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NASA Marshall Space Flight Center In-Space Manufacturing Project

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In-Space Manufacturing (ISM)





"If what you're doing is not seen by some people as science fiction, it's probably not transformative enough."

-Sergey Brin



ISM Objective

The AES In-space Manufacturing (ISM) project serves as Agency resource for identifying, designing, & implementing on-demand, sustainable manufacturing solutions for fabrication, maintenance, & repair during Exploration missions. TECHNOLOGY DEVELOPMENT

APPLICATIONS ISM Parts/Systems Design **Database & Test Articles**

Answers WHAT we need to make

- Top-down, quantitative analyses of ISM benefits to crew time, cost, mass, & reliability (w/EMC).
- Provide expertise to NASA **User community on AM** design optimization & materials.
- Test high-impact parts/systems to inform **Exploration technology** requirements (bottomsup).
- Develop In-space Parts Design Database, processes, & materials.



Part/System Requirements, Design, Materials & Processes

Unique Agency Expertise & Leveraging of Industry



'One-stop shop' for AM design, materials, & technology expertise for NASA **User Community.**



Leverage industry to meet NASA needs (i.e. Agency knowledgebase for terrestrial technology).

In-space Manufacturing provides Exploration mission benefits to cost, mass, crew time & reliability

Proactive influence during Exploration design phase required for meaningful implementation



ISM Technology Development & Testing

Answers HOW we will make it

- **Define NASA requirements** for ISM Technologies based on ISS & EMC Applications identified (micro-g effects, performance, & operations)
- Collaborate and establish mechanisms to leverage industry to develop the technologies needed for NASA missions.
- Utilize ISS as test-bed for developing 'FabLab' to serve as springboard for cis-lunar 'proving ground' missions.











In-Space Manufacturing (ISM) Path to Exploration

EARTH RELIANT

PROVING GROUND

EARTH INDEPENDENT

ISS Platform

- In-space Manufacturing Rack Demonstrating:
 - o 3D Print Tech Demo (plastic)
 - Additive Manufacturing Facility
 - Recycling
 - On-demand Utilization Catalogue
 - Printable Electronics
 - In-space Metals
 - Syn Bio & ISRU
 - External In-space Mfctr. & Repair

Demo Comme cial Cargo and Crew

Space Launch System

Planetary Surfaces Platform

- Additive Construction, Repair & Recycle/Reclamation Technologies (both Insitu and Ex-situ)
- Provisioning of Regolith Simulant Materials for Feedstock Utilization
- Execution and Handling of Materials for Fabrication and/or Repair Purposes
- Synthetic Biology Collaboration

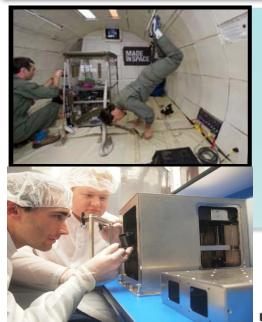
Asteroids

Earth-Based Platform

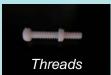
- Define Capacity and Capability Requirements (work with EMC Systems on ECLSS, Structures, Logistics & Maintenance, etc.)
- Certification & Inspection Process
- Material Characterization Database (in-situ & ex-situ)
- Additive Manufacturing Systems Automation Development
- Ground-based Technology Maturation & Demonstrations (i.e. ACME Project)
- Develop, Test, and Utilize Simulants & Binders for use as AM Feedstock



3D Printing in Zero G Technology Demonstration Mission



Potential Mission Accessories





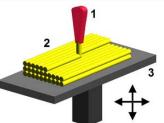








The 3D Print project delivered the first 3D printer on the ISS and investigated the effects of consistent microgravity on melt deposition additive manufacturing by printing parts in space.



