



Non- Topographic Space-Based Laser Remote Sensing

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Outline

- Introduction
- Space Laser Altimetry Instruments
- Progress on On-Going Programs
 - Earth Science
 - Planetary Science
 - Heliophysics
 - Astrophysics
- Summary



INTRODUCTION



EARTH SCIENCE APPLICATIONS

CO₂ LIDAR

CH₄ LIDAR



CARBON DIOXIDE LIDAR



NASA's ASCENDS Mission



Active Sensing of CO₂ Emissions over Nights, Days, and Seasons (ASCENDS) Mission

Science Mission Definition Study

Draft

ASCENDS Ad Hoc Science Definition Team:

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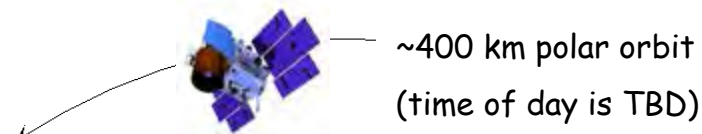
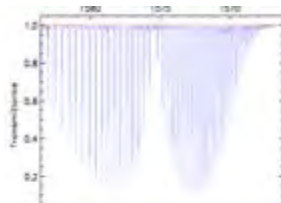
April 15, 2015

Avail from:

http://cce.nasa.gov/ascends_2015/index.html

Measures:

- CO₂ tropospheric column
- O₂ tropospheric column
- Cloud backscattering profile



~400 km polar orbit
(time of day is TBD)



Requirements for CO₂ Mixing Ratio:

Random error: ~ 1 ppm in ~100 km along track, or
~ 0.5 ppm in ~10 sec over deserts

Bias: < 0.5 ppm (< 1 part in 800)

Lower errors provide more benefit for flux est's.

NASA National Aeronautics and Space Administration | + NASA Homepage | + NASA Carbon Cycle and Ecosystems

ASCENDS 2015 Workshop

Home | Agenda and Presentations | Publications | ad hoc Science Definition Team

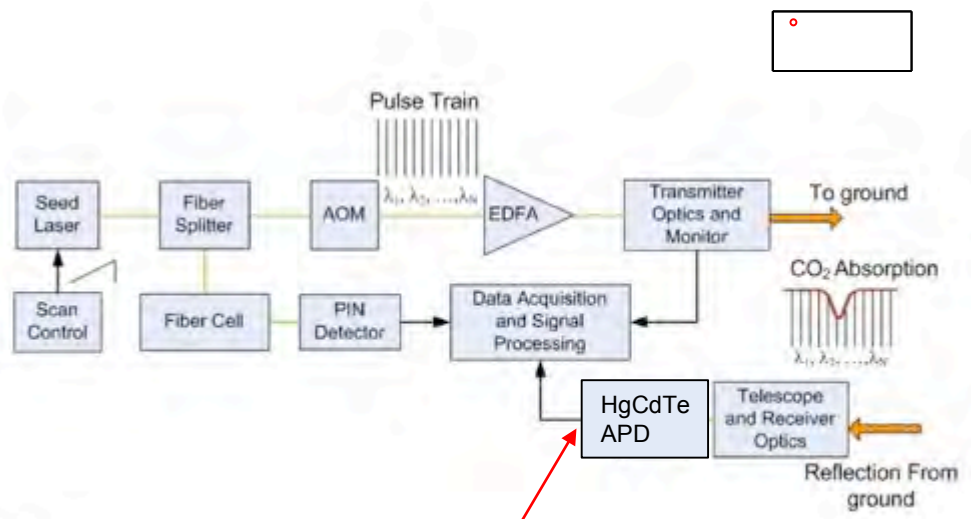
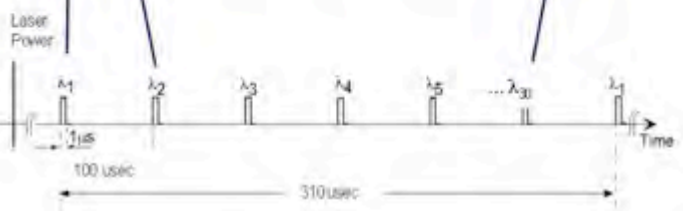
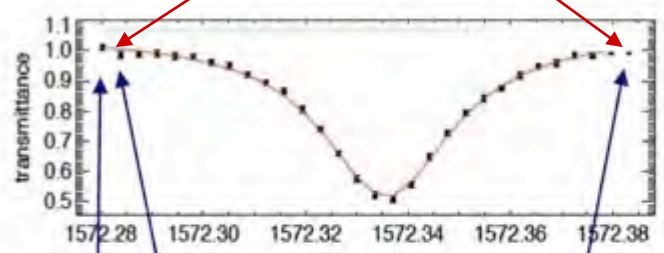
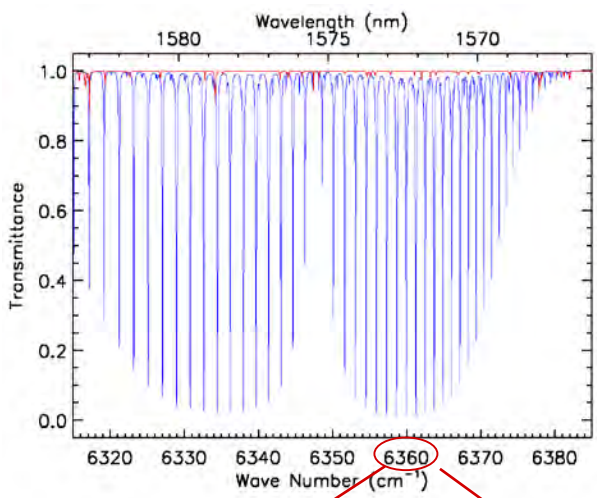
ASCENDS Workshop June 19, 2015 California Institute of Technology

This workshop will be held after the 11th International Workshop on Greenhouse Gas Measurements from Space at California Institute of Technology, Pasadena, California, USA, Friday, June 19th.
View their website for logistical information: <https://sites.google.com/site/wggms11/>



CO₂ Sounder Approach: Airborne CO₂ Line Sampling & Absorption line analysis

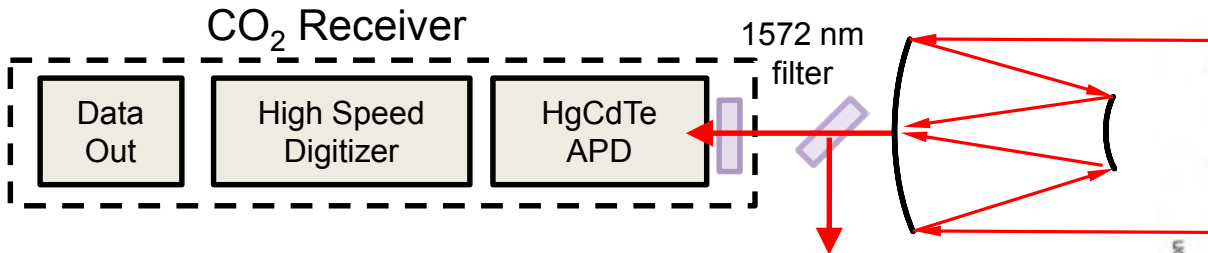
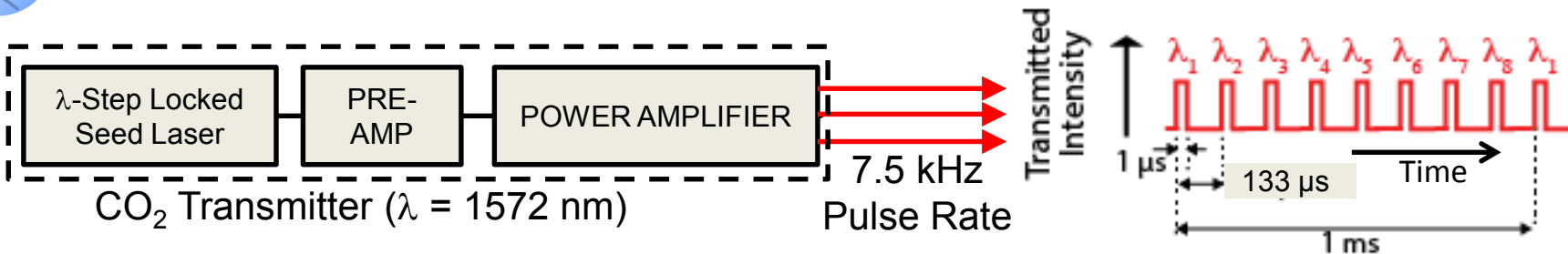
- *Line at 1572.33 nm*
- *Lidar* measures “dots” (wavelength samples) to all scattering surfaces
- *Post flight: Retrievals* (based on model atmosphere) calculates range, normalized line shapes & solves for best fit concentration



