



# Insights from Safety Tests with an On-Demand Internal Short Circuit Device in 18650 Cells

By

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International Battery Seminar Fort Lauderdale, FL 21-23 Mar 2017

#### **Outline**

- 5 Design Guidelines
- Trading thermal isolation vs heat dissipation
  - Full thermal isolation
  - Drawing heat from cell bottoms
  - Full can length interstitial heat sink approach
- Risk of side wall rupture during thermal runaway
- New cell designs with cell bottom vent from Sony and LG
  - Vent & burst pressure
  - Thermal runaway performance
- Summary of findings to date
- Future work

Some of NASA's Future Battery Applications

#### Robonaut 2

- To enhance and reduce frequency of manned spacewalks
- High energy density and high specific energy battery needed
- 90V, 4 kWh, 7 hour mission

#### Mars Rover Vehicle

- Terrestrial demonstration vehicle needing high voltage, power battery
- 400V, 4 kWh, 1 hour mission

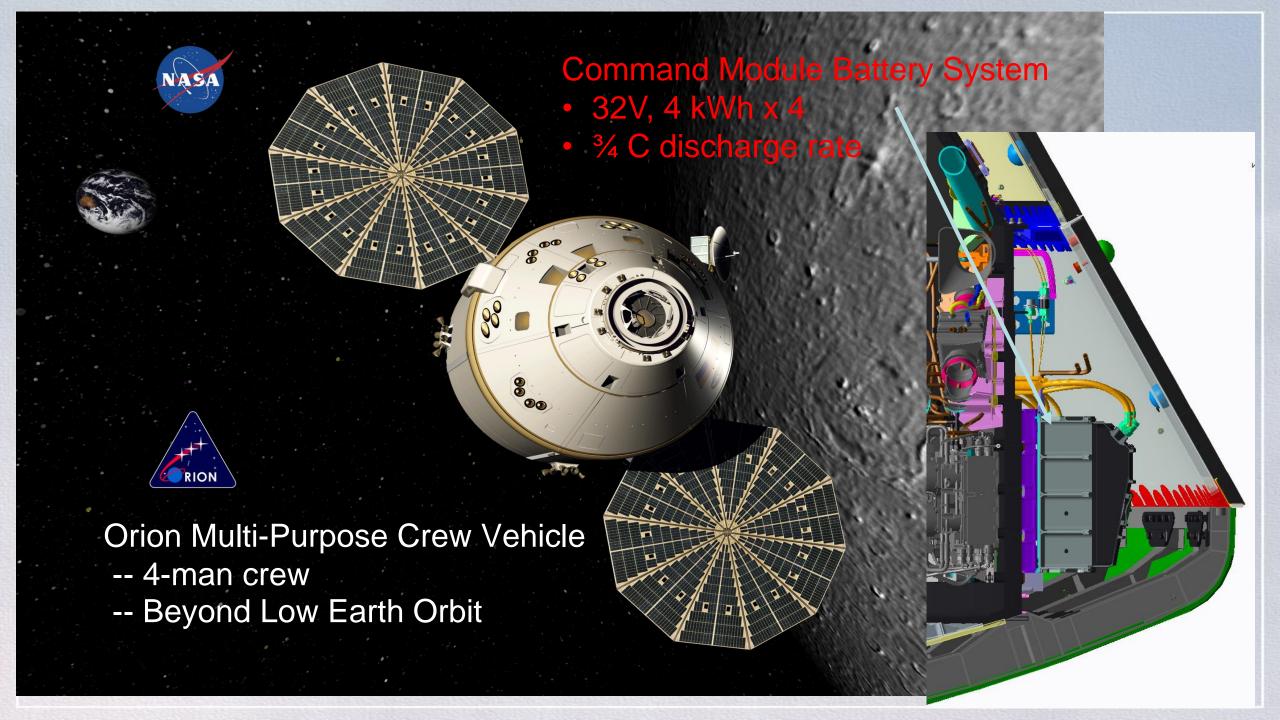
#### Valkyrie, RoboSimian

- Terrestrial dangerous operations robot
- 90V, 2kWh, 1 hour mission

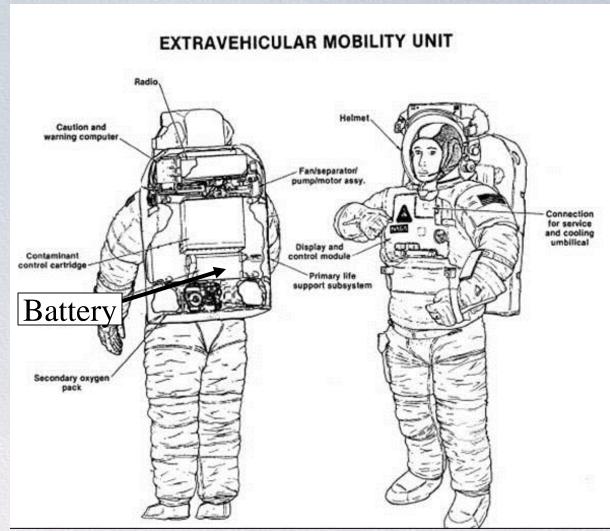
#### X-57 Electric Plane

- All electric aircraft demonstrating distributed electric propulsion
- 525V, 50 kWh, 1 hour mission





