

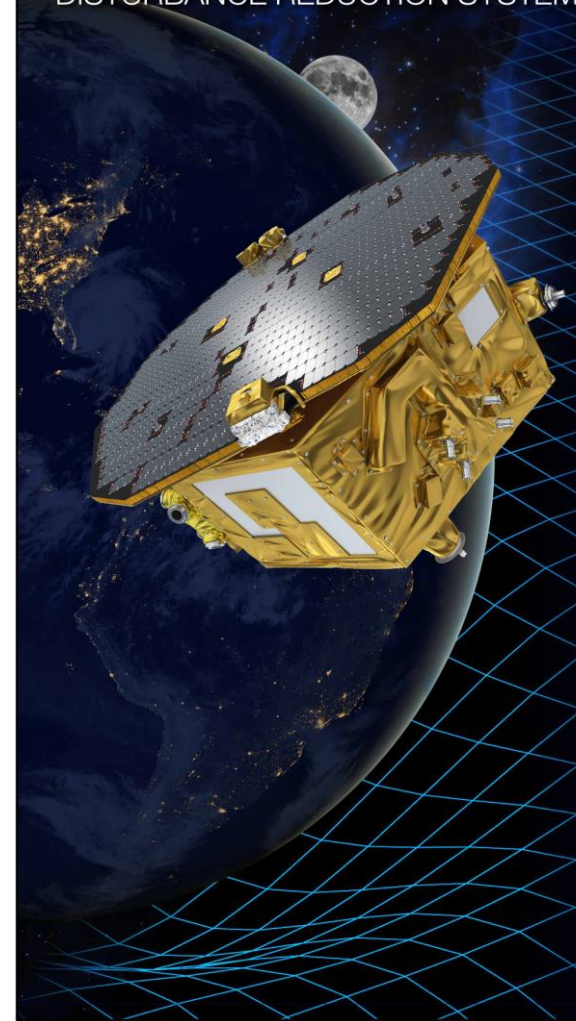
# Dynamic Control System Performance During Commissioning of the Space Technology 7 – Disturbance Reduction System Experiment on LISA Pathfinder

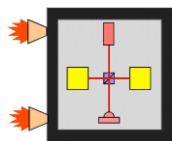
Oscar Hsu<sup>1</sup>, Peiman Maghami<sup>1</sup>,  
James R. O'Donnell Jr.<sup>1</sup>, John  
Ziemer<sup>2</sup>, Andrew Romero-Wolf<sup>2</sup>

<sup>1</sup> NASA Goddard Space Flight Center

<sup>2</sup> NASA/Caltech Jet Propulsion Lab

**ST7-DRS**  
**SPACE TECHNOLOGY 7**  
DISTURBANCE REDUCTION SYSTEM

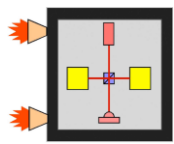




# Outline



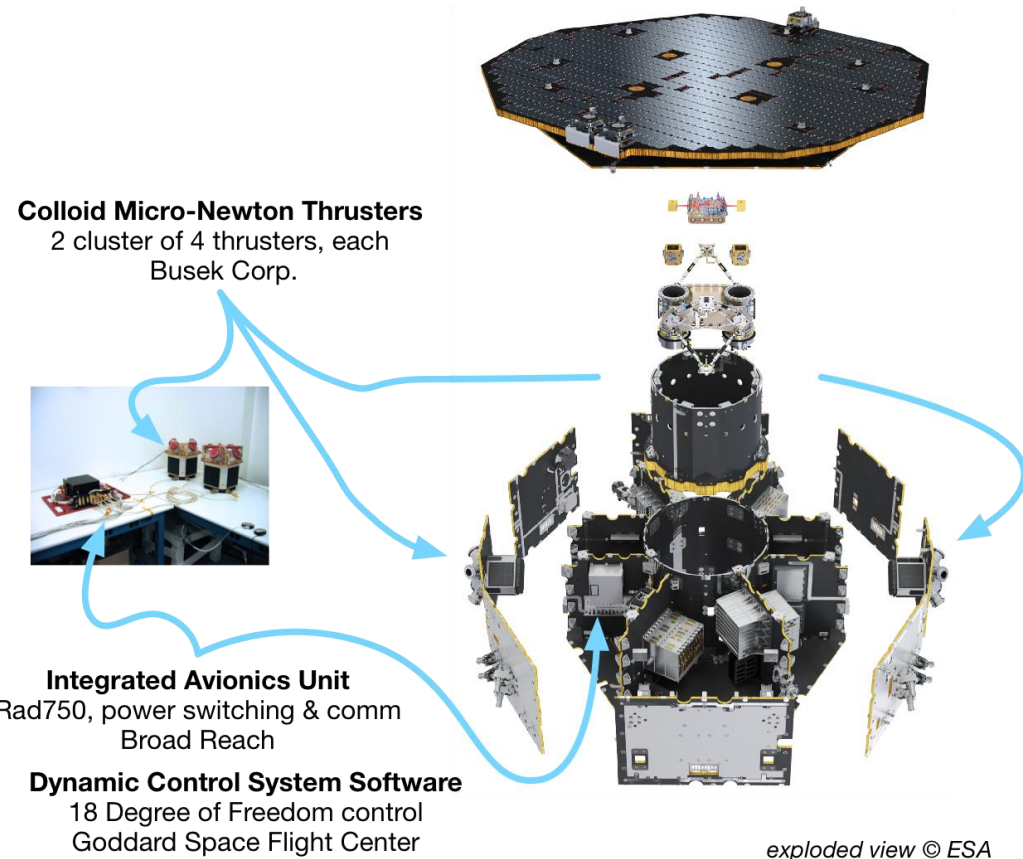
- LISA Pathfinder & DRS
- DRS Mission Timeline Highlights
- DRS Mission Modes
- Thruster Checkout
- Handover/Handback
- Zero-G Mode
- Drag-Free Low Force
- Thruster Anomaly & Resolution
- Acknowledgements
- Conclusion

