

# Plans & Preliminary Results of Fundamental Studies of Ice Crystal Icing Physics in the NASA PSL

June 15, 2016

**Peter Struk** 

NASA Glenn Research Center

**Jen-Ching Tsao, Tadas Bartkus** 

Ohio Aerospace Institute



## **Outline**

- Introduction & background
- NASA Fundamental Ice Crystal Icing Research Goals
  - Concepts using PSL
- Experimental Description Preliminary 2-day test May 2015
- Results
  - Review one case in detail
  - Look at general trends from sweeping
    - Total Water Content
    - Humidity
    - Spray Bar Water Temperature
- Conclusions



### Introduction

- NASA investigating the fundamental physics of ice crystal icing (ICI)
  - AEST → AATT
- Challenging to study ice-accretion physics directly inside the engine
  - Trying to simulating that environment without using engine
- Evaluating whether the NASA
   Propulsion Systems Lab (PSL), in addition to full-engine and motor-driven-rig tests, can be used for more fundamental ice-accretion studies
  - Paper presents concept & some preliminary experimental test results
  - Subsequent paper present complementary modelling work

Atmospheric Environments Safety
Technologies Project (AEST; 2009–2014)



Advance Air Transport Technology
Project (AATT; 2015 +)
Advanced Aircraft Icing (AAI) Subproject

#### **Technical Challenge:**

Expand engine aero-thermodynamic modeling capability to predictively assess the onset of icing in current and N+2/N+3 aircraft during flight operation (FY21).

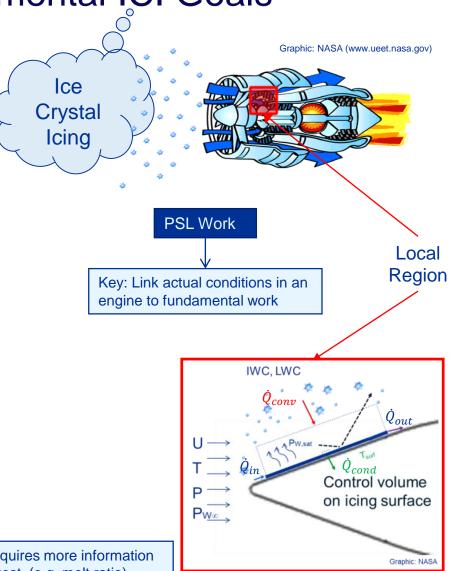
The simulation tools are well anchored in results from both fundamental physics studies and full engine tests.



## NASA Fundamental ICI Goals

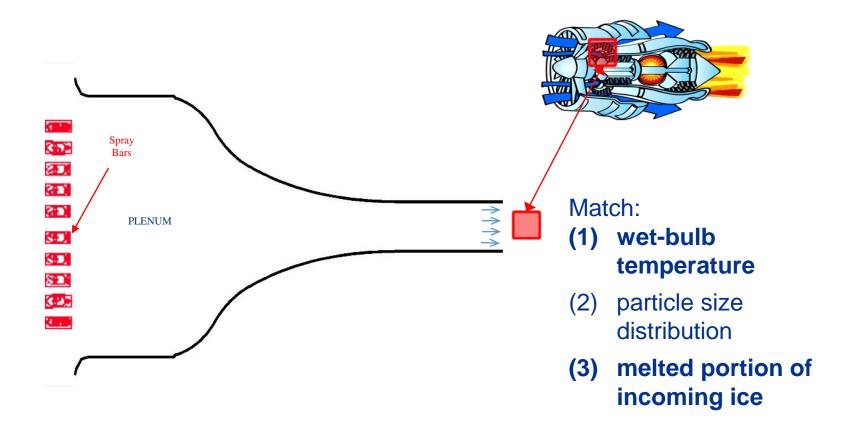
- 1. Identify and bound the conditions affecting ice-crystal ice accretion at the (local) accretion site
- 2. Generate & characterize (i.e. measure) those conditions including uniformity
- Gather data and develop models on ice-crystal icing factors
- 4. Scaling: develop & test scaling relations for ice-crystal icing

Local region requires more information than full-scale test (e.g. melt ratio)





## Concept Using PSL



Goal: Ability to generate a prescribed mixed-phase condition at the test section for fundamental ice-crystal icing research

