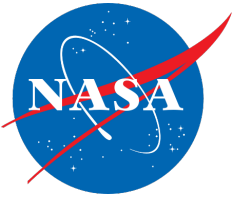


Lightweight and High-resolution Astronomical X-ray Optics Using Single Crystal Silicon

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NASA Goddard Space Flight Center



Next Generation X-ray Optics Team



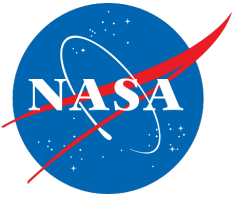
K. D. Allgood¹, M.P. Biskach¹, K.W. Chan², J.D. Kearney¹,
J.R. Mazzearella¹, R.S. McClelland¹, A. Numata¹, L.G. Olsen,
R.E. Riveros², T.T. Saha, M.J. Schofield¹, M.V. Sharpe¹,
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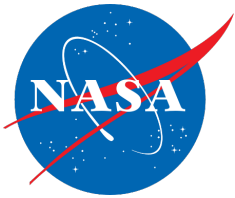
J.M. Carter, J.A. Gaskin, W.D. Jones, J.J. Kolodziejczak, S.L. O'Dell
NASA Marshall Space Flight Center



Challenges in Context



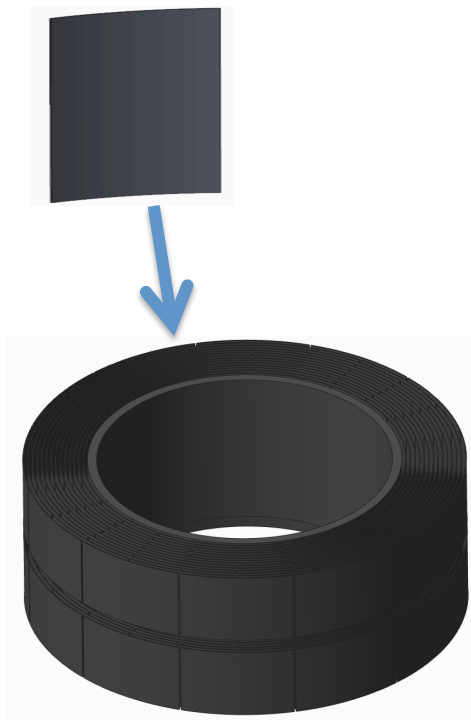
- **PSF**
 - Similar to or better than Chandra's: **~0.5"** HPD
- **Mass**
 - Similar to Chandra's: **~1,500** kg
- **Photon collecting area**
 - At least **10X** Chandra's
- **Cost**
 - Comparable to Chandra's in RY\$ or
 - Less than **0.5X** Chandra's in constant \$



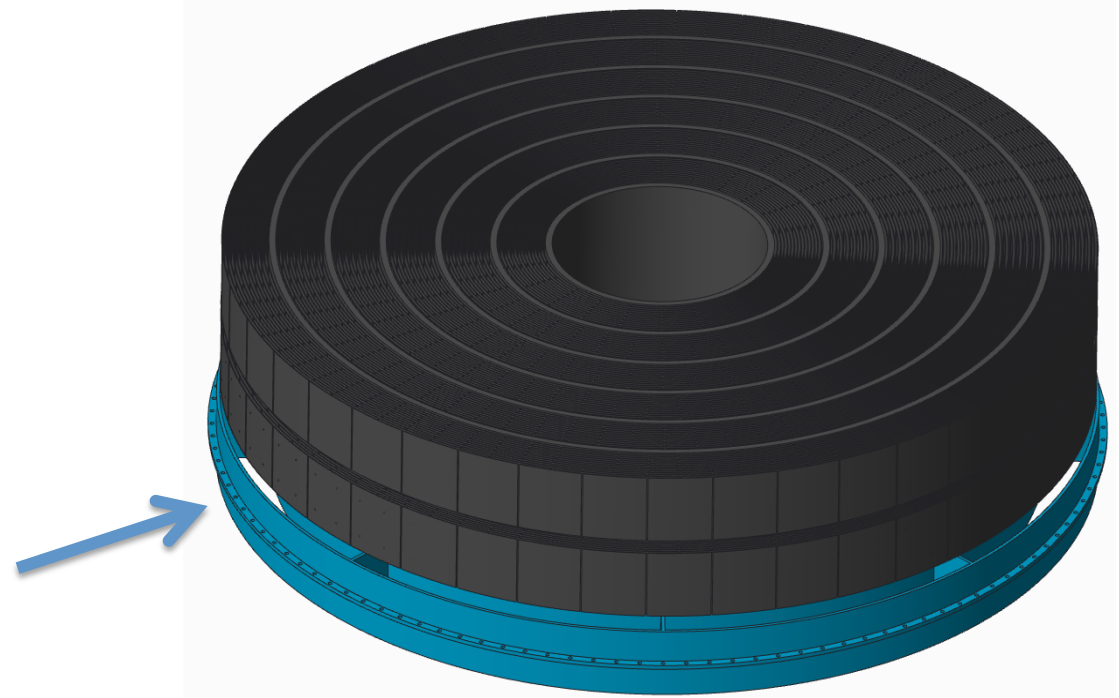
Approach: Mirrors → Meta-Shells → Assembly