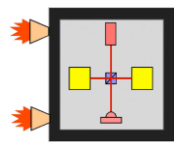


# LISA Pathfinder & DRS



- DRS is a NASA Contribution to the ESA LISA Pathfinder Mission
- Three Components:
  - Integrated Avionics Unit
  - Colloid Micro-Newton Thrusters
  - Dynamic Control System Software

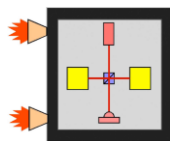




# DRS Mission Timeline Highlights



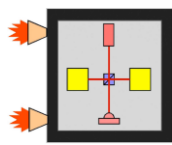
- 2002 – DRS Project Started
- 2006 – DCS Control Design Completed
- 2008 – DRS Technology Package Delivered to LISA Pathfinder
- 2015 – DCS algorithms updated.
- December 3, 2015 at 04:04 UTC – Launch of LISA Pathfinder
- December 12, 2015 – Transfer to Sun-Earth L1 Point Begins
- **January 2-10, 2016 – DRS Thruster Checkout**
- January 22, 2016 – Arrive at L1 Point/Propulsion Module Separation
- March 1, 2016 – LISA Pathfinder Science Mission Starts
- June 24, 2016 – LISA Pathfinder Completes first operations Phase
- **June 27-July 7, 2016 – DRS Instrument Checkout**
- **July 8, 2016 – DRS Thruster Anomaly**
- **August 8, 2016 – DRS Recommissioning Begins**
- August 12, 2016 – Final day of recommissioning, 18-DOF Control Achieved
- **August 13, 2016- DRS Experiment Phase Begins!!**



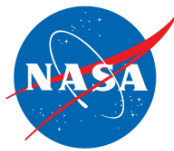
# DRS Mission Modes



DRS Mission Mode	Spacecraft Control Mode	Reference Test Mass Control Mode	Reference Test Mass Force Mode	Non-Reference Test Mass Control Mode	Non-Reference Test Mass Force Mode
Standby	Standby	DFS Standby	N/A	DFS Standby	N/A
Attitude Control	Attitude-Only	DFS Accelerometer	High Force	DFS Accelerometer	High Force
Zero-G	Accelerometer				
Drag Free Low Force	Drag Free 1	DFS Drag Free 1	Low Force	Suspended Drag Free 1	Low Force
18-DOF Transitional					
18-DOF					



# Thruster Checkout



- Two periods for Thruster Checkout
  - January 2-10, 2016
    - Propulsion Module still attached
  - June 27 – July 1, 2016
- Activities:
  - Thruster Impedance
  - Thruster Bubble & Dissipation Test
  - Thruster Functional Test
  - Thruster Characterization Tests

