



Jet-Surface Interaction – High Aspect Ratio Nozzle Test Test Summary

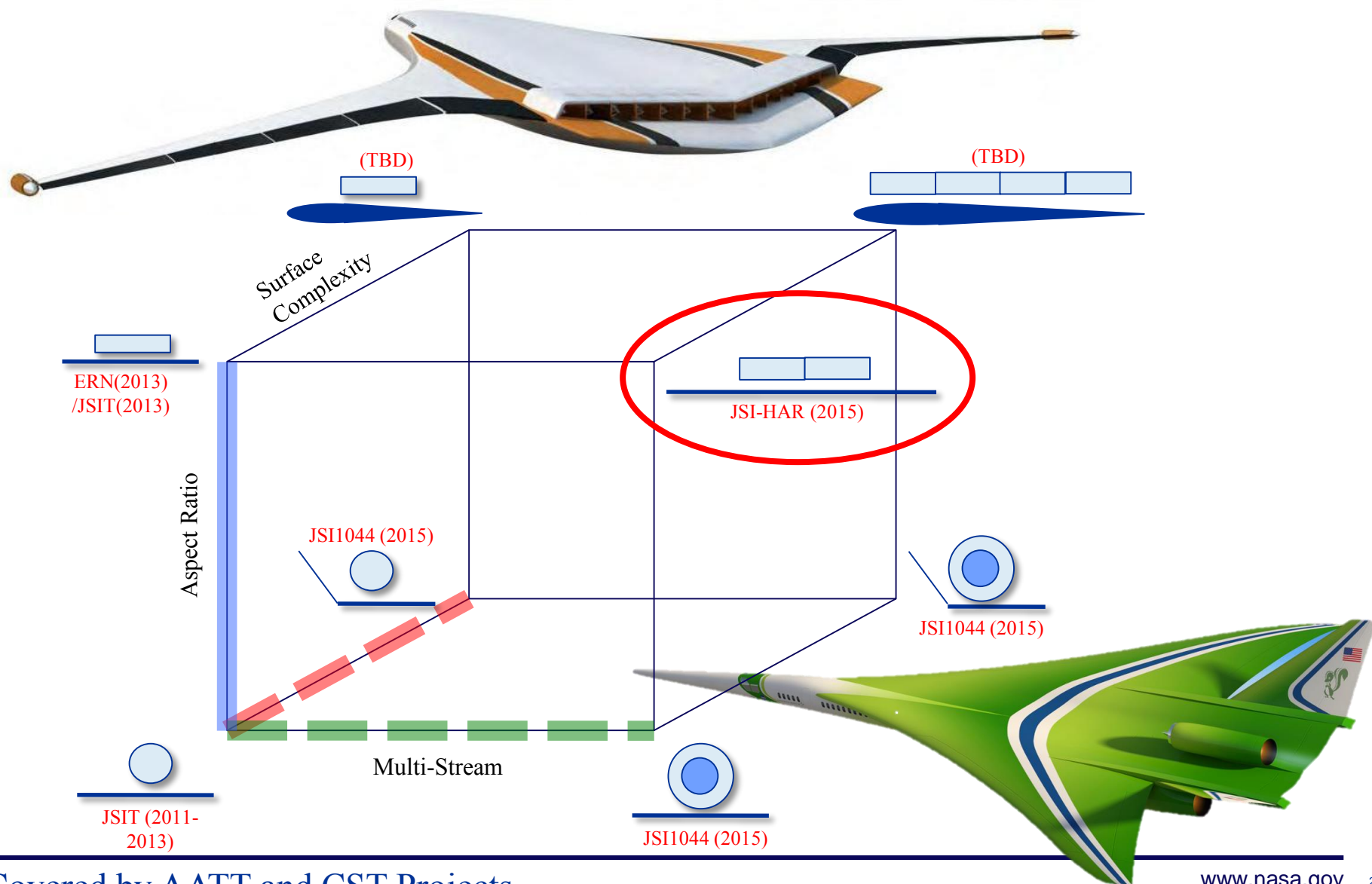
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Jet-Surface Interaction Noise Test Programs



Motivation:

Turbo-electric Distributed Propulsion Concept (TeDP)



- 32:1 aspect ratio slot
- Divided into 2:1 at exit
- Electric fan has low pressure ratio, low temperature ratio
- Aft deck extends (estimated) 1-4 slot heights downstream

Goals for JSI-HAR

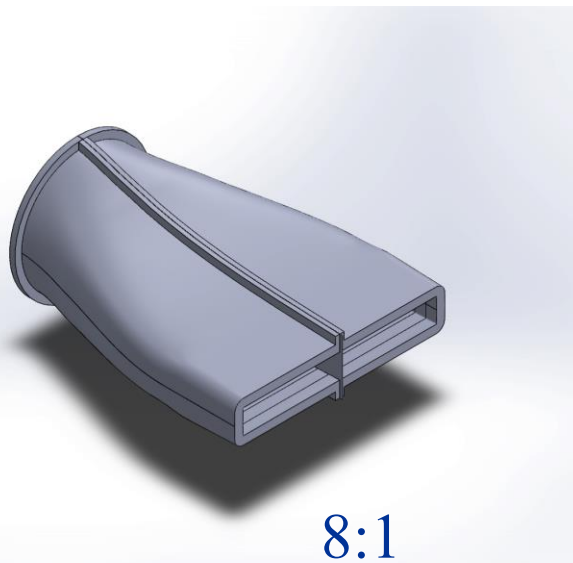
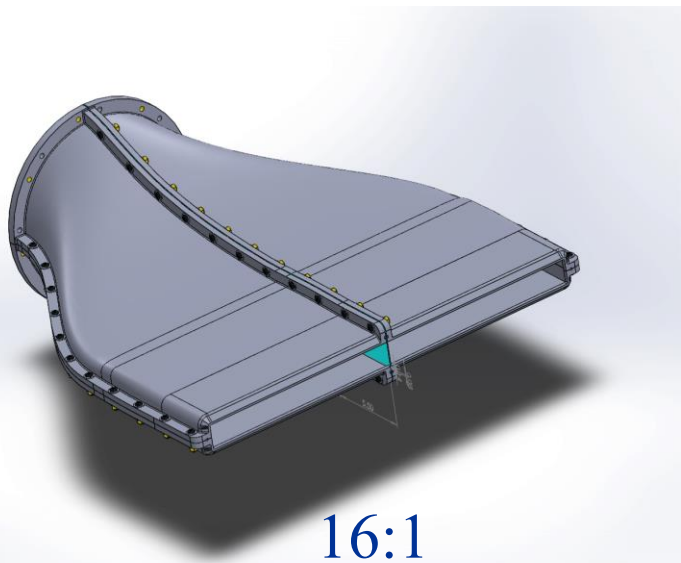
1. Extend current database to larger aspect ratio nozzles
2. Verify / connect current small-scale database to larger-scale rectangular nozzles near surfaces
3. Acquire data suitable for creating / validating empirical jet-surface interaction noise models
4. Investigate the effect of nozzle septa on the jet-mixing and jet-surface interaction noise sources



