

#### **Overview of Proposed ISRU Technology Development**

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#### Introduction



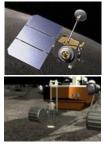
- NASA's Advanced Exploration Systems (AES) in the Human Exploration and Operations Mission Directorate (HEOMD) is assessing options for ISRU technology and system maturation focused on
  - Volatiles resource acquisition
  - Volatiles and atmospheric processing into propellants and other consumable products
- This presentation outlines preliminary definition of the objectives and approach of such an ISRU Technology Development plan

## AES ISRU Technology Development Focus: Acquisition, Processing, Consumables Production



#### ISRU involves any hardware or operation that harnesses and utilizes 'in-situ' resources (natural and discarded) to create products and services for robotic and human exploration

#### Resource Assessment (Prospecting)



Assessment of physical, mineral/ chemical, and volatile/ water resources, terrain, geology, and environment (orbital and local)

#### **Resource Acquisition**

Extraction, excavation, transfer, and preparation before processing

#### Resource Processing/ Consumable Production





In Situ Construction

Processing resources into products with immediate use or as feedstock for construction and/or manufacturing ➤ Propellants, life support gases, fuel

cell reactants, etc.

#### In Situ Manufacturing



Production of replacement parts, complex products, machines, and integrated systems from feedstock derived from one or more processed resources

#### Civil engineering, infrastructure emplacement, and structure construction using materials produced from in situ resources

 Radiation shields, landing pads, roads, berms, habitats, etc.

#### In Situ Energy



Generation and storage of electrical, thermal, and chemical energy with in situ derived materials

Solar arrays, thermal wadis, chemical batteries, etc.

- 'ISRU' is a capability involving multiple elements to achieve final products (mobility, product storage and delivery, power, thermal management, etc.)
- 'ISRU' does not exist on its own. By definition it must connect and tie to users/customers of ISRU products and services



- Focus on Mars O<sub>2</sub>/CH<sub>4</sub> end-to-end system demonstration as primary long-term objective with the following rationale:
  - Regolith acquisition and processing should / could be similar for Mars and for lunar icy soils, so we will be advancing that subsystem for both locations simultaneously
    - Responds to current EMC requirements/plans for Mars
    - Keeps options open for ISRU proving ground demos on the Moon/asteroids and for using lunar volatiles as a resource
  - Atmosphere processing technology development can be structured around common components/subsystems needed for both CH<sub>4</sub> production from atmosphere/soil-water and O<sub>2</sub>-only production from atmosphere
    - Leverage on-going work and maintain flexibility to evolving architecture
    - CO<sub>2</sub> acquisition and compression, water separation/condensation, water electrolysis, and gas / gas separation and recirculation subsystems all needed in multiple production system options
    - Advancing and scaling up Mars atmosphere-only subsystem and building off of/using lessons learned from MOXIE
  - Both regolith processing and CH<sub>4</sub> production can attract outside industries



