, Table III, Group II, tests 11 to 18.

Program: <u>NESC</u> Type of Test: <u>Pyrotechnic Shock Development Test</u>
Test Purpose: <u>To capture the acceleration time histories for group II – Test 11 to 18 tests.</u>

The Pyrotechnic Shock Facility is located in Rooms 170, 170A and 170B of Building 4619. Room 170A is designated as the Control Room. The area between Room 169 and 170 is used for storage of secondary pyrotechnic devices. Room 170B is used for storage of initiators. All detonation of pyrotechnic devices will be in Room 170.

#### 1.2 SCOPE

This document contains the steps and/or references the procedure to conduct the test.

#### 2.0 SAFETY

Follow all emergency and safety requirements specified in ET01-DYN-SHK-FOP-001.

#### 2.1 Responsibilities

The Test Engineer will be responsible for all activities occurring in the hazardous test area and for the safety of personnel involved in the test activities. It is the responsibility of each individual in a test program to fully comply with the requirements of this document and to report any individual not complying. Failure to do so could lead to serious personnel injuries or death.

#### 3.0 TEST REQUIREMENTS AND INFORMATION

#### 3.1 DOCUMENTS

#### 3.1.1 APPLICABLE DOCUMENTS

Test Requirements: Pyroshock Response Characterization of Composite Materials Test Plan Revision B, NESC Task # TI-12-0783 (SLS ADO-21), 5/17/2013

Test Procedure: ET01-DYN-SHK-FOP-001 Pyrotechnic Shock Tests



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

 $97M00200\text{-}\mathrm{GRP}$  II-PANEL 11-18 COMPOSITE TEST PANEL, GROUP II, TEST #11-#18

97M00202 LSC BACKING PLATE, COMPOSITE TEST PANEL PATHFINDER 97M00203-MOD LSC PLATE, COMPOSITE TEST PANEL PATHFINDER 97M00204-MOD-2-10 LSC SHIM, COMPOSITE TEST PANEL 10 GPF LSC 97M00204-MOD-2-22 LSC SHIM, COMPOSITE TEST PANEL 22 GPF LSC

#### 3.1.2 REFERENCED DOCUMENTS

ET01-DYN-OWI-001 Documentation Control

ET01-DYN-OWI-002 Test Operation Procedure Preparation and Change Control

- 3.2 TEST INFORMATION
- 3.2.1 The instrumentation locations are given in the drawings listed in the applicable documents and appendix A for the test.
- 3.2.2 Pyrotechnic shock tests may be performed on the test article in the order and configuration directed by the test requester.
- 3.2.3 The shock test will be performed on a room temperature test article.
- 3.3 TEST REQUIREMENTS
- 3.3.1 The Test Engineer will be in charge of all test preparations and activities.
- 3.3.2 All activities will be coordinated with the Test Engineer.
- 3.3.3 All changes to the procedure will be coordinated with the Test Engineer.
- 3.3.4 The development test articles will be tested with pyrotechnic shock test runs as directed by the test requester. The test article information will be recorded in this TCP.

#### 4.0 TEST DATA

- a. The test data includes a time history of the real time shock recorded over a 20 millisecond or longer interval and the units are g's peak versus time.
- b. The second plot is a Shock Response Spectrum (SRS) using 5% damping and a 1/6 octave shock spectrum analyzer. The SRS is computed over the frequency band from 50 to 10,000 Hertz. The SRS units are g's versus frequency.
- c. The data will be acquired on a Nicolet BE256LE data acquisition system and the SRS analysis will be performed using a personal computer and the Shock Analysis Tool Analysis Software.
- d. Sample rate of 1 million samples per second will be used for response from the



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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

ET40 / Vibrati	on, Acoustics, and Shock Te	am -
Composite Materials Pyroshock Development Test	NESC-DEV-13-026	Revision: Baseline
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#### accelerometers.

#### **TEST SETUP** 5.0

#### TEST ARTICLE AND SHOCK PLATE SETUI

- a. The test setup is shown in Appendix A.
- b. Suspend the shock plate from ceiling using straps or cables and shackles.
- c. Suspend 1 accelerometer near the plate. Connect to data system for recording.
- d. At the start of each test day, complete ET01-DYN-SHK-FOP-001, section 6.

#### TEST OPERATION

#### COMPOSITE MATERIALS PYROSHOCK DEV. TEST

Record and verify the test information below and in appendix C.

Group II - Test No.: 11 Panel #11 Date: 9-31-2013 Test Article Desc.: IM7/ TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape PID# 03324 611 Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4'5"

- b. Verify that the shock plate is ready for testing per section 5.1.
- Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to  $25\pm5$  in.-lb.

  Torque wrench: 1658783 Torque value: 25 Due: 10-4-2013 Cal.: 4-4-2013
- d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 55±5 ft-lb. 240 in-16. 95-31-303
- Torque wrench: M658396 Torque value: 240 Due: 10-8-203 Cal: 4-8-203

  e. D-ring ½-13 bolt's torque to 28 ft-lb. 240/a-1ls. LSC flute 3-coloss, Freed 9-coloss 220/a-1ls.

  Torque wrench: m658396 Torque value: 2220 Due: 10-8-203 Cal: 4-8-203

  f. Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.

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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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#### 6.2 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group II Test No.: 12 Panel #14 Date: 4-28-2013 Test Article Desc.: 1M7/
  TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Rohacell
  Foam & Tape PID# 0332 Ap /4
  Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al
  Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.

  Torque wrench: M658783 Torque value: 25 Due: 10-4-20/3 Cal.: 4-4-
- Torque wrench: M658783 Torque value: 25 Due: 10-4-2013 Cal.: 4-4-2013
  d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55±5 ft-lb: 240 in the Torque wrench: M658376 Torque value: 240 Due: 10-6-2013 Cal: 4-8-2013
- Torque wrench: M658396 Torque value: 240 Due: 10-6-2013 Cal: 4-8-2013

  e. D-ring ½-13 bolt's torque to 28-ft-lb. 240 inlb. LSC Plate D-rings; Fac and D-rings 220 in-lb. 1CA

  Torque wrench: M658396 Torque value: 249/20 Due: 16-8-2013 Cal: 4-8-2013

  5-31-2013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

QCE 6-28-2013

#### START OF THE COMPOSITE MATERIALS PYROSHOCK DEV. TEST

6.3

- a. Record and verify the test information below and in appendix C.

  Group II Test No.: 13 Panel #13 Date: 7-16-213 Test Article Desc.: IM7/
  TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Rohacell
  Foam and Tape PID# 0332 A 013
  Shock Source. LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al
  Actual Length Used: 4' 0.5"
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.

  Torque wrench: M658783

  Torque value: 25 Due: 10-4-2013 Cal.: 4-4-6
- Torque wrench: <u>M658783</u> Torque value: <u>25</u> Due: <u>10-4-2013</u> Cal.: <u>4-4-2013</u> d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to <u>55±5 ft-lb.</u> <u>#240in-lb.</u> <u>10-2013</u> Cal: <u>4-8-2013</u> Cal: <u>4</u>
- Torque wrench: M6583% Torque value: 240 Due:10-8-2013 Cal:4-8-2013

  e. D-ring ½-13 bolt's torque to 28 ft-lb. 240 in-lb 450 ftek D-rings; For end D-rings 220 in-lb 100

  Torque wrench: M658396 Torque value: 240 Due:10-8-2013 Cal:4/8/013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

JCB 7-16-2013



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#### START OF THE COMPOSITE MATERIALS PYROSHOCK DEV. TEST 6.4

- a. Record and verify the test information below and in appendix C. Group II - Test No.: 14 Panel #12 Date: 7-17-203 Test Article Desc.: 1M7/ TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb and Tape PID# 0332A012 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.
- Torque wrench: M658783 Torque value: 25 Due: 10-4-2013 Cal.: 4-4-2013 d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 55.15 ft-lb. 240 in 16 25.15 ft-lb.
- Torque wrench: M(58396 Torque value: 240 Due: 10-8-203Cal: 4-8-2013

  e. D-ring ½-13 bolt's torque to 28 ft-lb. 240 in-b. 450 Note 10-1:03; For and 10-1:03 220

  Torque wrench: M(58396 Torque value: 240 Due: 10-8-20)Cal: 4-8-2013
- Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

#### 6.5 START OF THE COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C. Group II - Test No.: 15 Panel #16 Date: 7-18-243 Test Article Desc.: IM7/ TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape PID# 0332 A016 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.

  Torque wrench: M658783 Torque value: 25 Due: 10-4-2013 Cal.: 4-4-2013

  d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55±5 ft-lb. 240 /a-16
- Torque wrench: M658396 Torque value: 240 Due: 10-9-1013 Cal: 4-8-2013
- D-ring 1/2-13 bolt's torque to 28 ft-lb. 240 in-lb. LSC flate D-ring; Fax and D-rings 220 in-lb.
  Torque wrench: M658396 Torque value: 24/220 Due: 10-8-2011 Cal: 4-8-2013.
- Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.



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#### START OF THE COMPOSITE MATERIALS PYROSHOCK DEV. TEST 6.6

- a. Record and verify the test information below and in appendix C. Group II - Test No.: 16 Panel #18 Date: 7-23-2413 Test Article Desc.: IM7/ TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Rohacell Foam & Tape PID# @ 332A @ 18 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.
- Torque wrench: M65 87 83 Torque value: 25 Due: 10-4-2013 Cal.: 4-4-2013 d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55.15 ft-lb. 240in-lb. 3-51-2013 Torque wrench: M65 8376 Torque value: 240 Due: 10-8-2013 Cal: 4-8-2013
- e. D-ring 1/2-13 bolt's torque to 28 ft lb. 240in-lh. LSc flat D-mass; Freed P-mass 220in-h. 25-31-411
  Torque wrench: #65 8396 Torque value: 24/220Due: 10-8-2013 Cal: 4-8-2013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

### Vgca 7/23/2013

#### 6.7 START OF THE COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C. Group II - Test No.: 17 Panel #15 Date: 8-9-2013 Test Article Desc.: IM7/ TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape PID# 03324015 Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4' 0.5"
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb. Torque wrench: M658783 Torque value: 25 Due: 10-4-20/3 Cal.: 4-4-20/3
- d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55±5-ft-lb. 240 in-lb.
- Torque wrench; M658396 Torque value: 246 Due: 10-8-2013 Cal: 4-8-2013
  e. D-ring 1/2-13 bolt's torque to 28-8-16. 240 in-16. LSC flate D-rings; Far and D-rings 220in-16 Torque wrench: 1658396 Torque value: 40/20 Due: 10-8-2013 Cal: 4-8-2013
- Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.

VACD 8/1/2013



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#### START OF THE COMPOSITE MATERIALS PYROSHOCK DEV. TEST 6.8

a. Record and verify the test information below and in appendix C.

Group II - Test No.: 18 Panel #17 Date: 8-19-2013 Test Article Desc.: IM7/ TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Rohacell Foam and Tape PID# **0332A017** Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4' 0.5"

- Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.

  Torque wrench: M658783 Torque value: 25 Due: 0-4-26/3 Cal.: 4-4-26/3

  d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55±5 ft-lb. 240/-
- Torque wrench: 1658396 Torque value: 240 Due: 10-8-2413 Cal: 4-8-243

  e. D-ring ½-13 bolt's torque to 28-8-16. 240 in-16. LSC Plate D-rings; For end D-rings 220 in

  Torque wrench: 1658396 Torque value: 242 Due: 10-8-2613 Cal: 4-8-2613 16. 1c4. 5-31-2013
- Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.

#### 7.0 RECORDS

The test report for this test will control and include the following records:

- a. This "AS RUN" TCP.
- b. The test data and the equipment list.

The test report is controlled by ET01-DYN-OWI-001, Documentation Control. However, due to the ITAR designation for the test results, the test report and data will be securely controlled. The test report will be available no later than 30 days after test completion. The Test Requirements will not be included in this TCP or in the report, but a copy may be filed with the report for future reference.

#### 8.0 TOOLS, EQUIPMENT, AND MATERIALS

The equipment used during this test will be listed in a table as part of the test report. The list will include test equipment calibration due dates.

#### PERSONNEL TRAINING AND CERTIFICATION 9.0

Personnel certified as Propellant and Explosive Handler are required to conduct this



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#### POST-TEST VERIFICATION

The Test and Check-out Procedure NESC-DEV-13-026 has been satisfactorily completed and documented.

Test Engineer / ET40 Date

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### APPENDIX A

TEST SETUP

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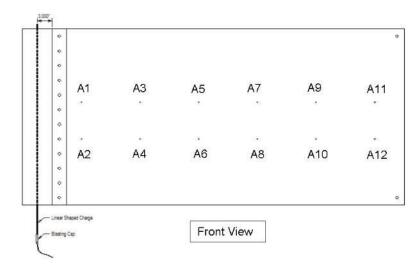
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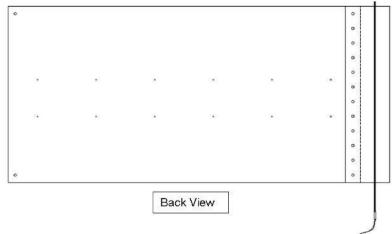
**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

Composite Materials Pyroshock Development Test

Test Article Panel: Sandwich Composite, Vertical Position

Supports: Straps and Shackles





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### APPENDIX B

GROUP II - TEST #11 - #18 PLY LAYUP

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Group II Tests #11-#18 Fill and Ply Layup

Panel Number	Test Number	Material	Panel Thickness	Fill/Ply	Orientation	Туре	LSC Core
11	11	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Al Honeycomb & Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10
12	14	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Al Honeycomb & Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22
13	13	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Rohacell Foam &Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10
14	12	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Rohacell Foam &Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22
15	17	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Al Honeycomb & Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10
16	15	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Al Honeycomb & Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22
17	18	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Rohacell Foam &Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10
18	16	Composite, IM7/TC350	8 Ply/ 1- inch thick fill	Rohacell Foam &Tape	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22



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### APPENDIX C

TEST DATA SHEET

(Reference Appendix B for Test Number and Corresponding Test Panel Number)



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#### TEST DATA SHEET

Group: II - Test No.: 11 Panel #11 Date: 5-31-2013 Test Article Desc.: IM7/TC350 Sandwich

Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 11 Material: IM7/TC350 PID# 0332A69

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4 25"

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N									
1	C02	31334	2	D02	43026	3	D02	43628	4	C02	3/33/
5	C02	31328	6		43029	7	D02	43179	8	C02	31351
9	C02	31330	10	D02	43180	11	D02	43181	12	C02.	40274
13	B02	11439						17.7			

Aluminum LSC panel severance: (Yes No) Not Completely

Post-test visually inspected observations: Bolts to good between 150-240 in-16. The Conter

bolts were at 150 in-16. The accels both torqued to at least 20 in-16.

Group: <u>II</u> - Test No.: <u>12 Panel #14</u> Date: <u>6/28/2013</u> Test Article Desc.: <u>1M7/TC350 Sandwich Composite Panel</u>, see appendix B for ply layup, 3'x6', Rohacell Foam & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 14 Material: IM7/TC350 PID# 0332A014

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: \_\_\_\_\_ LSC Core Load: 22 GR/FT Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 41

Loc.	Model	S/N									
1	D02	43026	2	C02	31340	3	C02	31338	4	D02	43028
5	D02	43029	6	C02	3/333	7	C02	40292	8	D02	43179
9	D02	43180	10	C02	40295	11	C02	31336	12	D02	43181
13	B02	11439				-	100		1		

Aluminum LSC panel severance: (Yes) No )

Post-test visually inspected observations: No loss bolts or accels. after test



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#### TEST DATA SHEET, cont.

Group: II - Test No.: 13 Panel #13 Date: 7-16-2013 Test Article Desc.: IM7/TC350 Sandwich

Composite Panel, see appendix B for ply layup, 3'x6', Rohacell Foam & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 13 Material: IM7/TC350 PID# 0322A8/3

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4' 05"

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	C02	31334	. 2	D02	43026	3	D02	43028	+ 4 ·	C02	31331
5.	C02	31328	6 .	D02	43029	7	D02	43179	. 8 -	C02	31351
9 .	C02	31330	10	D02	43180	-11	D02	43(8)	12	C02	40274
13	B02	11439	S			17377330	Lower	2.0			

Aluminum LSC panel severance: (Yes) No )

Post-test visually inspected observations: Bolts to qual to 220 in 16. Accels. torqual to

at least 20 in: 16. after test.

Group: II - Test No.: 14 Panel #12 Date: 7-17-2013 Test Article Desc.: IM7/TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 12 Material: IM7/TC350 PID# 0332 A012

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder
Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: \_\_\_\_\_ LSC Core Load: 22 GR/FT. Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1.	D02	43026	2	C02	31340	3	C02	3) 338	4	D02	43028
5	D02	43029	6	C02	3/333	7	C02	40292	8:	D02	43179
9	D02	43180	10	C02	40295	11	C02	31336	12	D02	43181
13	B02	11439			100			17 H			

Aluminum LSC panel severance: (Yes)/ No )

Post-test visually inspected observations: No loss accels

less than 210 in-16. The far side D-rings were not loose



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#### TEST DATA SHEET, cont.

Group: II - Test No.: 15 Panel #16 Date: 7-18-20 Test Article Desc.: IM7/TC350 Sandwich

Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 16 Material: IM7/TC350 PID# 0332A014

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

LSC Core Load: 22 GR/FT Explosive Material: CH-6 Shock Source: LSC L/N: none

Sheath: Aluminum Actual Length Used: 4

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
.1	. C02	31334	. 2 .	D02	43026	.3	D02	43028	: 4.,	C02	31331
5	C02	31328	,6	D02	43029	1.7	D02	43179	8	, C02	31351
9	C02	31330	10	D02	43180	11.	D02	43181	12	C02	40274
13	B02	11439	5	, - B					4885		7

Aluminum LSC panel severance: (Yes)/ No )

Post-test visually inspected observations: The accels. to good to atleast 20 int b.

Bolts torqued slightly less than 210 in-16. on LSC Plate.

Group: II - Test No.: 16 PANEL #18 Date: 7-23-243 Test Article Desc.: IM7/TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Rohacell Foam & Tape Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 18 Material: IM7/TC350 PID# 0332A018

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder
Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 22 GR/FT Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
. 1	D02	43026	.2	C02	31340	3	C02	31338	4	D02	43028
5	D02	43029	6 :	C02	31333	7	C02	40292	8	D02	43/79
9	D02	43180	10	C02	40295	.11	C02	31336	12	D02	43181
13	B02	11439				3 .			1447652		- 0

Aluminum LSC panel severance: (Yes) No )

Post-test visually inspected observations: No loose bolts or accels, after test.



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

ET40 / Vibrati	on, Acoustics, and Shock Te	am
Composite Materials Pyroshock Development Test	NESC-DEV-13-026	Revision: Baseline
Group II - Tests 11 to 18	Date: 5/30/2013	Page 18 of 18

#### TEST DATA SHEET, cont.

Group: II - Test No.: 17 Panel #15 Date: 8-9-2013 Test Article Desc.: IM7/TC350 Sandwich

Composite Panel, see appendix B for ply layup, 3'x6', Al Honeycomb & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 15 Material: IM7/TC350 PID# 0332A015

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder
Test Article Drawing #: 97M00203-MQD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4' 0.5"

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	C02	31334	2	D02	43026	3	D02	43028	4	C02	31331
5	C02	3/328	6	D02	43029	7	D02	43179	8	C02	31351
9	C02	31330	10	D02	43180	11	D02	43181	12	C02	40274
13	B02	11439					(c = 5) = 5		197		ilis v

Aluminum LSC panel severance: (Yes/No)

Post-test visually inspected observations: Bolts and accels. distant loose torque.

Group: II - Test No.: 18 Panel #17 Date: 8-19-2013 Test Article Desc.: IM7/TC350 Sandwich Composite Panel, see appendix B for ply layup, 3'x6', Rohacell Foam & Tape

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP II-PANEL 17 Material: IM7/TC350 PID# 8332A017

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: \_\_note LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4' 0.5"

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	D02	43026	2	C02	31340	3	C02	31338	4	D02	43028
- 5	D02	43029	6	C02	31333	7	C02	40292	8	D02	43179
9	D02	43180	10	C02	40 295	11	C02	31336	12	D02	43 181
13	B02	11439		W Romes							

Aluminum LSC panel severance: (Yes)/No)

Post-test visually inspected observations: No losse accels. o- bolts



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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-026 Composite Materials Shock Test

**Equipment List** 



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

		NESC-I	DEV-13-026		
		Equip	ment List		
Description	Manufacturer	Model/Version	ID/Serial Number	Location	Cal Due Date
Ch 1 A 1 1 1	ET40	Las		D.	U-16-15/21/2000
Shock Analysis Tool FEAM256	Nicolet	1.2.5 7.20			Verified 5/21/2009
			**********		Verified 7/26/2012
orque Wrench	Proto	6066C	M658396		10/8/2013
orque Wrench	Precision Instruments	44620	M658783		10/4/2013
ower Supply	Endevco	2793	M652262	Channels 1-15	1/10/2014
Data Acquisition System	Nicolet	BE256LE	2011288	Channels 2-8, 10-15	7/26/2013, 8/6/2014
Channel 2	Nicolet	614CB	001-2	A1	7/26/2013, 8/6/2014
Channel 3	Nicolet	614CB	001-3	Λ2	7/26/2013, 8/6/2014
Channel 4	Nicolet	614CB	001 4	A3	7/26/2013, 8/6/2014
Channel 5	Nicolet	614CB	002-1	A4	7/26/2013, 8/6/2014
Channel 6	Nicolet	614CB	002-2	A5	7/26/2013, 8/6/2014
Channel 7	Nicolet	614CB	002-3	A6	7/26/2013, 8/6/2014
Channel 8	Nicolet	614CB	002-4	A7	7/26/2013, 8/6/2014
Channel 10	Nicolet	614CB	003-2	A8	7/26/2013, 8/6/2014
Channel 11	Nicolet	614CB	003-3	AS	7/26/2013, 8/6/2014
Channel 12	Nicolet	614CB	003-4	A10	7/26/2013, 8/6/2014
Channel 13	Nicolet	614CB	004-1	A11	7/26/2013, 8/6/2014
Channel 14	Nicolet	614CB	004-2	A12	7/26/2013, 8/6/2014
Channel 15	Nicolet	614CB	004-3	Λ13	7/26/2013, 8/6/2014
Accelerometer	PCB	350B02	11439	A13	4/29/2014
Accelerometer	PCB	350C02	31334	Set 1 A1	4/23/2014
Accelerometer	PCB	350C02	31331	Set 1 A4	4/24/2014
Accelerometer	PCB	350C02	31328	Set 1 A5	4/24/2014
Accelerometer	РСВ	350C02	31351	Set 1 A8	4/24/2014
Accelerometer	РСВ	350C02	31330	Set 1 A9	4/24/2014
Accelerometer	РСВ	350C02	40274	Set 1 A12	4/24/2014
Accelerometer	РСВ	350C02	31340	Set 2 A2	4/23/2014
Accelerometer	РСВ	350C02	31338	Set 2 A3	4/24/2014
Accelerometer	PCB	350C02	31333	Set 2 A6	4/24/2014
Accelerometer	РСВ	350C02	40292	Set 2 A7	4/24/2014
ccelerometer	PCB	350C02	40295	Set 2 A10	4/24/2014
ccelerometer	PCB	350C02	31336	Set 2 A11	4/24/2014
ccelerometer	PCB	350D02	43026	Set 1 A2, Set 2 A1	4/23/2014
ccelerometer	PCB	350D02	43028	Set 1 A3, Set 2 A4	4/23/2014
Accelerometer	PCB	350D02	43029	Set 1 A6, Set 2 A5	4/23/2014
Accelerometer	PCB	350D02	43179	Set 1 A7, Set 2 A8	4/22/2014
Accelerometer	PCB	350D02	43180	Set 1 A10, Set 2 A9	4/23/2014
Accelerometer	PCB	350D02	43181	Set 1 A11, Set 2 A12	4/23/2014

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ASA Engineering and Safety Center	
<b>Technical Assessment Report</b>	

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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-026 Composite Materials Shock Test

## **Data Acquisition Setups**



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

```
TEAM256 SETTINGS
                                                                                                 TESTS 1,3,7,8
  Date: 08-09-2013
  Time: 08:32:44
  Storage Path:
                               C:\TEAMPRO
  Filename:
                               001
C:\TEAM256
  File Number:
  Settings Path:
Settings File:
                               NES217.SET
 Export Path:
Export Format:
                               D:\ATEST\NESC_4\NEST2-17\RAWDAT~1
                               FAMOS
  Average Blocks:
  Between Cursors: No
  ********************** RECORDER SETTINGS *******************
  BE1
   Frequency A :
Pre Trigger :
                                               1,0000 MHz(Internal)
                                               48000 Samples (48.00 ms)
1000576 Samples (1.001 s)
   Segment A :
Number of Blocks :
Digital Event Channels :
   Analog Channels :
      Nr. Name Min
1 XXX_1 -6.000
2 NES_2 -28.85
                                       Max Units
6.000 kg's pk
28.85 kg's pk
                                                                                      Filter Trig
33.00 k Off
33.00 k Basi
                                                                  Coup. Amp.
GND +
                                                                                                      Basic
             NES_3 -30.93
                                        30.93 kg's pk
                                                                                       33.00
                                                                                                      Basic
             NES_3 -30.93

NES_4 -31.25

NES_5 -27.78

NES_6 -28.04

NES_7 -30.30

NES_8 -30.61

XXX_9 -6.000

NES_10 -28.04

NES_11 -27.03

NES_12 -31.58

NES_13 -30.61
                                                                                       33.00 k
33.00 k
33.00 k
                                                                   DĊ
DC
                                        31.25 kg's pk
27.78 kg's pk
                                                                                                      Basic
                                                                                                       Basic
                                        28.04 kg's
                                                                                                       Basic
                                                                                      33.00 k
33.00 k
33.00 k
                                       30.30 kg's pk
30.61 kg's pk
6.000 kg's pk
                                                                   DC
DC
                                                                                                      Basic
                                                                                                      Off
                                                                   DC
DC
DC
                                                                                      33.00
33.00
33.00
                                       28.04
27.03
       10
                                                  kg's pk
                                                                                                      Off
                                                  kg's pk
kg's pk
                                                                                                      off
                                        31.58
                                                                                                      Off
             NES_13 -30.61
NES_14 -27.27
                                       30.61 kg's pk
27,27 kg's pk
                                                                   DC
DC
DC
                                                                                       33.00 k
                                                                                                      Off
                                                                                      33.00
                                                  kg's pk
       14
                                                                                                      Off
              NES_15 -8.929
                                        B.929
                                                                                                      Off
           ROC_16 -55.56

ROC_17 -55.56

ROC_18 -55.56

ROC_19 -55.56

ROC_20 -55.56
                                        55.56 kg's pk
55.56 kg's pk
                                                                                       33.00 k
33.00 k
33.00 k
                                                                   DC
DC
                                                  kg's pk
                                                                                                      Off
                                        55.56 kg's pk
                                                  kg's pk
                                                                                       33.00 k
  20 ROC_20 -55.56 55.56 kg s px

Engineering Units Scaling

XXX_1 0 + 1.0000 k * Voltage (g's pk)

NES_2 0 + 9.6154 k * Voltage (g's pk)

NES_3 0 + 10.309 k * Voltage (g's pk)

NES_4 0 + 10.417 k * Voltage (g's pk)

NES_5 0 + 9.2593 k * Voltage (g's pk)

NES_6 0 + 9.3458 k * Voltage (g's pk)

NES_7 0 + 10.101 k * Voltage (g's pk)

NES_7 0 + 10.204 k * Voltage (g's pk)
                                        55.56 kg's pk
                                                                                       33.00 k
```



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

TESTS 2, 4, 5, 6 TEAM256 SETTINGS Date: 07-23-2013 Time: 08:21:22 Storage Path: C:\TEAMPRO Filename: 001 C:\TEAM256 File Number: Settings Path: Settings File: NES216.SET Export Path: Export Format: Average Blocks: D:\ATEST\NESC\_4\NEST2-16\RAWDAT~1 FAMOS Between Cursors: No \* RECORDER SETTINGS \*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\* BE1 1.0000 MHz(Internal) Frequency A: Pre Trigger : 48000 Samples (48.00 ms) Segment A : Number of Blocks : Digital Event Channels : 1000576 Samples (1.001 s) Analog Channels : Max Units 6.000 kg's pk 30.93 kg's pk 31.58 kg's pk Filter 33.00 k 33.00 k Nr. Name Min 1 XXX 1 -6.000 Coup. Amp. Trigger NES\_2 -30.93 DC DC DC DC Basic NES\_3 -31.58 NES\_4 -30.00 33.00 33.00 Basic 30.00 kg's pk 31.25 kg's pk 30.30 kg's pk 33.00 k Basic NES\_5 -31.25 NES\_6 -30.30 NES\_7 -28.57 NES\_8 -28.30 DC DC DC 28.57 kg's pk 28.30 kg's pk 6.000 kg's pk Basic Basic NES\_8 -28.30 XXX\_9 -6.000 NES\_10 -30.61 NES\_11 -31.58 NES\_12 -28.30 NES\_13 -27.78 NES\_14 -30.61 NES\_15 -8.929 30.61 kg's pk 31.58 kg's pk 28.30 kg's pk 10 11 Off Off 33.00 k 27.78 kg's pk 30.61 kg's pk 8.929 kg's pk 13 Off Off 14 15 14 NES\_14 -30.61 30.61 kg's pk DC
15 NES\_15 -8.929 kg's pk DC
16 ROC\_16 -55.56 55.56 kg's pk DC
17 ROC\_17 -55.56 55.56 kg's pk DC
18 ROC\_18 -55.56 55.56 kg's pk DC
19 ROC\_19 -55.56 55.56 kg's pk DC
20 ROC\_20 -55.56 55.56 kg's pk DC
Engineering Units Scalling
XXX 1 0 + 1.0000 k \* Voltage (g's pk)
NES\_2 0 + 10.309 k \* Voltage (g's pk)
NES\_3 0 + 10.526 k \* Voltage (g's pk)
NES\_4 0 + 10.000 k \* Voltage (g's pk)
NES\_5 0 + 10.417 k \* Voltage (g's pk)
NES\_5 0 + 10.101 k \* Voltage (g's pk)
NES\_6 0 + 10.101 k \* Voltage (g's pk)
NES\_7 0 + 9.5238 k \* Voltage (g's pk)
XXX 9 0 + 1.0000 k \* Voltage (g's pk)
XXX 9 0 + 1.0000 k \* Voltage (g's pk)
NES\_10 0 + 10.204 k \* Voltage (g's pk)
NES\_11 0 + 10.526 k \* Voltage (g's pk)
NES\_12 0 + 9.4340 k \* Voltage (g's pk)
NES\_13 0 + 9.2593 k \* Voltage (g's pk)
NES\_14 0 + 10.204 k \* Voltage (g's pk)
NES\_15 0 + 8.9286 k \* Voltage (g's pk)
NES\_15 0 + 8.9286 k \* Voltage (g's pk)
NES\_15 0 + 8.9286 k \* Voltage (g's pk)
NES\_17 0 + 9.2593 k \* Voltage (g's pk)
NES\_18 0 + 9.2593 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk)
NES\_19 0 + 10.204 k \* Voltage (g's pk) off Off Off (g's pk) (g's pk) (g's pk) ROC\_20 0 + 9.25 Trigger Settings : 9.2593 k \* Voltage (g's pk) Auto Trigger:

A CONTRACTOR OF THE PARTY OF TH	NASA Engineering and Safety Center Technical Assessment Report	Document #:  NESC-RP- 12-00783	Version: 1.0
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Composite Materials Subjected to Pyroshock Loading			

### **B4.** Group III (Sandwich Panel Tests)

The test report documenting the test results for the Group III composite sandwich panel tests are documented in the attachment below.

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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

National Aeronautics and Space Administration George C. Marshall Space Flight Center Marshall Space Flight Center, AL 35812



January 9, 2015

Reply to Altrn of: ET40-15-003

TO:

EV32/David O. Ordway

FROM:

ET40/Timothy C. Driskill

SUBJECT:

Composite Materials Pyroshock Development Test, Group III - Tests 19 to 28

and Group I Retests 1 to 4, NESC-DEV-13-037

REF:

ED73.1.

The solid and sandwich composite panels test articles were tested in the ET40 Pyrotechnic Shock Facility, building 4619, room 170. Testing was completed on June 11, 2014. The test was run in accordance with Test and Checkout Procedure, (TCP) NESC-DEV-13-037. Ten sandwich composite panels were tested and 4 solid composite panels were retested. A total of 14 tests were performed.

The accelerometer test setup is shown in appendix A of the TCP and in the photographs section of this report. Three composite and 2 aluminum LSC panels did not sever. Four tests noted a small degree of post-test torque loss to some bolts or accelerometers.

Please direct any questions or comments regarding this test to Mr. Craig Garrison at (256) 544-7197 or craig.garrison@nasa.gov.

Timothy C. Driskill

Branch Chief

Structural Dynamics Test Branch

Enclosure

cc

ET01/File (w/o enclosure)

ET40/File

C105/Steve Gentz

ES22/David Parsons



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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-037 Composite Materials Shock Test

**General Test Setup** 



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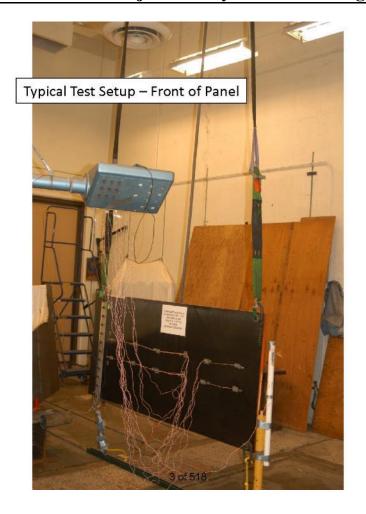
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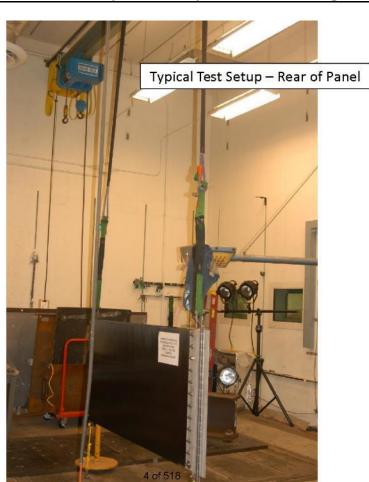
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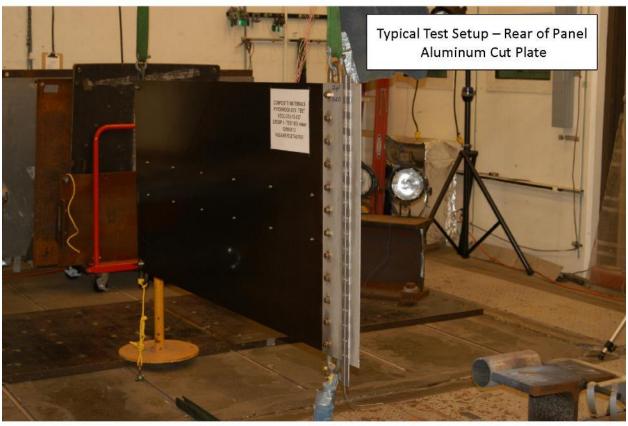
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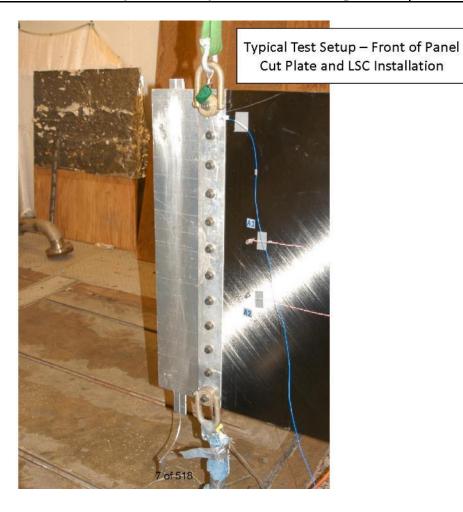
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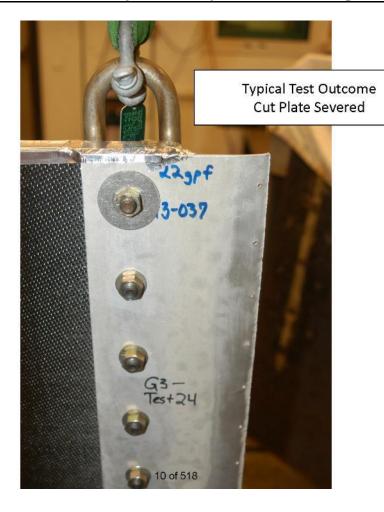
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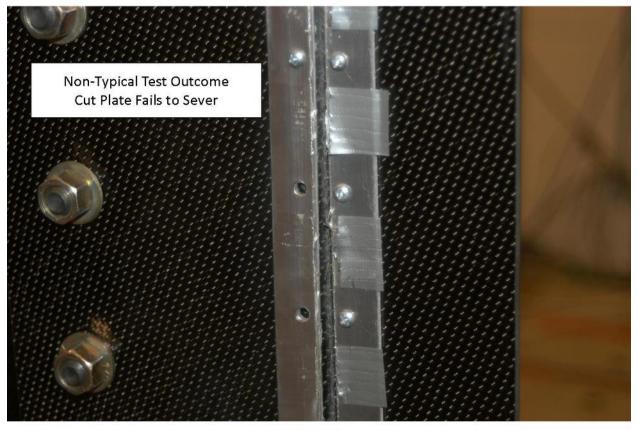
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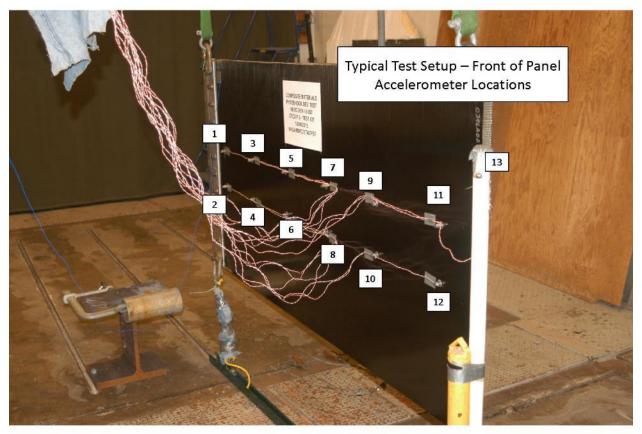
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

Test Matrix As Run				
Test	Date	Panel ID	Test ID	
1	10-31-13	0346A019	Group 3 - Test 19	
2	11-19-13	0346A020	Group 3 - Test 20	
3	11-25-13	0346A022	Group 3 - Test 22	
4	12-4-13	0346A021	Group 3 - Test 21	
5	12-12-13	0320A004	Group 1 - Test 4 Retest	
6	12-16-13	0320A002	Group 1 - Test 2 Retest	
7	12-18-13	0346A026	Group 3 - Test 26	
8	1-16-14	0326A010	Group 1 - Test 10 Retest	
9	1-23-14	0326A009	Group 1 - Test 9 Retest	
10	1-24-14	0346A028	Group 3 - Test 28	
11	4-18-14	0346A023	Group 3 - Test 23	
12	4-22-14	0346A024	Group 3 - Test 24	
13	6-10-14	0346A025	Group 3 - Test 25	
14	6-11-14	0346A027	Group 3 - Test 27	

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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

Panel Location	Set 1 Accelerometers	Set 2 Accelerometers		
1	350C02 31334	350D02 43026		
2	350D02 43026	350C02 31340		
3	350D02 43028	350C02 31338		
4	350C02 31331	350D02 43028		
5	350C02 31328	350D02 43029 350D02 43373 *		
6	350D02 43029 350D02 43373 *	350C02 31333		
7	350D02 43179	350C02 40292		
8	350C02 31351	350D02 43179		
9	350C02 31330	350D02 43180		
10	350D02 43180	350C02 40295		
11	350D02 43181	350C02 31336		
12	350C02 40274	350D02 43181		
13	350B02 11439	350B02 11439		
Set 1 used in tests 1, 4, 6, 8, 11, 13, and 14				

Set 1 used in tests 1, 4, 6, 8, 11, 13, and 14

Set 2 used in tests 2, 3, 5, 7, 9, 10, and 12 \* 43029 replaced with 43373 prion to 18 to 1



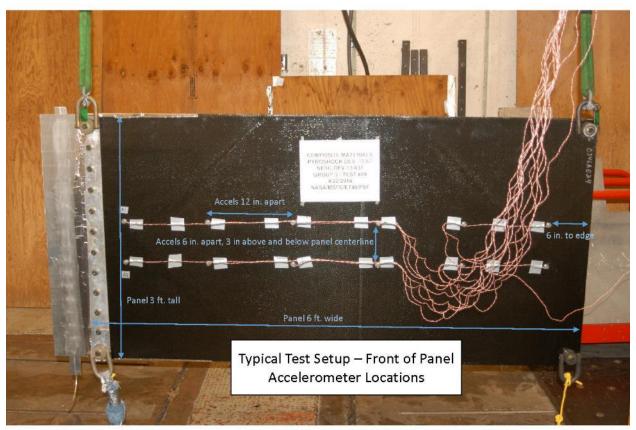
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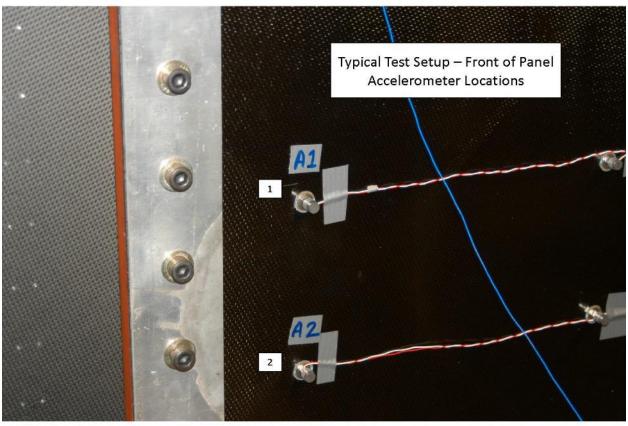
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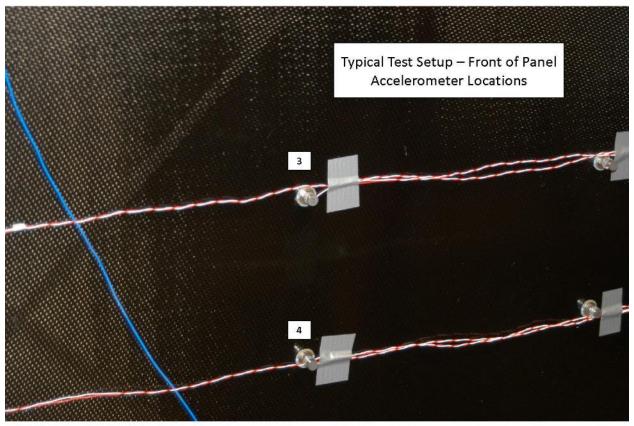
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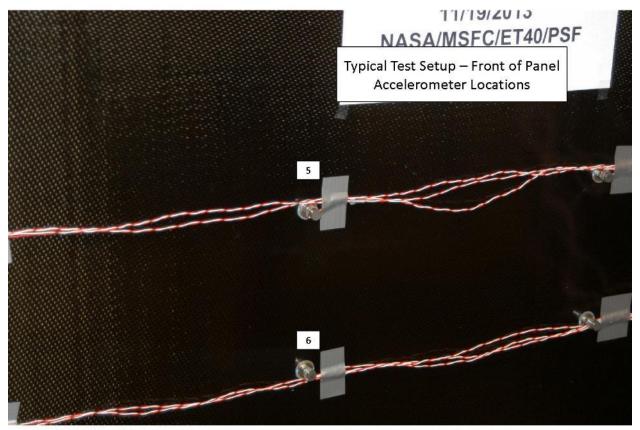
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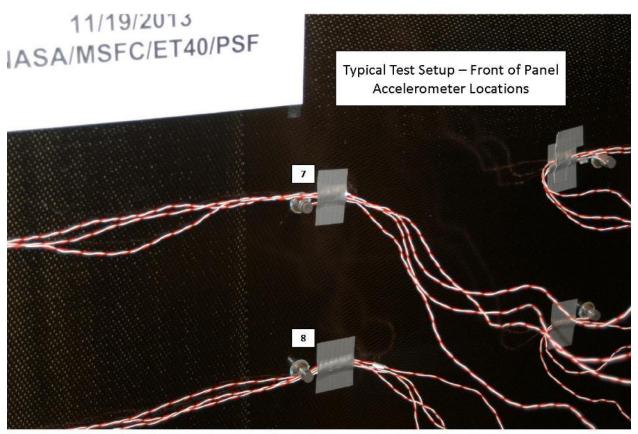
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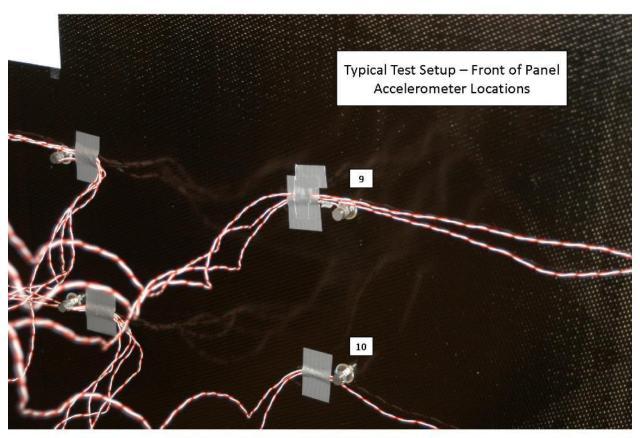
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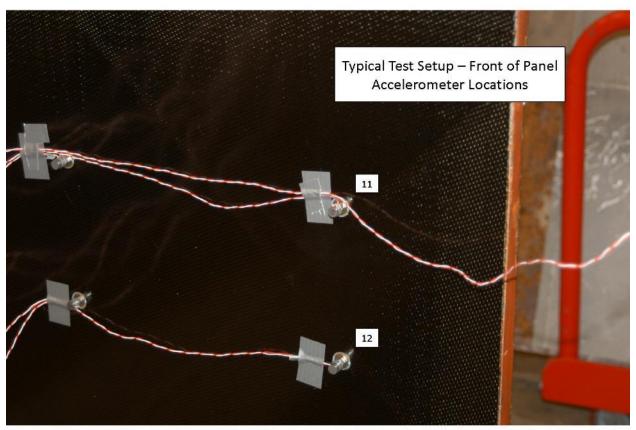
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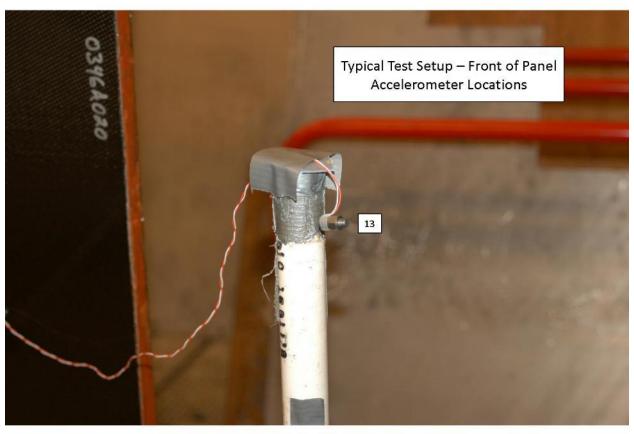
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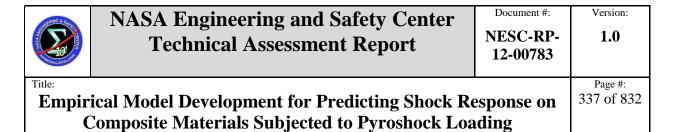
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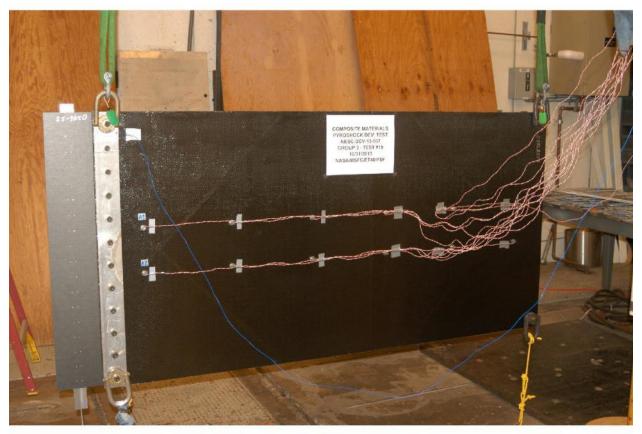
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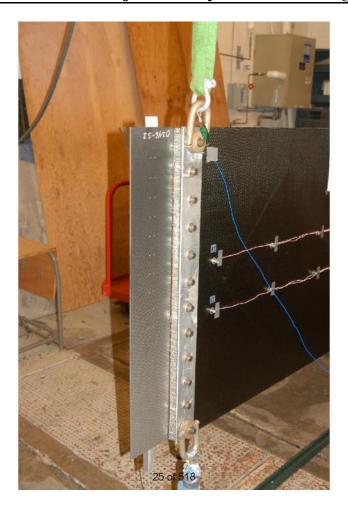
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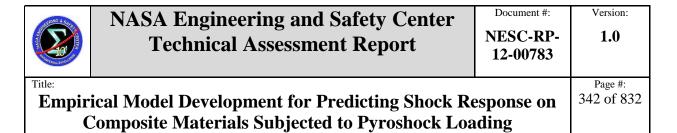
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# NESC-DEV-13-037 Composite Materials Shock Test

Test #2 Setup

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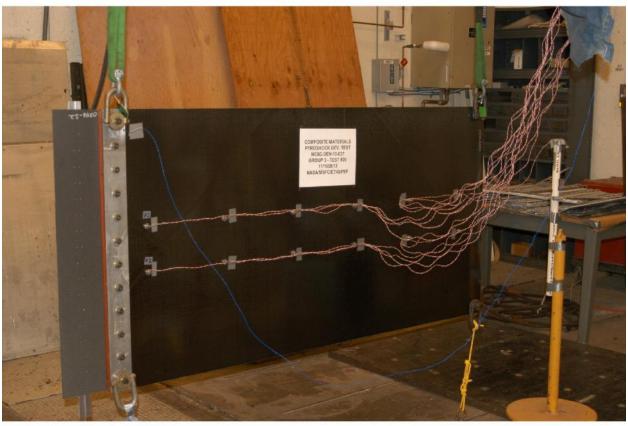
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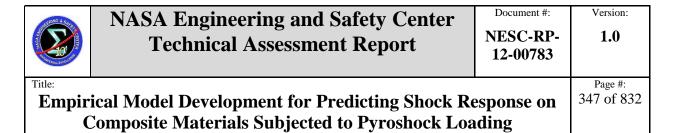
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# NESC-DEV-13-037 Composite Materials Shock Test

Test #3 Setup

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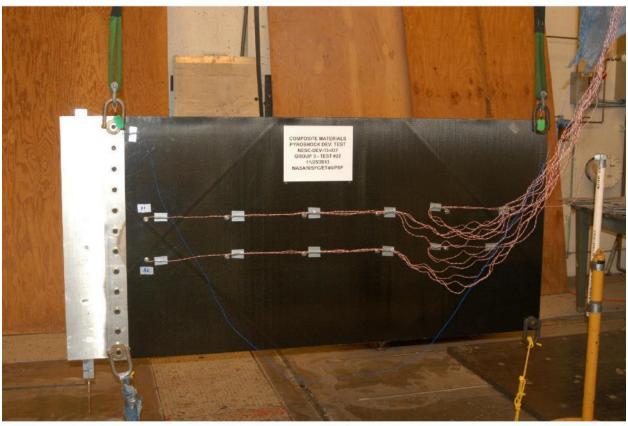
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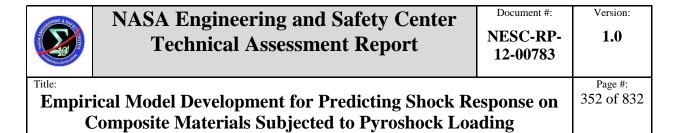
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# NESC-DEV-13-037 Composite Materials Shock Test

Test #4 Setup

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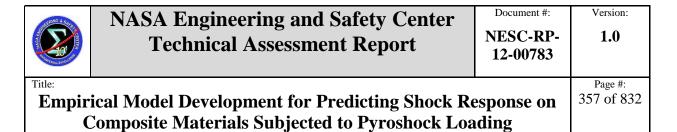
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Test #5 Setup