Mass

Fused deposition modeling:

- 1) nozzle ejecting molten plastic,
- deposited material (modeled part),
- 3) controlled movable table

3D Print Specifications

Dimensions 33 cm x 30 cm x 36 cm Print Volume 6 cm x 12 cm x 6 cm

20 kg (w/out packing material or

spares)

Est. Accuracy 95 % **Resolution** .35 mm

Maximum Power 176W (draw from MSG)

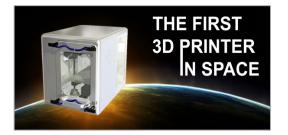
Software MIS SliceR

Traverse Linear Guide Rail

Feedstock ABS Plastic



Microgravity Science Glovebox (MSG)





Phase I Operations Timeline





- Technology Demonstration Mission via a Small Business Innovation Research contract with Made in Space, Inc.
- Ground Control Samples were made in May 2014 on the flight unit in the MSG mock-up facility at MSFC
- The 3D Print Tech Demo launched to ISS on SpaceX-4 (September 2014)
- Installed in the Microgravity Science Glovebox on ISS in November 2014
- Flight Samples were made in November December 2014
- Specimens underwent testing from May-September 2015
 - Small number of specimens make comparison between ground and flight specimens difficult
- Data from 3DP phase I out-briefed at a technical interchange meeting at NASA MSFC on Dec. 2-3, 2015
- Results will be published as a NASA technical publication in summer 2016



Phase I Prints

Completed Phase 1 Technology Demonstration Goals

- Demonstrated critical operational function of the printer
- Completed test plan for 42 ground control and flight specimens
- Identified influence factors that may explain differences between data sets

Phase II - June/July 2016

Better statistical sampling

Mechanical Property Test Articles







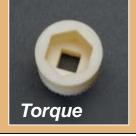
Functional Tools





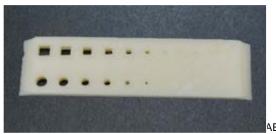






Printer Performance Capability







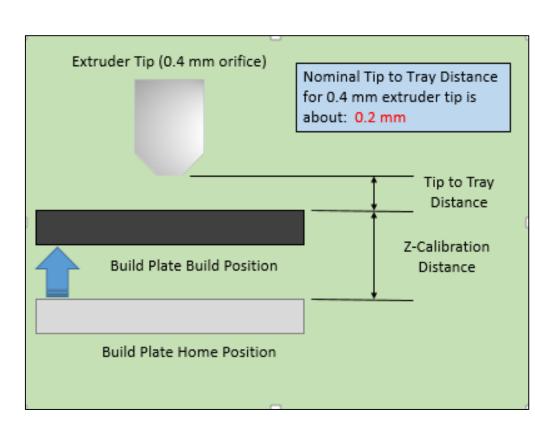






Notes on Printer Operations

- Feedstock for ground and flight are the same material and originate from the same manufacturing lot, but are from different canisters
- Flight feedstock 5-6 months older than ground feedstock at time of printing
- Changes in build tray over course of prints
 - Four separate build trays used for flight prints
- Z-calibration distance (and tip to tray distance, which is determined by the z-calibration setting) was changed slightly during the course of flight prints based on visual feedback
 - Z-Calibration was held constant for ground prints





Testing of Phase I Prints

Photographic and Visual Inspection

Inspect samples for evidence of:

- Delamination between layers
- Curling or deformation of samples
- · Voids or pores
- Sample removal damage

Mass Measurement

Measure mass of samples:

- Laboratory scale accurate to 0.01 mg
- Note any discrepancy between flight and ground samples

Structured Light Scanning

Scan external geometry of samples:

- Accurate to \pm 12.7 μ m
- Compare scan data CAD model to original CAD model
- · Measure volume from scan data
- Measure feature dimensions: length, width, height, diameter, etc.

