

National Aeronautics and
Space Administration



Application of High Speed Digital Image Correlation in Rocket Engine Hot Fire Testing

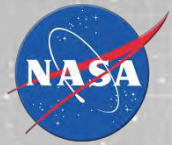
November 7-10, 2016
iDICs 2016 Conference
Philadelphia, PA

Paul Gradl, NASA
Tim Schmidt, Trilion

NASA MSFC
256.544.2455
Paul.R.Gradl@nasa.gov



MARSHALL
SPACE FLIGHT CENTER



Abstract

Application of High Speed Digital Image Correlation in Rocket Engine Hot Fire Testing

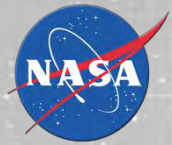
Paul R. Gradl

NASA Marshall Space Flight Center

Tim Schmidt

Trillion Quality Systems LLC

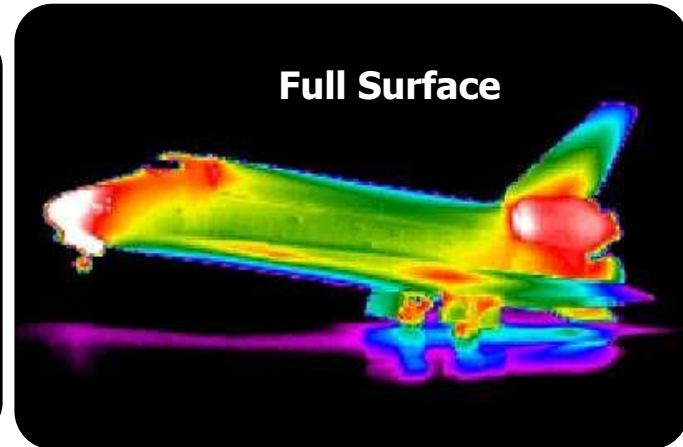
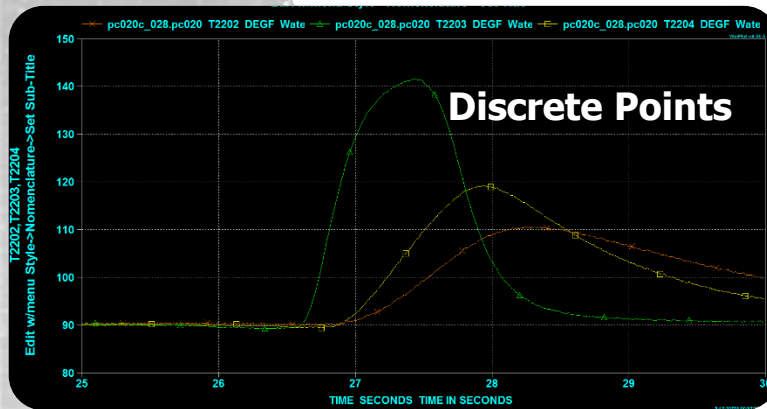
Hot fire testing of rocket engine components and rocket engine systems is a critical aspect of the development process to understand performance, reliability and system interactions. Ground testing provides the opportunity for highly instrumented development testing to validate analytical model predictions and determine necessary design changes and process improvements. To properly obtain discrete measurements for model validation, instrumentation must survive in the highly dynamic and extreme temperature application of hot fire testing. Digital Image Correlation has been investigated and being evaluated as a technique to augment traditional instrumentation during component and engine testing providing further data for additional performance improvements and cost savings. The feasibility of digital image correlation techniques were demonstrated in subscale and full scale hotfire testing. This incorporated a pair of high speed cameras to measure three-dimensional, real-time displacements and strains installed and operated under the extreme environments present on the test stand. The development process, setup and calibrations, data collection, hotfire test data collection and post-test analysis and results are presented in this paper.



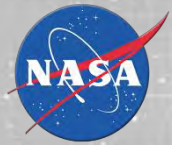
Motivation for Technology

- Subscale and Full-scale testing requires expensive and labor intensive instrumentation to better understand hardware performance
 - Design Modifications and Performance Predictions based on “discrete” point instrumentation
 - Thermocouples, Pressure Transducers, Accelerometers, Strain Gages
- **Challenge: Measure highly dynamic elevated temperature components**

Full Surface > Point
IR > Thermocouple
D.I.C. > Strain Gage

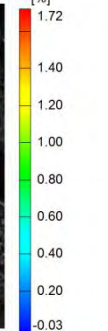
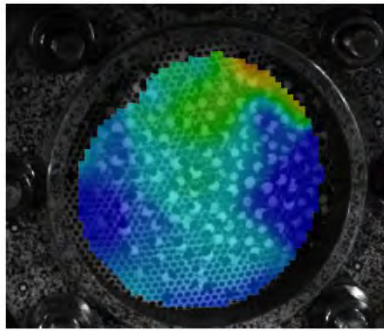


Goal: Augment Traditional Gages to gain a better understanding of hardware and environment loads to design more efficient components and systems



Applications and Development work for Digital Image Correlation at NASA

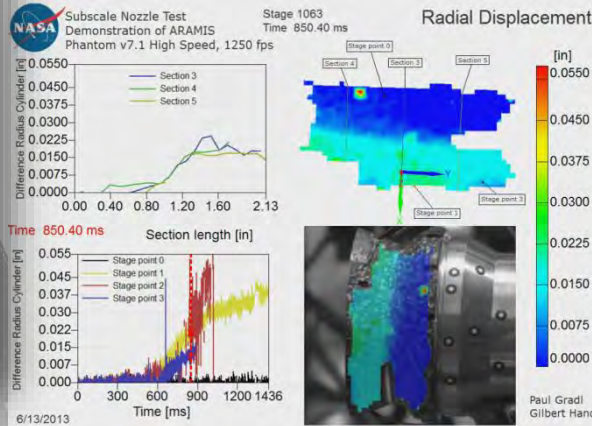
Test 91 April 3, 2013 300 SS 0.005" Half H2O Major Strain



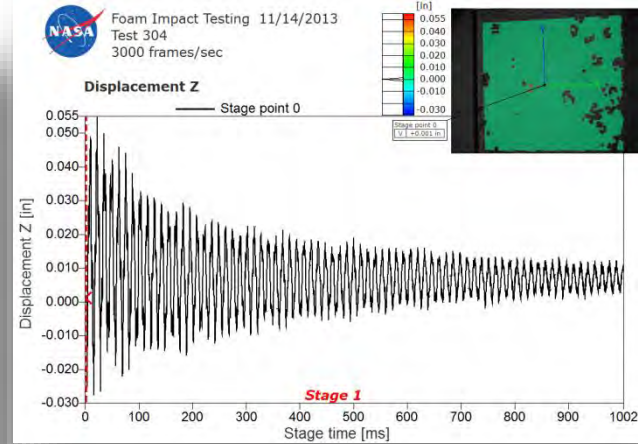
Last Frame Before Perforation



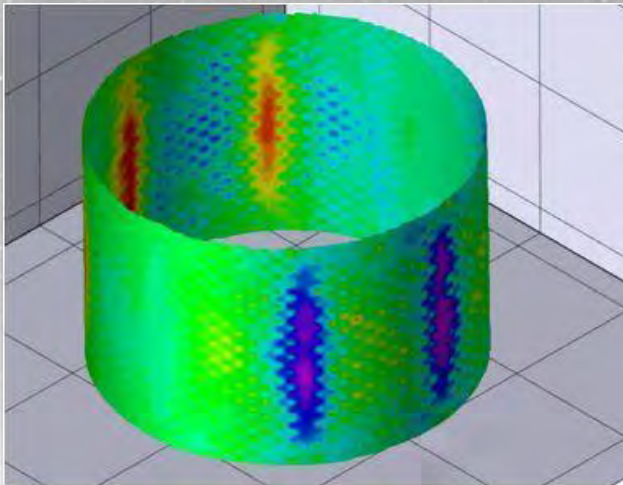
Blast Pressure Wave Tracking at 70,000 fps