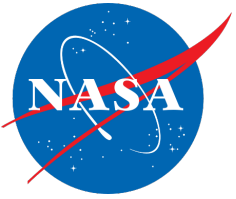
Single Crystal Silicon Mirrors



Meta-shell: mirrors bonded onto a silicon structural shell using silicon spacers and epoxy



Assembly: Many meta-shells aligned and flexure-mounted onto an aluminum or composite spider web

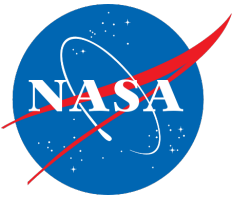


Notional Design of an X-ray Surveyor Mirror Assembly



- Focal length: **10 m**
- Outer diameter: **2 m**
- Inner diameter: **0.3 m**
- No. of mirror layers/shells: **~400**
- Physical mirror surface area: **~250 m²**
(cf. Chandra's 19 m²)
- Effective area at 1 keV: **~1.2 m²**
(cf. Chandra's 0.08 m²)
- Diffraction limit at 1 keV: **0.36"** (90% Power Diameter)

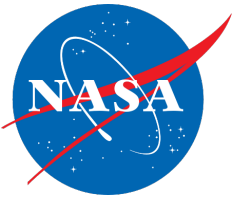
→ **25,000 mirrors, each 100 x 100 x 0.5 mm³**



Magnitude of the Challenge



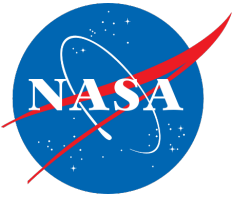
- **Fabricate ~25,000 mirrors**
 - Typical mirror: 100mm by 100mm by 0.5mm
 - **Technical, cost, and schedule** challenges
- **Align and bond these mirrors onto ~20 meta-shells**
 - **Technical and schedule** challenges
- **Integrate ~20 meta-shells into an assembly**
 - **No challenge**. Substantially similar to XMM-Newton's and Chandra's mirror integration.



Mirror Fabrication



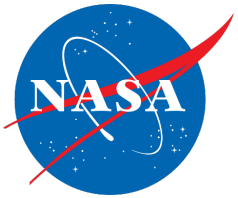
- So far the **ONLY** way to fabricate mirrors that meet requirements is precision polishing.
- Polishing has two problems
 - It has only made thick mirrors
 - Typical aspect ratio (size/thickness): **~6 to 10**
 - X-ray Surveyor requirement (size/thickness): **~200**
 - It is slow and expensive
- We are developing two solutions
 - Use single crystal silicon
 - Adopt mass production techniques