Data Obtained

- Thorough documentation of sample quality
- Archival Photographs

Average Sample Mass

- Geometric Accuracy
- Average Sample Volume

Average Sample Density

- Internal structure
- Densification
- Mechanical Properties
- Comparison to ABS characterization data
- Microstructure data
- Layer adhesion quality
- Microgravity effects on deposition

CT Scanning / X-Ray

Inspect internal tomography of samples:

- Internal voids or pores
- Measure layer thickness / bead width
- Note any discrepancy in spacing between filament lines

Mechanical (Destructive) Testing

Mechanical Samples only:

- ASTM D638: Tensile Test
- ASTM D790: Flexural Test
- ASTM D695: Compression Test

Optical / SEM Microscopy

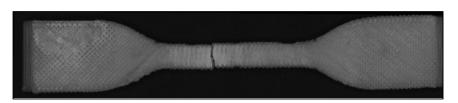
Inspect for discrepancies between flight and ground samples:

- External anomalies noted in previous tests
- microstructure
- Areas of delamination
- Fracture surface of tensile samples

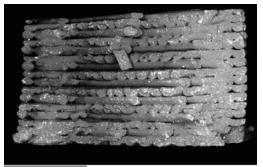


Testing of Phase I Prints





Optical microscope image of tensile specimen post-mechanical testing



Flight tensile fracture surface



Closeup of ground tensile fracture surface

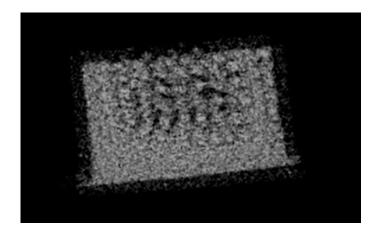


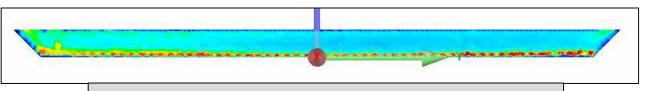
Image from CT scan of flight tensile specimen



Compression specimen



Bottom Surface Crowfoot (Flight Specimen)



Structured Light Scan of Flight Flexural Specimen



3DP Phase I Key Observations: Material Properties

Density

- Flight specimens slightly more dense than ground specimens
- Compression specimens show opposite trend
- Gravimetric density strongly correlated with other mechanical properties

Tensile and Flexure

 Flight specimens stronger and stiffer than ground counterparts

Compression

• Flight specimens are weaker than ground specimens



Optical microscope image of tensile specimen

Mechanical Properties

Material Property	Percent Difference (WRT Ground)	Coefficient of Variation (Flight)	Coefficient of Variation (Ground)
Ultimate tensile strength (KSI)	17.1%	6.0%	1.7%
Modulus of Elasticity (MSI)	15.4%	6.1%	2.7%
Fracture Elongation (%)	-30.4%	26.3%	9.9%
Compressive Strength (KSI)	-25.1%	3.1	5.0
Compressive Modulus (MSI)	-33.3%	9.4%	4.2%
Flexural Strength (PSI)	25.6%	9.3%	6.0%
Flexural Modulus (KSI)	22.0%	9.6%	3.9%

Density

Specimen Type	Percent Difference (WRT Ground)	
Tensile	3.4%	
Compression	-2.6%	
Flexure	5.6%	

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3DP Phase I Key Observations: XRay and CT

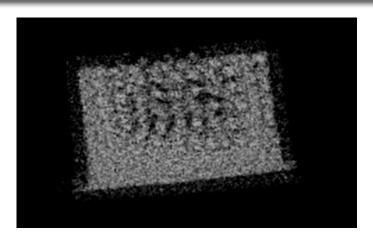
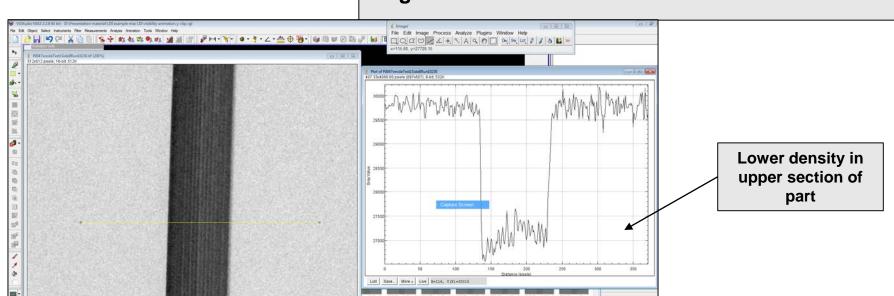