Subscale Nozzle Displacements at 1700F

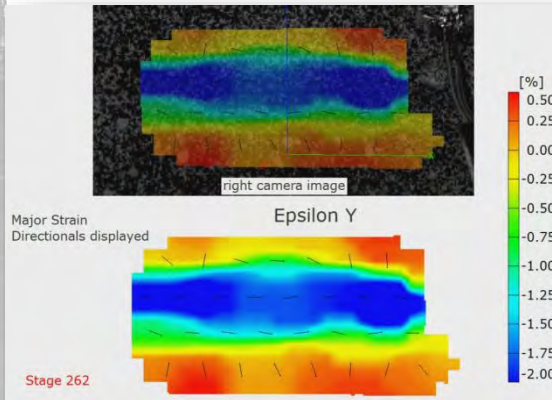


Debris Impact Testing – Eliminated Strain Gages

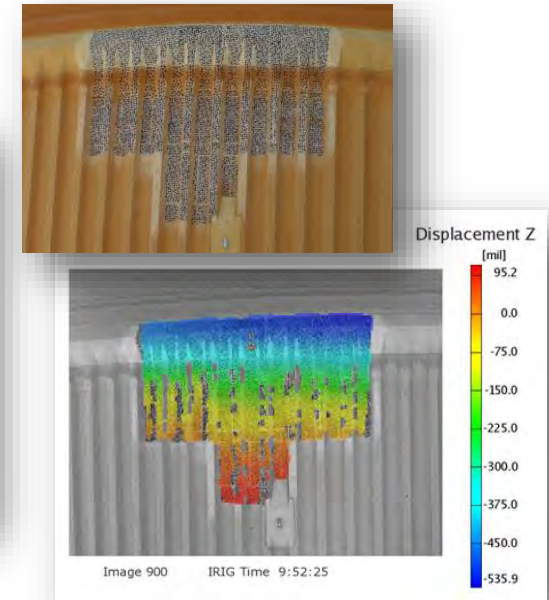


Full-Field Strain and Displacements of 18-ft Dia Tank

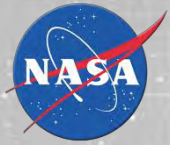
Ref: Todd Boles, MSFC/ET30



High Speed Composite Compression – Direct Application of Major Strain



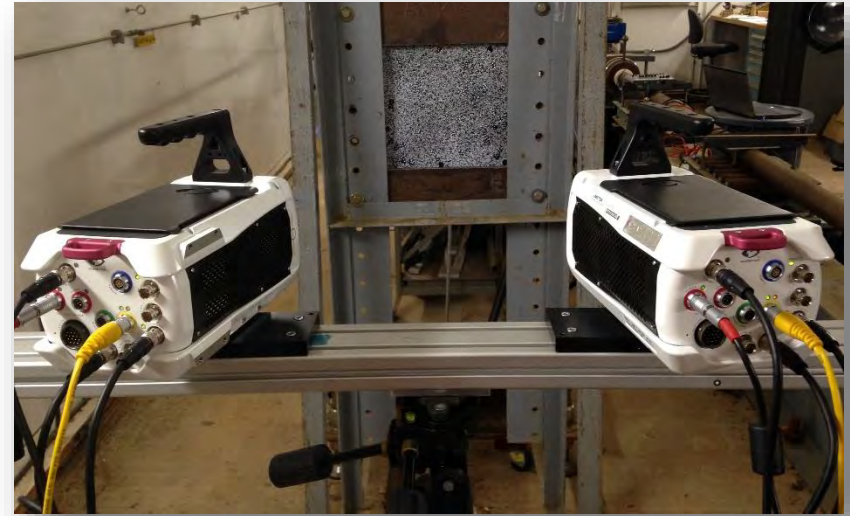
ET (on Pad) Cryo tanking test to observe stringer displacement



Digital Image Correlation - Overview of Technology

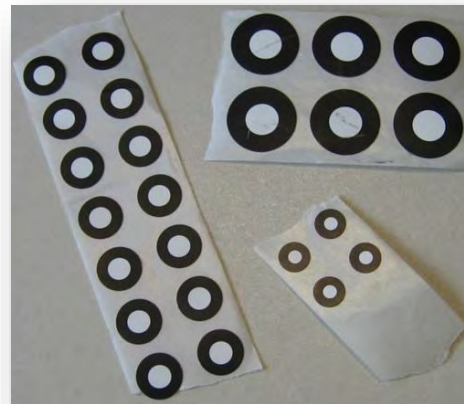
Photos by: Paul Gradl and Gilbert Handley

- Uses paired high speed video cameras calibrated to a volume to full field surface data
- Post-processing of paired images to determine **Displacement of surface, strains, acceleration, velocity**
- High Speed cameras can provide high frame rate although frame rate limited by duration of test and current post-processing techniques (tremendous amounts of data)

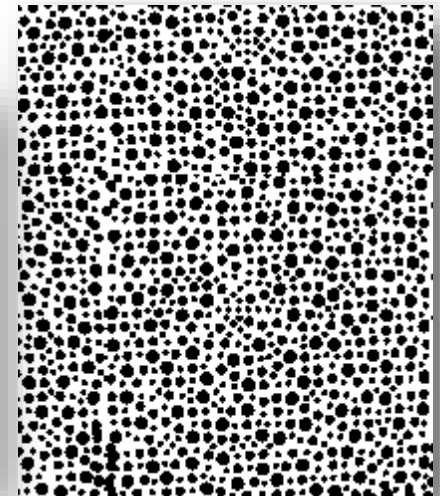


ARAMIS

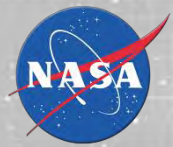
PONTOS



Discrete Point Setup

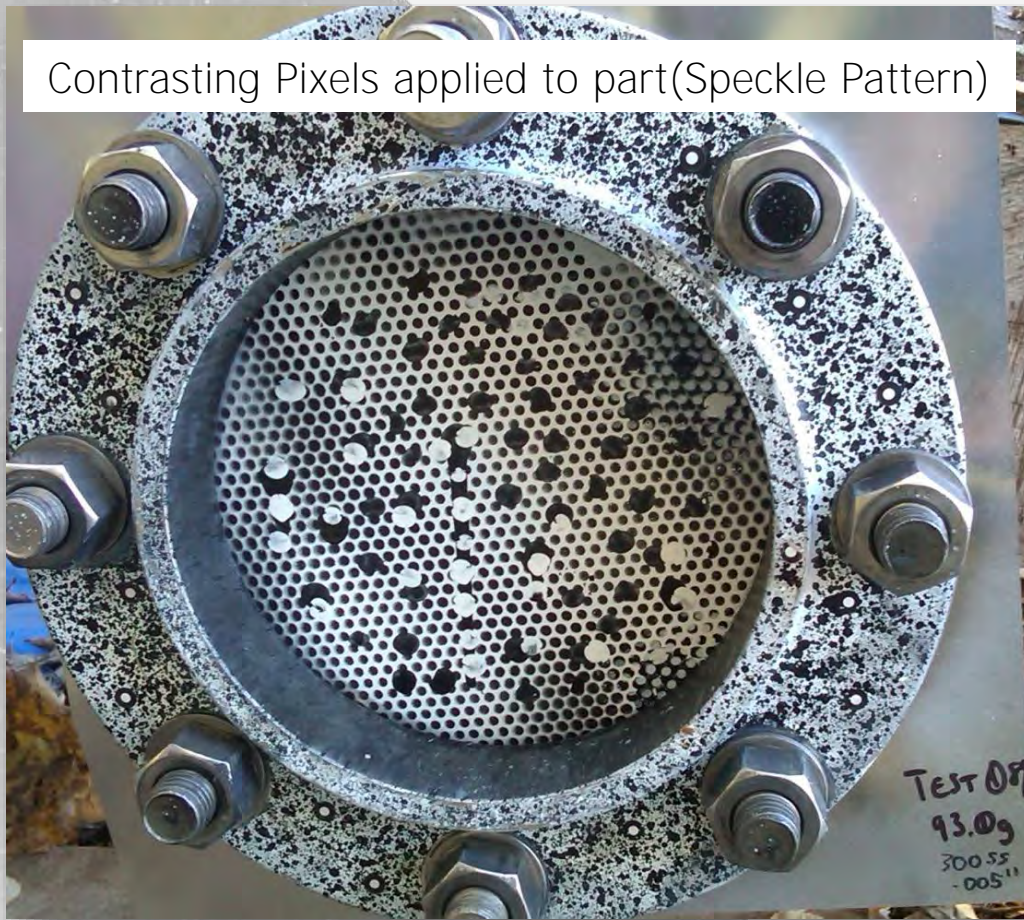


Full Surface Setup



What is Digital Image Correlation?

Contrasting Pixels applied to part (Speckle Pattern)



Stereo Camera Triangulation

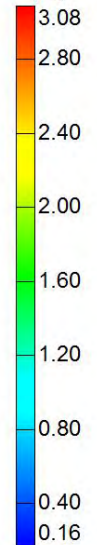
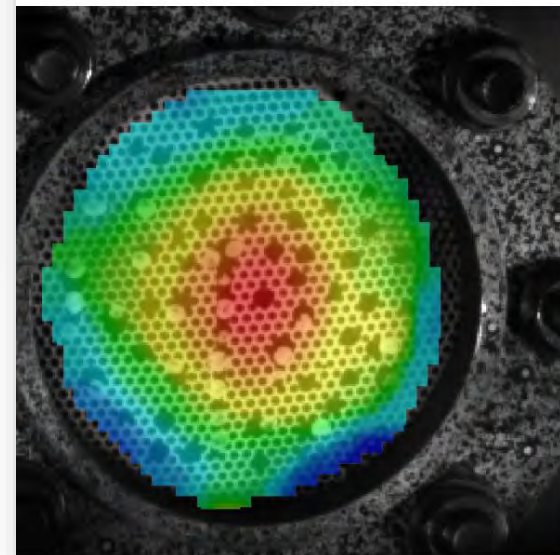


Photo Provided by: Tim Schmidt / Trillion

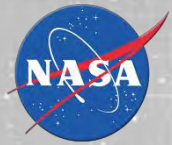
Oct 24, 2012 300 SS 0.005"