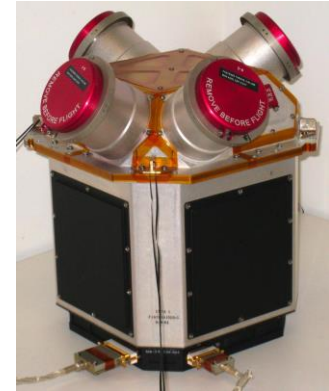


Image Credit: ESA/NASA/JPL-Caltech

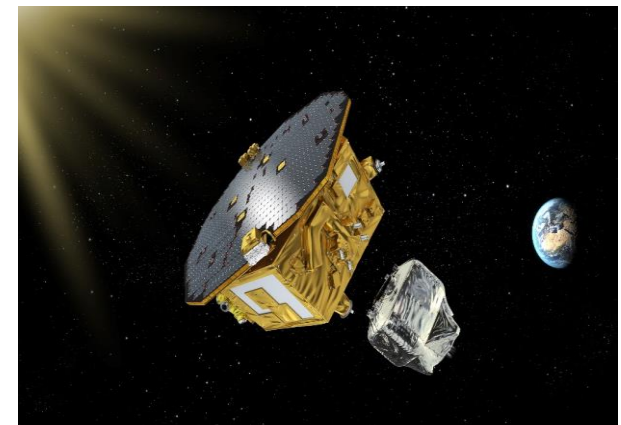
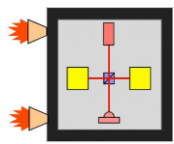
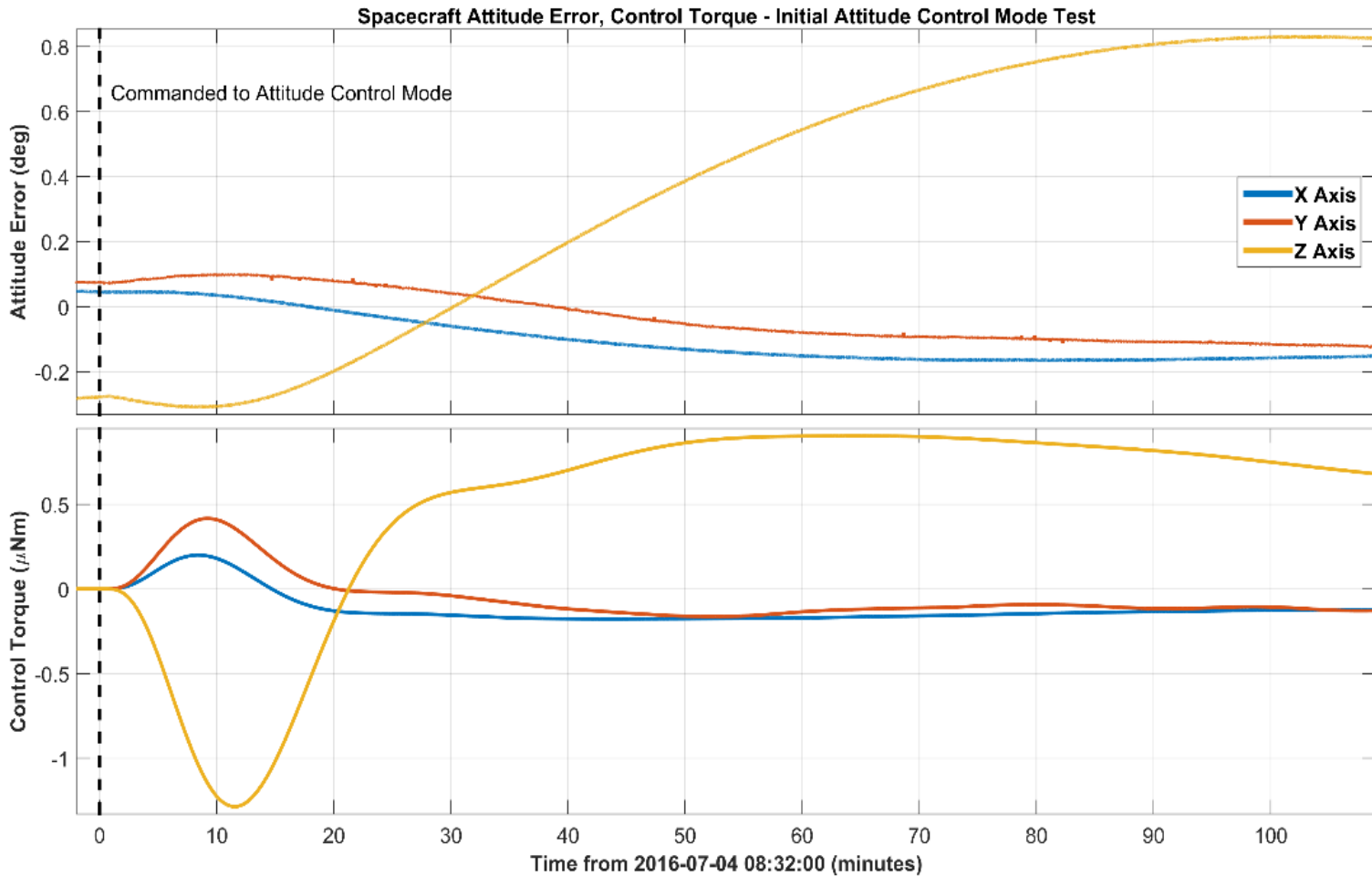
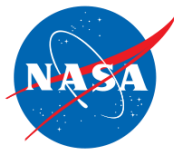


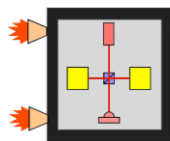
Image Credit: ESA/C. Carreau



# Handover/Handback Test



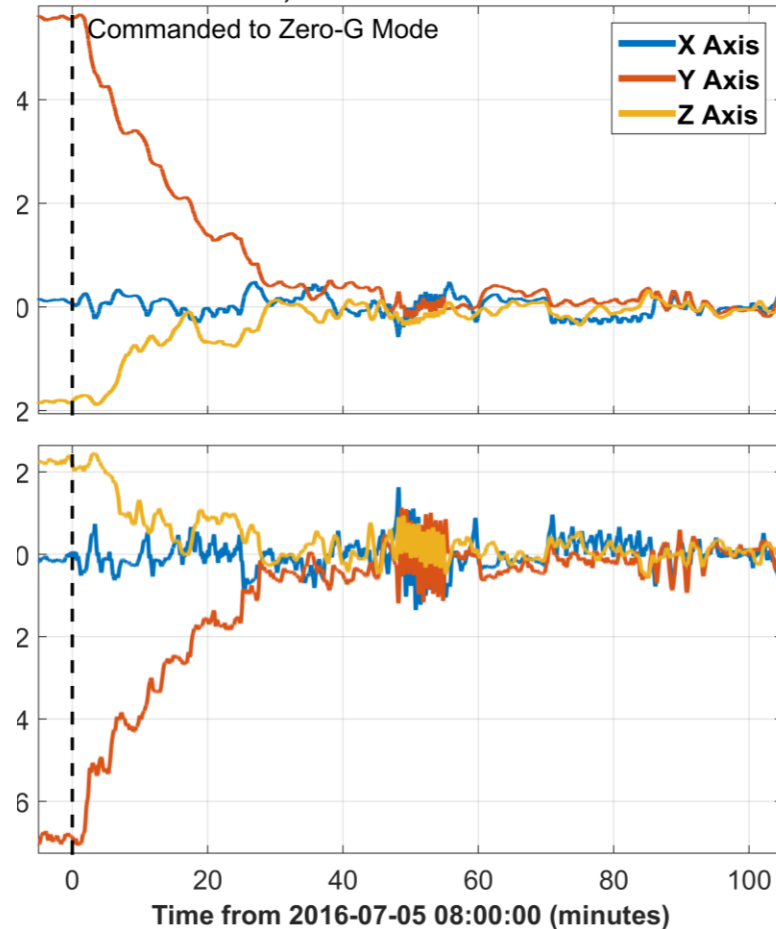
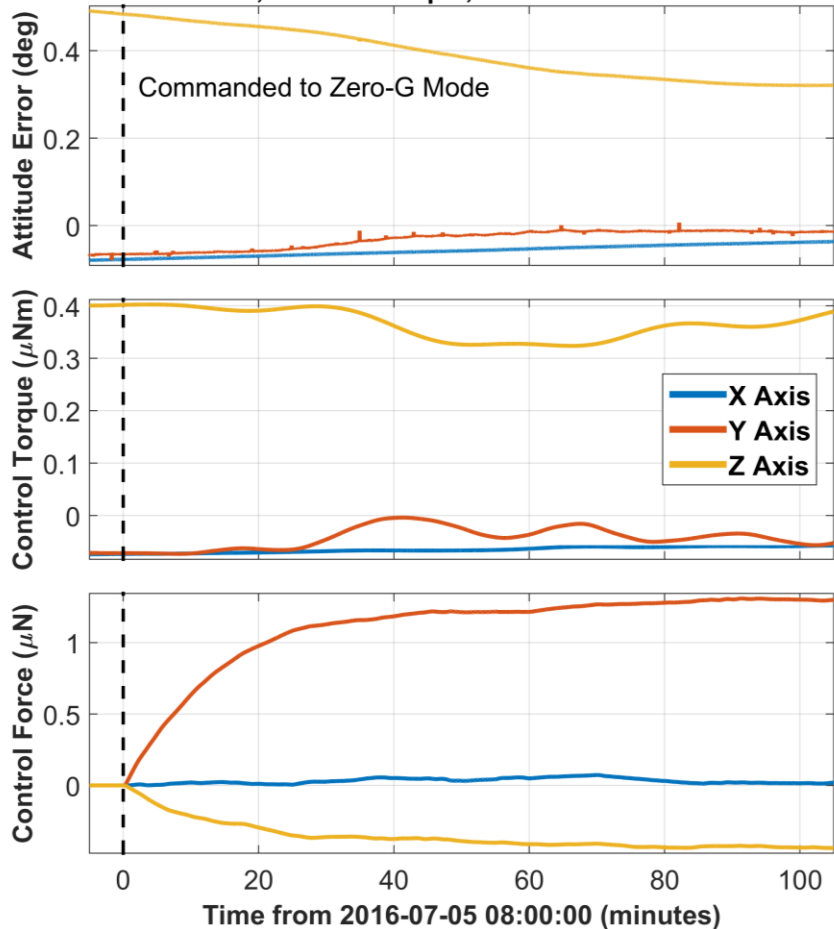
DRS given control on July 4, 2016 at 08:32 UTC