## ISRU is a disruptive capability

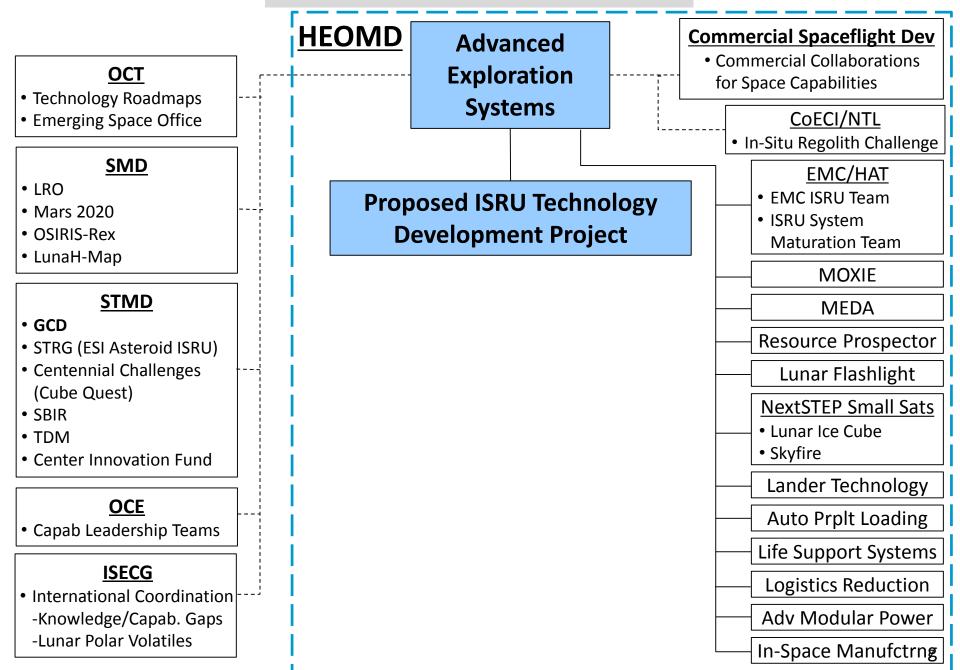
- Enables more affordable exploration than today's paradigm
- Allows more sustainable architectures to be developed
- Understand the ripple effect in the other Exploration Elements
  - MAV: propellant selection, higher rendezvous altitude (higher DV capable with ISRU propellants)
  - EDL: significantly reduces required landed mass
  - Life Support: reduce amount of ECLSS closure, reduce trash mass carried through propulsive maneuvers
  - Power: ISRU drives electrical requirements, reactant and regeneration for fuel cells for landers, rovers, and habitat backup

## **ISRU** Technology Development Needed Now (cont)



- Every Exploration Element except ISRU has some flight heritage (power, propulsion, habitats, landers, life support, etc.)
  - ISRU will require flight demonstration missions before it will be included in the critical path
    - Mission needs to be concluded at least 10 years before first human landed mission to ensure lessons learned can be incorporated into final design
  - ISRU Formulation team has generated a (still incomplete) list of over 75 technical questions on more than 40 components and subsystems that need to be answered before the 'right' ISRU system will be ready for this flight demo

#### **Current NASA ISRU Landscape**



## ISRU State-of-the-Art: Resource Acquisition, Processing, Consumables Production



- Significant work has been performed to demonstrate feasibility of ISRU concepts and develop components and technologies (TRL 1-3)
  - Moon/Mars
    - Mars atmosphere collection, separation, and processing into  $O_2$  or  $O_2/CH_4$
    - Lunar regolith excavation, beneficiation, and processing to extract  $O_2$
    - Civil engineering/soil stabilization
  - Asteroid
    - Acquisition concept work is just starting through STMD-ESI, BAAs, and SBIR/STTRs
- Some development & testing has been performed at the system level (TRL 4-6)
  - Moon (Lab, Analog sites)
    - RESOLVE, PILOT, ROxygen
  - Mars (Lab, Environment)
    - Portable Mars Production Plant (early '90s), MIP (flight experiment for cancelled Mars '01)
- However, significant work is needed to mature these technologies
  - Development & testing much closer to full-scale for human mission needs
  - Much longer operational durations
  - Much more testing outside the laboratory to validate performance under relevant environmental conditions
  - Integrate many components and subsystems into system prototypes
  - Realize synergy between ISRU and other system technologies, such as life support/fuel cell, power, surface mobility

## ISRU Critical Challenges That Need to Be Addressed



- What is the 'right' set of components and subsystems to enable production of mission consumables from either regolith or atmospheric resources at a variety of destinations?
- What is the performance and life that can be expected from the ISRU system in the actual environment?
- How does the ISRU system integrate and interact with other systems (e.g., power, lander, life support, etc.)?
  - ConOps
  - Power sharing
  - Total surface thermal management
  - Maintenance and refurbishment