Major Strain

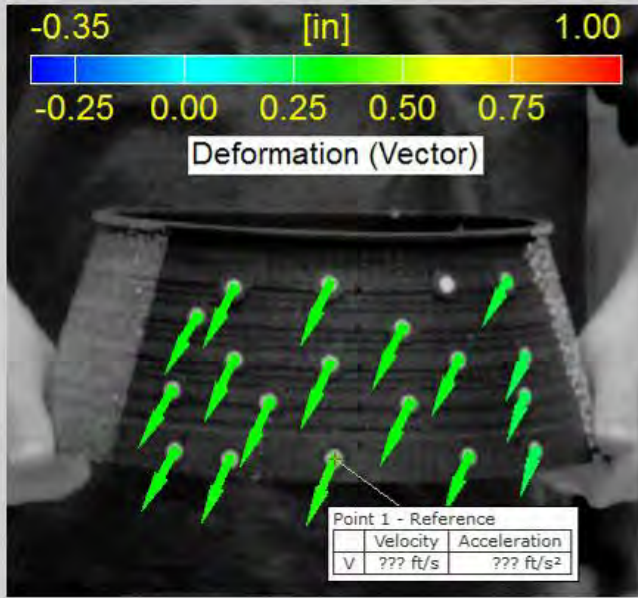
[%]



= Full Field
Displacement and
Strain Measurements



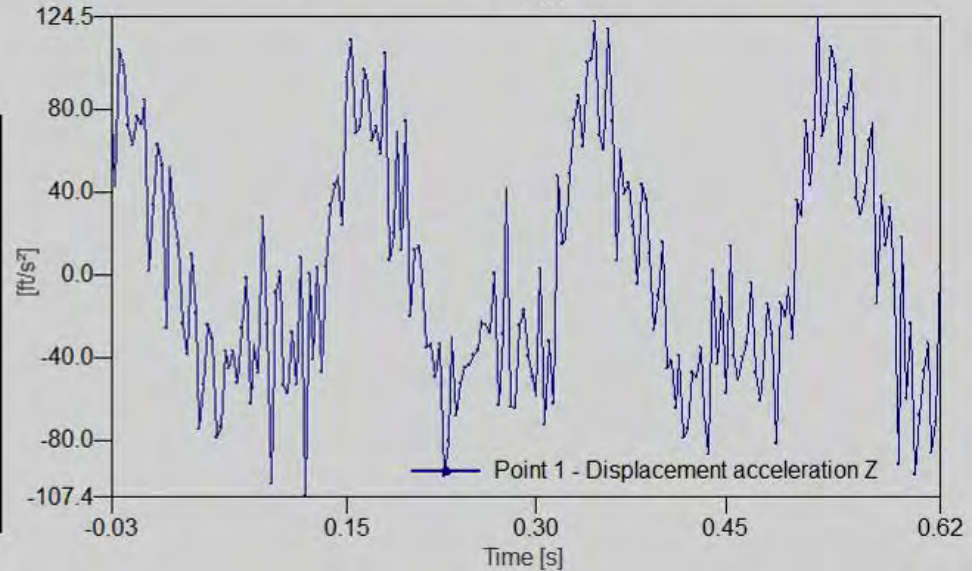
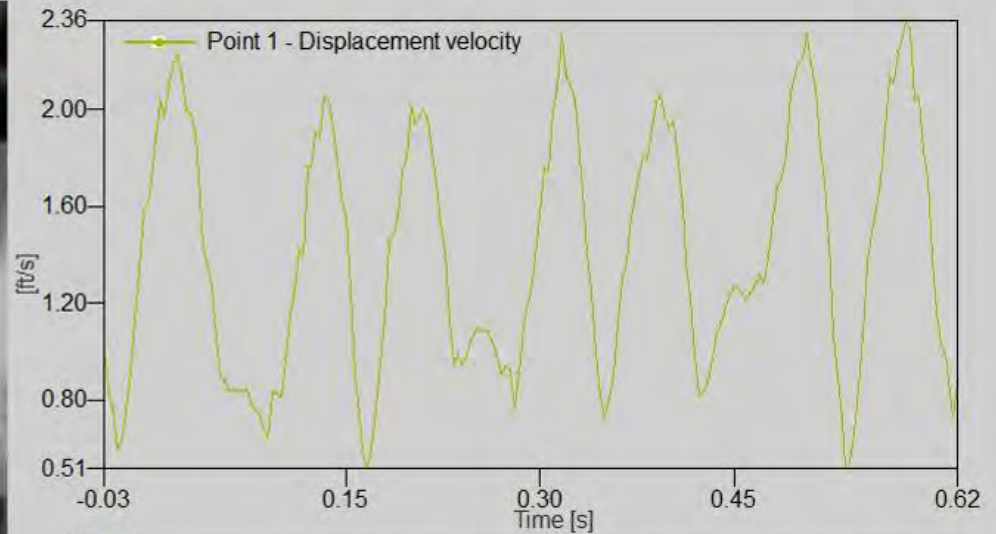
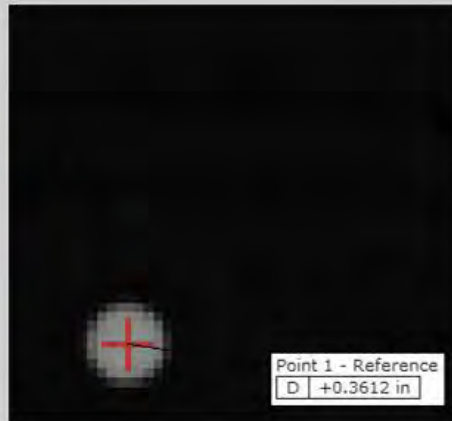
PONTOS Lab Experiments

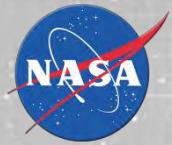


Nozzle Deformation
 Date: 4/2/2013
 0.000000 sec
 Phantom 7.1M HS
 50mm lenses



Paul Gradl
 Gilbert Handley

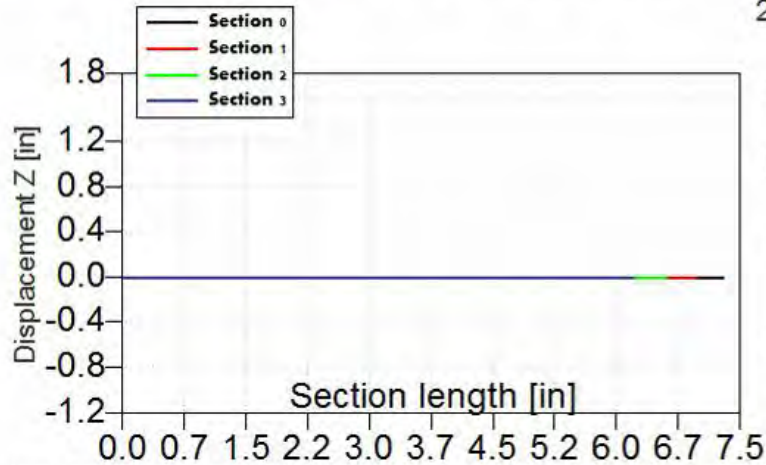




ARAMIS Lab Experiments – Displacement

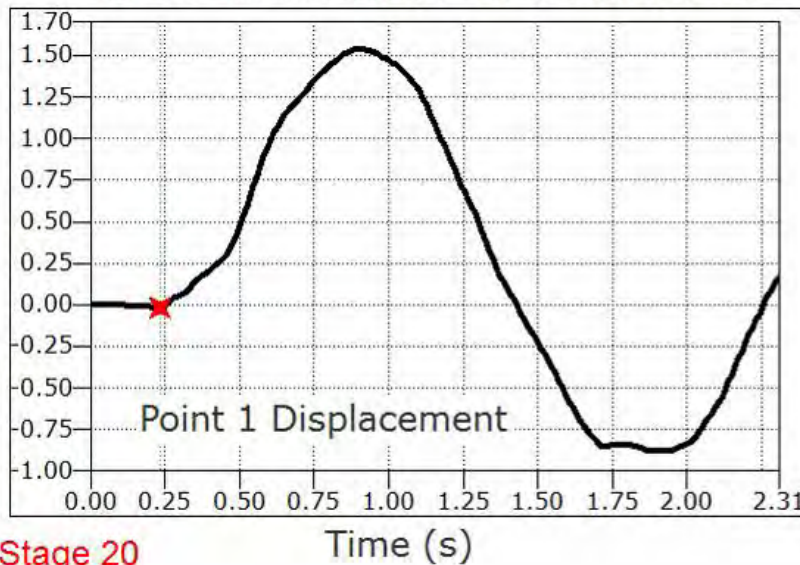
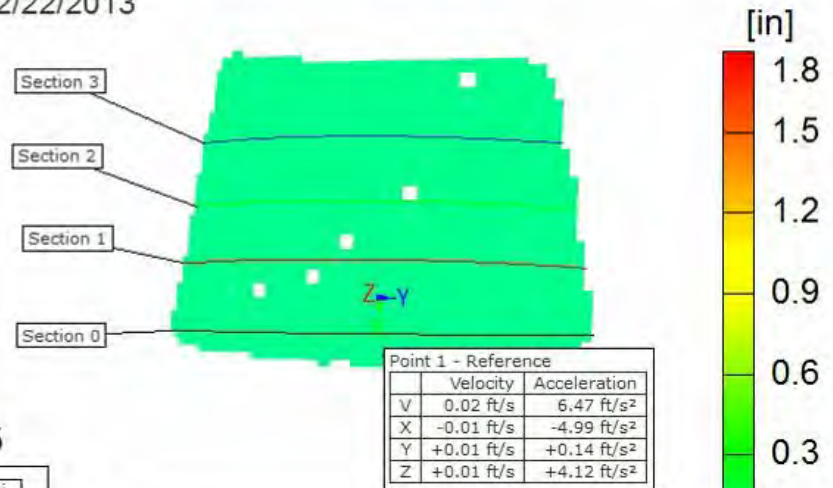
Stage 20