# Zero-G

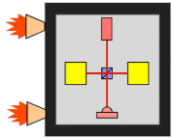


Spacecraft Attitude Error, Control Torque, Control Force - Initial Zero-G Mode Test Test Mass 1 Position, Control Force - Initial Zero-G Mode Test



Zero-G Mode commanded on July 5, 2016 at 08:00 UTC



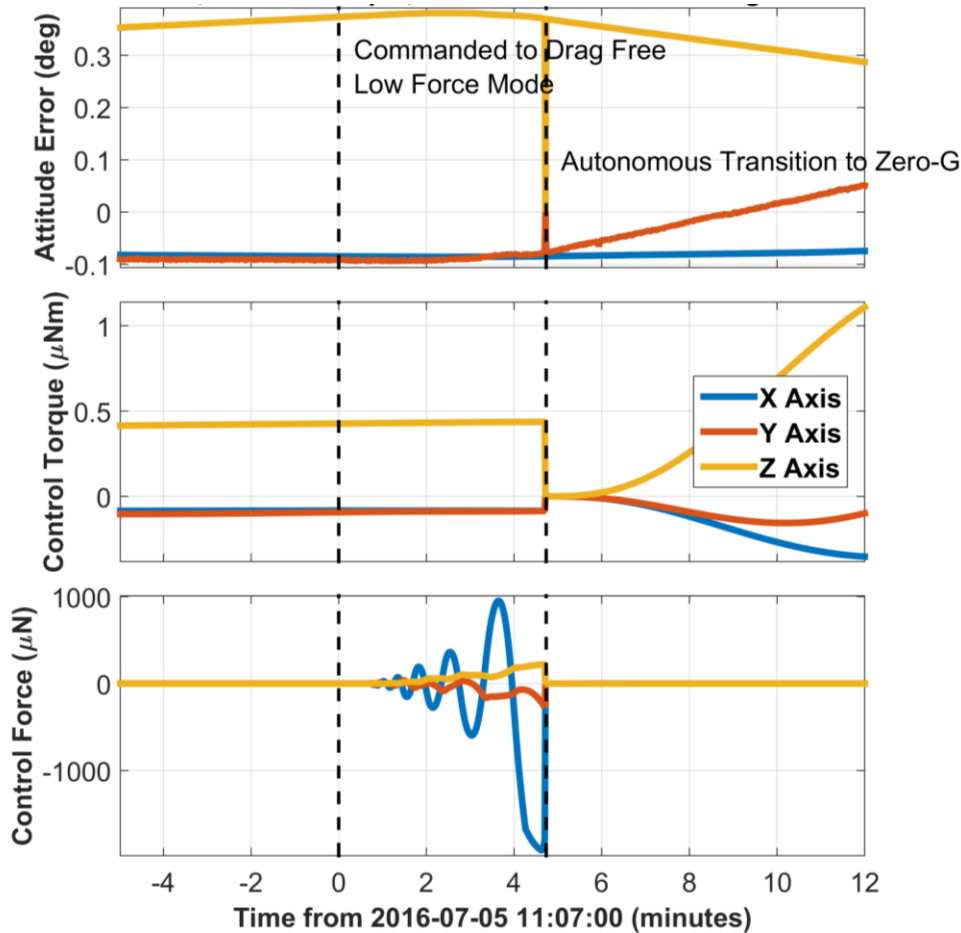


# Drag-Free Low Force: 1<sup>st</sup> Attempt

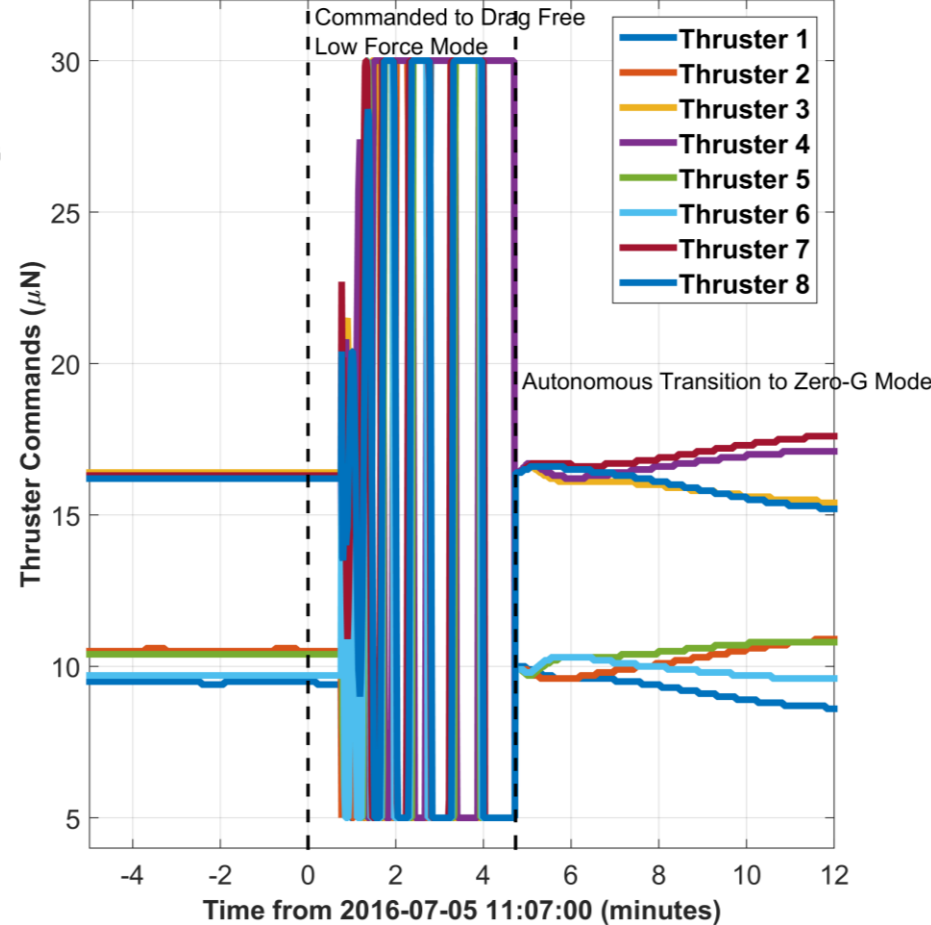