#### Overall Goal: System-level TRL 6 to support future Pathfinder missions

#### Initial focus

#### Critical Technology Gap Closure

 icy/hard soil excavation, soil-water extraction, microchannel reactors, gas and water cleanup and separations, solid oxide stacks and life

#### Component Development in relevant environment (TRL 5)

- Better control of test conditions to identify which operating parameters critically affect performance
- Provide parametric data to validate and improve analytical models
- Interim Goal: ISRU Subsystems Tests in relevant environment (Subsystem TRL 6)
  - Use subsystem testbeds (CO<sub>2</sub> acquisition and compression, O<sub>2</sub> production, CH<sub>4</sub> production, water electrolysis, icy and hydrated soil acquisition, water extraction) to evaluate multiple technology options before downselect
  - Continue to use testbeds for new component technologies and long-duration testing

#### End-Goals

End-to-end ISRU System Tests in relevant environment (System TRL 6)

Oxygen & fuel production from atmosphere and soil resources including liquefaction and storage

Integrated ISRU-Exploration Elements Demonstration in relevant environment

 E2E ISRU System integrated with power system (including excavator recharging station), thermal rejection system, autonomous control, and interfaces to life support, MAV/Lander

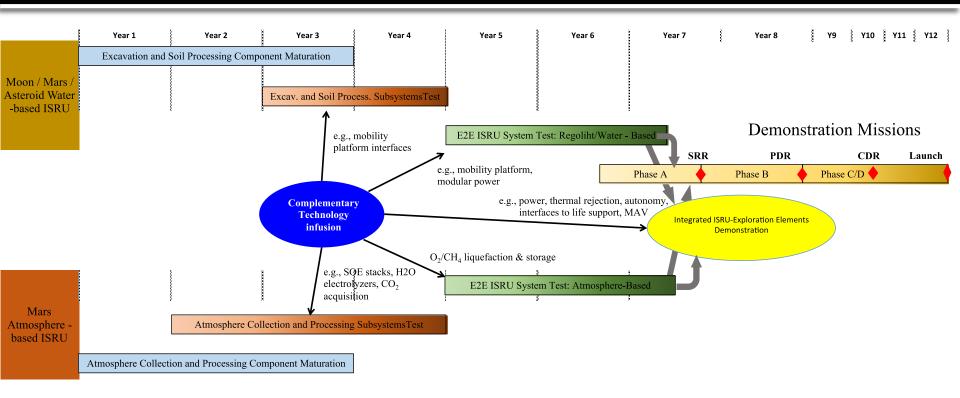
## Provide Exploration Architecture Teams with validated, highfidelity answers for mass, power, and volume of ISRU Systems





- Complementary technologies needed for Integrated ISRU Demo include:
  - Water electrolyzers and subsystem components
  - CO<sub>2</sub> acquisition/sequestration components
  - Advanced solid oxide stacks for CO<sub>2</sub> reduction
  - O<sub>2</sub> and CH<sub>4</sub> liquefaction and storage subsystems
  - Mobility platform
  - High specific-power batteries for excavation and delivery

#### **Possible Technology Development Schedule**



#### LEGEND





CO2 Freezer	Design Questions
Requirements: CO2 collection rate, outlet pressure (minimum), purity of the CO2	What is the optimal thickness of frozen CO2 to design cold head surface area to balance time to acquire and power efficiency? (i.e., power to freeze additional CO2 increases as thickness increases
<b>Requirements Drivers:</b> Outlet Pressure: comes from SOE (lower pressure), RWGS and Sabatier (higher pressure better)	
$CO_2$ collection rate: 1) $O_2$ production rate required, 2) $CO_2$ conversion efficiency ( $O_2$ production method and $CO_2$ recycling)	Size of vessel compared to cold head: Space between cold fins and wall - bigger vessel means lower max pressure upon thaw (i.e., becomes all vapor sooner) so thinner walls, but affects incoming flow around cold head and then heat transfer into cold head during thaw if walls too far away?
	How do you get good flow around the cold head as you scale up and the cold head becomes bigger and more complicated, and what is the scale limit? How do you control the flow (both pressure and mass flow) as you drain the pressurized CO <sub>2</sub> acquisition vessel, the temperature starts dropping along the liquid/vapor saturation curve)? Cryocooler scale up: efficiency needed? Type of cycle needed?