Nozzle Displacement Z



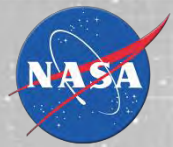
Stage 20
Time 0.23 s
2/22/2013

Displacement Z



Stage 20

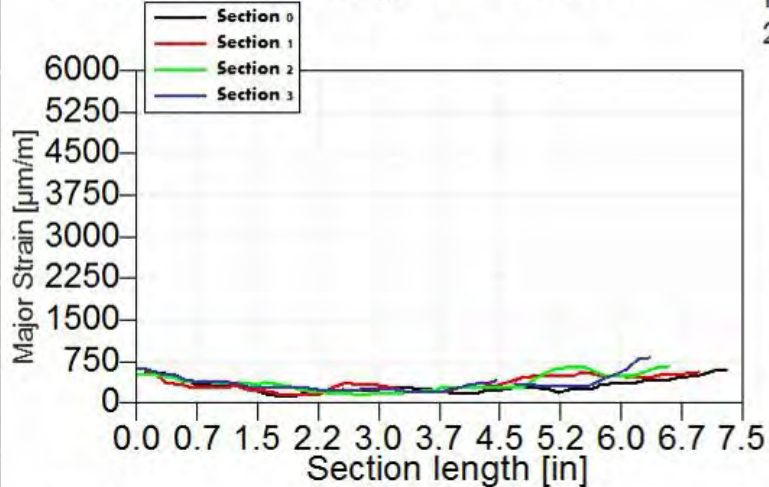




ARAMIS Lab Experiments – Principal Strain

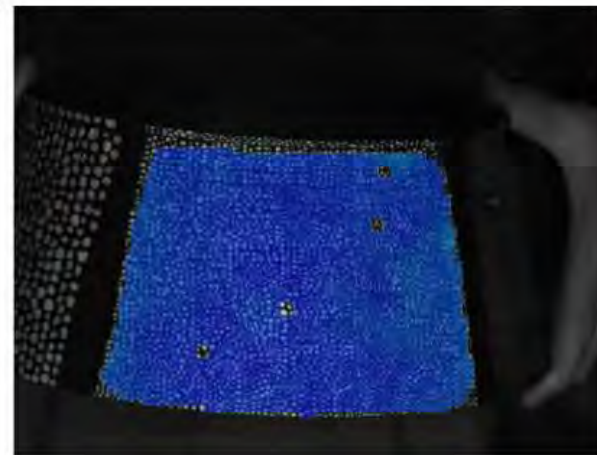
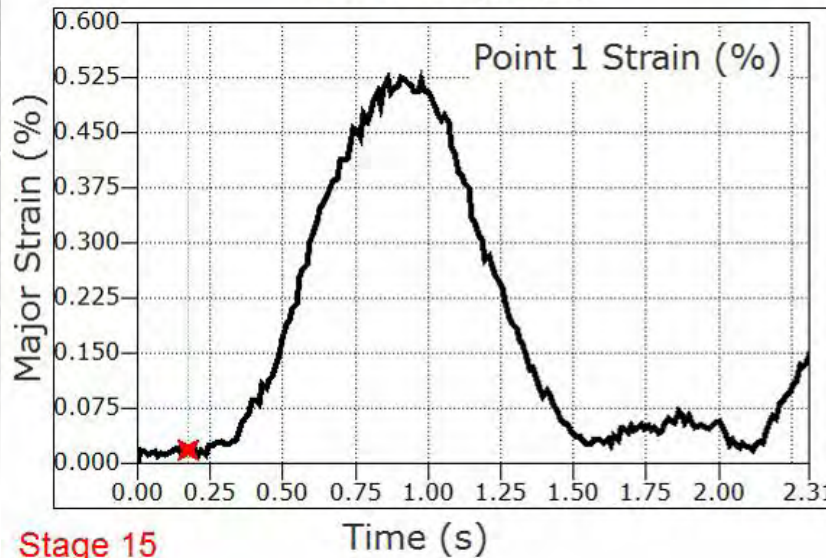
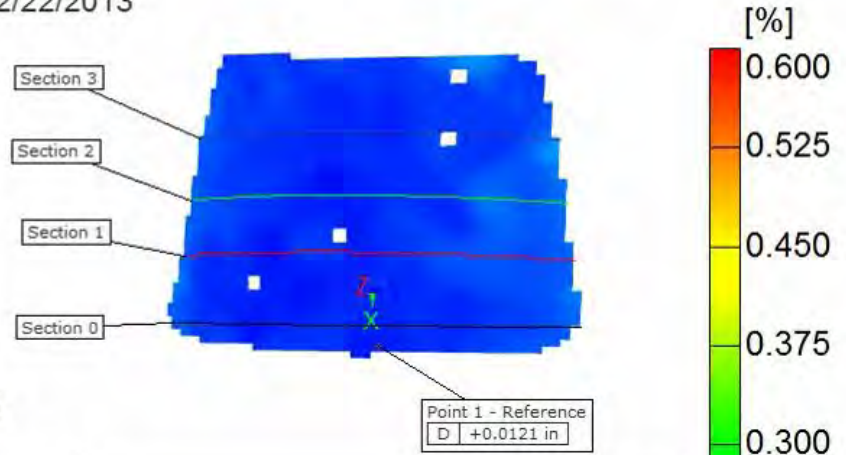
Stage 15

Nozzle - Major Strain



Stage 15
Time 0.17 s
2/22/2013

Major Strain

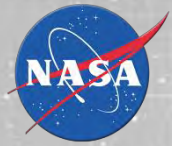


Stage 15

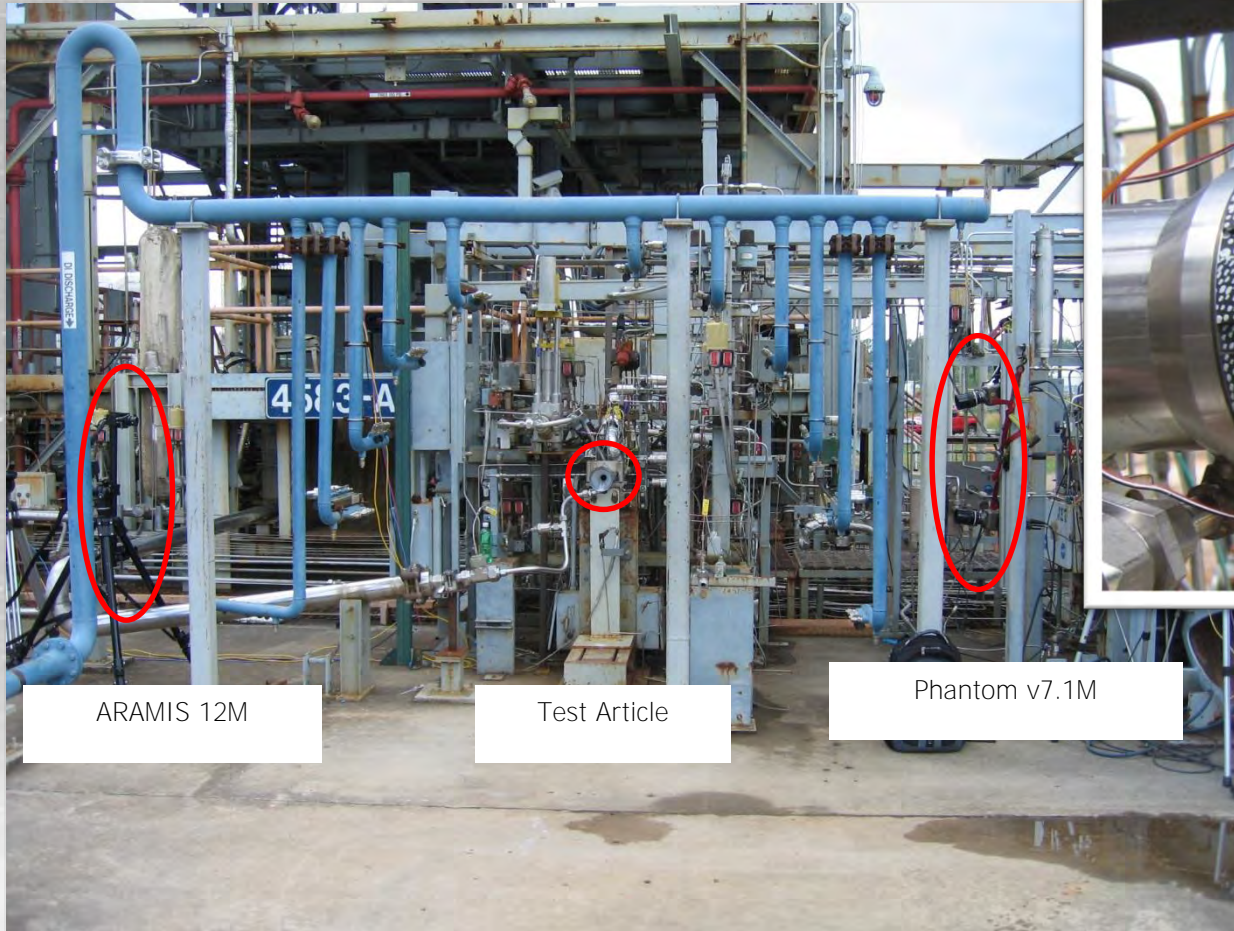
Time (s)