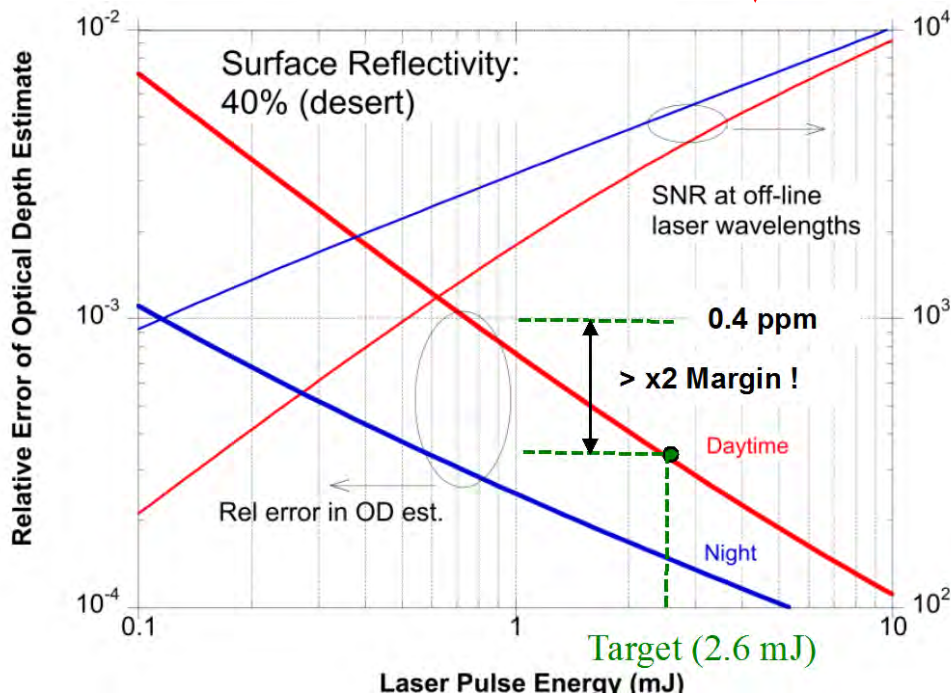
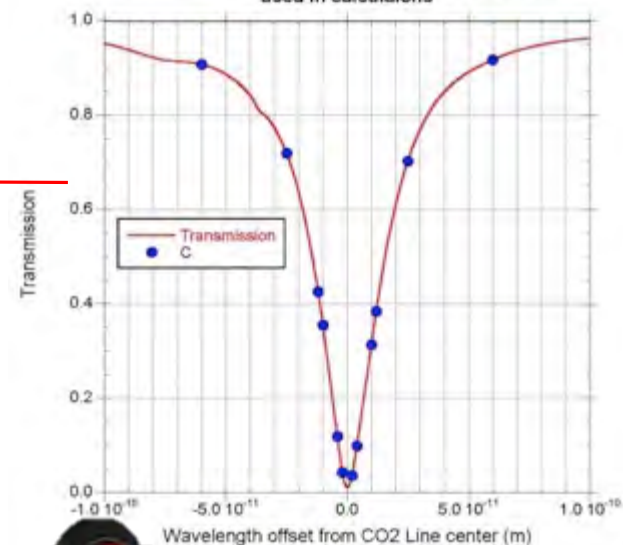
HgCdTe APD Detector in 2014 flights



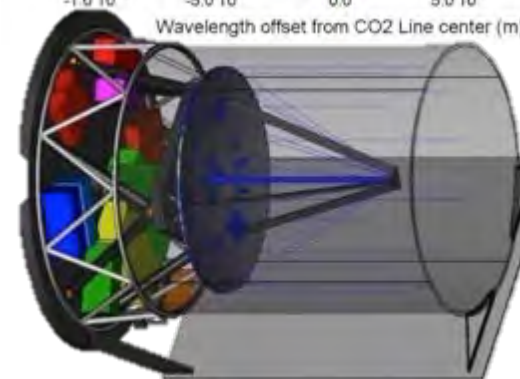
Scaling CO₂ Sounder Lidar to Space



Calculated CO₂ Line Shape & Wavelength Sampling used in calculations



SNR at off-line Wavelengths





Laser Requirements

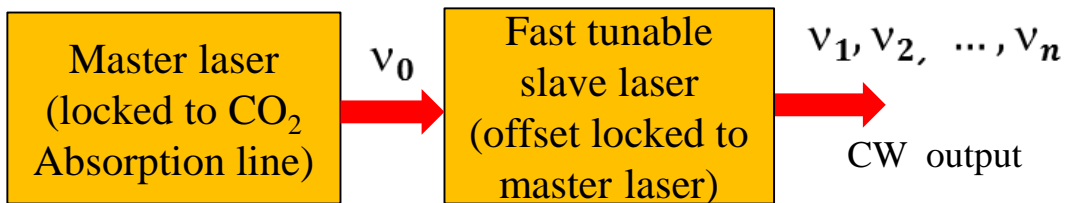
Performance Parameter	<u>Laser Transmitter</u>
Center Wavelength	Nominally centered at 1572.335 nm

Performance Parameter	<u>Laser Transmitter</u>
Center Wavelength	Nominally centered at 1572.335 nm
Linewidth (each wavelength channel)	≤ 100 MHz (TBR)
Pulse repetition frequency	7.5 KHz
Pulse Width	1-1.5 μs
Pulse Energy	>3.2 mJ/pulse (goal); >2.6 mJ/pulse (operating, 18% derating)
PER [TBR]	20 dB (TBR)
Wall-plug Efficiency	> 6%

Pulse energy stability (long term – 1 hr)	< 5%
Optical Output	Free space, PM, ~100 μ rad divergence, beams co-aligned to better than ~20 μ rad
Beam quality	$M^2 < 1.3$ (TBR)
PER [TBR]	20 dB (TBR)
Environmental	Launch to ISS (TBR)
Wall-plug Efficiency	> 6%

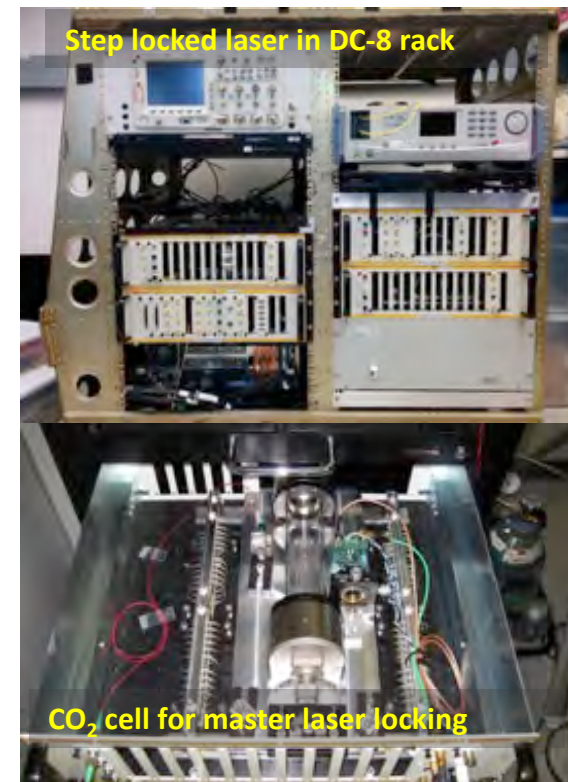


Seed laser approach



- DFB master laser is locked to CO₂ reference cell.
- Digital Supermode-DBR slave laser is dynamically offset-locked to the master DFB laser using an optical phase-locked loop (OPLL).
- Slave laser output is source for remaining laser stages.
- Frequency-stepped pulse train carved by MZM and subsequently amplified.
- Demonstrated laser frequency noise suppression (to < 0.2 MHz), tuning speed (< 40 μs) and tuning range (~32 GHz) – all satisfy or exceed ASCENDS requirements
- US Patent granted (July 2015).

Ref: Numata, K., et al., *Optics Express* Vol. 20, Issue 13, pp. 14234-14243 (2012).
 Numata, K., et al. *Applied Optics* 50.7 (2011): 1047-1056.
 Chen, et al., US Patent No.: US 9,065,242 B2 (45), Jun. 23, 2015.