### Summary



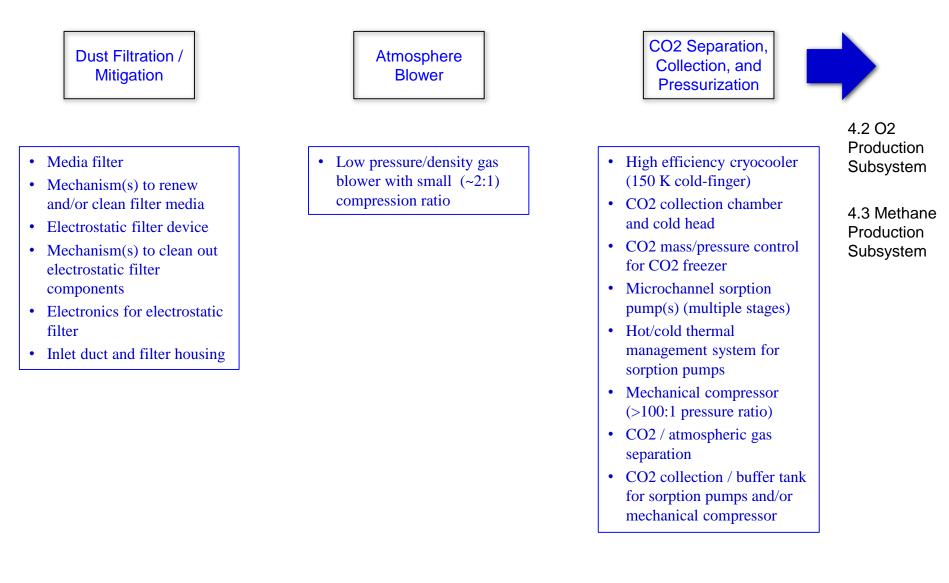
- An ISRU Technology Development plan has been proposed to the HEOMD AES program
  - Focus on acquisition and consumables production and storage
  - Component –> subsystem -> system progression
- Raise System-level TRL in preparation for potential demonstration mission
- Provide Exploration Architecture Teams with validated, highfidelity answers for mass, power, volume, and concept of operations



## **Back-up Slides**

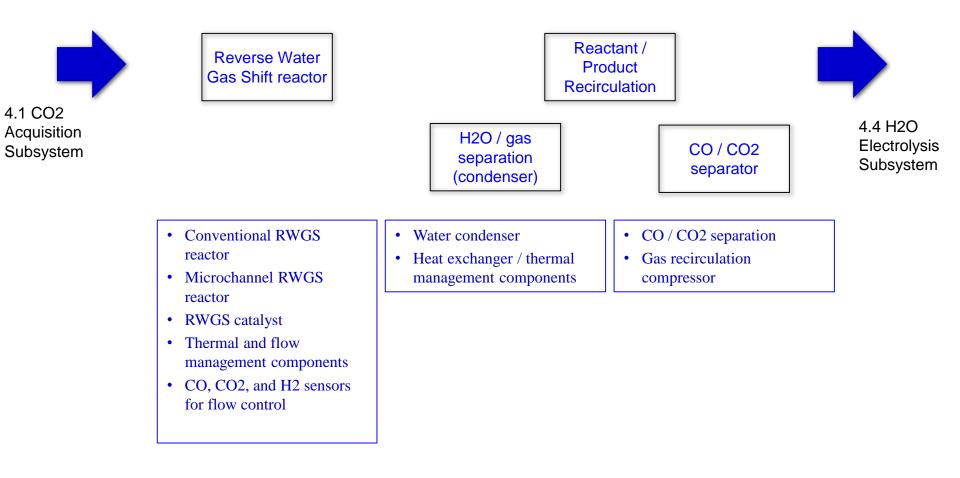
# 4.1 Atmosphere Carbon Dioxide (CO<sub>2</sub>) Collection Subsystem