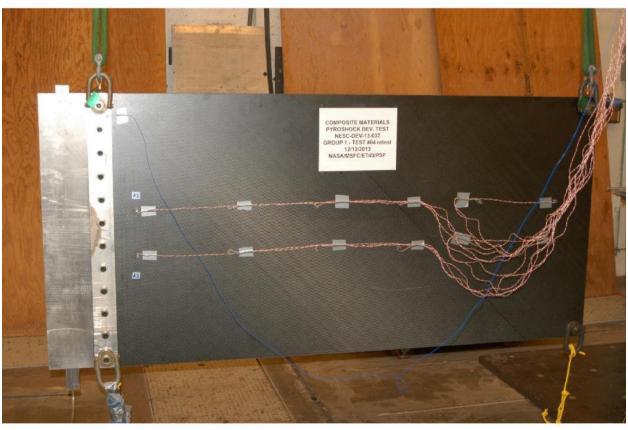
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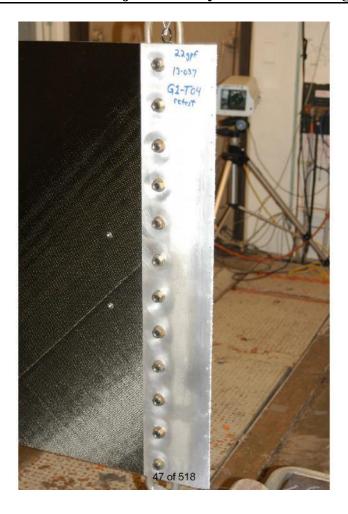
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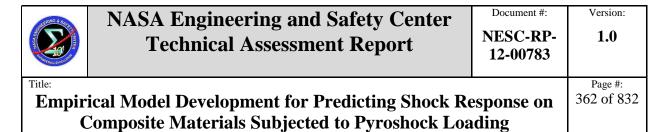
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Test #6 Setup



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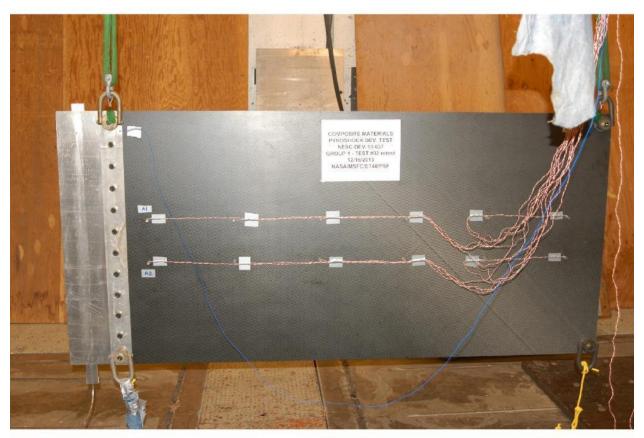
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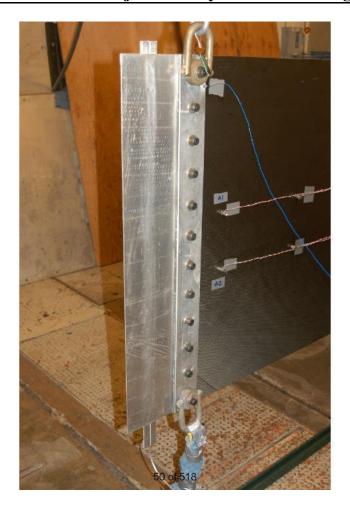
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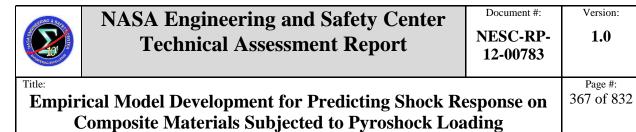
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# NESC-DEV-13-037 Composite Materials Shock Test

Test #7 Setup



Title:

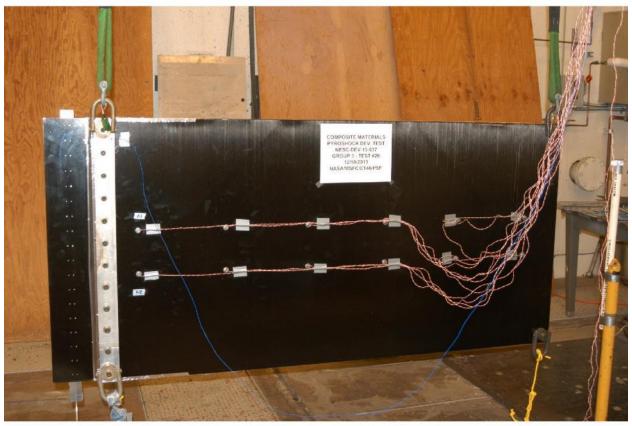
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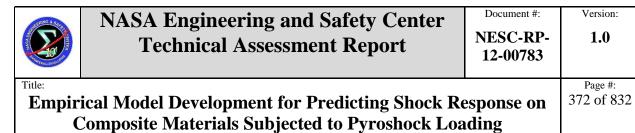
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Test #8 Setup



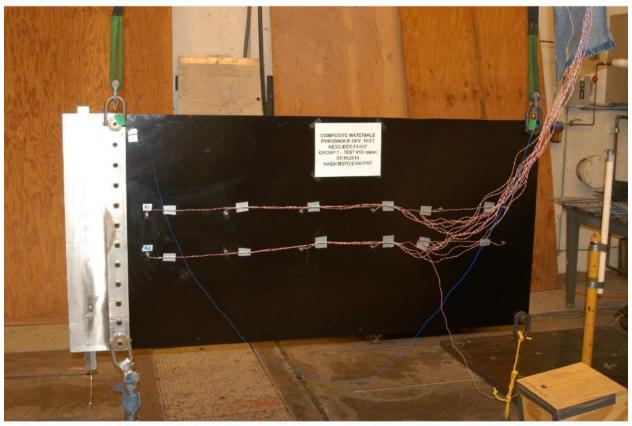
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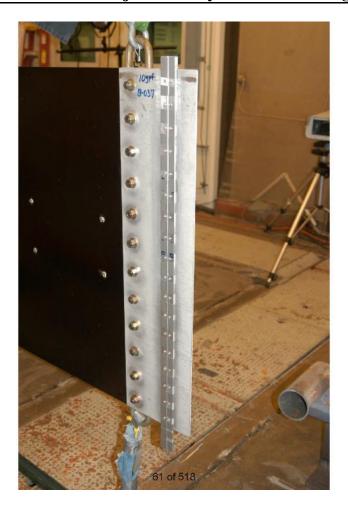


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# NESC-DEV-13-037 Composite Materials Shock Test

Test #9 Setup



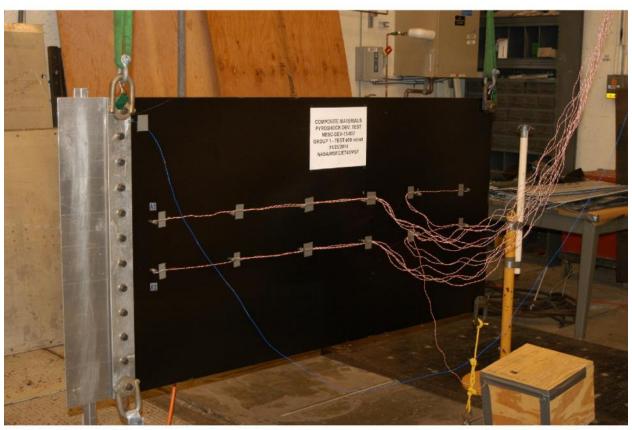
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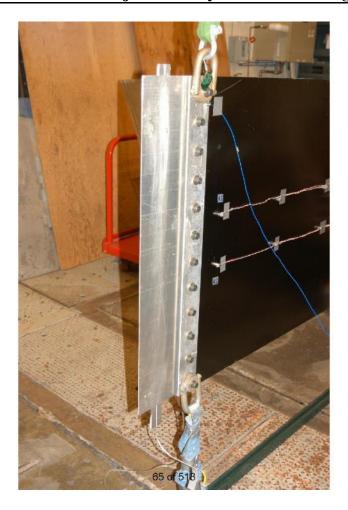
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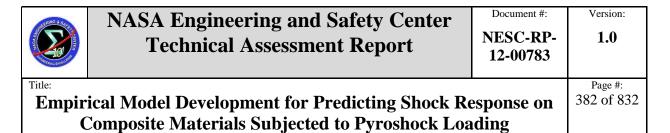
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# NESC-DEV-13-037 Composite Materials Shock Test

Test #10 Setup



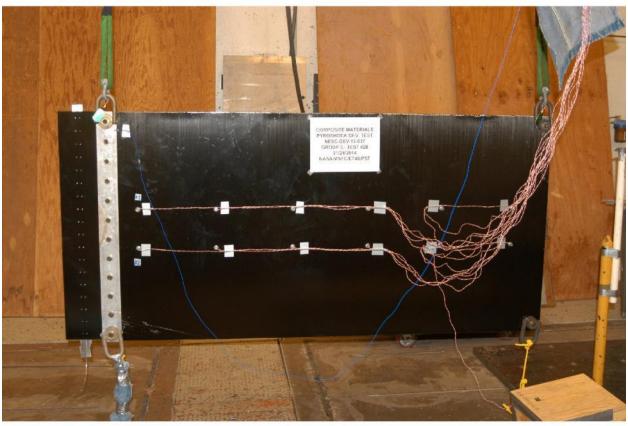
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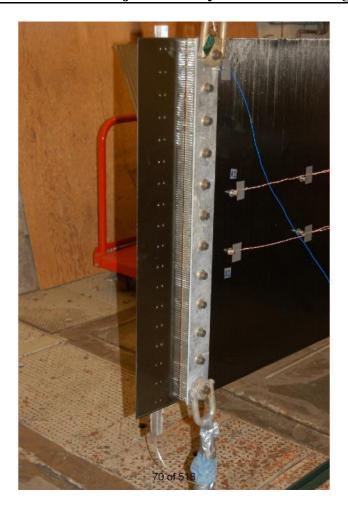
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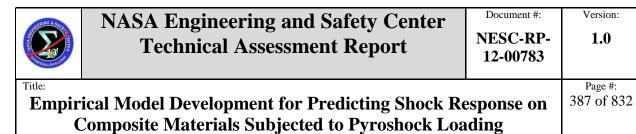
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Test #11 Setup



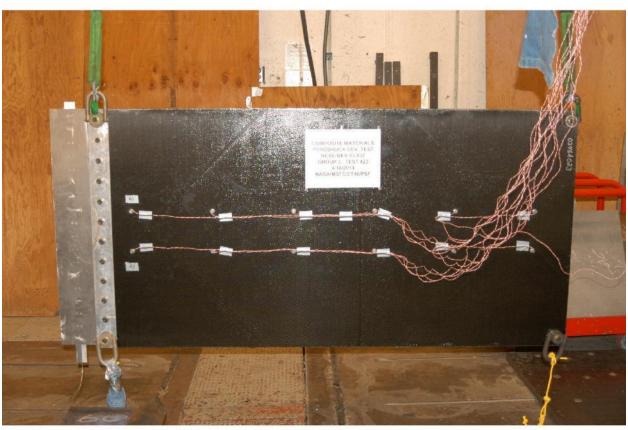
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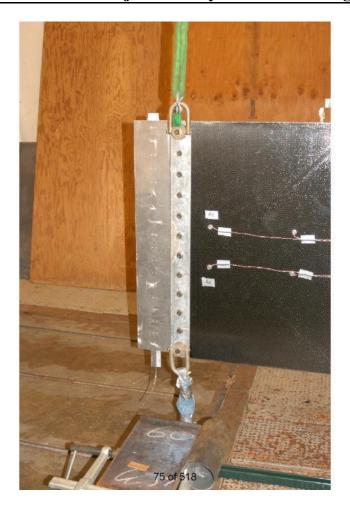


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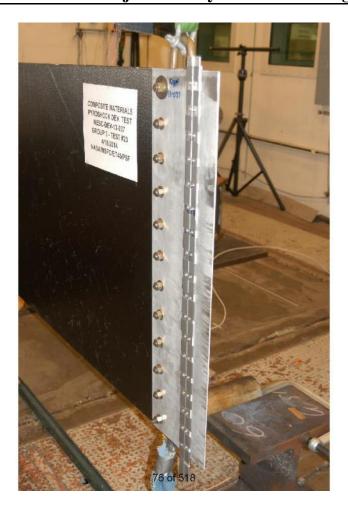
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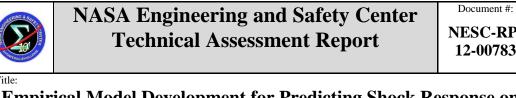
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# NESC-DEV-13-037 Composite Materials Shock Test

Test #12 Setup



Title:

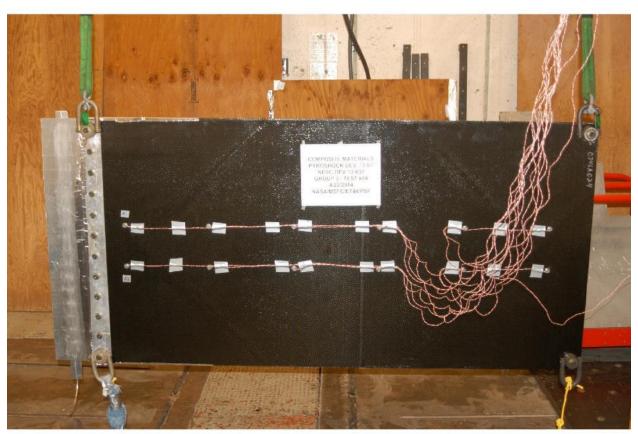
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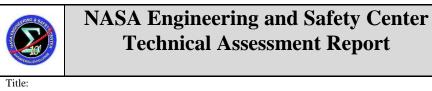
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Test #13 Setup



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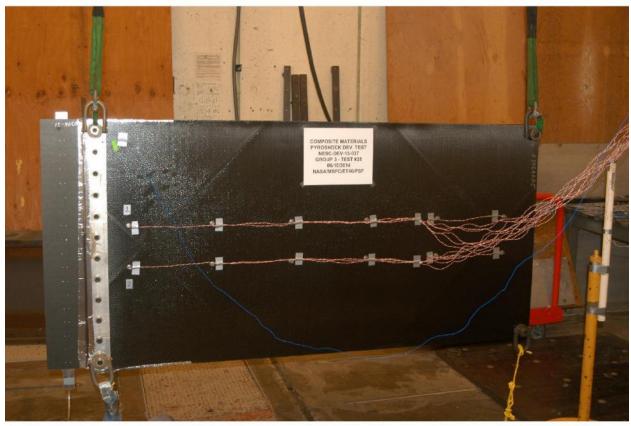
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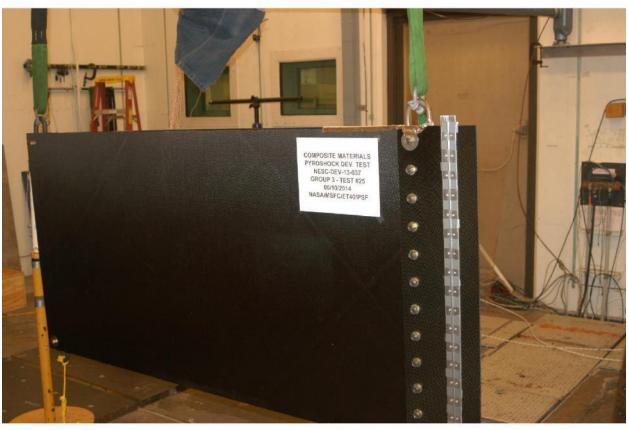
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### NESC-DEV-13-037 Composite Materials Shock Test

Test #14 Setup



Title:

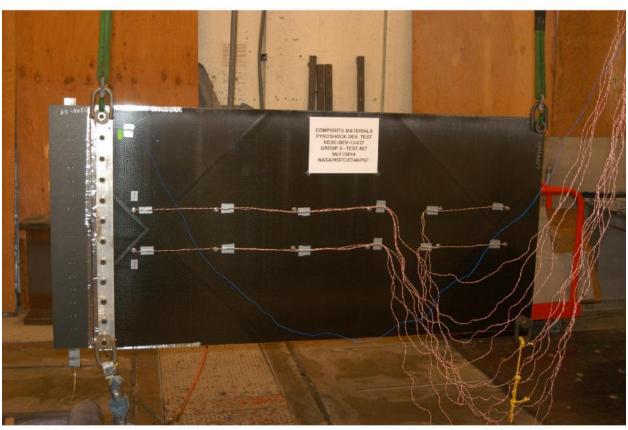
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### NESC-DEV-13-037 Composite Materials Shock Test

### Test #1 Accelerometer Data Panel 0346A019



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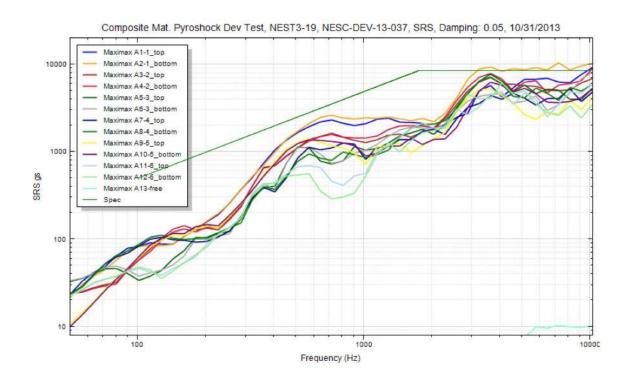
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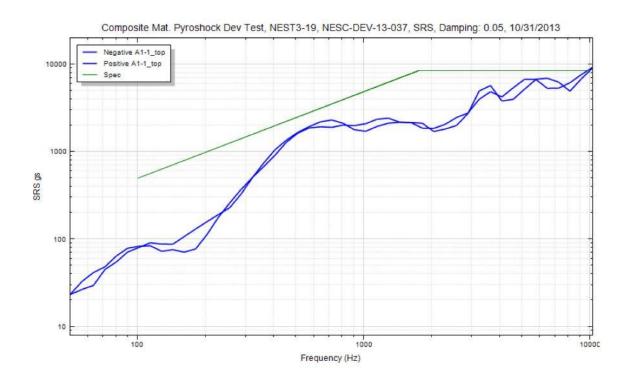
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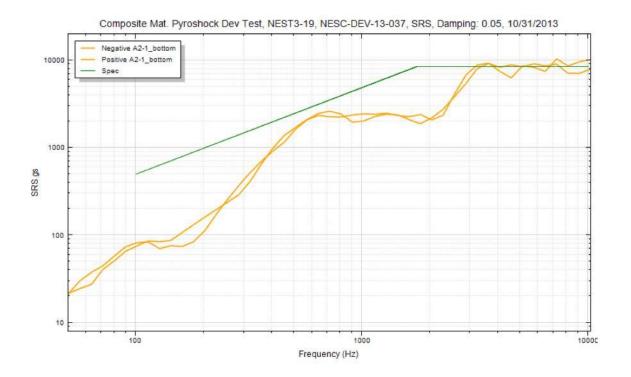
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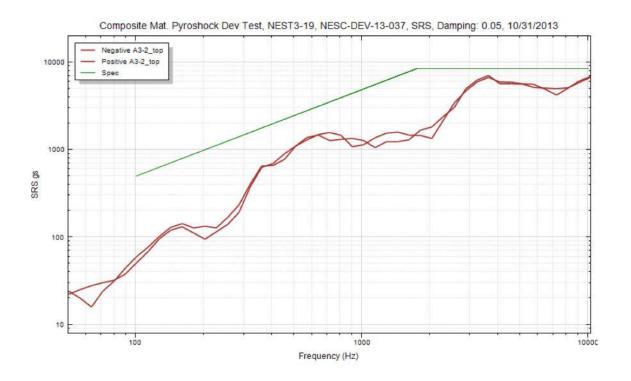
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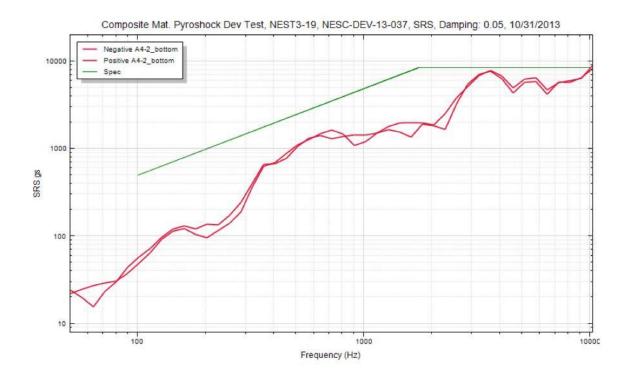
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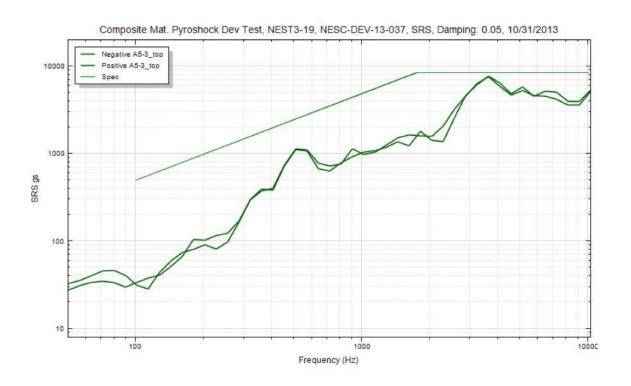
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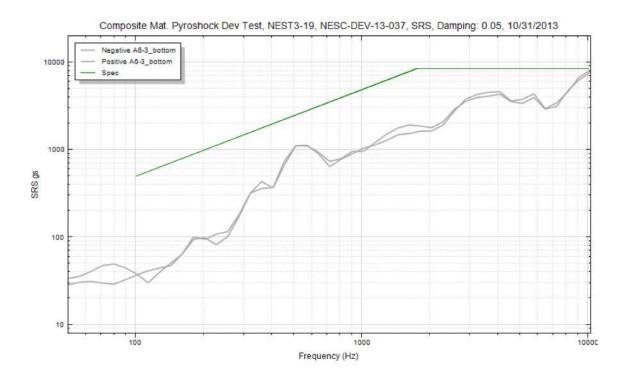
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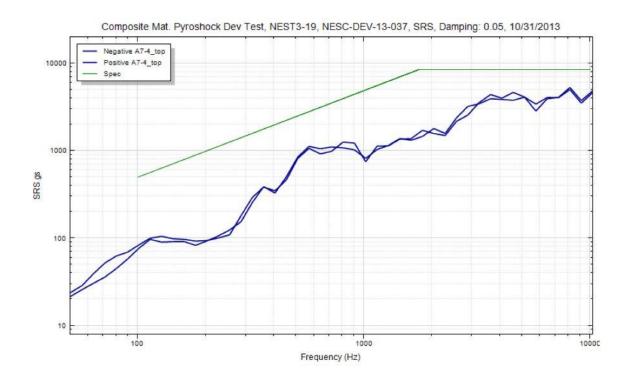
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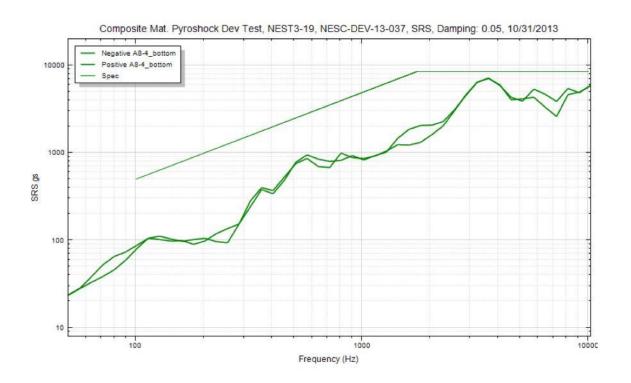
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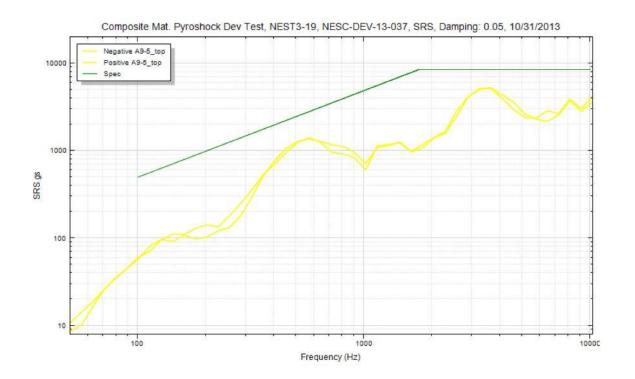
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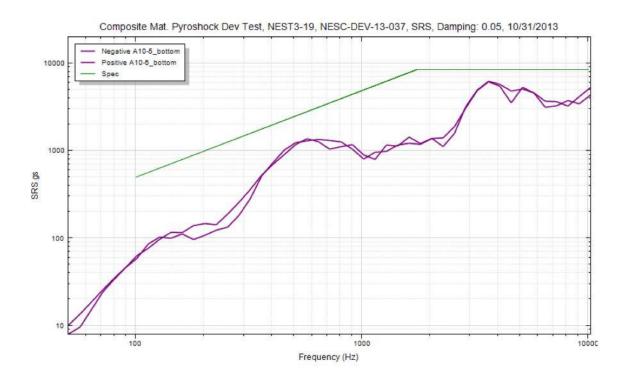
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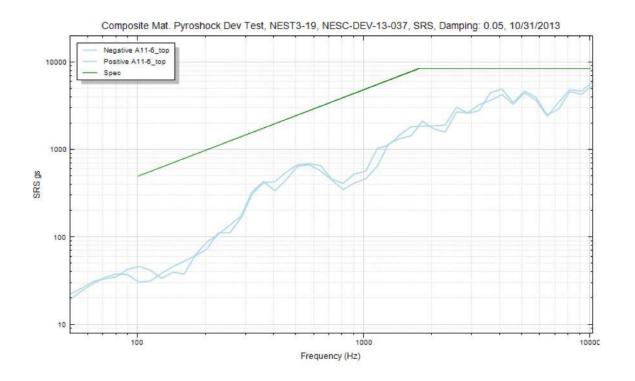
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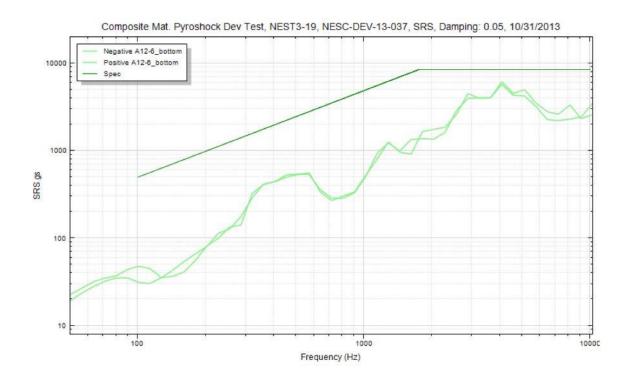
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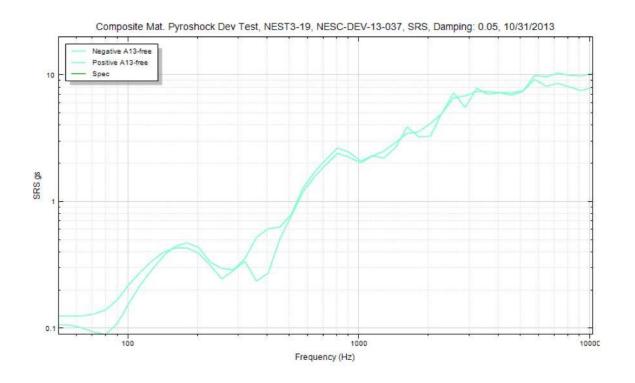
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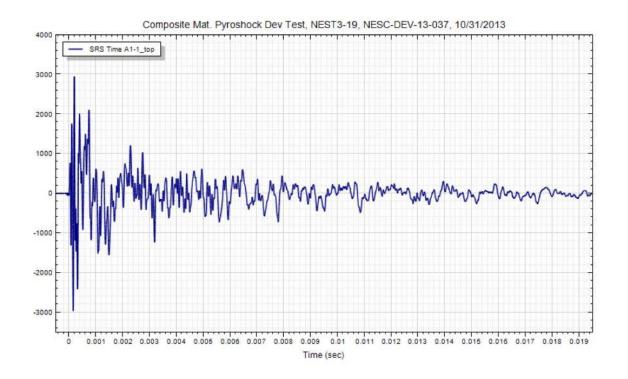
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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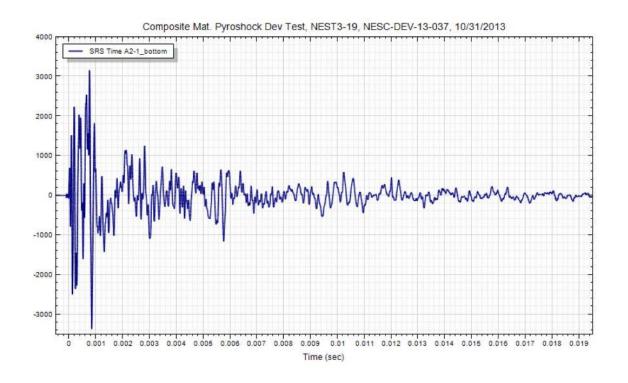
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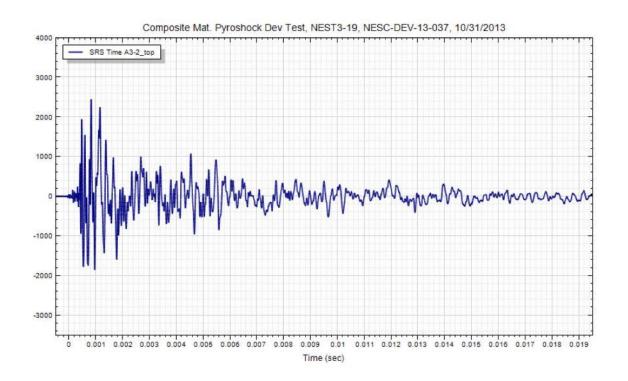
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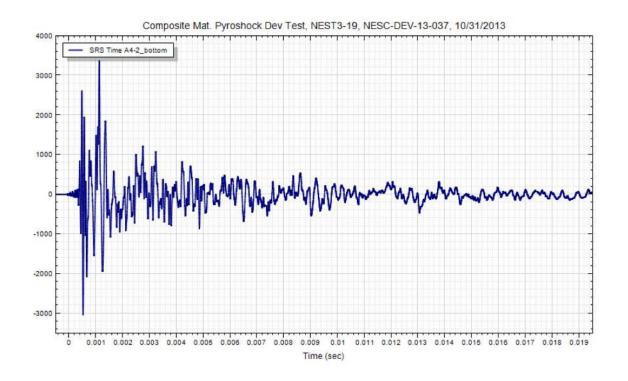
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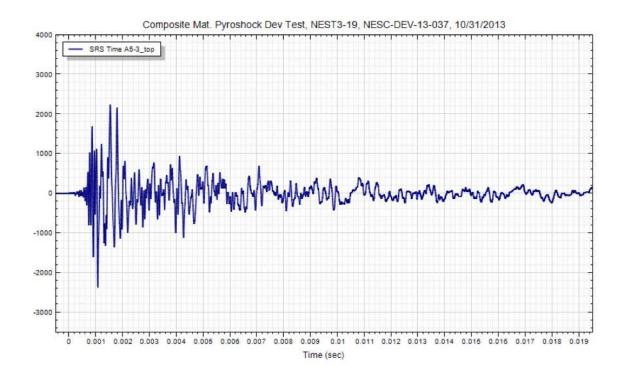
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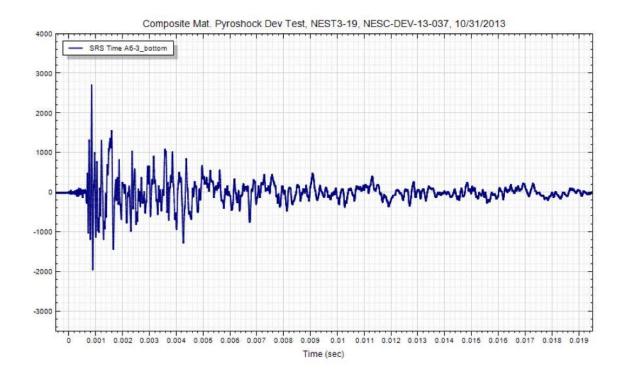
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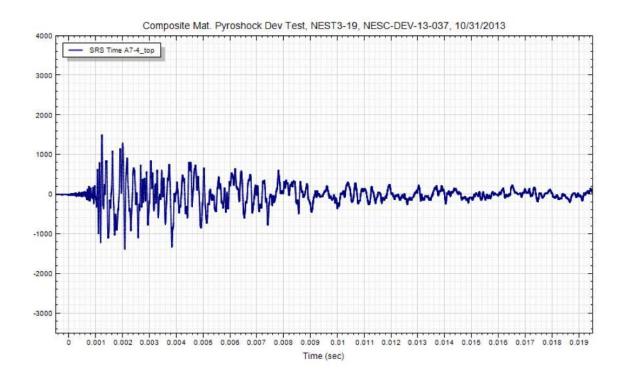
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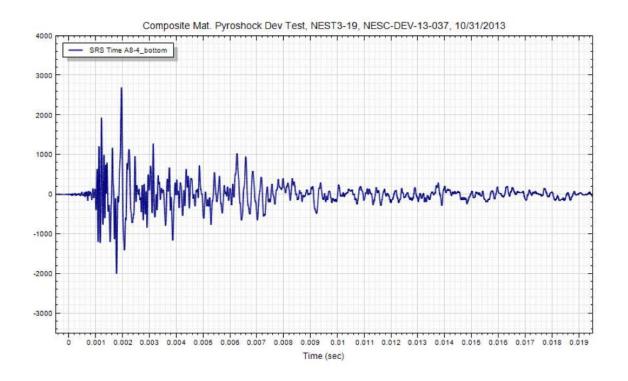
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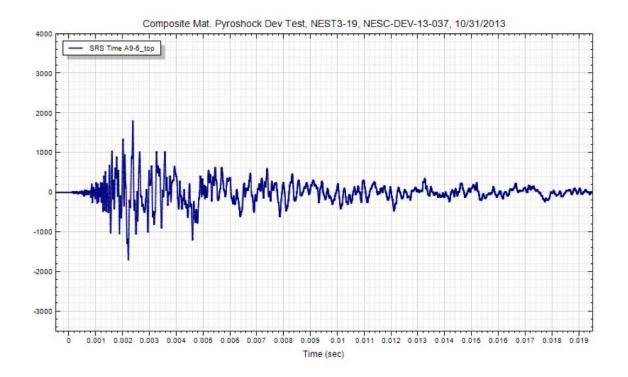
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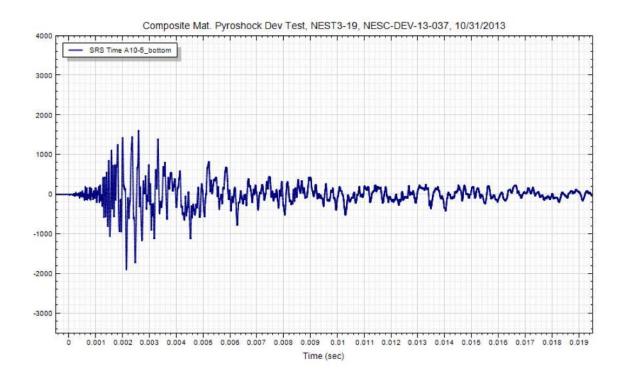
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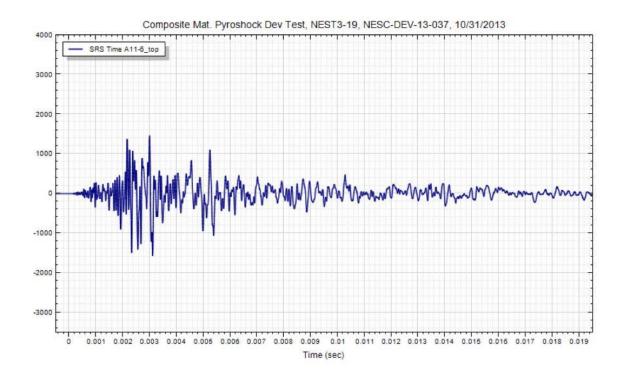
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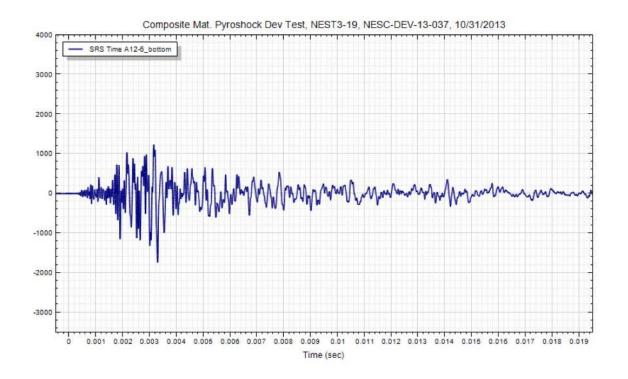
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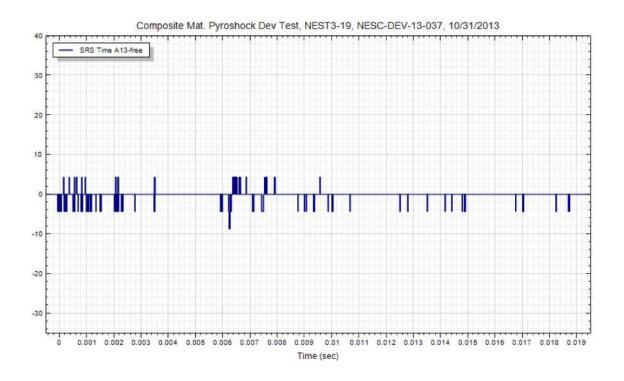
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #2 Accelerometer Data Panel 0346A020



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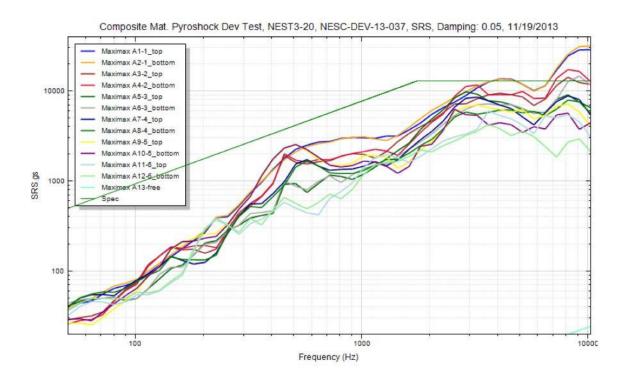
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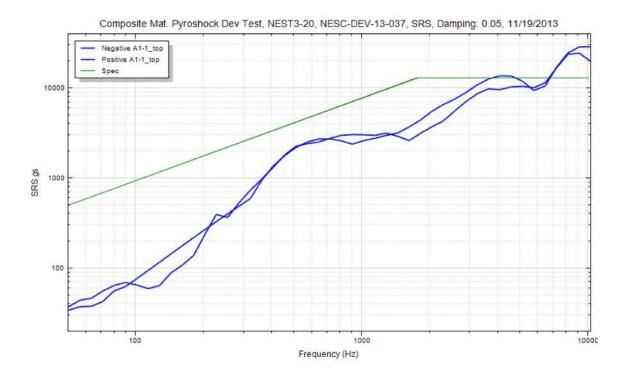
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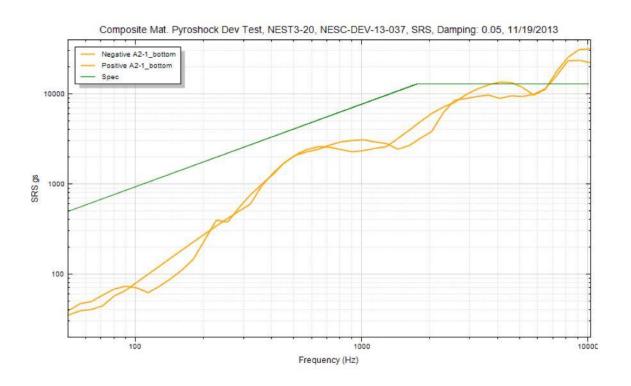
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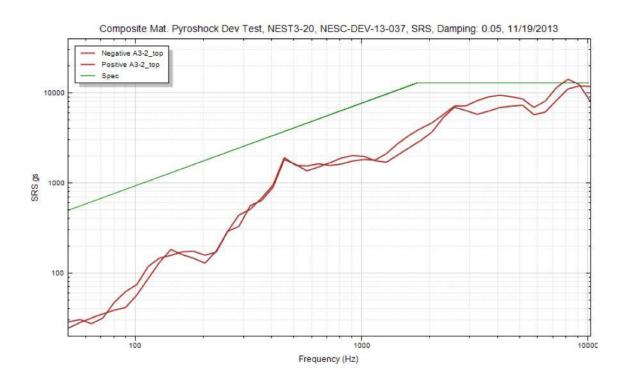
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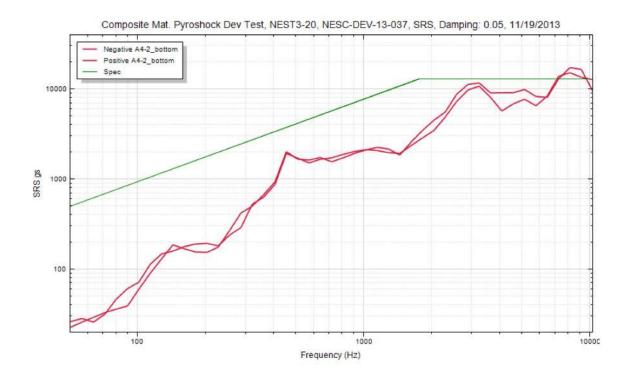
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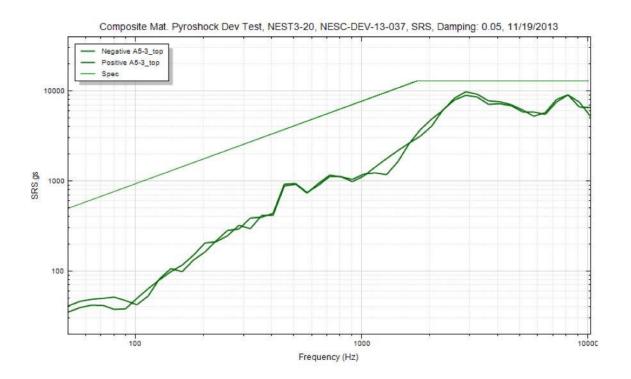
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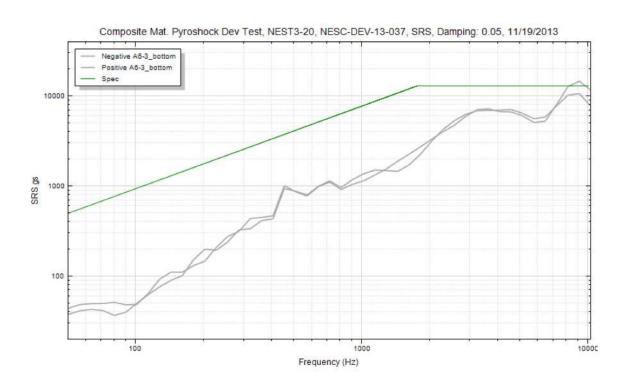
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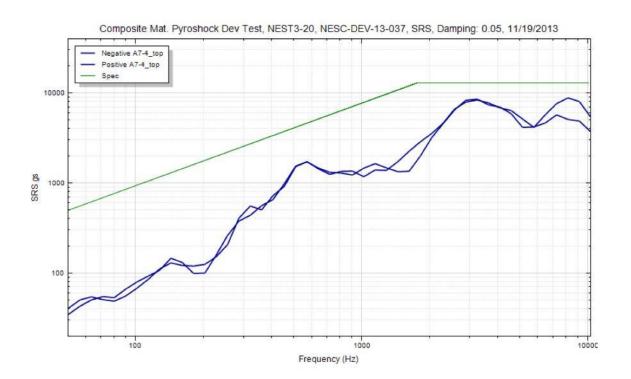
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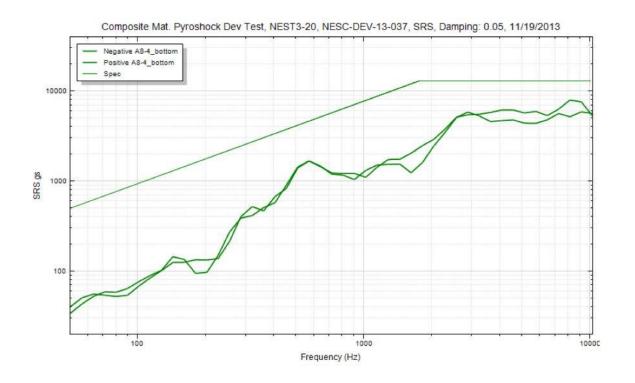
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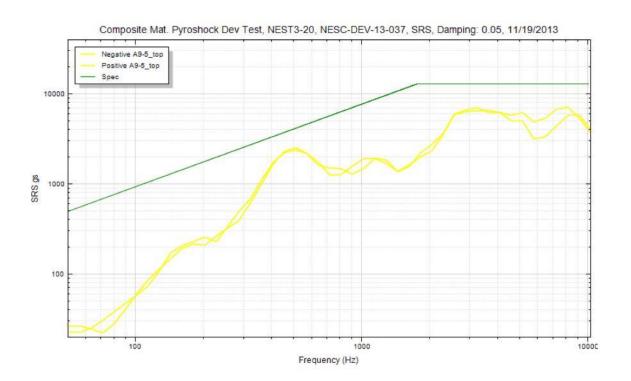
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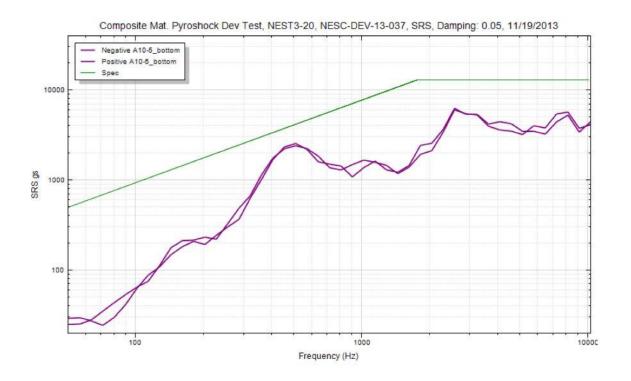
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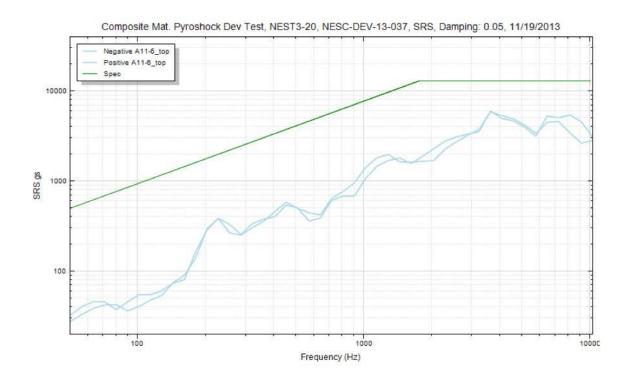
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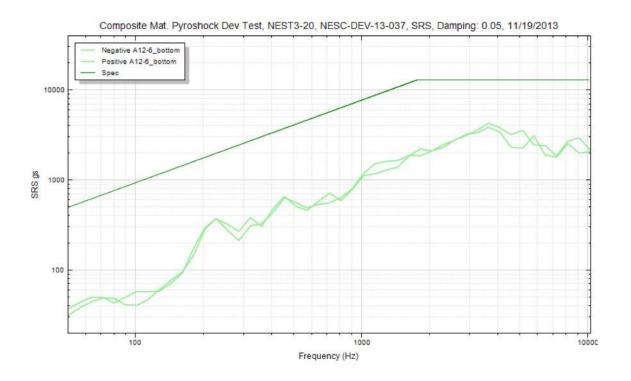
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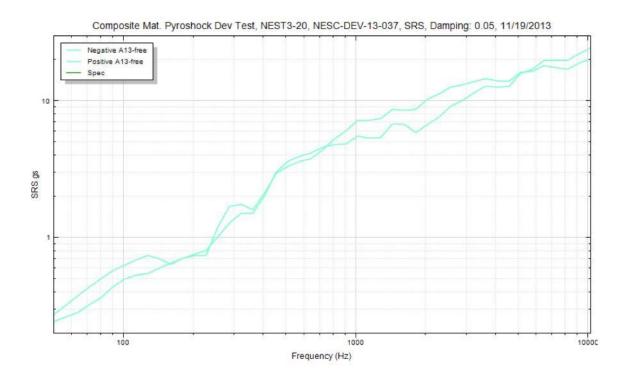
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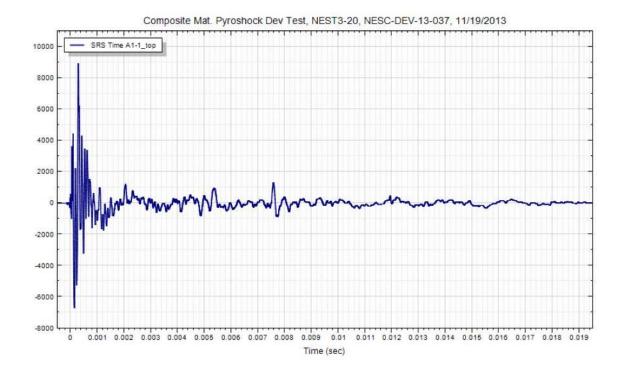
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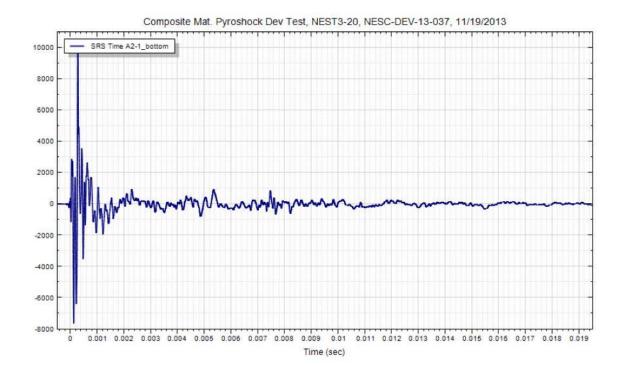
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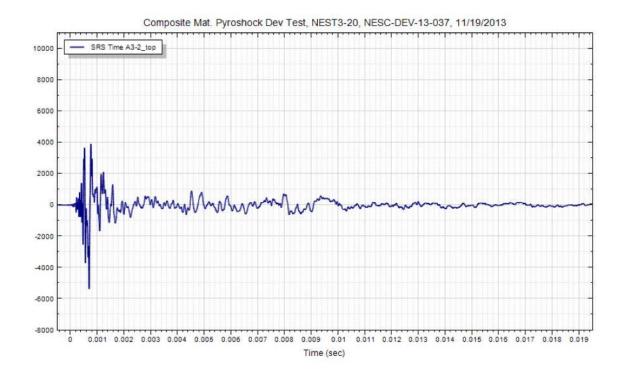
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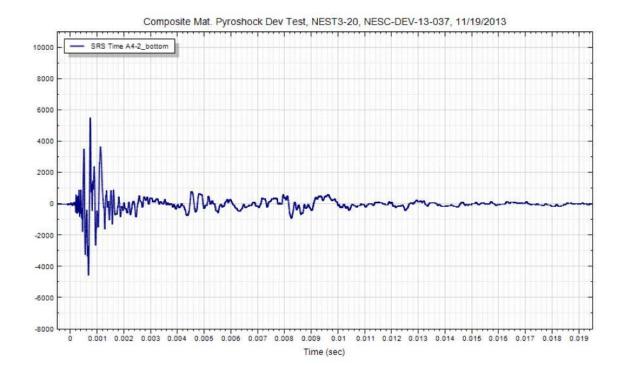
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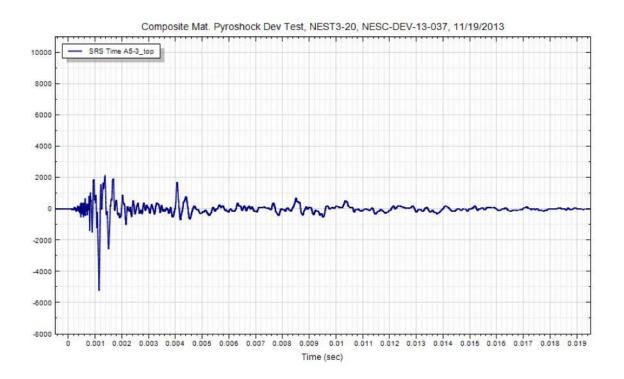
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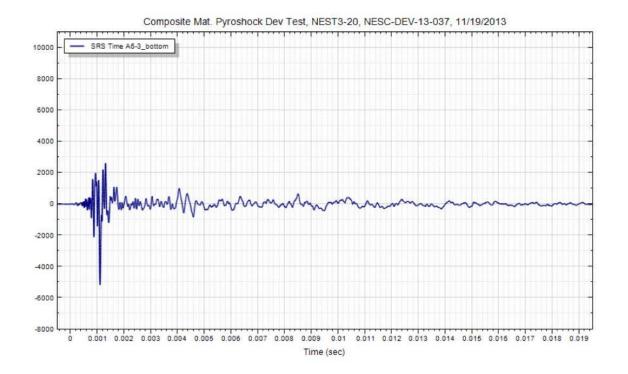
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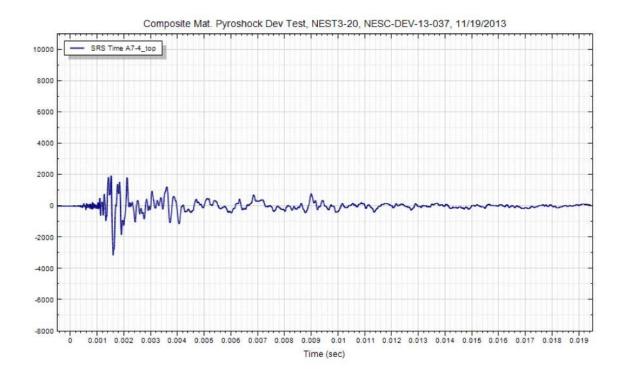
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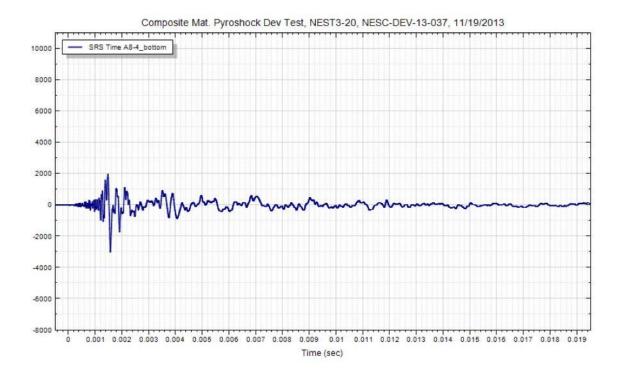
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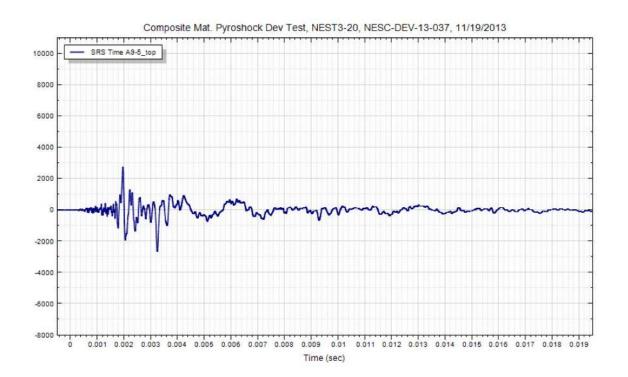
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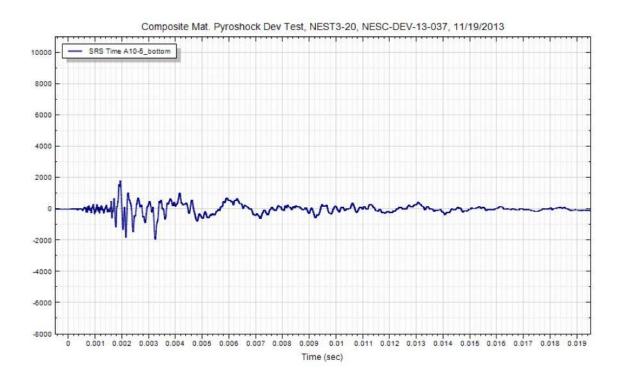
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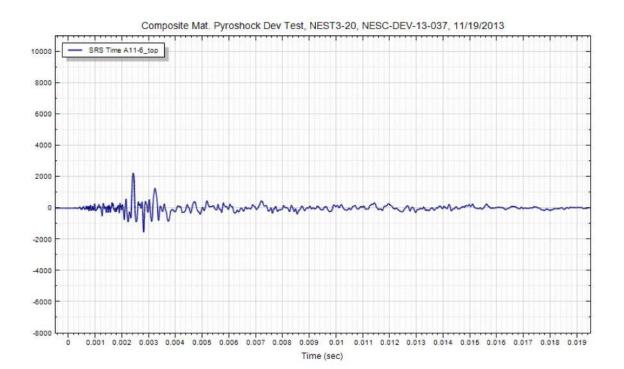
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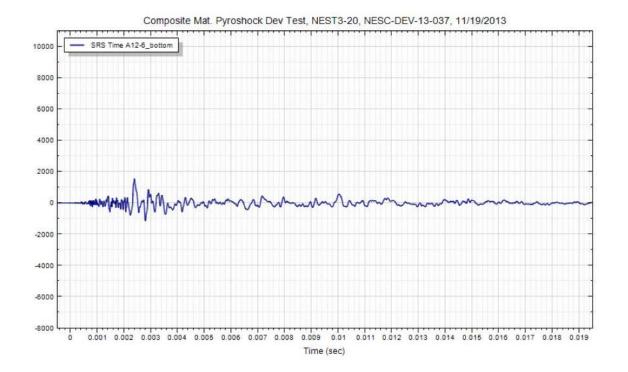
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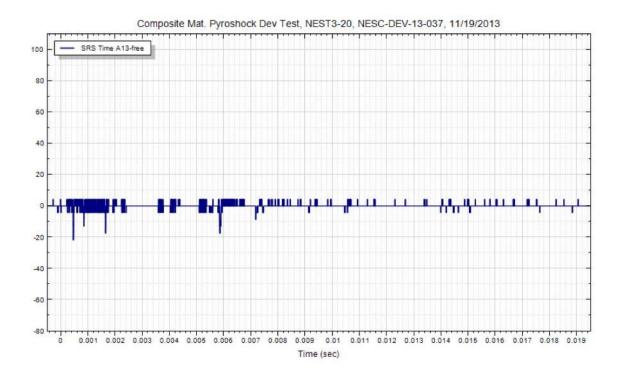
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### NESC-DEV-13-037 Composite Materials Shock Test