## Spacecraft Attitude Error, Control Torque, Control Force – Initial

### Drag Free Low Force

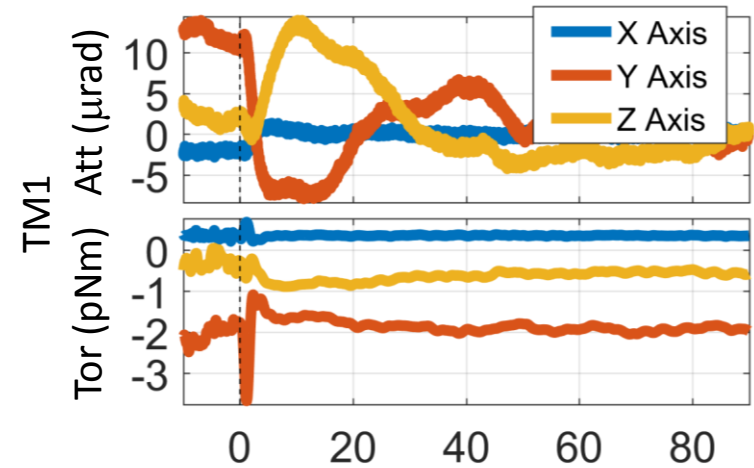
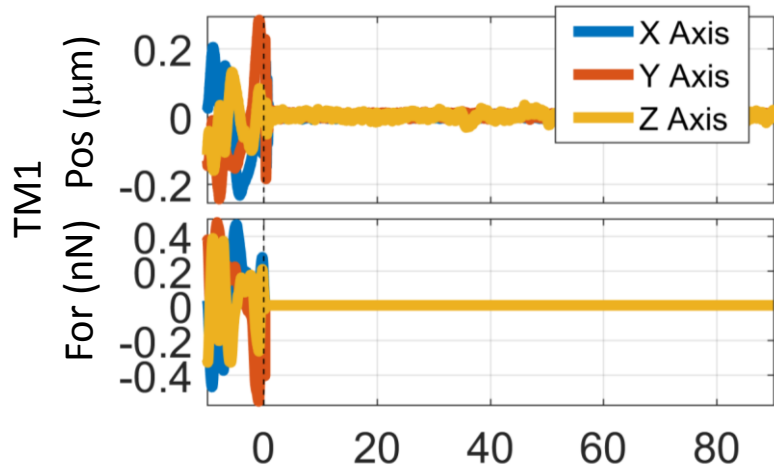
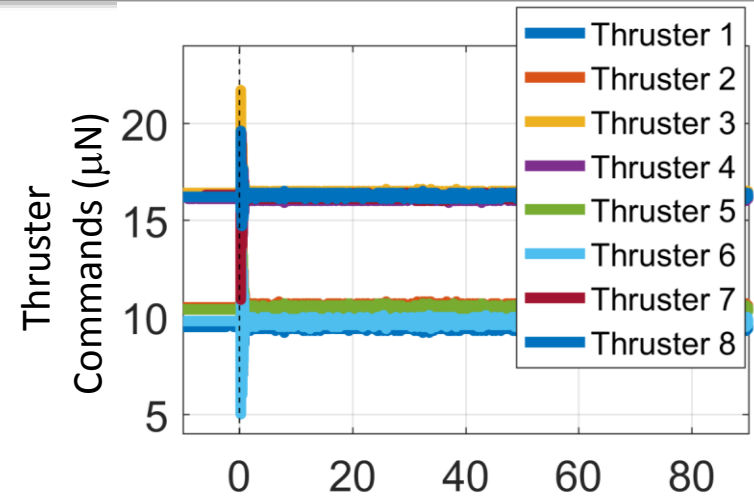
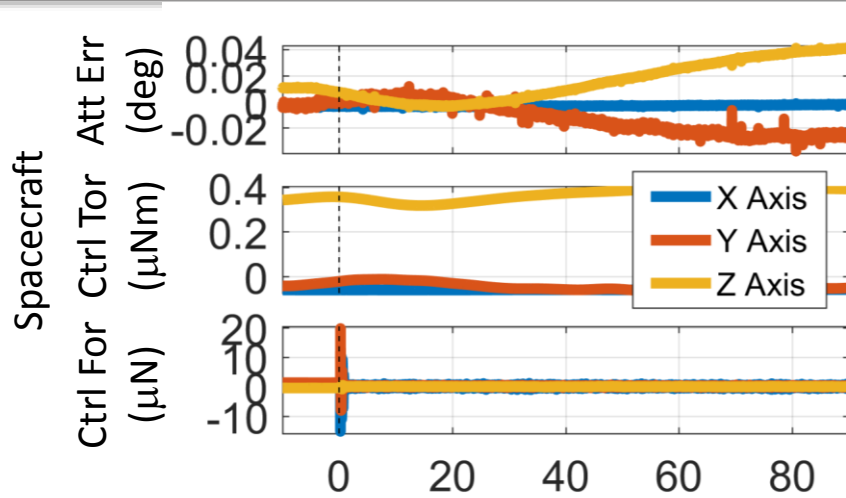
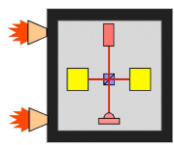