#### 4.2 Oxygen (O2) Production Subsystem (4.2.1 Reverse Water Gas Shift option)

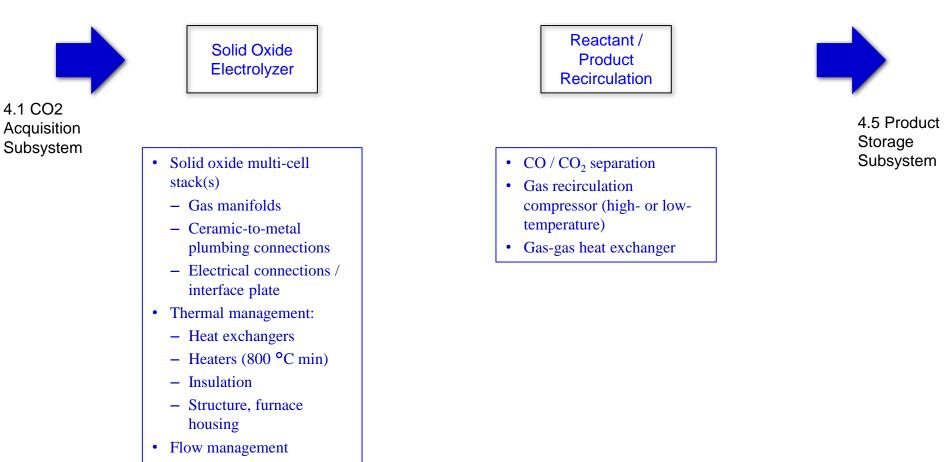






# 4.2 Oxygen (O2) Production Subsystem (4.2.2 SOE option)

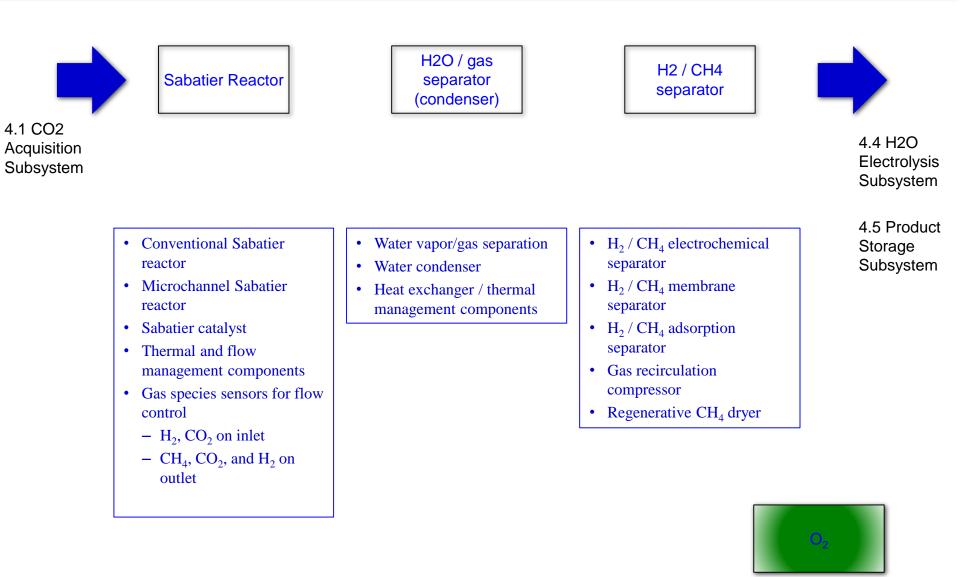




• O<sub>2</sub> and CO sensors on outlets

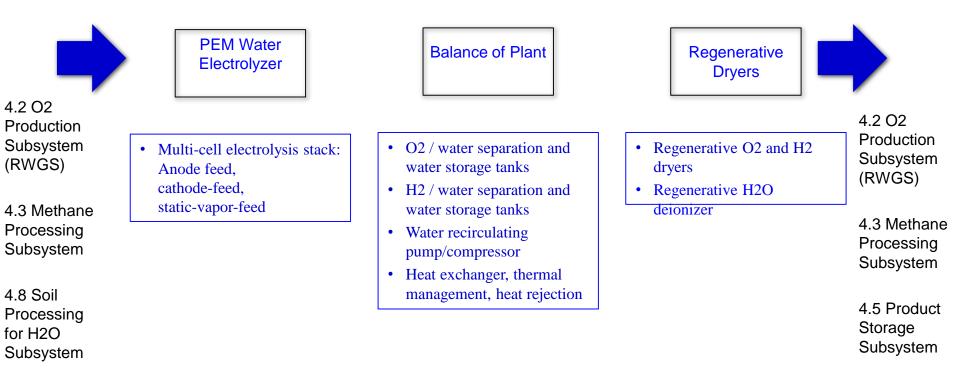