### Test #3 Accelerometer Data Panel 0346A022



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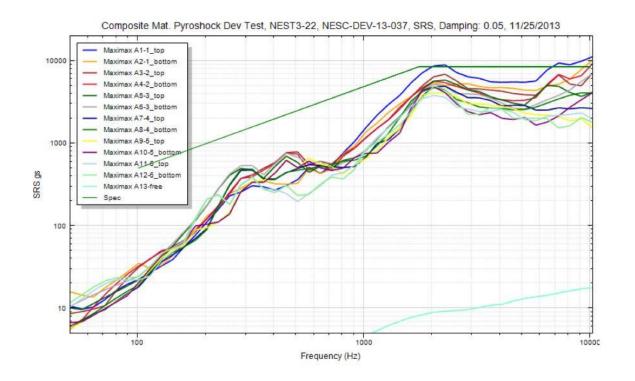
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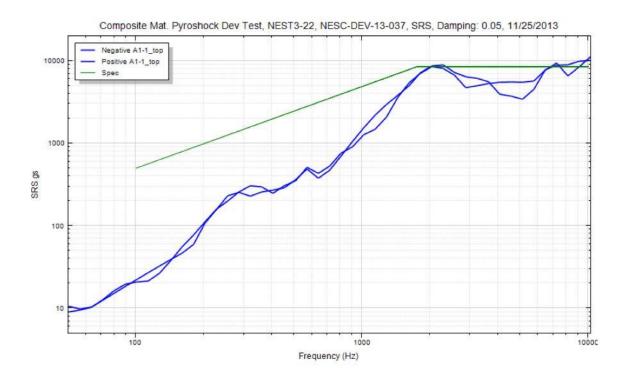
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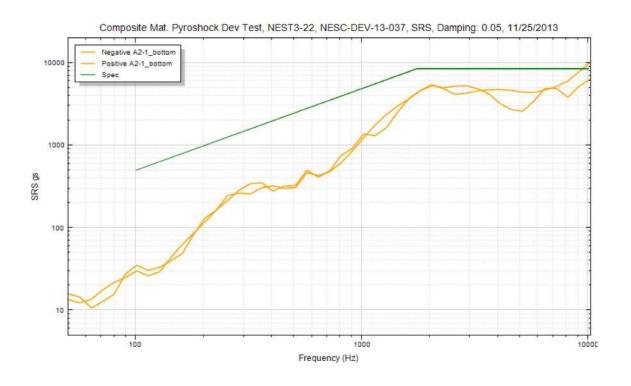
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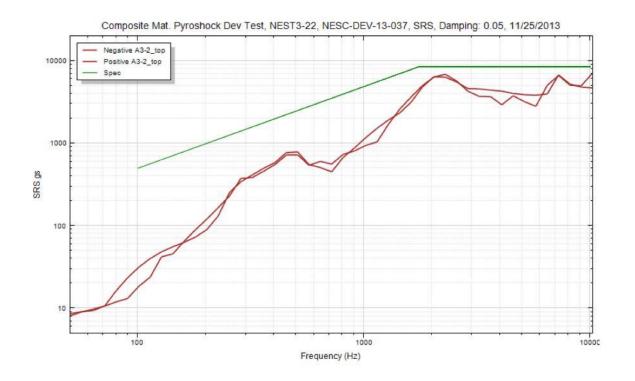
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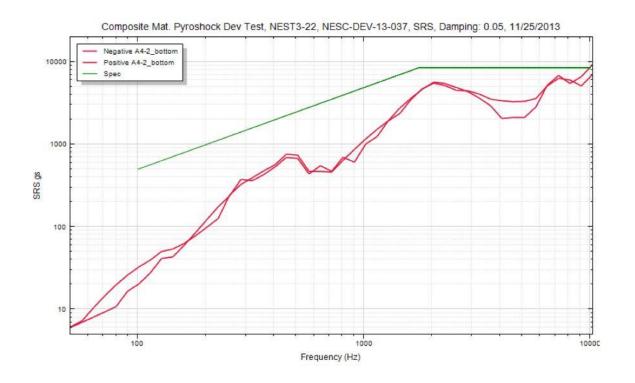
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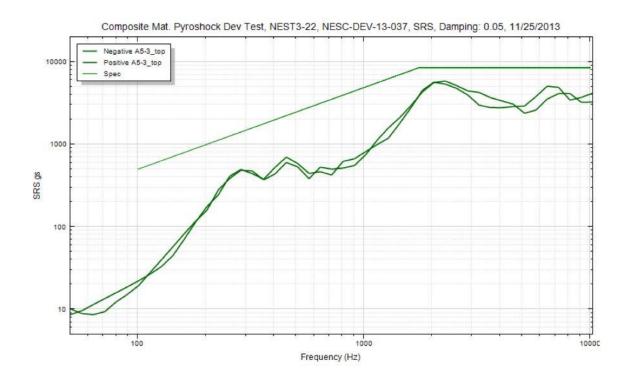
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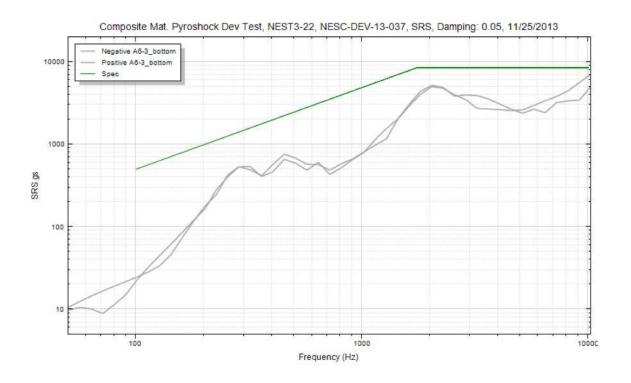
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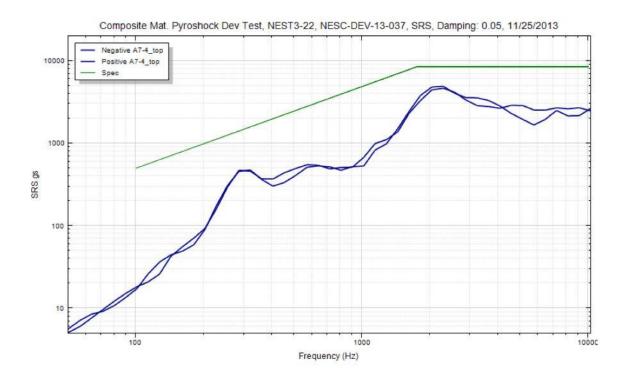
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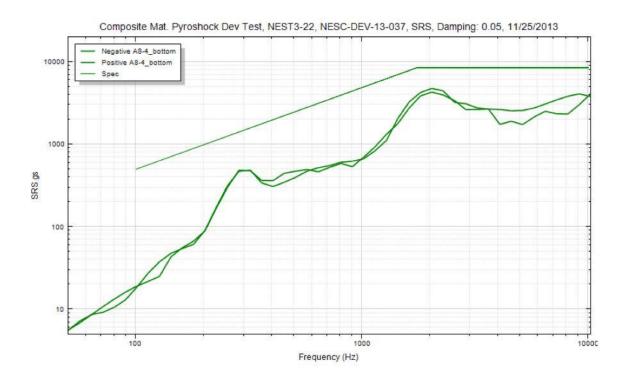
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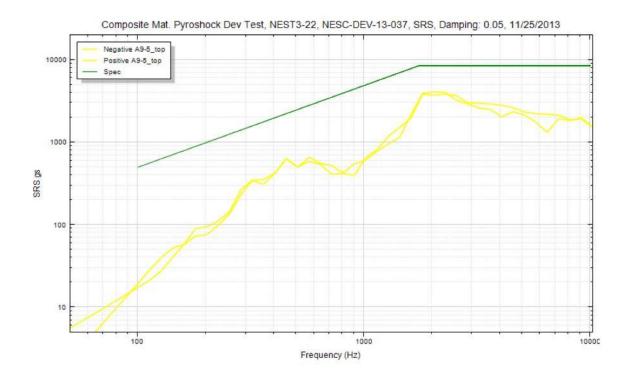
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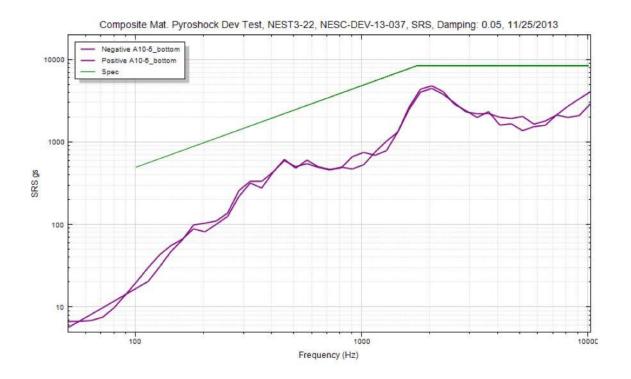
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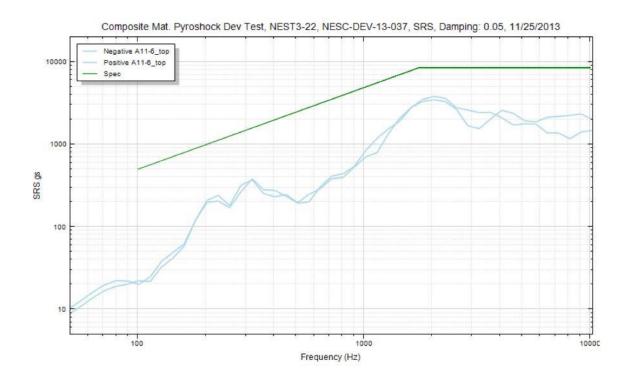
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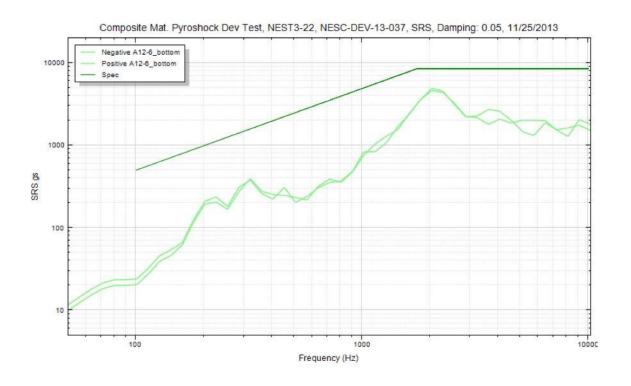
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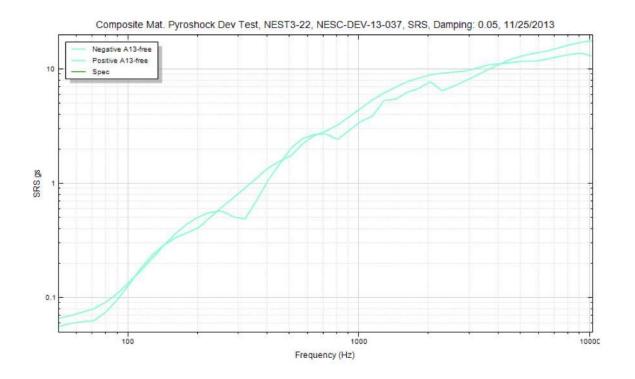
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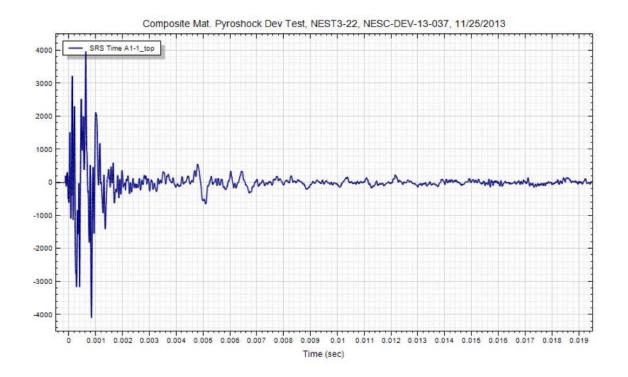
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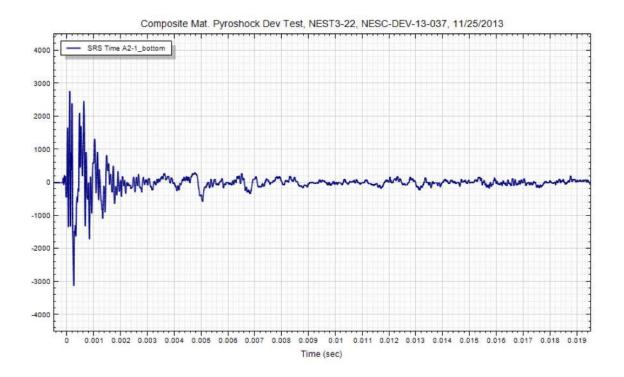
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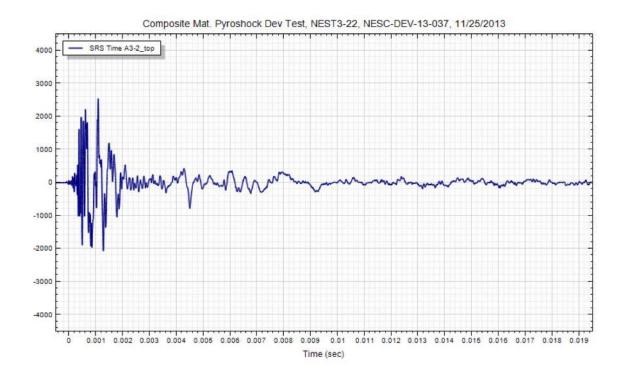
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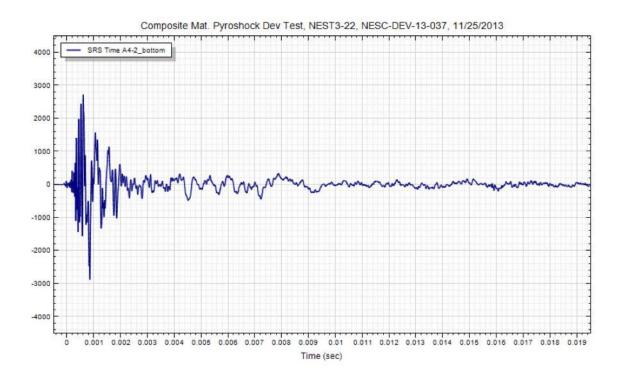
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





NESC-RP-12-00783

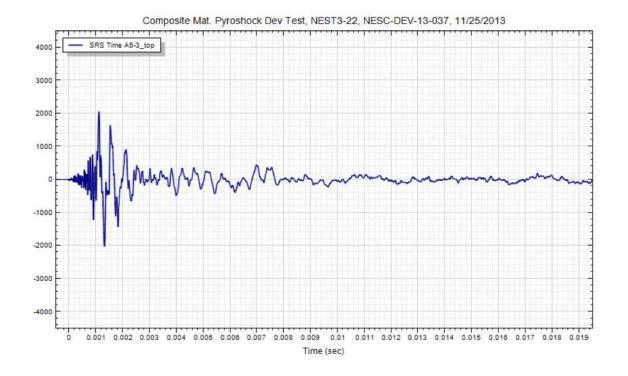
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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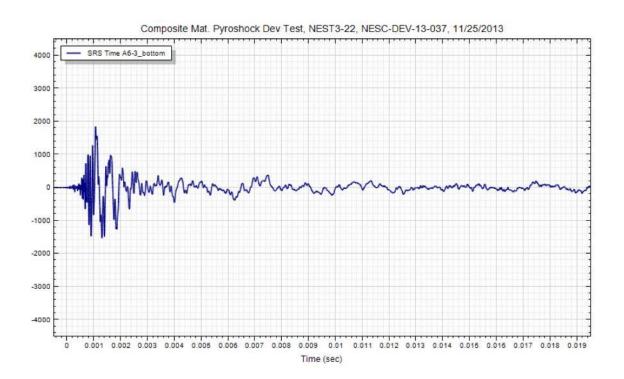
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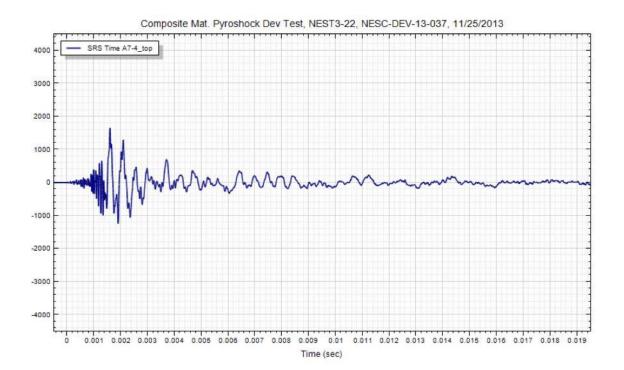
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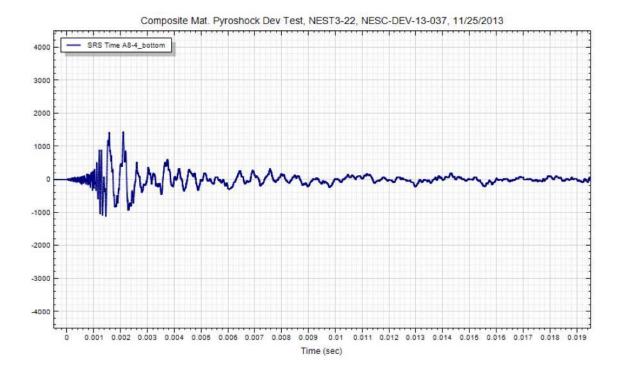
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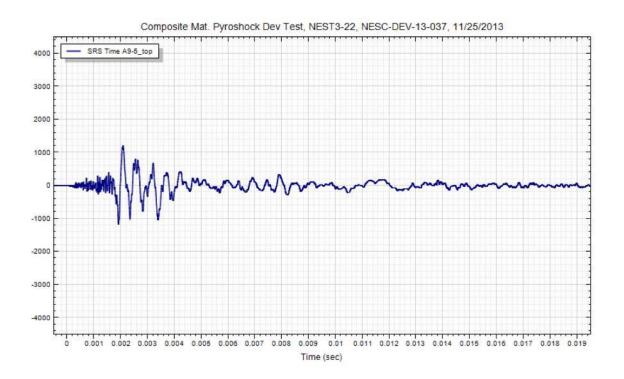
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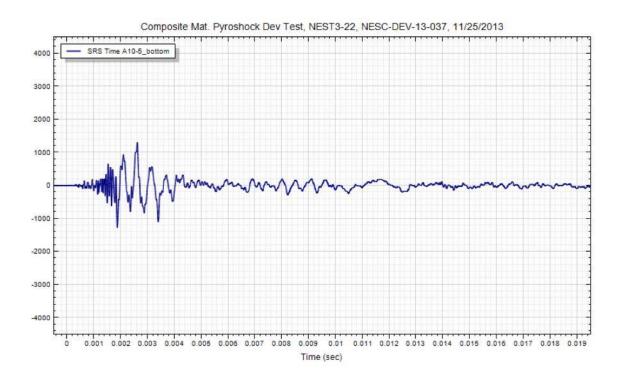
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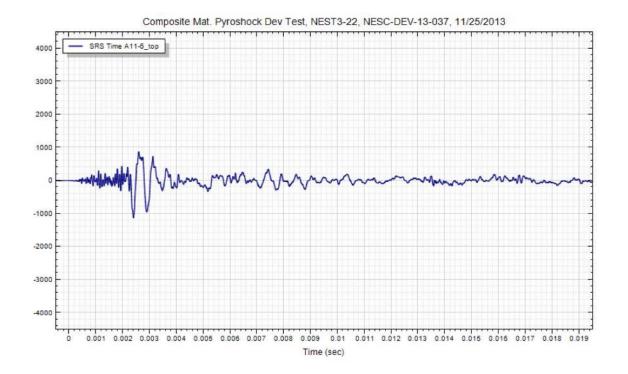
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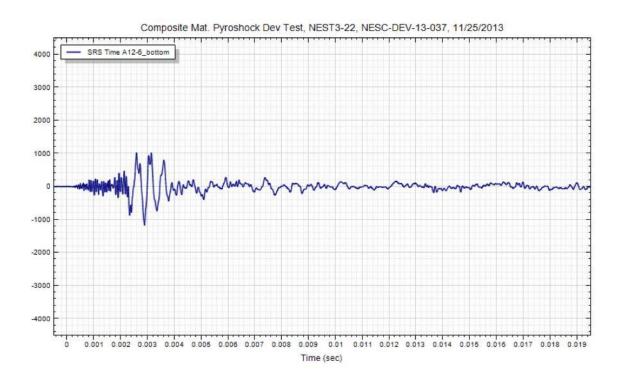
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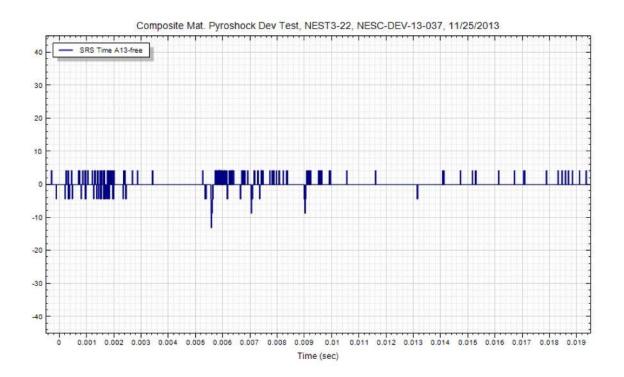
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### NESC-DEV-13-037 Composite Materials Shock Test

### Test #4 Accelerometer Data Panel 0346A021



NESC-RP-12-00783

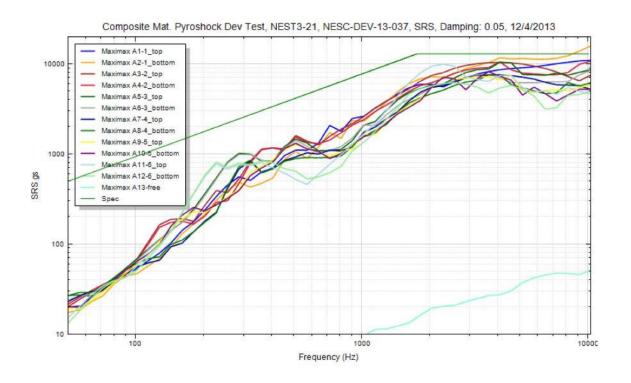
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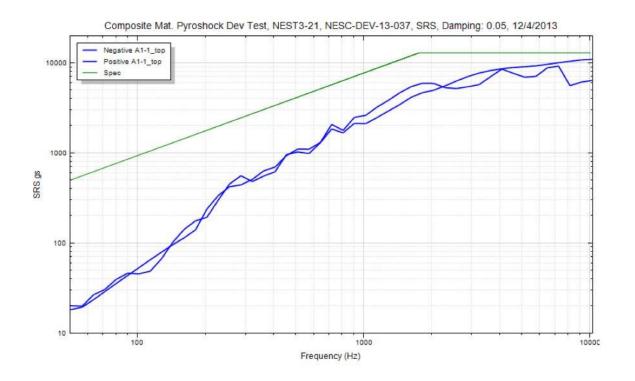
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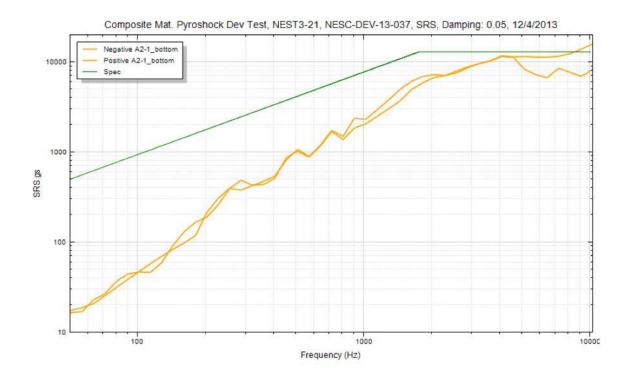
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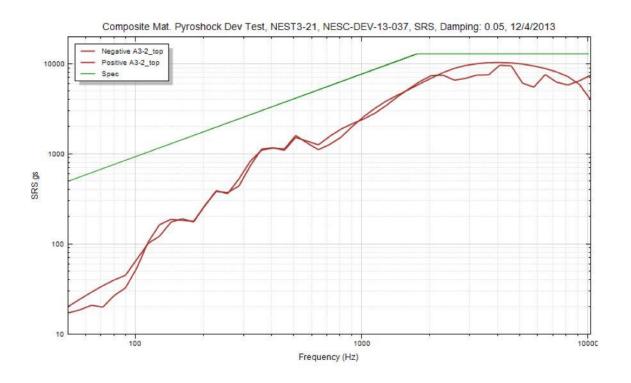
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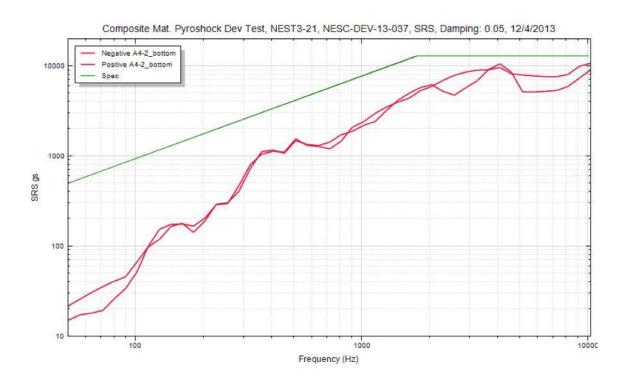
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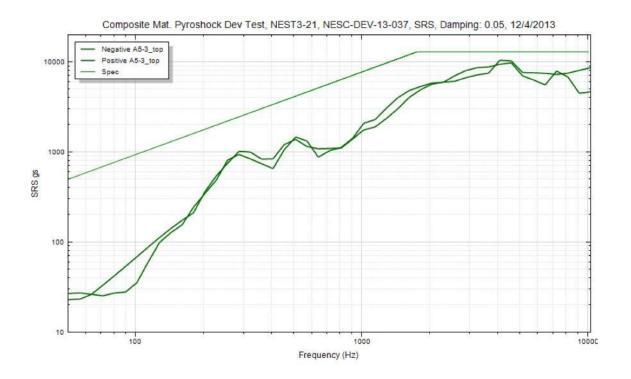
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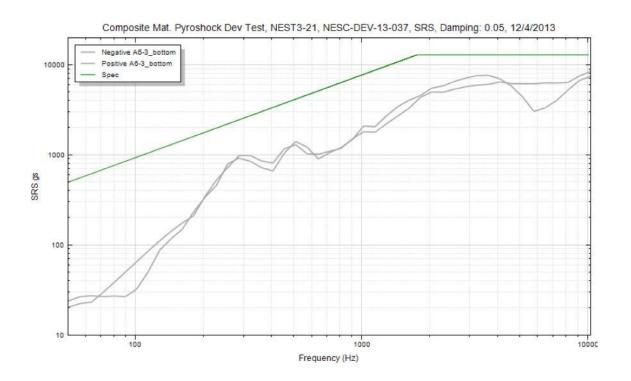
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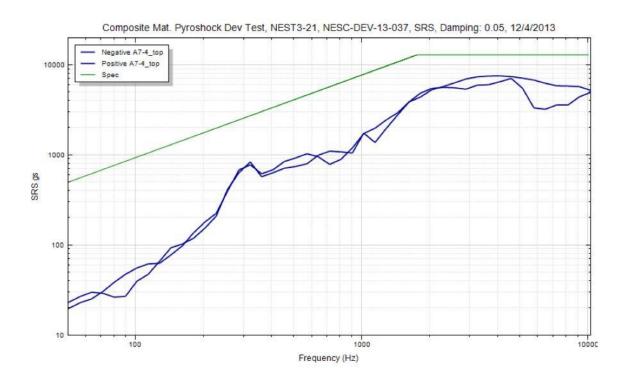
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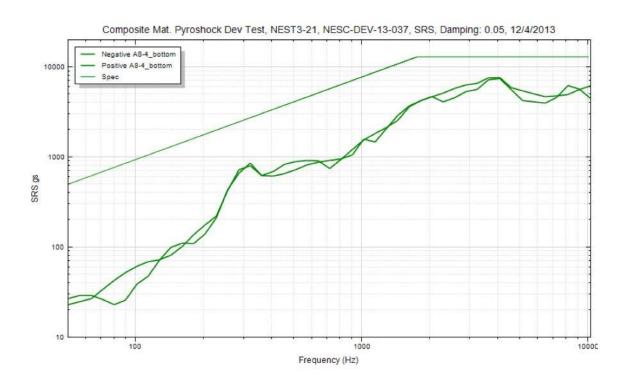
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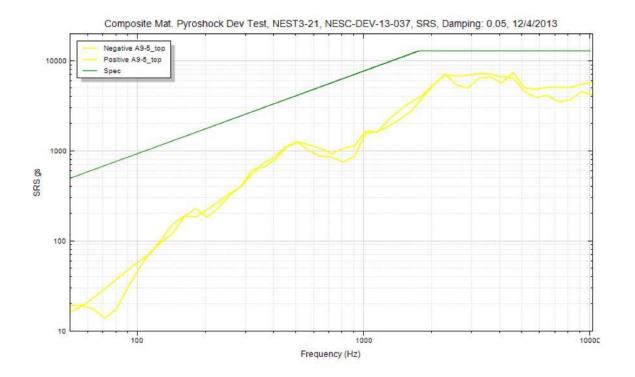
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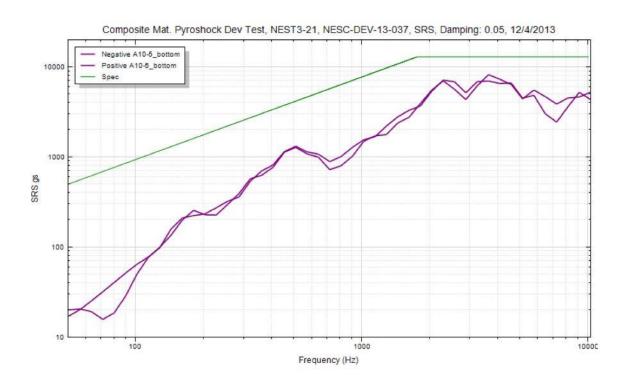
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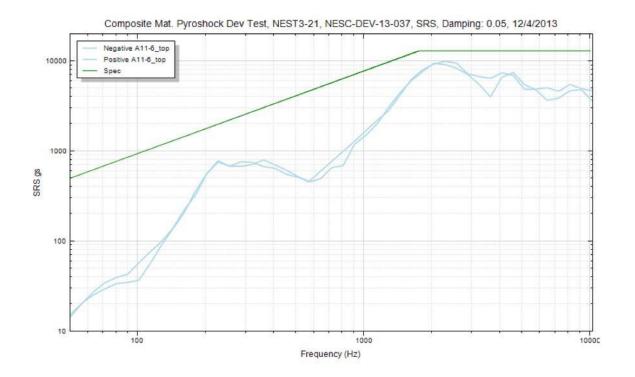
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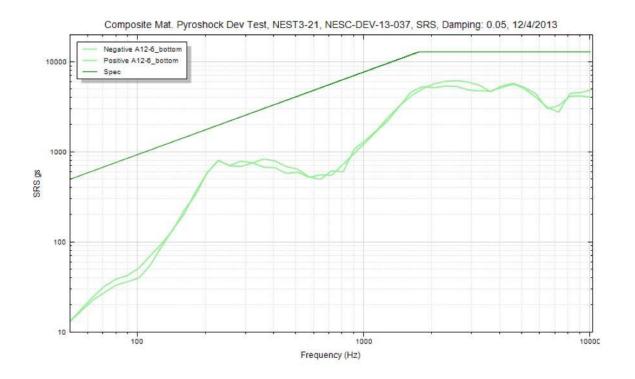
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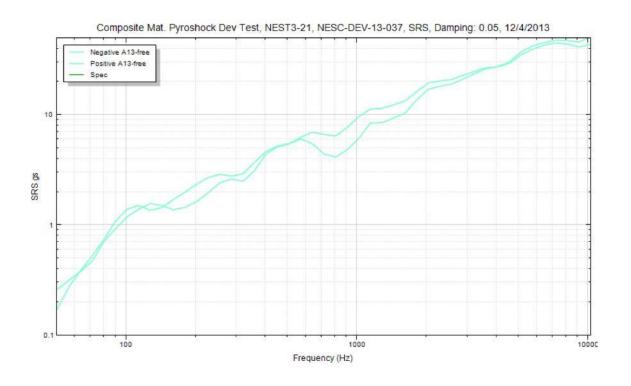
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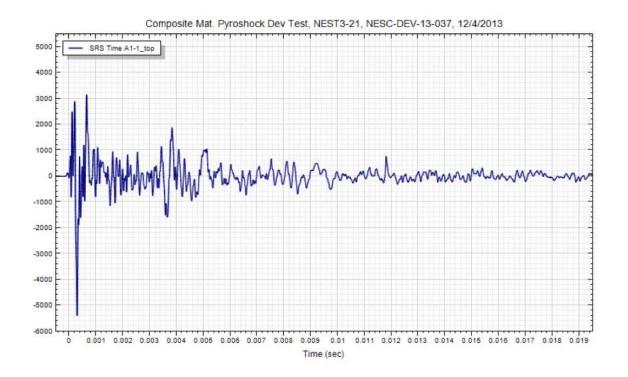
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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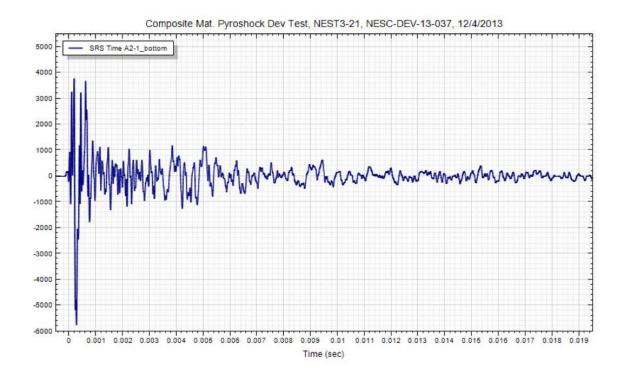
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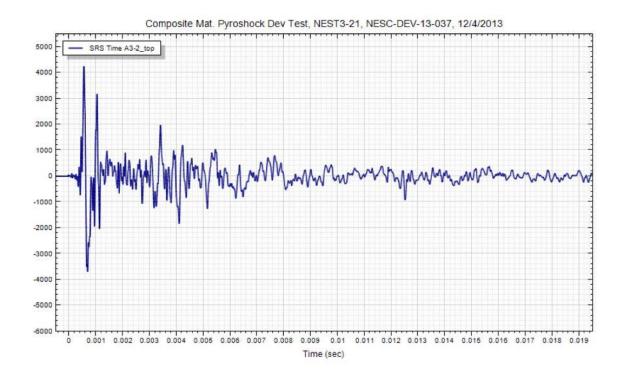
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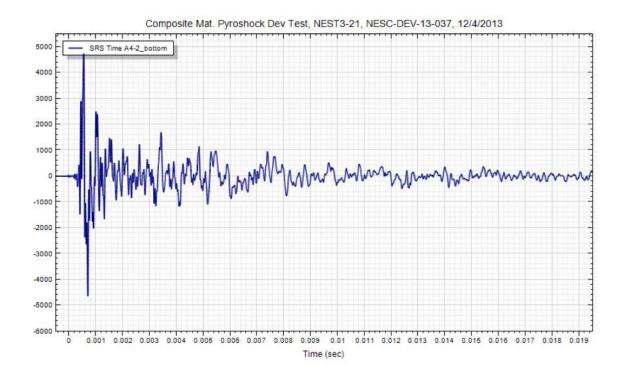
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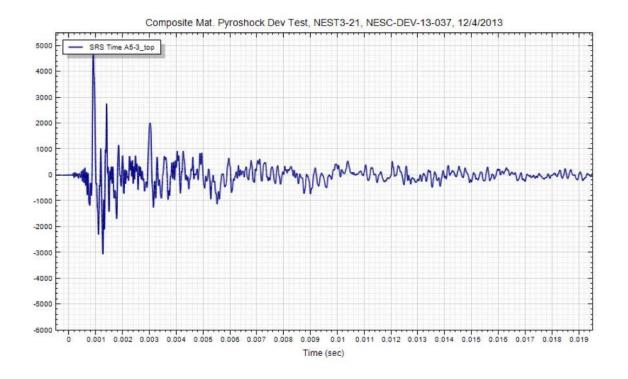
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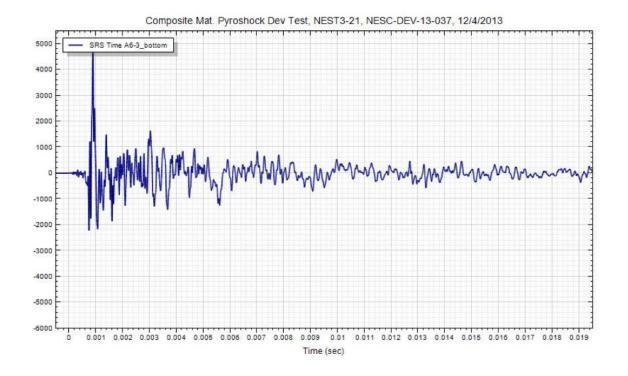
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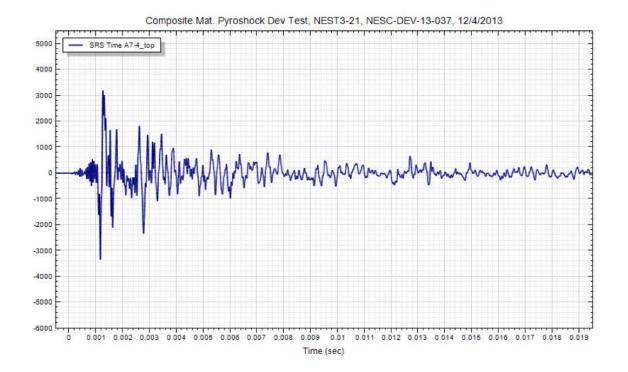
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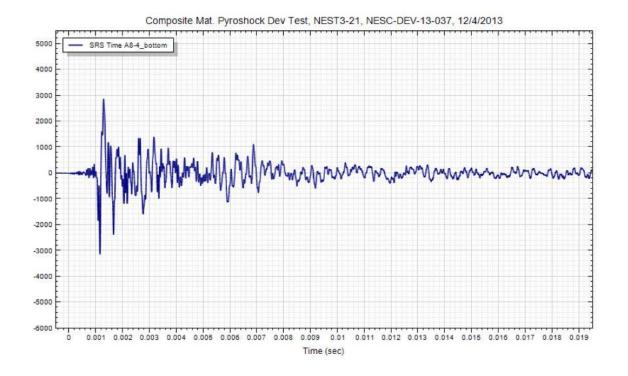
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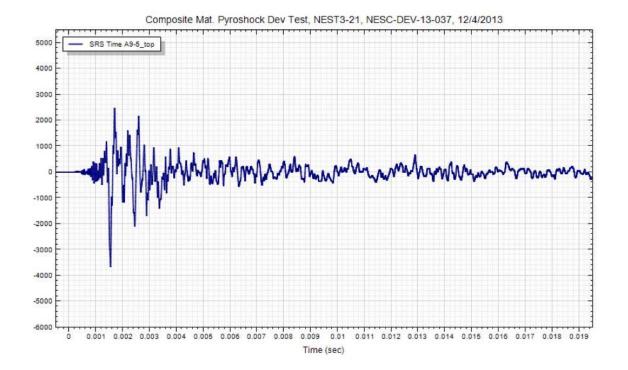
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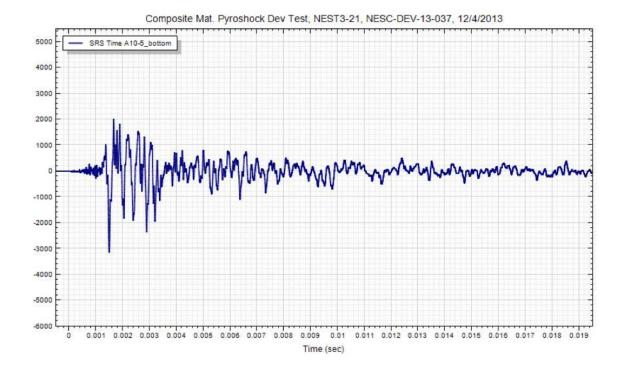
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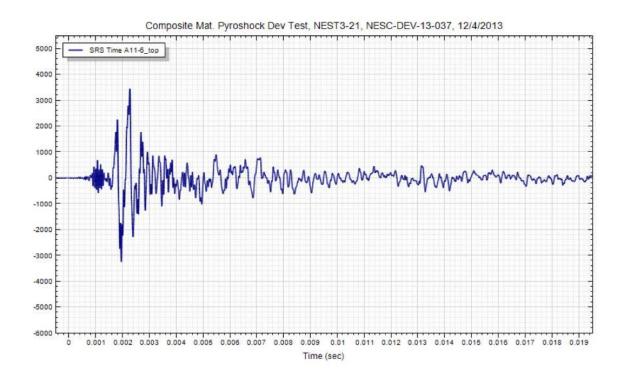
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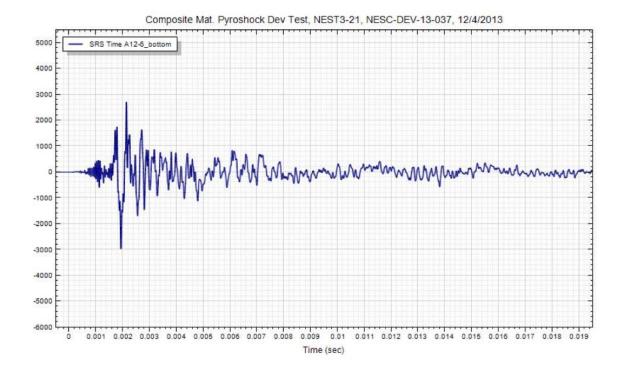
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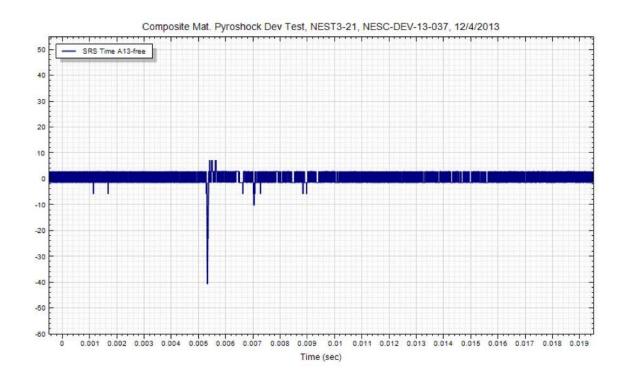
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### NESC-DEV-13-037 Composite Materials Shock Test