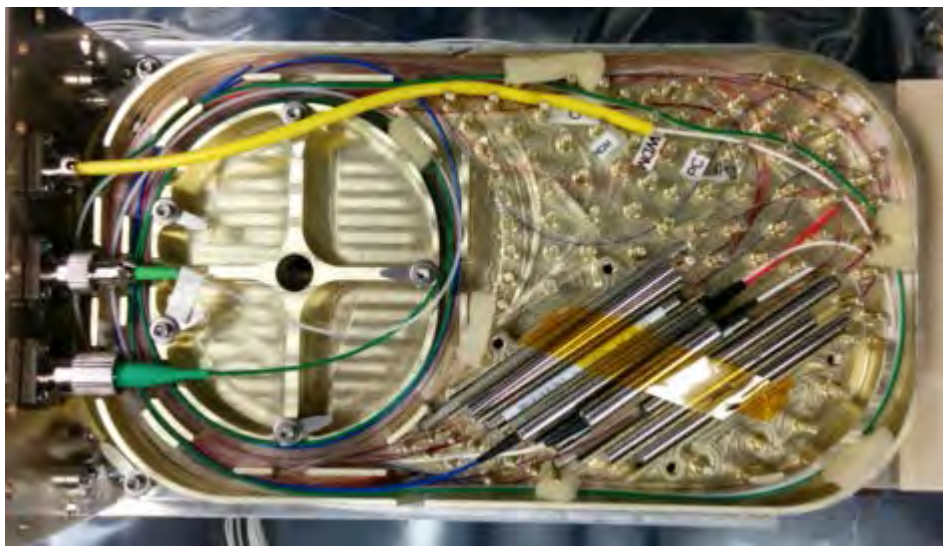
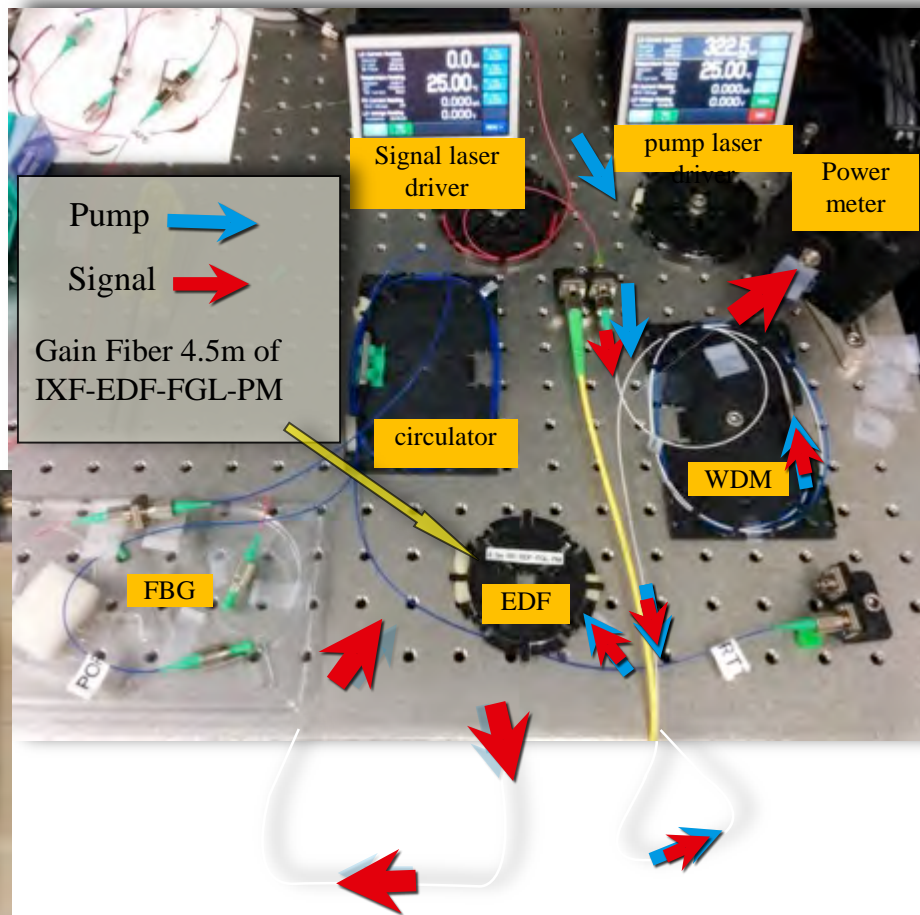




Pre-Amplifier Development

Preliminary Stage 1 Pre-amplifier breadboard Results

- 4.5 meter IXF-EDF-FGL-PM
- Core pumped
- Pump power 200mW @ 976 nm
- Input 9mW of 1572.3nm
- Measured Output ~80mW
- Measured Gain ~10dB
- ASE filtered with FBG

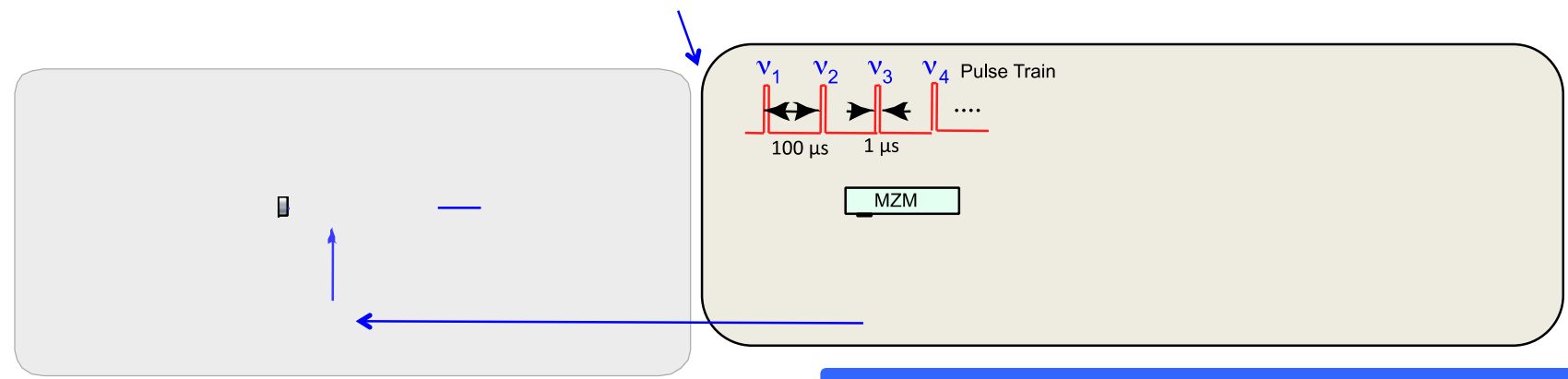


Packaging Approach



Power Amplifier

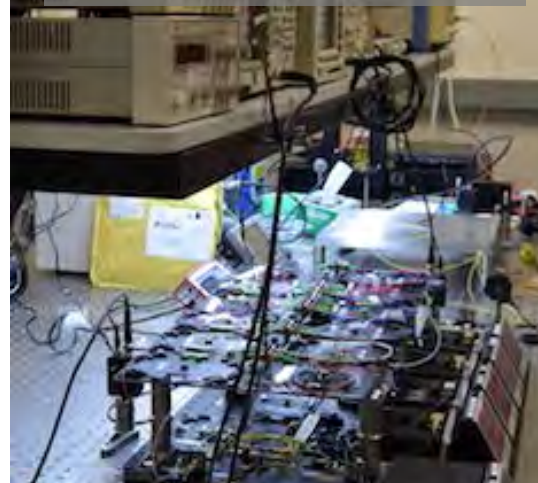
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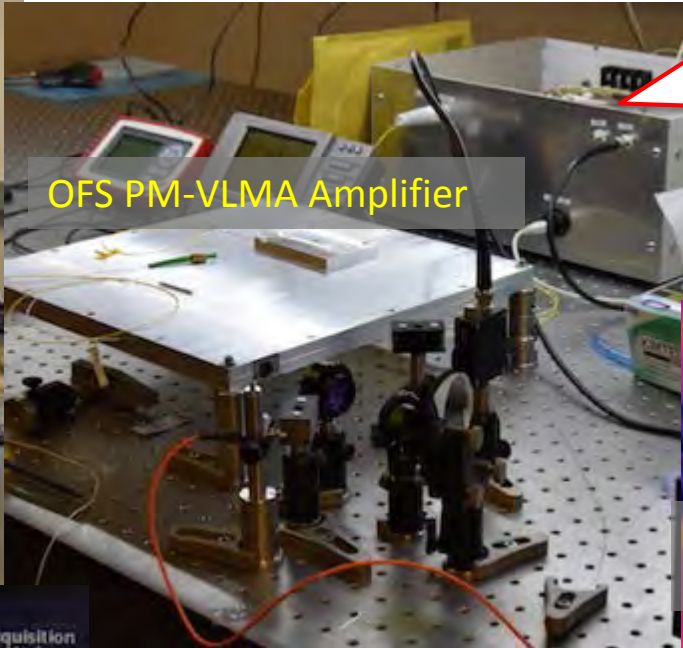


OFS PM-VLMA Power Amplifier Performance

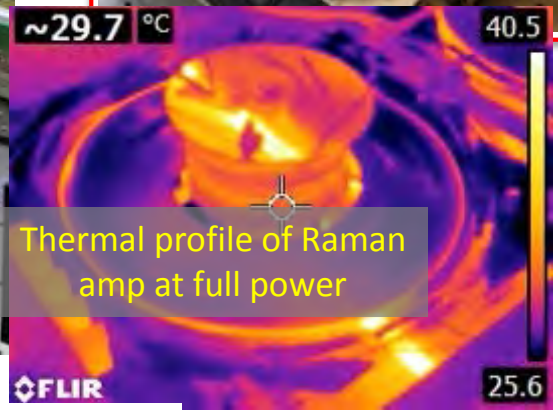
GSFC In-House Seed Laser & Pre-amplifier for VLMA test



