

Flow Boiling and Condensation Experiment (FBCE) for the International Space Station

Issam Mudawar, Chirag Kharangate, Lucas O'Neill & Chris Konishi Purdue University Boiling and Two-Phase Flow Laboratory (PU-BTPFL) Mechanical Engineering Building, 585 Purdue Mall West Lafayette, IN 47907

Mohammad Hasan, Henry Nahra, Nancy Hall & R. Balasubramaniam NASA Glenn Research Center Fluid Physics and Transport Branch 21000 Brookpark Rd, Cleveland, OH 44135

> Jeffrey Mackey Vantage Partners, LLC 3000 Aerospace Parkway Brook Park, OH 44142





Examples of Systems Demanding Predictive Models of Effects of Gravity on Two-Phase Flow and Heat Transfer





NASA-Supported Facilities at Boiling & Two-Phase Flow Laboratory



High-Capacity Condensation Facility



Mini/micro-channel Condensation Facility



One-g Flow Boiling Facility



Falling-Film Heating/Evaporation Facility



Parabolic Flight Flow Boiling Facility



Parabolic Flight Condensation Facility

3





The proposed research aims to develop an integrated two-phase flow boiling/condensation facility for the International Space Station (ISS) to serve as primary platform for obtaining two-phase flow and heat transfer data in microgravity.

Overriding objectives are to:

- 1. Obtain flow boiling database in long-duration microgravity environment
- 2. Obtain flow condensation database in long-duration microgravity environment
- 3. Develop experimentally validated, mechanistic model for microgravity flow boiling critical heat flux (CHF) and dimensionless criteria to predict minimum flow velocity required to ensure gravity-independent CHF
- 4. Develop experimentally validated, mechanistic model for microgravity annular condensation and dimensionless criteria to predict minimum flow velocity required to ensure gravity-independent annular condensation; also develop correlations for other condensation regimes in microgravity



Layout of FBCE



Consists of: nPFH sub-loop Water sub-loop **Contains three test modules:** Flow Boiling Module (FBM) **Condensation Module CM-HT for heat** transfer measurements **Condensation Module CM-FV for flow** visualization



 Both One g_e and Parabolic flight flow boiling experiments using single-sided and double-sided heat walls

2. Modeling of CHF for single-sided and double-sided heated walls at different orientations in Earth gravity and in microgravity

3. Computational modeling of condensing film



Flow Boiling Facility



ASGSR 2015 Boiling and Two-Phase Flow Laboratory (BTPFL) November 2015

7



Flow Boiling Module



8



Heated Wall Design





Horizontal Flow Boiling – Heated Wall Configurations



ASGSR 2015 Boiling and Two-Phase Flow Laboratory (BTPFL) November 2015 10

H2: q''_{w}

Vapor

Layer

W



Horizontal Flow Boiling – Flow Visualization

Slightly subcooled flow at low velocity









Mass Balance:

 $\begin{aligned} \text{Vapor layer 1:} \qquad & U_{g1} = \frac{q_w^{-1} z}{\rho_g \delta_1 \left(c_{p,f} \Delta T_{sub,in} + h_{fg} \right)} \\ \text{Vapor layer 2:} \qquad & U_{g2} = \frac{q_w^{''} z}{\rho_g \delta_2 \left(c_{p,f} \Delta T_{sub,in} + h_{fg} \right)} \\ \text{Liquid layer:} \qquad & U_f = \frac{UH}{H - \delta_1 - \delta_2} - \frac{2q_w^{''} z}{\rho_f \left(H - \delta_1 - \delta_2 \right) \left(c_{p,f} \Delta T_{sub,in} + h_{fg} \right)} \end{aligned}$

Momentum Balance:

 $\begin{array}{ll} \text{Heated wall}\\ \text{vapor layer 1:} & G^{2} \frac{d}{dz} \left[\frac{x_{1}^{2}}{\rho_{g} \alpha_{1}} \right] = -\alpha_{1} \frac{dp}{dz} - \frac{\tau_{w,g1}}{A} \frac{P_{w,g1}}{A} \pm \frac{\tau_{i1}}{A} - \rho_{g} \alpha_{1} g \sin\theta \\\\ \text{Central}\\ \text{Liquid layer:} & G^{2} \frac{d}{dz} \left[\frac{\left(1 - x_{1} - x_{2}\right)^{2}}{\rho_{f} \left(1 - \alpha_{1} - \alpha_{2}\right)} \right] = -\left(1 - \alpha_{1} - \alpha_{2}\right) \frac{dp}{dz} - \frac{\tau_{w,f}}{A} \frac{P_{w,f}}{A} \pm \frac{\tau_{i1}}{A} \frac{P_{i1}}{A} \pm \frac{\tau_{i2}}{A} - \rho_{f} \left(1 - \alpha_{1} - \alpha_{2}\right) g \sin\theta \\\\ \text{Heated wall}\\ \text{vapor layer 2:} & G^{2} \frac{d}{dz} \left[\frac{x_{2}^{2}}{\rho \alpha_{1}} \right] = -\alpha_{2} \frac{dp}{dz} - \frac{\tau_{w,g2}}{A} \frac{P_{w,g2}}{A} \pm \frac{\tau_{i2}}{A} - \rho_{g} \alpha_{2} g \sin\theta \end{array}$