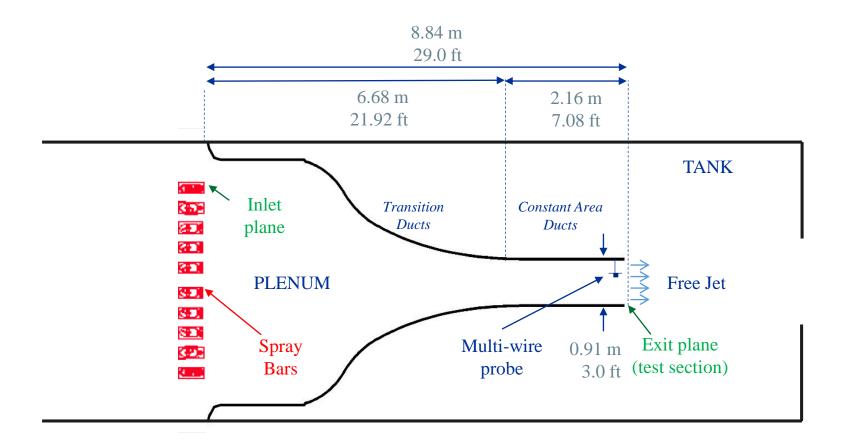


# **Preliminary Testing**

- 2 days of testing occurred in May 2015
- Objectives
  - Preparation for more extensive test scheduled for 2016
  - Examine spray bar and plenum parameters and how they affect the mixedphase at the exit of the free jet
  - Cloud characterization:
    - Melt ratio using SEA multiwire
    - Temperature & humidity measurements at test section (cloud on vs. cloud off) using custom probe
  - Observe ice accretion

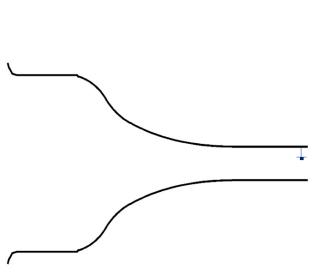


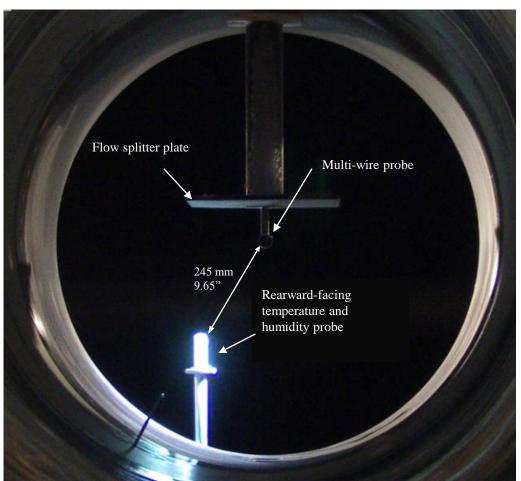
## **PSL** Configuration





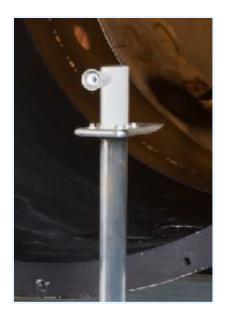
# PSL Configuration (cont.)







## Temperature and Humidity Measurement



Side View



Top View



Tunnel Flow

- Reward facing probe
  - Temperature
    - Resistance Temperature Device (RTD)
       placed inside probe inlet to prevent
       water / ice contamination
    - Small suction induced in probe
    - Calibrated to read total temperature given Mach
  - Humidity
    - Flow extracted via same probe inlet
    - Using Spectra Sensor Model WVSS-II
      - Tunable Diode Laser Absorption Spectroscopy (TDLAS)
  - Cameras imaged probe to observe any ice accretion