### Test #5 Accelerometer Data Panel 0320A004



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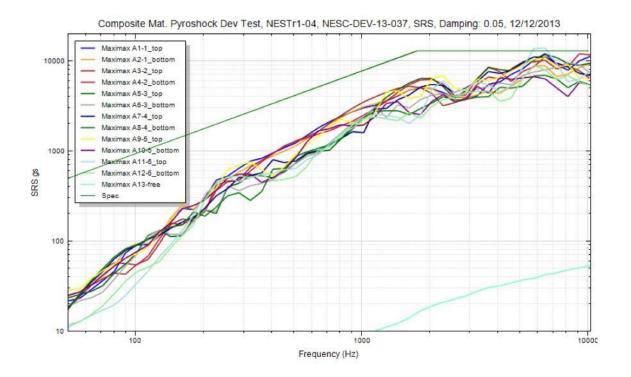
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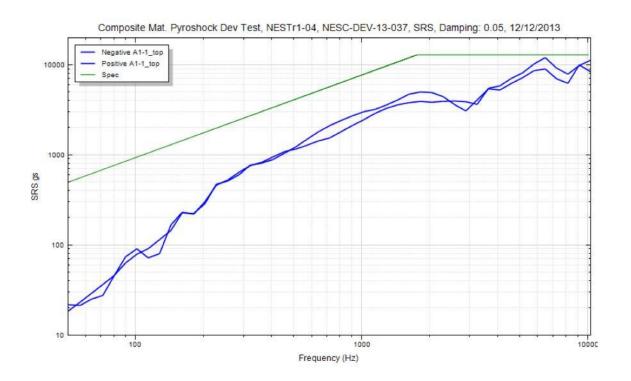
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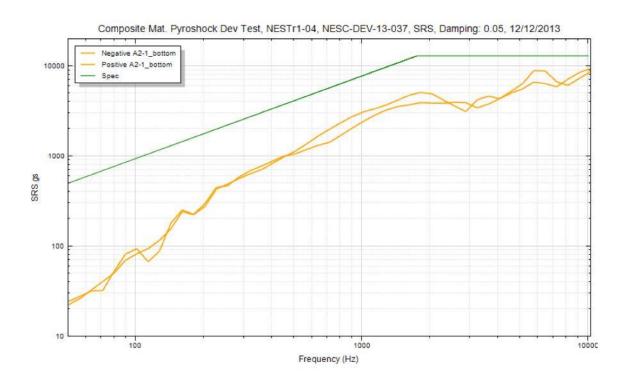
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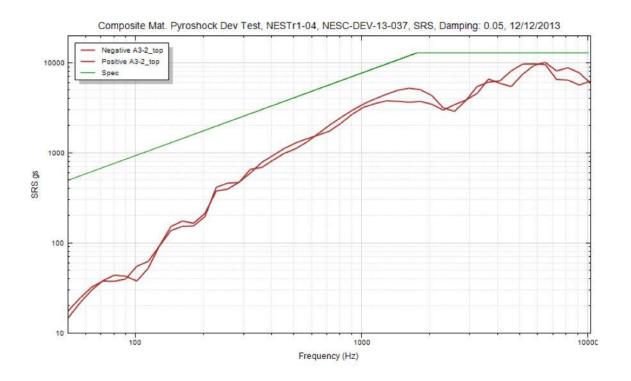
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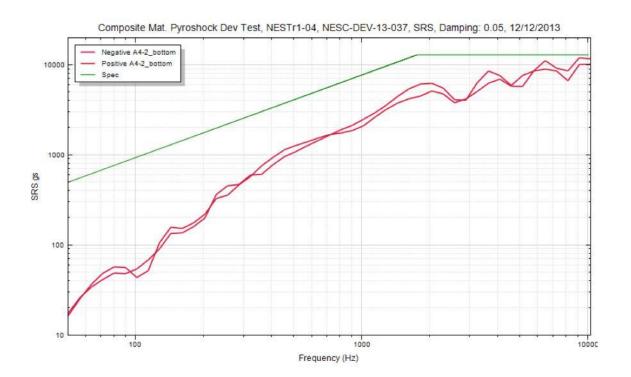
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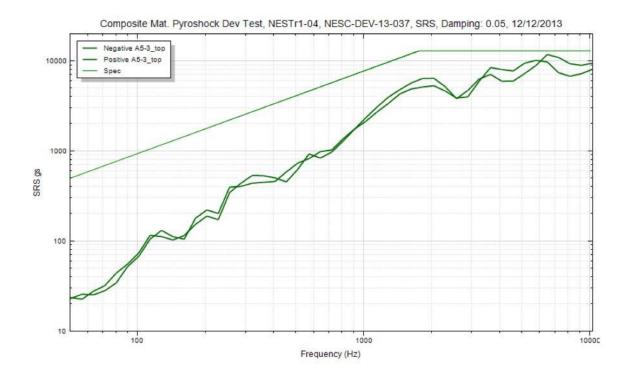
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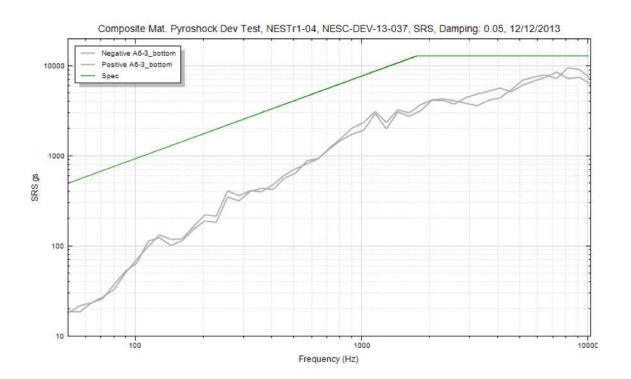
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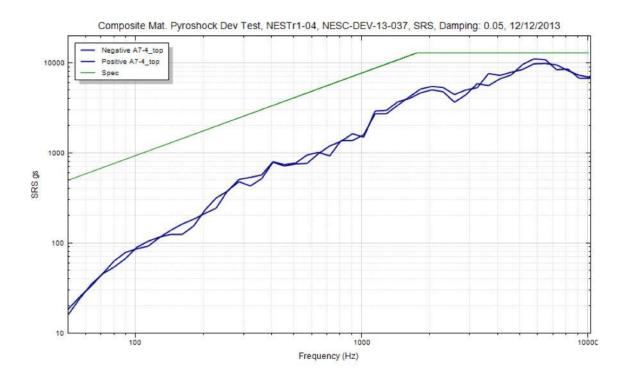
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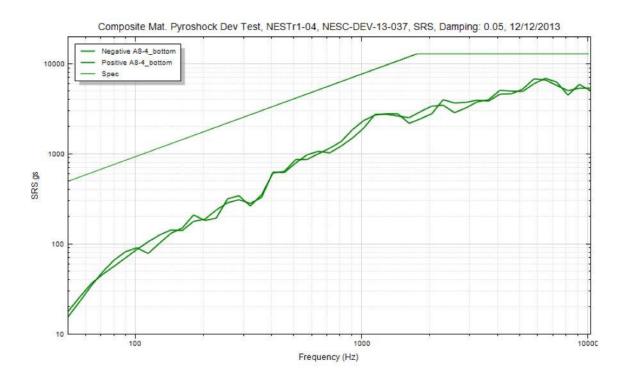
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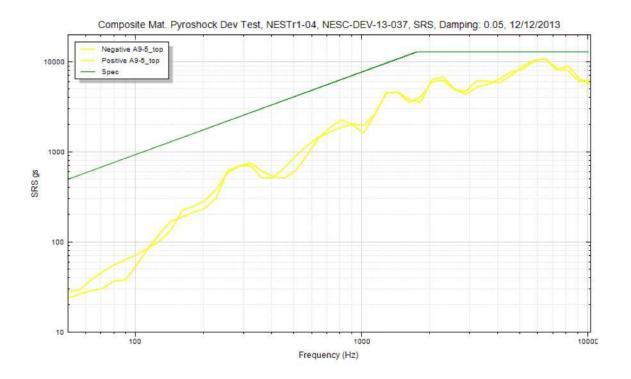
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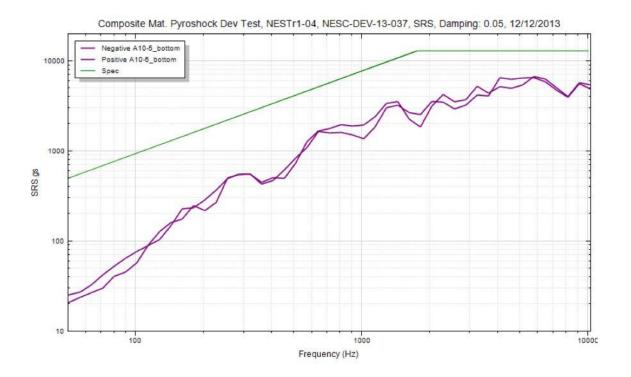
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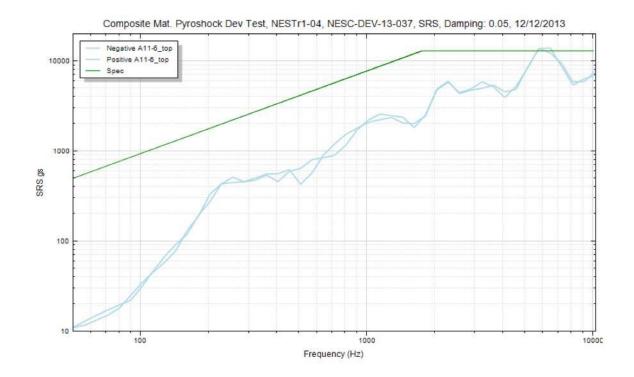
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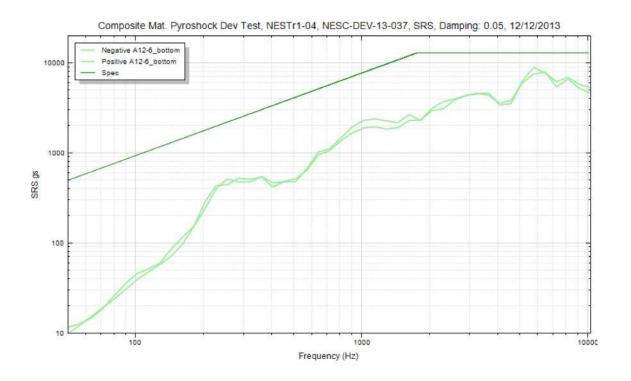
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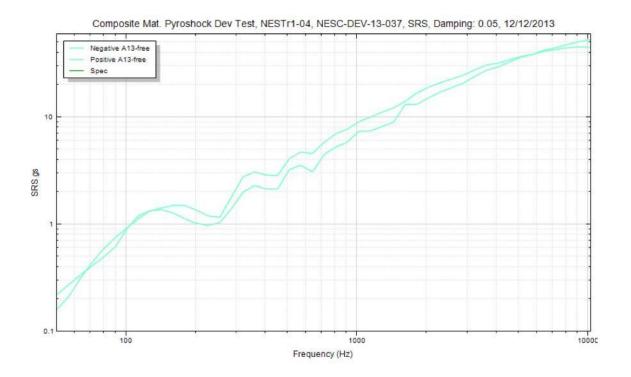
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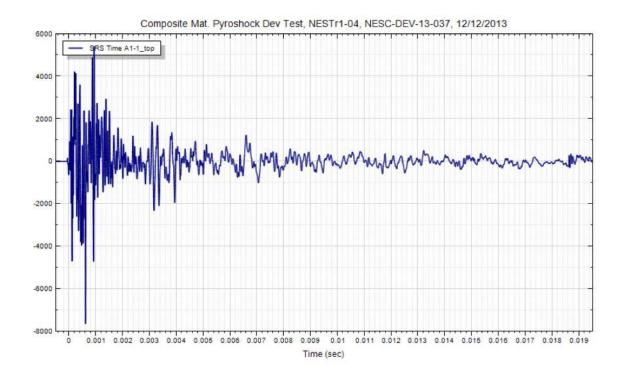
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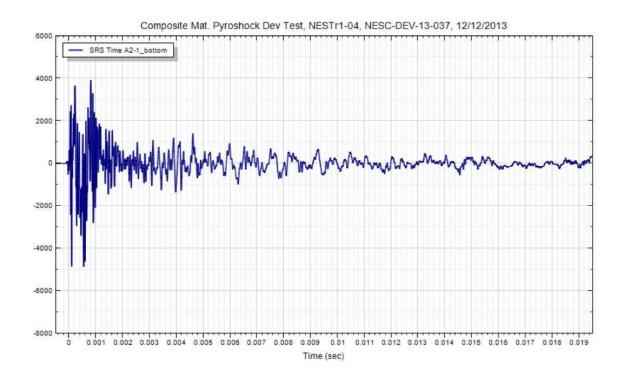
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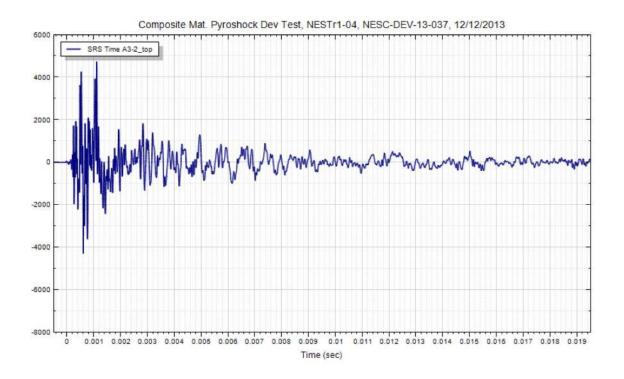
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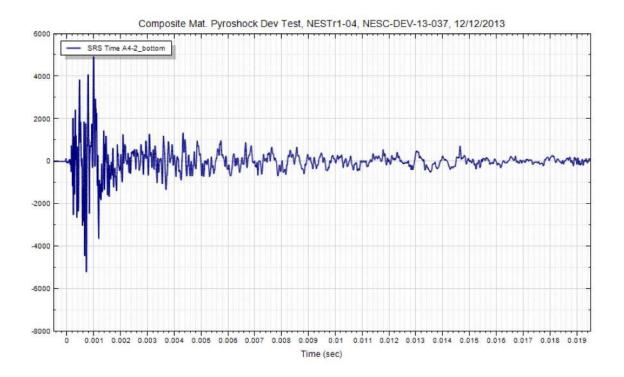
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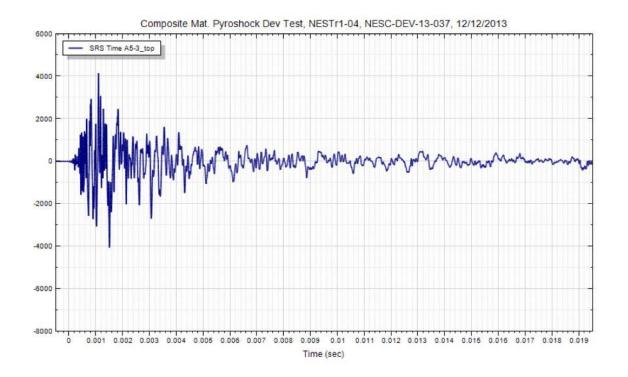
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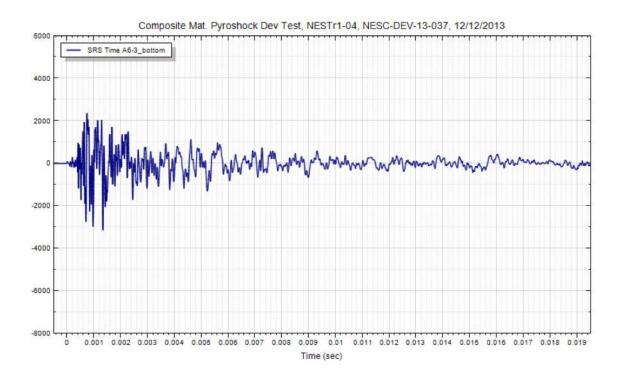
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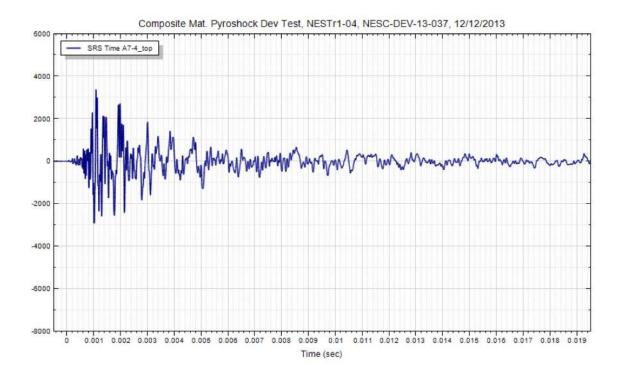
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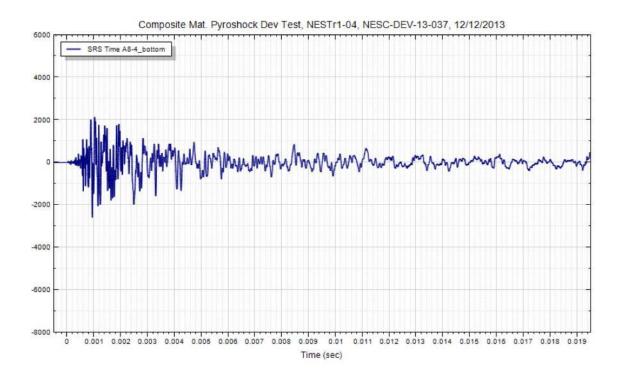
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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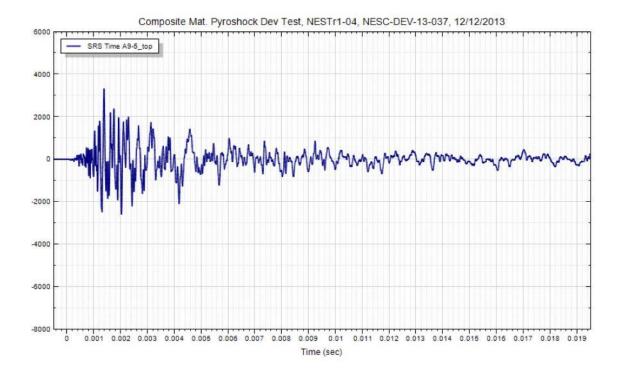
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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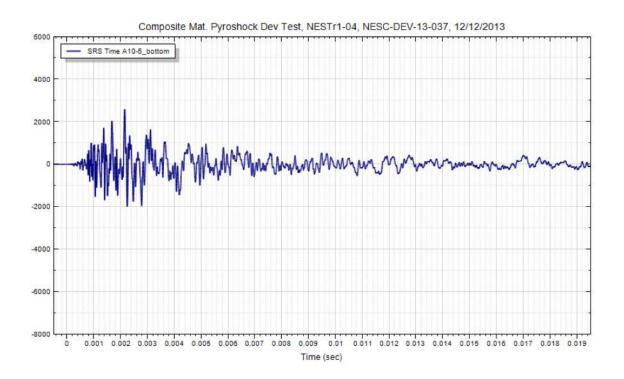
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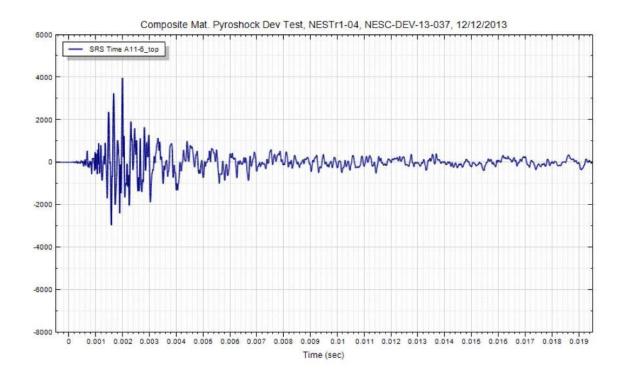
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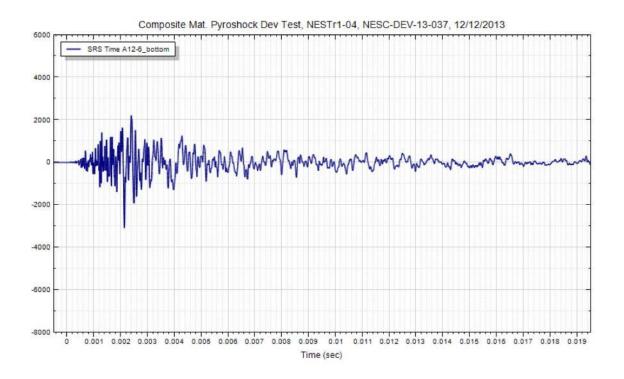
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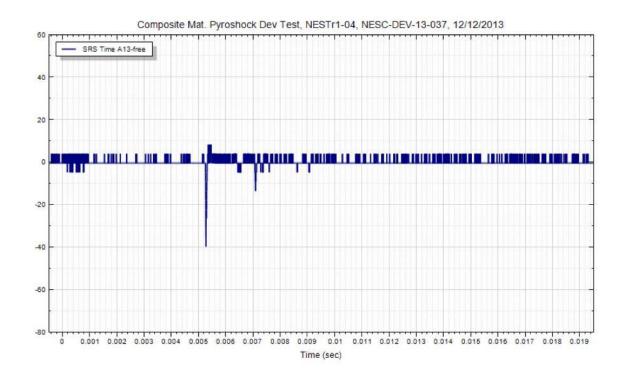
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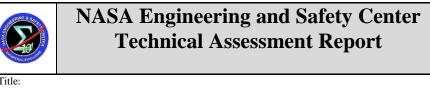
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #6 Accelerometer Data Panel 0320A002



NESC-RP-12-00783

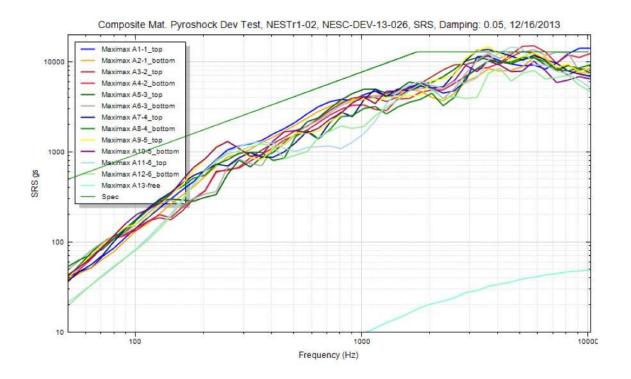
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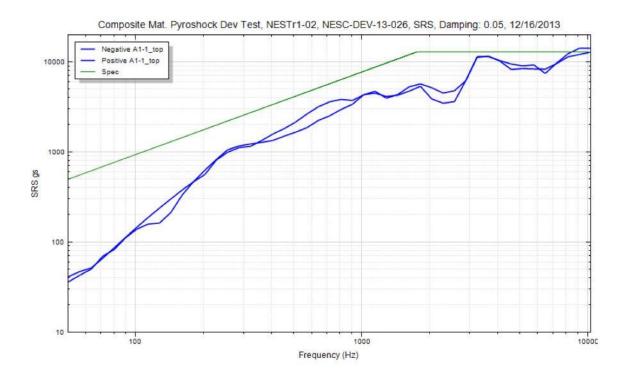
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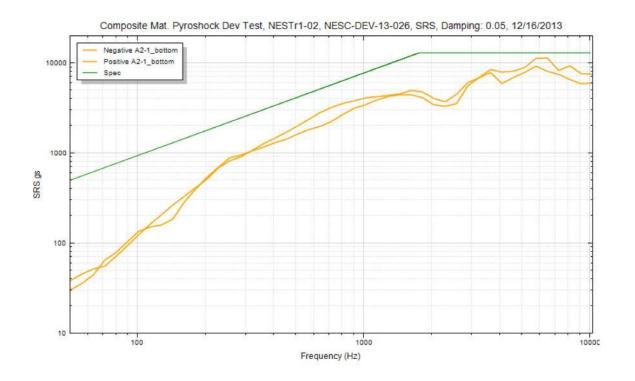
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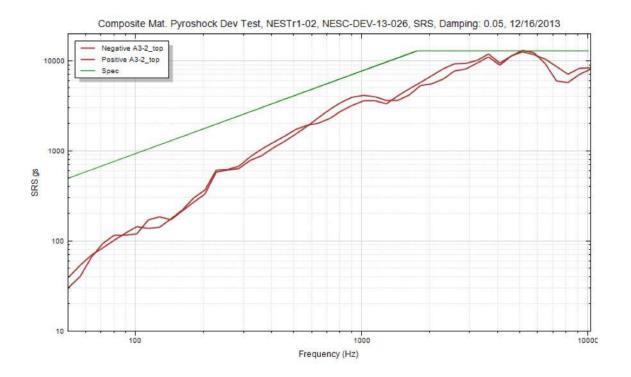
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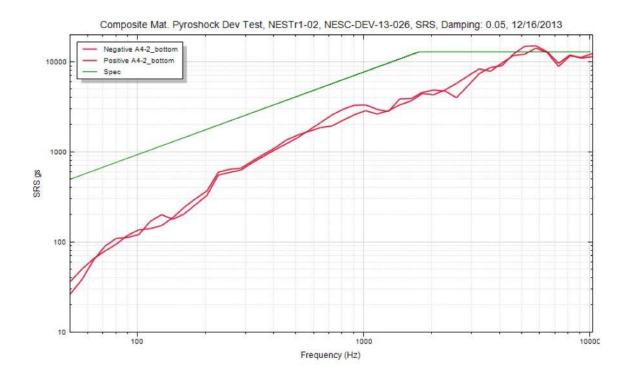
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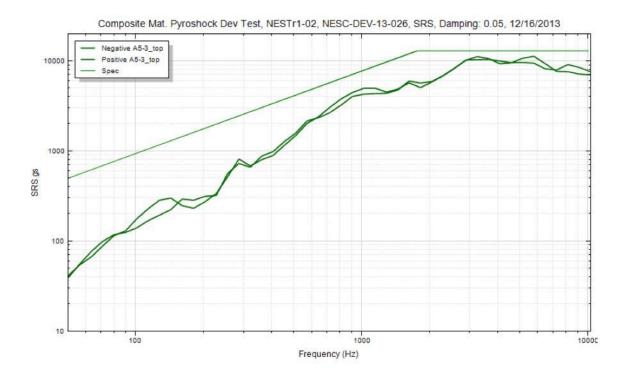
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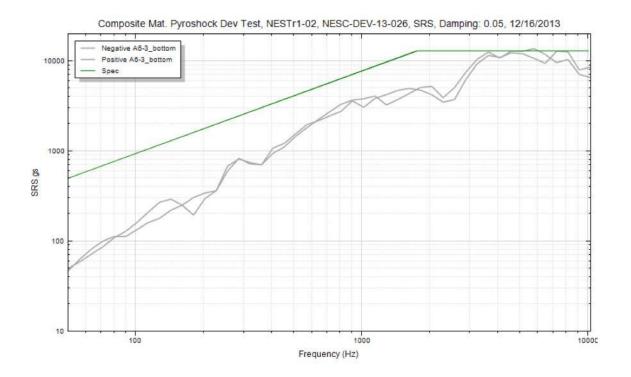
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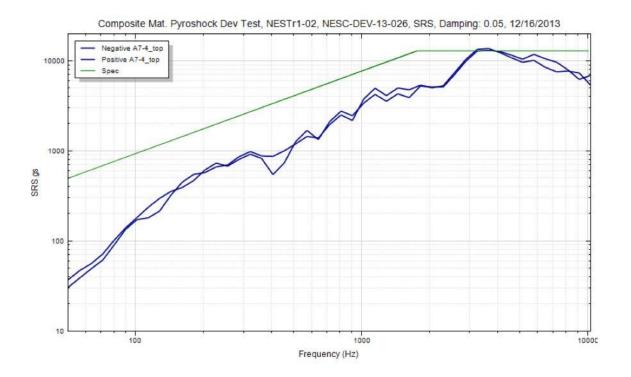
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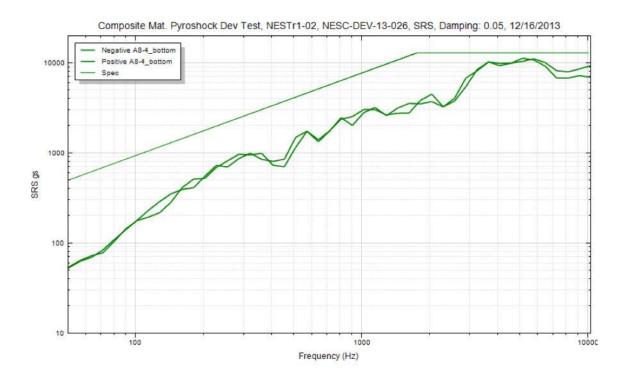
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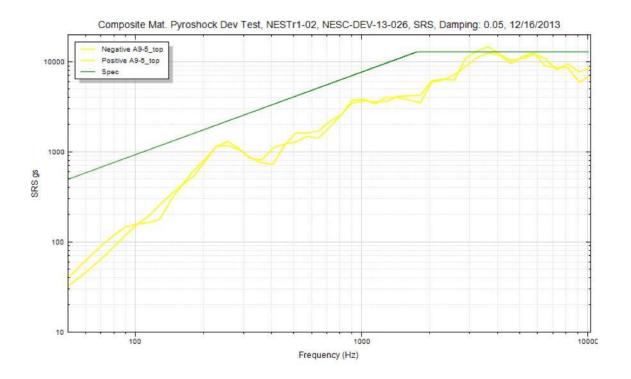
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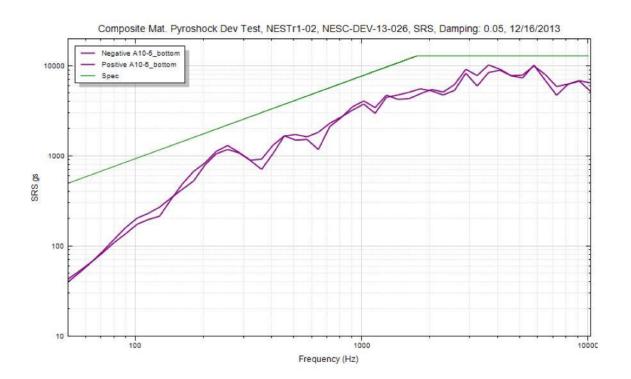
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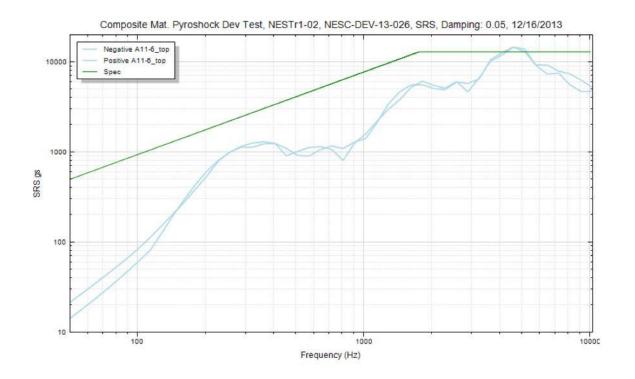
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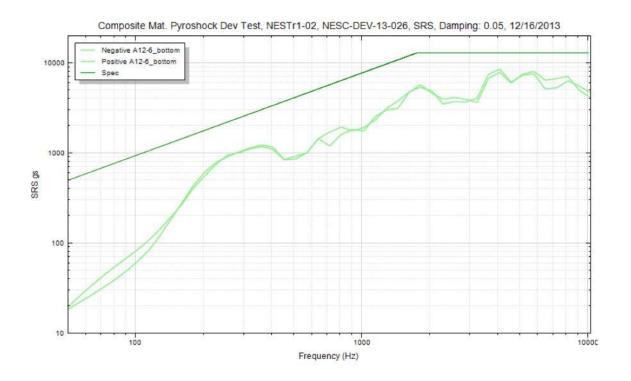
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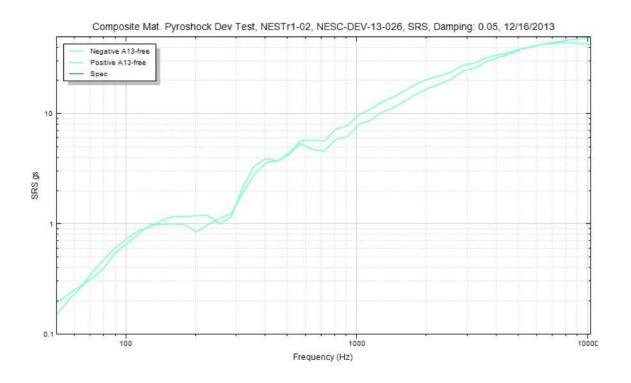
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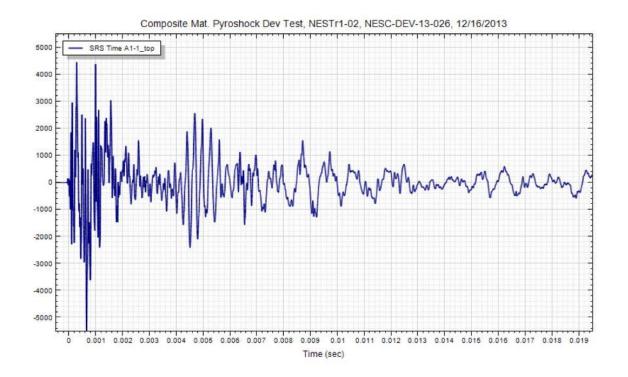
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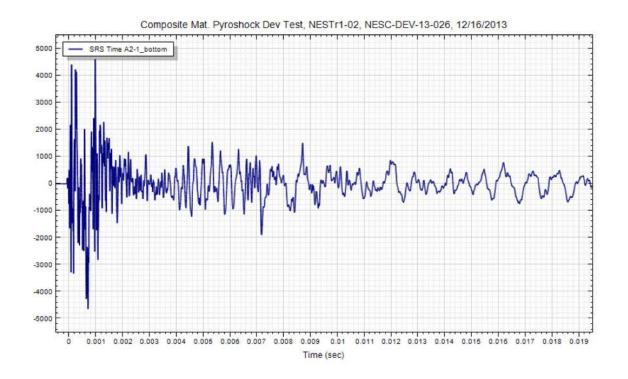
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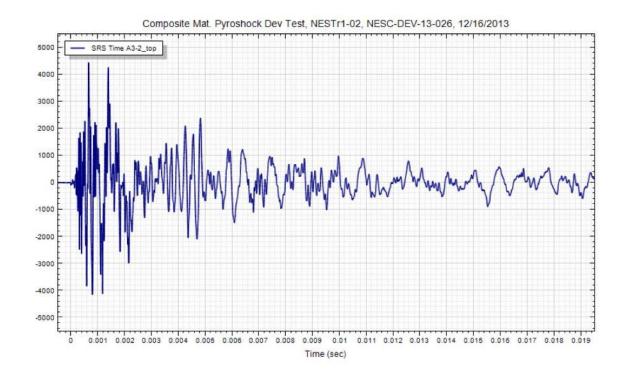
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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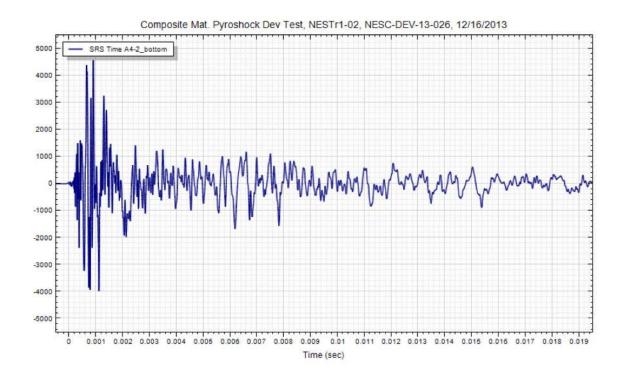
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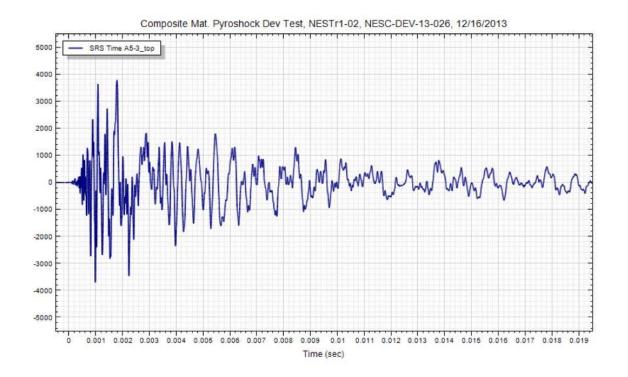
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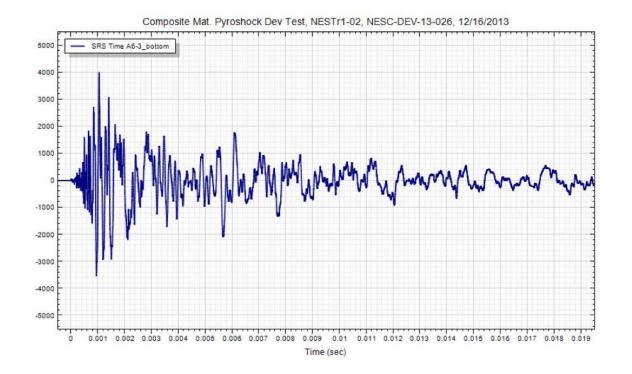
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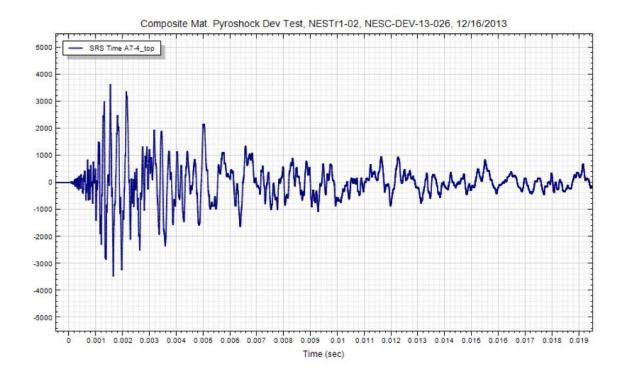
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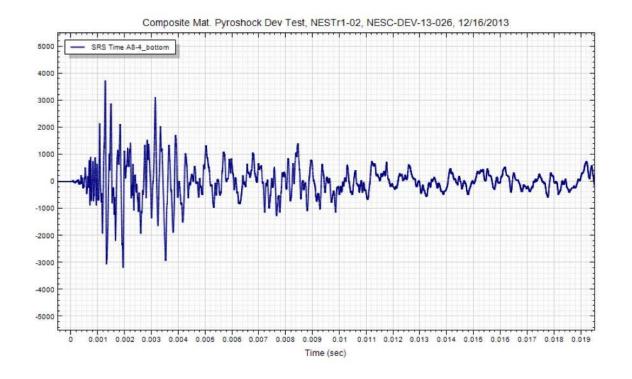
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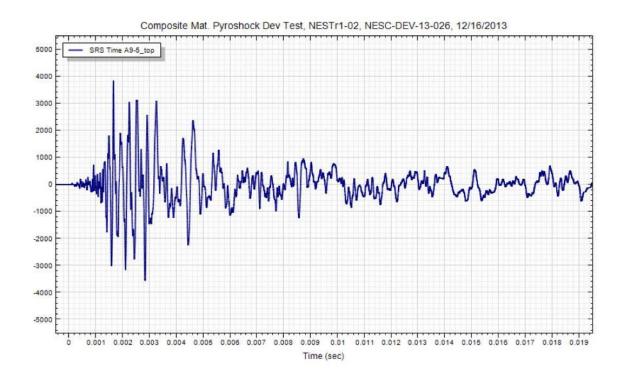
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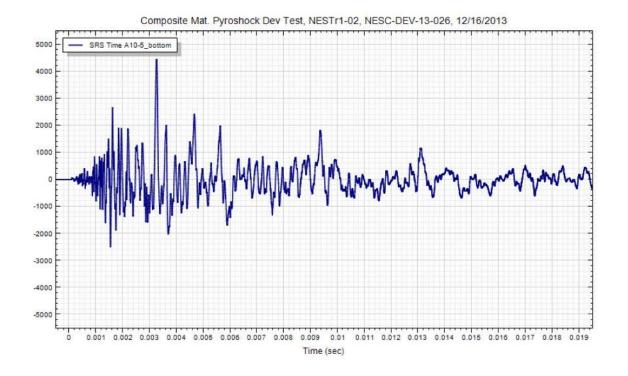
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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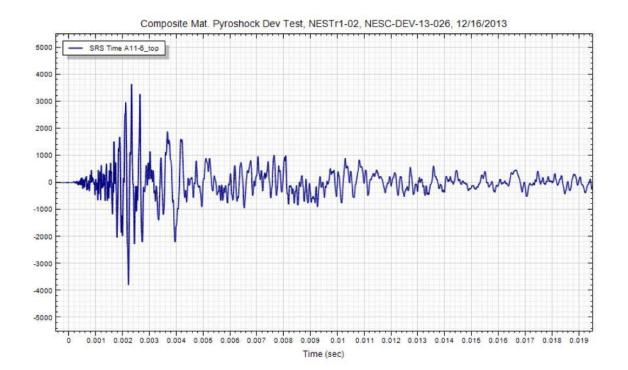
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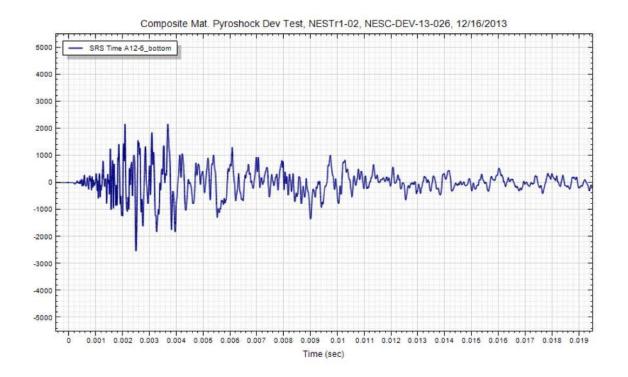
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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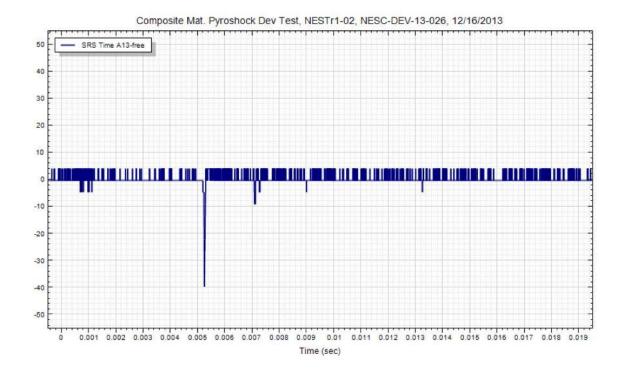
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #7 Accelerometer Data Panel 0346A026



NESC-RP-12-00783

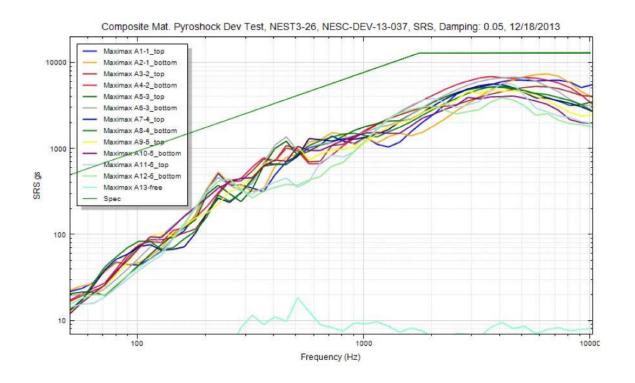
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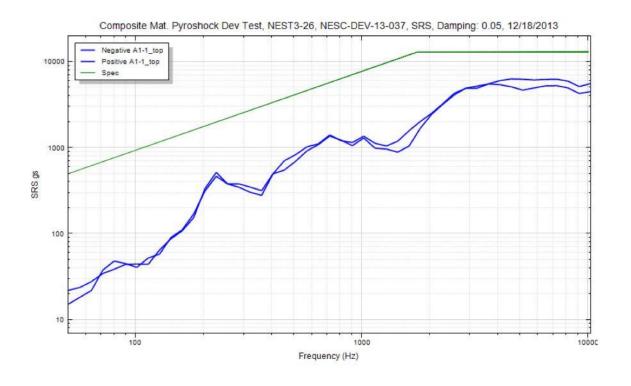
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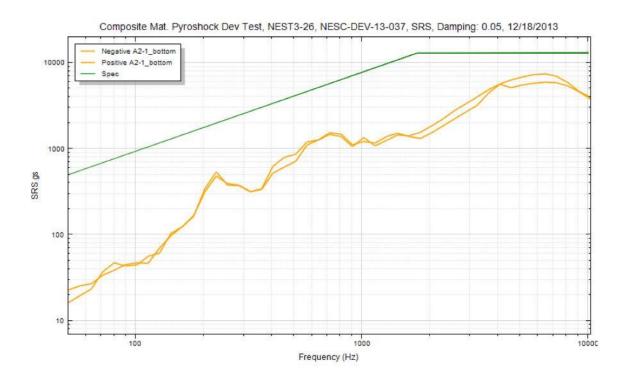
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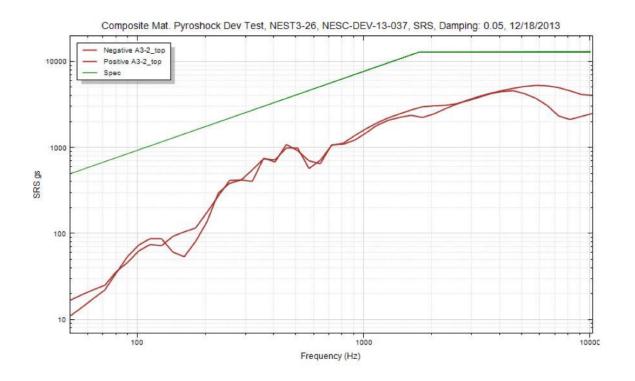
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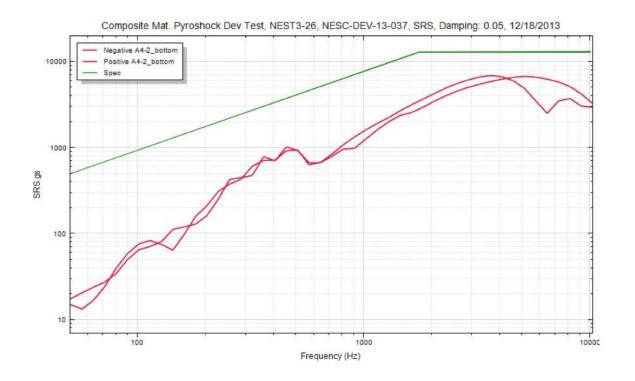
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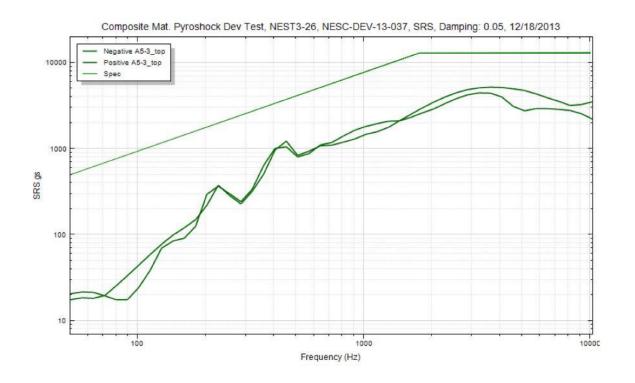
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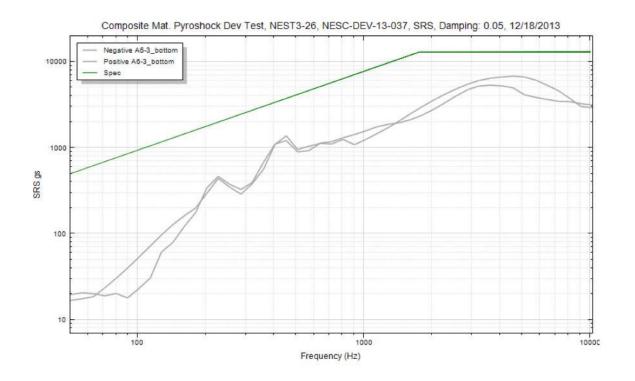
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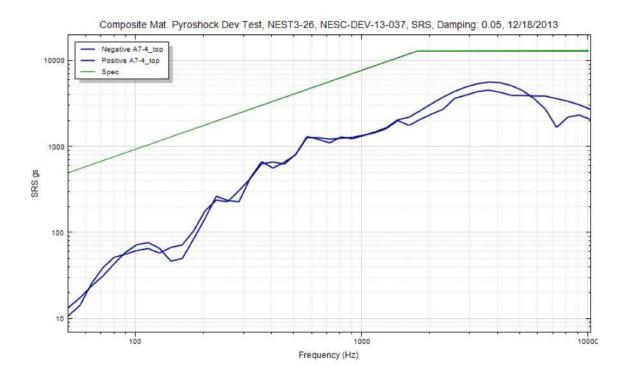
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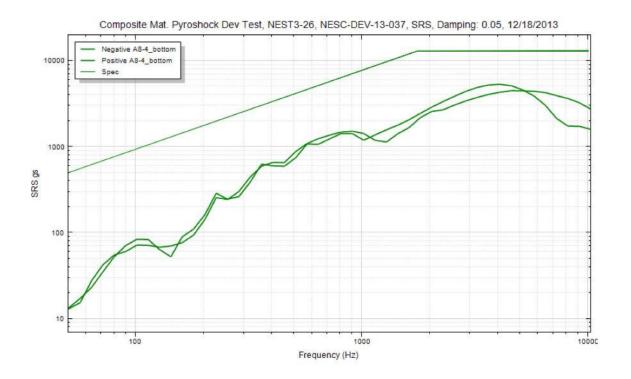
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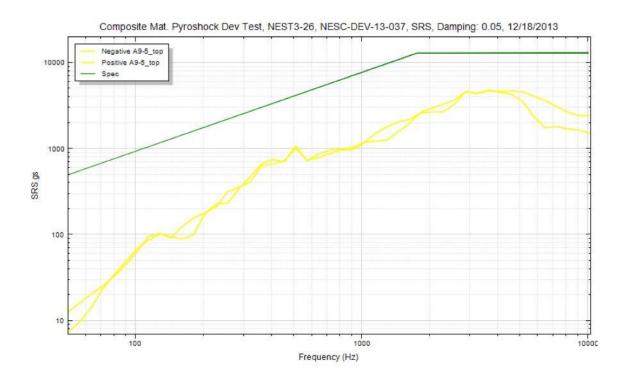
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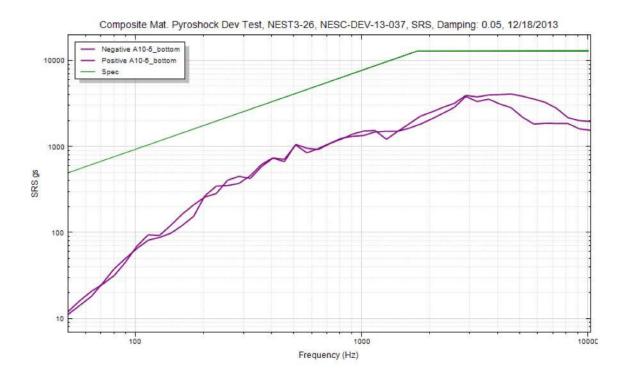
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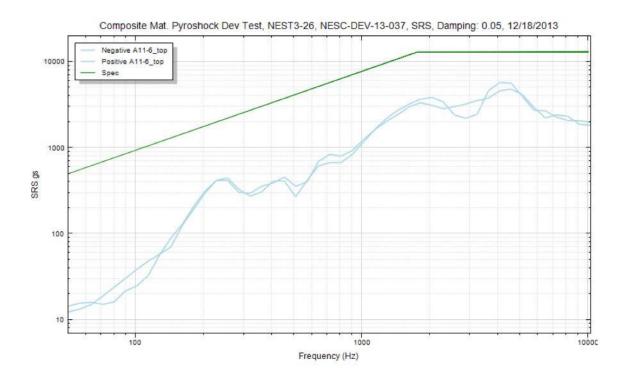
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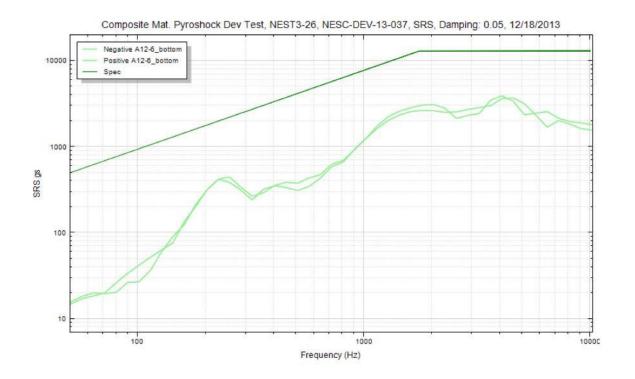
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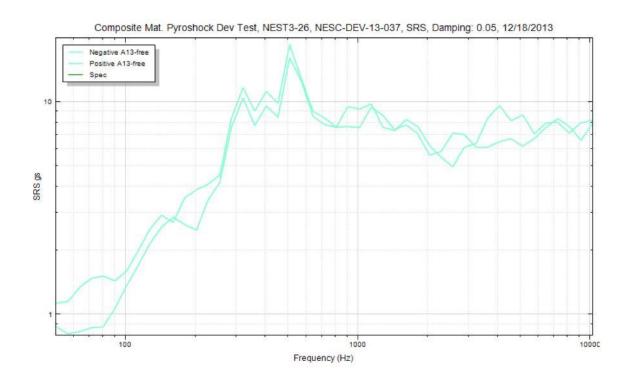
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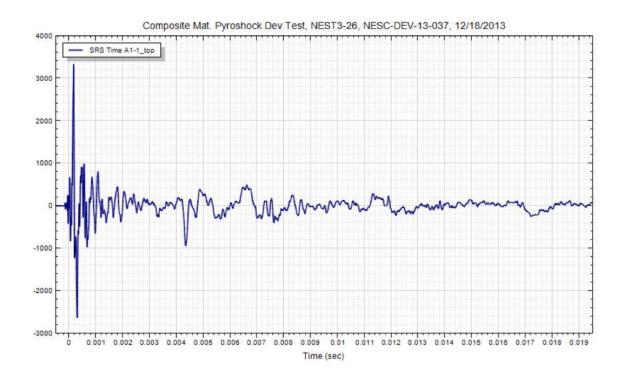
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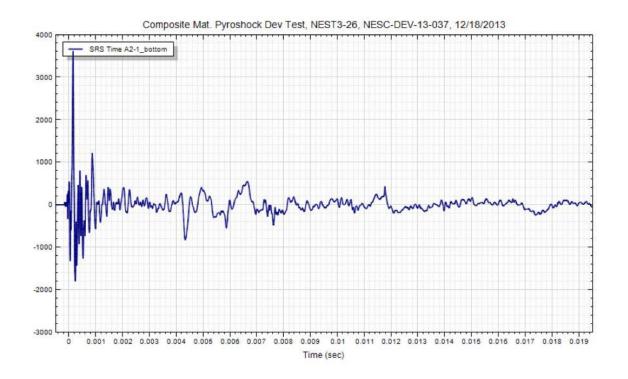
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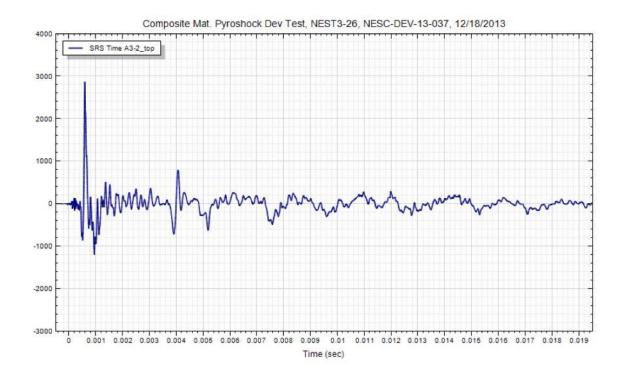
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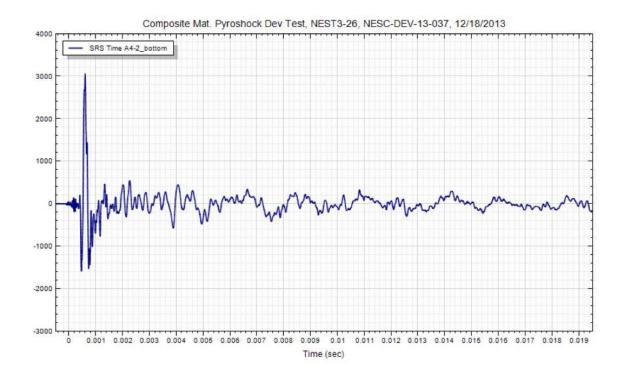
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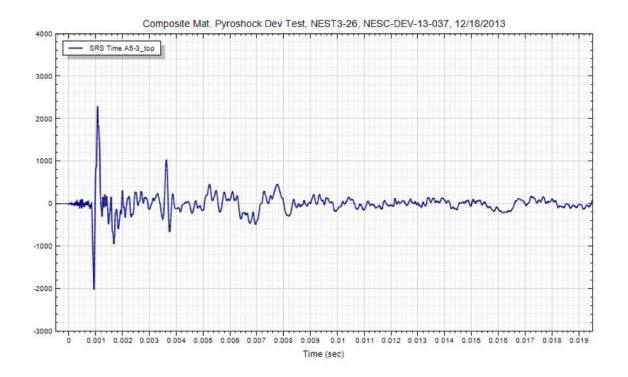
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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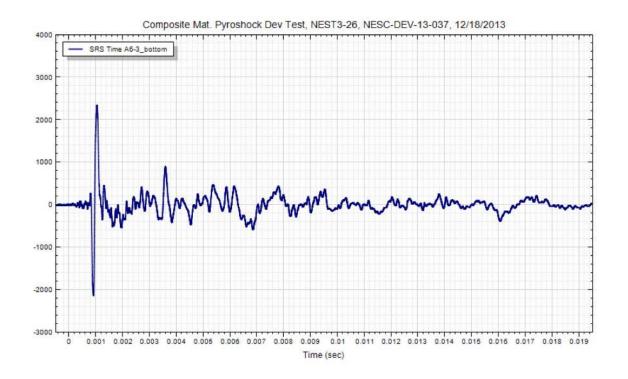
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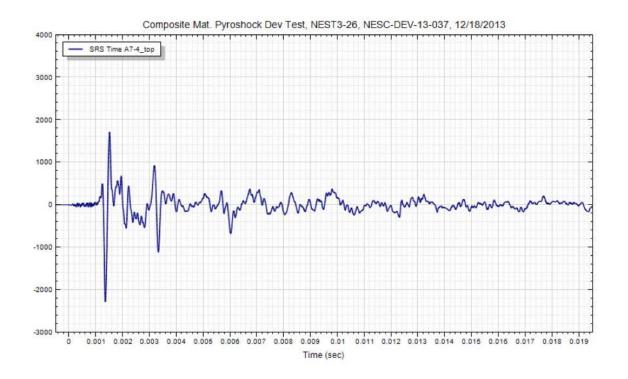
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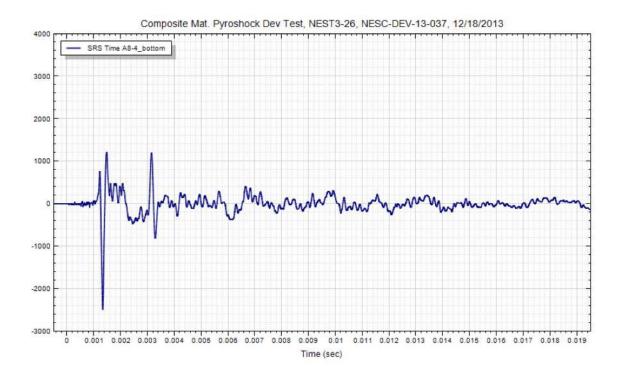
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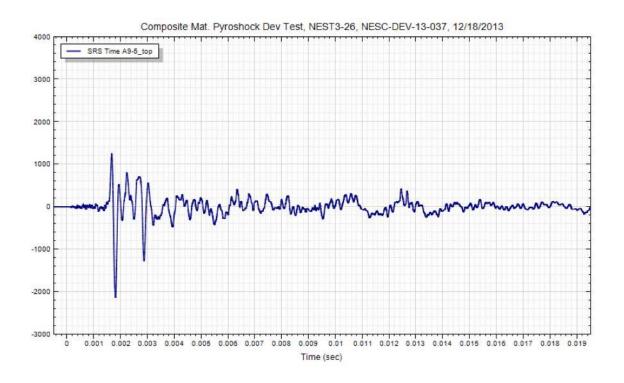
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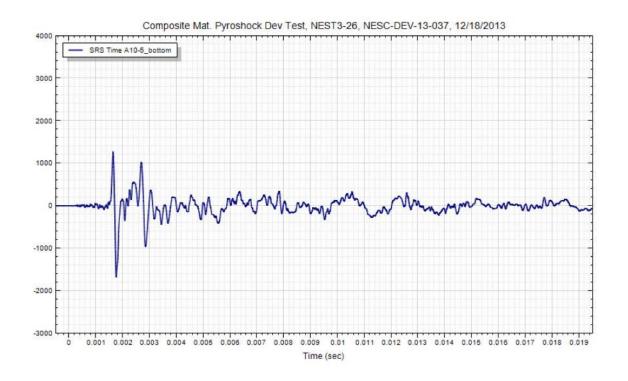
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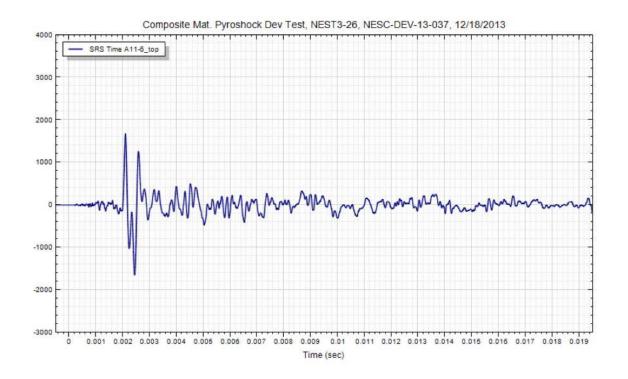
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