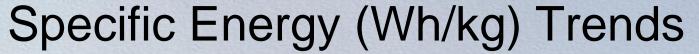
### Current Li-ion Spacesuit Battery

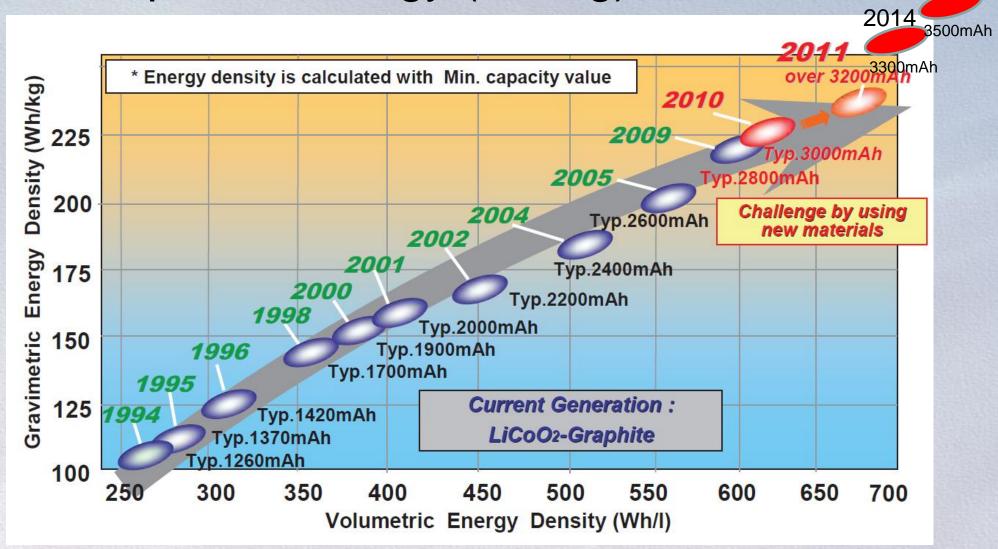




Used on over 22 spacewalks for far

2016

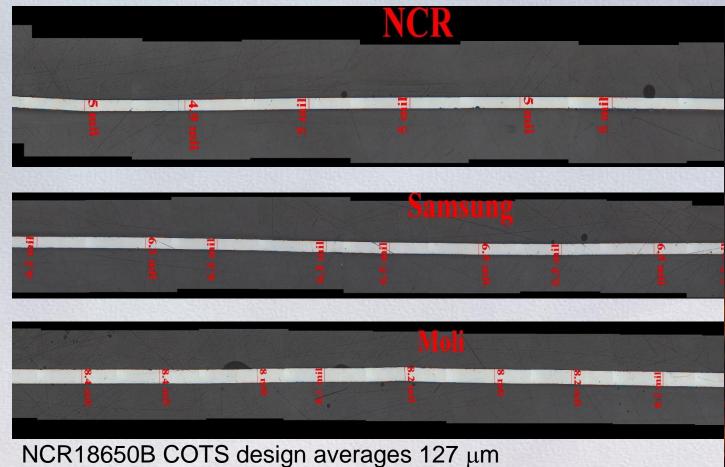


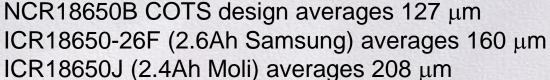


Source: Sanyo/Panasonic 2010

A high production rate design that achieves > 240 Wh/kg and > 660 Wh/L exists since 2012 Specify energy improvements are trending at 7-10% per year....should get to 300 Wh/kg by 2017

#### Cell Can Wall Cross Sections

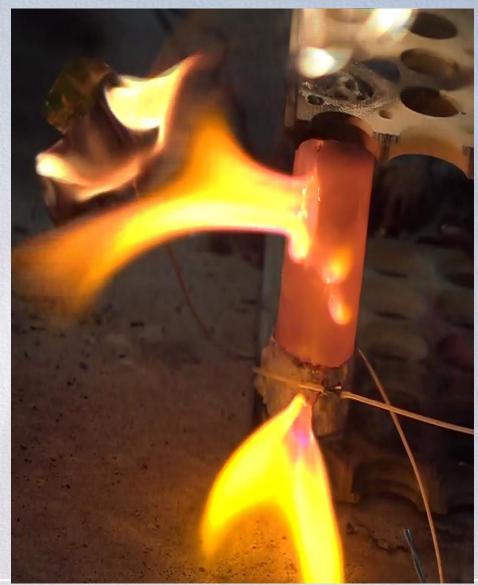




Thin can wall with >660 Wh/L → high propensity to side wall ruptures/breaching Other factors include high reaction kinetics and high header crimp burst pressure

### 5 Design Driving Factors for Reducing Hazard Severity from a Single Cell TR

- Reduce risk of cell can side wall ruptures
  - Without structural support most high energy density (>660 Wh/L) designs are very likely to experience side wall ruptures during TR
  - Battery should minimize constrictions on cell TR pressure relief
- Provide adequate cell spacing and heat rejection
  - Direct contact between cells nearly assures propagation
  - Spacing required is inversely proportional to effectiveness of heat dissipation path
- Individually fuse parallel cells
  - TR cell becomes an external short to adjacent parallel cells and heats them up
- Protect the adjacent cells from the hot TR cell ejecta (solids, liquids, and gases)
  - TR ejecta is electrically conductive and can cause circulating currents
- Prevent flames and sparks from exiting the battery enclosure
  - Provide tortuous path for the TR ejecta before hitting battery vent ports equipped flame arresting screens



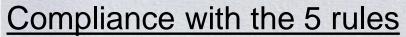
**Current Spacesuit Battery Design** 



**Design Features** 

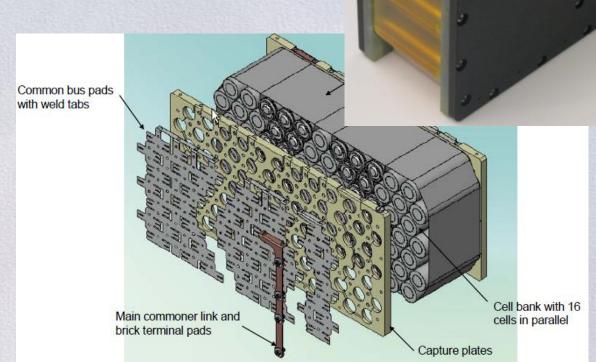
• 80 Li-ion cells (16p-5s)

 ICR-18650J from E-one Moli Energy (2.4Ah)



- Minimize side wall ruptures
- No direct cell-cell contact
- Individually fusing cell in parallel
- Protecting adjacent cells from TR ejecta
- Include flame arresting vent X ports





Solid Al side panels

block cell vents

#### Design Propagates TR – Catastrophic Hazard



Battery external surfaces reach 350°C Vented some sparks and much smoke for > 15 min

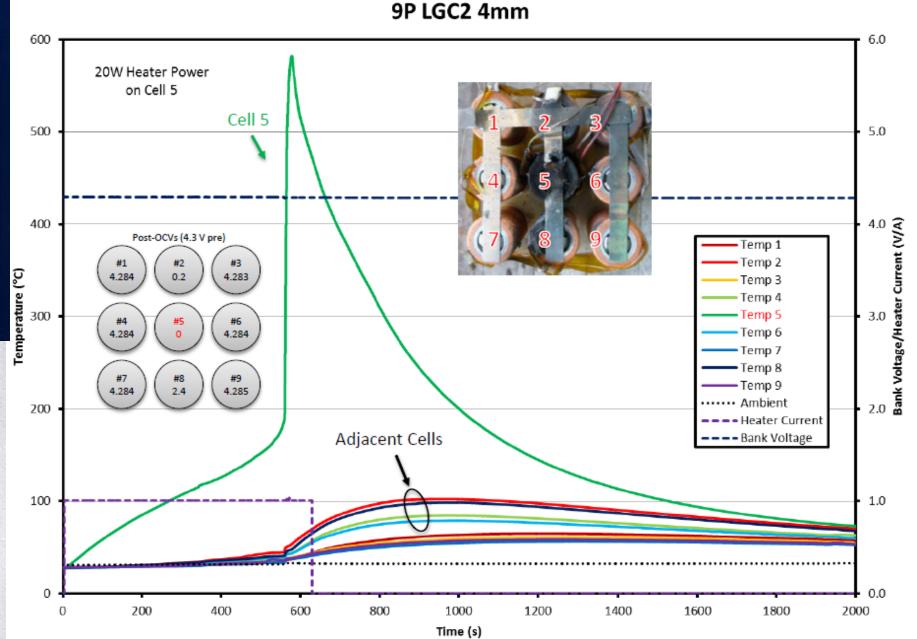


#### Thermal Isolation Example – 4mm air spacing between cells

#### Pre-Test



Jeevarajan et al. from 2014 Workshop showed that without any heat dissipation path except through electrical parallel connections, adjacent cells get damaged (shorted) with even 4 mm spacing



### VHS TR Test with Panasonic NCR18650B Cells



Side wall ruptures will even defeat very high flux heat rejection paths!