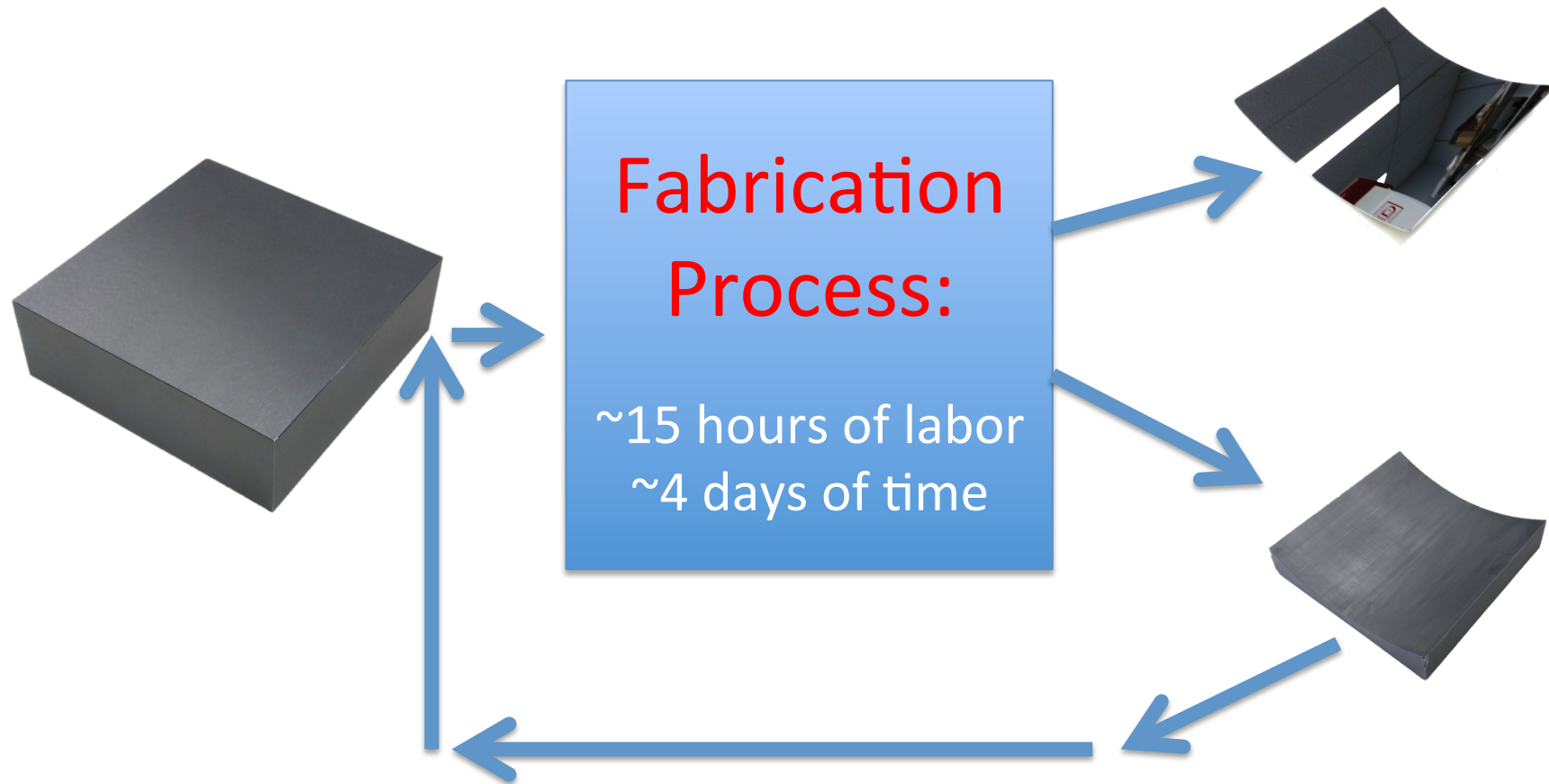
Why Single Crystal Silicon?