NESC-RP-12-00783

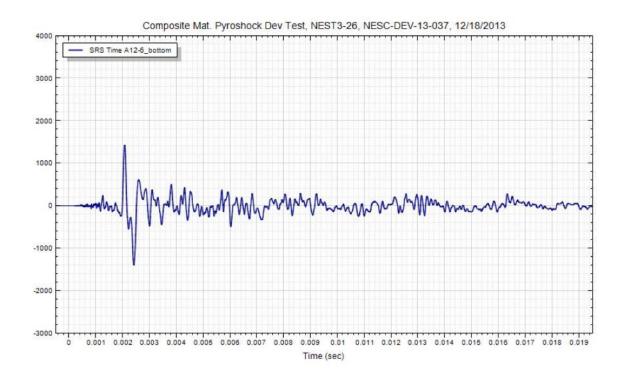
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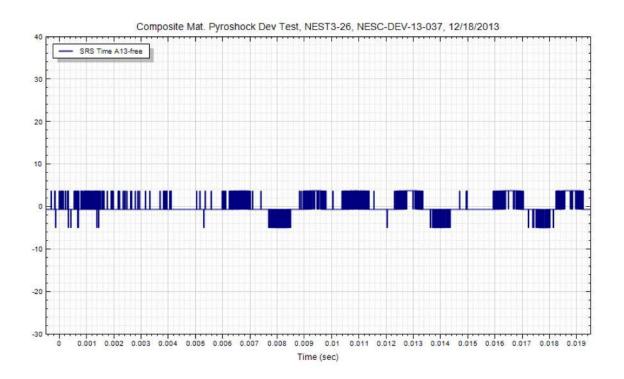
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #8 Accelerometer Data Panel 0326A010



NESC-RP-12-00783

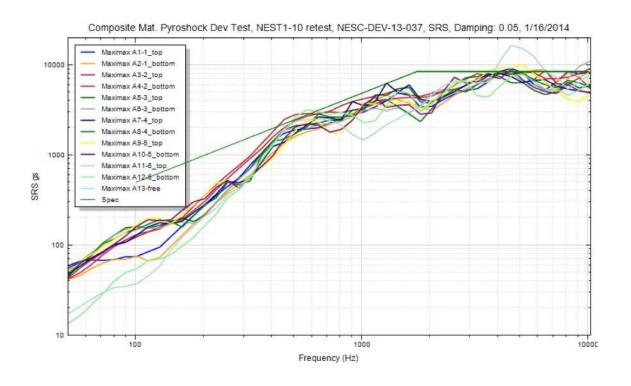
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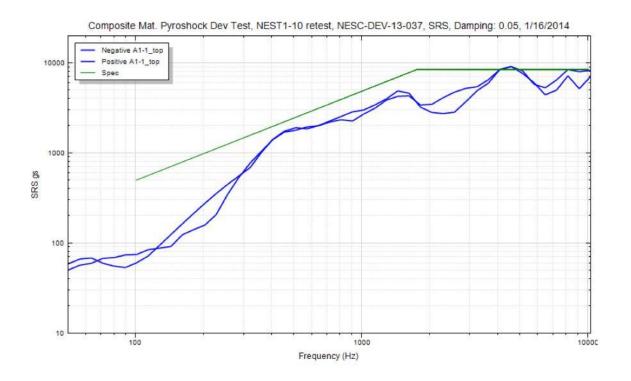
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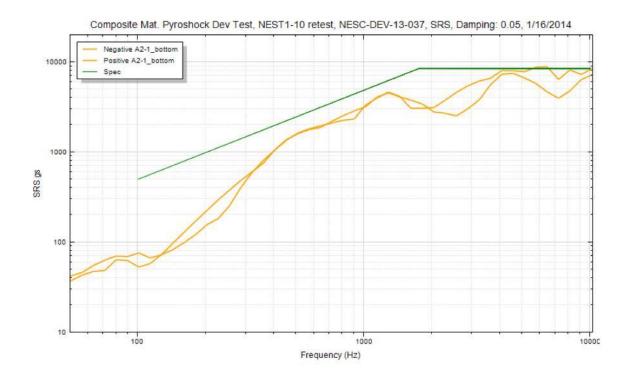
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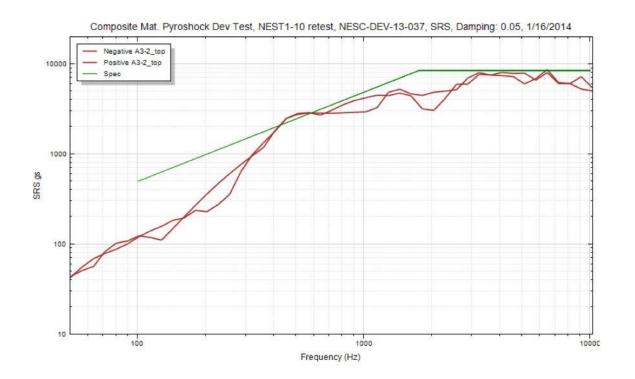
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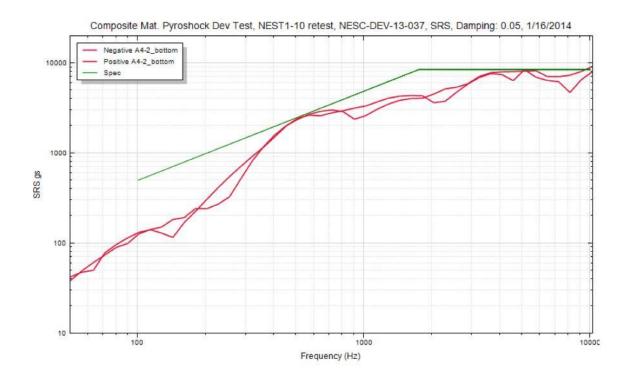
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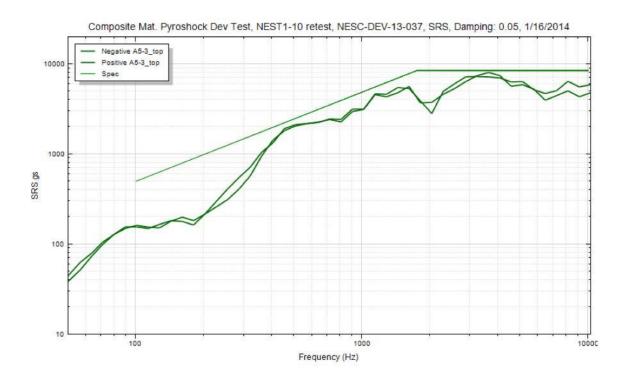
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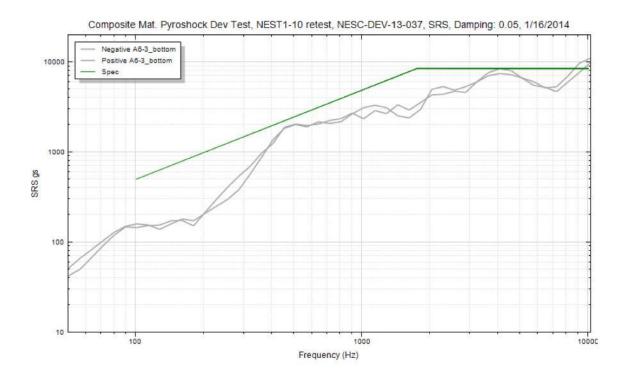
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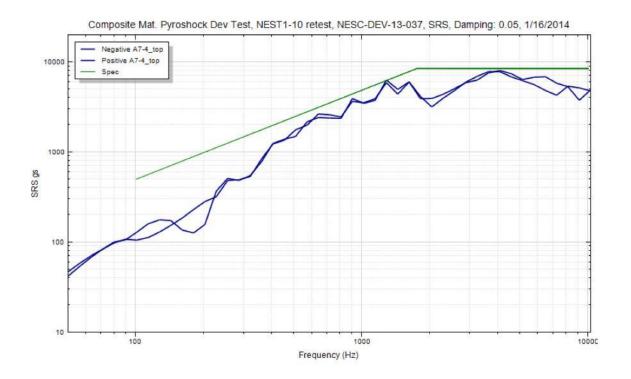
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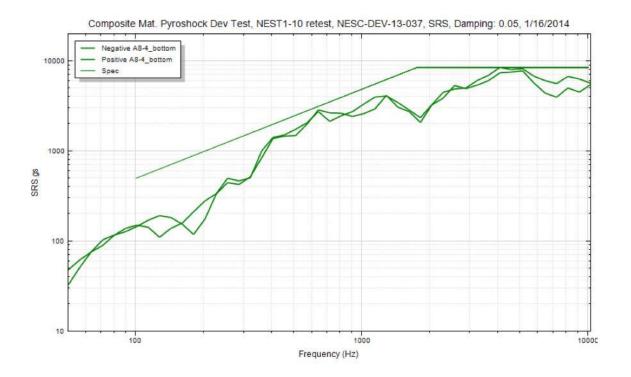
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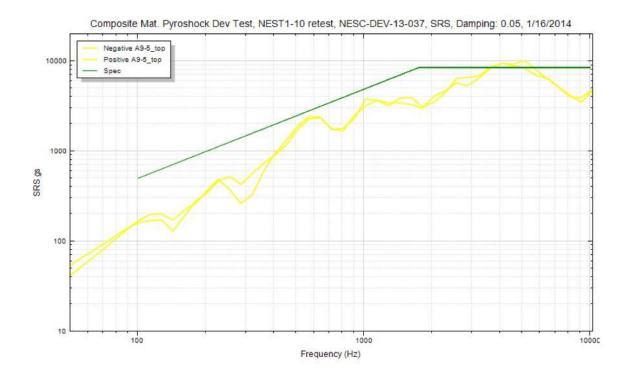
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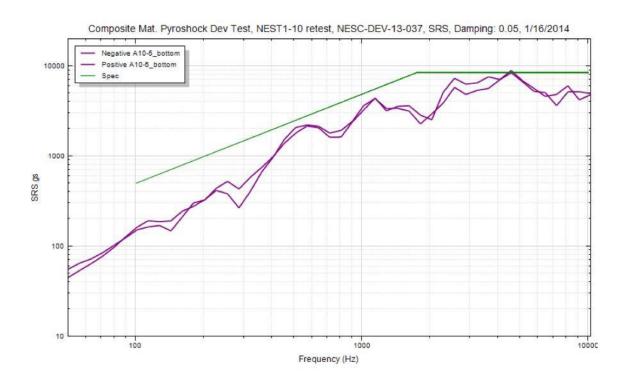
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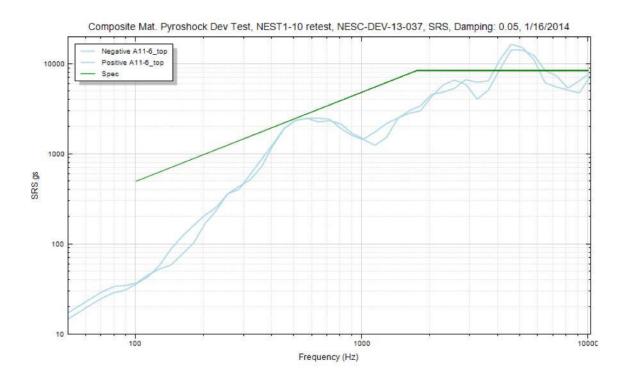
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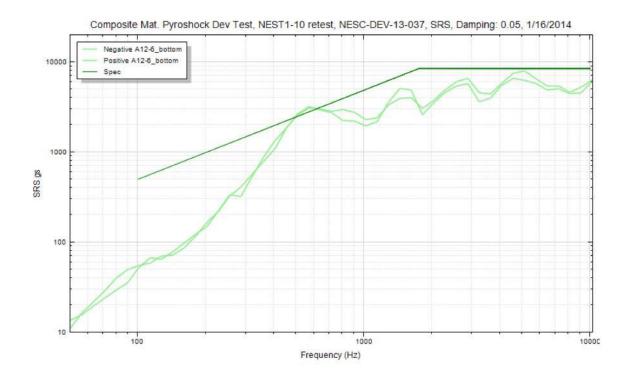
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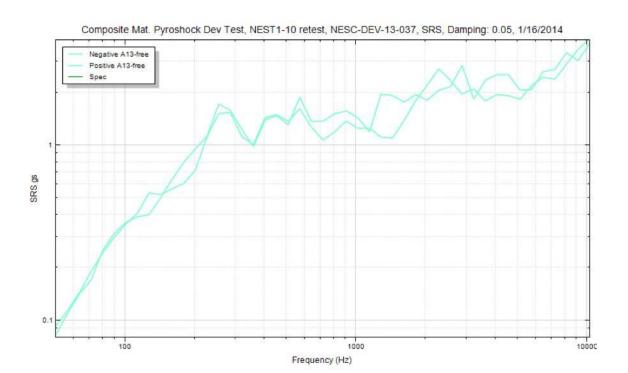
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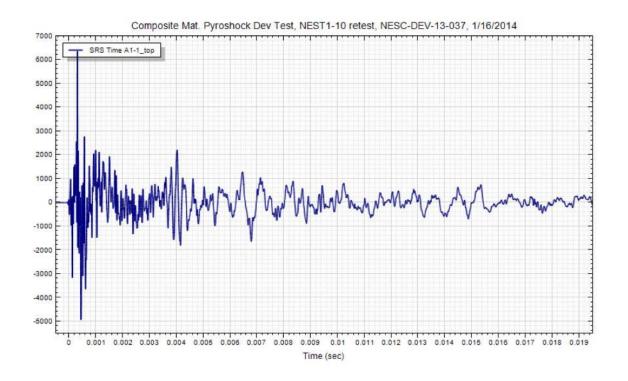
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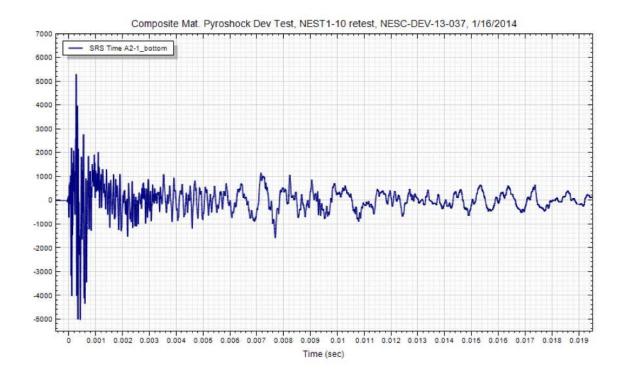
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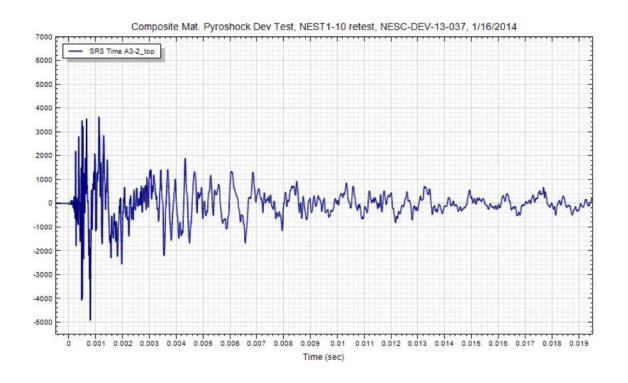
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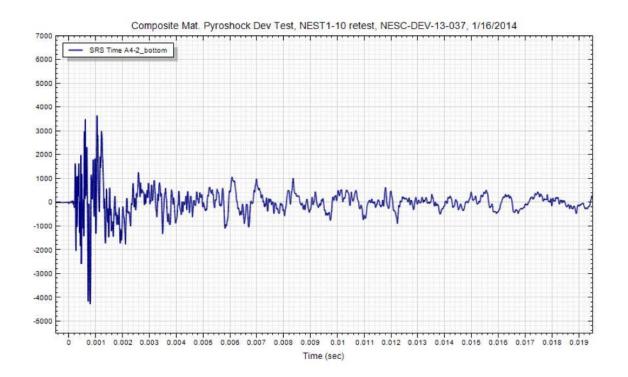
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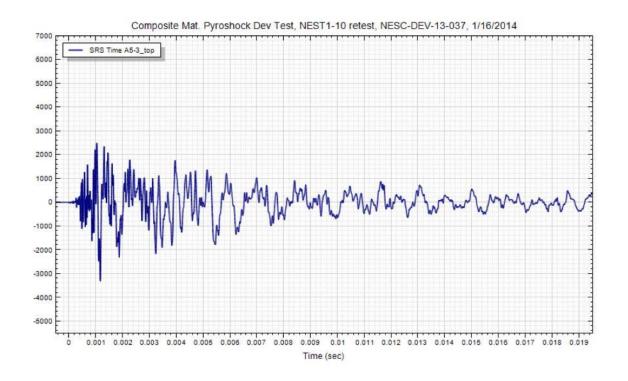
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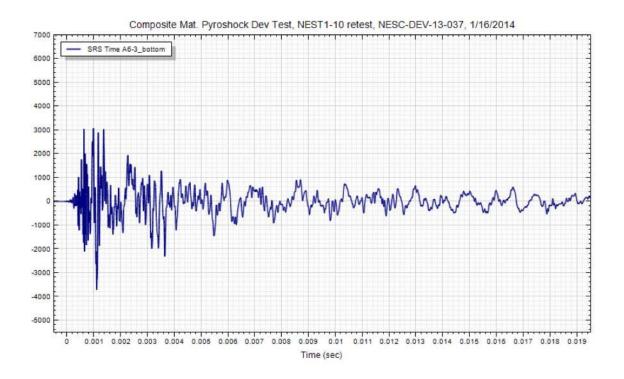
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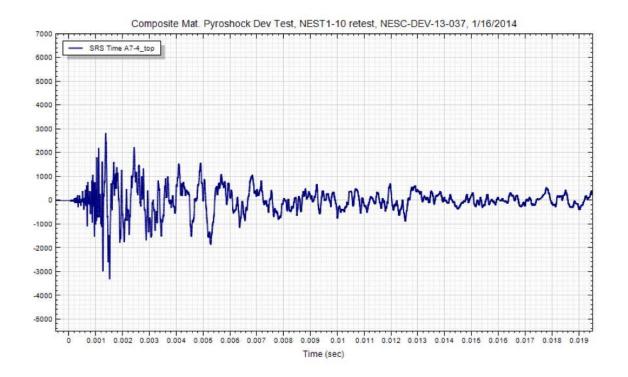
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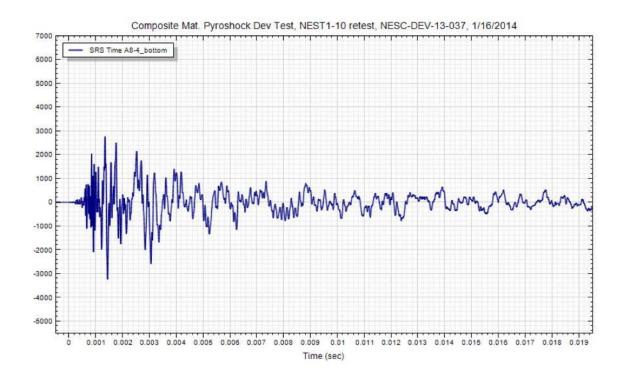
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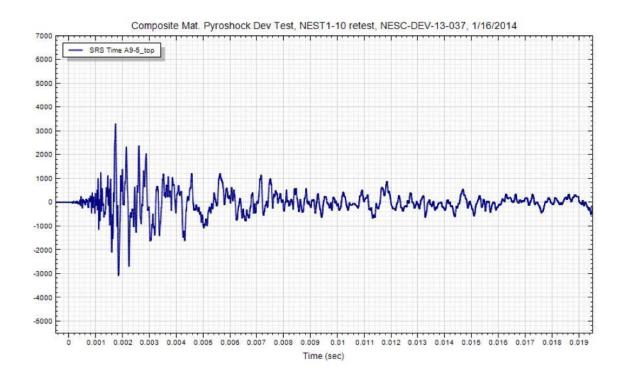
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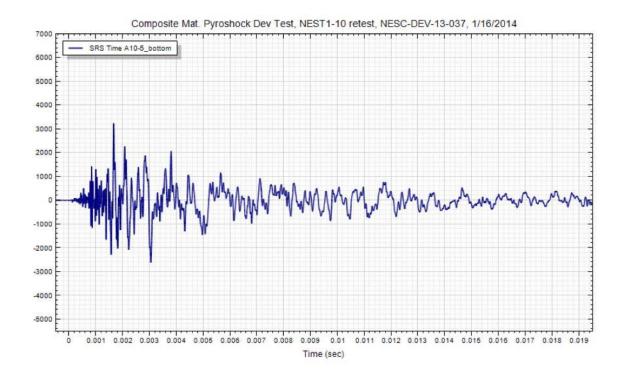
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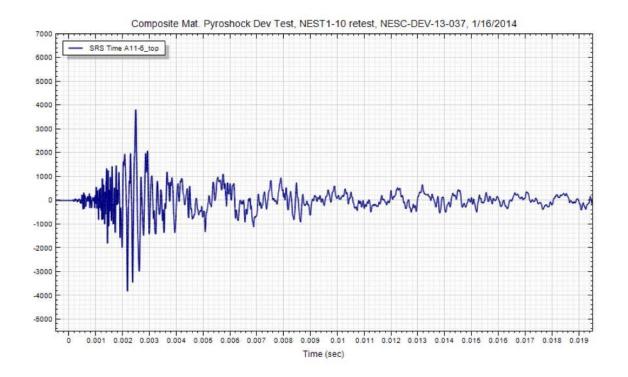
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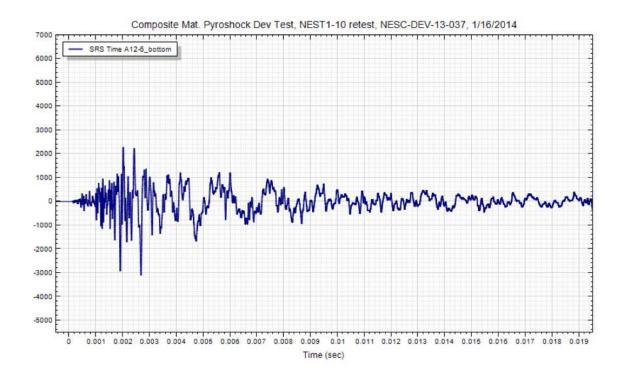
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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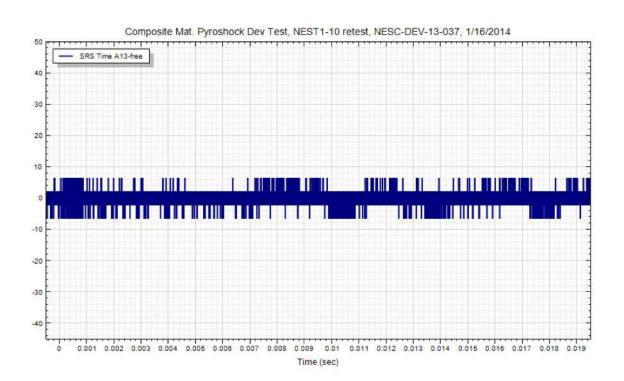
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #9 Accelerometer Data Panel 0326A009



NESC-RP-12-00783

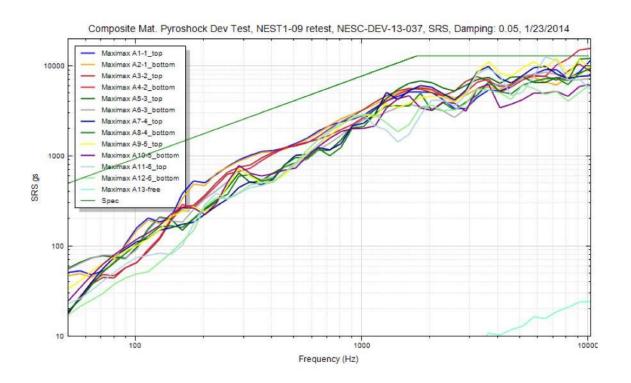
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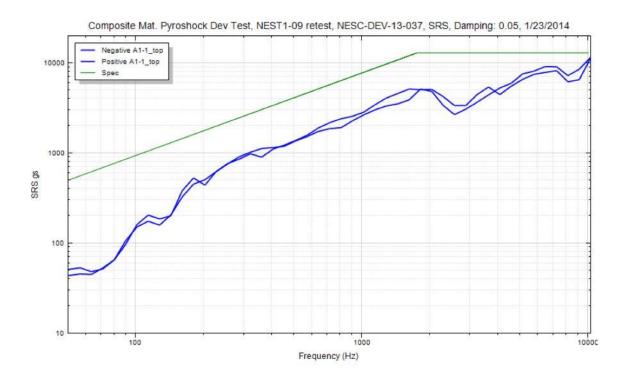
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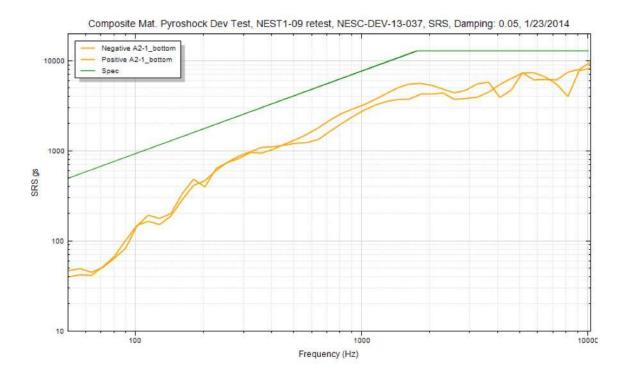
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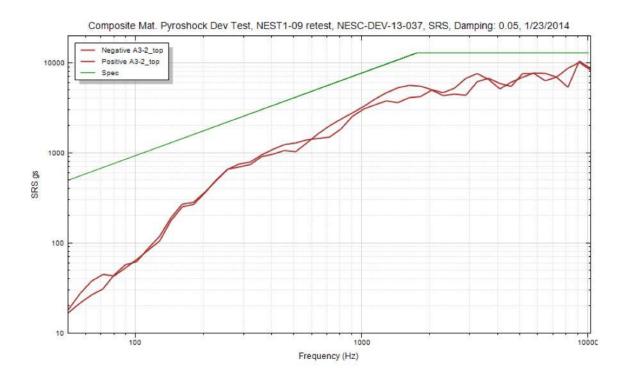
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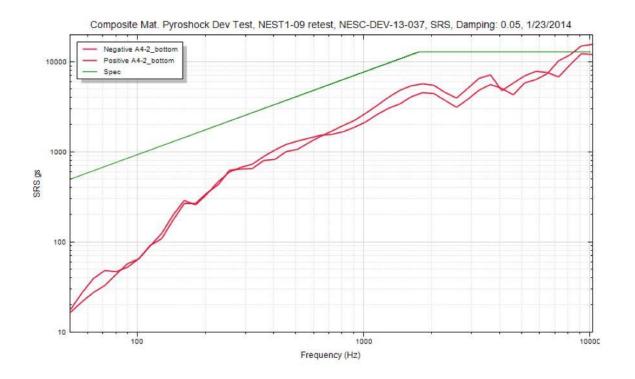
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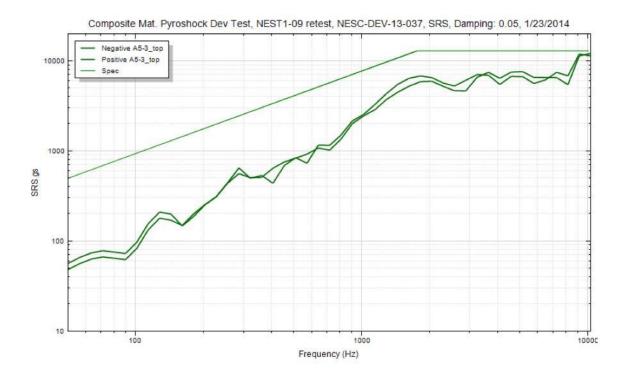
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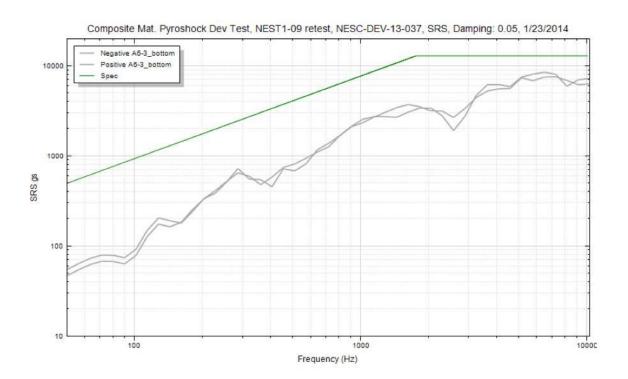
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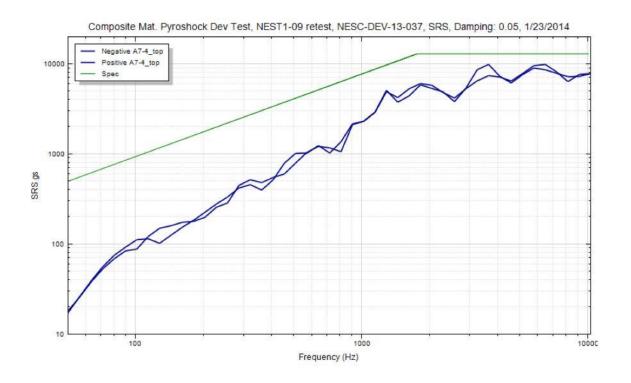
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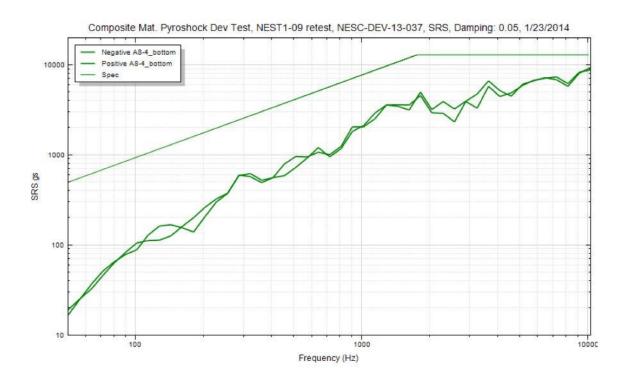
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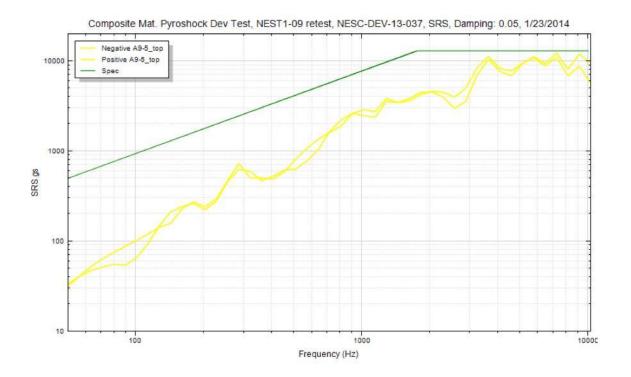
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

Composite Mat. Pyroshock Dev Test, NEST1-09 retest, NESC-DEV-13-037, SRS, Damping: 0.05, 1/23/2014

Negative A10-5\_bottom
Positive A10-5\_bottom
Spec

1000
Frequency (Hz)



NESC-RP-12-00783

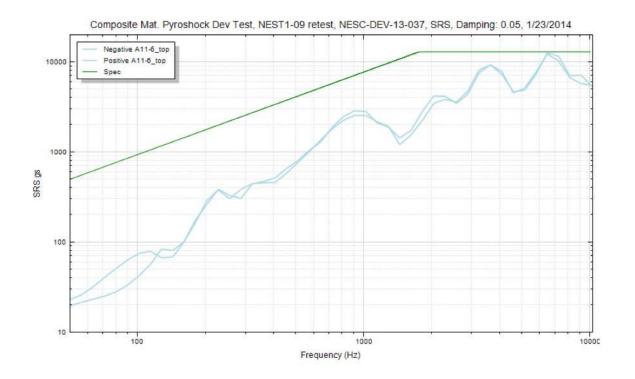
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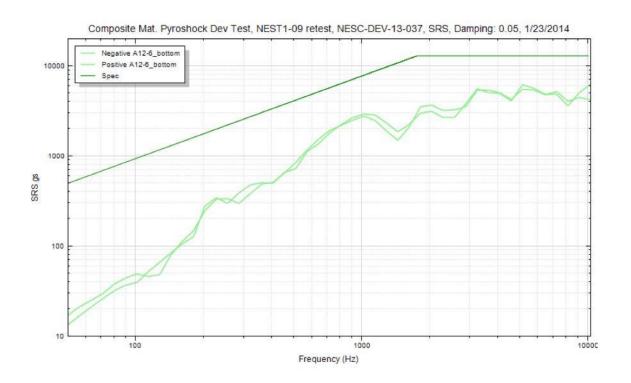
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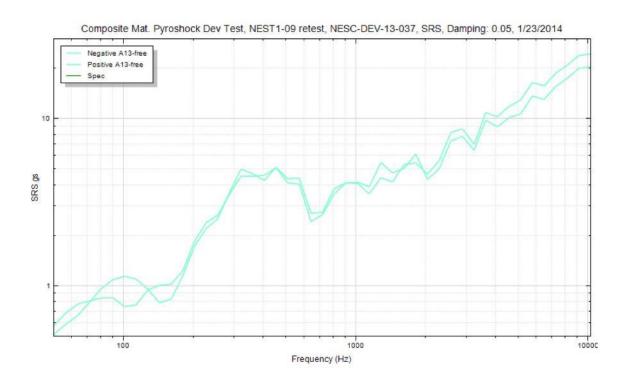
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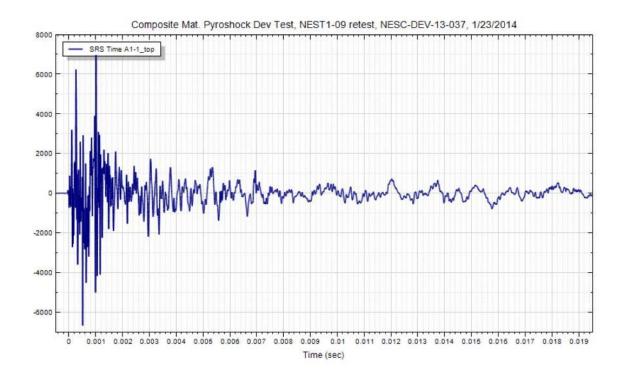
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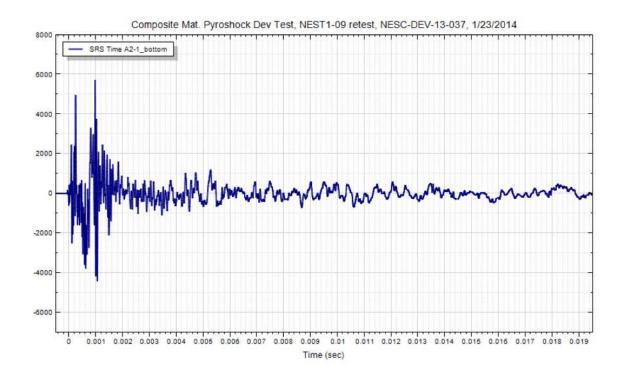
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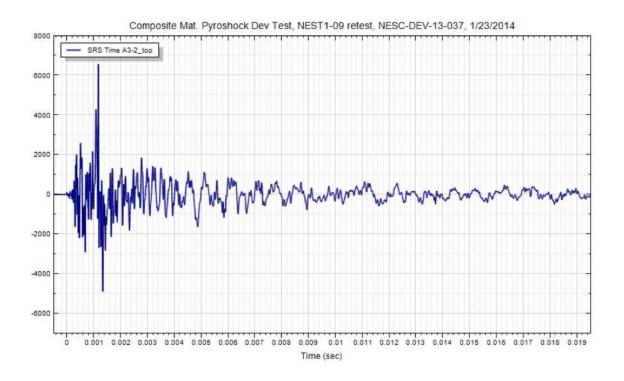
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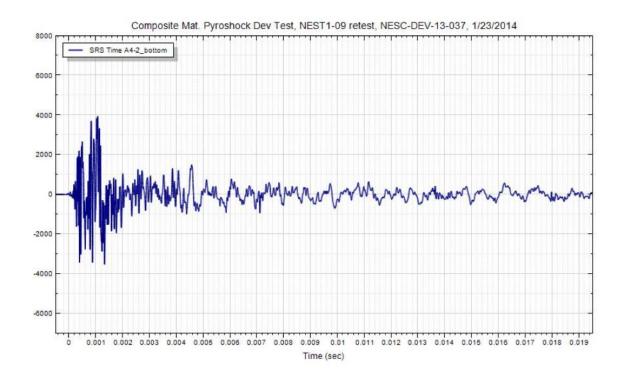
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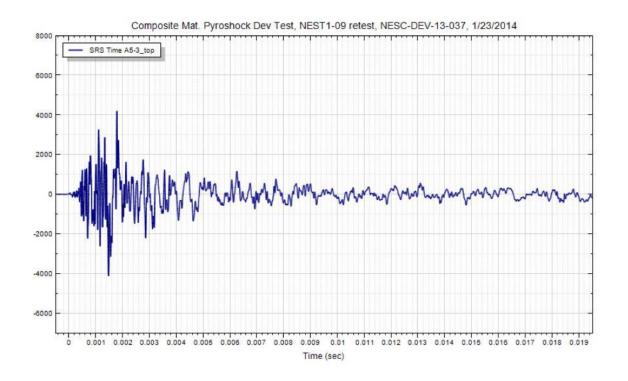
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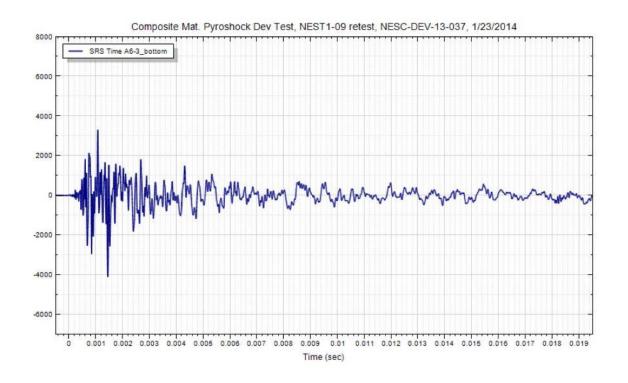
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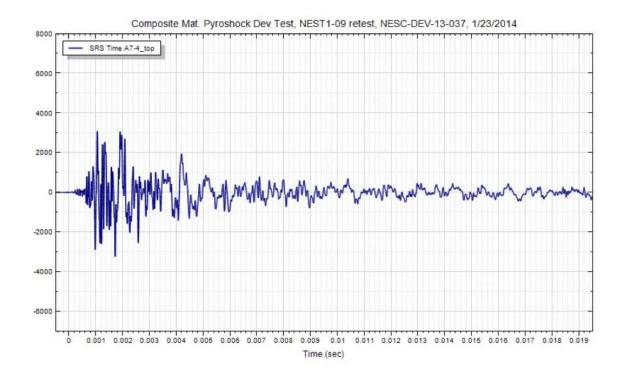
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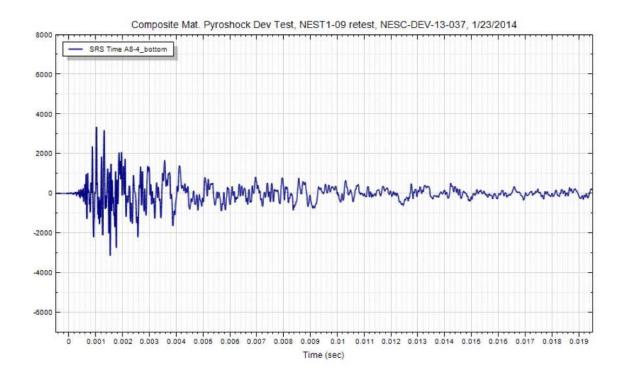
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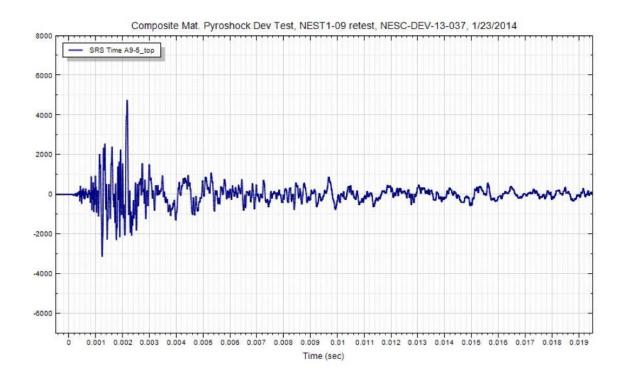
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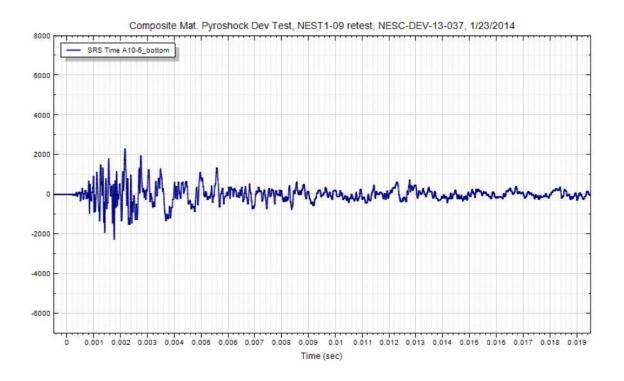
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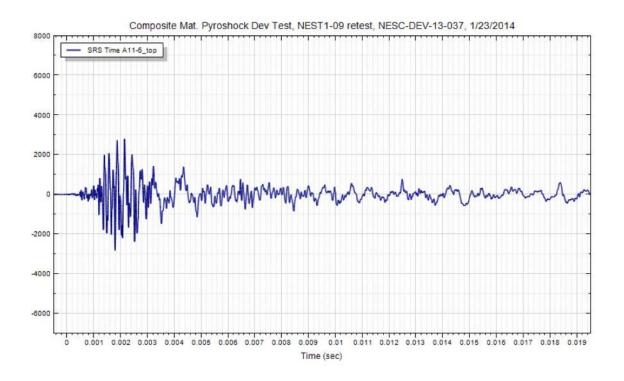
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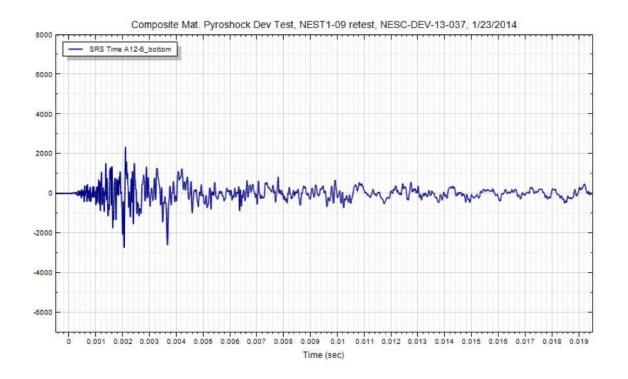
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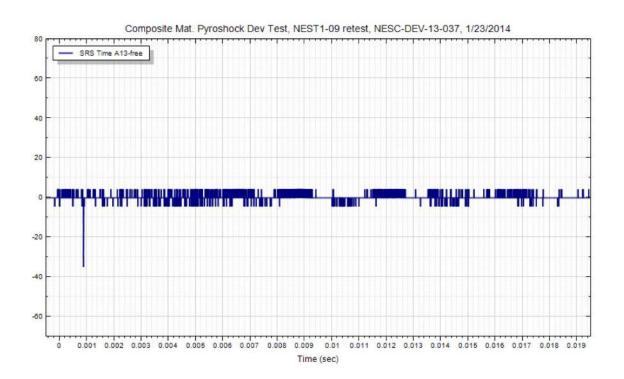
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #10 Accelerometer Data Panel 0346A028



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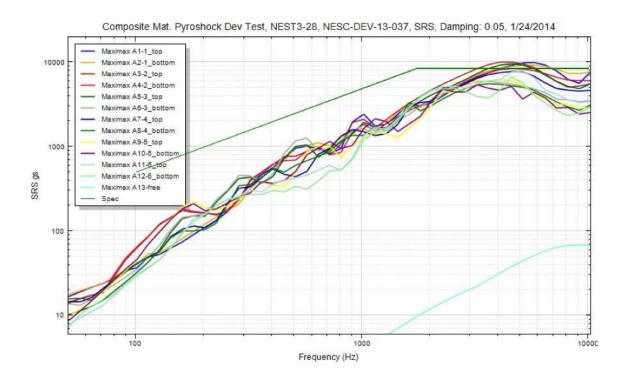
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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