Test Plan

1. Design and test 3 nozzles (listed by priority):

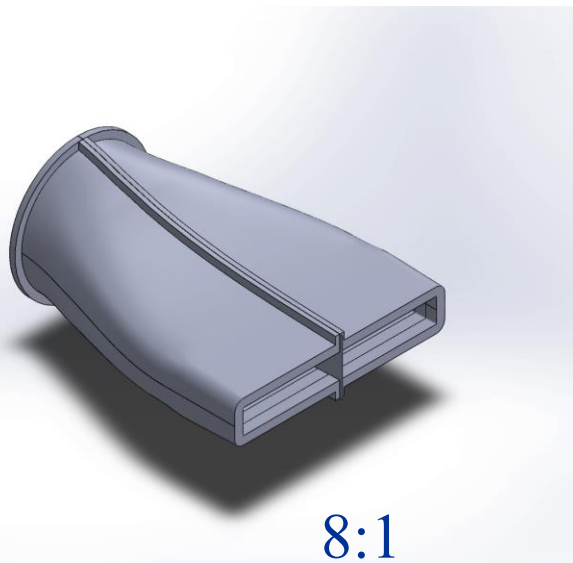
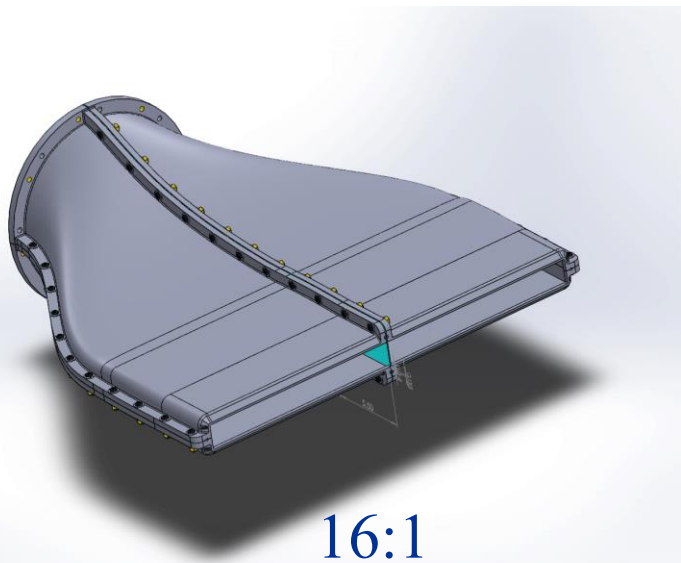
1. 16:1 aspect ratio – extend current database to higher aspect ratios
2. 8:1 aspect ratio – verify/connect small-scale database to larger-scale
3. 12:1 aspect ratio – midpoint to allow a second-order modeling



Test Plan

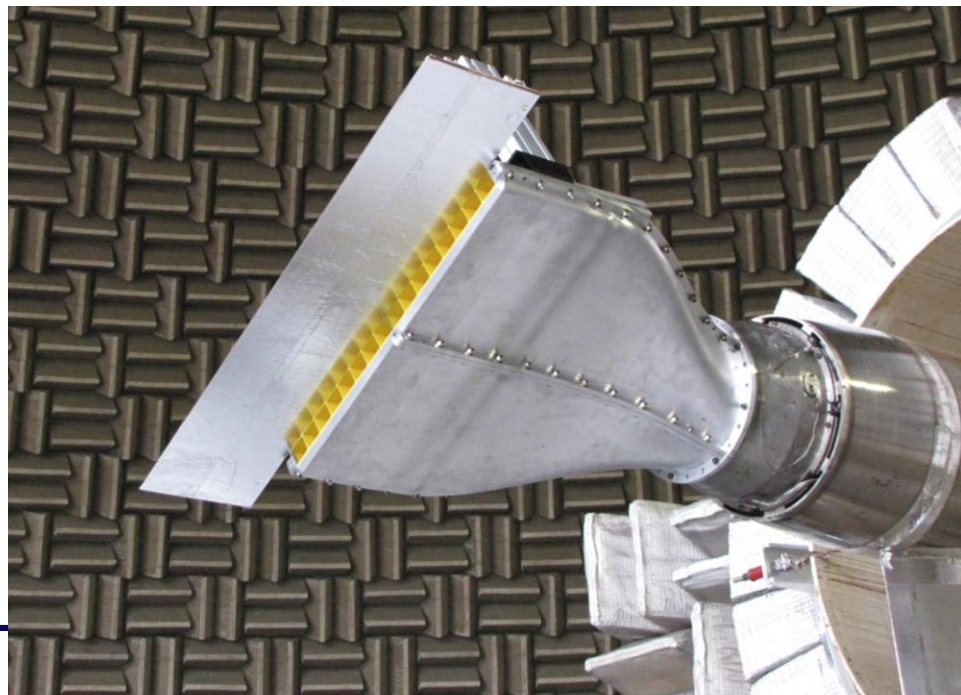
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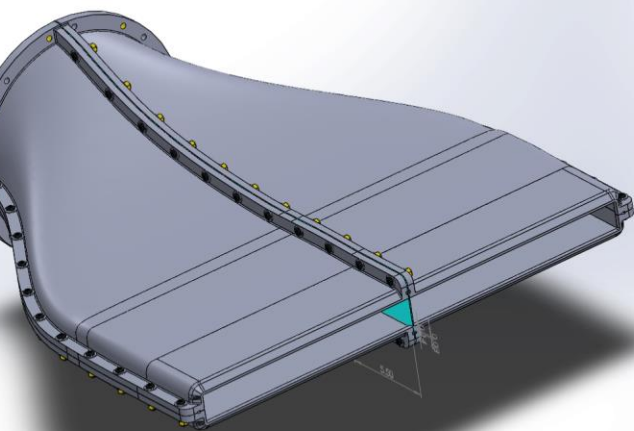
Test Plan

1. Design and test 16:1, ~~8:1~~, ~~12:1~~ aspect ratio nozzles
2. Add aft decks / surfaces onto nozzles
 1. Acquire data for modeling JSI source and shielding effect

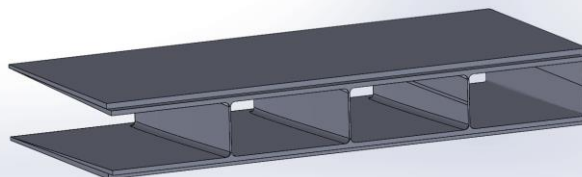


Test Plan

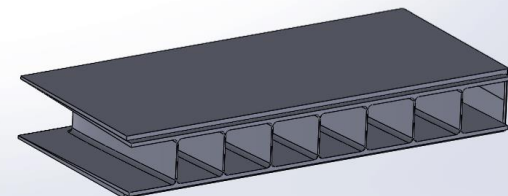
1. Design and test 16:1, ~~8:1~~, ~~12:1~~ aspect ratio nozzles
2. Add aft decks / surfaces onto nozzles
3. Design and test nozzle septa inserts
 1. “Open” no septa insert – effect of aspect ratio on jet mixing noise
 2. 2:1 / 7 septa inserts – similar to the TeDP concept
 3. 1:1 / 15 septa insert – effect of varying number of septa
 4. ~~Other variations~~