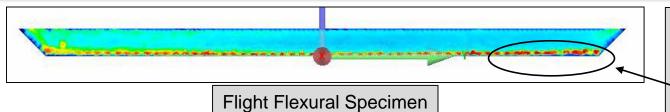
Image from CT scan of flight tensile specimen

- CT scans show an abrupt step change in density about halfway through the thickness of many specimens
 - More pronounced densification in lower half of flight specimens
 - Differences in densities (measured as mean CT) between upper and lower half of specimens is not statistically significant
- Probable voids detected throughout flight and ground articles; no significant difference in number or size of voids between the flight and ground sets

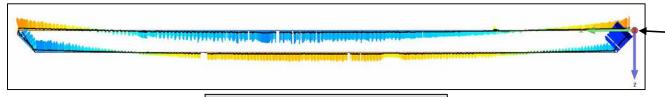




3DP Phase I Key Observations: Structured Light Scanning



Protrusions along bottom edges indicate that extruder tip may have been too close to the print tray (more pronounced for flight prints)



Ground Tensile Specimen

Warping of Samples

- may indicate inconsistent cooling of the specimen leading to internal stress build-up
- Damage sustained during specimen removal process



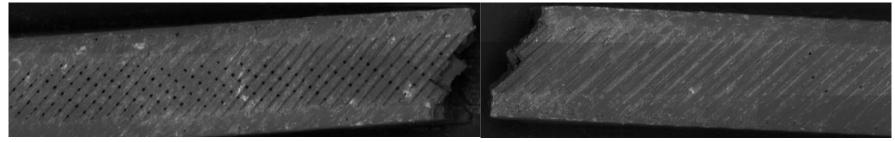
Roundness of Circular Samples

 Flight specimens slightly more out of round based on structured light scanning results

		Elliptical Cross- Sectional Area (mm²)	Percent Error of Cross-Section WRT CAD
Flight	0.14	121.7	4.11 %
Ground	0.12	123.0	2.96 %



- Structural differences are seen within both ground and flight specimen groups
- Ground sample surfaces are generally more "open" than flight specimens



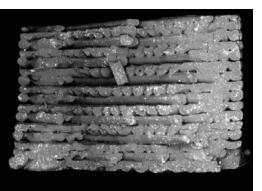
Ground tensile specimen surface

Flight tensile specimen surface

- Fracture surfaces for ground specimens have open central filaments and dense fiber agglomeration on sides
- Fracture surfaces for flight specimens have dense filament agglomeration on sides and bottom

Ground tensile fracture surface



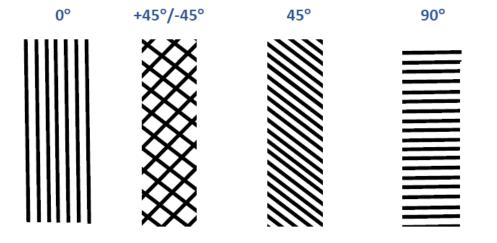


Flight tensile fracture surface

Image credit: Dr. Richard Grugel, NASA MSFC



Raster orientation	Mean yield strength (PSI)	
Longitudinal (0)	3700	
Diagonal (45)	2274	
Transverse (90)	2081	
Default (+/- 45)	2741	





Characteristic appearance of flight specimens

- Ground and flight specimens built with +/-45 orientation
- More filament bonding on bottom of flight specimens
- Likely explains increased strength of flight specimens and reduced elongation



Ground – G001 Cut Section View

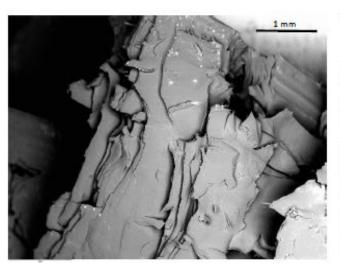
Flight – F001C Cut Section View

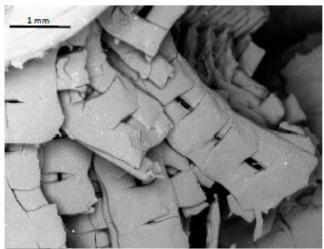
Variation in fiber diameter

Fiber "slump"

- Both calibration coupons (ground and flight) show evidence of filament slump.
- Results not suggestive of microgravity effect on materials processing, although differences in manufacturing processing conditions between flight and ground specimens preclude a definitive assessment.
- Phase II prints (completed July 16) will provide additional data.