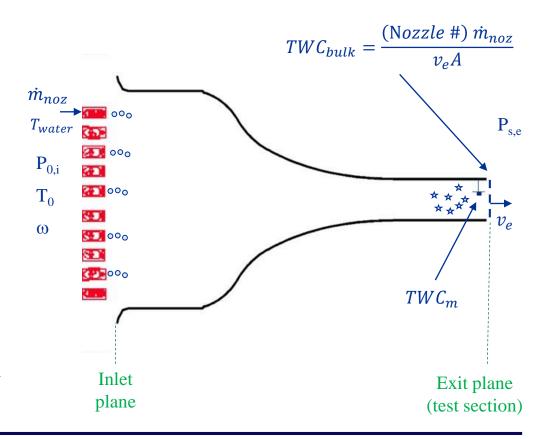


# Mixed-Phase Investigation

#### **Parameters**

- Plenum / test section
  - Total pressure,  $P_{0.i}$  (kPa)
  - Static pressure, P<sub>s.e</sub> (kPa)
    - Velocity,  $v_e$  (m/s)
  - Total temperature,  $T_{0,i}$  (°C)
  - Humidity,  $\omega_i$  (g / kg dry)
- Spray bar
  - TWC
    - $P_{air} \& P_{water} \rightarrow \dot{m}_{noz}$
    - Nozzle #
  - Particle Size
    - MVDi (IRT calibrated values)
  - Water / air temperatures, T<sub>water</sub>

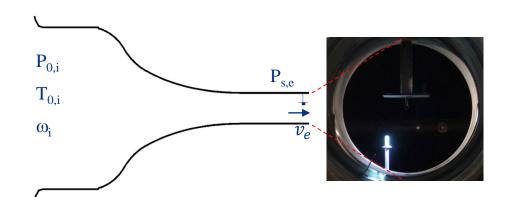
#### **Nomenclature**

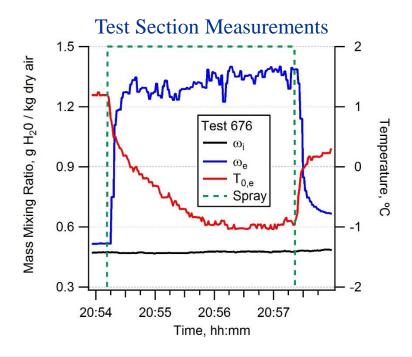




# Sample Test Data

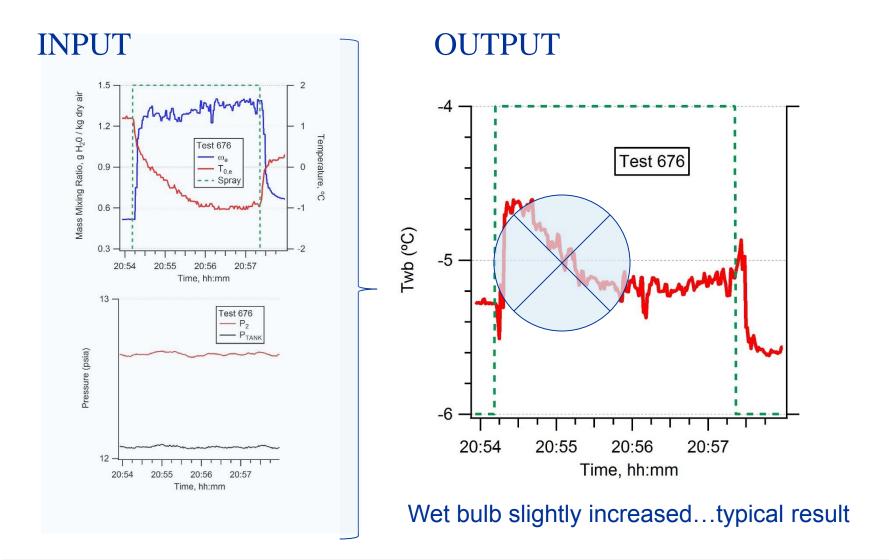
- Plenum / Test Section (targets)
  - $P_{0i} = 87.3 \text{ kPa}$
  - $P_{s,e} = 83.6 \text{ kPa } (1.6 \text{ km})$
  - $-v_e = 85 \text{ m/s}$
  - $-T_{0.i} = 1.8^{\circ}C$
  - $-\omega_{i} = 0.5 \text{ g/kg dry (RH_{Pl} = 10\%)}$
- Spray bar
  - $TWC_{bulk} = 1.4 \text{ g/m}^3$
  - $MVDi = 40 \mu m$
- Reported data
  - Temperature measurement lag likely due to thermal inertia of inlet
  - 30 second averages
    - Cloud off (0.52 g/kg, 1.2 °C)
    - Cloud on (1.37 g/kg, -0.9 °C)
  - $-\Delta T_{0,e} = T_{0,e,on} T_{0,e,off}$