O<sub>2</sub>

## 4.3 Methane (CH4) Fuel Production Subsystem

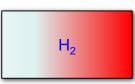


### 4.4 Water Electrolysis Subsystem



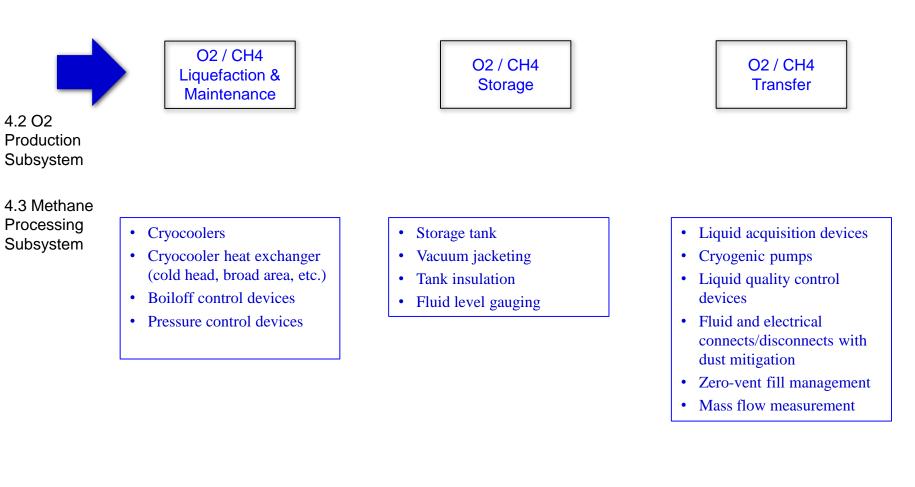






## 4.5 Product Storage and Distribution Subsystem

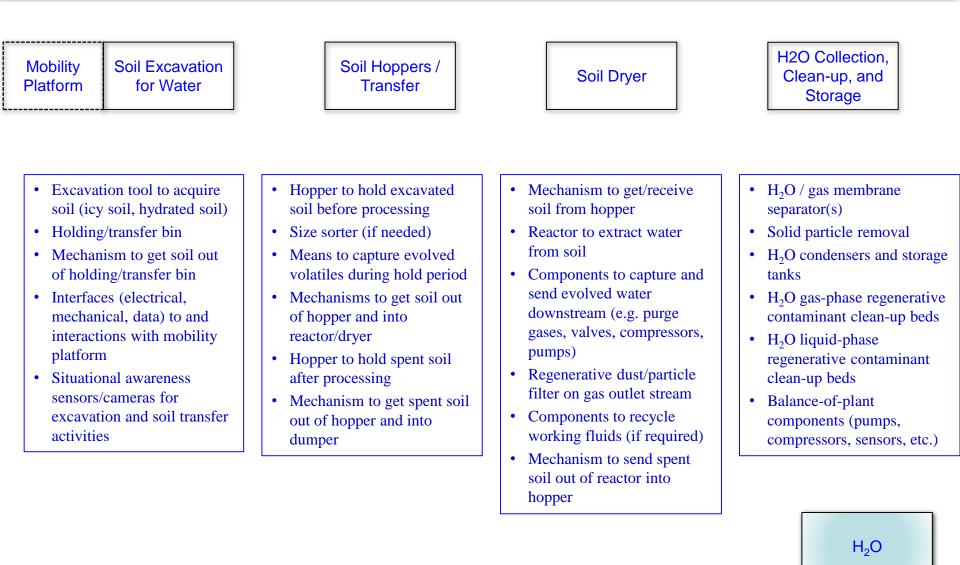