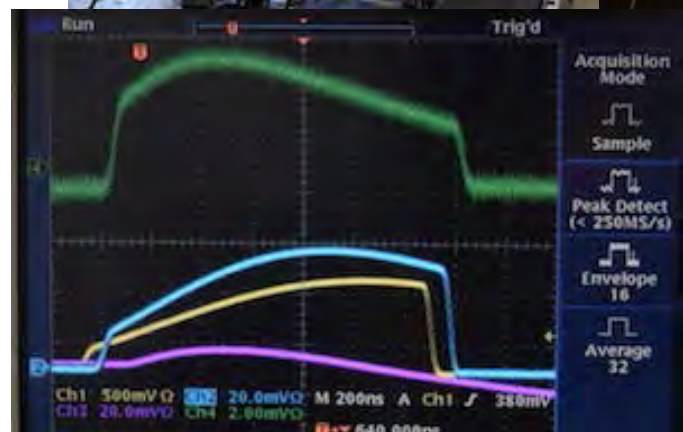
OFS PM-VLMA Amplifier



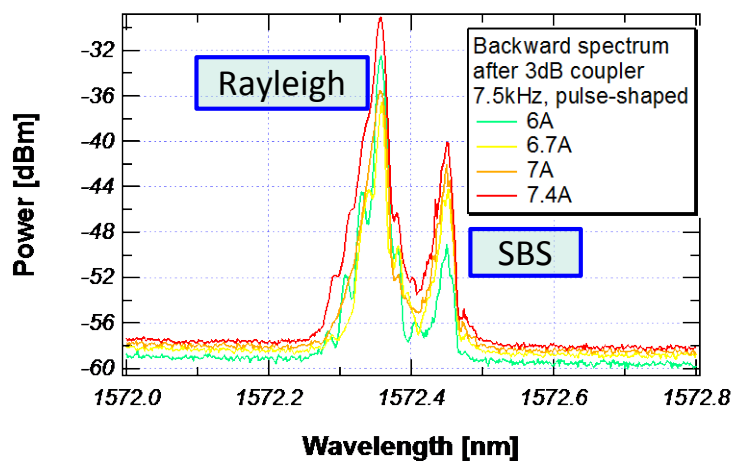
Raman Pump



Thermal profile of Raman amp at full power



Pre-shaped pulses from seed (yellow) and preamplifier (blue) and VLMA Output Pulse shape (Green); Red is ASE from VLMA



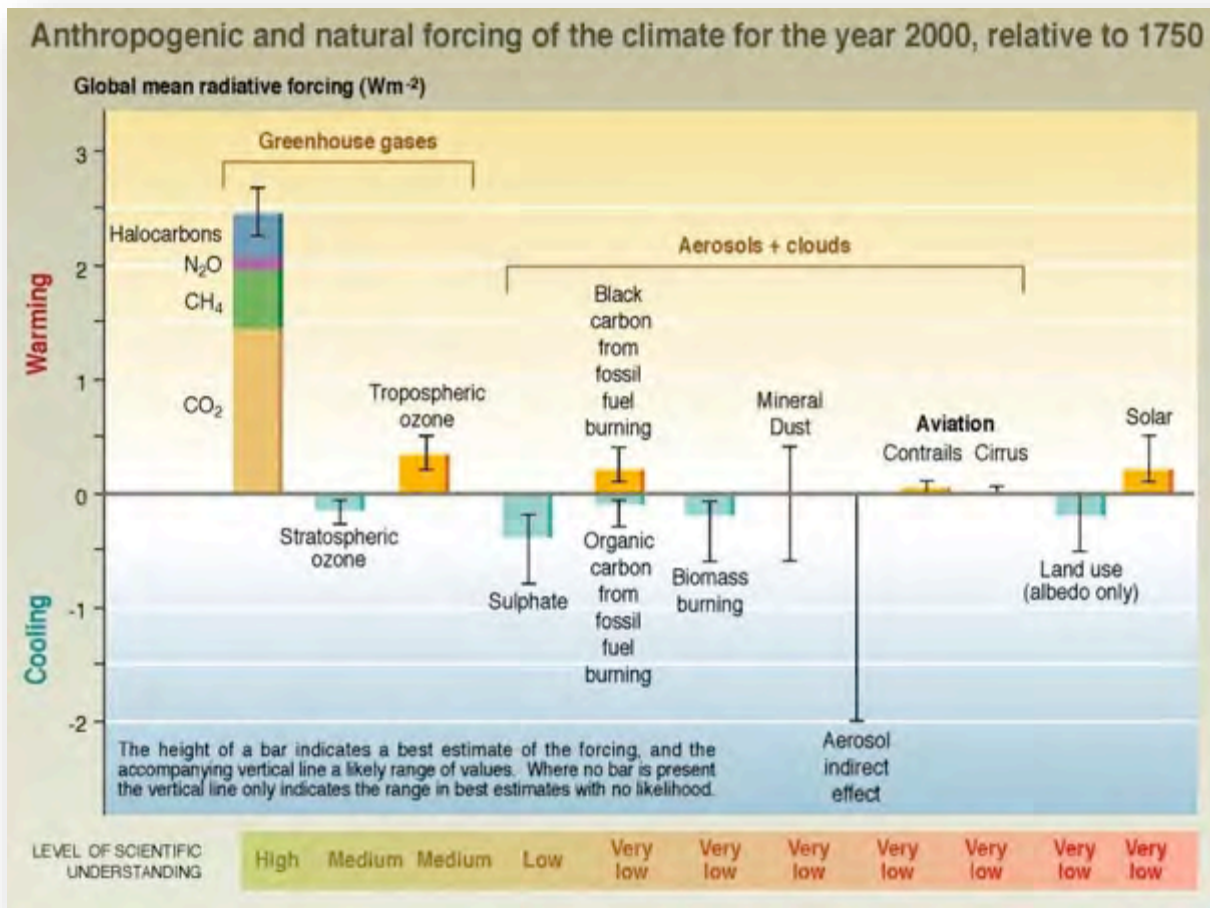
Breadboard Performance (w/o optimized pre-shape pulse – **460 μ J at 7.5 kHz, 1 μ s pulse**
6X would provide ~2.8 mJ



METHANE LIDAR



Methane radiative forcing



Source: IPCC Report 2007

- CH₄ is a strong greenhouse gas (~×23-25 higher radiative forcing than CO₂ on a per molecule basis).
- Earth Science Decadal Survey (NRC 2007):
 - “Ideally, to close the carbon budget, methane should also be addressed, but the required technology is not now obvious. **If appropriate and cost-effective methane technology becomes available, methane capability should be added.**”



Methane “Arctic Time Bomb”

- Increasing concern about CH₄ in the Arctic: ***“Is a Sleeping Climate Giant Stirring in the Arctic?”***. Large amounts of organic carbon are stored as CH₄ and CO₂ in the Arctic permafrost. Thawing Arctic permafrost soil, is a cause for concern as a rapid, positive greenhouse gas/climate feedback. In addition, large but uncertain amounts of CH₄ are sequestered as gas hydrates in shallow oceans and permafrost soils, which are also subject to potential rapid release.