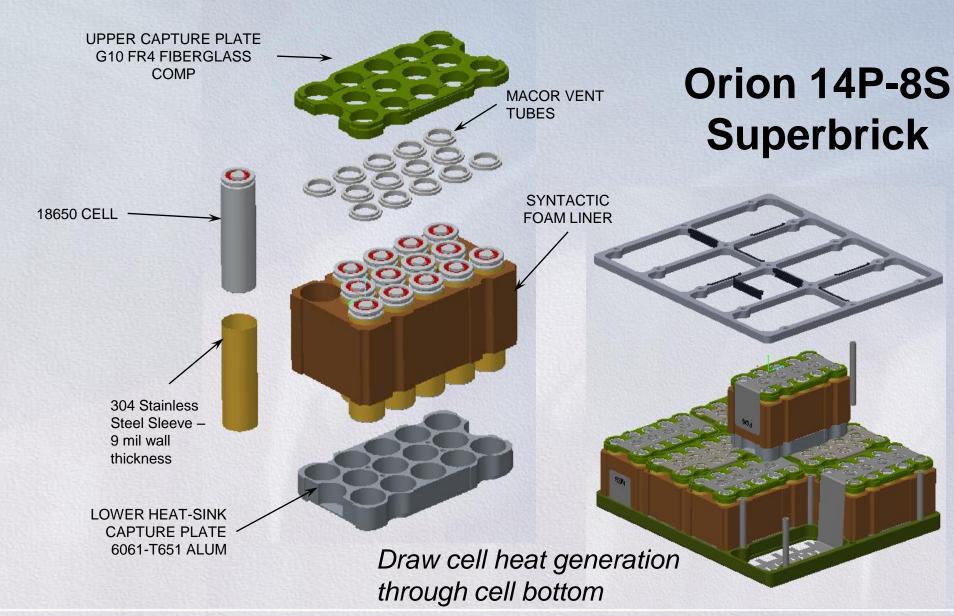
- Vaporizing Heat Sink (VHS) leaves 10mm of cell can wall bottoms exposed
- 2mm spacing between cells
- Trigger cell had side wall rupture in circumferential heater area which impinged TR ejecta into adjacent cell
- Resulted in propagation to two additional cells and damaged several others



#### **Orion Battery 14-cell Block**

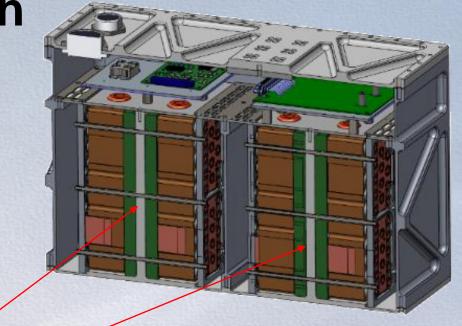


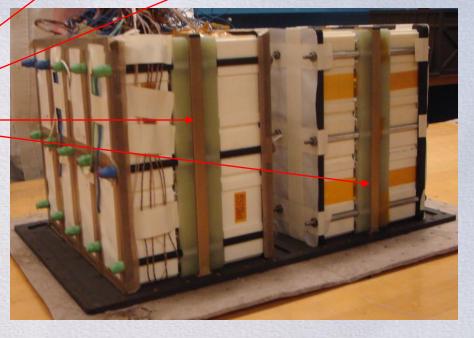




### Isolating vs Providing a heat path

- If you thermally isolate cells (air)
  - Adjacent cell ∆T rise 80-100°C
  - Limited to cell designs with little risk of side wall ruptures
  - Achieves 160-170 Wh/kg
- Orion Partially conductive (Draw heat from cell bottom)
  - Conduct heat to divider plate
  - Adjacent cell \( \Delta \T\) rise 60-70°C and shorter exposure
  - 14P-8S superbrick with SS sleeves achieves 150-160 Wh/kg





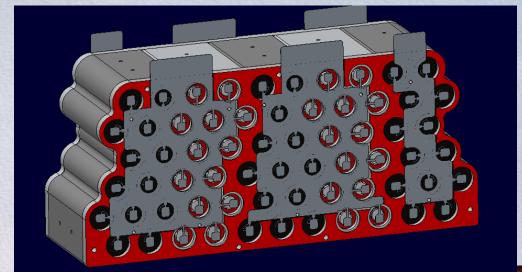
### Safer, Higher Performing Battery Design

#### Compliance with the 5 rules

- Minimize side wall ruptures
  - Al interstitial heat sink
- No direct cell-cell contact
  - 0.5mm cell spacing, mica paper sleeves on each cell
- Individually fusing cell in parallel
  - 12A fusible link
- Protecting adjacent cells from TR ejecta
  - Ceramic bushing lining cell vent opening in G10 capture plate
- Include flame arresting vent ports
  - Tortious path with flame arresting screens
  - Battery vent ports lined with steel screens

#### **Features**

- 65 High Specific Energy Cell Design 3.4Ah (13P-5S)
- 37Ah and 686 Wh at BOL (in 16-20.5V window)
- Cell design likely to side wall rupture, but supported











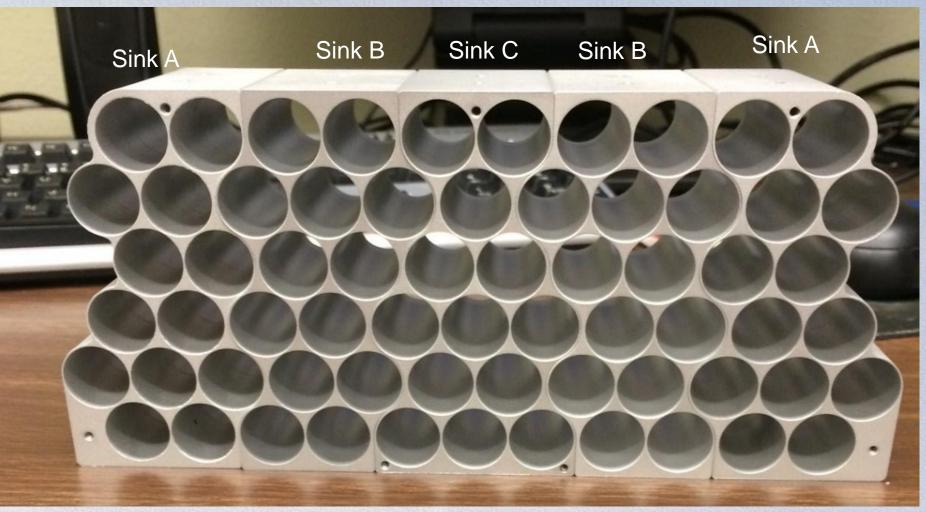




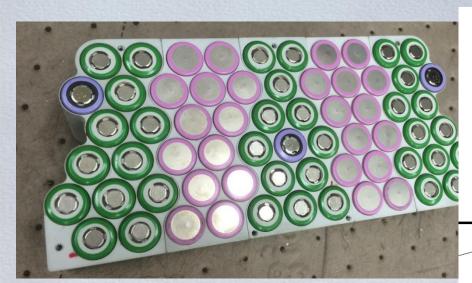
#### LLB2 Heat Sinks

No corner cells - Every cell has at least 3 adjacent cells





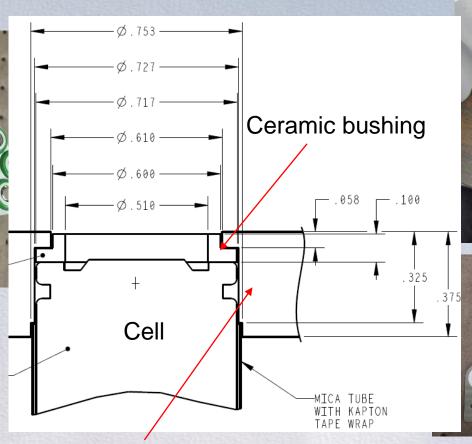
0.5mm cell spacing, Al 6061T6



 13P-5S Configuration with 3.4 Ah LG cell design yielding 37 Ah at 3.8 A mission rate.