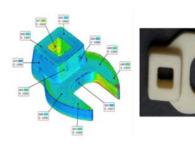


- Comparison of internal structure for ground compression specimen G013 (left) and flight compression specimen F016 (right) post-destructive testing.
- Ground compression specimens exhibit better fiber bonding.
- Likely explains difference comparative weakness of flight specimens.
- Source of structural variations may be changes in tip to tray distance for flight prints (follow-on ground based study and phase II prints will provide additional data)

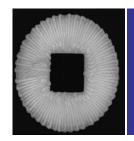


3DP Phase I Executive Summary

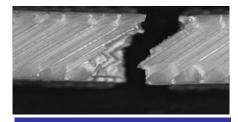
- The Phase I parts (first 21 parts printed) underwent testing and evaluation at the Materials and Processes Laboratory at NASA Marshall Space Flight Center and were compared with "ground truth" samples printed prior to printer's launch to ISS.
 - Phase I report published as NASA technical publication in summer 2016
- Differences noted in testing between the ground and flight specimens could not be definitively linked to microgravity as a processing variable
- Based on the Phase I results, the ISM team developed a go forward plan which includes: (1) Clear objectives defined for Phase II on-orbit prints and (2) Additional ground-based characterization work in order to address variables related to the 3DP data set
- Complementary microstructural and macrostructural modeling work of FDM at Ames Research Center underway
 - ISM team providing data for model validation



Structured Light Scan Data of Crowfoot Tool 3D Printed on ISS



Optical
Microscopy
of Ground
Control
Ratchet
Tool Head



Optical Microscopy of Break in Tensile Test Flight Specimen



3DP Phase I Follow-On Work

Ground Based Investigations

- Study of effect of tip-to-tray distance on part quality and performance
 - Systematic variation of this distance using 3DP backup flight unit
 - Study envelopes commanded values for ground and flight prints
 - Test regime includes surface metrology, mass measurement, structured light scanning, XRay/CT, ,mechanical testing and SEM
 - Complete by October 2016

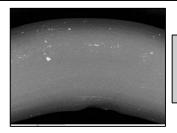


Further Analysis of Phase I Specimens

- Chemical composition analysis using Fourier Transform Infrared Spectroscopy
 - Demonstrated no significant chemical differences between ground and flight prints in terms of functional groups present and relative concentrations
- Scanning electron microscopy (SEM) of calibration coupons specimens (sparser fill) and SEM of layer quality (square column) specimens
 - No microgravity effects observed to date with SEM

On-Orbit Investigations

- Better statistical sampling with specimens from Phase II operations
- Phase II prints (34 additional specimens) completed in June and July 2016



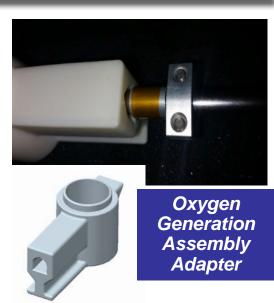
SEM Image

 Deformed ABS Filament with microcracks



Additional ISM Activities

- Interface with and design of components for ISS stakeholders
 - Oxygen Generation Assembly Adapter allows ISS crew to obtain consistent and accurate airflow velocity measurements for Environmental Control and Life Support Systems (ECLSS) hardware
 - Air Nozzle Adapter (will be used to inflate refillable stowage bags for ISS demo test)
 - Robonaut camera calibration mount (senior design project with Vanderbilt University)
 - OGA and air nozzle will be printed with Additive Manufacturing Facility (AMF)
- Defined phase II prints based on phase I results
 - Streamlined process for operations to conserve crew time
 - Phase II prints took place in June/July 2016
- Made in Space Additive Manufacturing Facility (AMF) commercial printer is now on ISS
 - Multi-user facility