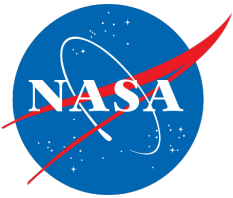


- **It has no internal stress**
 - Damage-free removal of material from the surface does not lead to unpredictable figure change, in contrast to thin sheet of glass.
- **It has excellent properties**
 - Low density
 - High thermal conductivity
 - Low thermal expansion
 - High elastic modulus
- **It is commercially and inexpensively available**



Mirror Fabrication





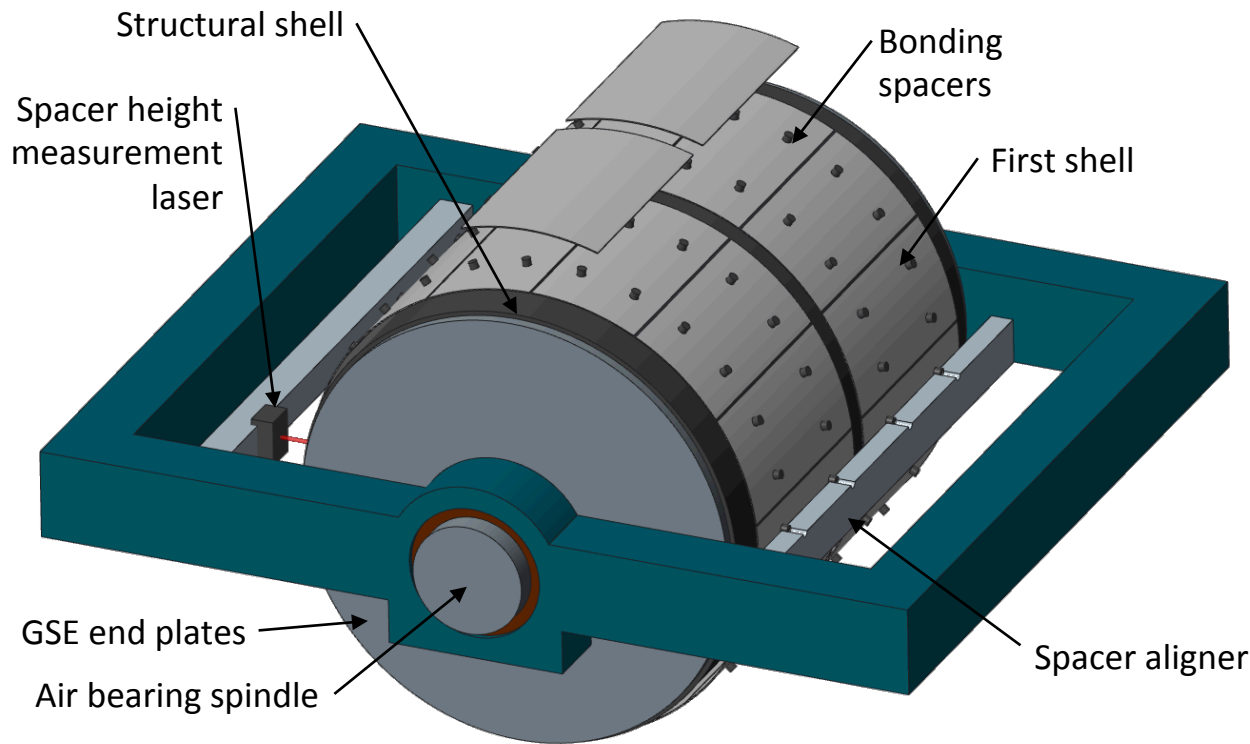
Mirror Fabrication Process



- **Generation:** setting radius and cone angle
- **Light-weighting:** removing the extra pounds
- **Acid etch:** removing damage and stress
- **Stress-polishing:** making precise optics
- **Trimming:** making it fit
- **Edge treatment:** preventing breakage
- **Metrology:** verifying figure quality