Aluminum interstitial heat sink, 0.5 mm spacing between cells

- Mica sleeves around shrink wrap, 2 FT
- The G10 capture plate houses the + and - ends of the cells and prevents the Ni bussing from shorting to the heat sinks.
- The ceramic Macor bushing acts as a chimney to direct ejecta outwards and protect the G10/FR4 capture plate



G10/FR4

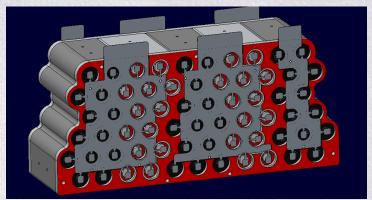






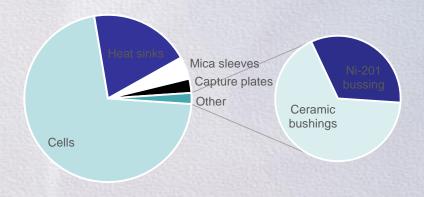
### Cell Brick Assembly > 180 Wh/kg

Mass Categories	g	%
3.4Ah 18650 Cells	3012.75	71.3%
Heat sinks	824.95	19.5%
Mica sleeves	182.31	4.3%
Capture plates	115.81	2.7%
Ceramic bushings	60.15	1.4%
Ni-201 bussing	29.71	0.7%
Total	4225.7	

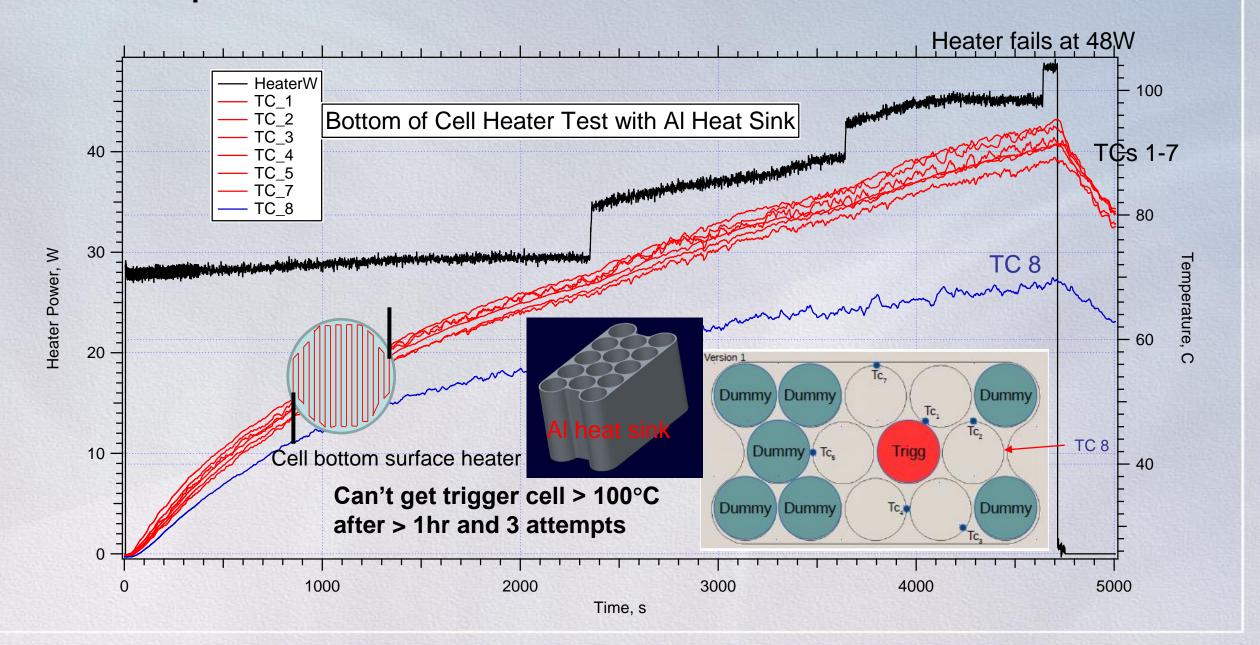


- With 12.41 Wh/cell, cell brick assembly achieves 191 Wh/kg
  - Assuming 12.41Wh per cell
- Design has 1.4 parasitic mass factor
  - Cell mass x 1.4 = Brick mass