“Open”

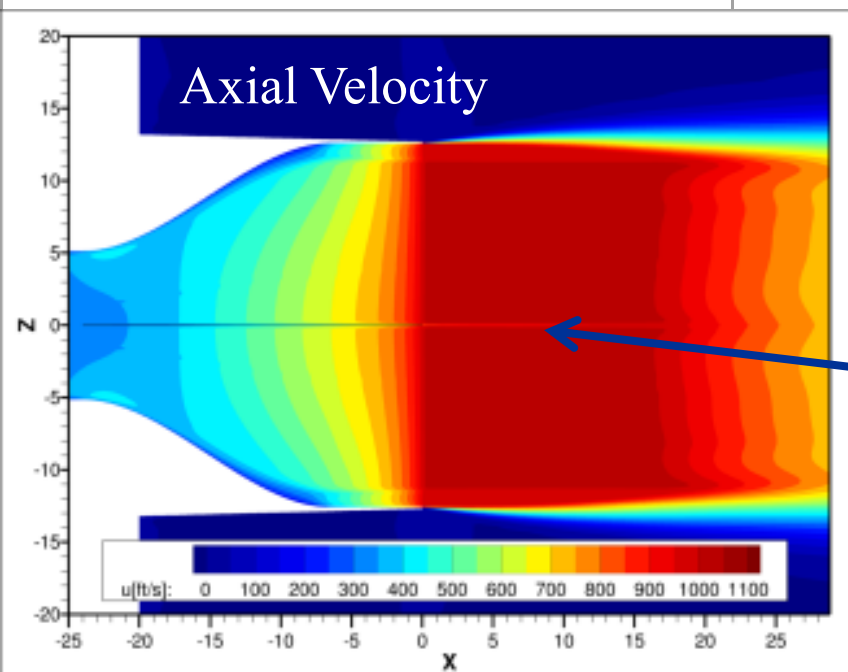
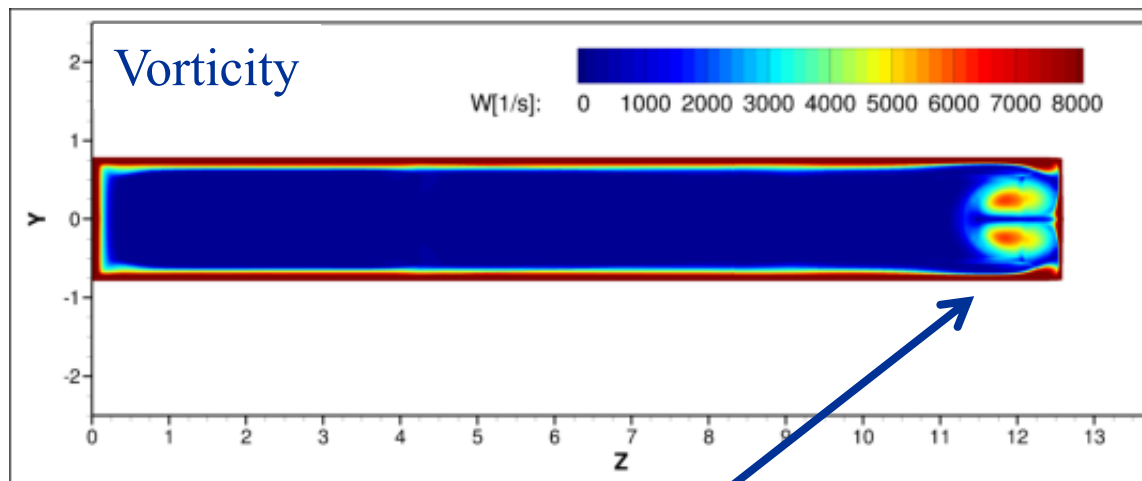
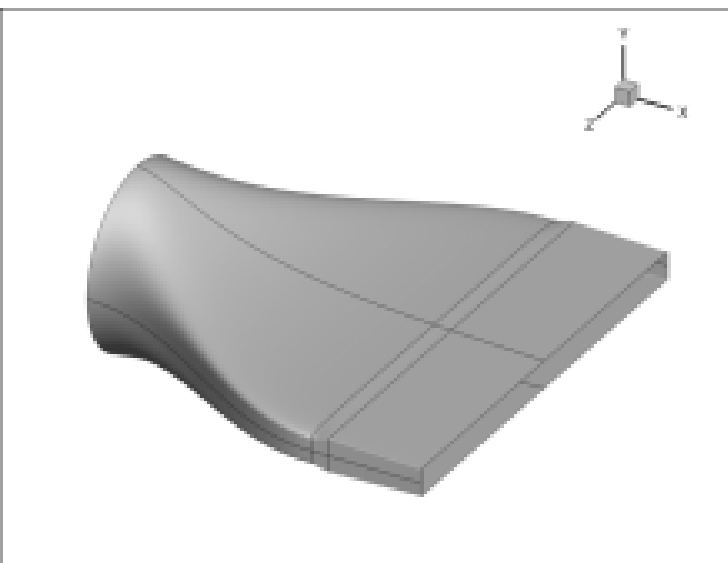