Paul Gradl
Gilbert Handley





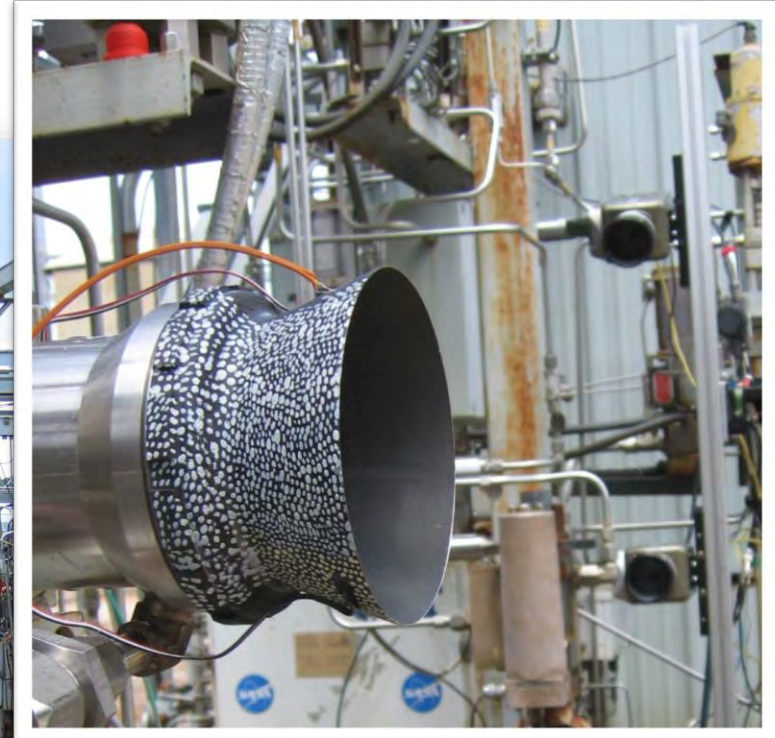
Subscale Hot-fire Nozzle Testing



ARAMIS 12M

Test Article

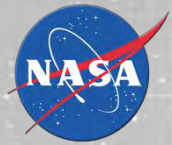
Phantom v7.1M



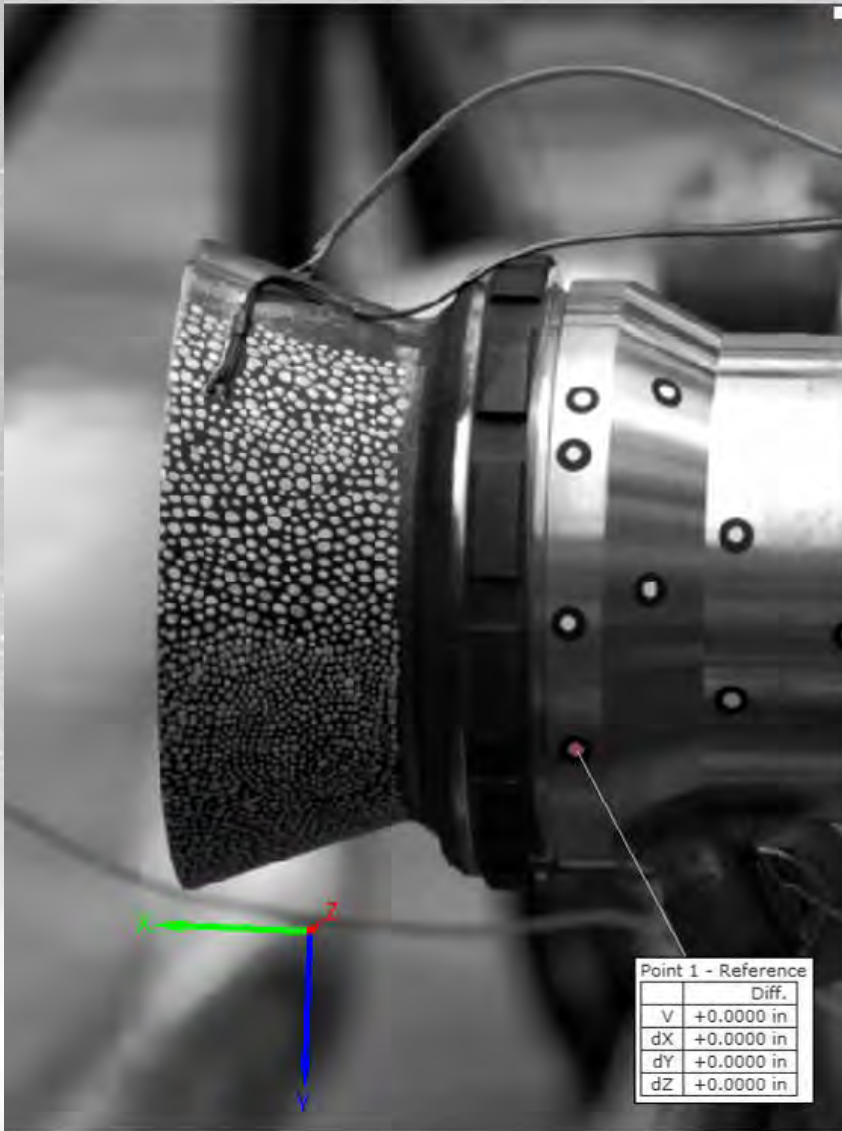
Additive Manufactured Radiative Cooled Nozzle Extension

Test Photos and Data Collection:

Paul Gradl
Gilbert Handley
Sandy Elam Greene



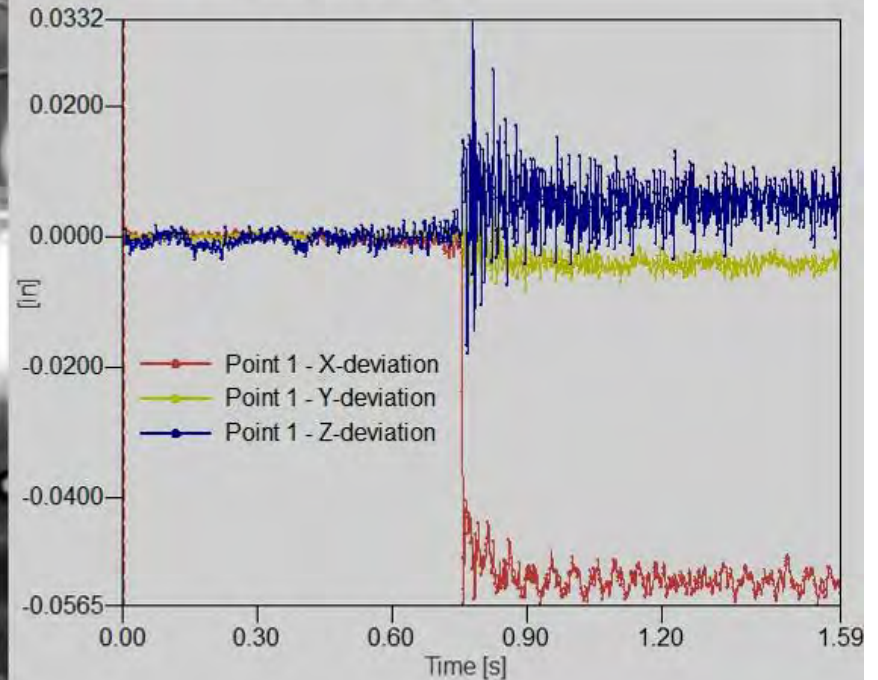
Bench Testing Doesn't Always Translate into the Field...