NESC-RP-12-00783

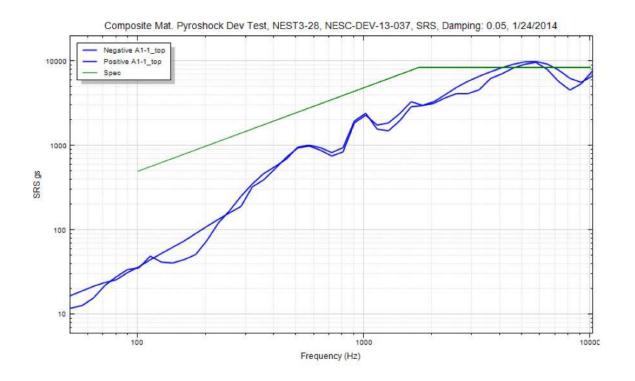
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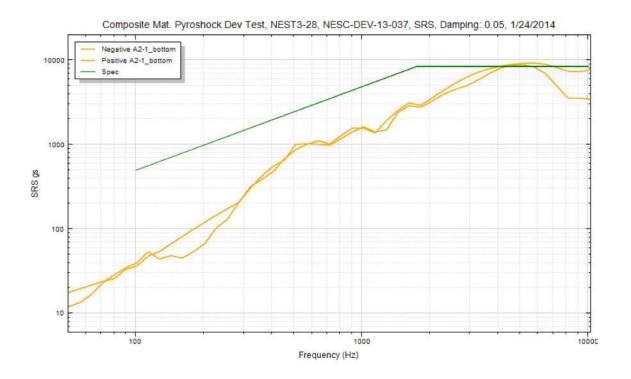
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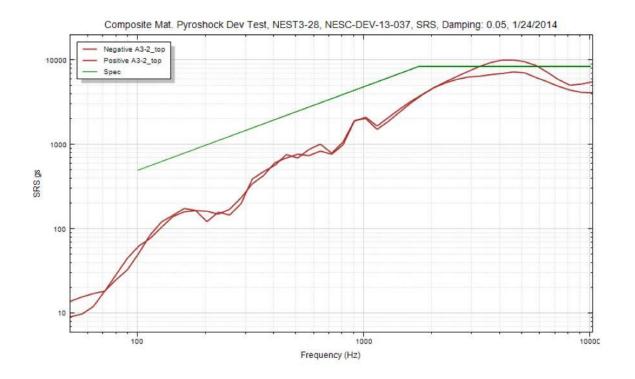
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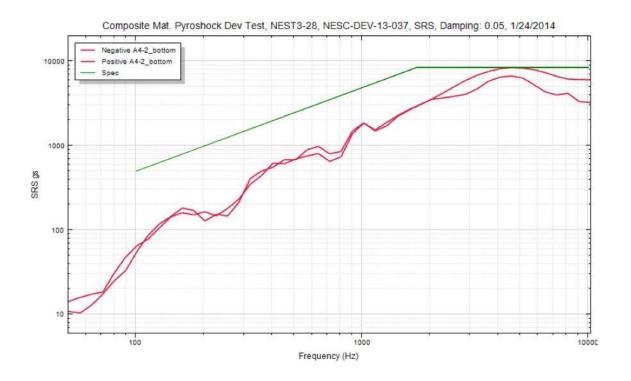
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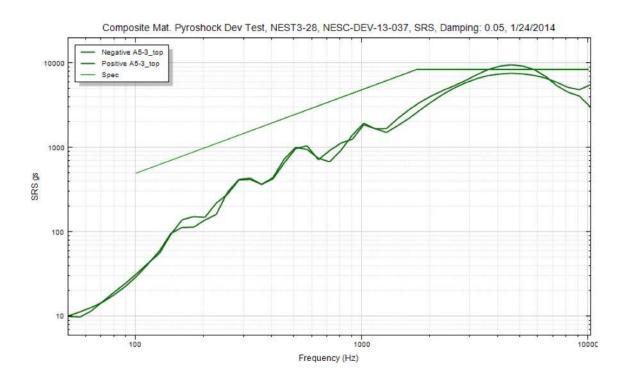
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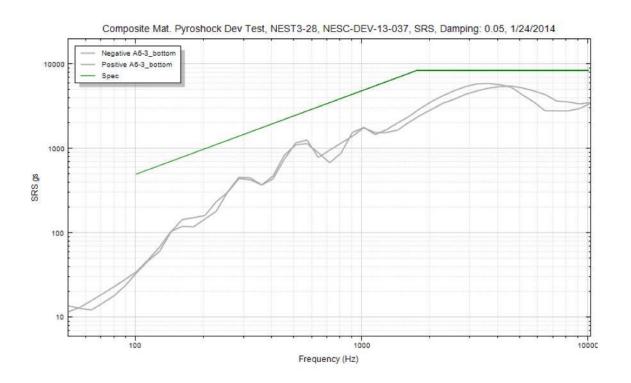
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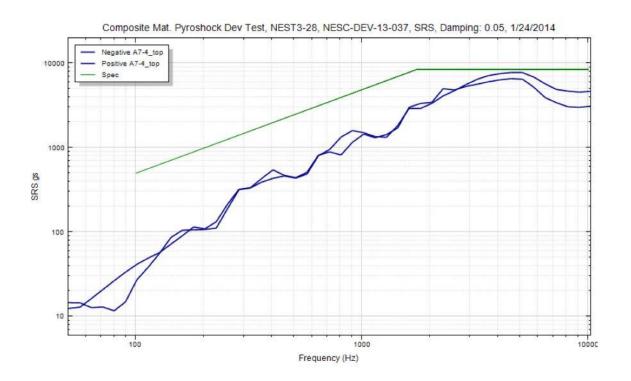
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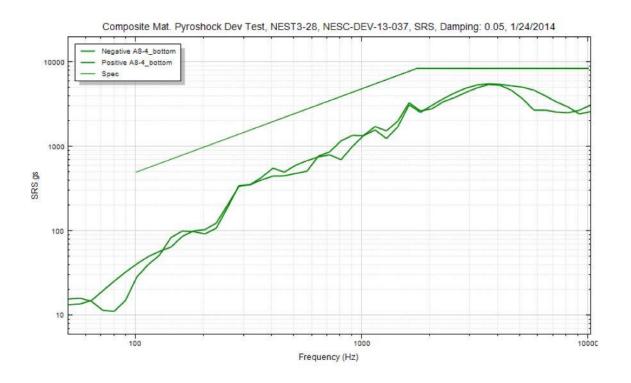
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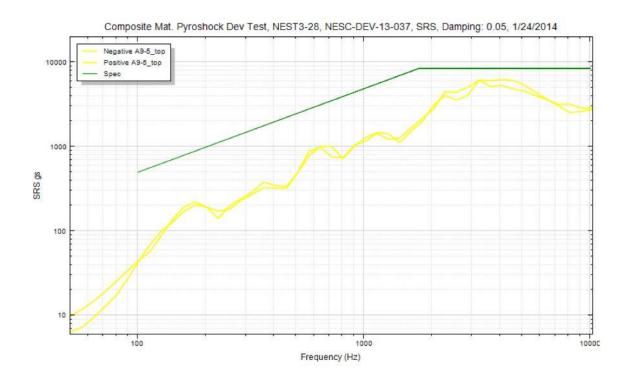
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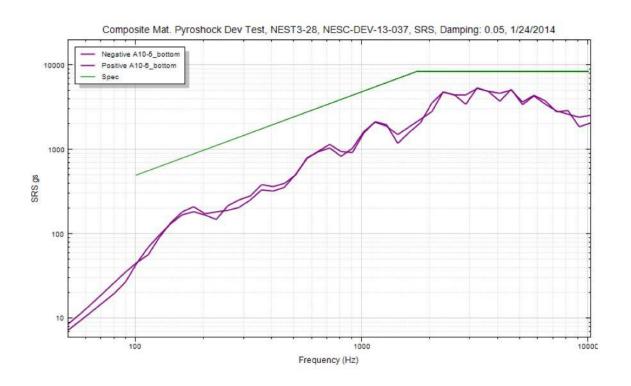
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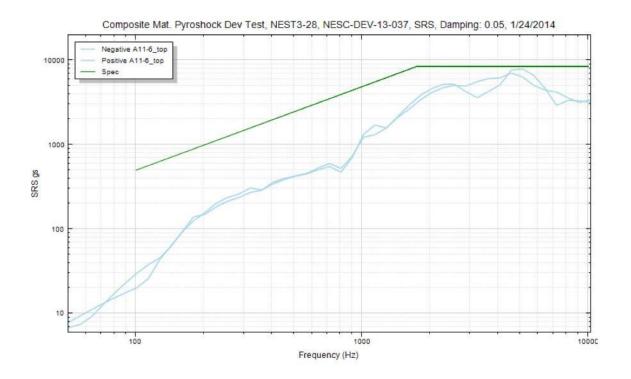
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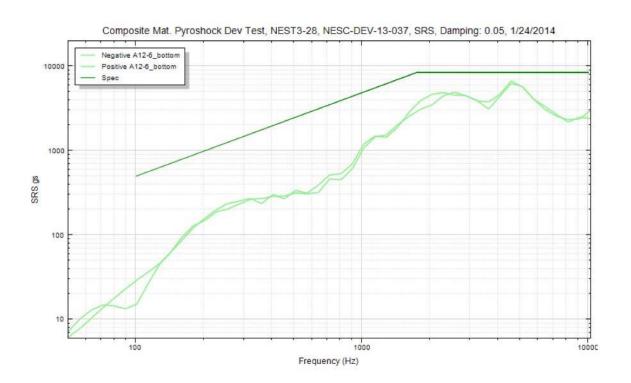
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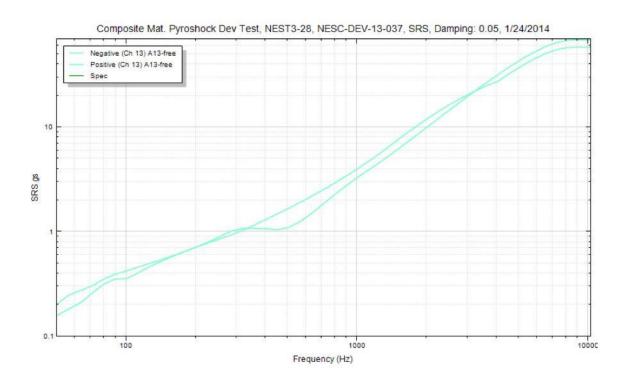
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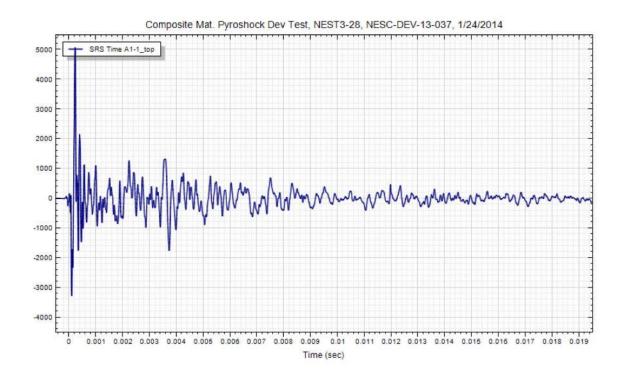
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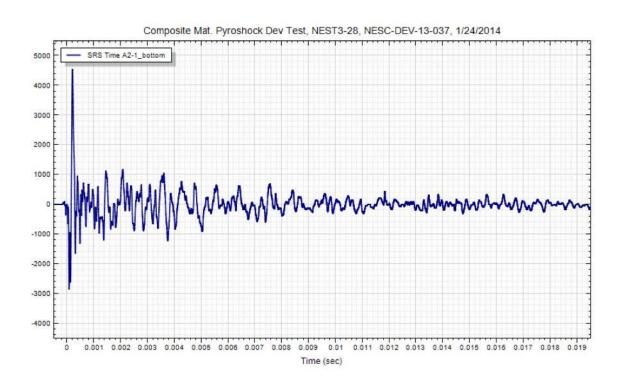
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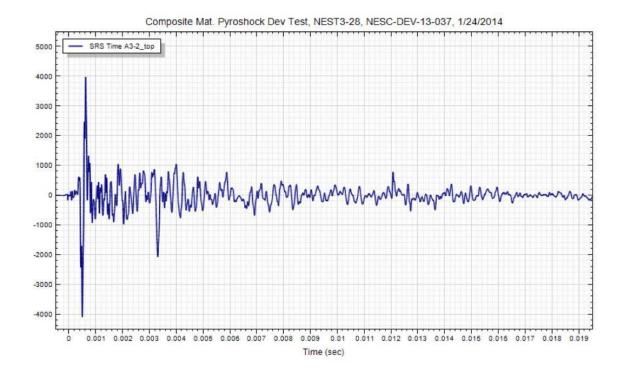
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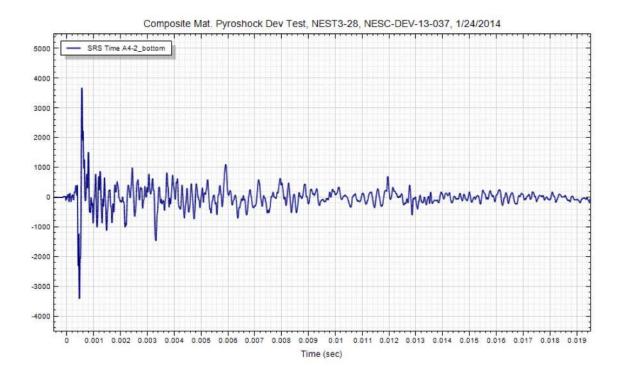
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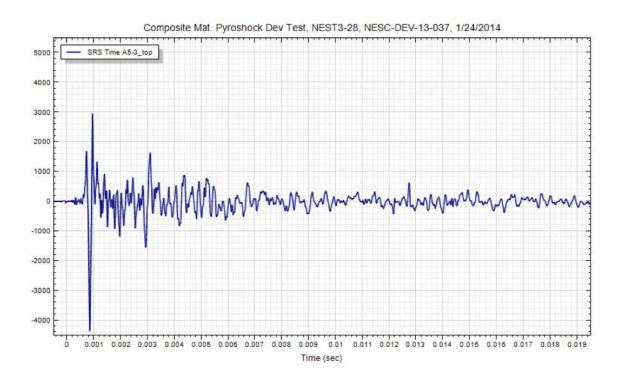
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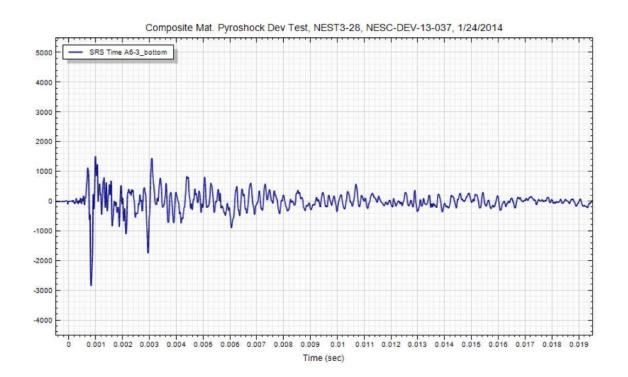
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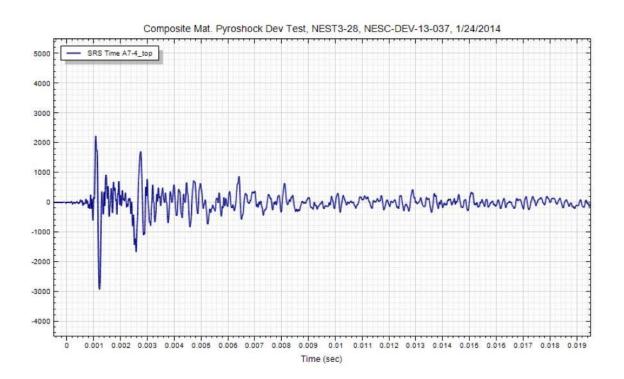
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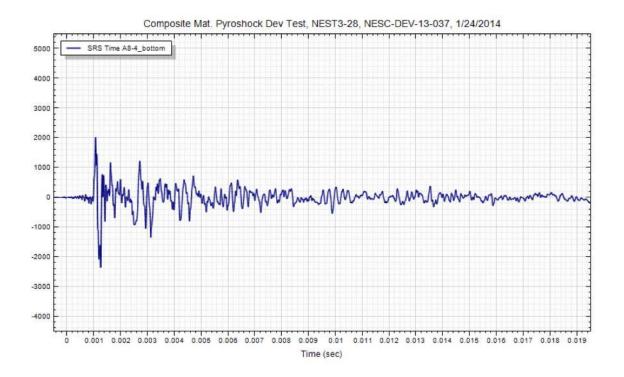
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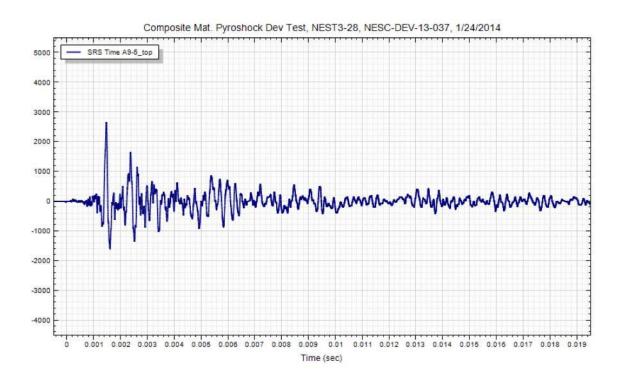
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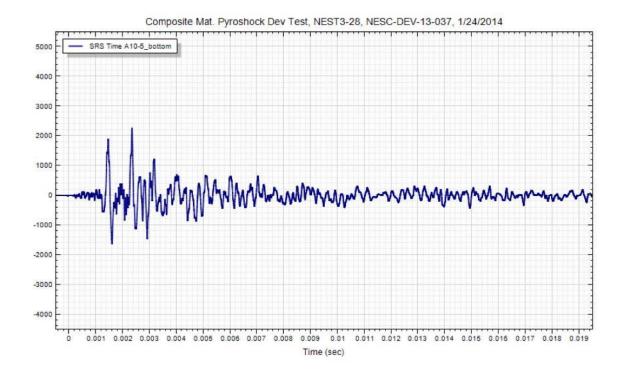
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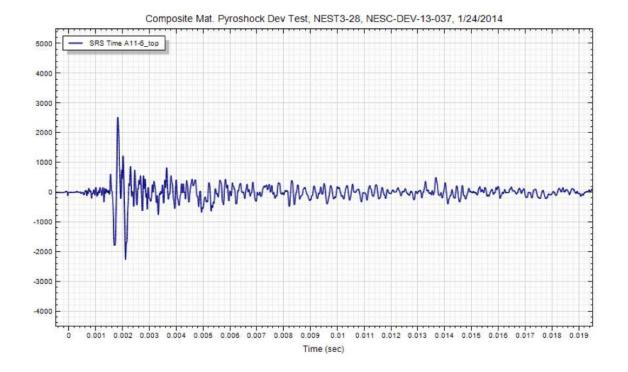
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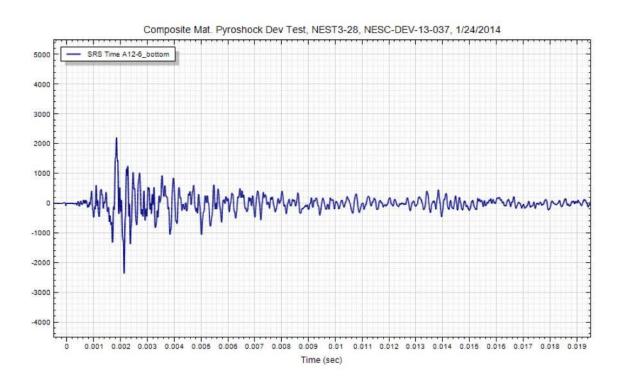
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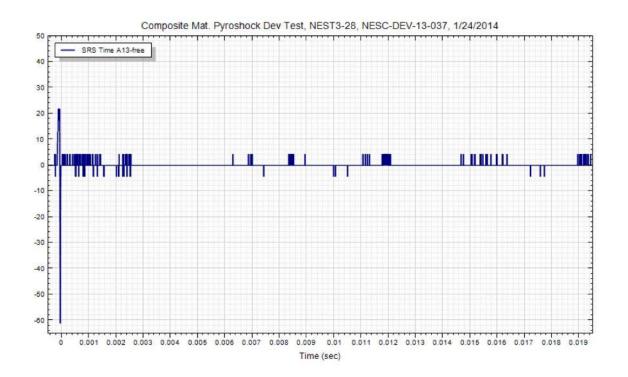
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# **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### NESC-DEV-13-037 Composite Materials Shock Test

### Test #11 Accelerometer Data Panel 0346A023



NESC-RP-12-00783

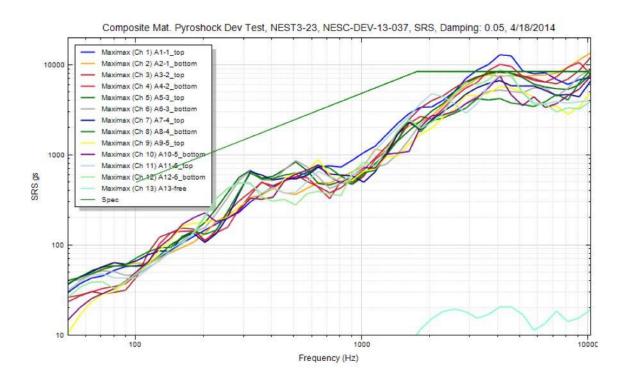
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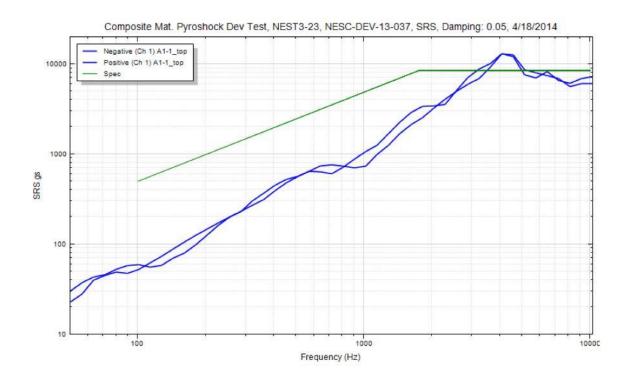
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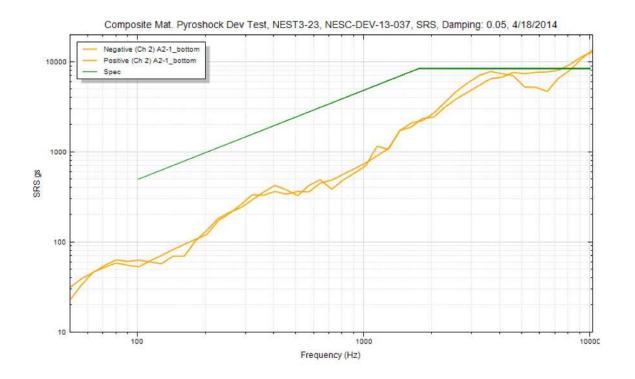
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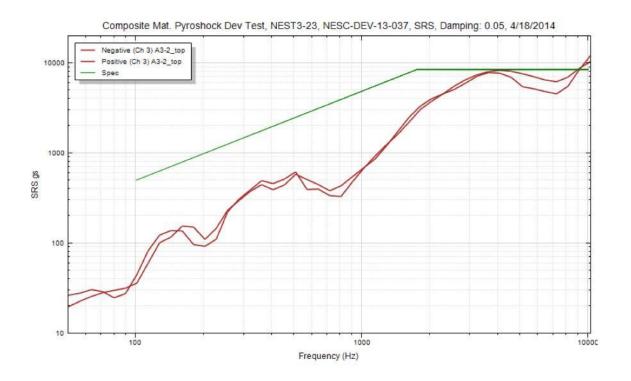
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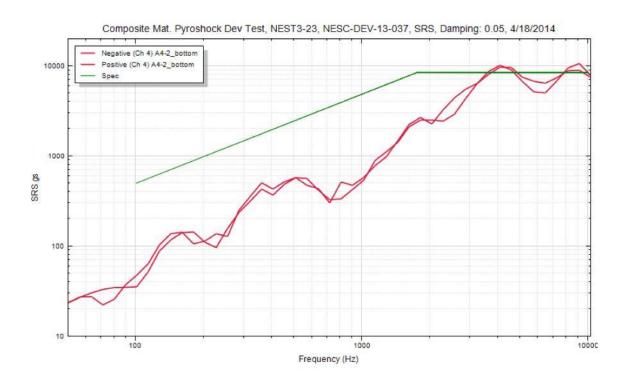
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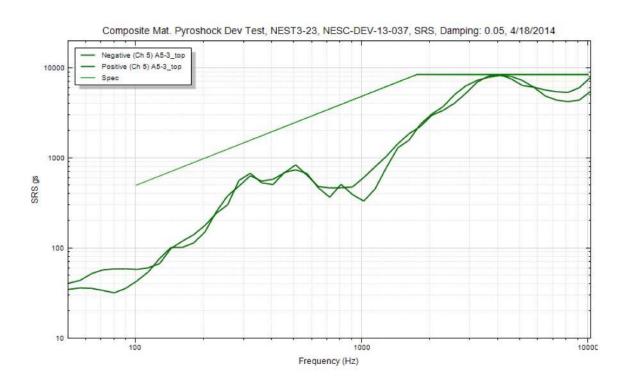
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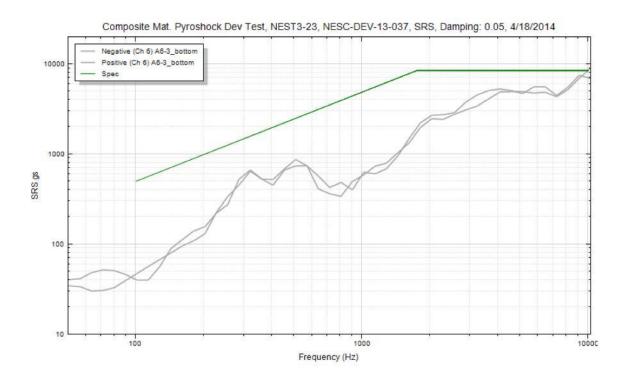
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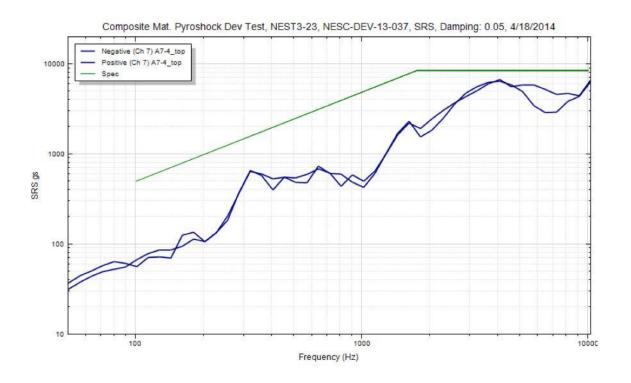
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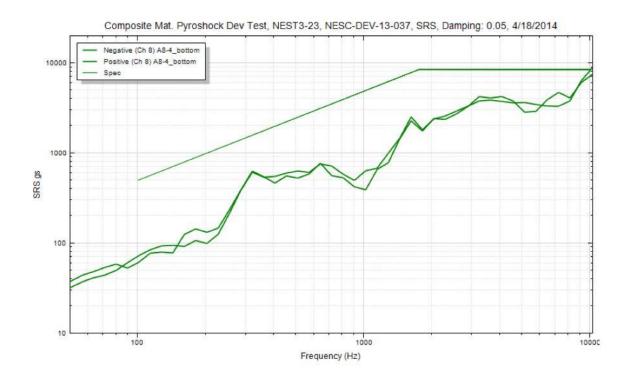
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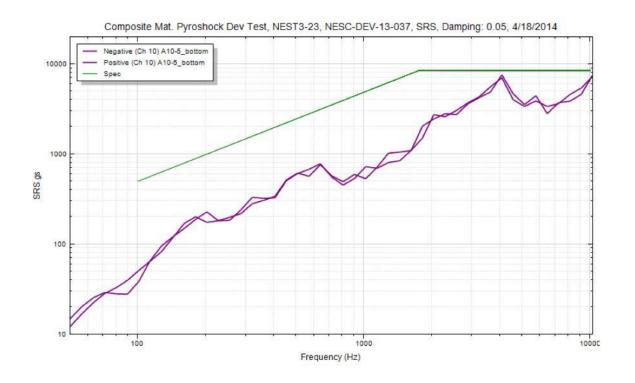
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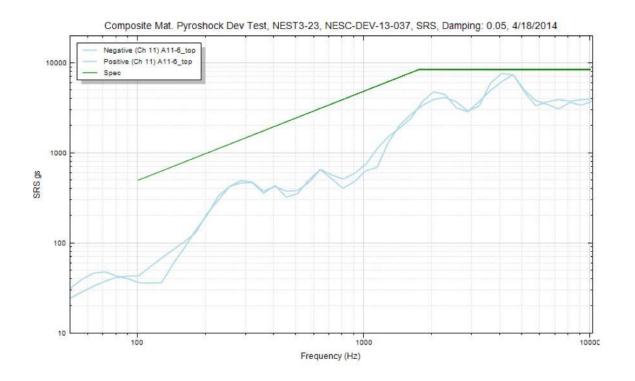
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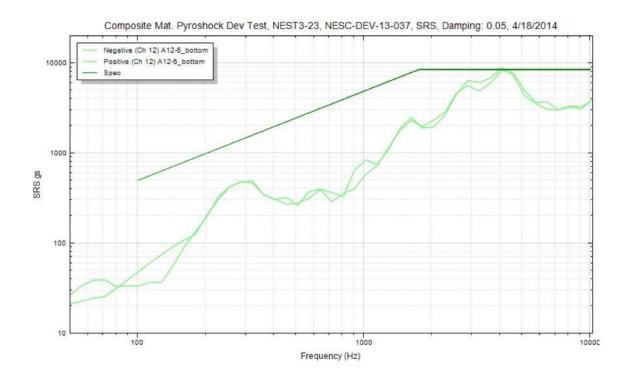
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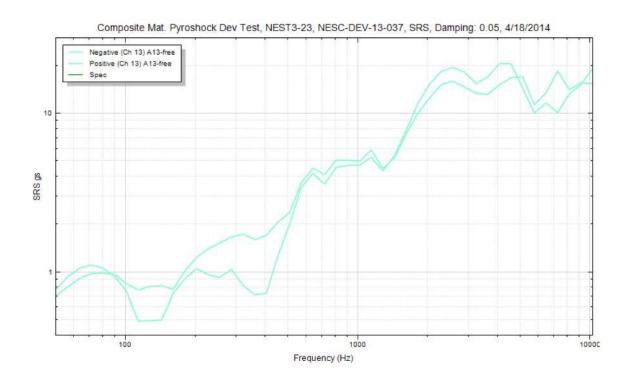
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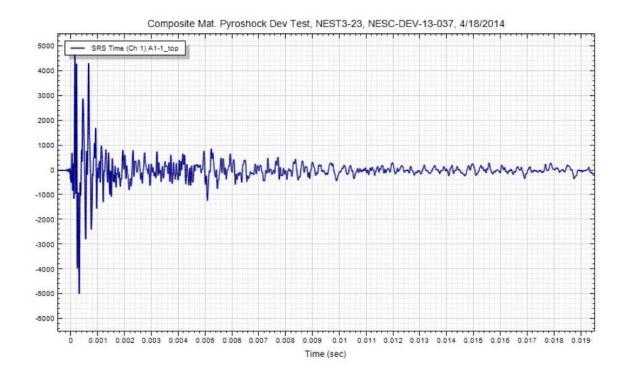
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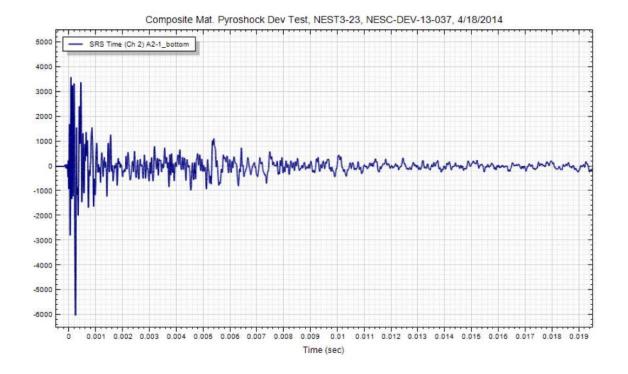
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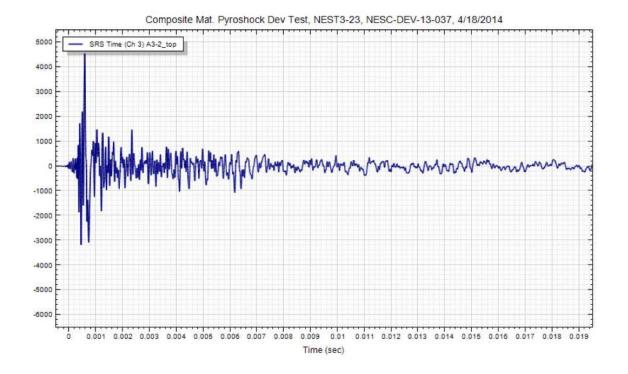
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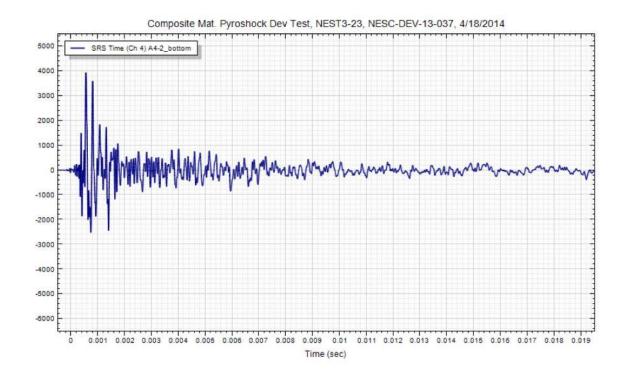
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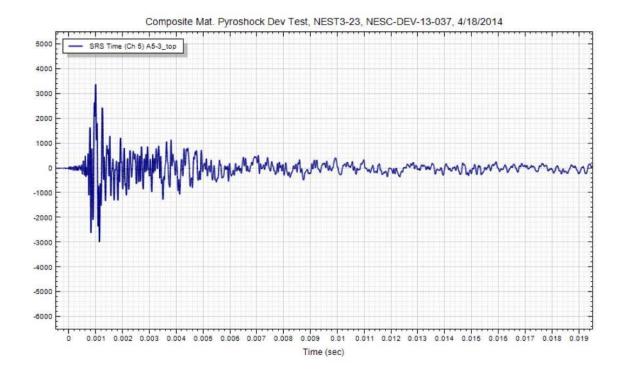
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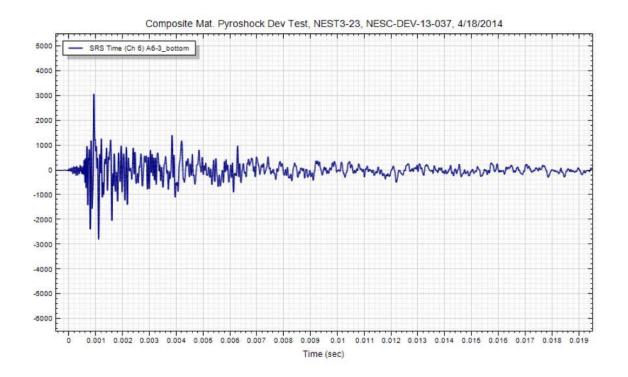
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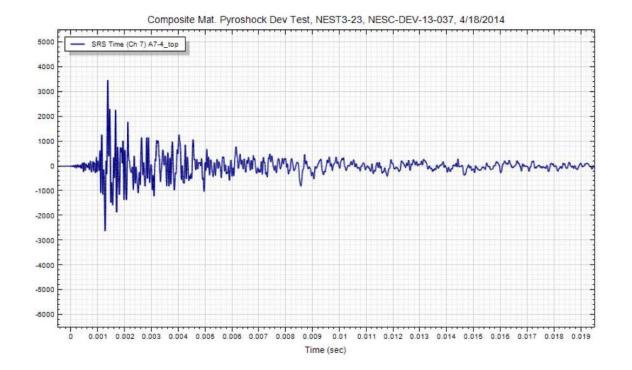
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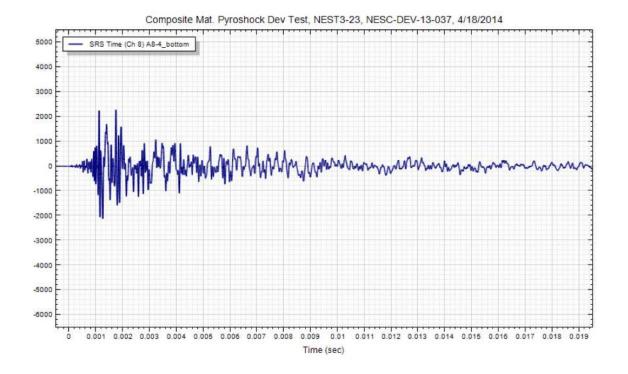
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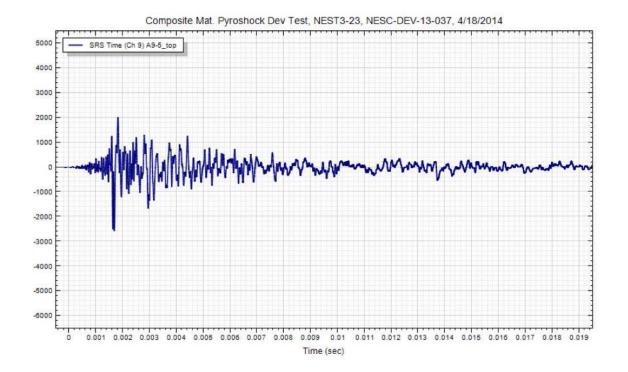
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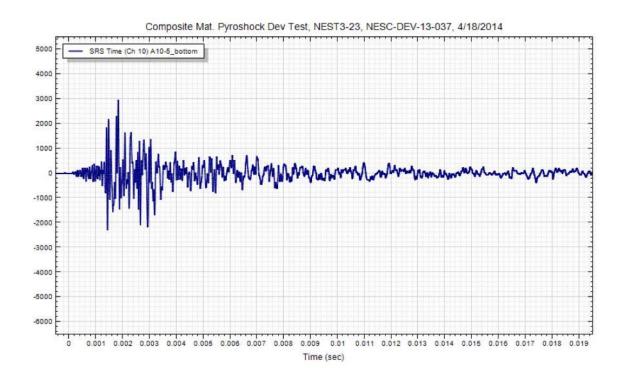
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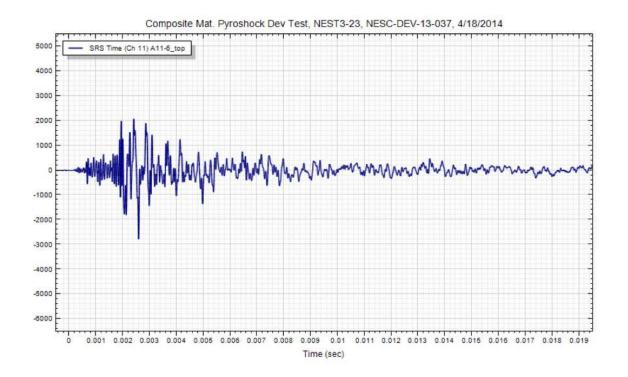
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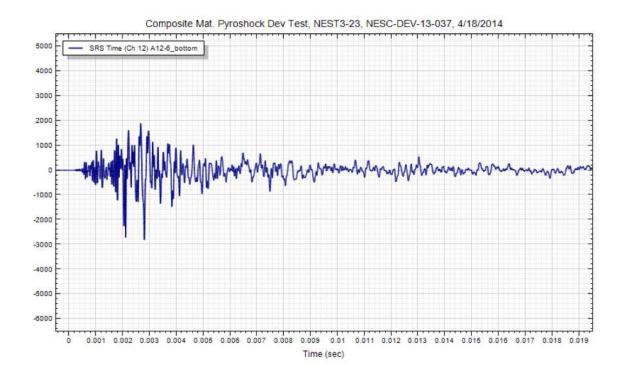
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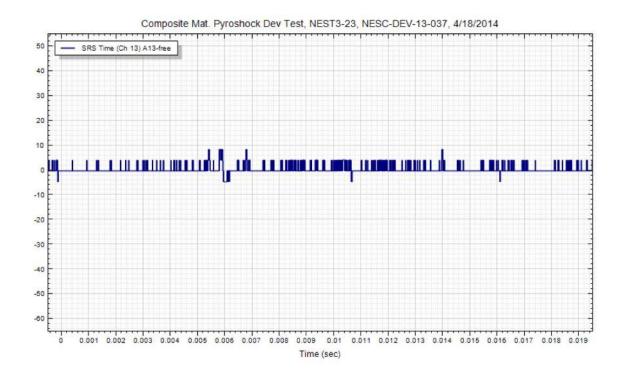
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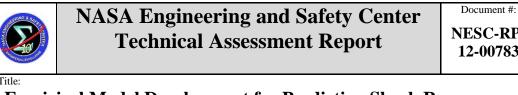
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

### **NESC-DEV-13-037 Composite Materials Shock Test**

### Test #12 Accelerometer Data Panel 0346A024



NESC-RP-12-00783

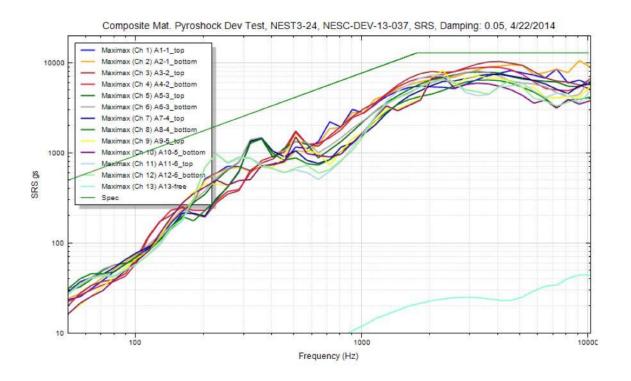
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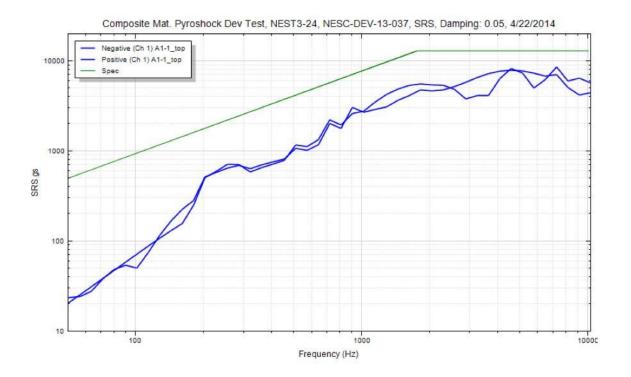
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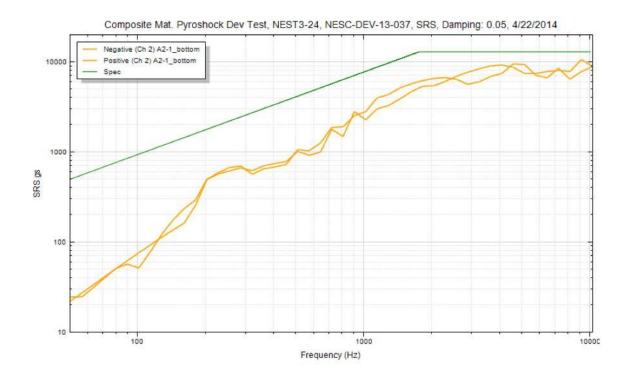
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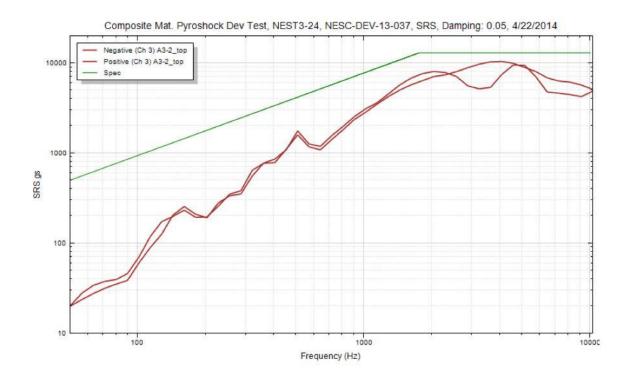
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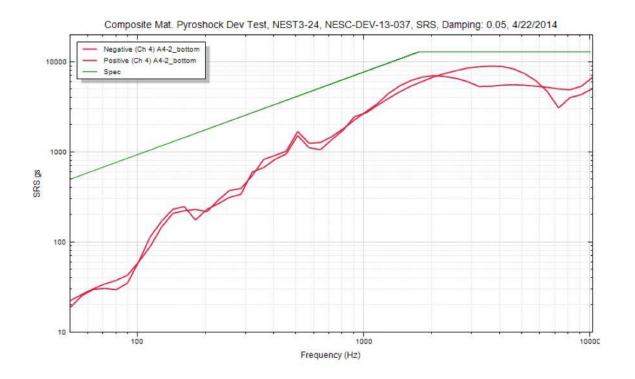
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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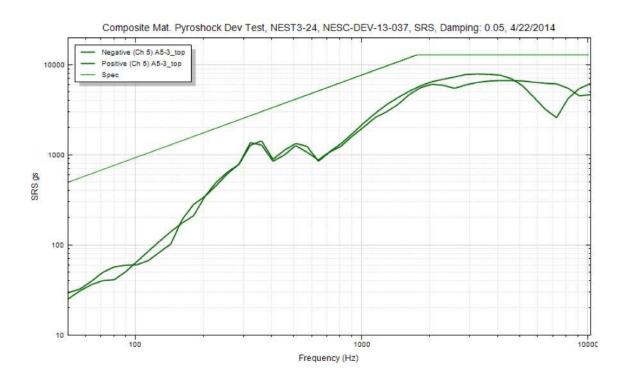
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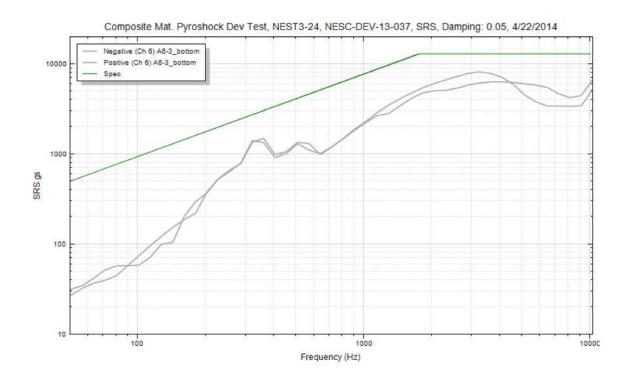
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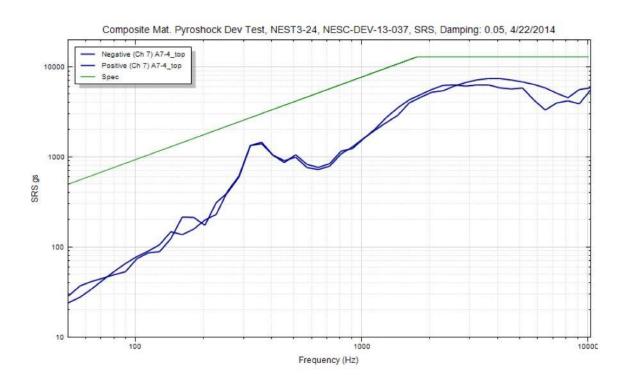
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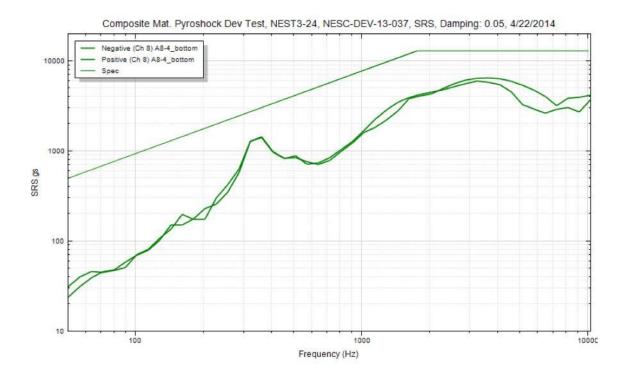
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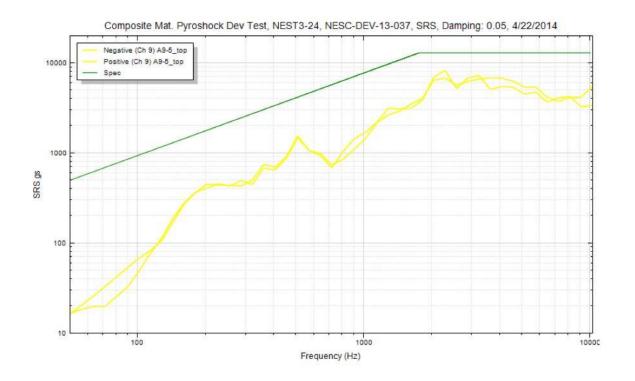
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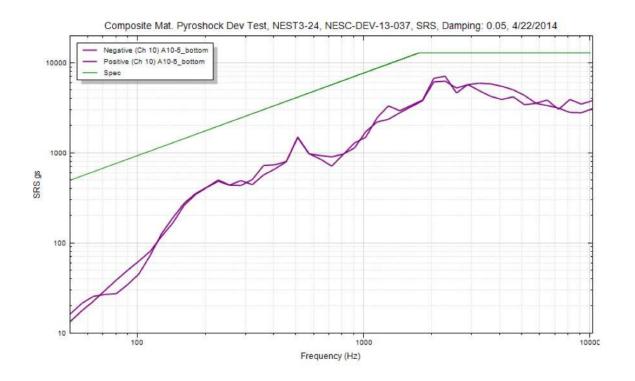
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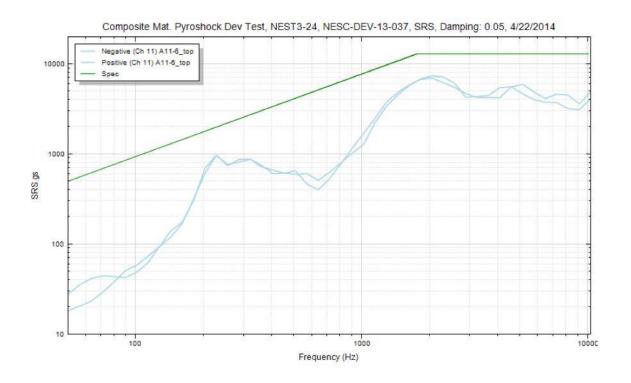
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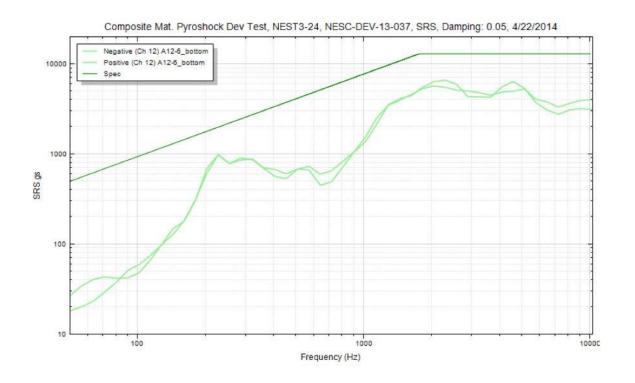
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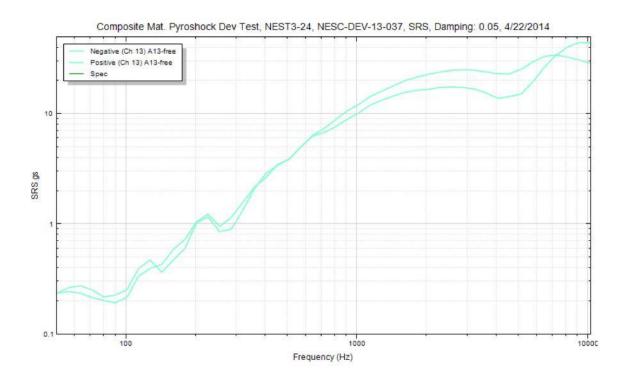
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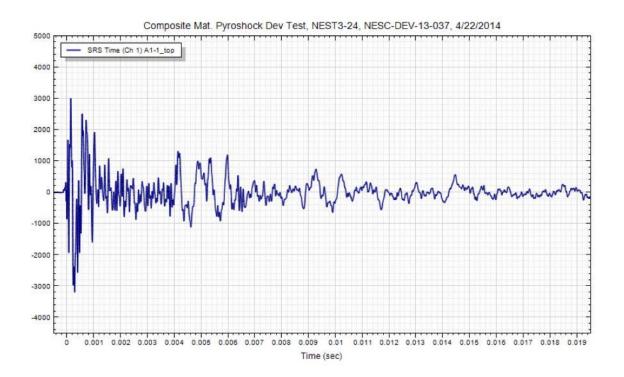
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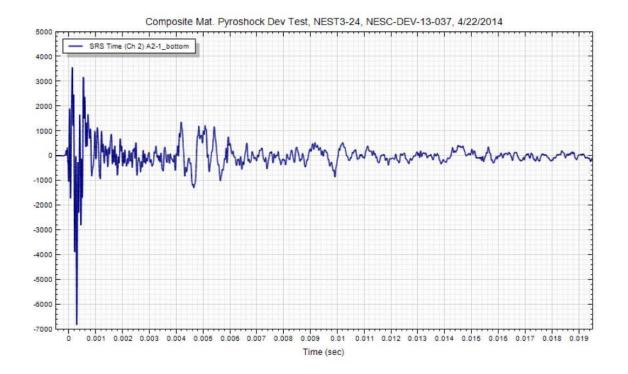
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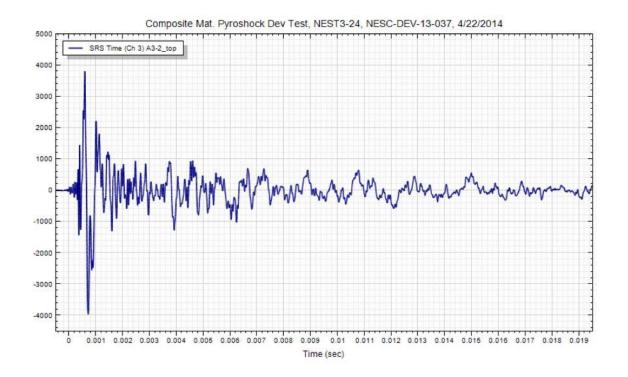
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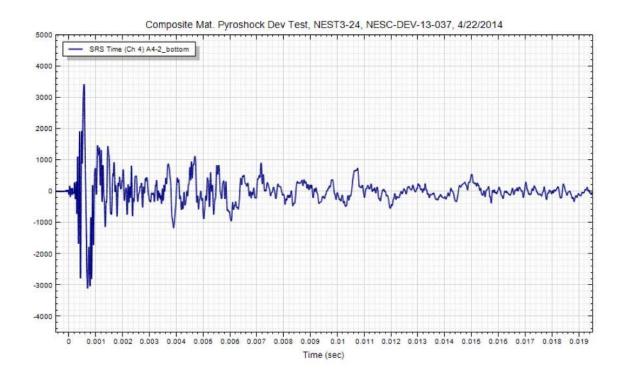
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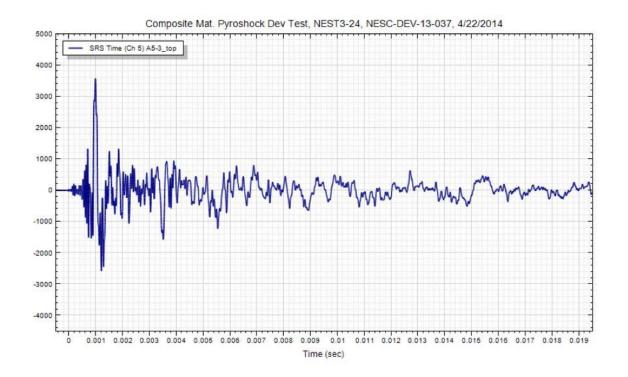
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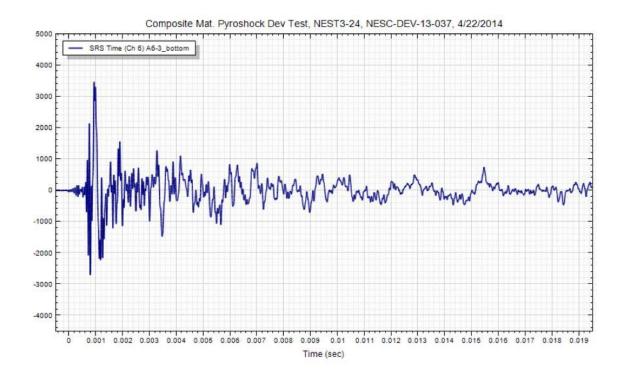
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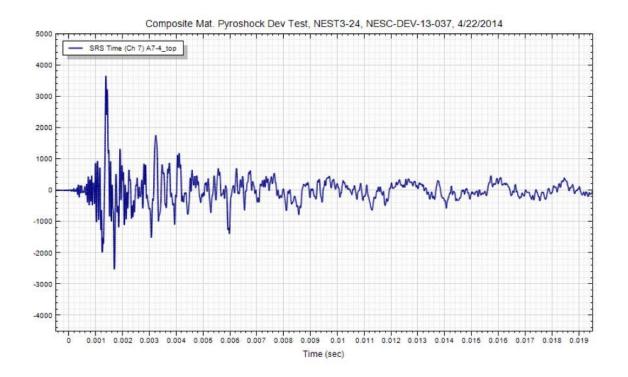
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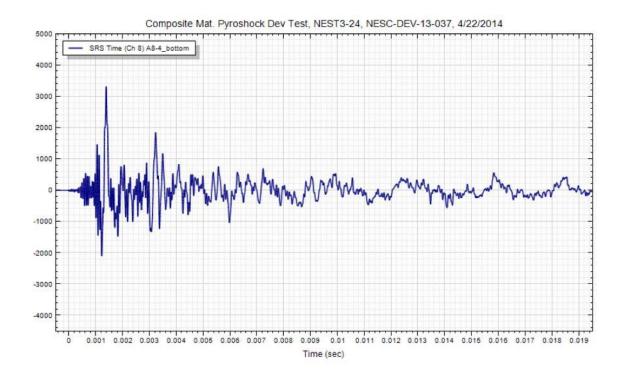
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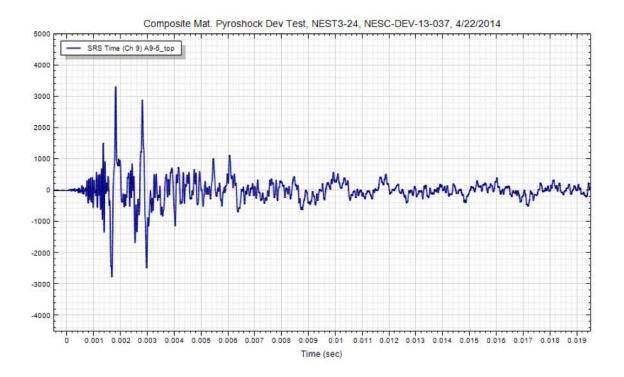
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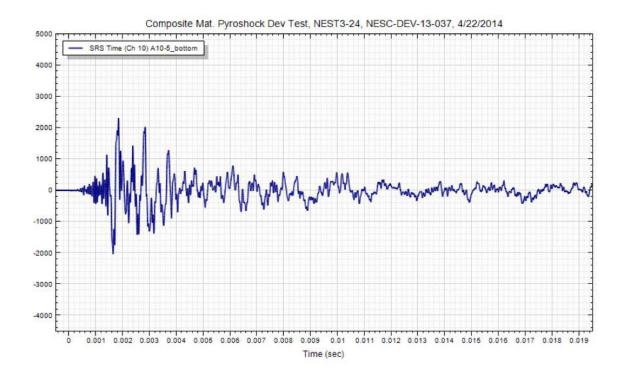
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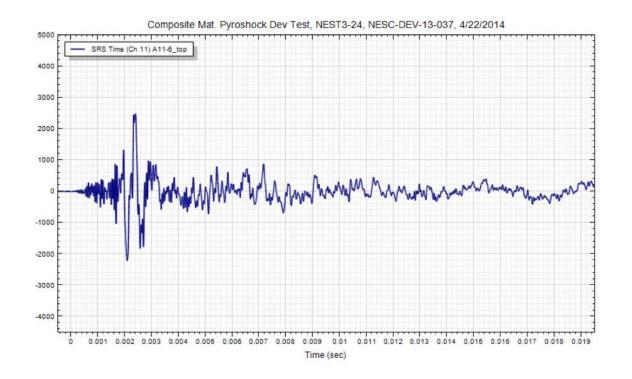
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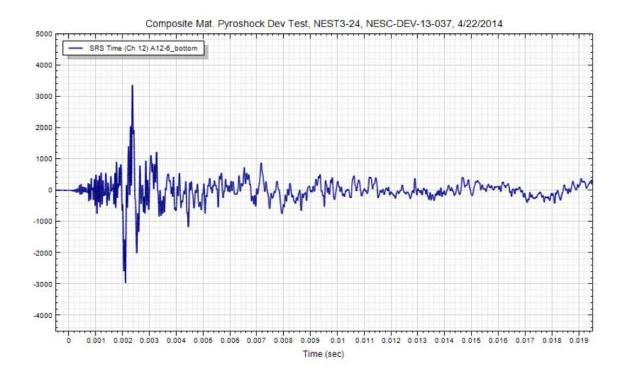
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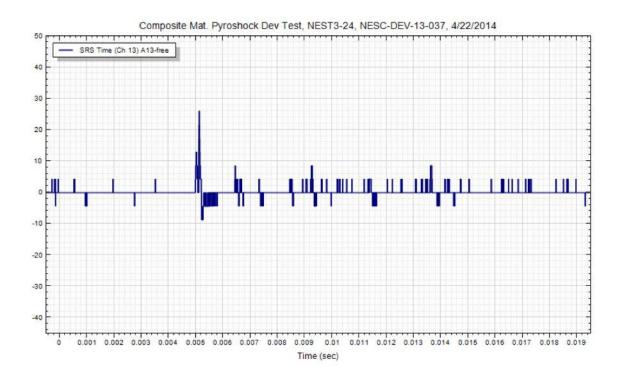
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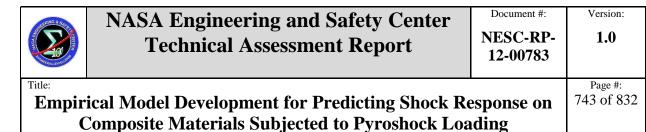
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### NESC-DEV-13-037 Composite Materials Shock Test