## 4.7 Excavation and 4.8 Soil Processing Subsystems



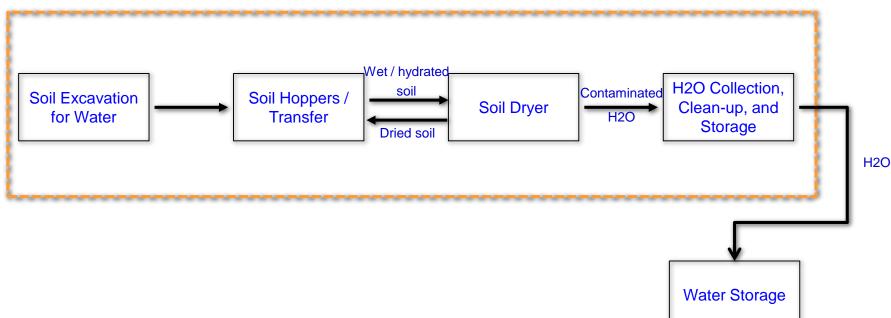




## **ISRU** Integrated Systems

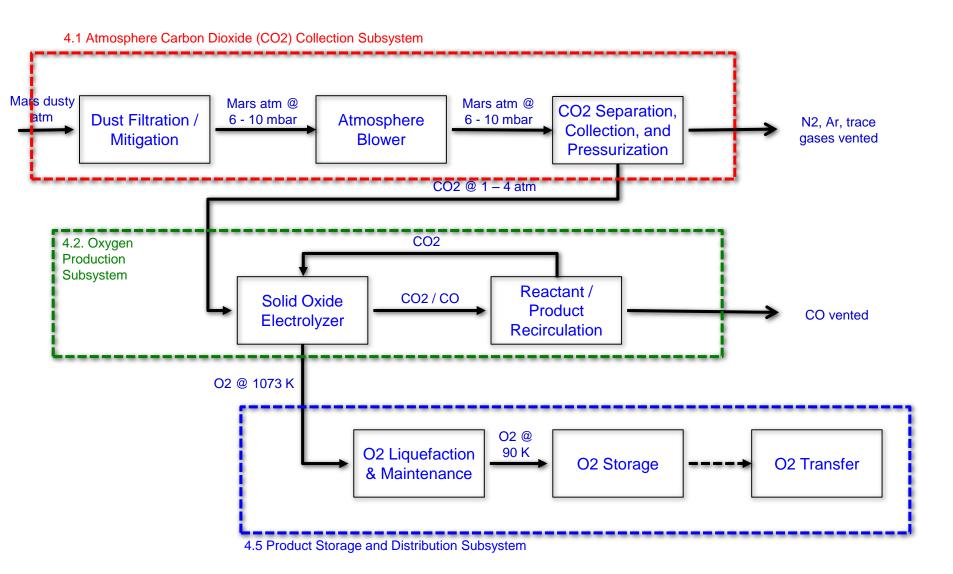


#### 4.7 & 4.8 Excavation and Soil Processing Subsystems



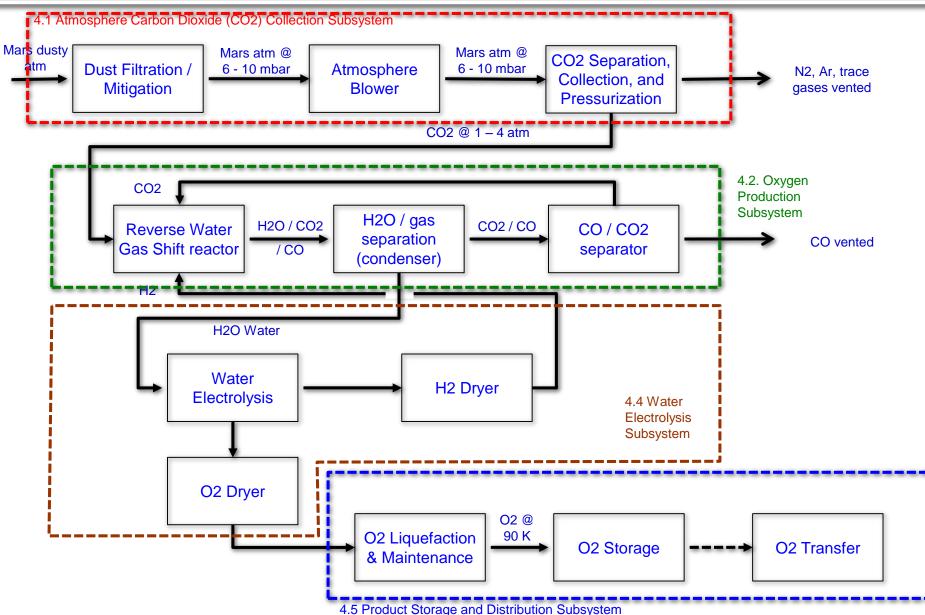