#### Mass Distribution

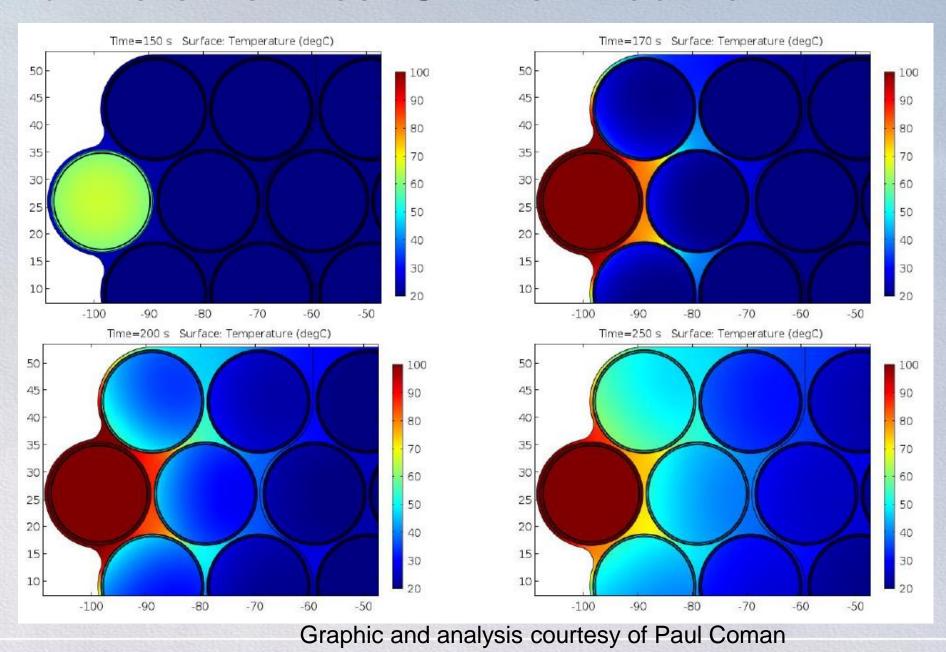


### Attempts to Drive TR with Cell Bottom Heater Fails

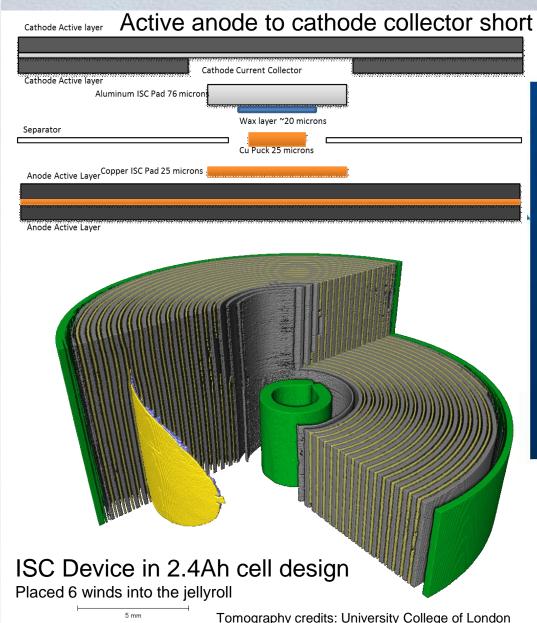


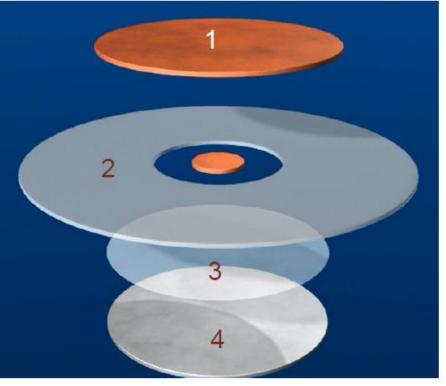
#### Metallic Interstitial Heat Sink is Effective

- Cell can isolated with mica paper sleeves and very small air gap
- Heat sink spreads heat more quickly through multiple layers than through mica and onto cells
- Heat from trigger cell is quickly dispersed and shared among more cells



### NREL/NASA ISC Device Design





Graphic credits: NREL

#### Top to Bottom:

- 1. Copper Pad
- 2. Battery Separator with Copper Puck
  - 3. Wax Phase Change Material
    - 4. Aluminum Pad

#### 2010 Inventors:

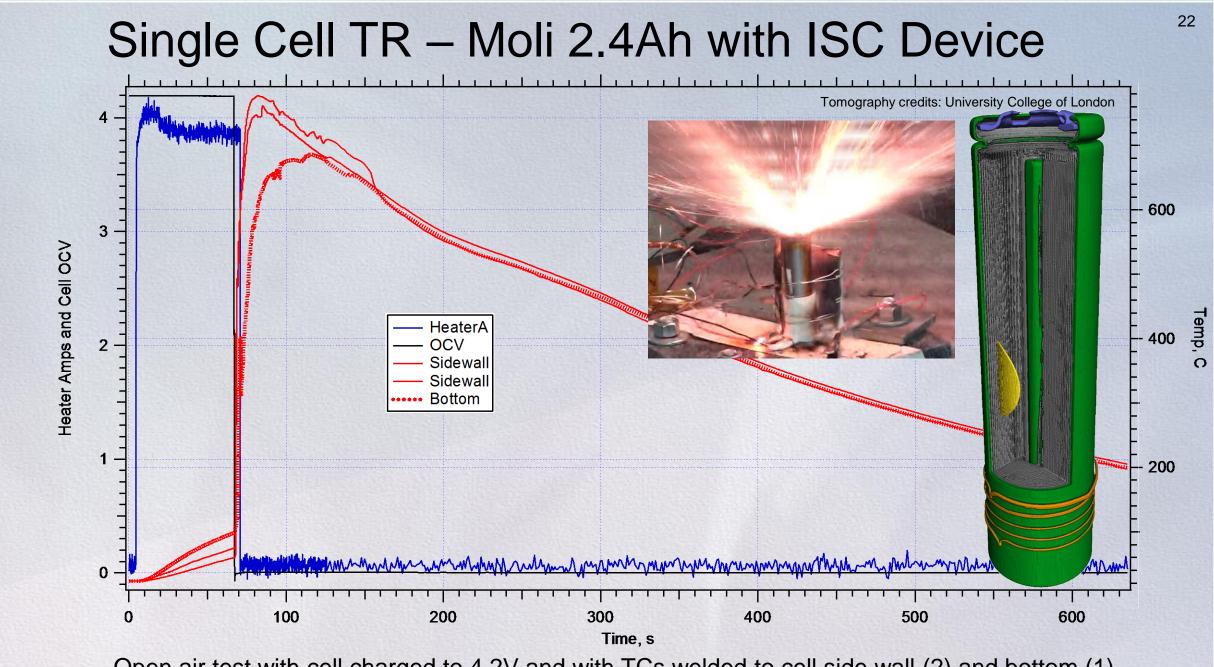
- Matthew Keyser, Dirk Long, and Ahmad Pesaran at NREL
- Eric Darcy at NASA

US Patent # 9,142,829 awarded in 2015

Wax formulation used melts ~57°C

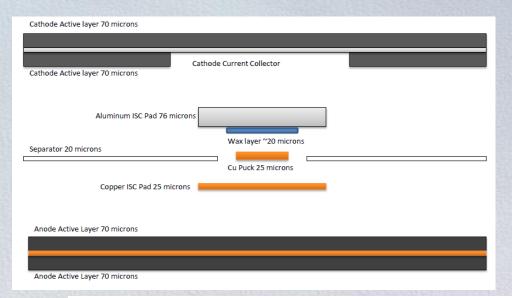
Thin (10-20 µm) wax layer is spin coated on Al foil pad

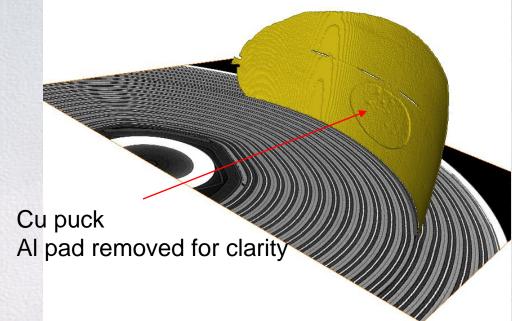


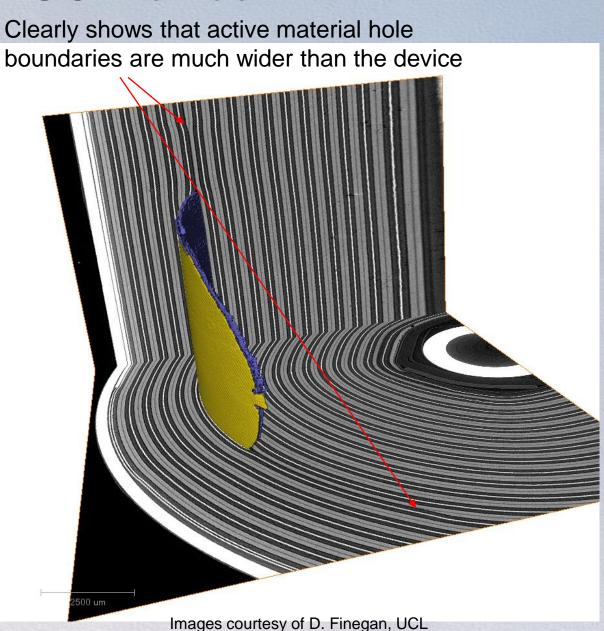


Open air test with cell charged to 4.2V and with TCs welded to cell side wall (2) and bottom (1)