Point 1 - Reference	
	Diff.
V	+0.0000 in
dX	+0.0000 in
dY	+0.0000 in
dZ	+0.0000 in

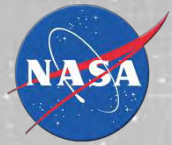


Nozzle Extension Skirt Buckling Test
Intentional Predicted Failure
May 22, 2013



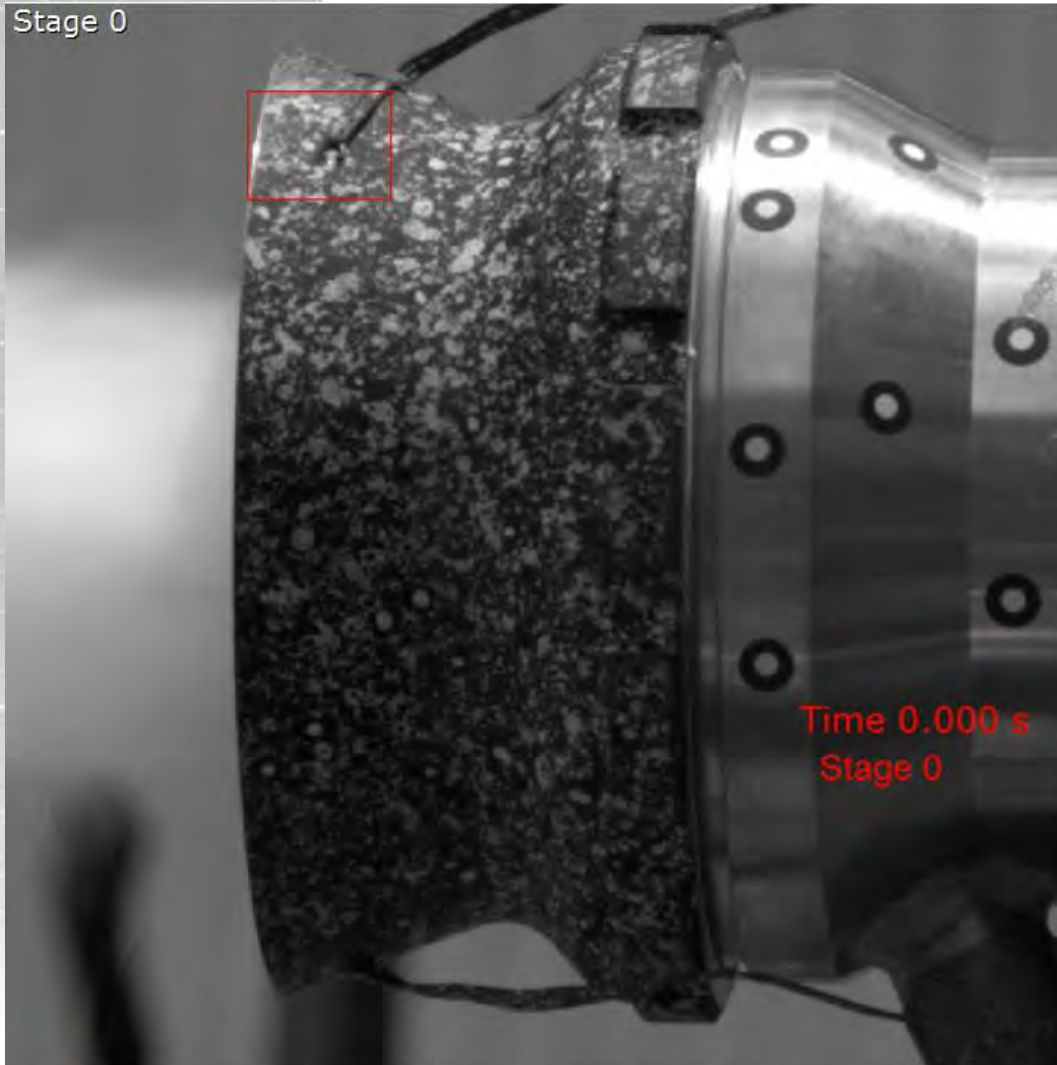
Time: 0.000000 s from trigger

Phantom High Speed v7.1M
750 fps
135mm lens @6ft

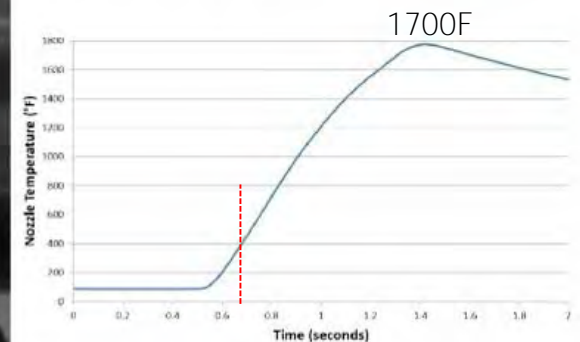
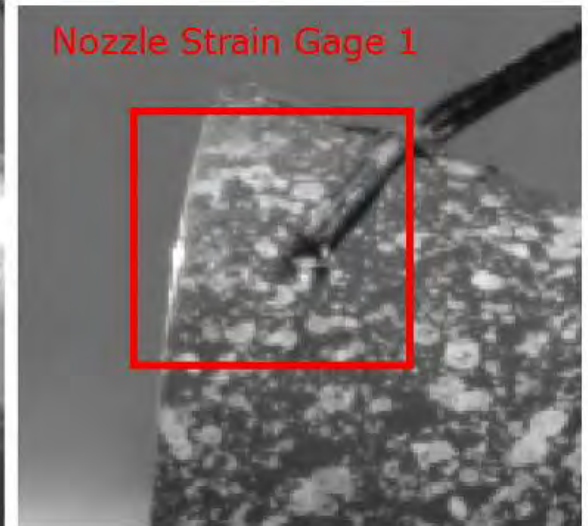


Motivation to Develop Technique

Stage 0



Nozzle Strain Gage 1



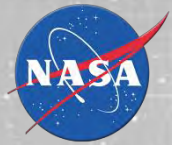
Strain Gage Failure at ~400F



Subscale Nozzle Hotfire Demonstration
Phantom v7.1 M, 1250 fps

6/13/2013

Paul Gradl
Gilbert Handley



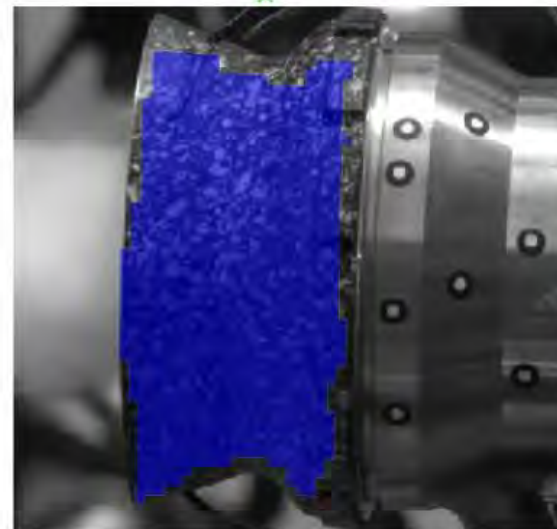
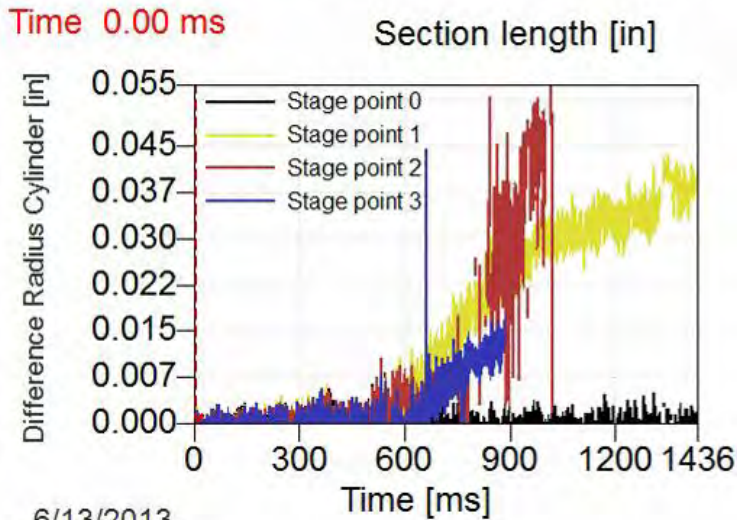
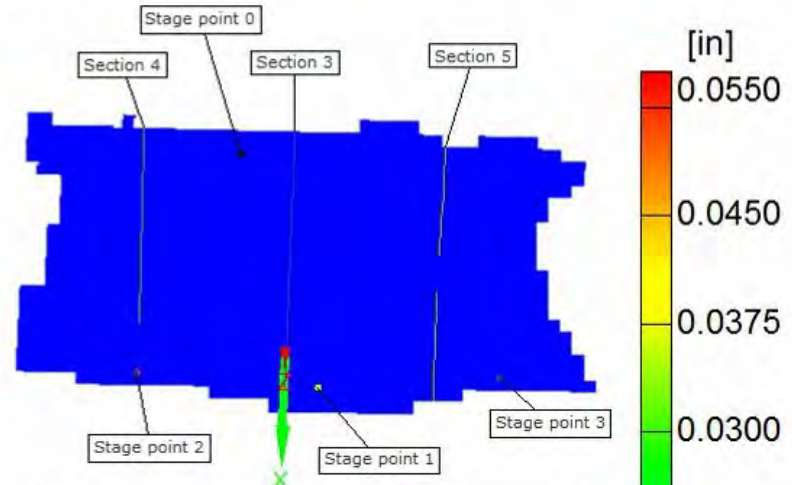
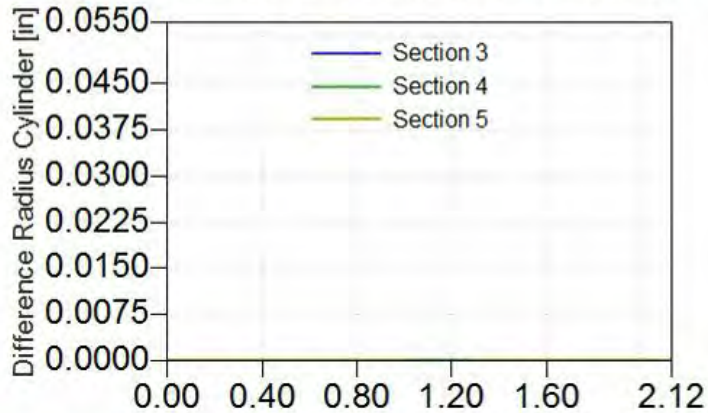
Subscale Hotfire Testing on Nozzle