### Test #13 Accelerometer Data Panel 0346A025



NESC-RP-12-00783

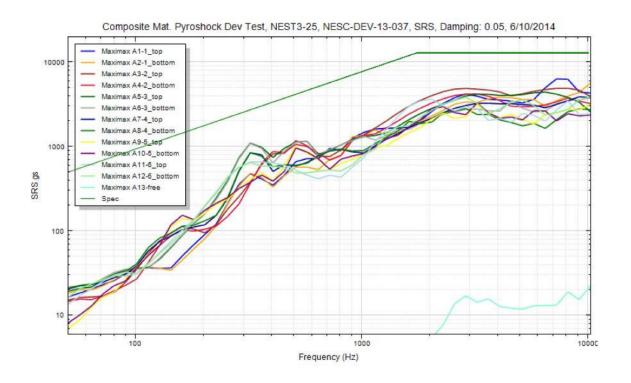
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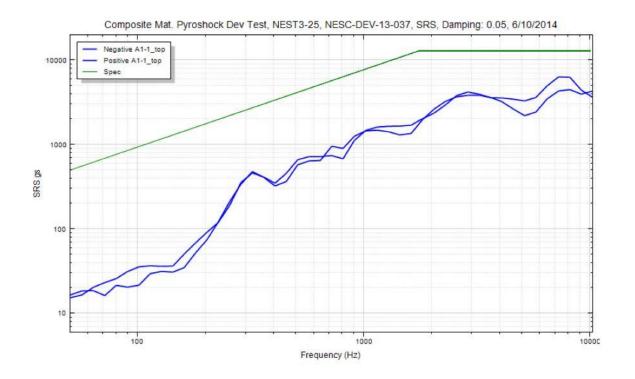
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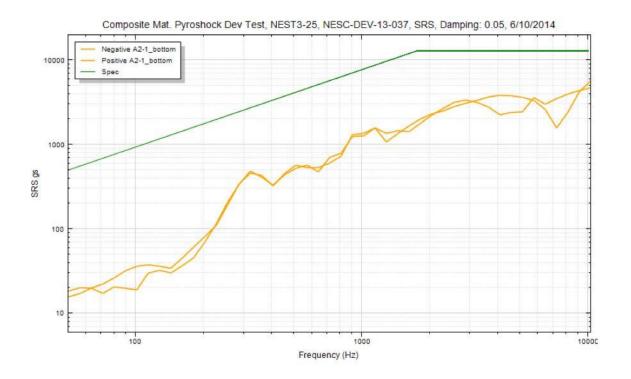
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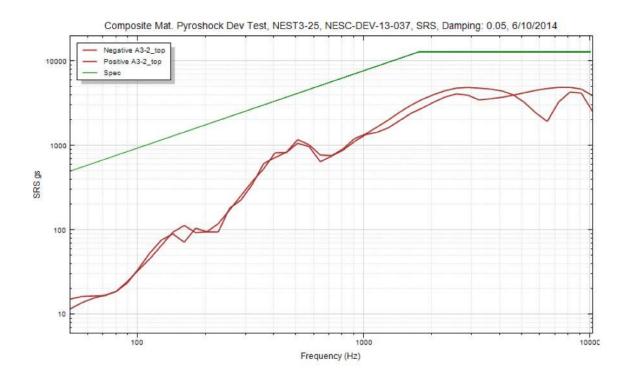
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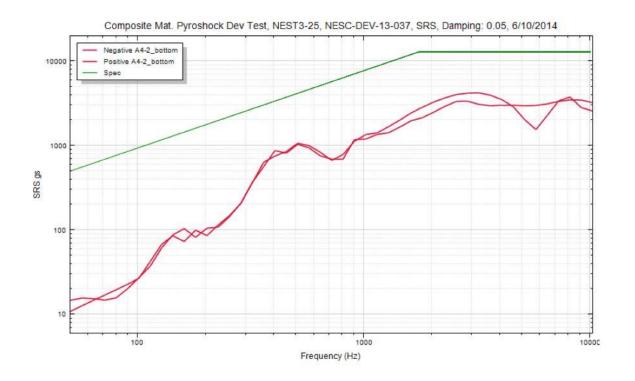
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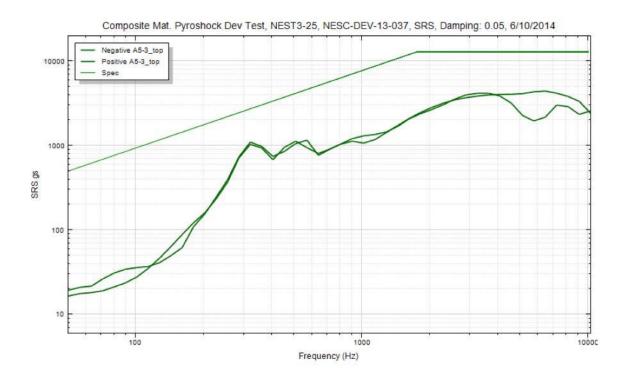
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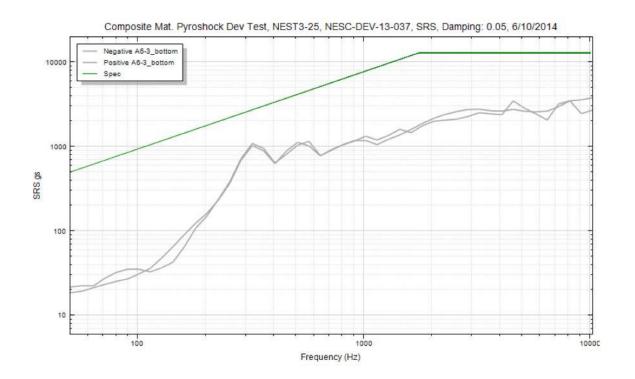
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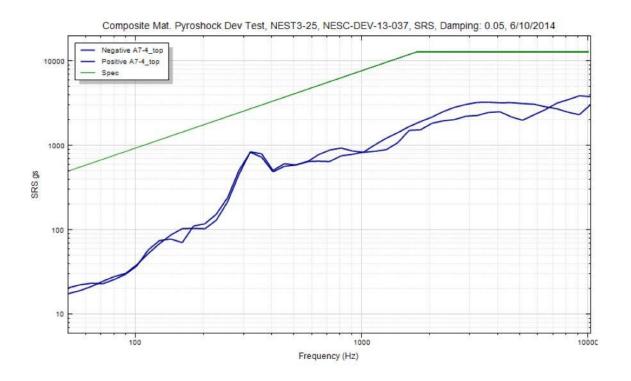
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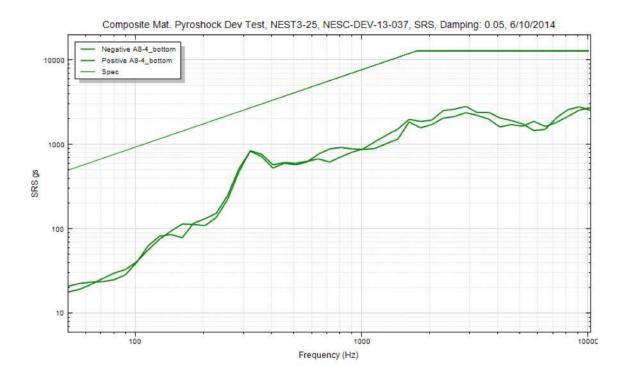
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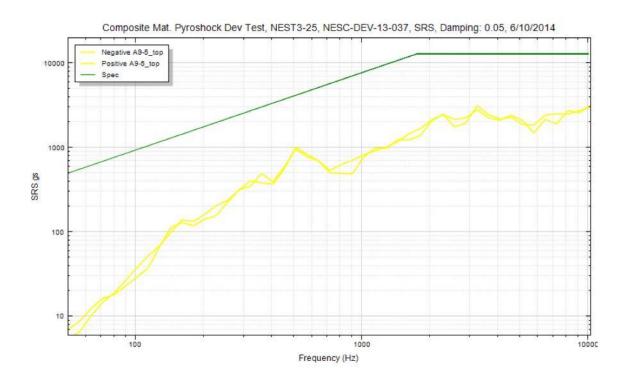
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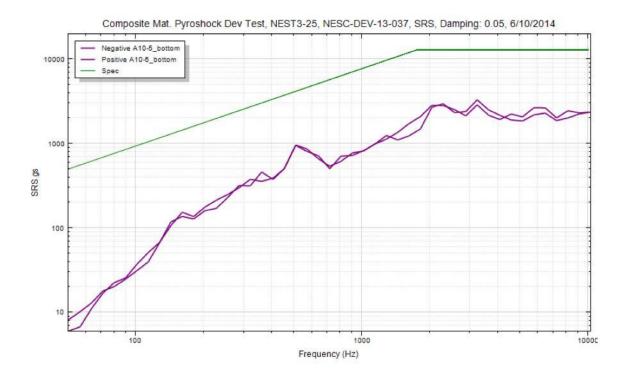
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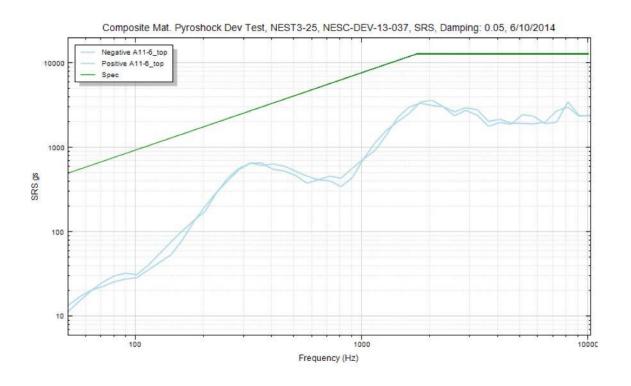
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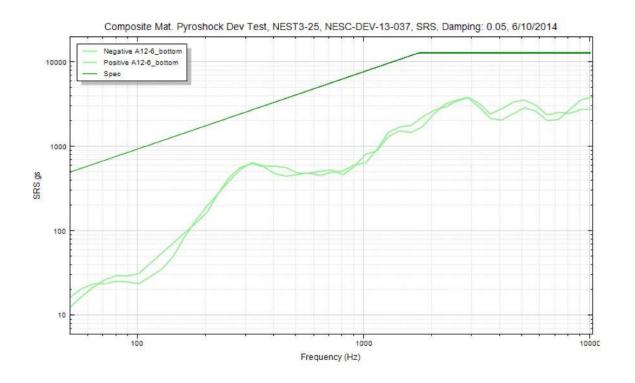
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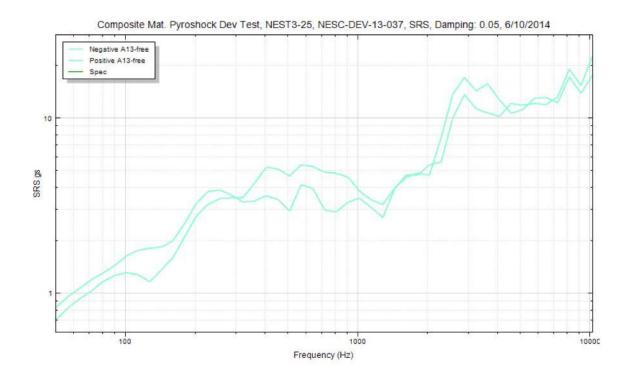
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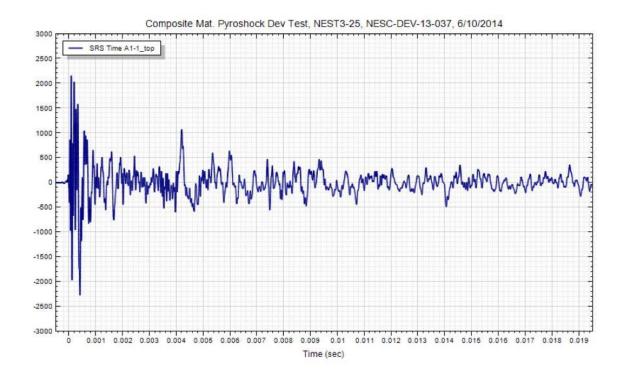
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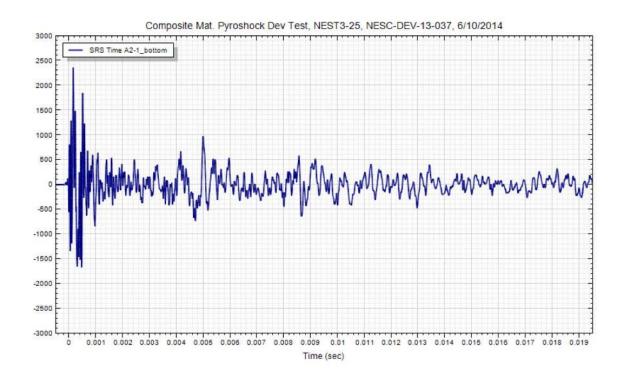
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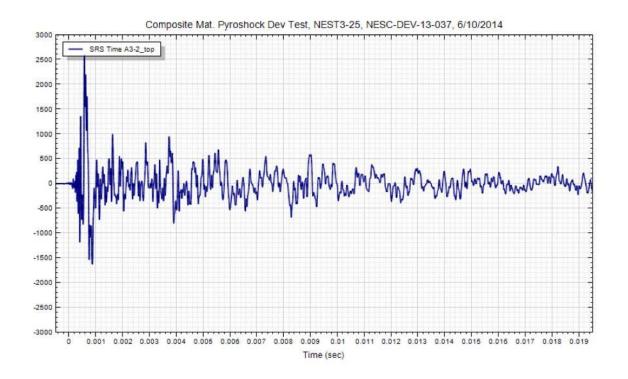
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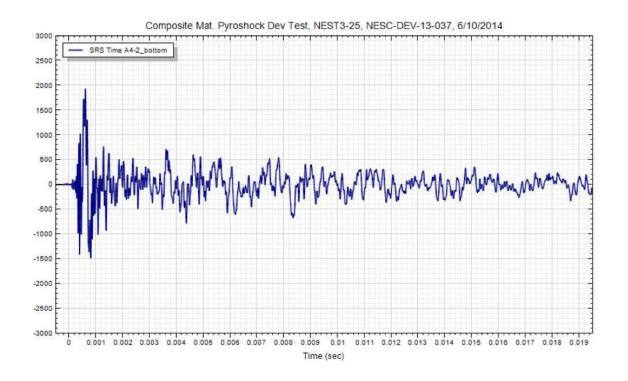
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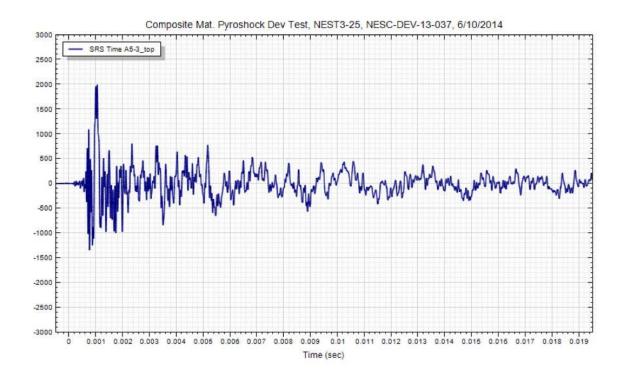
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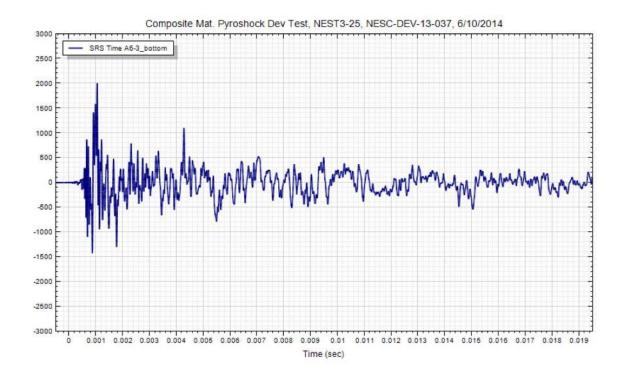
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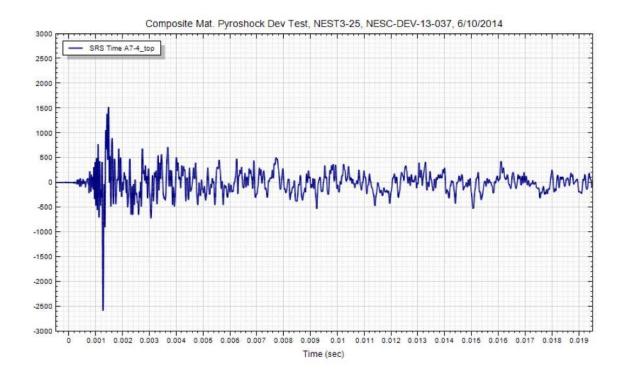
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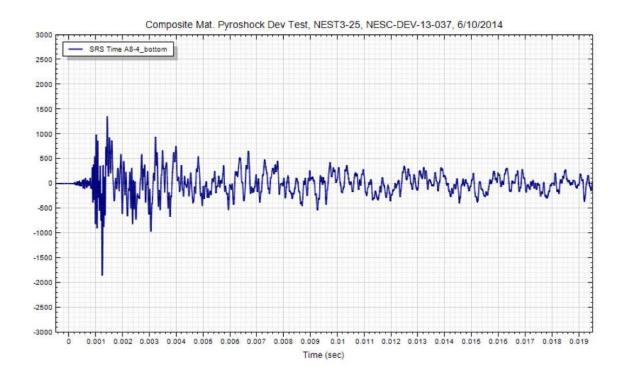
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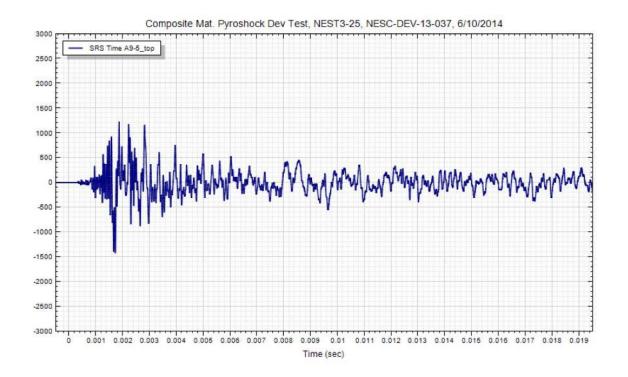
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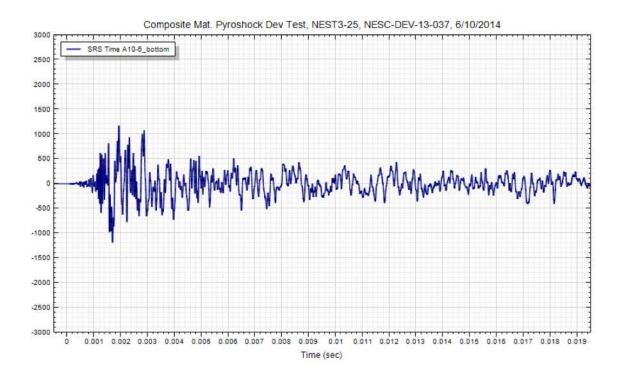
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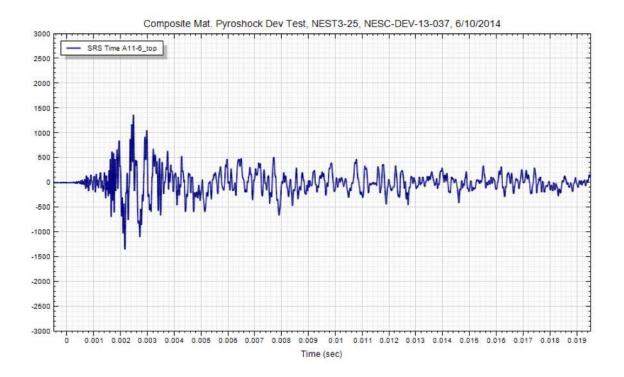
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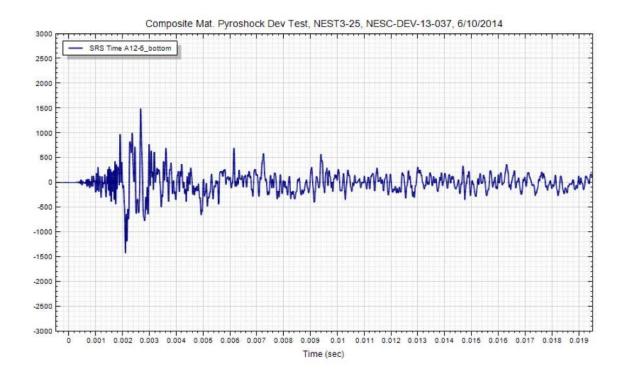
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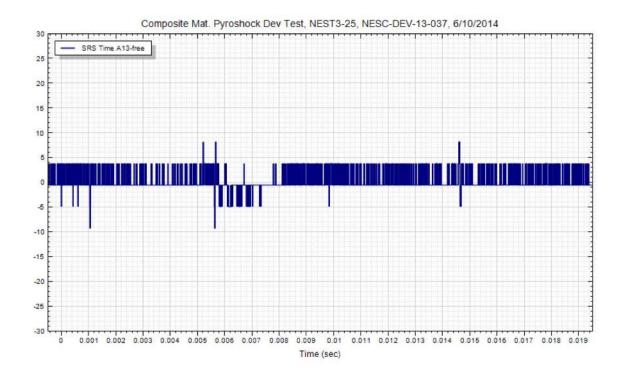
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### NESC-DEV-13-037 Composite Materials Shock Test

### Test #14 Accelerometer Data Panel 0346A027



NESC-RP-12-00783

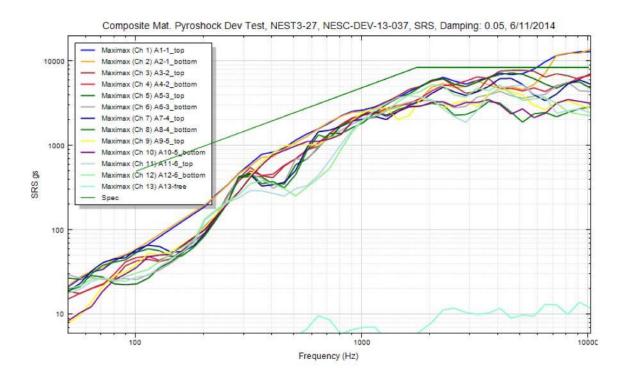
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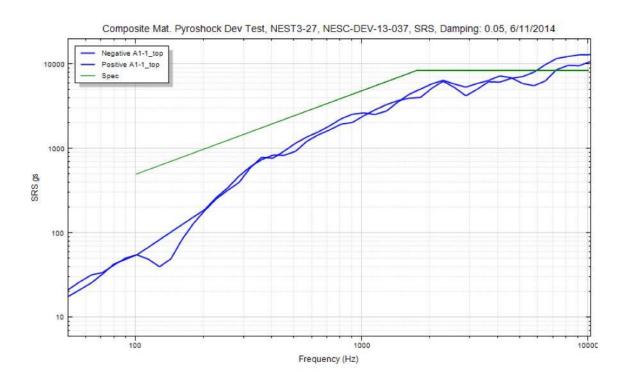
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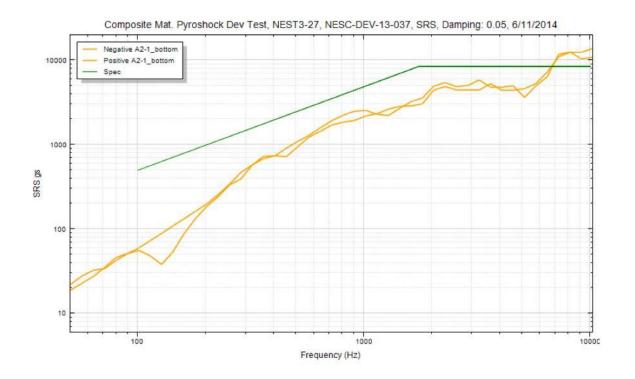
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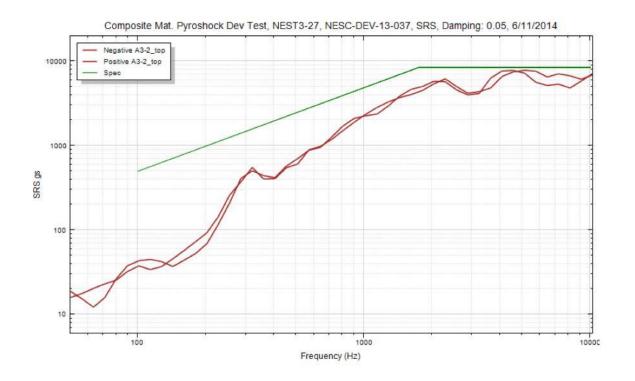
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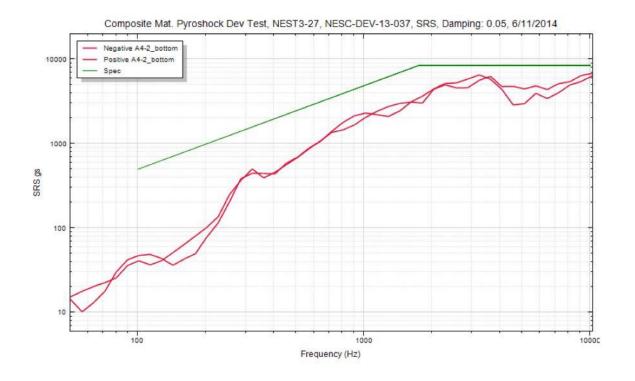
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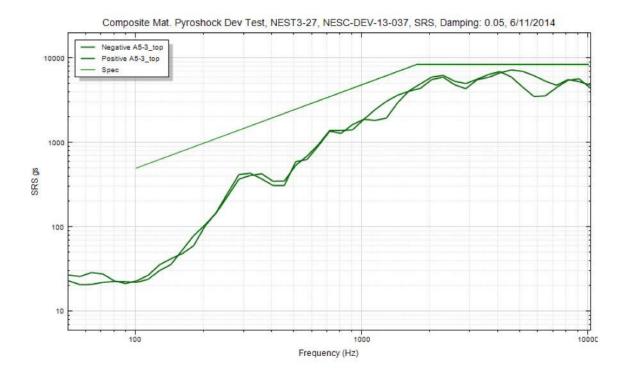
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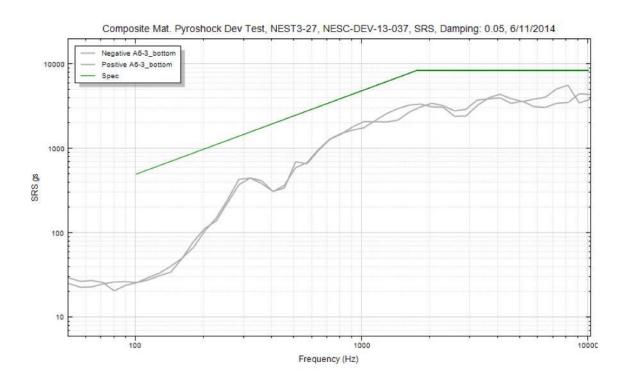
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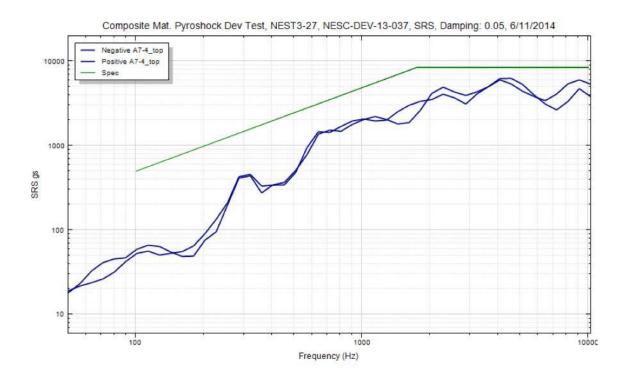
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 





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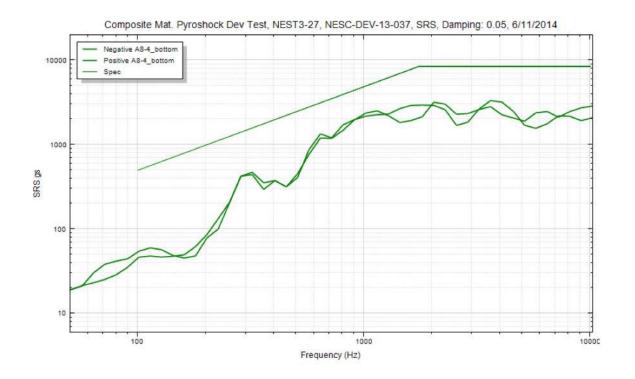
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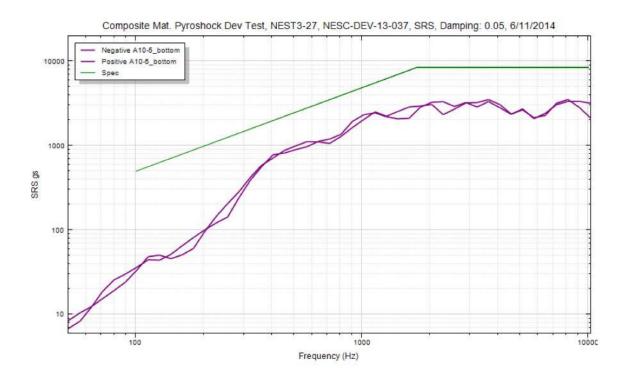
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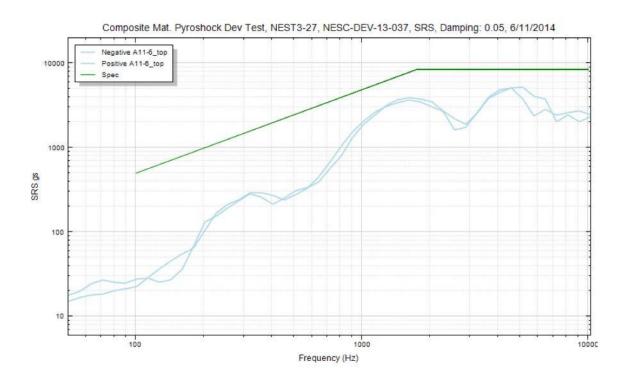
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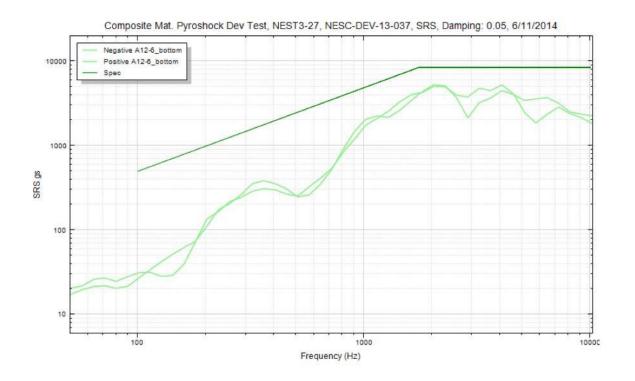
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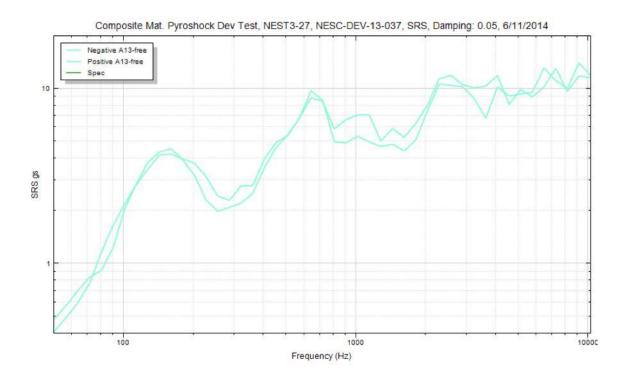
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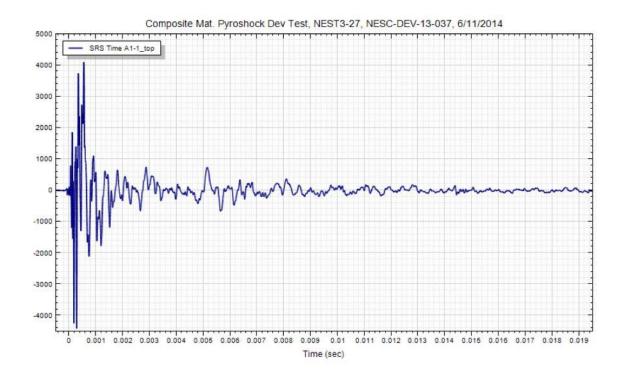
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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**





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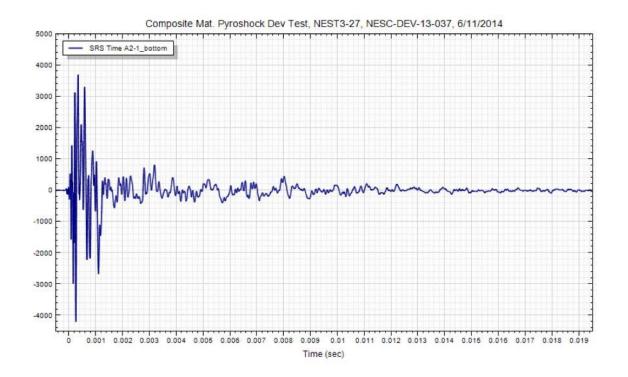
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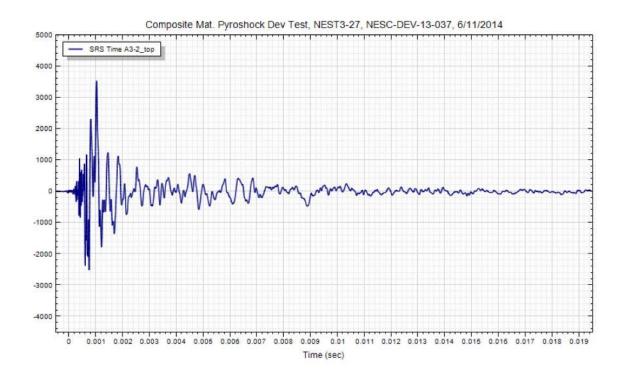
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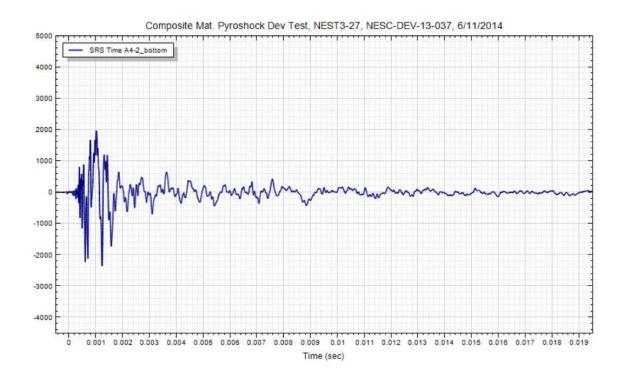
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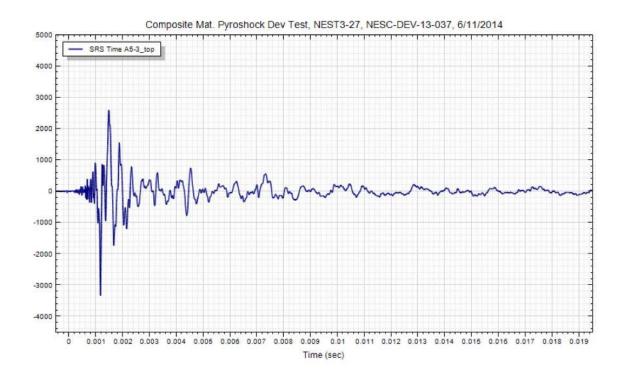
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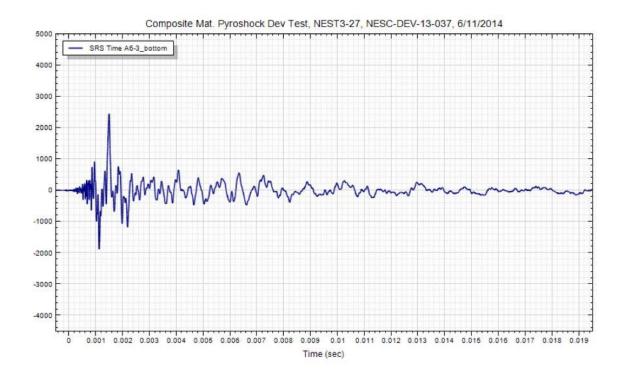
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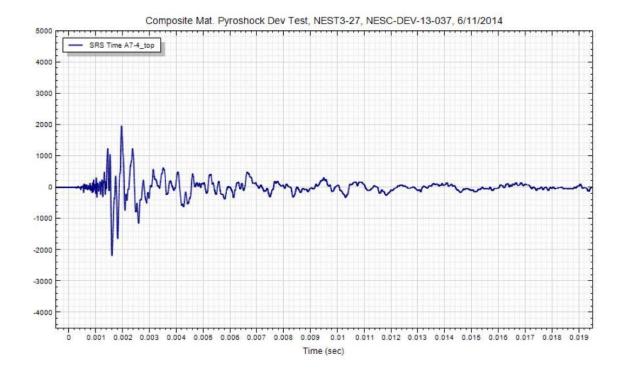
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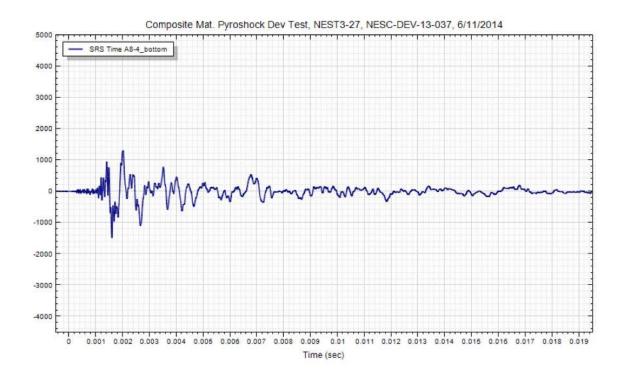
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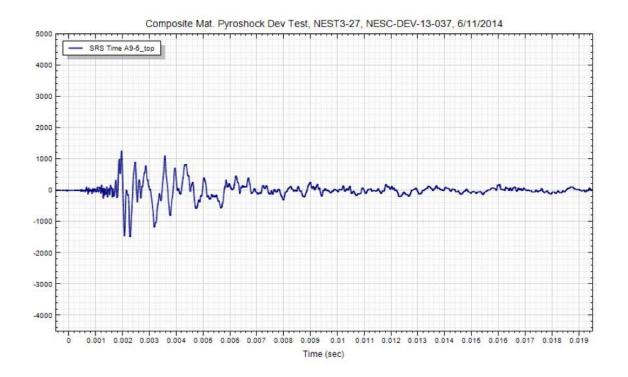
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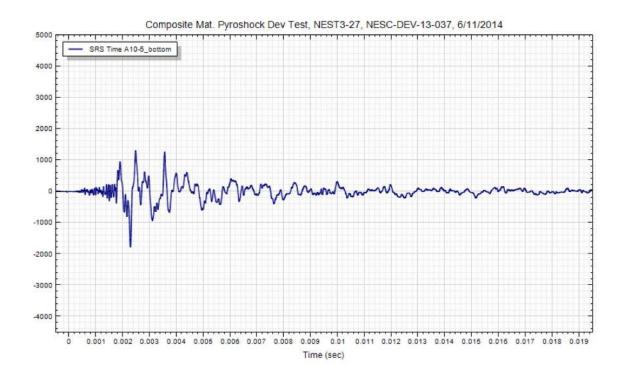
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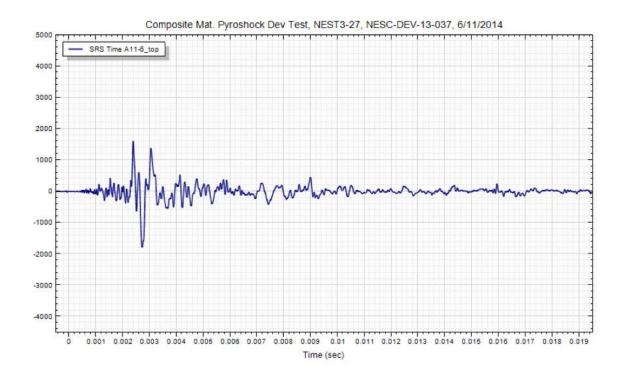
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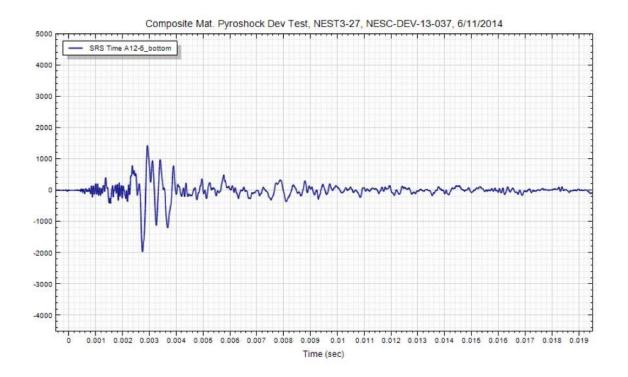
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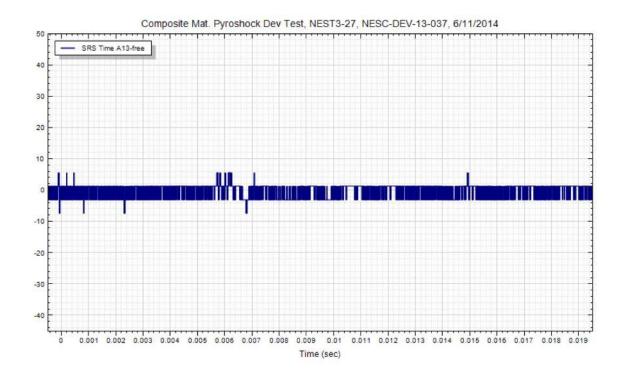
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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-037 Composite Materials Shock Test

**Test and Checkout Procedure** 



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AG RUN Record Copy

George C. Marshall Space Flight Center Marshall Space Flight Center, Alabama 35812 NESC-DEV-13-037 10/21/2013

### TEST AND CHECK-OUT PROCEDURE

ET40 / VIBRATION, ACOUSTICS, AND SHOCK TEAM

# COMPOSITE MATERIALS PYROSHOCK DEVELOPMENT TEST

Group III – Tests 19 to 28 Group I – Retests 1 to 4

This Procedure Describes Safety Critical Operations

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	on, Acoustics, and Shock Te	Am
Composite Materials Pyrosbock Development Test	NESC-DEV-13-037	Revision: Baseline
Group III - Tests 19 to 28, Group I - Retests I to 4	Date: 10/21/2013	Page I of 24
PREPARED BY: 1 Jhu Cain Sun John Craig Garrison / E	, , , , , , , , , , , , , , , , , , ,	10/29/2013 Daie
Test Engineer	140	Daie
APPROVED BY: Kathy O. fwofford / Kathy O. fwofford / Deputy Branch Chief Structural Dynamics Te	Mond 40 st Branch	10/29/1- Date
APPROVED BY		
David Ordway / EV32 Aerospace Engineer, Py Structural & Mechanica Test Requester		Date
PPROVED BY: David Barrons		2013-10-23

Mechanical, Thermal and Life Support Branch

David Parsons / ES22

Structural Dynamics



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

Composite Materials	Pyroshock Development Test	on, Acoustics, and Shock Team NESC-DEV-13-037	Revision; Baselin
	a 28, Group I - Retests 1 to 4	Date: 10/21/2013	Page ! of 24
PREPARED BY:			
	John Craig Garrison / E Test Engineer	T40	Date
APPROVED BY:	Kathy O. Swofford / ET Deputy Branch Chief Structural Dynamics Te		Date
APPROVED BY:	DAVID ORDWA	Digitally signed by DAVID ORDIVAY  DN: C-US, 0-US, Government, ou =NASA, otl=Postler, on=DAVID ORDIVAY, 45: 2342-19700100, 100, 1.1 =docdway Date, 2013, 102, 15: 25: 26: 0-5007	10/22/2013
	David Ordway / EV32 Acrospace Engineer, Py Structural & Mechanica Test Requester		Date

Mechanical, Thermal and Life Support Branch

David Parsons / ES22

Structural Dynamics



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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Composite Materials Pyroshock Development Test	NESC-DEV-13-037	Revision: Baseline
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#### 1.0 INTRODUCTION

#### 1.1 PURPOSE

The purpose of this procedure is to define the steps necessary to perform a pyrotechnic shock test in the Pyrotechnic Shock Facility in Building 4619 using pyrotechnic devices.