2:1 / 7 septa



1:1 / 15 septa

16:1 Nozzle Design



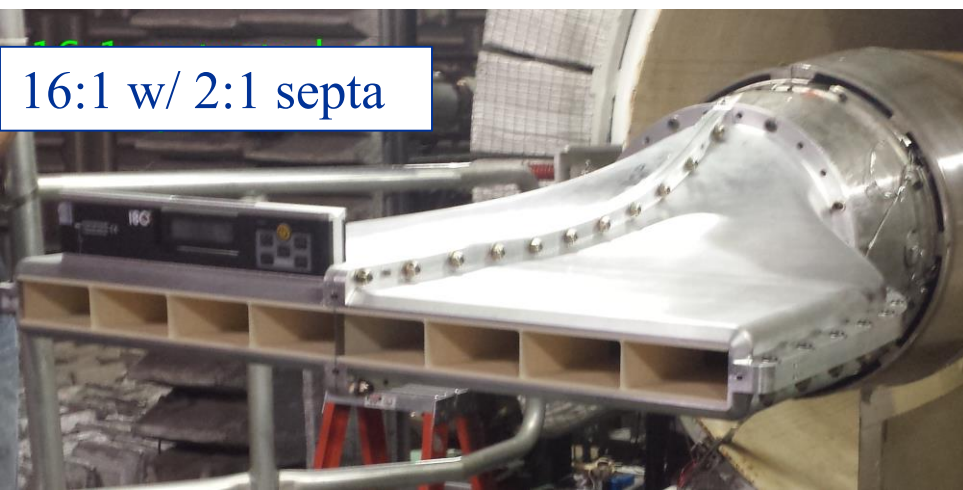
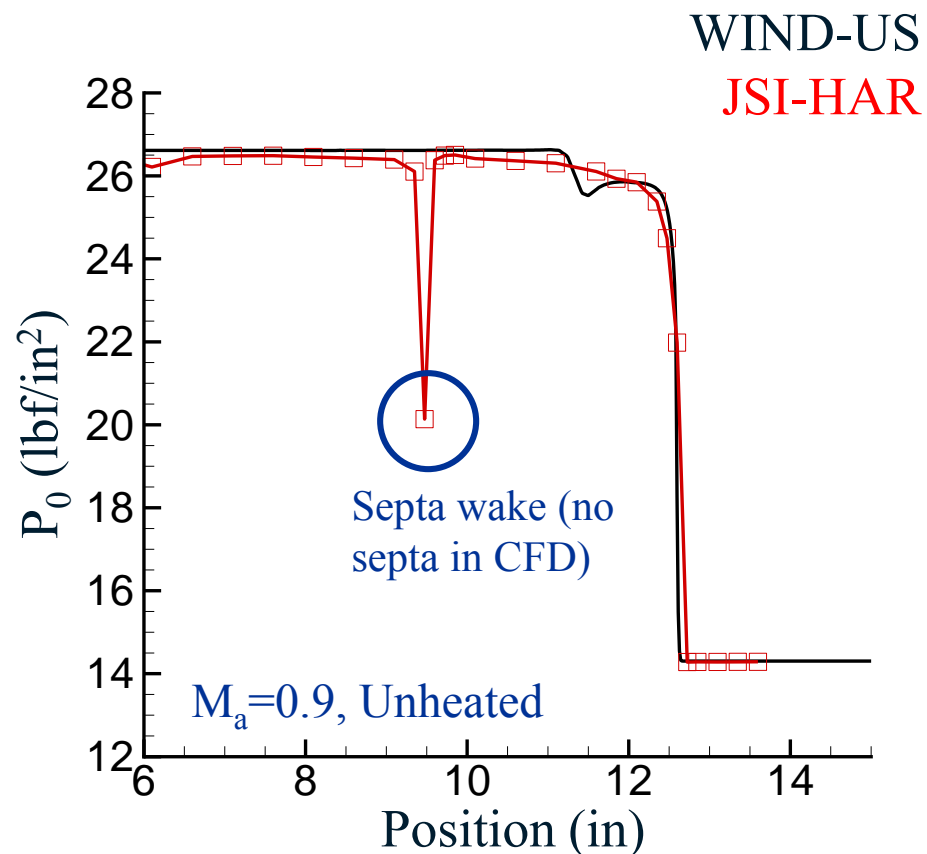
- Significant vorticity near corners
- Attached flow along outboard edge of major axis (BL thickness still significant)
- No normal shocks at nozzle exit
- Continuous area contraction helps
- Significant wake from center vane (added for structural support)

* Brown & Dippold, TWG Fall 2015

* Dippold, V., "Design and Analyses of High Aspect Ratio Nozzle for Distributed Propulsion Acoustics Measurements", AIAA Aviation 2016 Conference

Flow Profile at Nozzle Exit

- 2:1 / 7 septa insert installed for **JSI-HAR** but not in WIND-US
- Total pressure measured 0.25" downstream of nozzle exit
- No indication of vortex in **JSI-HAR**
 - 1 Hz averaged pressure data would not likely pick this up even if present
- Flat profile between septa
- Losses slightly higher in **JSI-HAR** data



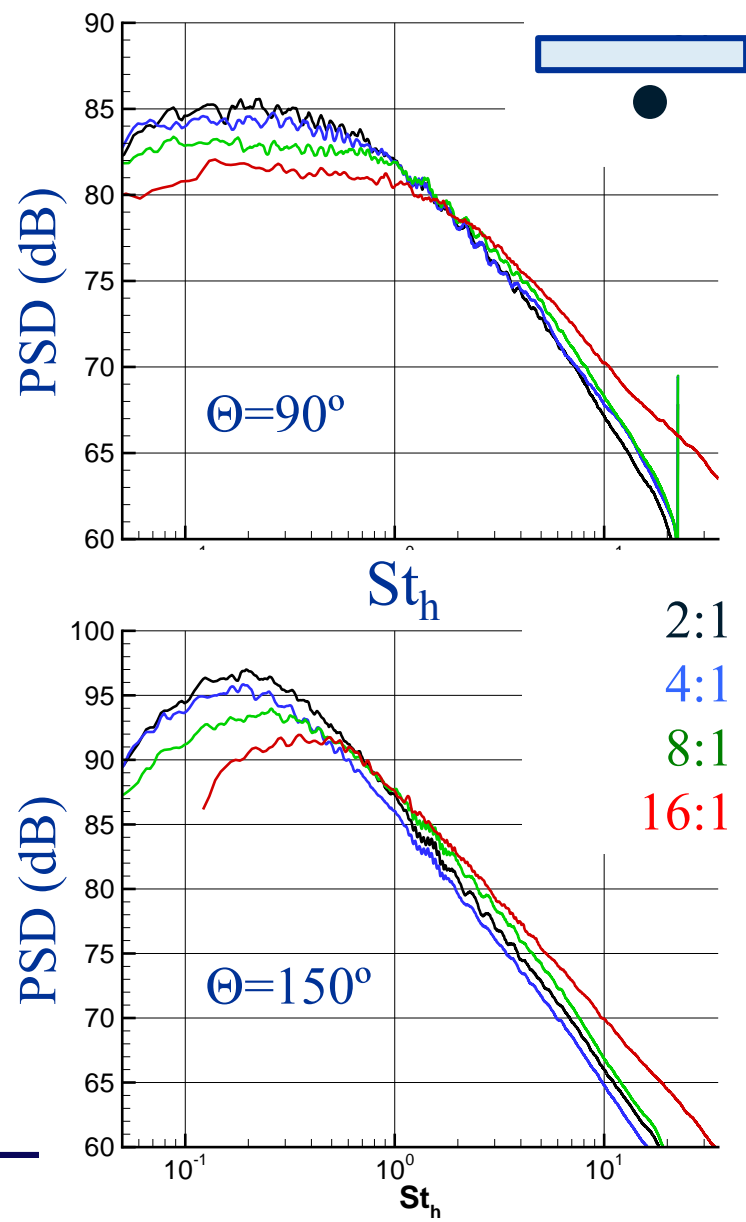
Goals for JSI-HAR

1. Extend current database to larger aspect ratio nozzles
2. Verify / connect current small-scale database to larger-scale rectangular nozzles near surfaces
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Extend to Higher Aspect Ratios

- Compare spectra to:
 - Show effect of nozzle aspect ratio
 - Connect to existing database via trends
- Similar to TeDP jet exit condition
 - Mach 0.7, unheated
- Nozzles with different sizes
 - 2:1, 4:1, 8:1 -> Area = 3.57 in²
 - 16:1 -> Area = 33.7 in²
- Scale:
 - Frequency as Strouhal number based on nozzle height
 - Distance to 100 equivalent jet diameter
- Trends follow from small to large scale across test programs