Subscale Nozzle Test
Demonstration of ARAMIS
Phantom v7.1 High Speed, 1250 fps

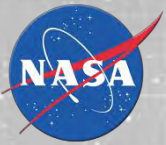
Stage 0
Time 0.00 ms

Radial Displacement

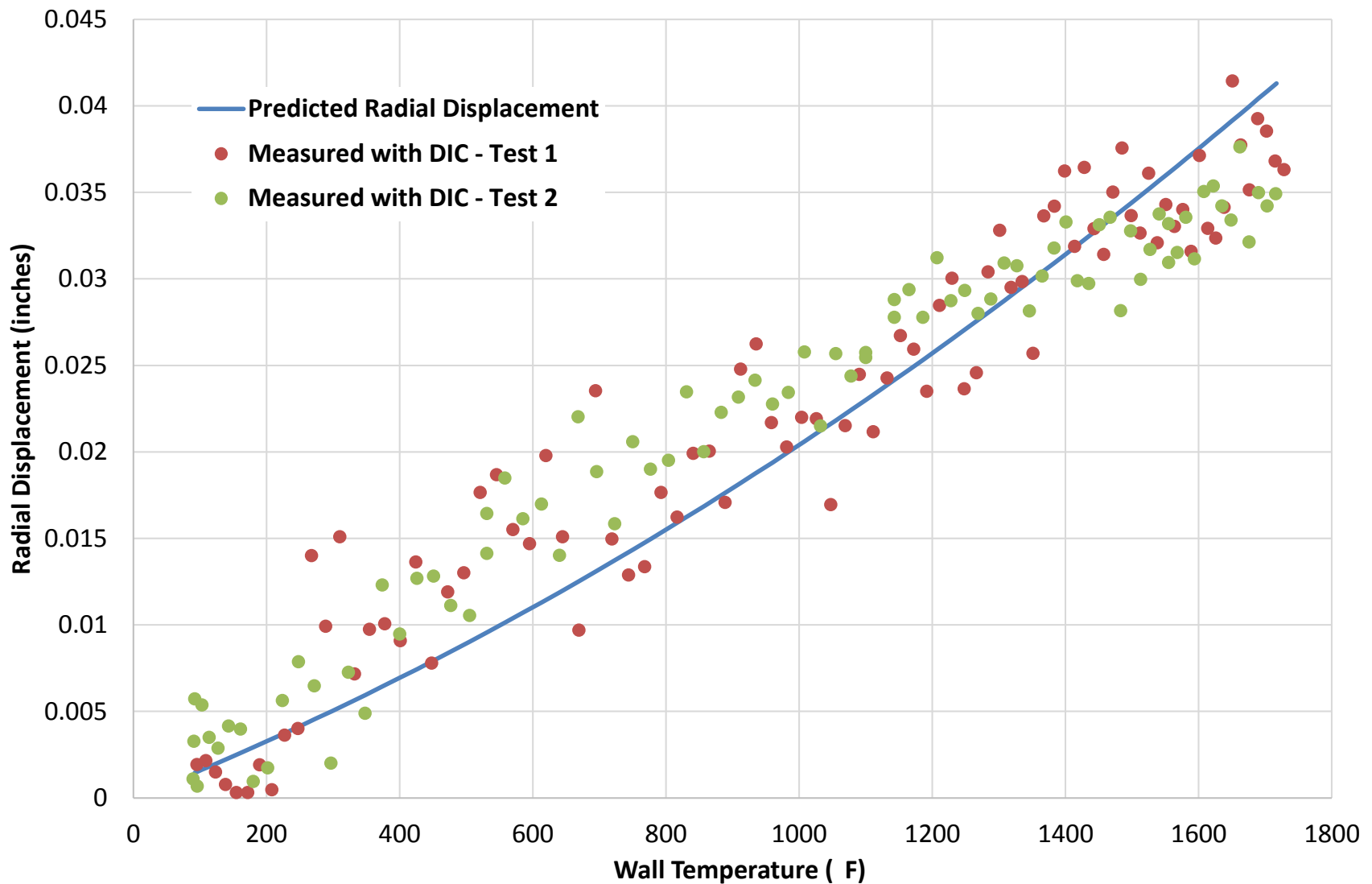


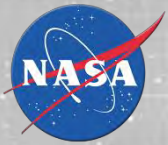
Paul Gradl
Gilbert Handley

6/13/2013

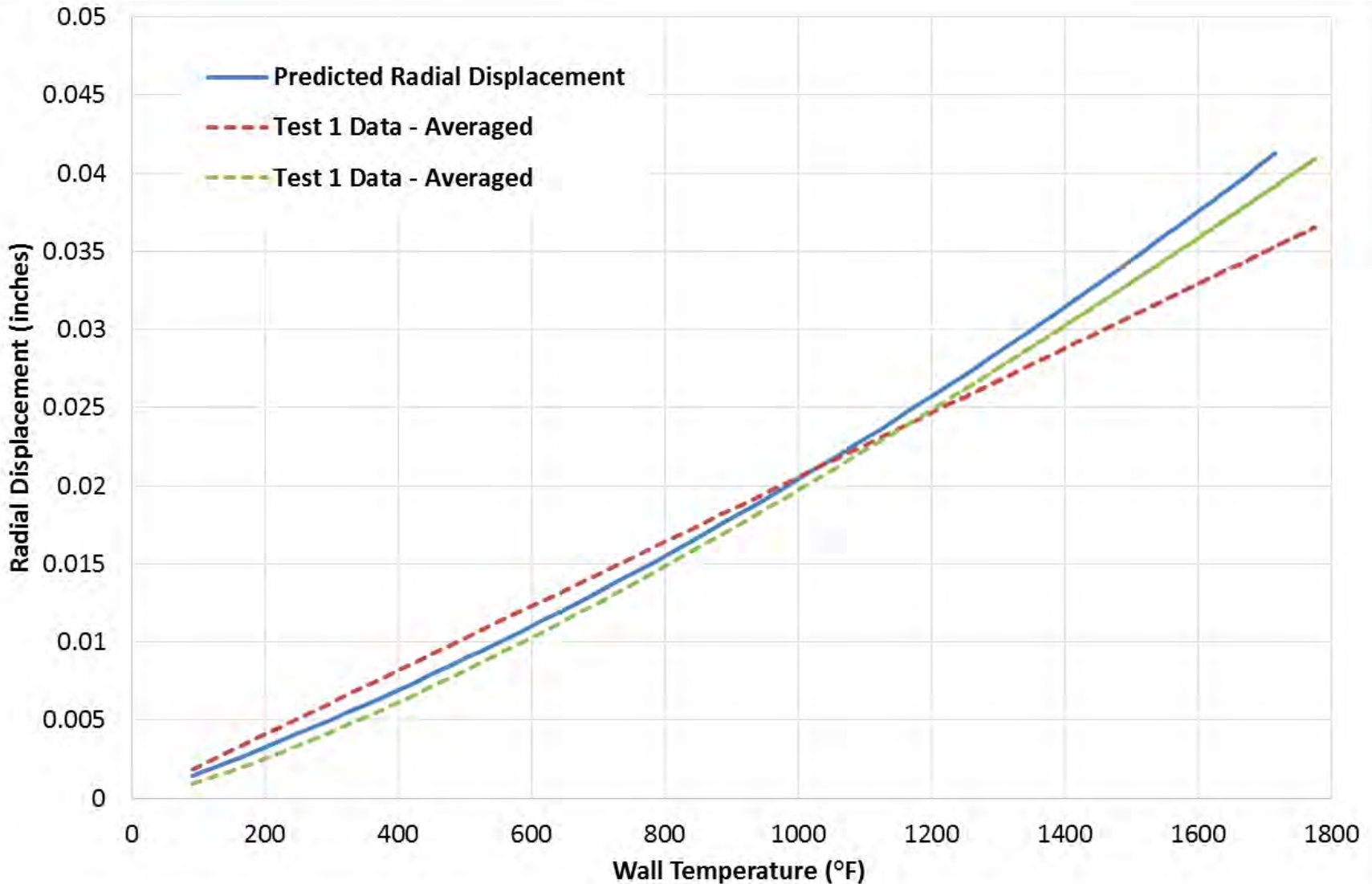


Subscale Hotfire Testing – Data Analysis

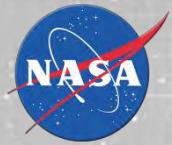




Subscale Hotfire Testing – Averaged Data



Optical test data tracking closely with predictions; error grows at elevated temperatures