## Wet-bulb temperature





## Multiwire Results



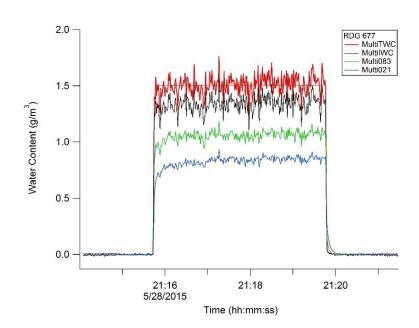
$$-$$
 MVDi = 40  $\mu$ m

#### Multiwire data

- 30 second averages
- $TWC_{m} = 1.50 \text{ g/m}^{3}$
- LWC<sub>m,2.1</sub> = 1.06 g/m<sup>3</sup>
- LWC<sub>m.0.5</sub> = 0.83 g/m<sup>3</sup>
- Melt ratio,  $\eta_e$

$$\eta_e = \frac{\max(LWCm)}{TWC_m} = \frac{1.06}{1.50} = 0.70^*$$

\* more detailed analysis anticipated to be applied later





## Ice Accretion Examples

Case 677 ( $\eta_e = 0.70$ )



Case 663 ( $\eta_e = 0.20$ )

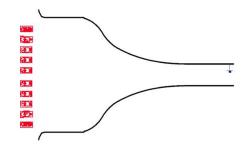


8X actual speed



## Parameter Sweeps

- Paper presents parameter sweeps for the following variables:
  - $TWC_{bulk} (0.5 5 \text{ g/m}^3)$
  - Plenum RH (10 50%)
  - Spray bar temperature (7°C, 43°C, and 82°C)
- Within each sweep, additional variations on:
  - **MVDi**
  - Wet-bulb temperature



→ Table 2. Facility target conditions and select measurements during two TW				
Test Series ->	TWC Sweep 3			
Facility Target Condition				
P <sub>0,i</sub> (kPa)	87.3			
P <sub>s,e</sub> (kPa)	83.6			
ve (m/s)	85			
Altitude (km)	1.6			
T <sub>0,i</sub> (°C)	4.2			
T <sub>s,e,off</sub> (°C)	0.6			
RH <sub>0,i</sub> (%)	10			
$\omega_i (g/kg)$	0.6			
Twb <sub>0,e,off</sub> (°C)	-3			
$Twb_{s,e,off}(^{\circ}C)$	-6			
T <sub>water,i</sub> (°C)	7			
$TWC_{bulk}(g/m^3)$	0.78	1.4	2.3	5.0
MVD <sub>i</sub> (μm)	40			
	Measurements			
Test #	670	671	672	673
$TWC_m (g/m^3)$	1.4	2.9	4.3	10.0
η e (-)	0.69	0.66	0.23	0.27
$\Delta\omega_{\rm e}({\rm g/kg})$	0.55	0.87	1.3	2.3
$\Delta T_{0,e}$ (°C)	-1.7	-2.3	-2.9	-3.7
Twb <sub>0,e,on</sub> (°C)	-3	-4	-3	-2
Twb <sub>s,e,on</sub> (°C)	-6	-6	-5	-4
Ice Accr. (Y/N)	Y	Y	Y	Y



## Conclusions

- NASA conducting research on fundamentals of ICI with following goals:
  - Identify and bound the conditions at the (local) accretion site
  - Generate & characterize conditions
  - Develop models & gather data on ice-crystal icing factors
  - Scaling: develop & test scaling relations for ice-crystal icing
- Generate environment outside of an engine to facilitate study
  - Evaluating PSL as potential test bed
- Presented data from an initial 2-day test effort in May 2015
  - Parameter sweeps on TWC, Plenum RH, and T<sub>water</sub>
    - More limited variation on initial particle size and Twb
  - Saw both expected trends; harder-to-explain trends; new insights
    - Measurement uncertainties, cloud uniformity, and additional data required
  - Preparatory work for 2016 testing
- 2-week test campaign occurred in March 2016
  - Data analysis beginning



# Acknowledgement

- Financial support:
  - NASA's, Advanced Air Vehicle's program
    - Advance Air Transport Technology Project (AATT)
      - Advanced Aircraft Icing (AAI) Mr. Tony Nerone, Project Manager
- Special thanks to:
  - Staff of the NASA PSL
  - Kyle Zimmerle, Michael Oliver, and Judy Van Zante who provided information presented in this paper.
  - Mr. Chris Lynch for his excellent imaging work.