Goals for JSI-HAR

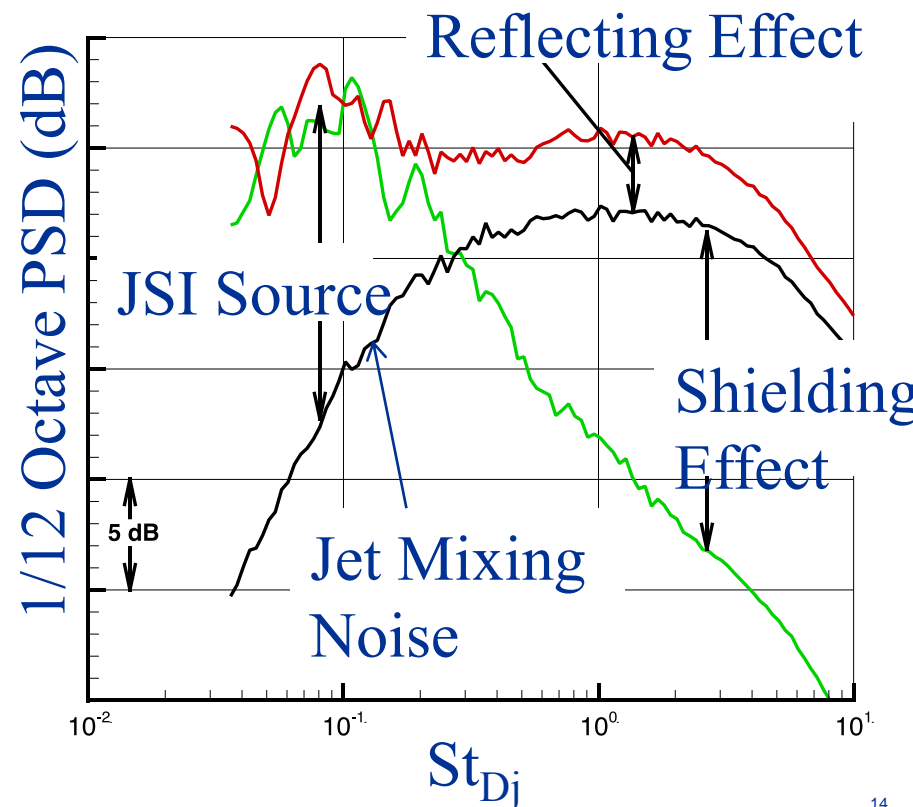
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Jet-Surface Interaction (JSI) Noise Sources and Effects

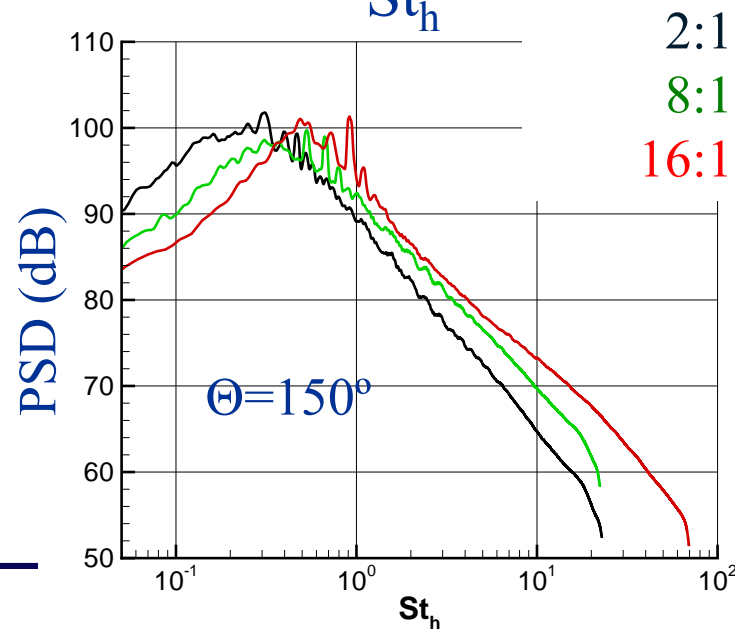
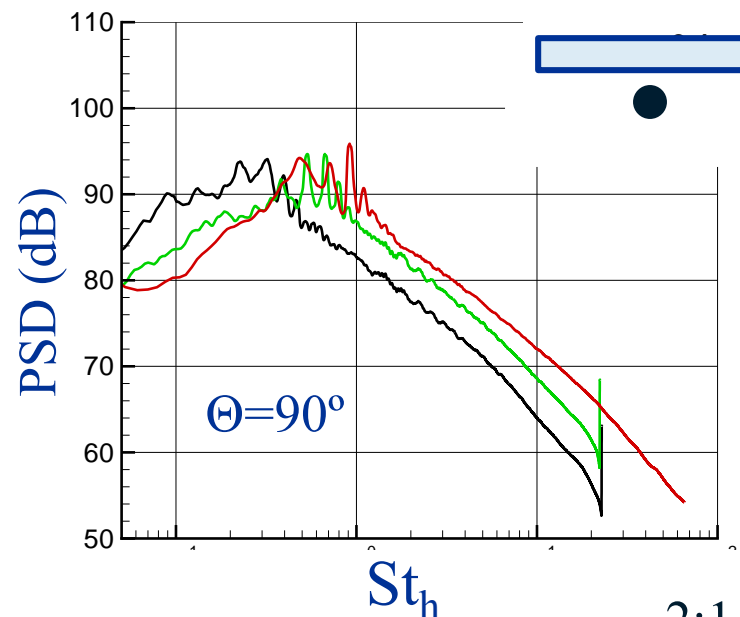


- Measured far-field noise includes:
 - Jet-surface interaction noise sources
 - Jet mixing noise (isolated)
 - Shielding/Reflecting effect
- Types of JSI noise sources
 - Surface loading (“scrubbing”) noise
 - Trailing edge (“scattering”) noise
 - Surface vibration noise
- Data acquired for surface lengths $x_E/h = 0.83, 2, 4, 6, 8$, zero standoff