Energy Balance:

$$\frac{dx_1}{dz} = \frac{dx_2}{dz} = \frac{q''_w W}{\dot{m} \left(c_{p,f} \Delta T_{sub,in} + h_{fg} \right)}$$

13



Interfacial Lift-off CHF Model

Use separated flow model to determine axial variations of:

$U_{g1,} U_{g2}$	Near-wall vapor layer velocities
U_f	Liquid layer velocity
δ_1, δ_2	Near-wall vapor layer thicknesses

Critical interfacial wavelength

$$k_{c} = \frac{2\pi}{\lambda_{c}} = \frac{\rho_{f}^{"} \rho_{g}^{"} \left(U_{g} - U_{f}\right)^{2}}{2 \sigma \left(\rho_{f}^{"} + \rho_{g}^{"}\right)} + \sqrt{\left[\frac{\rho_{f}^{"} \rho_{g}^{"} \left(U_{g} - U_{f}\right)^{2}}{2 \sigma \left(\rho_{f}^{"} + \rho_{g}^{"}\right)}\right]^{2}} + \frac{\left(\rho_{f} - \rho_{g}\right) g_{n}}{\sigma}$$
where $\rho_{f}^{"} = \rho_{f} \coth\left(2\pi H_{f}/\lambda_{c}\right)$ and $\rho_{g}^{"} = \rho_{g} \coth\left(2\pi H_{g}/\lambda_{c}\right)$
Earth Gravity: $g_{n,l} = g_{e} \cos\theta$ and $g_{n,2} = g_{e} \cos\left(\theta + \pi\right) = -g_{e} \cos\theta$
Microgravity: $g_{n,l} = g_{n,2} = \mu g_{e} \cong \theta$
 $\lambda_{c} = \frac{2\pi \sigma \left(\rho_{f}^{"} + \rho_{g}^{"}\right)}{\rho_{f}^{"} \rho_{g}^{"} \left(U_{g} - U_{f}\right)^{2}}$

Mean pressure difference across wetting front

$$\overline{p_f - p_g} = \frac{4\pi\sigma\delta}{b\lambda_c^2} sin(b\pi)$$

where b = 0.20 is ratio of wetting front length to wavelength Interfacial lift-off criterion

$$\overline{p_f - p_g} = \rho_g \left[\frac{q_w''}{\rho_g h_{fg}} \right]^2$$

Surface energy balance

ASGSR 2015

NASA

$$q_m'' = b q_w''$$



Critical heat flux

$$q_m'' = \rho_g \left(c_{p,f} \Delta T_{sub,in} + h_{fg} \right) \left[\frac{4 \pi \sigma b \sin(b \pi)}{\rho_g} \right]^{1/2} \frac{\delta^{1/2}}{\lambda_c} \bigg|_{z^*}$$

Boiling and Two-Phase Flow Laboratory (BTPFL) November 2015

14







Single-sided Heating

Double-sided Heating











Data Sharing Plans





- NASA Office of Physical Science Informatics (PSI) tasked with organizing and distributing databases to researchers in the field
- Large databases (terabytes of data) will be generated from FBCE ISS experiment
- Purdue-Glenn team will create organization structure for FBCE databases to be provided to (PSI)
- Purdue is presently exploring most effective means for packaging data for ease of use by other researchers using recent FBM Earth's gravity data as example









Data Folders

Contain four filetypes:

- Text Files containing raw sensor data output by data acquisition system
- Matlab Scripts for processing raw data
- Excel Spreadsheets containing all relevant parameters (e.g., pressure drop, heat transfer coefficient, CHF) output by processing script
- Image Files for flow visualization

With subfolders used to group data by operating conditions





Computational Modeling





ANSYS Fluent Modeling

Objectives

- Study flow condensation using CFD solver Fluent
- Select an appropriate phase change model
- Study heat transfer and fluid flow characteristics over a broad range of Reynolds numbers
- Lay foundation for future computational modeling of complicated flow boiling processes

NASA





- Lagrangian
 - Smoothed-Particle Hydrodynamics (SPH) Method: Gingold & Monaghan (1977), Lucy (1977)
 - Multiphase Particle-in-Cell (MP-PIC) Method: Harlow (1955)
- Eulerian
 - Level-Set Method (LSM): Osher & Sethian (1955)
 - Volume-Of-Fluid (VOF) Method: Hirt & Nichols (1981)
 - Coupled Level-Set/Volume-Of-Fluid (CLSVOF) Method: Sussman & Puckett (2000), Enright *et al.* (2002), Tomar *et al.* (2005)

Eulerian-Lagrangian

- Front Tracking Method: Unverdi & Tryggvason(1992), Tryggvason et al. (2001)