### CT Images of ISC Device



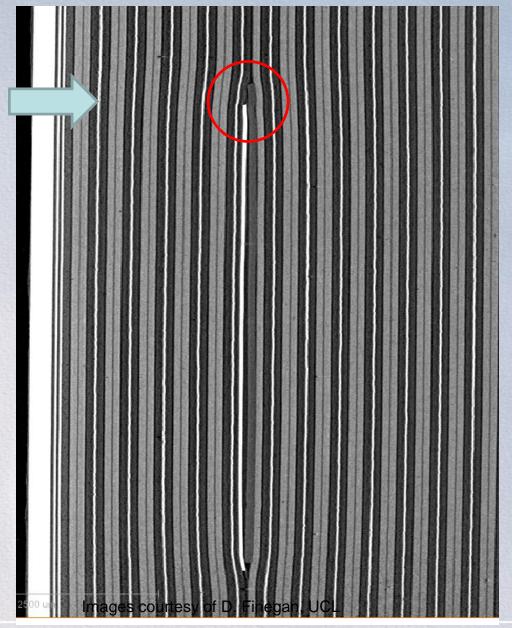




## CT images (cont.)

Misalignment of Cu and Al pads creates stress zones on the separator and could explain the damage initiation at the ISC device edge in some videos

Image picks up tweezer marks during fabrication on the Cu puck



No TR propagation, max adjacent T = 92 ℃

500

600

However, trigger cell was only 2.4Ah cell

400

50

Heater power ~42W for 180s. Onset of TR (OTR) occurs 180s after power on and coincides with trigger bank OCV dip. Adjacent cell1 has  $\Delta T = 58.9^{\circ}$ C to max of 92.0°C, while adjacent cells 2 & 3 have  $\Delta T = 48^{\circ}$ C to max of 76.0°C

Time, s

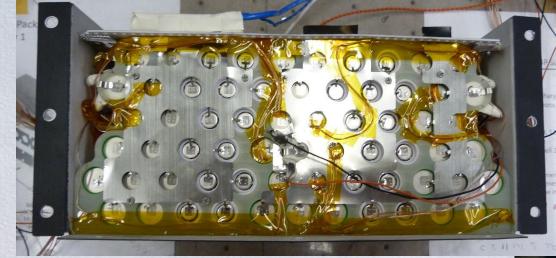
300

100

200

### No TR Propagation, Only Smoke Exits Battery





Mesh 40 & 30 steel screens arrest flames and sparks

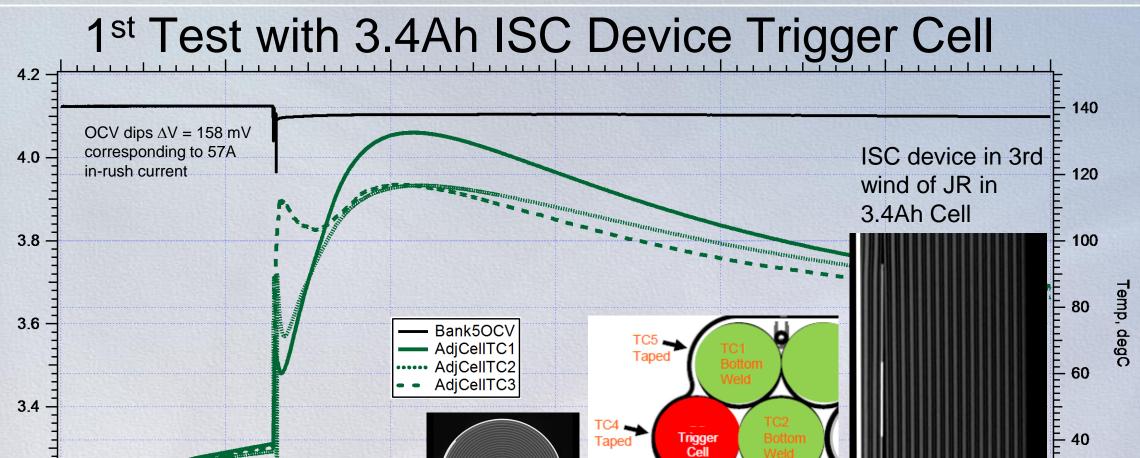


However, trigger cell was only 2.4Ah cell

20

600

500



Bank OCV

3.2

3.0

Adjacent cell temperatures TC1, TC2, and TC3 peak at  $133^{\circ}$ C,  $117^{\circ}$ C, and  $117^{\circ}$ C in 77-87s from onset temperatures of 39°C, 37°C, and 38°C for  $\Delta T = 94^{\circ}$ C, 77°C, and 78°C, respectively.