Extend to Larger Scale

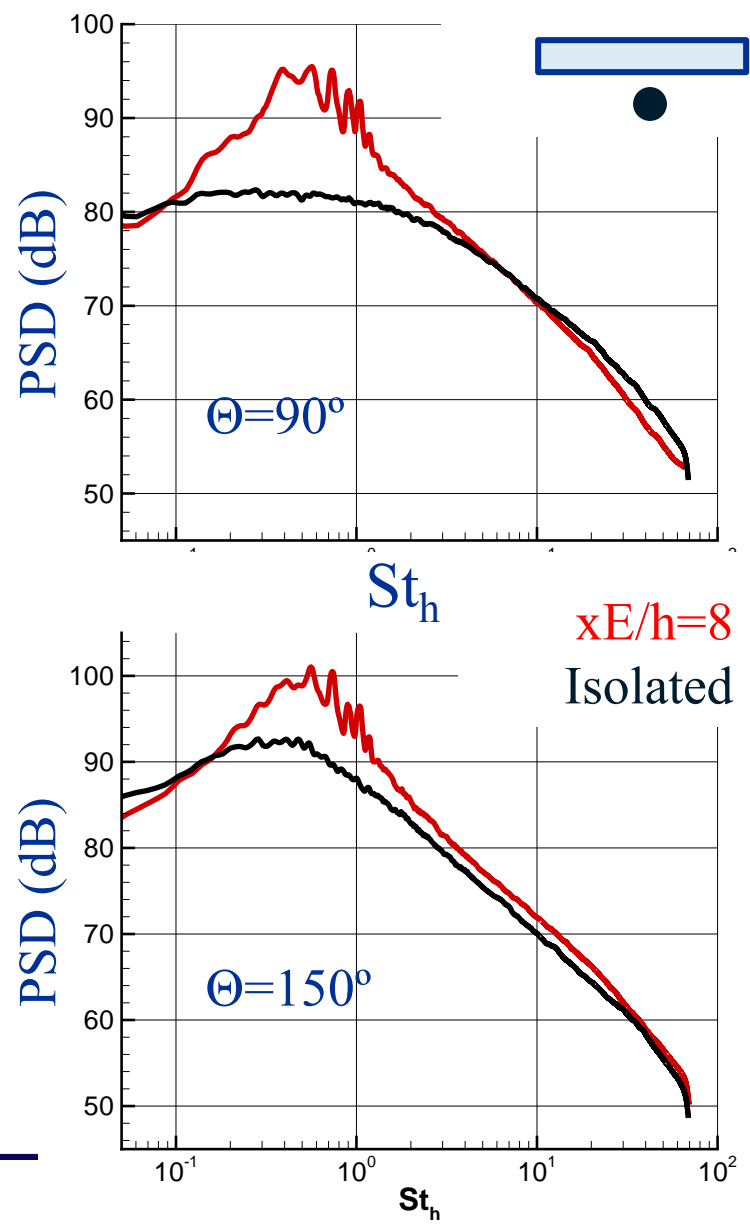
- Compare spectra to:
 - Effect surface at increase aspect ratios
 - Connect to existing database via trends
- Similar to TeDP jet exit condition
 - Mach 0.7, unheated
- Surface length, $x_E/h = 6$
- Scale:
 - Frequency as Strouhal number based on nozzle height
 - Distance to 100 equivalent jet diameter
- Trends follow from small to large scale across test programs





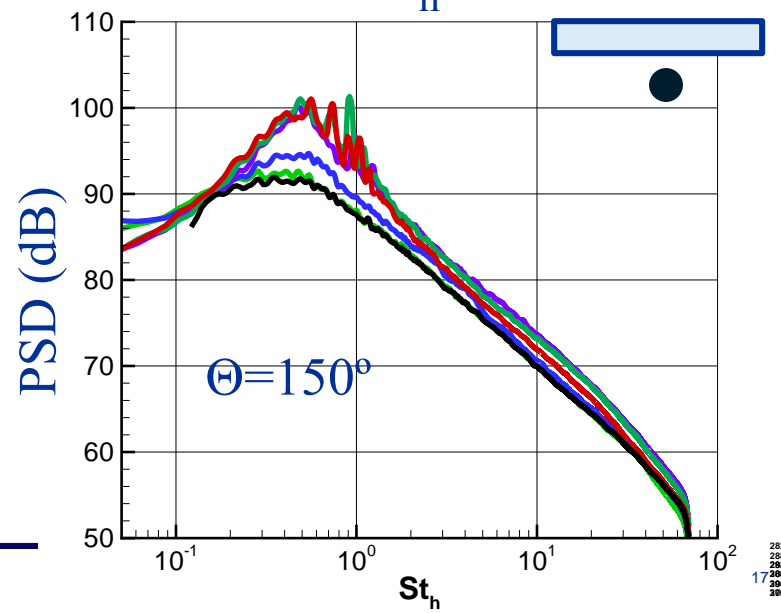
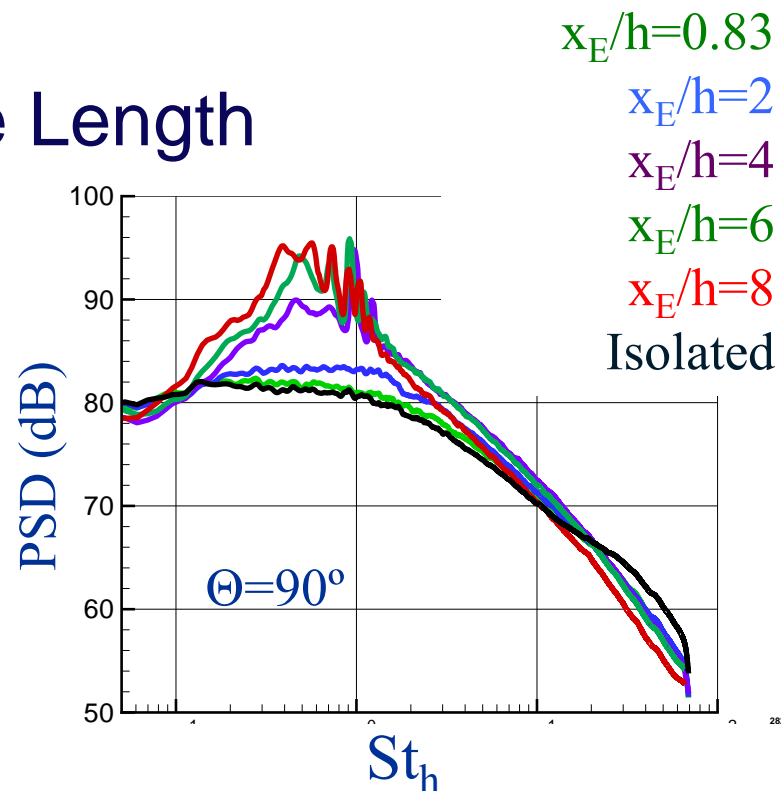
Noise Impact of Surface

- Compare spectra to:
 - Show effect of adding surface
- Similar to TeDP jet exit condition
 - Mach 0.7, unheated
- Aspect ratio 16:1
- Surface length, $x_E = 8h$
- JSI source maybe large relative to shielding
- Model to full-scale factor matters