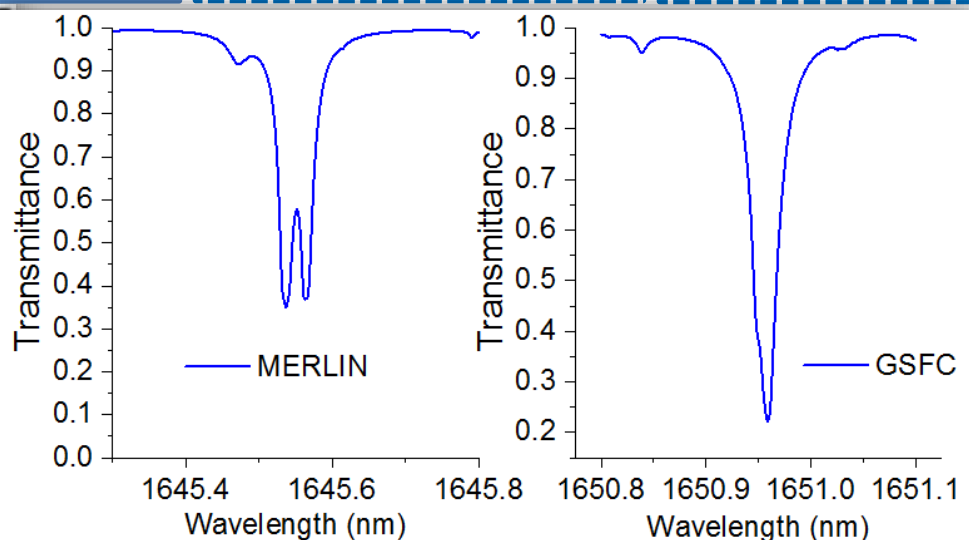
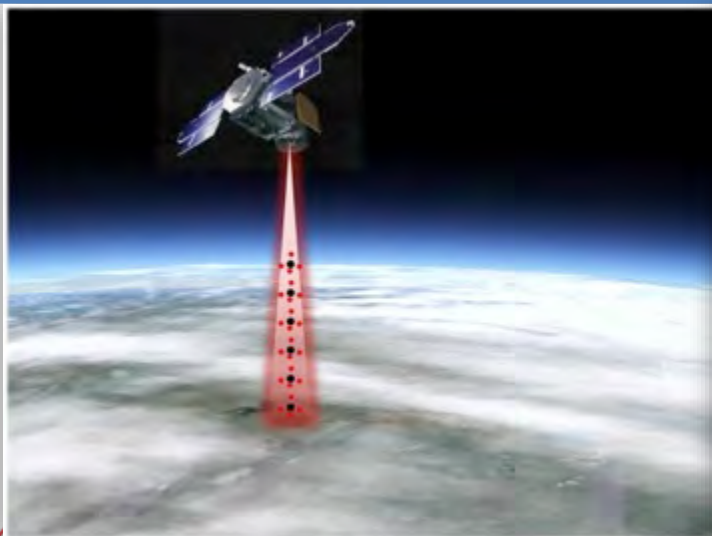
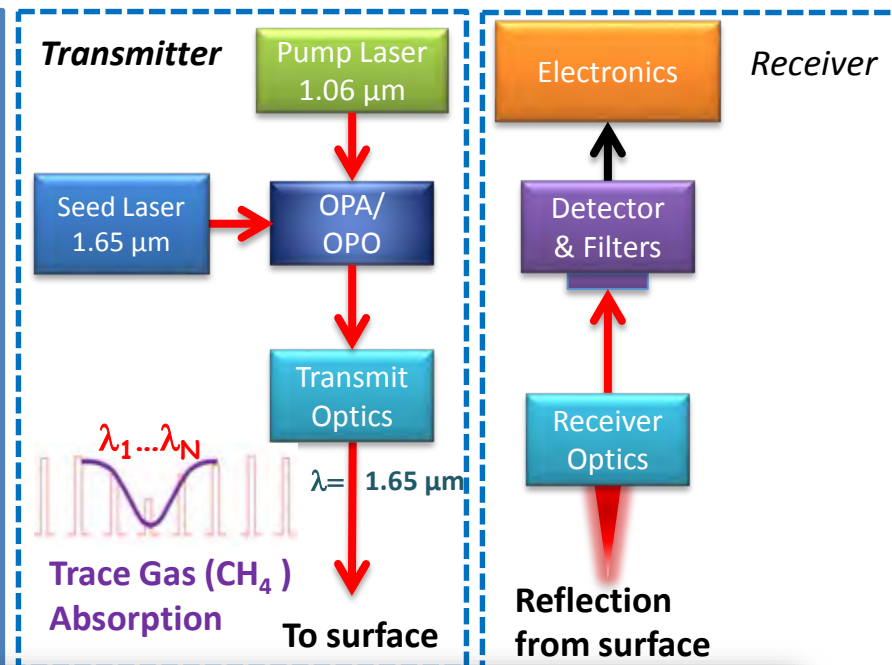
CH₄ Detection with Lidar

- Transmitter (Laser) technology

- Current (optimum) Wavelength for CH₄ Earth Detection: ~1.64-1.66 μm
- Optical Parametric Oscillators (OPO) and Optical Parametric Amplifiers (OPA) are the best solutions currently available for a transmitter.
- Other options (Er:YAG and Er:YGG) also possible

- Receiver (Detector) Technology

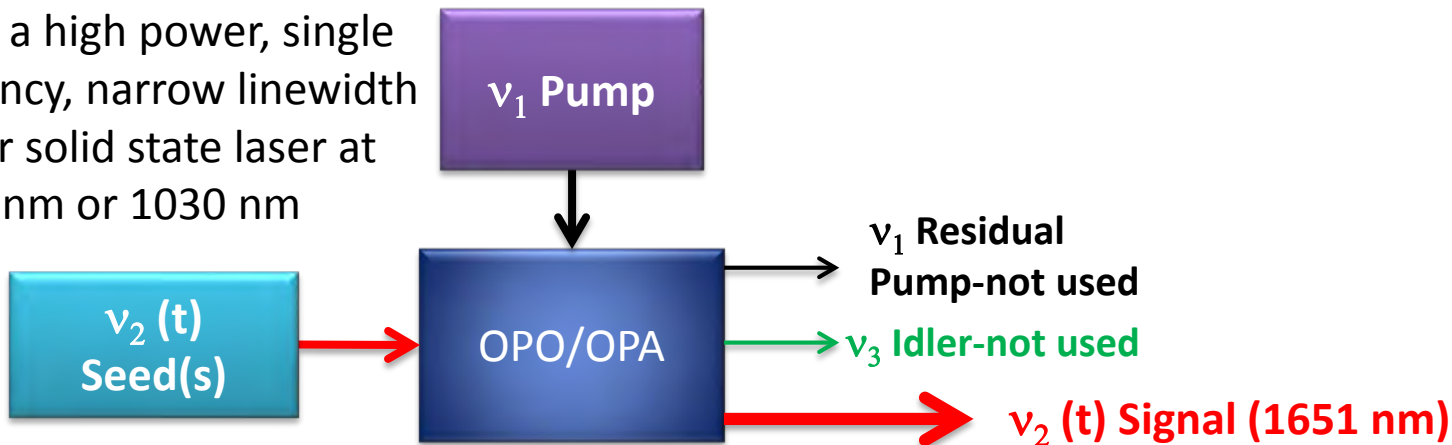
- DRS e-APD





Methane Transmitter Components

Pump: a high power, single frequency, narrow linewidth fiber or solid state laser at ~1064 nm or 1030 nm



Seed: a (low) power, single frequency diode laser at 1651 nm.

Optical Parametric Oscillator (OPO) or Optical Parametric Amplifier (OPA).

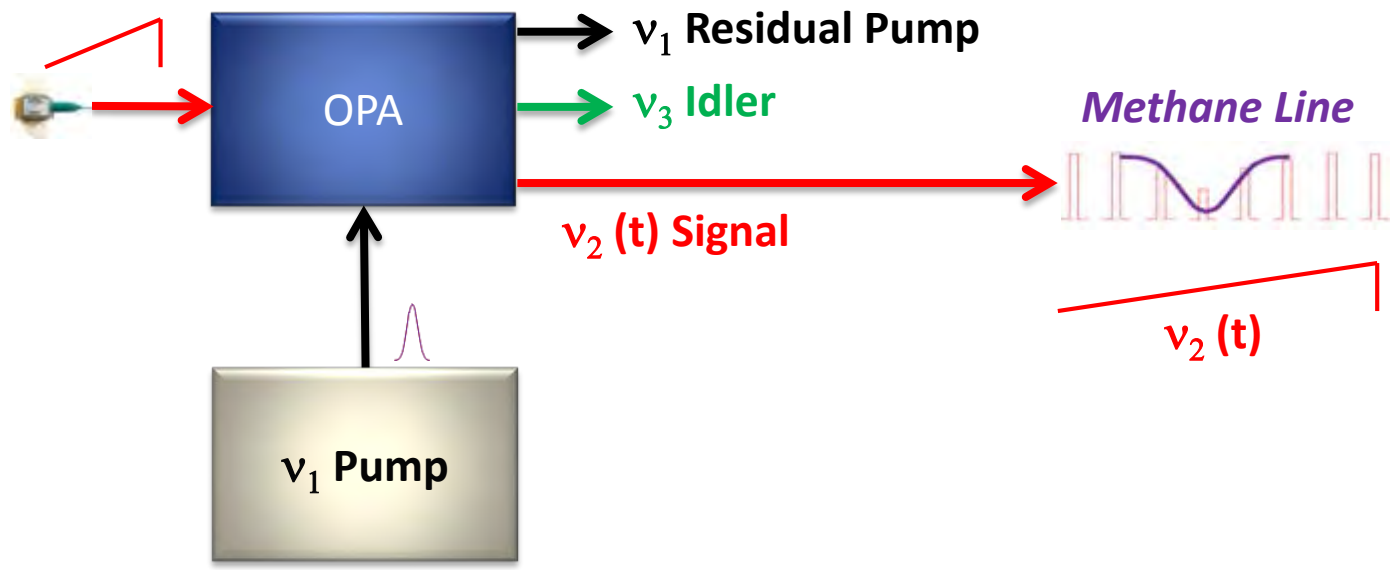
A non-linear crystal that amplifies the seed laser to the energy needed for space **without** degrading the spectral characteristics



All components are critical and require technology development



Transmitter Technology - OPA



OPA: OPA samples the CH_4 line at several wavelengths using a single, continuously tuned seed laser.