Test Matrix Test Articles: <u>Ten 3'x6' Sandwich composite panels with LSC plate and LSC backing plate</u>. <u>There are also 4 retest panels from Group I, tests 10, 4, 2 and 9 that are 3'x6' solid composite panels with LSC plate and LSC backing plate</u>. <u>The 2 test matrix are described in test plan, Table IV, Group III, tests 19 to 28 and Table VI, Group I retests.</u>

Program: NESC Type of Test: Pyrotechnic Shock Development Test
Test Purpose: To capture the acceleration time histories for group III – Test 19 to 28 and 4 retests from the group I test series.

The Pyrotechnic Shock Facility is located in Rooms 170, 170A and 170B of Building 4619. Room 170A is designated as the Control Room. The area between Room 169 and 170 is used for storage of secondary pyrotechnic devices. Room 170B is used for storage of initiators. All detonation of pyrotechnic devices will be in Room 170.

#### 1.2 SCOPE

This document contains the steps and/or references the procedure to conduct the test.

#### 2.0 SAFETY

Follow all emergency and safety requirements specified in ET01-DYN-SHK-FOP-001.

#### 2.1 Responsibilities

The Test Engineer will be responsible for all activities occurring in the hazardous test area and for the safety of personnel involved in the test activities. It is the responsibility of each individual in a test program to fully comply with the requirements of this document and to report any individual not complying. Failure to do so could lead to serious personnel injuries or death.

#### 3.0 TEST REQUIREMENTS AND INFORMATION

#### 3.1 DOCUMENTS

#### 3.1.1 APPLICABLE DOCUMENTS

Test Requirements: Pyroshock Response Characterization of Composite Materials Test Plan Revision C, NESC Task # TI-12-0783 (SLS ADO-21), 8/15/2013



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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Test Procedure: ET01-DYN-SHK-FOP-001 Pyrotechnic Shock Tests

97M00200-02x PANEL TESTS 19-28 COMPOSITE SANDWICH PANEL GROUP III TESTS 19-28 (where dash number -02x is -020 to -029)
97M00200-GRP I-TEST 2,4 COMPOSITE TEST PANEL, GROUP I, TEST #2 & #4
97M00200-GRP I-TEST 9-10 COMPOSITE TEST PANEL, GROUP I, TEST #9&#10
97M00202 LSC BACKING PLATE, COMPOSITE TEST PANEL PATHFINDER
97M00203-MOD LSC PLATE, COMPOSITE TEST PANEL PATHFINDER
97M00203-MOD-COMPOSITE COMPOSITE LSC PLATE, COMPOSITE TEST PANEL TEST
PANEL TEST
97M00204-MOD-2-10 LSC SHIM, COMPOSITE TEST PANEL 10 GPF LSC
97M00204-MOD-2-22 LSC SHIM, COMPOSITE TEST PANEL 22 GPF LSC

3.1.2 REFERENCED DOCUMENTS

ET01-DYN-OWI-001 Documentation Control

ET01-DYN-OWI-002 Test Operation Procedure Preparation and Change Control

- 3.2 TEST INFORMATION
- 3.2.1 The instrumentation locations are given in the drawings listed in the applicable documents and appendix A for the test.
- 3.2.2 Pyrotechnic shock tests may be performed on the test article in the order and configuration directed by the test requester.
- 3.2.3 The shock test will be performed on a room temperature test article.
- 3.3 TEST REQUIREMENTS
- 3.3.1 The Test Engineer will be in charge of all test preparations and activities.
- 3.3.2 All activities will be coordinated with the Test Engineer.
- 3.3.3 All changes to the procedure will be coordinated with the Test Engineer.
- 3.3.4 The development test articles will be tested with pyrotechnic shock test runs as directed by the test requester. The test article information will be recorded in this TCP.
- 4.0 TEST DATA
  - a. The test data includes a time history of the real time shock recorded over a 20 millisecond or longer interval and the units are g's peak versus time.
  - b. The second plot is a Shock Response Spectrum (SRS) using 5% damping and a



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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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1/6 octave shock spectrum analyzer. The SRS is computed over the frequency band from 50 to 10,000 Hertz. The SRS units are g's versus frequency.

- c. The data will be acquired on a Nicolet BE256LE data acquisition system and the SRS analysis will be performed using a personal computer and the Shock Analysis Tool Analysis Software.
- d. Sample rate of 1 million samples per second will be used for response from the accelerometers.

#### 5.0 TEST SETUP

#### 5.1 TEST ARTICLE AND SHOCK PLATE SETUP

- a. The test setup is shown in Appendix A.
- Suspend the shock plate from ceiling using straps or cables and shackles.
- c. Suspend 1 accelerometer near the plate. Connect to data system for recording.
- d. At the start of each test day, complete ET01-DYN-SHK-FOP-001, section 6.

#### 6.0 TEST OPERATION

#### 6.1 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

a. Record and verify the test information below and in appendix C.

Group III - Test No.: 19 Panel #19 Date: 10-31-2013 Test Article Desc.: IM7/ TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face Sheets, Fabric LSC plate PID# 0346A019 Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4'

- b. Verify that the shock plate is ready for testing per section 5.1.
- Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb. Torque wrench: 1634527 Torque value: 25 Due: 3-12-2014 Cal.: 9-12-2013

d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 240±10 in-lb.

- Torque wrench: 1644 173 Torque value: 240 Due: 3-6-247 Cal: 7-6203

  e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb. Torque wrench: 16473 Torque value: 240/200 Due: 3-6-247 Cal: 7-20-201

  f. Photograph the locations and orientations of all accelerations.
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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#### 6.2 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group III Test No.: 20 Panel #20 Date: II-19-243 Test Article Desc.: IM7/
  TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell
  Foam and Fabric Face Sheets, Fabric LSC plate PID# 0346 #0 20

  Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al
  Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.

  Torque wrench: #134517 Torque value: 25 Due: 3-12-2014 Cal.: 9-12-2013
- d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 240±10 in-lb.

  Torque wrench: 1644973 Torque value: 240 Due: 3-16-2614 Cal: 9-16-18
- e. LSC plate D-ring 1/2-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb.

  Torque wrench: Torque value: 240/20 Due: 240/20
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- Photograph the test sctup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

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#### 6.3 COMPOSÍTE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group III Test No.: 21 Panel #21 Date: 12-4-20/3 Test Article Desc.: 1M7/
  TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al
  Honeycomb & Tape Face Sheets, Al LSC plate PID# 0346 4021
  Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al
  Actual Length Used: 4'
- h. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.
   Torque wrench: M634527 Torque value: 25 Due: 3-12-2014 Cal.: 9-12-2013
- d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 240±10 in-lb. Torque wrench: M644973 Torque value: 240 Due: 3-16-2014 Cal: 9-16-2013
- e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb.

  Torque wrench: M659926 Torque value: 200 Due:3-30-20 Cal: 9-30-2013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

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#### 6.4 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group III Test No.: 22 Panel #22 Date: 11-25-2013 Test Article Desc.: 1M7/
  TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell
  Foam and Fabric Face Sheets, Al LSC plate PID# 0346 A 022
  Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb. Torque wrench: M634527 Torque value: 25 Due: 3-/2-20/4 Cal.: 9-12-20/3
- d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 240±10 in-lb.

  Torque wrench: M644 973 Torque value: 240 Due: 3-16-2014 Cal: 9-16-2015
- e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb.

  Torque wrench: M659926 Torque value: 200 Due: 3-30-14 Cal: 9-30-2013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

11-25-2013

#### 6.5 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group III Test No.: 23 Panel #23 Date: 4-18-2014 Test Article Desc.: 1M7/TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face Sheets, Al LSC plate PID# 0346 A023

  Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.
   Torque wrench: M658783 Torque value: 25 Due: 5-21-204 Cal.; Il-27-2013
- d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 240±10 in-lb.

  Torque wrench: M627361 Torque value: 240 Due: 5-27-24 Cal: 11-27-263
- e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb. Torque wrench: 1627361 Torque value: 246/220 Due: 5-27-20/Cal: 11-29-2013
- f. Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

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#### 6.6 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C. Group III - Test No.: 24 Panel #24 Date: 4-22-204 Test Article Desc.: IM7/ TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face Sheets, Al LSC plate PID# 0344A024 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4°
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.

  Torque wrench: 19658 783 Torque value: 25 Due: 5-29-2019 Cal.: 11-29-2013

  d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 240±10 in-lb.
- Torque wrench: M659395 Torque value: 240 Due5-21-2014 Cal: 11-21-2013
- e. LSC plate D-ring 1/2-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb. Torque wrench: <u>M658395</u> Torque value: <u>244220</u> Due: <u>5-21-14</u> Cal: <u>11-21-2613</u>
- Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

#### 6.7 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C. Group III - Test No.: 25 Panel #25 Date: 6-10-2014 Test Article Desc.: IM7/ TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face Sheets. Fabric LSC plate PID# @344A#25 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb. Torque wrench: M634527 Torque value: 25 Due: 5-9-2014 Cal.: 11-9-2014
- d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 240±10 in-lb. Torque wrench: M650749 Torque value: 240 Due: 10-3-2014 Cal: 4-3-2014
- e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb. Torque wrench: 17450749 Torque value: 240/220 Due: 10-3-14 Cal: 4-3-2014

  f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.



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#### 6.8 COMPOSITE MATERIALS PYROSIIOCK DEV. TEST

- a. Record and verify the test information below and in appendix C. Group III - Test No.: 26 Panel #26 Date: 12-18-2013 Test Article Desc.: IM7/ TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell Foam and Tape Face Sheets, Tape LSC plate PID# 0346A026 Shock Source LSC Core Load:22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 41
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb. Torque wrench: 1658783 Torque value: 25 Due: 5/29/2014 Cal.: 11/29/2013
- d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 240±10 in-lb. Torque wrench: 1644973 Torque value; 240 Due: 3/16/2014 Cal: 9/16/2013
- e. LSC plate D-ring 1/2-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb. Torque wrench: 1659926 Torque value: 200 Due:3/30/2014Cal: 9/30/2013
- f. Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SIIK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.

ICA 12-18-2013

#### COMPOSITE MATERIALS PYROSHOCK DEV. TEST 6.9

- a. Record and verify the test information below and in appendix C. Group III - Test No.: 27 Panel #27 Date: 6-11-2014 Test Article Desc.: IM7/ TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell Foam and Fabric Face Sheets, Fabric LSC plate PID# @346 A @ 27 Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb. Torque wrench: M634527 Torque value: 25 Due: 11-9-2014 Cal.: 5-9-2014
- d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 240±10 in-lb. Torque wrench: M650749 Torque value: 240 Due: 10-3-2014 Cal: 4-3-2014
- e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb.

  Torque wrench: M650749 Torque value: 240/210 Due: 10-3-14 Cal: 4-3-2044

  f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.



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#### 6.10 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group III Test No.: 28 Panel #28 Date: I-24-26/4 Test Article Desc.: IM7/
  TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al
  Honeycomb & Tape Face Sheets, Tape I.SC plate PID# 0346 A 028
  Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al
  Actual Length Used: 41
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Use a #12 washer between the accelerometer and the insert. Each accelerometer's torque will be to 25±5 in.-lb.
- Torque wrench: 1658783 Torque value: 25 Due: 5-29-2014 Cal.: 11-29-2013
  d. LSC plate & LSC backer plate to Test Panel 1/4-13 bolt's torque to 240±10 in-1b.
  Torque wrench: 1644973 Torque value: 240 Due: 3-16-2014 Cal: 9-16-2013
- e. LSC plate D-ring ½-13 bolt's torque to 240 in-lb. Right side D-rings are 220 in-lb.

  Torque wrench: 1644973465792 Torque value: 240/200 Due 13240 Cal: 30202013
- Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

1-24-2014

#### 6.11 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.

  Group I (retest of #10) Test No.: 01 Panel #10 Date: I-16-2014 Test Article
  Desc.: IM7/TC350 Solid Tape Composite Panel, 3'x6'x0.2" PID# 0326A 010

  Shock Source LSC Core Load: 10 GR/FT Explosive Material: RDX Sheath: Al
  Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Each accelerometer's torque will be to 30±5 in.-lb.

  Torque wrench: M658783 Torque value: 30 Due: 5/21/204 Cal.: 11-29-2013
- d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55±5 ft-lb.

  Torque wrench: #644 9 73 Torque value: 55 Due: 3-16-2014 Cal: 9-16-2013
- e. D-ring ½-13 bolt's torque to 28 ft-lb.
  Torque wrench: M644973 Torque value: 28 Due: 3-16-2014Cal: 9-16-2013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

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#### 6.12 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C. Group I (retest of #4) - Test No.: 02 Panel #4 Date: 12-12-2013 Test Article Desc.: IM7/TC350 Solid Fabric Composite Panel, 3'x6'x0.3" PID# 0320 A 004 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- c. Instrument shock plate. Each accelerometer's torque will be to 30±5 in.-lb. Torque wrench: M658783 Torque value: 30 Due: 5/29/14 Cal.: 11/29/13
- d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 55±5 ft-lb.

  Torque wrench: 1/644973 Torque value: 55 Due: 3/14/2014 Cal: 9/14/2013
- e. D-ring 1/2-13 bolt's torque to 28 ft-lb. Torque wrench: 1644973 Torque value: 28 Due: 3/14/4H Cal: 9/14/2013
- f. Photograph the locations and orientations of all accelerometers.
- g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- Verify that the test run has been completed.

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#### 6.13 COMPOSITE MATERIALS PYROSHOCK DEV. TEST:

- a. Record and verify the test information below and in appendix C. Group I (retest of #2) - Test No.: 03 Panel #2 Date: 12-16-2013 Test Article Desc.: IM7/ TC350 Solid Fabric Composite Panel, 3'x6'x0.2" PID# 03204002 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: Al Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- Instrument shock plate. Each accelerometer's torque will be to 30±5 in.-lb. Torque wrench: M658783 Torque value: 30 Due: 5/27/2019 Cal.: 11/27/20/3
- d. LSC plate & LSC backer plate to Test Panel 1/2-13 bolt's torque to 55±5 ft-lb. Torque wrench: 1644 973 Torque value: 55 Due: 3/14/2014 Cal: 9/16/2013
- e. D-ring ½-13 bolt's torque to 28 ft-lb. Torque wrench: 14644 973 Torque value: 28 Due: 4/6/2014 Cal: 9/16/2013
- Photograph the locations and orientations of all accelerometers.
- Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
- h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.
- i. Verify that the test run has been completed.

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#### 6.14 COMPOSITE MATERIALS PYROSHOCK DEV. TEST

- a. Record and verify the test information below and in appendix C.
   Group I (retest of #9) Test No.: 04 Panel #9 Date: I-23-2014 Test Article Desc.: IM7/ TC350 Solid Tape Composite Panel, 3'x6'x0.3" PID# 0326 A 209 Shock Source LSC Core Load: 22 GR/FT Explosive Material: CH-6 Sheath: ΔI Actual Length Used: 4'
- b. Verify that the shock plate is ready for testing per section 5.1.
- (1) c. Instrument shock plate. Each accelerometer's torque will be to 30±5 in.-lb, Torque wrench: M658783 Torque value: 30 Due: 5-29-2014 Cal.: 11-29-2013
  - d. LSC plate & LSC backer plate to Test Panel ½-13 bolt's torque to 55±5 ft-lb.

    Torque wrench: 1644973 Torque value: 55 Due 3-16-2014 Cal: 9-16-2013
  - e. D-ring 1/2-13 bolt's torque to 28 ft-lb.
  - Torque wrench: M644973 Torque value: 28 Due: 3-16-2014Cal: 9-16-2013
  - f. Photograph the locations and orientations of all accelerometers.
  - g. Perform the test per sections 7, 8, and 9 of ET01-DYN-SHK-FOP-001.
  - h. Photograph the test setup after the test. Photograph and document any post-test visually inspected observations under this test number in appendix C.

i. Verify that the test run has been completed.

1-23-2014(1) Acceleranter A7 only torqued to 20 in-16. and insert started to more Acceleranter A9 only torqued to 25 in-16. and insert started to more Test Requester Afground to proceed.

7.0 RECORDS

The test report for this test will control and include the following records:

a. This "AS RUN" TCP.

b. The test data and the equipment list.

The test report is controlled by ET01-DYN-OWI-001, Documentation Control. However, due to the ITAR designation for the test results, the test report and data will be securely controlled. The test report will be available no later than 30 days after test completion. The Test Requirements will not be included in this TCP or in the report, but a copy may be filed with the report for future reference.

#### 8.0 TOOLS, EQUIPMENT, AND MATERIALS

The equipment used during this test will be listed in a table as part of the test report. The list will include test equipment calibration due dates.

#### 9.0 PERSONNEL TRAINING AND CERTIFICATION

Personnel certified as Propellant and Explosive Handler are required to conduct this test.



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#### POST-TEST VERIFICATION

The Test and Check-out Procedure NESC-DEV-13-037 has been satisfactorily completed and documented.

Test Engineer / ET40 Date



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## APPENDIX A

TEST SETUP



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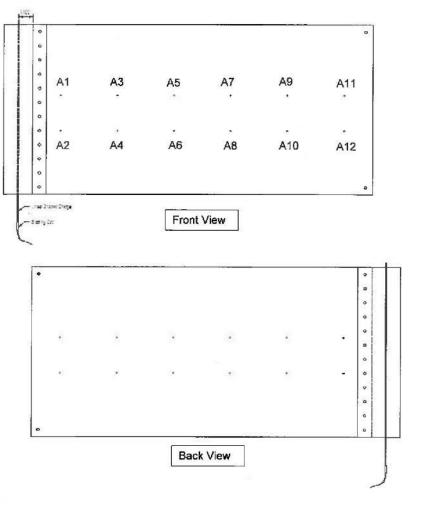
#### Composite Materials Pyroshock Development Test

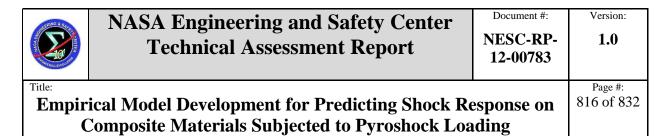
Test Article Panel:

Sandwich or Solid Composite, Vertical Position

Supports:

Straps and Shackles





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### APPENDIX B

GROUP III – TEST #19 - #28 PLY LAYUP (from the test plan)



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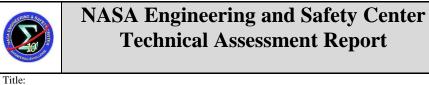
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Group III Tests #19-#28 Fill and Ply Layup

Panel and Test Number	Material	Panel Thickness	Fill/Ply	Orientation	Туре	LSC (gpf)	LSC Plate
19	Composite, IM7/TC350		Al Honeycomb & Fabric Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10	Composite, IM7/TC350
20	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Rohacell Foam & Fabric Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22	Composite IM7/TC350
21	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Al Honeycomb & Tape Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22	Aluminum
22	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Rohacell Foam & Fabric Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10	Aluminum
23	Composite, IM7/TC350			10	Aluminum		
24	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Al Honeycomb & Fabric Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22	Aluminum
25	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Al Honeycomb & Fabric Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22	Composite, IM7/TC350
26	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Rohacell Foam & Tape Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	22	Composite, IM7/TC350
27	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Rohacell 90°/+45°/-45°/0°/0°/ 3 Ply/ 1-inch Foam & 45°/+45°/0°		10	Composite, IM7/TC350	
28	Composite, IM7/TC350	8 Ply/ 1-inch thick fill	Al Honeycomb & Tape Face Sheets	90°/+45°/-45°/0°/0°/ -45°/+45°/90°, 8 ply both faces	Sandwich	10	Composite, IM7/TC350



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### APPENDIX C

**TEST DATA SHEET** 



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#### TEST DATA SHEET

Group: III - Test No.: 19 Panel #19 Date: 10-31-2013 Test Article Desc.: 1M7/TC350
Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face

Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-020 PANEL TESTS 19-28 Material: IM7/TC350

PID# 0346A019

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD-COMPOSITE Material: Composite S/N: Fabric 0346-53

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder
Shock Source: LSC L/N: none LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N									
Į.	C02	3/334	2	D02	43026	3	D02	43028	4	C02	31331
5	C02	31328	6	D02	43029	7	D02	43179	8	C02	3/351
9	C02	3/330	10	D02	43180	11	D02	43181	12	C02	40274
13	B02	11439								0.00	

Congoix Aluminum I.SC panel severance: (Yes No) 1-2 layers of compain several.

Post-test visually inspected observations: Acrels did not loace torque. Bolts did not loase

torque except, from top-left, 1,23,89,12 were less thanks in-like.

Group: III - Test No.: 20 Panel #20 Date: 11-19-2013 Test Article Desc.: IM7/IC350
Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell Foam and Fabric Face

Sheets Test Article Configuration: hanging
Test Article Drawing #: 97M00200-021 PANEL TESTS 19-28 Material: IM7/TC350

PID# 0346A0 20
Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD-COMPOSITE Material: Composite S/N: Fabric 0346-51

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: hone LSC Core Load: 22 GR/FT Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	D02	43026	2	C02	31340	3	C02	31338	4	D02	43028
5	D02	43029	6	C02	3 133 3	7	C02	40292	8	D02	43179
9	D02	43/80	10	C02	40295	11	C02	31336	12	D02	43 181
13	B02	11439									

Composite Aluminum LSC panel severance: (Yes/No) de laminated LSC panel and severad.

Post-test visually inspected observations: Bolts and accels did not loose torque.

Several layers of the LSC panel delaminated toward the congests sandwick panel. The cut end of the LSC panel had I layer completely separate.



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#### TEST DATA SHEET, cont.

Group: III - Test No.: 21 Panel #21 Date: 12-4-208 Test Article Desc.: IM7/TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Tape Face

Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-022 PANEL TESTS 19-28 Material: IM7/TC350

PID# 0346 A 021

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder
Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 22 GR/FT Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 4' Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	C02	3(334	.2 .	D02	43026	. 3 .	D02	43028		C02	31331
5	C02	31328	. 6	D02	43029	7	D02	43/79	. 8	C02	31351
9	C02	31330	10	D02	43180	11	D02	43/81	12	C02	40274
13	B02	11439	- 8	-	2						

Aluminum LSC panel severance: (Yes)/ No )

Post-test visually inspected observations: No loose bolts or necels

Group: III - Test No.: 22 Panel #22 Date: (1-25-2013 Test Article Desc.: IM7/TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell Foam and Fabric Face Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-023 PANEL TESTS 19-28 Material; IM7/TC350 PID# 03464022

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: LSC Core Load: 10 GR/FT | Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX Loc. Model S/N Loc. Model S/N Model S/N D02 43026 2 31340 C02 3 C02 31338 4 D02 43028 43029 6 5 D02 C02 31333 7 C02 8 D02 43179 40292 9 D02 43180 10. C02 40295 C02 D02 31336 43/81 B02 11439

Aluminum LSC panel severance: (Yes)/No)

Post-test visually inspected observations: No losse balts. Al and A2 18ih-16e., A3-A12. 20-29in-4e



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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Composite Materials Pyroshock Development Test	NESC-DEV-13-037	Revision: Baseline	
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#### TEST DATA SHEET, cont.

Group: III - Test No.: 23 Panel #23 Date: 4-18-2014 Test Article Desc.: IM7/TC350 Composite

Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face Sheets

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-024 PANEL TESTS 19-28 Material: IM7/TC350

PID# 0346 A023

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	C02	31334	. 2	D02	43026	. 3	D02	43028	. 4	C02	3/331
5	C02	3/328	. 6	D02	43373	7	D02	43179	8	C02	31351
. 9 .	C02	31330	10	D02	43180	→11 ·	D02	43181	-12	C02	40274
13	B02	11439						12000 00000			

Aluminum LSC panel severance: (Yes 100) Agroz. 1/2 panel was severed.
Post-test visually inspected observations: Boths and accels did not loss to gree.

Group: III - Test No.: 24 Panel #24 Date: 4-22-2014 Test Article Desc.: IM7/TC350
Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honeycomb & Fabric Face
Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-025 PANEL TESTS 19-28 Material: IM7/TC350 PID# 0344A024

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: none LSC Core Load: 22 GR/FT Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N									
1	D02	43026	2	C02	3 340	- 3	C02	3/338	. 4	D02	43028
5	D02	43373	6	C02	31333	7	C02	40292	8	D02	43179
9	D02	43180	10	C02	40295	11	C02	31334	12	D02	43/8/
13	B02	11439				7.9					

Aluminum LSC panel severance: (Yes/No)

Post-test visually inspected observations: Bolts and accels did not loose torque.



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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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#### TEST DATA SHEET, cont.

Group: III - Test No.: 25 Panel #25 Date: 6-10-2014 Test Article Desc.: IM7/TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Al Honcycomb & Fabric Face

Sheets Test Article Configuration: hanging Test Article Drawing #: 97M00200-026 PANEL TESTS 19-28 Material: IM7/TC350

PID# 0346 A025

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD-COMPOSITE Material: Composite S/N: Fabric

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

LSC Core Load: 22 GR/FT Explosive Material; CH-6 Shock Source: LSC L/N: none

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N									
. 1	C02	3/334	2	D02	43026	3	D02	43028	4	. C02	3/331
-5	C02	3/328	.6	D02	43373	7	D02	43179	8	C02	31351
9	C02	31330	10	D02	43180	1.1	D02	43181	12	C02	40274
13	B02	11431			100						

Composite Aluminum LSC panel severance: (Yes)/ No.)

Post-test visually inspected observations: Bolts and a ccols. did not lesse torque.

Group: III - Test No.: 26 Panel #26 Date: 2-18-2013 Test Article Desc.: IM7/TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell Foam and Tape Face Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-027 PANEL TESTS 19-28 Material: IM7/TC350 PID# 0346A026

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD-COMPOSITE Material: Composite S/N: Tape 5P-5

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: \_none LSC Core Load: 22 GR/FT Explosive Material: CH-6

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

					7 42 47 4						
Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
. 1	D02	43026	2	C02	2340	: 3	C02	31338	. 4	D02	43028
. 5	D02	43029	6	C02	31333	7	C02	40292	. 8	D02	43/79
9 .	D02	43180	10	C02	40295	. 11	C02	31336	12.	D02	43181
13	B02	11439								9	

Composite Aluminum-LSC panel severance: (Yes) No )

Post-test visually inspected observations: No loose a ccels or bolts.



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### **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

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#### TEST DATA SHEET, cont.

Group: III - Test No.: 27 Panel #27 Date: 6-11-2014 Test Article Desc.: IM7/TC350 Composite Sandwich Panel, see test plan for ply layup, 3'x6', Rohacell Foam and Fabric Face

Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-028 PANEL TESTS 19-28 Material: IM7/TC350

PID# 0346 4 927

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD-COMPOSITE Material: Composite S/N: Fabric Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder LSC Core Load: 10 GR/FT Explosive Material: RDX Shock Source: LSC L/N: none

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	C02	3/334	2	D02	43026	3	D02	43028	4	- C02	31331
5	C02	31328	- 6	D02	43373	17	D02	43179	- 8	C02	31351
9	C02	31330	10	D02	43180	11	D02	43181	12	C02	40274
13.,	B02	(1439.							989) 8 - 119		

Corposite Aluminum LSC panel severance: (Yes /(Nd)

Post-test visually inspected observations:

Group: III - Test No.: 28 Panel #28 Date: 1-24-2614 Test Article Desc.: IM7/TC350 Composite Sandwich Panel, see test plan for ply layup. 3'x6', Al Honeycomb & Tape Face Sheets Test Article Configuration: hanging

Test Article Drawing #: 97M00200-029 PANEL TESTS 19-28 Material: IM7/TC350

PID# <u>0346A0</u>28

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD-COMPOSITE Material: Composite S/N: Tape Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: LSC Core Load: 10 GR/FT Explosive Material: RDX

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX S/N Loc. Loc. Model S/N Loc. Model S/N Model Loc. Model D02 43026 2 31340 31338 4 C02 C02 D02 48028 5 D02 43373 C02 31333 C02 D02 40292 43179 D02 43/80 9 10 C02 11 C02 31336 40295 D02 43 181 13

B02 1/439

Comparie Aluminum LSC panel severance: (Yes No)
Post-test visually inspected observations: The majority of the fanel was cot almost through. There was also delamination.



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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

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#### TEST DATA SHEET, cont.

Group: I (retest of #10) - Test No.: 01 Panel #10 Date: 1-/6-2014 Test Article Desc.: IM7/

TC350 Solid Tape Composite Panel, 3'x6'x0.2"

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP I-TEST 10 Material: IM7/TC350 PID# 0328A010

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-10 Material: Aluminum S/N: Pathfinder A leal

Shock Source: LSC L/N: none LSC Core Load: 10 GR/FT Explosive Material: RDX 125

Sheath: Aluminum Actual Length Used: 4' Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	C02	31334	2	D02	43026	3	D02	43028	4	C02	3/331
5	· C02	31328	6	_ D02	4362	7	D02	43179	. 8	C02	3/35/
9	C02	31330	1.0	D02	43110	· 11.	D02	43/81	12	C02	40274
13	B02	11439	4	- No.		Turn i	54000	50 S	* I.	1870 E	

Aluminum LSC panel severance: (Yes (No) 2 5 mall blow throughs in the middle of the plate Post-test visually inspected observations: Boltz and accels. did not losse targue.

Group: I (retest of #4) - Test No.: 02 Panel #4 Date: 12-12-2013 Test Article Desc.: IM7/ TC350 Solid Fabric Composite Panel, 3'x6'x0.3"

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP I-TEST 4 Material: IM7/TC350 PID# 0320 A004

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder Shock Source: LSC L/N: <u>Mone</u> LSC Core Load: <u>22 GR/FT</u> Explosive Material: <u>CH-6</u> Sheath: <u>Aluminum</u> Actual Length Used: <u>**4**<sup>t</sup></u>

Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N	Loc.	Model	S/N
1	D02	43026	2	C02	31340	3	C02	\$1338	4	D02	43028
5	D02	43029	6	C02	31333	7	C02	40292	8	D02	43179
9	D02	43180	10	C02	40215	11	C02	31336	. 12	D02	43181
13	B02	11484					9.5		J		7/307

Aluminum LSC panel severance: (Yes) No)

Post-test visually inspected observations: No loose bolts. No loose accels, except A5, which had 20-25 in-lbs after the test. Aliasert broke loose from panel while removing accel.



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#### TEST DATA SHEET, cont.

Group: I (retest of #2) - Test No.: 03 Panel #2 Date: 12-16-2013 Test Article Desc.: IM7/

TC350 Solid Fabric Composite Panel, 3'x6'x0.2"

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP I-TEST 2 Material: IM7/TC350 PID# 6320A662

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder Shock Source: LSC L/N: none LSC Core Load: 22 GR/FT Explosive Material: CII-6

Sheath: Aluminum Actual Length Used: 4'

Accelerometer MFG: PCB Model: 350XXX

Loc.	Model	S/N									
1	C02	31334	2	D02	43026	3	D02	43028	4	C02	31331
5	C02	31338	6	D02	43029	7	D02	43/79	8	C02	31351
9	C02	3/330	10	D02	43180	11	D02	43181	12	C02	40274
13	B02	11439		_	1	- 8					

Aluminum LSC panel severance: (Yes)/No)

Post-test visually inspected observations: Bolts and accels did not loose to rque.

Group: I (retest of #9) - Test No.: 04 Panel #9 Date: 1-23-2014 Test Article Desc.: 1M7/ TC350 Solid Tape Composite Panel, 3'x6'x0.3"

Test Article Configuration: hanging

Test Article Drawing #: 97M00200-GRP I-TEST 9 Material: IM7/TC350 PID# 0326 A 809

Test Article Drawing #: 97M00202 Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00203-MOD Material: Aluminum S/N: Pathfinder

Test Article Drawing #: 97M00204-MOD-2-22 Material: Aluminum S/N: Pathfinder

Shock Source: LSC L/N: \_\_nene\_\_ LSC Core
Sheath: Aluminum Actual Length Used: \_\_4' LSC Core Load: 22 GR/FT Explosive Material: CH-6

Accelerometer MFG: PCB Model: 350XXX Loc. Model S/N Loc. Model S/N Loc. Model S/N Loc. 43026 2 43373 6 D02 C02 3 31340 C02 31338 4 D02 43028 D02 C02 31333 C02 8 D02 40292 43179 D02 43180 31336 10 C02 C02 D02 43181 B02 11439

Aluminum LSC panel severance: (Yes)/ No )

Post-test visually inspected observations: No loss access or bolts. Ab held torque but Started debonding at the end of the torque check. A7 had 15 in-16. & A9 had 20 in-16s.



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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-037 Composite Materials Shock Test

**Equipment List** 



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

			DEV-13-037 Oment List				
20 st							
Description	Manufacturer	Model/Version	ID/Serial Number	Location	Cal Due Date		
Shock Analysis Tool	ET40	1.2.5			Verified 5/21/2009		
TEAM256	Nicolet	7.20	L. Constitution of the Con		Verified 7/26/2012		
Forque Wrench (tests 1-10)	Prata	6012	M644973	- î	3/16/2014		
Forque Wrench (tests 1-4,7,10)	Proto	6062C	M659926		3/30/2014		
Torque Wrench (tests 1-4)	Snap-Or	TEC3FUA	M634527		3/12/2014		
Torque Wrench (tests 13,14)	Snap-Or	TEC3FUA	M634527		11/9/2014		
Forque Wrench (tests 5-12)	Precision Instruments	44620	M658783		5/29/2014 5/29/2014		
Forque Wrench (test 11)	Snap Or	TE25 FUA	M627361 M658395				
forque Wrench (test 12)	Proto	6066C			5/21/2014		
forque Wrench (tests 13,14)	Proto Endevco	6006F 2793	M650749 M652262	Channels 1-15	10/3/2014 1/10/2014		
Power Supply (tests 1-7)		2793	M652262		1/8/2015		
Power Supply (tests 8-15) Data Acquisition System	Endevco Nicolet	BE256LE	2011288	Channels 1-15 Channels 2-8, 10-15	8/6/2014		
Channel 2	Nicolet	614CB	001-2	A1	8/6/2014		
Channel 3	Nicolet	614CB	001-3	A2	8/6/2014		
Channel 4	Nicolet	614CB	001-4	A3	8/6/2014		
Channel 5	Nicolet	614CB	002-1	A4	8/6/2014		
Channel 6 Nicolet		614CB	002-2	A5	8/6/2014		
Channel 7	Nicolet	614CB	002 3	A6	8/6/2014		
Channel 8	Nicolet	614CB	002-4	A7	8/6/2014		
Channel 10	Nicolet	614CB	003-2	A8	8/6/2014		
Channel 11	Nicolet	614CB	003-3	A9	8/6/2014		
Channel 12	Nicolet	614CB	003-4	A10	8/6/2014		
Channel 13	Nicolet	614CB	004-1	A11	8/6/2014		
Channel 14	Nicolet	614CB	004-2	A12	8/6/2014		
Channel 15	Nicolet	614CB	004-3	A13	8/6/2014		
Accelerometer	PCB	350B02	11439	A13 (tests 1-12)	4/29/2014		
Accelerometer	PCB	350B02	11439	A13 (tests 13,14)	6/5/2015		
Accelerometer	PCB	350C02	31334	Set 1 A1 (tests 1,4,6,8)	4/23/2014		
Accelerometer	PCB	350C02	31334	Set 1 A1 (tests 11,13,14)	1/22/2015		
Accelerometer	PCB	350C02	31331	Set 1 A4 (tests 1,4,6,8)	4/24/2014		
Accelerometer	PCB	350C02	31331	Set 1 A4 (tests 11,13,14)	1/22/2015		
Accelerometer	PCB	350C02	31328	Set 1 A5 (tests 1,4,6,8)	4/24/2014		
Accelerometer	PCB	350C02	31328	Set 1 A5 (tests 11,13,14)	1/22/2015		
Accelerometer	PCB	350C02	31351	Set 1 A8 (tests 1,4,6,8,11)	4/24/2014		
Accelerometer	PCB	350C02	31351	Set 1 A8 (tests 13,14)	6/5/2015		
Accelerometer	PCB	350C02	31330	Sct 1 A9 (tests 1,4,6,8,11)	4/24/2014		
Accelerometer	PCB PCB	350C02 350C02	31330 40274	Set 1 A9 (tests 13,14)	6/5/2015		
Accelerometer	The second secon			Set 1 A12 (tests 1,4,6,8,11)	4/24/2014		
Accelerometer Accelerometer	PCB PCB	350C02 350C02	40274 31340	Set 1 A12 (tests 13,14) Set 2 A2	6/5/2015 4/23/2014		
Accelerometer Accelerometer	PCB	350C02 350C02	31340	Set 2 A2 Set 2 A3	4/24/2014		
Accelerameter	PCB	350C02 350C02	31333	Set 2 A6	4/24/2014		
Accelerometer	PCB	350C02	40292	Set 2 A7	4/24/2014		
Accelerometer	PCB	350C02	40295	Set 2 A10	4/24/2014		
Accelerometer	PCB	350C02	31336	Set 2 All	4/24/2014		
Accelerometer	PCB	350D02	43026	Set 1 A2, Set 2 A1 (tests 1-8)	4/23/2014		
Accelerometer	PCB	350D02	43026	Set 1 A2, Set 2 A1 (tests 9 14)	1/22/2015		
Accelerometer	PCB	350D02	43028	Set 1 A3, Set 2 A4 (tests 1 8)	4/23/2014		
Accelerometer	PCB	350D02	43028	Set 1 A3, Set 2 A4 (tests 9-14)	1/22/2015		
Accelerometer	PCB	350D02	43029 (tests 1-7)	Set 1 A6, Set 2 A5	4/23/2014		
Accelerometer	PCB	350D02	43373 (tests 8-14)	Set 1 A6, Set 2 A5	10/17/2014		
Accelerometer	PCB	350D02	43179	Set 1 A7, Set 2 A8 (tests 1-8)	4/22/2014		
Accelerometer	PCB	350D02	43179	Set 1 A7, Set 2 A8 (tests 9 14)	1/22/2015		
Accelerometer	PCB	350D02	43180	Set 1 A10, Set 2 A9 (tests 1-8)	4/23/2014		
Accelerometer	PCB	350D02	43180	Set 1 A10, Set 2 A9 (tests 9-14)	1/22/2015		
Accelerometer	PCB	350D02	43181	Set 1 A11, Set 2 A12 (tests 1-8)	4/23/2014		
Accelerometer	PCB	350D02	43181	Set 1 A11, Set 2 A12 (tests 9-14)	1/22/2015		



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**Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading** 

## NESC-DEV-13-037 Composite Materials Shock Test

**Data Acquisition Setups** 



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

TEAM256 SETTINGS TESTS 1,4,6 Date: 11-05-2013 Time: 08:54:26 Storage Path: C:\TEAMPRO Filename: 001 C:\TEAM256 File Number: Settings Path: C:\TEAM256
Settings File: NES319.SET
Export Path: D:\ATEST\N.
Export Format: FAMOS D:\ATEST\NESC\_5\NEST3-19\RAWDAT-1 Average Blocks: Between Cursors: No \* RECORDER SETTINGS \*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\* Frequency A : Pre Trigger : 1.0000 MHz(Internal) 48000 Samples (48.00 ms) 1000576 Samples (1.001 s) Segment A : Number of Blocks : Digital Event Channels : Analog Channels : Nr. Name 1 XXX 1 Coup. Amp. Filter GND + 33.00 k DC + 33.00 k Units Max Trigger -6.000 6.000 kg's pk -28.85 28.85 kg's pk Off DC DC NES\_2 Basic NES\_3 -30.93 NES\_4 -31.25 30.93 kg's pk 31.25 kg's pk 27.78 kg's pk 33.00 k 33.00 k Basic Basic NES\_5 DC 33.00 k Basic 33.00 k 33.00 k 33.00 k 33.00 k 33.00 k 33.00 k NES\_6 NES\_7 NES\_8 -28.04 -30.30 -30.61 DC DC DC 28.04 kg's pk 30.30 kg's pk 30.61 kg's pk Basic Off XXX\_9 -6.000 NES\_10 -28.04 NES\_11 -27.03 6.000 kg's pk 28.04 kg's pk 27.03 kg's pk 10 DC DC Off Off NES\_12 -31.58 DC DC DC DC 33.00 k 33.00 k 33.00 k NES\_13 -30.61 NES\_14 -27.27 30.61 kg's pk 27.27 kg's pk 13 Off 14 kg's pk Off 8.929 kg's pk 55.56 kg's pk NES\_15 -8.929 33.00 k 15 NES\_15 -8.929 8.929 kg's pk DC
16 ROC\_16 -55.56 55.56 kg's pk DC
17 ROC\_17 -55.56 55.56 kg's pk DC
18 ROC\_18 -55.56 55.56 kg's pk DC
19 ROC\_19 -55.56 55.56 kg's pk DC
20 ROC\_20 -55.56 55.56 kg's pk DC
Engineering Units Scaling
XXX\_1 0 + 1.0000 k \* Voltage (g's pk)
NES\_2 0 + 9.6154 k \* Voltage (g's pk)
NES\_3 0 + 10.309 k \* Voltage (g's pk)
NES\_3 0 + 10.417 k \* Voltage (g's pk)
NES\_4 0 + 10.417 k \* Voltage (g's pk)
NES\_5 0 + 9.2593 k \* Voltage (g's pk)
NES\_6 0 + 9.3458 k \* Voltage (g's pk)
NES\_7 0 + 10.101 k \* Voltage (g's pk)
NES\_7 0 + 10.101 k \* Voltage (g's pk)
NES\_10 0 + 9.3458 k \* Voltage (g's pk)
NES\_11 0 + 9.0000 k \* Voltage (g's pk)
NES\_11 0 + 9.0000 k \* Voltage (g's pk)
NES\_11 0 + 10.526 k \* Voltage (g's pk)
NES\_12 0 + 10.526 k \* Voltage (g's pk)
NES\_13 0 + 10.204 k \* Voltage (g's pk)
NES\_14 0 + 9.0909 k \* Voltage (g's pk)
NES\_15 0 + 8.9286 k \* Voltage (g's pk)
NES\_16 0 + 9.2593 k \* Voltage (g's pk)
ROC\_16 0 + 9.2593 k \* Voltage (g's pk)
ROC\_17 0 + 9.2593 k \* Voltage (g's pk)
ROC\_19 0 + 9.2593 k \* Voltage (g's pk)
ROC\_19 0 + 9.2593 k \* Voltage (g's pk)
ROC\_20 0 + 9.2593 k \* Voltage (g's pk)
ROC\_20 0 + 9.2593 k \* Voltage (g's pk)
ROC\_20 0 + 9.2593 k \* Voltage (g's pk)
ROC\_20 0 + 9.2593 k \* Voltage (g's pk)
ROC\_20 Trigger: Off ROC\_16 -55.56 ROC\_17 -55.56 ROC\_18 -55.56 DC DC DC 16 17 33.00 k 33.00 k 33.00 k 33.00 k Off off 33.00 k Trigger Settings : Auto Trigger: Off



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

TEAM256 SETTINGS TESTS 2,3,5,7 Date: 11-19-2013 Time: 09:39:35 C:\TEAMPRO Storage Path: Filename: File Number: 001 Settings Path: C:\TEAM256 Settings File: Export Path: Export Format: NES320.SET D:\ATEST\NESC\_5\NEST3-20\RAWDAT-1 FAMOS Average Blocks: Between Cursors: No Frequency A : Pre Trigger : 1.0000 MHz(Internal) 48000 Samples (48.00 ms) 1000576 Samples (1.001 s) Segment A: Number of Blocks: Digital Event Channels: Analog Channels : Nr. Name 1 XXX 1 Units Coup. Amp. Filter Man max units -6.000 6.000 kg's pk -30.93 30.93 kg's pk -31.58 31.58 kg's pk -30.00 30.00 kg's pk GND 33.00 k 33.00 k Off DC DC DC DC NES 2 Basic 33.00 k Basic NES\_4 33.00 k 33.00 k 33.00 k 33.00 k 33.00 k 33.00 k -31.2531.25 kg's pk 30.30 kg's pk NES 5 Basic NES\_6 -30.30 28.57 kg's pk 28.30 kg's pk DC DC -28.578 -28.30NES 8 Basic -6.000 6.000 kg's pk GND Off NES\_10 -30.61 NES\_11 -31.58 NES\_12 -28.30 31.58 kg's pk 28.30 kg's pk 27.78 kg's pk DC DC DC 33.00 k 33.00 k 33.00 k Off NES\_13 -27.78 33.00 k 33.00 k 33.00 k DC DC DC NES\_14 -30.61 30.61 kg's pk NES\_15 -8.929 8.929 kg's pk 55.56 kg's pk 15 55.56 kg's pk 55.56 kg's pk 55.56 kg's pk DC DC DC 33.00 k 33.00 k 33.00 k Off 20 ROC\_19 -35.56 55.56 kg's pk DC
20 ROC\_20 -55.56 55.56 kg's pk DC
Engineering Units Scaling
XXX 1 0 + 1.0000 k \* Voltage (g's pk)
NES\_2 0 + 10.309 k \* Voltage (g's pk)
NES\_3 0 + 10.526 k \* Voltage (g's pk)
NES\_4 0 + 10.000 k \* Voltage (g's pk)
NES\_5 0 + 10.417 k \* Voltage (g's pk)
NES\_6 0 + 10.101 k \* Voltage (g's pk)
NES\_7 0 + 9.5238 k \* Voltage (g's pk)
NES\_8 0 + 9.4340 k \* Voltage (g's pk)
NES\_10 0 + 10.000 k \* Voltage (g's pk)
NES\_11 0 + 10.526 k \* Voltage (g's pk)
NES\_12 0 + 9.4340 k \* Voltage (g's pk)
NES\_13 0 + 9.2593 k \* Voltage (g's pk)
NES\_13 0 + 9.2593 k \* Voltage (g's pk)
NES\_14 0 + 10.204 k \* Voltage (g's pk)
NES\_15 0 + 8.9286 k \* Voltage (g's pk)
ROC\_16 0 + 9.2593 k \* Voltage (g's pk)
ROC\_17 0 + 9.2593 k \* Voltage (g's pk)
ROC\_18 0 + 9.2593 k \* Voltage (g's pk)
ROC\_19 0 + 9.2593 k \* Voltage (g's pk) ROC\_19 0 + 9.2593 k \* Voltage (g's pk) ROC\_20 0 + 9.2593 k \* Voltage (g's pk) Trigger Settings : Auto Trigger:



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

```
3
                                               TEAM256 SETTINGS
                                                                                          TESTS 8, 11, 13, 14
  Date: 01-16-2014
  Time: 10:09:26
  Storage Path:
                            C:\TEAMPRO
                           Data
001
  Filename:
  File Number:
  Settings Path:
                            C:\TEAM256
  Settings File:
                            NESR110.SET
  Export Path:
Export Format:
                           D:\ATEST\NESC_5\NESR1-10\RAWDAT~1
                           FAMOS
  Average Blocks:
  Between Cursors: No
  Frequency A :
                                          1.0000 MHz (Internal)
                                         48000 Samples (48.00 ms)
1000576 Samples (1.001 s)
   Pre Trigger :
   Segment A :
Number of Blocks :
Digital Event Channels :
   Analog Channels:
Nr. Name Min
1 XXX_1 -6.000
                      Min Max Units
-6.000 6.000 kg's pk
                                                                          Filter
33.00 k
                                                         Coup. Amp.
                                                          GND
                                                                                         Off
                                                                  +
                                                                           33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
           NES_2 -28.85
NES_3 -30.93
NES_4 -31.25
                                                           DC
DC
DC
DC
                                  28.85 kg's pk
                                                                                         Basic
                                  30.93 kg's pk
31.25 kg's pk
                                                                                         Basic
       4567
                                                                                         Basic
                     -31.25
-27.78
-28.04
-31.91
-30.61
            NES_5
                                   27.78
                                           kg's pk
                                                                                         Basic
           NES_6
NES_7
                                  28.04 kg's pk
31.91 kg's pk
30.61 kg's pk
                                                           DC
DC
                                                                            33.00 k
                                                                                         Basic
                                                                           33.00 k
                                                                          33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
           XXX_9 -6.000
NES_10 -28.04
                                  6.000 kg's pk
28.04 kg's pk
27.03 kg's pk
      10
                                                           DC
DC
                                                                                         OFF
            NES_11 -27.03
                                                                                         Off
           NES_12 -31.58
NES_13 -30.61
NES_14 -27.27
                                                          DC
DC
DC
      12
13
                                  31.58 kg's pk
                                  30.61 kg's pk
27.27 kg's pk
8.929 kg's pk
      14
                                                                                         OFF
           NES_15 -8.929
ROC_16 -55.56
ROC_17 -55.56
      15
                                  55.56 kg's pk
55.56 kg's pk
      16
17
                                                          DC
DC
                                                                                         OFF
          ROC_18 -55.56
ROC_19 -55.56
ROC_20 -55.56
      18
                                  55.56
                                           kg's pk
                                                                                         Off
                                  55.56 kg's pk
55.56 kg's pk
      19
                                                                            33.00
  20
                                                                           33.00 k Off
     ROC_16 0 + 9.2593 k * Voltage (g's pk)
ROC_17 0 + 9.2593 k * Voltage (g's pk)
ROC_18 0 + 9.2593 k * Voltage (g's pk)
ROC_19 0 + 9.2593 k * Voltage (g's pk)
ROC_20 0 + 9.2593 k * Voltage (g's pk)
rigger Settings :
  Trigger Settings :
Auto Trigger:
```



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## **Empirical Model Development for Predicting Shock Response on Composite Materials Subjected to Pyroshock Loading**

```
9
                                                TEAM256 SETTINGS
                                                                                        TESTS 9,10,12
  Date: 01-23-2014
 Time: 11:26:54
  Storage Path:
                           C:\TEAMPRO
  Filename:
                            Data
  File Number:
  Settings Path:
                            C:\TEAM256
 Settings File:
Export Path:
Export Format:
                           NESR109.SET
D:\ATEST\NESC_5\NESR1-09\RAWDAT~1
 Average Blocks:
 Between Cursors: No.
  ******************* RECORDER SETTINGS *****************
   Frequency A :
Pre Trigger :
                                          1.0000 MHz(Internal)
                                              48000 Samples (48.00 ms)
  Segment A :
Number of Blocks :
                                          1000576 Samples (1.001 s)
  Digital Event Channels :
Analog Channels :
Nr. Name Min Max
                                    Max
                                            Units
                                                          Coup. Amp. Filter
                                                                                           Trigger
           NAME MAX ONLES

XXXX_1 -6.000 6.000 kg's pk

NES_2 -30.93 30.93 kg's pk

NES_3 -31.58 31.58 kg's pk

NES_4 -30.00 30.00 kg's pk
                                                           DC
DC
                                                                             33.00 k
                                                                                           Basic
                                                                                           Basic
                                                                             33.00
                                                                                           Basic
                     -31.25
-31.91
                                  31.25 kg's pk
31.91 kg's pk
28.57 kg's pk
                                                           DC
DC
           NES_5
                                                                            33.00 k
33.00 k
                                                                                           Basic
           NES_6
NES_7
                                                                                           Basic
                     -28.57
                                                                             33.00
                                                                                           Basic
                                  28.30 kg's pk
6.000 kg's pk
30.61 kg's pk
31.58 kg's pk
           NES_8 -28.30
XXX_9 -6.000
NES_10 -30.61
       8
                                                                             33.00
                                                                            33.00 k
                                                          GND
DC
DC
DC
DC
DC
DC
DC
                                                                                           Off
                                                                                           Off
           NES_11 -31.58
NES_12 -28.30
NES_13 -27.78
     12
                                  28.30 kg's pk
27.78 kg's pk
30.61 kg's pk
                                                                                          Off
Off
     13
           NES_14 -30.61
     15 NES 15 -8.929
16 ROC 16 -55.56
17 ROC 17 -55.56
18 ROC 18 -55.56
19 ROC 19 -55.56
20 ROC 20 -55.56
                                  8.929 kg's pk
55.56 kg's pk
55.56 kg's pk
                                                                                          Off
Off
                                                                            33.00 k
33.00 k
33.00 k
33.00 k
33.00 k
                                                                                          off
                                   55.56 kg's pk
                                  55.56 kg's pk
55.56 kg's pk
 ROC_20 0 + 9.25
Trigger Settings :
Auto Trigger:
```

#### REPORT DOCUMENTATION PAGE

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#### 13. SUPPLEMENTARY NOTES

#### 14. ABSTRACT

The NASA Engineering and Safety Center (NESC) received a request to develop an analysis model based on both frequency response and wave propagation analyses for predicting shock response spectrum (SRS) on composite materials subjected to pyroshock loading. The model would account for near-field environment (~9 inches from the source) dominated by direct wave propagation, mid-field environment (~2 feet from the source) characterized by wave propagation and structural resonances, and far-field environment dominated by lower frequency bending waves in the structure. This document contains appendices to the Volume I report.

#### 15. SUBJECT TERMS

NASA Engineering and Safety Center; Shock response spectrum; Pyroshock loading; Model development

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