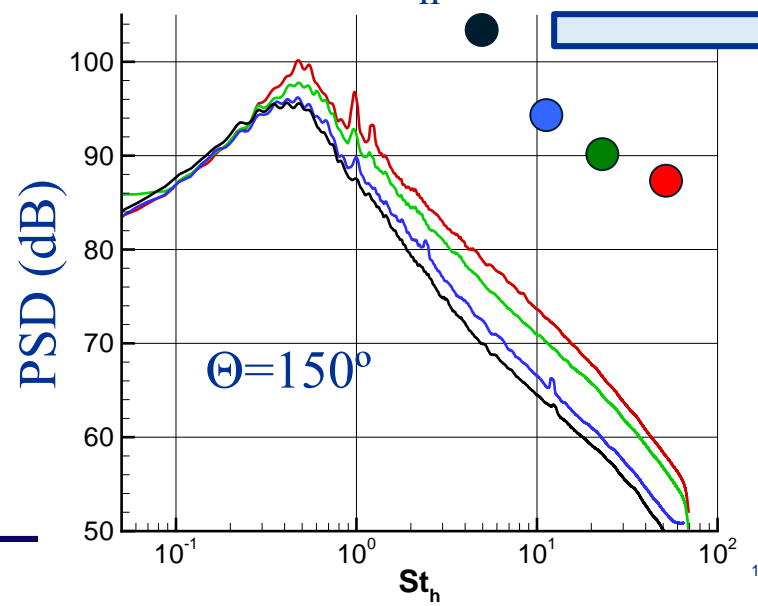
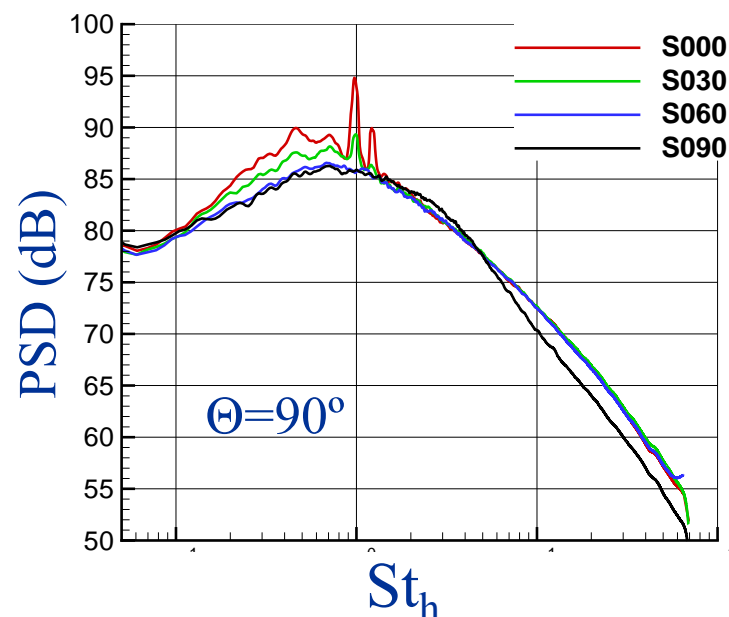
Noise Impact of Surface Length

- Compare spectra to:
 - Show effect of surface length
- Similar to TeDP jet exit condition
 - Mach 0.7, unheated
- Aspect ratio 16:1
- Shorter surface may give high frequency shielding with smaller low frequency penalty at 90°
- All surfaces produce more high frequency noise than isolated at 150°



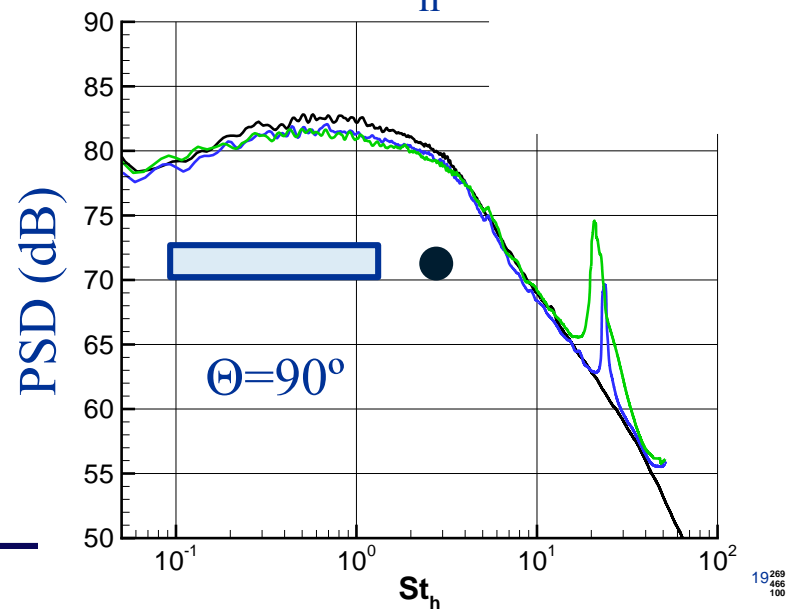
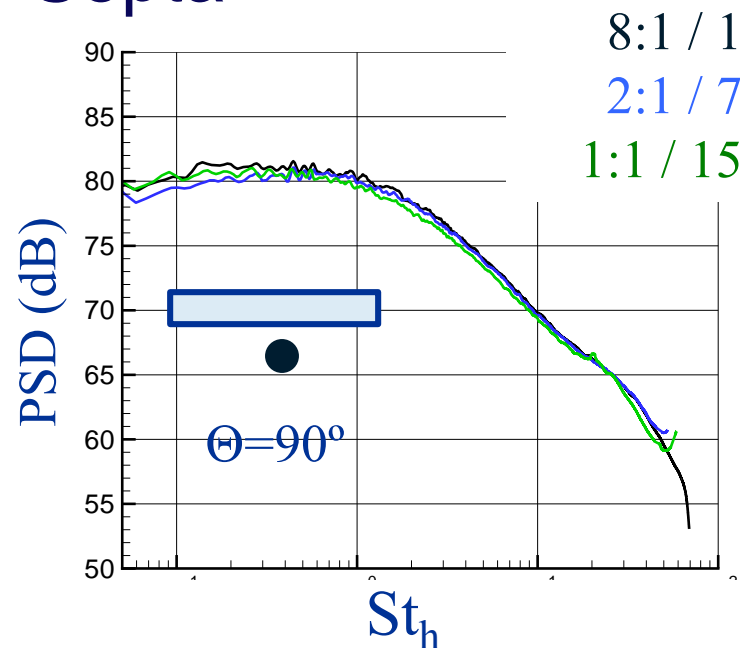
Noise Impact of Observer Azimuthal Angle

- Compare spectra to:
 - Show effect sideline
- Similar to TeDP jet exit condition
 - Mach 0.7, unheated
- 16:1, $x_E/h = 4$
- Significant changes at downstream observer angles as azimuthal angle changes



Noise Impact of Nozzle Septa

- Compare spectra to:
 - Show effect nozzle septa
- Similar to TeDP jet exit condition
 - Mach 0.7, unheated
- 16:1, no surface
- Septa create tone to major axis observer that grows with number of septa





Summary of JSI-HAR

1. Extend current database to larger aspect ratio nozzles
 - Acquired data with 16:1 nozzle
2. Verify / connect current small-scale database to larger-scale rectangular nozzles near surfaces
 - Trends with and without surfaces appear to follow from previous work
3. Acquire data suitable for creating / validating empirical jet-surface interaction noise models
 - Acquired data over a range of surface lengths
4. Investigate the effect of nozzle septa on the jet-mixing and jet-surface interaction noise sources
 - Data acquired with 3 septa configurations
- What's next?