## Oxygen Production from Atmosphere Integrated System (SOE Option)





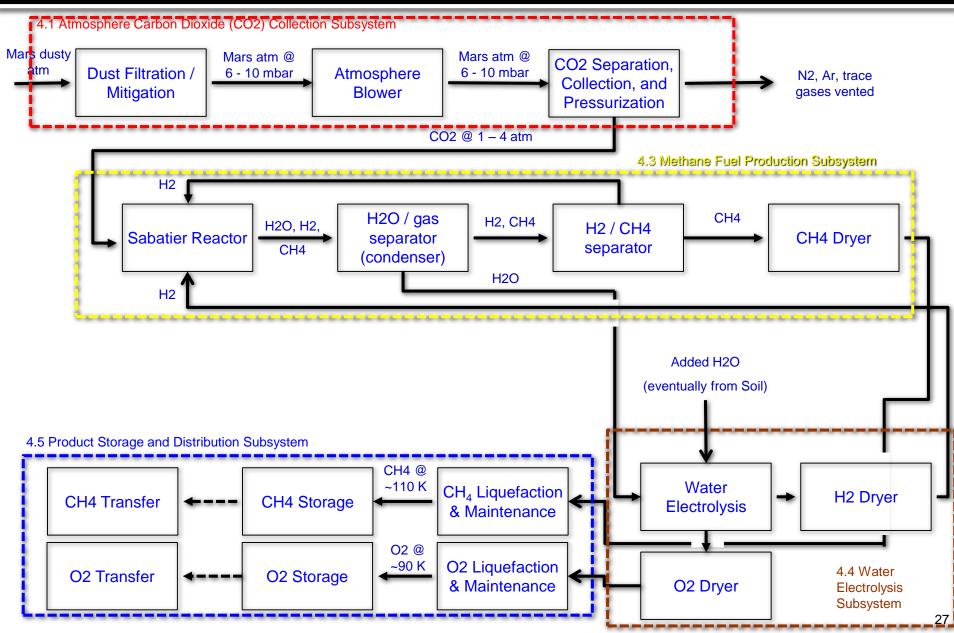
## Oxygen Production from Atmosphere Integrated System (RWGS Option)





## **Fuel and Oxygen Production Integrated System**





## ISRU Fuel and Oxygen Production End-to-End Integrated System