300

Time, s

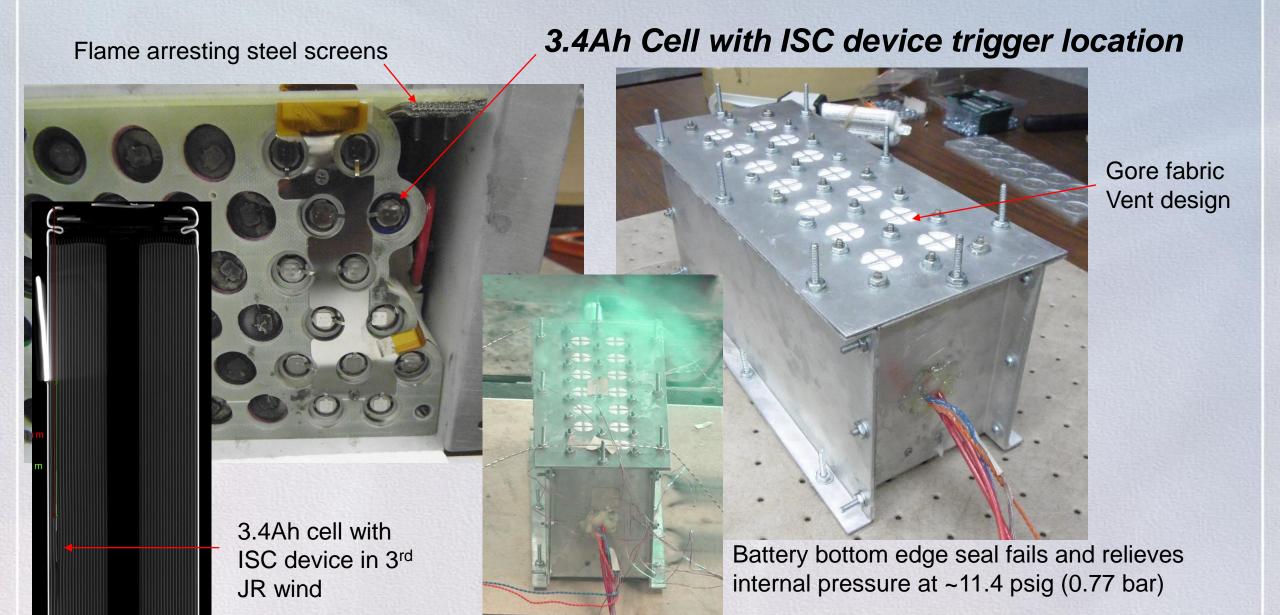
200

100

Taped

400

#### No TR Propagation - Only Clean Smoke Exits Gore Vent



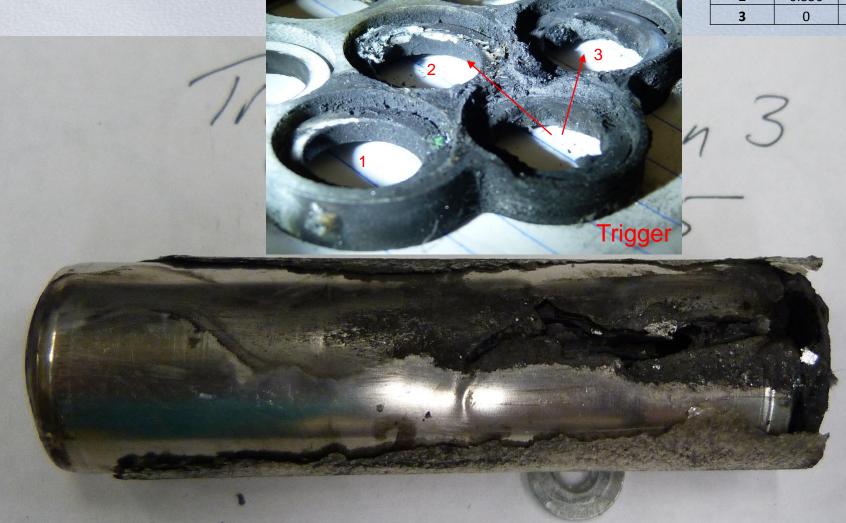
### 3.4 Ah Trigger Cell Experienced a Side Wall Rupture

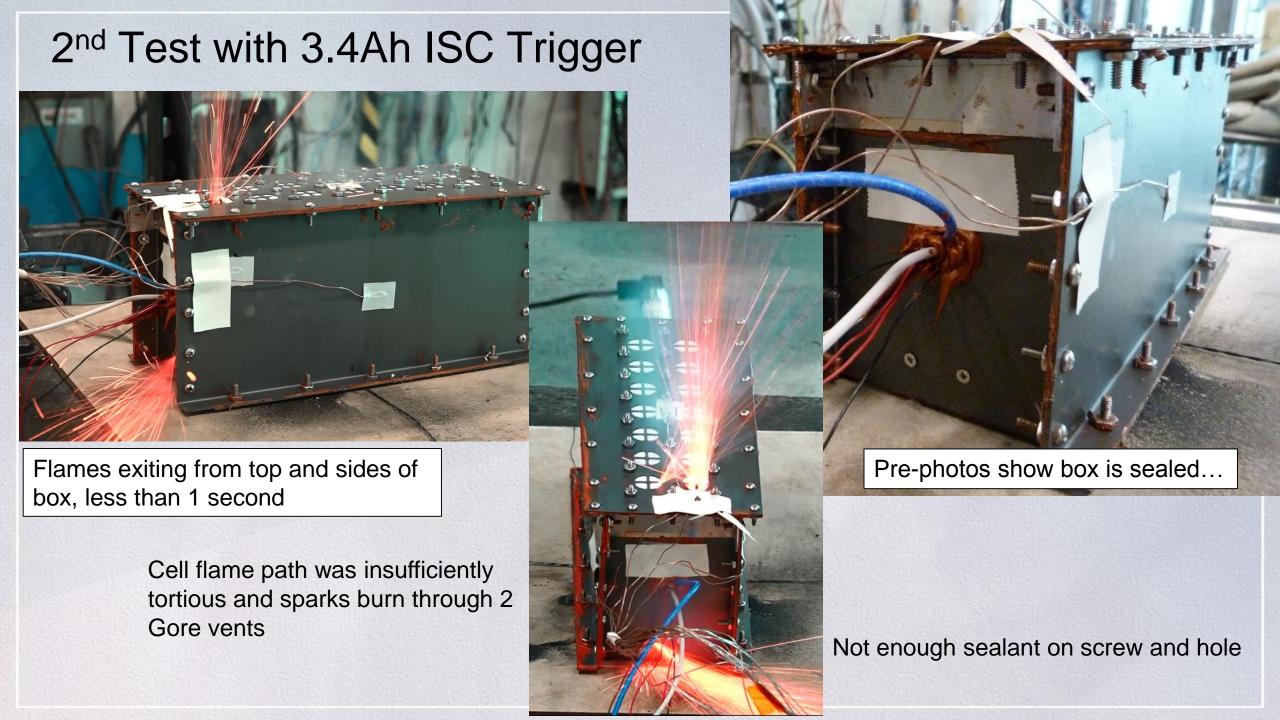
Trigger cell was a struggle to extract from heat sink.

The mica insulation was severely damaged adjacent to rupture

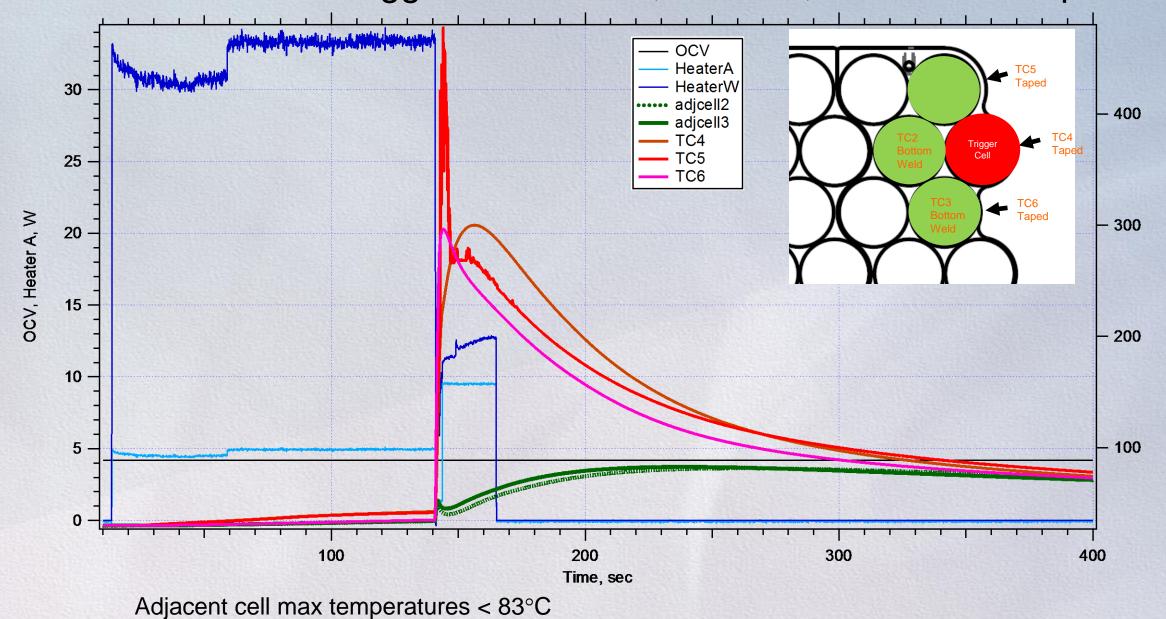
	Cell	OCV (V)	Mass (g)
	Trigger	0	17.161
	1	3.474	46.801
	2	0.336	46.691
1	3	0	46.671