Rankine-Hugoniot jump condition

Mao (2009). Michita & Thome(2010), Sun et al. (2012)

$$q_{i}^{\prime\prime} = -k_{eff} \nabla T_{i} \times \vec{n} = \vec{m}^{\prime\prime} h_{fg}$$

$$S_{g} = -S_{f} = \vec{m}^{\prime\prime} \left| \nabla \alpha_{g} \right| = \frac{k_{eff} \left(\nabla \alpha \times \nabla T \right)}{h_{fg}}$$
where $k_{eff} = \partial_{g} k_{g} + \partial_{f} k_{f}$

Schrage Model (1953)

Kartuzova & Kassemi (2011), Magnini et al. (2013) ...

$$S_{g} = -S_{f} = \dot{m}'' \left| \nabla \alpha_{g} \right|$$

where $\dot{m}'' = \frac{2}{2 - \gamma_{c}} \sqrt{\frac{M}{2\pi R}} \left[\gamma_{c} \frac{P_{g}}{\sqrt{T_{g}}} - \gamma_{e} \frac{P_{f}}{\sqrt{T_{f}}} \right]$

Lee Model (1980)

NASA

Wu et al.(2007). Yang et al.(2008), Fang et al.(2010) ...

$$S_{g} = -S_{f} = r_{i} \partial_{g} r_{g} \frac{\left(T - T_{sat}\right)}{T_{sat}} \quad \text{for condensation} (T < T_{sat})$$
$$S_{g} = -S_{f} = r_{i} \partial_{f} r_{f} \frac{\left(T - T_{sat}\right)}{T_{sat}} \quad \text{for evaporation} (T > T_{sat})$$





25



Rankine-Hugoniot jump condition

1. Ease of implementation

Mao (2009). Michita & Thome (2010), Sun *et al.* (2012) ...

Pros:

Cons:

- 1. Allows for phase change only along interface
 - 2. Cannot maintain saturation temperature

Schrage Model (1953)

Kartuzova & Kassemi (2011), Magnini et al. (2013)...

Pros:

Cons:

- 1. Successfully used for evaporating & condensing films
- 1. Requires use of empirical coefficient y
- 2. Allows for phase change only along interface

Lee Model (1980)

Wu et al. (2007). Yang et al.(2008), Fang et al. (2010) ...

Pros:

1.

- Ease of implementation
- 2. Successfully used for condensation processes

Cons:

- 1. Not applicable for subcooled boiling
- 2. Requires use of empirical coefficient r_i



Computational Domain, Governing Equations and Boundary Conditions

Computational Domain



Governing Equations

Continuity Equations
 Liquid Phase:

$$\frac{\partial}{\partial t} (\alpha_f \rho_f) + \nabla \times (\alpha_f \rho_f \vec{u}_f) = S_f$$

- Vapor Phase: $\frac{\partial}{\partial t} (\alpha_g \rho_g) + \nabla \times (\alpha_g \rho_g \vec{u}_g) = S_g \qquad S_g = -S_f = r_i \partial_g r_g \frac{(T - T_{sat})}{T_{sat}}$
- Momentum Equation $\frac{\partial}{\partial t} (\rho \vec{u}) + \nabla \times (\rho \vec{u} \vec{u}) = -\nabla P + \nabla \times [\mu (\nabla \vec{u} + \nabla \vec{u}^T)] + \rho \vec{g} + \vec{F}$
- Energy Equation $\frac{\partial}{\partial t} (\rho E) + \nabla \times (\vec{u} (\rho E + P)) = \nabla \times (k_{eff} \nabla T) + Q \quad (Q = h_{fg}S_f)$

Boundary Condition

- Axisymmetric centerline
- k-ω SST turbulence model
- Inlet uniform velocity from experimental data
- Wall heat flux from experimental data



Void Fraction Results: Climbing Film Regime





Climbing film









Future Computational Modeling





- Fluent is able to accurately replicate experimental results, but "tuning" necessary for phase change model means it is not a reliable predictive tool
- Difficult to work with Fluent because solver code is proprietary
- Fluent very robust and can tackle wide range of problems, making it slower than a dedicated research code
- Fluent does not utilize cutting edge multi-phase computational techniques





Current Work

- Working with Prof. Carlo Scalo's group at Purdue University to develop a 2-D code using best available computational techniques
- Performing comparisons with Fluent to quantify in-house solver performance



Future Work

- Use proposed 2-D code to run select cases which can be represented reasonably well by 2-D domains (e.g., axi-symmetric flow condensation, slug flow)
- Scale 2-D code up to 3-D, highly parallelized solver, which will include turbulence effects, to be run on Purdue supercomputing cluster
- Begin comparing data for transient cases with 3-D geometry to prior experimental studies



The Flow Boiling and Condensation Experiment is supported by NASA grant NNX13AB01G.

This work was supported by NASA Space Technology Research Fellowship NNX15AP29H.

Thanks to Professor Scalo and Victor Sousa for creating the Fluentcomparison figures.