## Thruster Commands - Initial Drag Free Low Force Mode Test



Drag-Free Low Force Commanded on July 5, 2016 at 11:07 UTC  
Transition Aborted at 11:12 UTC

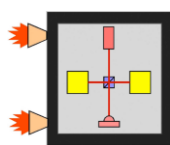
# Drag-Free Low Force: 2<sup>nd</sup> Attempt



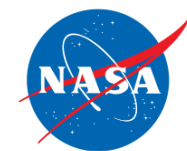
Time from 2016-07-06 05:50:00 (minutes)

Two-Pronged Approach for transition:

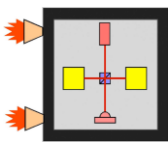
- Transition RTM to Low-Force earlier (July 6, 2016 at 05:30 UTC)
- Limit Spacecraft Force Command to 5 micron per axis. (July 6, 2016 at 11:00 UTC)



# Final Commissioning Tests and Bonus Activities



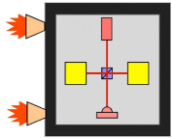
- Final Commissioning Tests executed on July 7-8, 2016
  - Purpose: Verify remaining command sequences needed for the experiment phase:
  - Activities:
    - Test Mass Signal Injection
    - Single Thruster Open Loop Test
    - Freeze Thruster Open Loop Test
    - System Level Delay Measurements
    - DRS use Optical Measurement System Data for Control
    - Change Reference Test Mass to the 2<sup>nd</sup> test mass.
- All Tests were successful with the exception of the system level delay test.
- Following week will be transitioning to 18-DOF Mode



# Thruster Command Anomaly



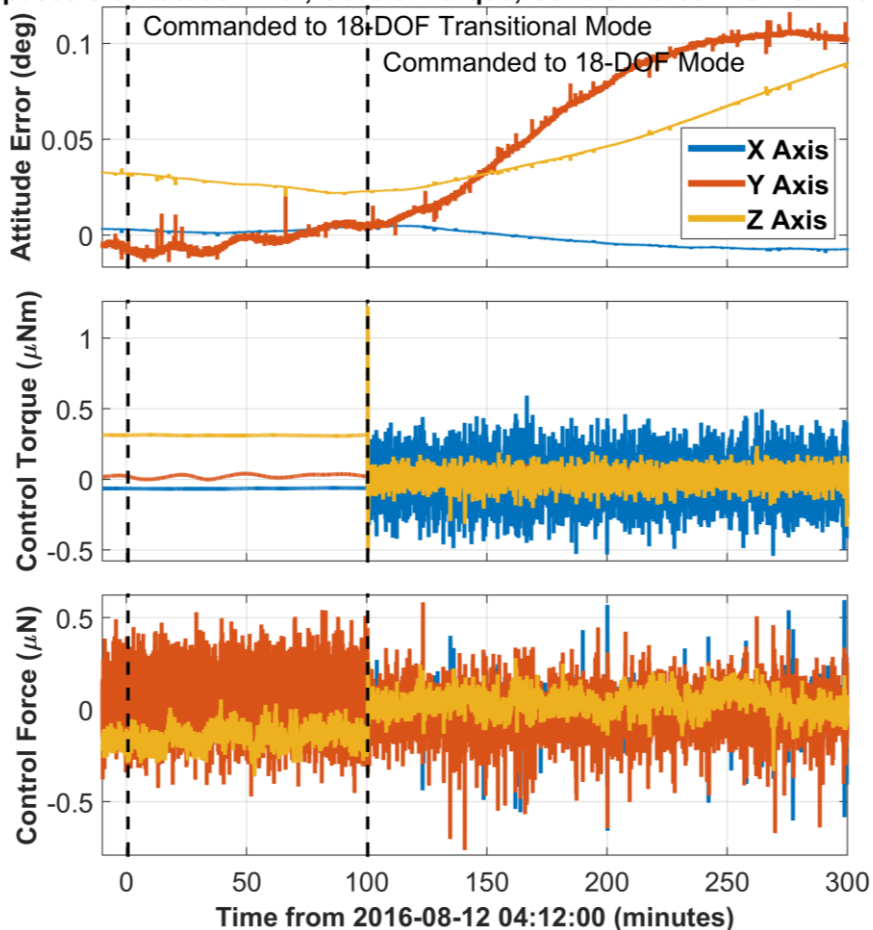
- July 9, 2016 – Acquisition of Signal showed that DRS was no longer in control of LISA Pathfinder.
- Upon Investigation, Reset of Thruster Cluster 2 occurred due to single event effect. The Single Event Effect caused the HW Logic to reset the cluster upon receipt of thruster commands (Thruster Command Mode).
- DRS FSW and Thrusters include a second a method to command thrusters (Thruster Diagnostic Mode).
  - Used during thruster commissioning.
  - Never used for close-loop control.
- Needed changes tested over 3 week period by JPL and GSFC engineers.