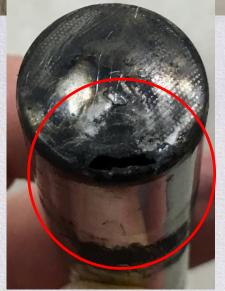
#### 2<sup>nd</sup> Test 3.4Ah ISC Trigger Cell – OCV, Heaters, & Interior Temps





Post-Test Mass: 25.3g











Bottom breach

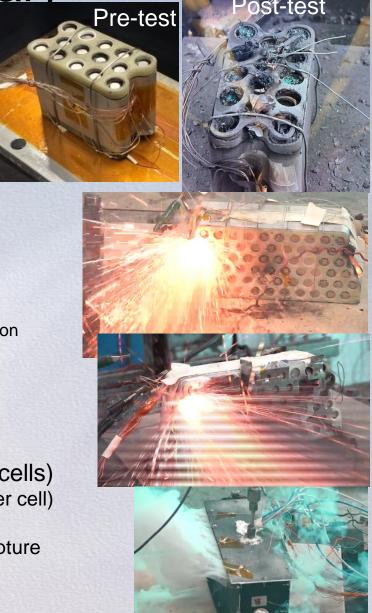
Spin groove is stretched

### Findings from 2<sup>nd</sup> Test with 3.4Ah ISC Trigger Cell

- ISC device in 3.4Ah 18650 cell triggered in 127 seconds with bottom heater at 32W average
  - Very similar initiation time (1st run was in 119s)
  - Very similar biasing of adjacent cells (34-35°C) at onset of TR (1st run at 37-39°C)
- No propagation of TR
  - Despite bottom rupture of trigger cell, which damaged the G10/FR4 negative capture plate
  - Reusing the same heat sinks from the first test undamaged after both tests
- Max adjacent cell temperatures < 83°C</li>
  - Adjacent cell temperature rise was 46-47°C, significantly lower than 1<sup>st</sup> run (77-94°C)
  - Bottom rupture yields a much less severe impact than side wall rupture

Spacesuit Prototype Battery Test Summary

- Al Heat Sink Tests
  - 4 attempts to drive > 250Wh/kg cell into TR All failures
    - 2 with Panasonics, 2 with LGs, all with home made bottom heaters
  - 5 attempts with 2.4Ah ISC device cells No propagation of TR
    - · 1 dud and 4 success with the 2.4Ah ISC cell driven into TR
  - 2 heat to vent tests with 5 fully charged 3.4Ah cells each
    - · No side wall ruptures in areas supported by the sink
- LLB2 brick tests (All six 2.4Ah ISC cells successfully driven to TR)
  - 3 no-Ni bussing brick tests
    - No TR propagation and no OCV changes to adjacent cells with excellent temp margins
      - Interior cell trigger ΔT ~ 19°C (one run)
      - Edge cell trigger ΔT ~ 42°C (two runs)
    - Interior cell trigger are less vulnerable than edge cells based on temperature rise (max-onset T) on adjacent cells
  - 3 Ni bussing (13P5S)
    - No propagation of TR, no impact on adjacent cell OCVs
    - Very good temperature margins (vs onset of TR temperature)
      - Interior cell trigger: ΔT ~ 30°C (one run)
      - Edge cell trigger ΔT ~ 48°C (one valid run)
- LLB2 full scale tests (4 runs 2 w/ 2.4Ah, 2 with 3.4Ah ISC device implanted cells)
  - No propagation of TR (even with side wall rupture of trigger cell in 1st test w/ 3.4Ah trigger cell)
  - Maximum adjacent cell temperature rise with 2.4Ah trigger cell was 55-58°C
  - Maximum adjacent cell temperature rise with 3.4Ah trigger cell was 94°C w/ side wall rupture and 46°C with bottom rupture
  - Gore vent design needs more flame arresting protection to handle 3.4Ah cell TR output
  - Screened vents were demonstrated as a successful flame arresting solution



### ISC Device Location Reveals Side Wall Rupture Risk

- 3.4Ah cell can thickness
  - 165 microns
  - No bottom vent
- Unsupported oven heating test
  - No side wall ruptures (30 cells)
  - Slow external heating to TR
- Unsupported circumferential heater test
  - No side wall ruptures (5 cells) at ~30W
  - 1 of 3 side wall rupture at ~60W
- With ISC device (11 tested so far)
  - 8 sidewall ruptures
    - 5 unsupported
    - 3 supported by Al interstitial heat sink
  - 1 bottom rupture
    - Supported by Al interstitial heat sink
  - 2 vented through header
    - Supported by Fe tubes



ISC device in 3rd wind



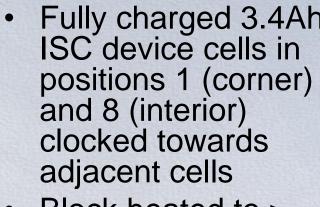
Circumferential heater near bottom of can wall

#### How Effective Are Steel Tubes?

Fully charged 3.4Ah ISC device cells in positions 1 (corner) and 8 (interior)

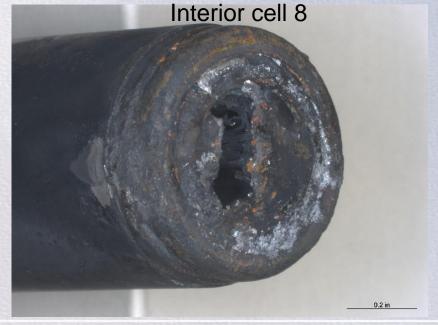
Block heated to > 60°C to activate ISC devices

- Corner cell wrapped with 0.015" (381 μm) SS tube experienced side wall rupture outside of tube
  - Dissection of tube found no cell can side wall ruptures inside tube area
- Interior cell wrapped with 0.009" (229 μm)
  - No side wall ruptures outside or inside tube





Corner cell 1





Orion 14-cell assembly with cell,

tubes, foam

### **Summary Findings**

- ISC device enables critical battery safety verification
  - With the aluminum interstitial heat sink between the cells, normal trigger cells can't be driven into TR without excessive temperature bias of adjacent cells
  - With an implantable, on-demand ISC device, TR tests show that the conductive heat sinks very effectively protected adjacent cells from propagation
    - Even with >700 Wh/L cell design experiencing side wall or bottom rupture (4 test runs)
  - 3.4Ah 18650 cell design shown susceptible to side and bottom rupture with ISC device
    - Note that no side wall ruptures occurred during slow heat to TR testing (unsupported, 30 cells tested)
- High heat dissipation and structural support of Al heat sinks show high promise for safer, higher performing batteries
  - Battery brick design achieving > 190Wh/kg demonstrated to be safe
- Preliminary results on bottom vents are inconclusive
  - TR testing with ISC device is needed

#### Future work

- Will examine impact of the location of the ISC device in the JR
- Will examine merits of cell designs with bottom burst disk vent feature to reduce side wall rupture risk
  - Is it a better solution than thicker can and/or lower header burst pressure?

#### Acknowledgements

- M. Keyser, National Renewable Energy Labs, for making the ISC devices
- M. Shoesmith, E-one Moli Energy, for successfully implanting the ISC device in their 2.4Ah cell design
- D. Finegan, University College of London, for tomography and high speed X-ray videos
- · P. Coman, University of South Denmark, for battery design guidance through thermal analysis