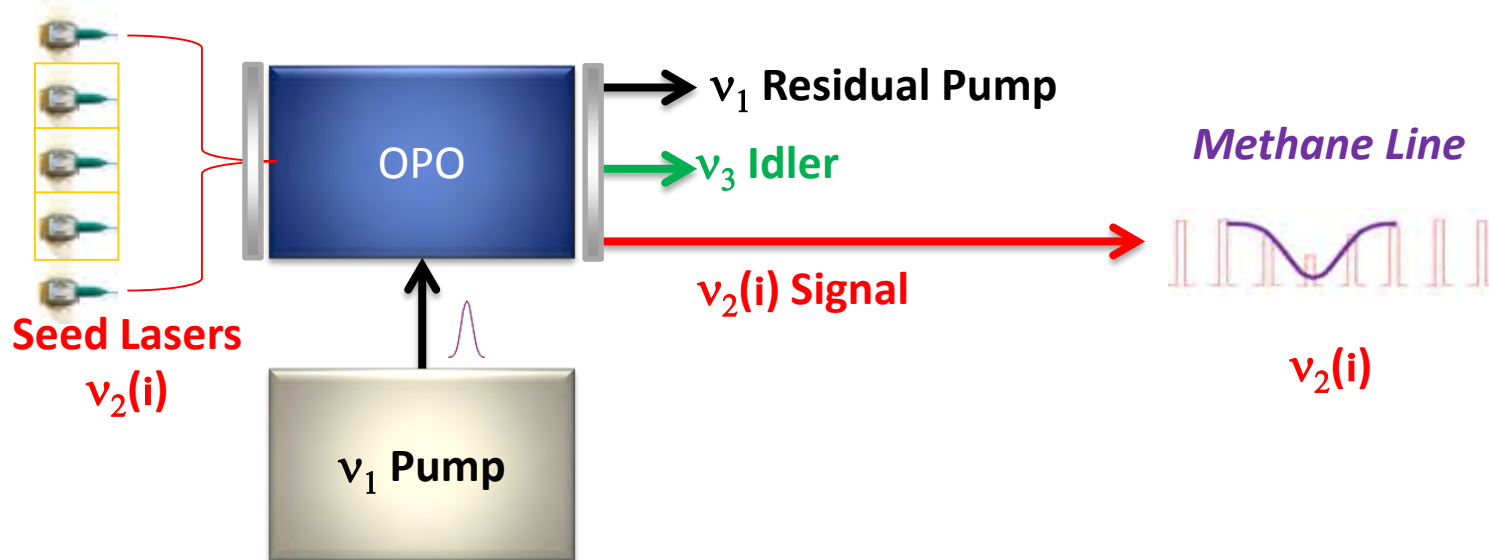
Easy to align, easy to tune, hard to achieve power scaling while maintaining narrow linewidth.

Need to increase seed laser power.

Burst mode pump laser an alternative approach.



Transmitter Technology - OPO



- ***OPO samples the CH_4 line at several discrete wavelengths using multiple seed lasers.***
- ***Complicated to align and tune; power scaling easier to achieve while maintaining narrow linewidth.***
- ***Master laser locked to the CH_4 absorption***
- ***Cavity locked to master laser***
- ***Remaining seed lasers offset-locked by integral number of cavity modes***



Summary

- Methane measurements are needed over all latitudes and seasons.
- Airborne demonstration with DRS detector and two candidate architectures for a CH₄ transmitter: 20-wavelength OPA and 5-wavelength OPO.
 - DRS detector worked well during flight. Its performance is close to CO₂ (as expected).
 - Both OPA and OPO performance was better than expected during flight.
 - 20 wavelength OPA gave better results (precision) but there is no simple path to scale the power of the transmitter to space.
 - 5 wavelength OPO gave good results (when detector was not saturated). Can use more wavelengths but is already close to the energy we need for space.
- Transmitter Plan - Many different approaches and options for laser transmitter were investigated.
 - Er:YGG/Er:YAG (Baseline)
 - OPO (Backup):
 - Burst Mode OPA (no easy path to space)
 - Single Pulse OPA (no easy path to space)
- **Plan will provide at least one baseline configuration (Er:YGG) and at least one backup configuration (OPO) that can be flown.**
- Leveraged SBIR, IRAD, and Tipping Point programs.
- Significant progress has been made with modest investments.

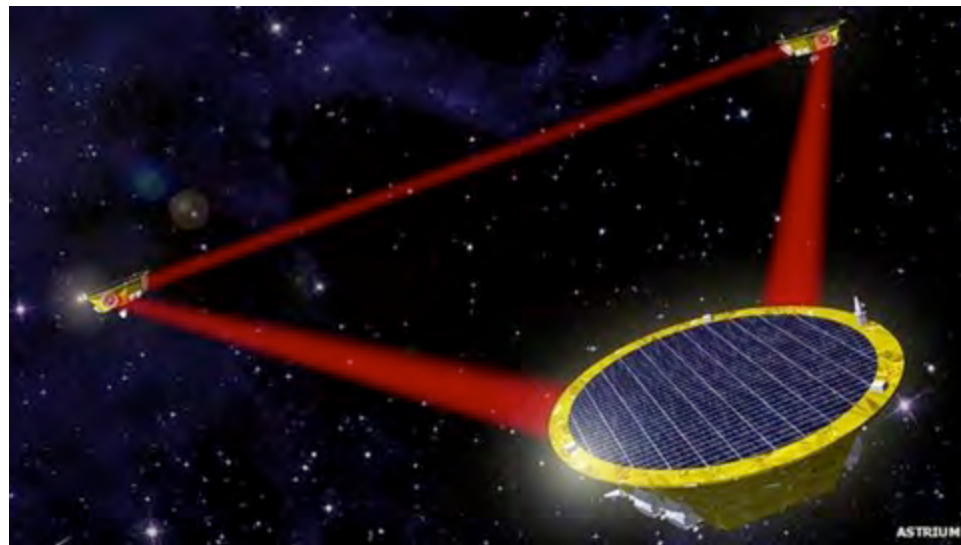
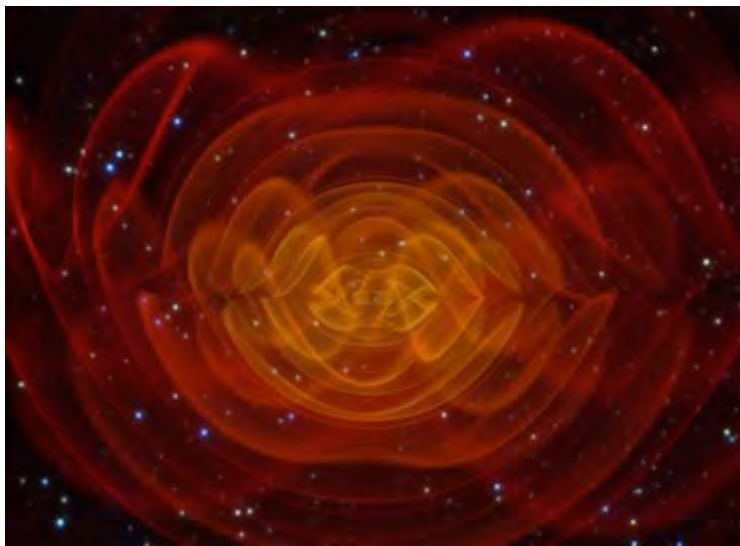


ASTROPHYSICS



Science of Gravitational Waves/eLISA

- eLISA (evolved Laser Interferometer Space Antenna)
 - Planned late 2020s ~ early 2030s launch
 - <https://www.elisascience.org>
- New way to observe the universe
 - Expected astrophysical GW sources in eLISA
 - Black holes and galaxy formation
 - Merging massive black holes in galaxies at all distances
 - Massive black holes swallowing smaller compact objects





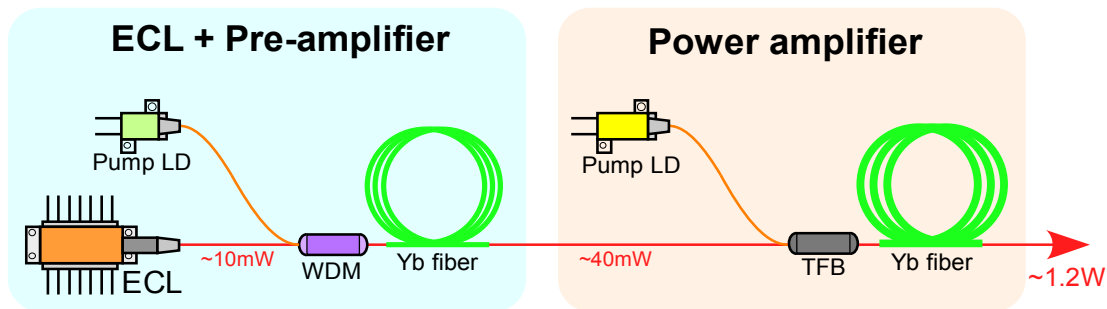
eLISA laser requirement

- Unique requirements for precision interferometry
 - Continuous-wave laser
 - Single-mode & frequency
 - Extreme stability in frequency & intensity