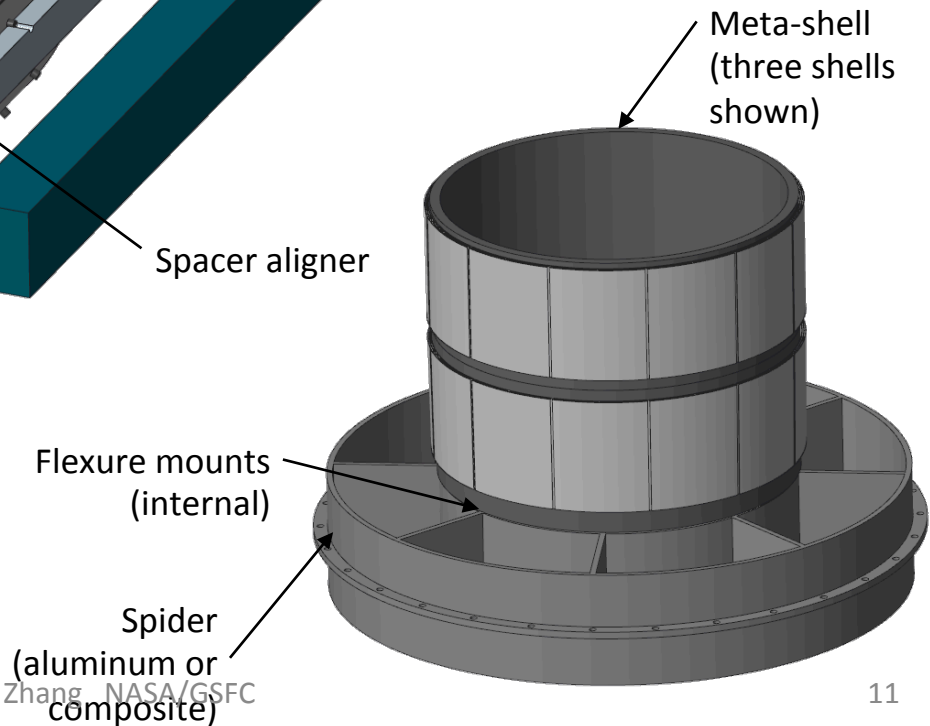


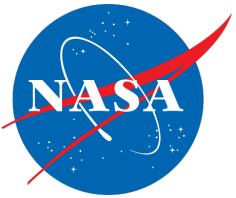
Building a Meta-shell



Meta-shell Makeup:
Silicon mirrors, Silicon spacers, and trace amounts of iridium and epoxy

- **Spindle defines the optical axis.**
- **Each mirror is bonded at 4 locations.**

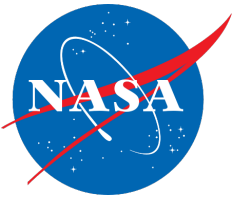




System Level Issues



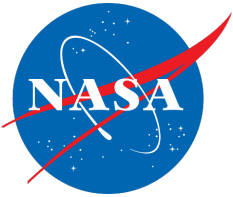
- **Structural**
 - A meta-shell of many mirrors bonded with Hysol 9309 epoxy can withstand generic launch load.
- **Thermal**
 - A meta-shell can achieve better than 1" PSF performance under typical on-orbit thermal conditions.
- **Gravity release**
 - A meta-shell constructed with its optical axis in the horizontal direction can achieve better than 1" PSF once gravity is released.



Areas of Development



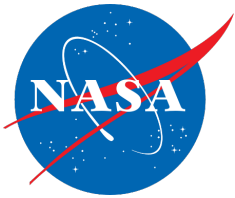
- **Mirror Fabrication**
 - Figure quality improvement (currently at ~3" HPD)
 - Fabrication time reduction
- **Coating**
 - Atomic layer deposition or magnetron sputter
 - Reduction/elimination of figure distortion
- **Alignment and Bonding**
 - Precision machining of spacers
 - Fast application and cure of epoxy
- **System level studies**
 - Complete end-to-end structural, thermal, and optical performance (**STOP**) analysis
 - Construction and test of meta-shells: **performance** and **environmental**



Prospects



- **2016-2017**
 - Demonstrate **possibility** of making sub-arcsec lightweight single crystal silicon mirrors
 - Build **mirror stacks** that can produce X-ray images close to 1" HPD
- **2018-2019**
 - Demonstrate **mass-production** process for making sub-arcsec mirrors; Team up with industry for implementation
 - Build and test **meta-shells** that meet X-ray Surveyor requirements: performance and environmental (TRL-6); Team up with industry for systems engineering



Acknowledgement

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NASA through
APRA and SAT Programs.**