Goals for JSI-HAR

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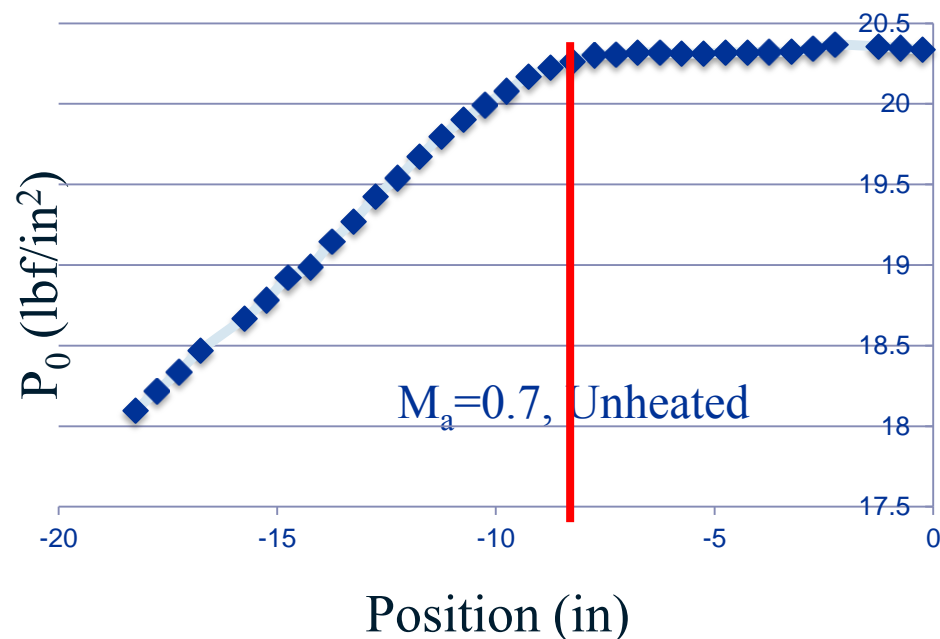


JSI Source and Effect Modeling

- Empirical models have been developed for round nozzles near surfaces
- First-order modeling for rectangular nozzles based on these round nozzle models suggest:
 - Scaling distances and frequency on nozzle height
 - Adjusting potential core length
- Jet potential core length is nondimensionalizing parameter
 - Data were acquired with 16:1 nozzle to estimate potential core length

Jet Potential Core Length

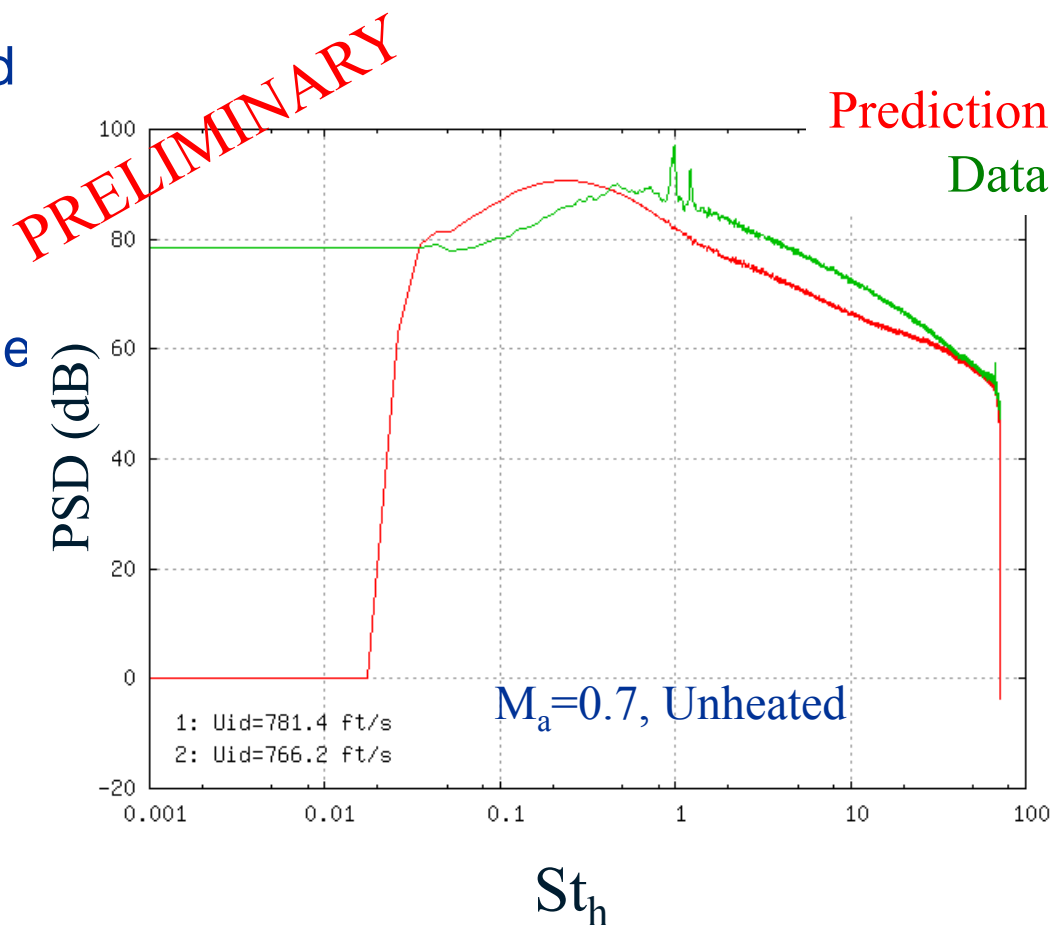
- JSI source and shielding effect models both depend on jet potential core length (x_C)
- Surface length in model is x_E/x_C
- Jet potential core length is approximately 7.75" for $Ma=0.7$, unheated jet
- Model for round jet would give $x_C/D_e \approx 5.13$
- If rectangular nozzle scales by h instead of D_e ,
 $x_C/h \approx 5.13 \rightarrow x_C \approx 7.7$ "





Jet Potential Core Length

- Modeled prediction with adjusted scaling parameters for rectangular nozzles
- Peak frequency shift
- Approximate right peak amplitude (JSI source driven)
- Spectral shape off at high frequencies
- **More development needed!**





Questions?

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Summary

- A round-to-rectangular convergent nozzle with aspect ratio 16:1 was designed for acoustic measurements
 - Minimized potential noise sources from: (1) internal flow separation and (2) shock cells
- 16:1 aspect ratio nozzle fabricated for testing
 - Inserts to simulate TeDP concept details (septa) rapid prototyped
- Pressure traverse at nozzle exit shows expected flow profile
- Preliminary analysis of noise data consistent with previous experiments
 - JSI noise source prominent at low frequencies
 - Shielding at only the highest frequencies
- Test on-going through October
 - Baseline (no septa), 2:1 / 7 Septa inserts planned

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