Power	λ (nm)	Intensity noise (/Hz ^{1/2})	Frequency noise (Hz/Hz ^{1/2})	Differential phase noise (rad/Hz ^{1/2})	Lifetime
1.5 W	1064	10 ⁻⁴ (@ 10 ⁻³ Hz) 10 ⁻⁸ (@ 10 ⁷ Hz)	300 (@ 10 ⁻² Hz)	6x10 ⁻⁴ (@ 10 ⁻² Hz)	2.5 years

- eLISA laser configuration

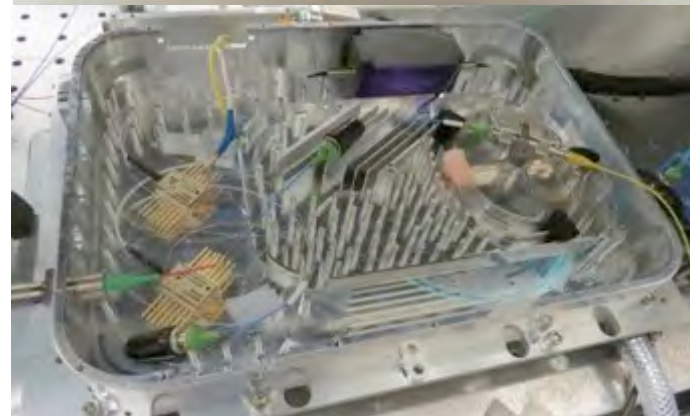
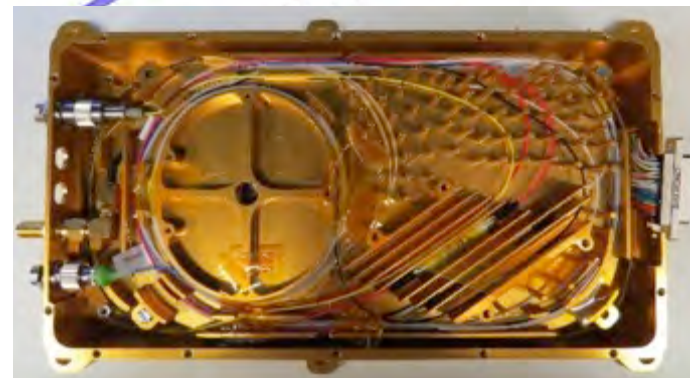
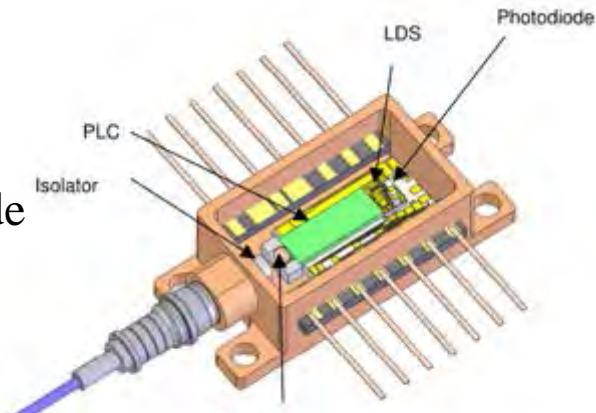
- All-fiber master oscillator power amplifier (MOPA)





Current Activities & Future Work

- **Master oscillator**
 - Working with RIO to build low noise planar-waveguide ECL
 - Environmental test to be completed soon at RIO
 - On-chip phase modulator planned
- **Pre-amplifier**
 - ~150mW CW output
 - Prototype was built, packaged, and tested
 - Environmental testing being done
- **Power amplifier**
 - All-fiber design, ~2.4W CW output
 - Noise measurement & stabilization being done
 - Full laser system noise & reliability tests planned





PLANETARY SCIENCE

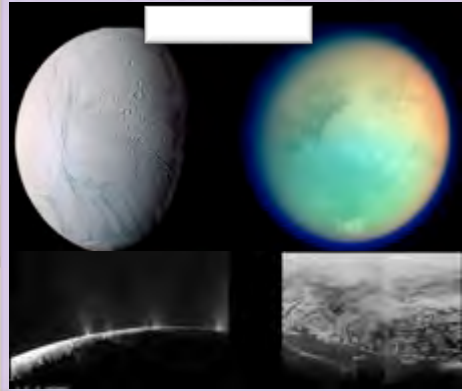


TIME OF FLIGHT MASS SPECTROMETER

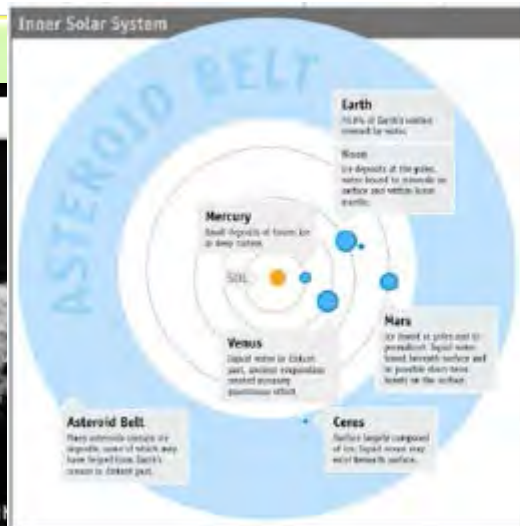


Science Motivation

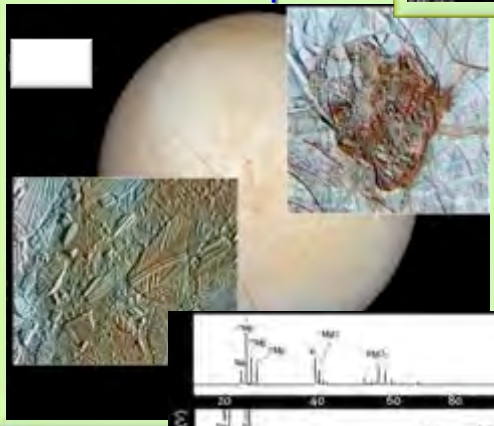
Enceladus & Titan



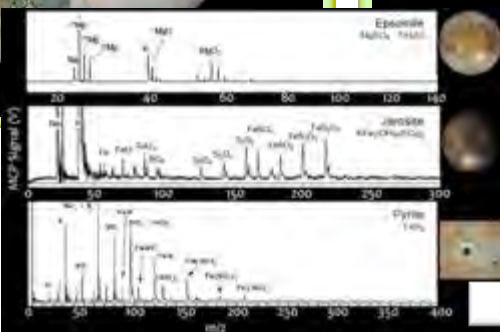
Comets



Europa



Ceres



All these bodies contain volatiles, including water just waiting to be analyzed

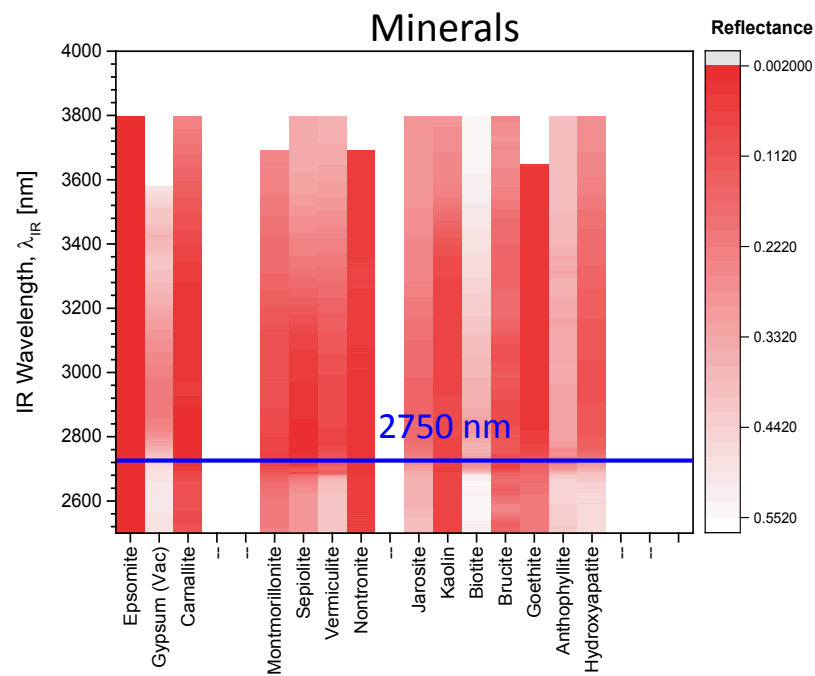
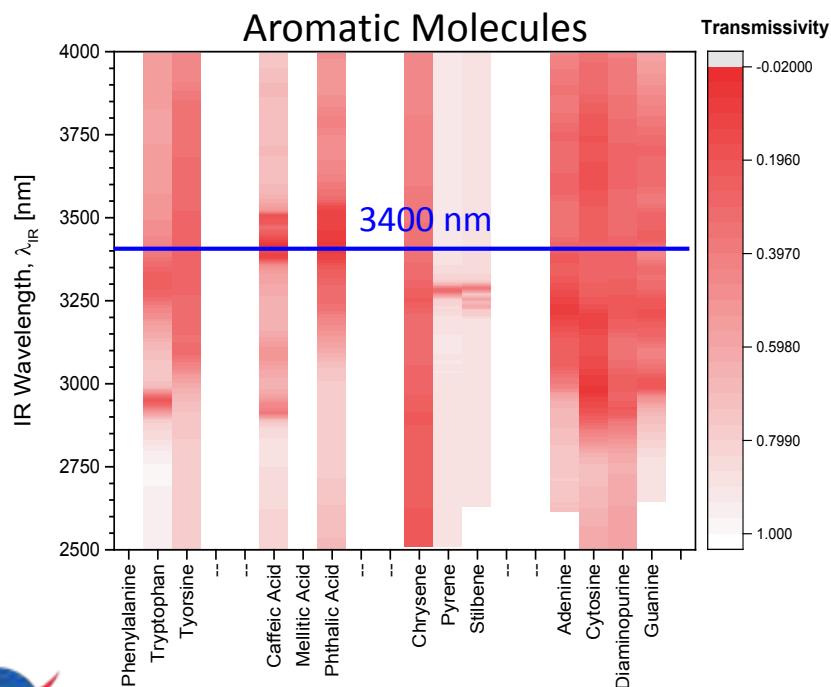
Our science motivation is to seek a solution with “Universal” detector for

