A SPACE-TIME DISCONTINUOUS-GALERKIN APPROACH FOR SEPARATED FLOWS

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Background

- Developing a spectral-element DG capability for separated flows over the past few years
 - Led by investment from TTT/RCA
 - Diosady & Murman AIAA 2013-2870, 2014-2784, 2015-0294
- Effort has grown recently w/ collaboration from other projects
 - Desired synergy between R&D and engineering
- Opportunity to share broader vision of effort & fill in technical gaps for AIAA audience





Context

- No single optimal algorithm/method/solver
 - Pareto front of optimal choices
- Different groups prioritize differently
- Our priorities are derived from the needs of numerous projects within NASA - ARMD, HEOMD, STMD - and industrial partners
- Current NASA technology based primarily on RANS/DES
 - Works well for many engineering tasks
- Supplement existing capability w/ scale-resolving methods



Target Applications

- Complex geometry unstructured mesh
- Complex physics scale-resolving methods
- High-Re, combustion, chemistry fully implicit methods
- Computational intensive high-order, adaptive methods
- Multi-disciplinary, multi-physics robust, extensible methods







- DG spectral-element formulation
 - Unstructured arbitrary order
 - Variational Multiscale Method (VMM) for scale-resolving
- Fully implicit space-time
 - Entropy-stable, consistent all-speed scheme
 - *h*-*p* adaptation in space and time
- Galerkin formulation
 - Demonstrated success for relevant applications





- Three main thrusts
 - New algorithms and methods
 - Optimized for next-gen exascale hardware
 - Novel physical models
- Informally known as the eddy solver
- Currently lower TRL than production methods







Turbomachinery Benchmarks



- Developed PML approach for non-reflective BC
 - Garai et al. AIAA 2016-1338
- Developed physics-based approach for freestream turbulence
- Garai et al., ASME GT2015-42773, GT2016-56700





- We want to approach spectral limit in space and time
- Leads to efficiency gains and improved physical models
- Better match for current/future hardware
 - Less data movement, more flops for the same level of accuracy

Hardware-optimized Kernels

- Current algorithms achieve < 5% of machine peak
- Spectral elements a good match for current & future hardware









ernels ne peak & future hardware

Hardware-optimized Kernels

- Tensor-product sum-factorization linear algebra kernels
- Benchmark represents ~20% of code







Time-parallel

- Exploit multiple levels of parallelism
 - Parallel in space across nodes (MPI)
 - Parallel in time within node (OpenMP)
 - Parallel within loops on chip (SIMD vectorization)



• 500 Gflops per Haswell node for 8th-order



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Scale-resolving Models

- Improved numerics changes how we *do* CFD
 - Efficiency, automation, error estimates
- Consistent predictive models would change how we *use* CFD Certification through simulation
- Need to prioritize new modeling approaches
 - Tighter coupling of numerics and modeling
- Current work is not a DG solver development it is a framework for examining scale-resolving models and methods









- Explicit separation of scales (Hughes et al., 1998, Collis, 2001)
- Filtering is variational projection operator
- Assume unresolved scales only interact w/ finest resolved scales
- Extended VMM to dynamic procedure, varying coefficient in space & time



Dynamic Modeling

- Dynamic (parameter-free) models are a necessity for complex flows
 - Automatically adjust to physics, numerics
 - $C_S = 0.18$ for HIT, $C_S = 0.065$ for shear flow 10x change in eddy viscosity
- Successful approaches have been built upon strong physical understanding
 - Scale similarity, homogeneity, local isotropy, near-wall asymptotics
- New approaches need to leverage these lessons learned





Dynamic VMM Model

$$\tau\left(u, \bar{w}\right) \simeq -2\left(\left(C_1 \Delta\right)^2 \|\check{S}_{i,j}\|\right)$$

- Variational Leonard stresses
 - Requires high-order (N \geq 4)

$$\left(\bar{u}_{i}^{h}\bar{u}_{j}^{h} - \bar{u}_{i}^{H}\bar{u}_{j}^{H}, \bar{w}_{i,j}^{H}\right) = -2\left(\left(C_{1}\Delta\right)^{2} \|\check{S}_{i,j}^{h}\| \check{S}_{i,j}^{h}, \check{w}_{i,j}^{H}\right) +$$

- Using state as test function gives variational analogue to Germano procedure
- Can also provide analogue to Lilly's least-square
- Entropy-stable compressible formulation in full paper





 $|\check{S}_{i,j},\check{w}_{i,j}
ight)$

$2\left(\left(C_{1}\Delta\right)^{2} \|\check{S}_{i,j}^{H}\|\check{S}_{i,j}^{H},\check{w}_{i,j}^{H}\right)$

Channel Flow



- Practical simulations never have sufficient resolution
- Examine behavior on realistic coarse mesh
 - $\text{Re}_{\tau} = 544$
 - 4th-order in time, 8th-order in space

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$$\Delta t^+ = 1$$
, $\Delta x^+ = 100$, $\Delta y^+ = 1$, $\Delta z^+ = 50$



APS 2014

Channel Flow

• VMM w/ fixed coefficient degrades performance





Idealized Behavior

- ILES always resolves lower Re
- Dynamic approach resolves inertial range uses model for dissipation scales
- Requires non-dissipative scheme (e.g. skew symmetry)
- Entropy-stable schemes inherently dissipative
- Completely remove numerical dissipation as first test







Channel Flow

- Dynamic procedure least sensitive to current mesh resolution
 - Examine trends w/ changing resolution through higher Re





t mesh resolution gh higher Re eynolds Stress

Dynamic VMM Model

- Expected value in log layer
- Approach zero towards wall
- Decays towards centerline







Summary

- Working prototype to experiment w/ scale-resolving methods for complex flows
- Existence proof that spectral-elements can take advantage of modern hardware
- Initial experiments w/ VMM encouraging
- Current work is extending to relevant flight geometry and conditions
 - Wall-modeled LES/VMM
 - Complex geometry (AIAA 2015-0294)
 - Relative motion/FSI capability



Backup



Space-time Adjoint



- Variable-order produces lower error at same cost
 - Ceze et al. AIAA 2016-0833
- Currently extending to space-time *h*-*p* adaptation and error estimates





Current Status

- Laslo Diosady moving body, shock capturing
- Anirban Garai turbomachinery, LES AIAA 2016-XXXX
- Marco Ceze adjoint, mesh adaptation AIAA 2016-XXXX
- Corentin Carton de Wiart wall modeling, hybrid-RANS



AA 2016-XXXX AA 2016-XXXX hybrid-RANS

Hardware-optimized Kernels

• Optimization counterbalances increase in cost for high order