Additional ISM Activities

- Tethers Unlimited (TUI) developing an in-space recycler and printer for recycling of printed parts into feedstock
- NASA Science Technology Mission Directorate (STMD)
 External In-space Manufacturing Tipping Point Project
 with Made in Space, Inc. entitled "Versatile In-Space
 Robotic Precision Manufacturing and Assembly System"
- Additive Construction by Mobile Emplacement (ACME)
 - project is in conjunction with the Army Corps of Engineers and is co-led by MSFC and KSC
 - Development of additive construction technologies for use with in-situ resources
- Procurement of Nscrypt machine
 - Multimaterial 3D printer
 - printable electronics capability
- Ongoing development work toward ISS "FabLab"
 - Trade studies of manufacturing processes for inspace applications
 - Logistics analyses
 - Material characterization activities to understand machine and material capabilities and inform requirements development



Feedstock recycler from TUI



ACME "B-Hut"





ISM Technology Portfolio

IN-SPACE POLYMERS





- ISS On-demand Mfctr. w/polymers.
- 3D Print Tech Demo
- Additive Manufacturing Facility with Made in Space, Inc.
- Material
 Characterization &
 Testing

IN-SPACE RECYCLING





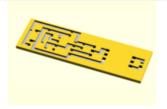
- Refabricator ISS
 Demo with Tethers
 Unlimited, Inc. (TUI)
 for on-orbit 3D
 Printing & Recycling.
- Multiple SBIRs underway on common-use materials & medical/food grade

MULTI-MATERIAL 'FAB LAB' RACK



- Develop Multimaterial Fabrication Laboratory Rack as 'springboard' for Exploration missions
- In-space Metals ISS
 Demo
- nScrypt Multi-material machine at MSFC

PRINTED ELECTRONICS



- MSFC Conductive & Dielectric Inks patented
- Designed & Tested RFID Antenna, Tags and ultra-capacitors
- 2017 ISM SBIR subtopic
- Collaboration w/Ames on plasma jet

IN-SPACE V&V PROCESS





Develop & Baseline on-orbit, in-process certification process based upon the DRAFT Engineering and Quality Standards for Additively Manufactured Space Flight Hardware

EXPLORATION DESIGN DATABASE & TESTING





- Develop design-level database for applications
- Materials dev. & characterize for feedstocks (intransit & surface) in MAPTIS DB.
- Design & test highvalue components for ISS & Exploration (ground & ISS

NASA

Acknowledgements

- Made in Space, Inc.
- Niki Werkheiser, In-Space Manufacturing Project Manager, NASA MSFC
- Dr. Raymond "Corky" Clinton, Deputy Manager, NASA MSFC Science and Technology Office
- Richard Ryan, Chief Engineer, In-Space Manufacturing
- Quincy Bean, Technology Discipline Lead Engineer for In-Space Manufacturing
- Steve Newton, In-Space Manufacturing Deputy Project Manager
- Dr. Frank Ledbetter, Senior Technical Advisor for In-Space Manufacturing
- Personnel who worked on testing and analysis of phase I prints:
 - Dr. Terry Rolin
 - Dr. Ron Beshears
 - Steven Phillips
 - Catherine Bell
 - Dr. Richard Grugel
 - Erick Ordonez
 - Lewis "Chip" Moor€



Questions







Backup Slides



ISM Education & Public Outreach 'Scrapbook' (Oct, 2015 – April, 2016)



3D Print

included as Top

15 ISS events

for the ISS 15th

Anniversary

Infographic

Released

11/2/15

Future Engineers listed as 'Breakthrough Award' in Nov. Issue of Popular Mechanics





"Design Consultation" with FE Winner, R.J. Hillan, NASA ISM team members, and MIS Design Lead, Mike Snyder 12/4/15



Media Event with ISM and Former ISS Commander Butch Wilmore 11/16/15

National FE Challenge Teen Winner, Ryan B., at California Science Center with Astronaut Leland Melvin 10/27/15

FE Junior Division Winner, Emily T., with her winning design, the Flower Tea Cage

NASA **Systems** Eng. Excellence Award for 3D Print Demo