- Comprehensive sample analysis
- Flexibility to adapt for different mission architectures including flybys, orbiters, landers, and/or rovers!



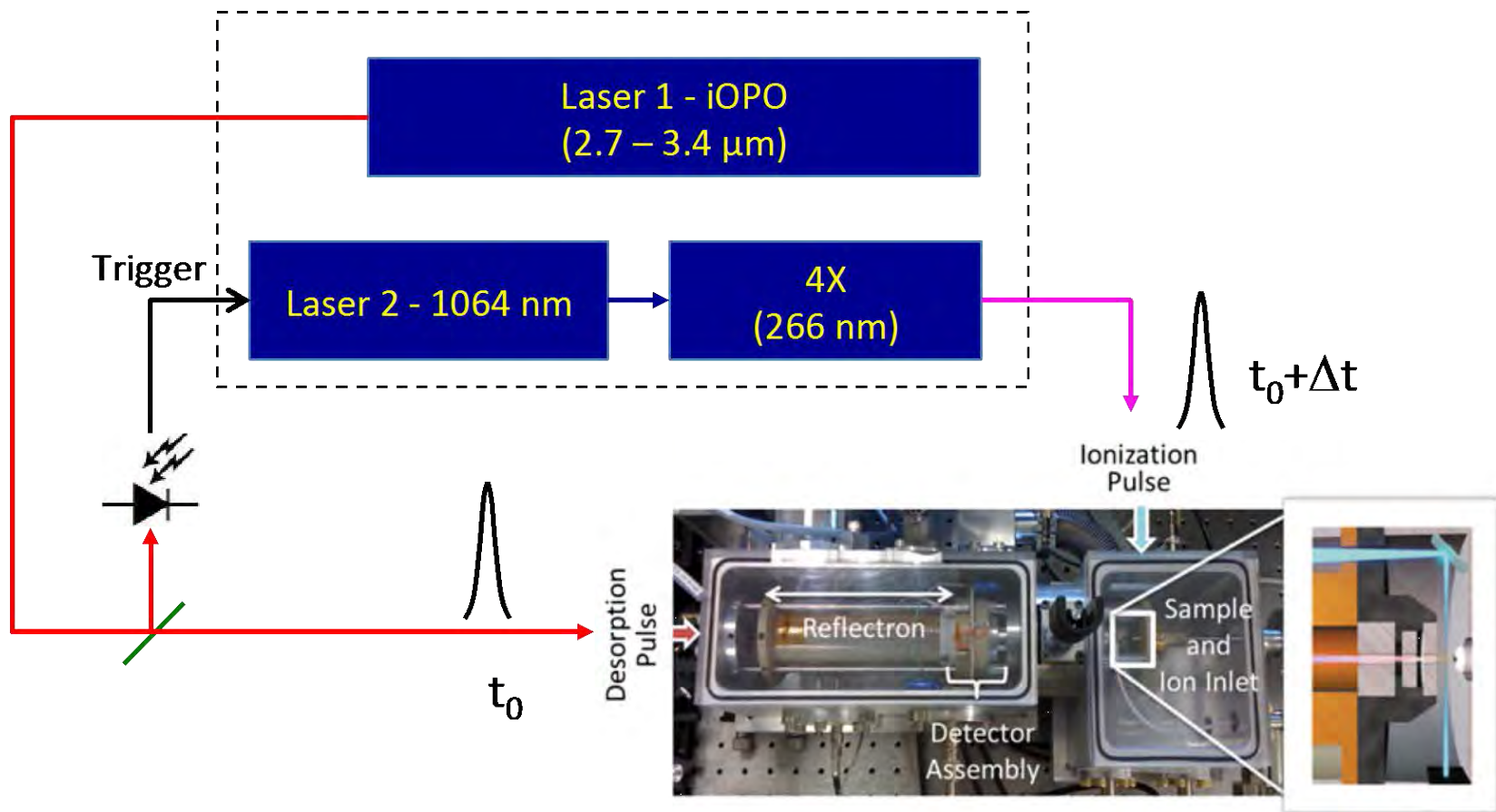
L2MS Instrument Overview

- Wavelengths are selected based on key vibrational and electronic resonances in the targeted species aligned with the organic diversity and mineralogy expected for future planetary missions of high priority to NASA
 - 2.75 μm - IR vibrational resonances of hydrated minerals
 - 3.4 μm - C-H vibration resonance of organic species
 - 266 nm - coincides with a short-lived metastable state in many aromatic molecules
- Matching MIR laser wavelength allows for selective desorption





L2MS Instrument Overview



- Typical delay between Laser 1 and Laser 2 (Δt) range between 0.3-2 μs



L2MS Laser Requirements

Lasers Requirement	MIR Laser	UV Laser
Pulse Repetition Frequency (PRF)	1 – 20 Hz	1 – 20 Hz
Wavelength	2.8X μm and $3.40\pm 0.05 \mu\text{m}$	266 nm
Energy	$\sim 100 \mu\text{J}$	$\sim 18 \mu\text{J}$
Pulse Width	$< 7 \text{ ns}$	$< 7 \text{ ns}$
Peak Power	$\sim 14 \text{ kW}$	$\sim 2.5 \text{ kW}$
Peak Intensity (assuming 100 μm beam diameter)	180 MW/cm^2	$\sim 30 \text{ MW/cm}^2$
Spectral Width	Few GHz	Few GHz
Timing	t_0	$t_0 + \Delta t$; $\sim 100 \text{ ns} < \Delta t < \text{few } \mu\text{s}$
Laser Lifetime	3 year mission at 10% duty cycle $\sim 64 \text{ Mshots @ } 20 \text{ Hz}$	



Future Work

Laser 1 – MIR Laser

- Test 3.4 μm breadboard with L2MS laboratory instrument and compare with commercial OPO
- Complete 2.75 μm breadboard laser
- Finalize design for dual wavelength (2.75 μm and 3.4 μm) concept

Laser 2 – UV Laser

- Investigate other non-linear optical crystals for SHG and FHG – leverage ICESat-2/ATLAS LBO aging study
- Optimize overall 4th harmonic conversion efficiency
- Test breadboard with L2MS laboratory instrument and compare with commercial UV laser
- Develop epoxy-free opto-mechanical design for mounting optics to minimize UV induced contamination on optical surfaces

Laser Transmitter

- Improve packaging of the laser transmitter for space flight
- Build brass board laser transmitter that will generate both MIR and UV wavelengths on a single laser bench



Preparing for the Future

- Use nonlinear optical processes to generate specific wavelengths for laser spectroscopy
- New approaches in laser architectures
 - High rep rate, lower pulse energy, sub-ns pulse for high resolution mapping (8-10 kHz, >50 μ J, <2 ns)
 - High efficiency laser systems (>15% wall plug)
 - Highly reliable laser systems (multi-Billion shots)
 - High sensitivity detector and detector arrays with low-noise, high speed ROICs
 - linear mode PC in the NIR because of its wavelength advantages
 - Data volume management
 - On-board data processing
 - Data compression
 - High data downlink rate
- Common requirements for all laser based instruments
 - Lifetime
 - Reliability
 - Efficiency