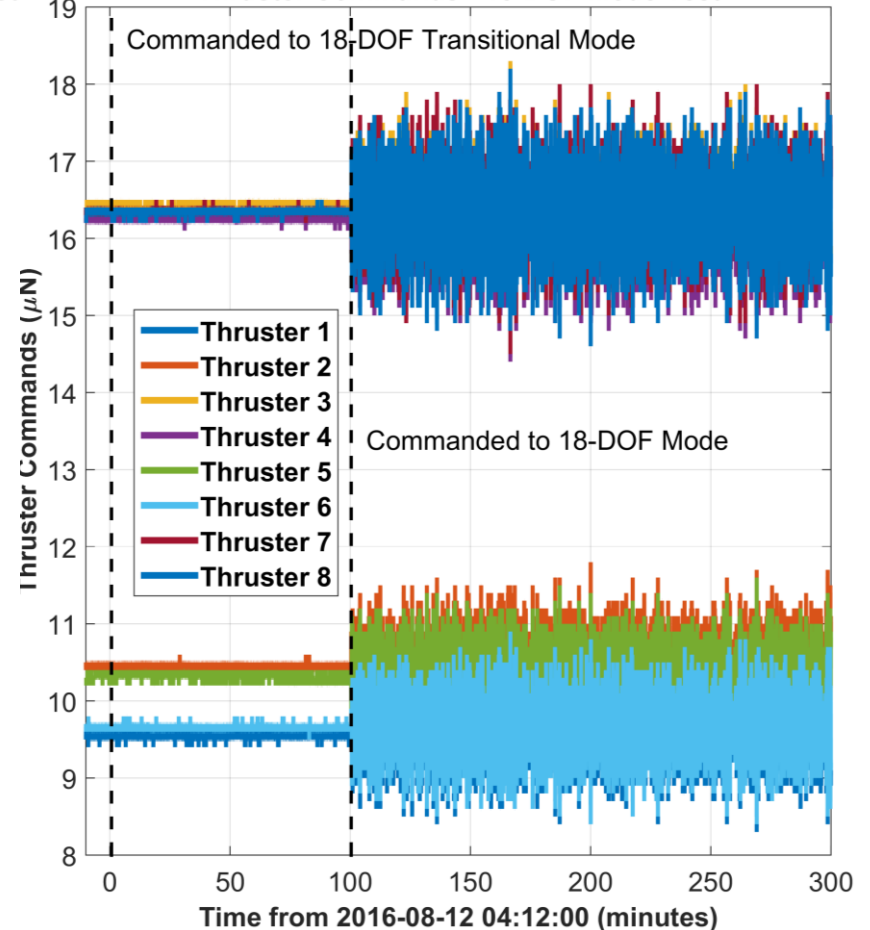
# Recommissioning and 18-DOF



Spacecraft Attitude Error, Control Torque, Control Force - 18-DOF Mode Test

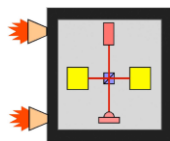


Thruster Commands - 18-DOF Mode Test

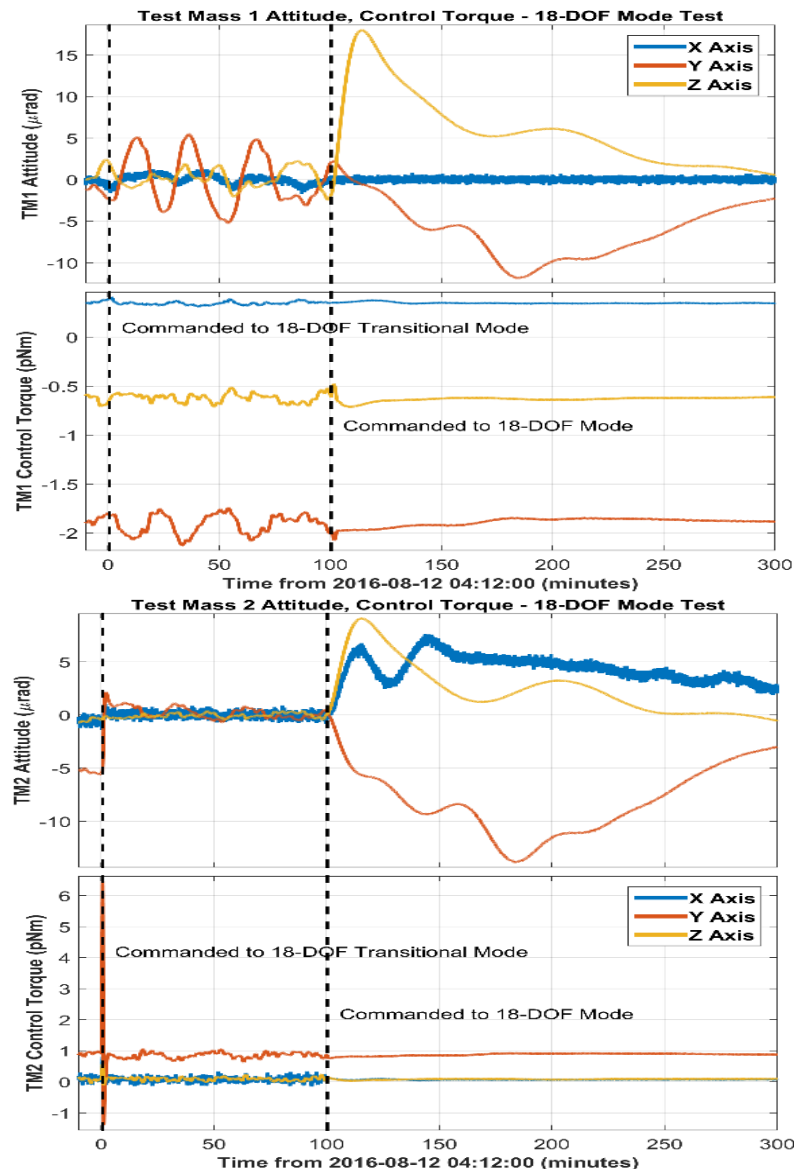
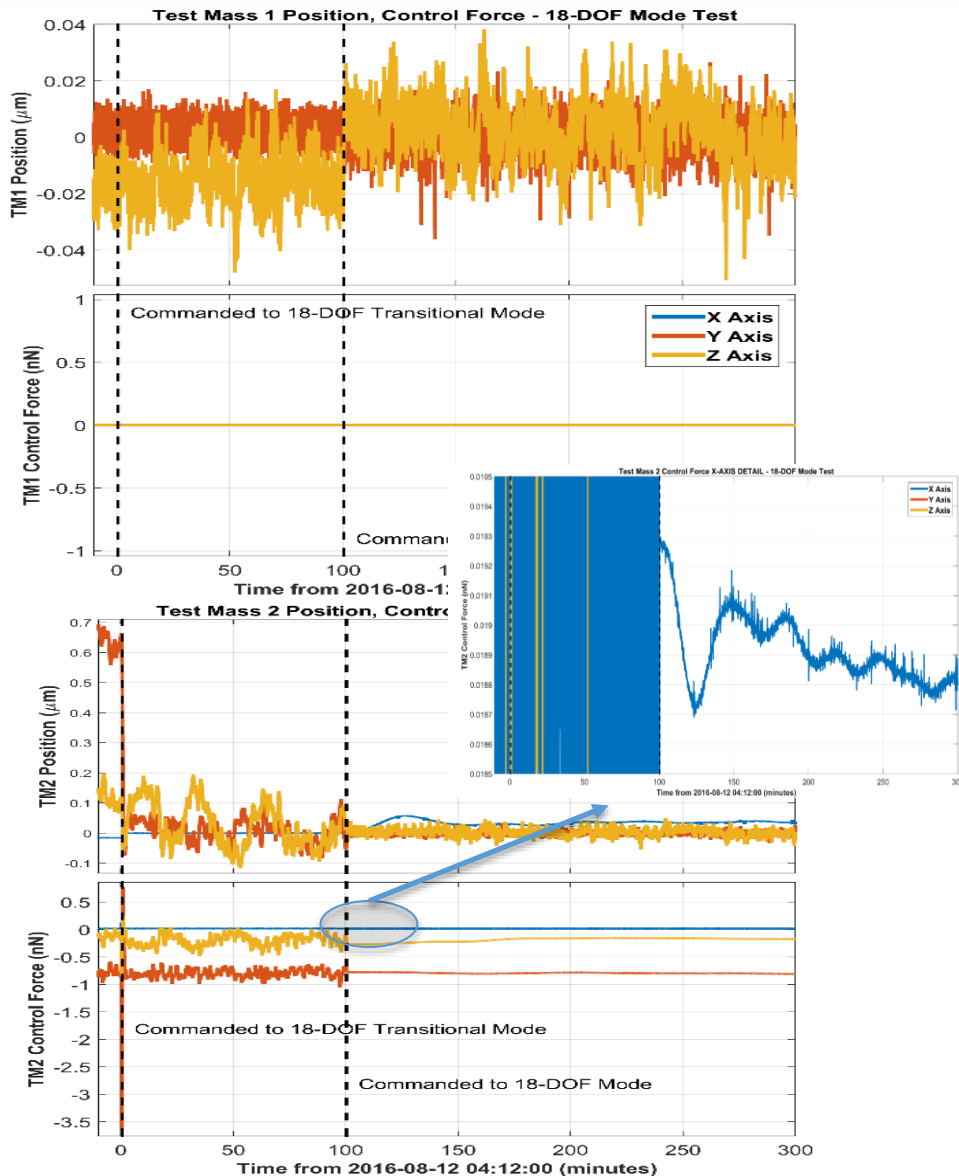


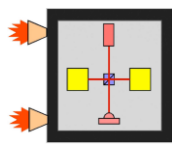
Recommissioning Activities – August 8-12, 2016

Completed with successful transition to 18-DOF Mode on August 12, 2016 at 05:52 UTC

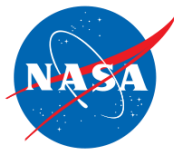


# 18-DOF Test Masses

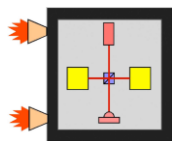




# Acknowledgements



- the Hammers Company
- Busek Co.
- Airbus Defense and Space in Stevenage, UK
- European Space Operations Centre in Darmstadt, Germany
- LISA Pathfinder Spacecraft and Science Team

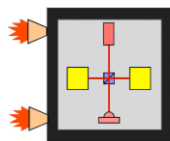


# Conclusion



- DRS launched aboard LISA Pathfinder on December 3, 2015.
- Three highly successful commissioning periods with two originally planned a third added after thruster anomaly.
- DCS Modes and Mode Transitions successfully verified.





# Questions?