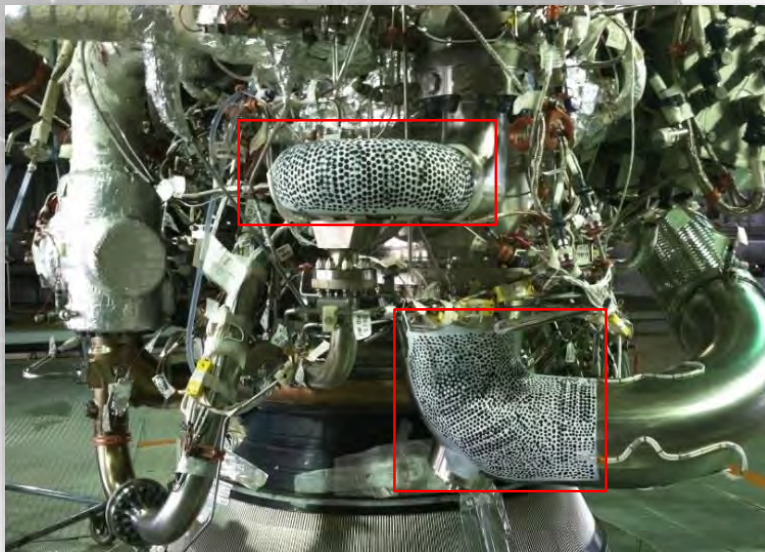


Large Scale D.I.C. for Engine Hotfire Testing

MSFC has developed new optical measurement techniques to augment or replace traditional gages in harsh environment engine testing or manufacturing operations

Stereo high-speed cameras measure full-surface displacements and strains using "speckle pattern" (calibrated triangulation)

- Leveraged basic techniques from NESC Shell Buckling Test and NASA & industry experts
- Developed speckle pattern and initial vibration damping in subscale hotfire testing at MSFC
- J-2X provided the test-bed environment to develop camera stability damping
- Industry-first attempt for high temperature, high vibration environments where traditional gages do not operate reliably



Stereo Cameras installed and Speckle Pattern Applied at Stennis A1 Stand

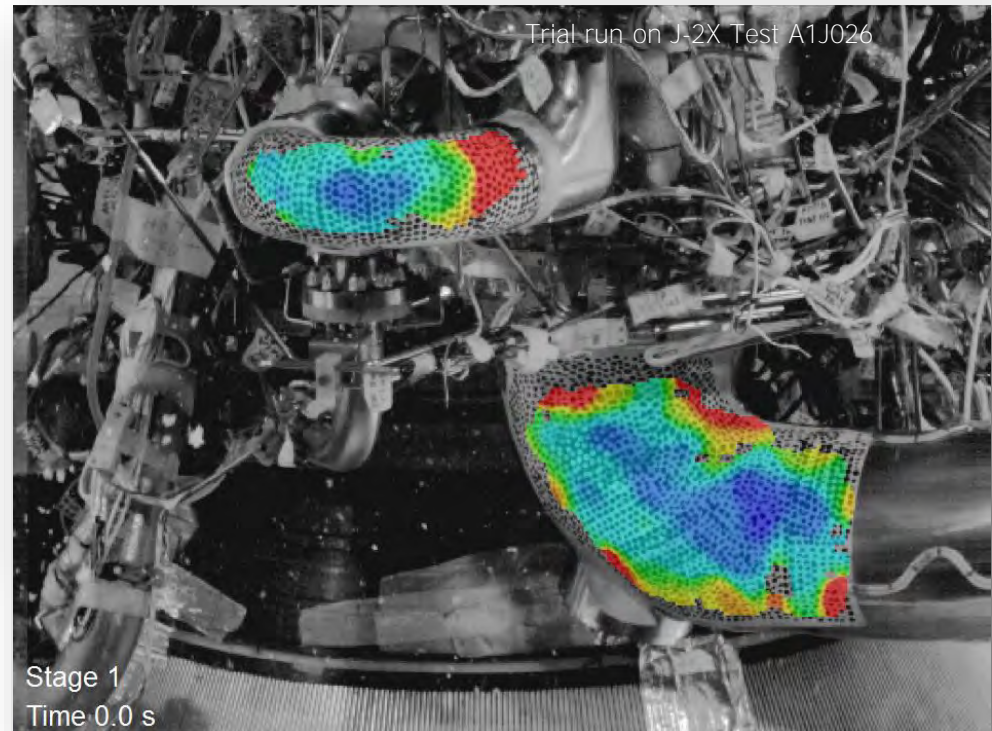
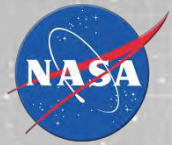




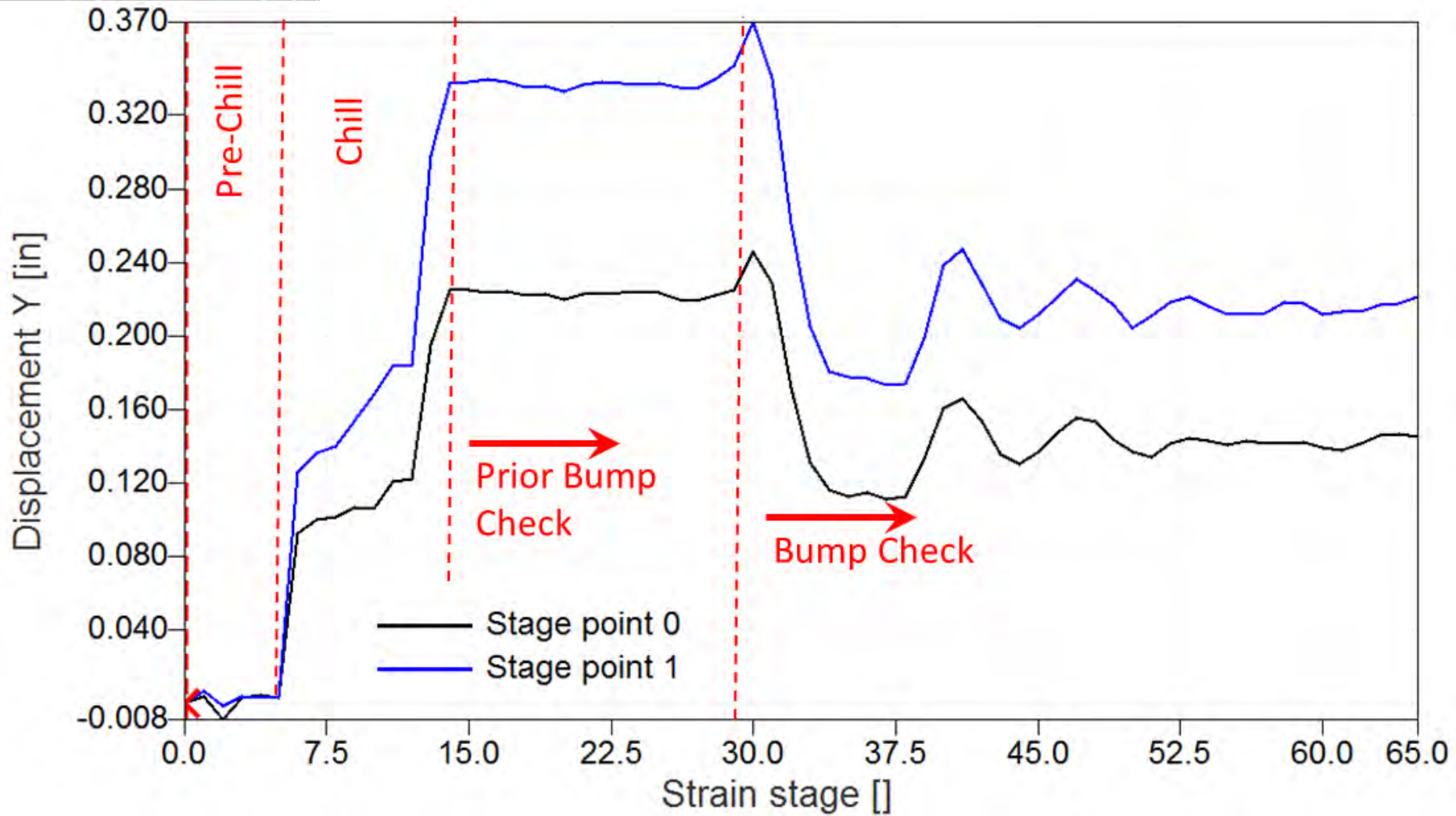
Photo Credit: Dan Goade

Test Data Collection:
Paul Gradl, Gilbert Handley, Brian West

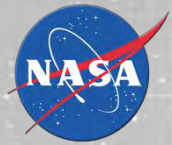
ARAMIS high speed cameras



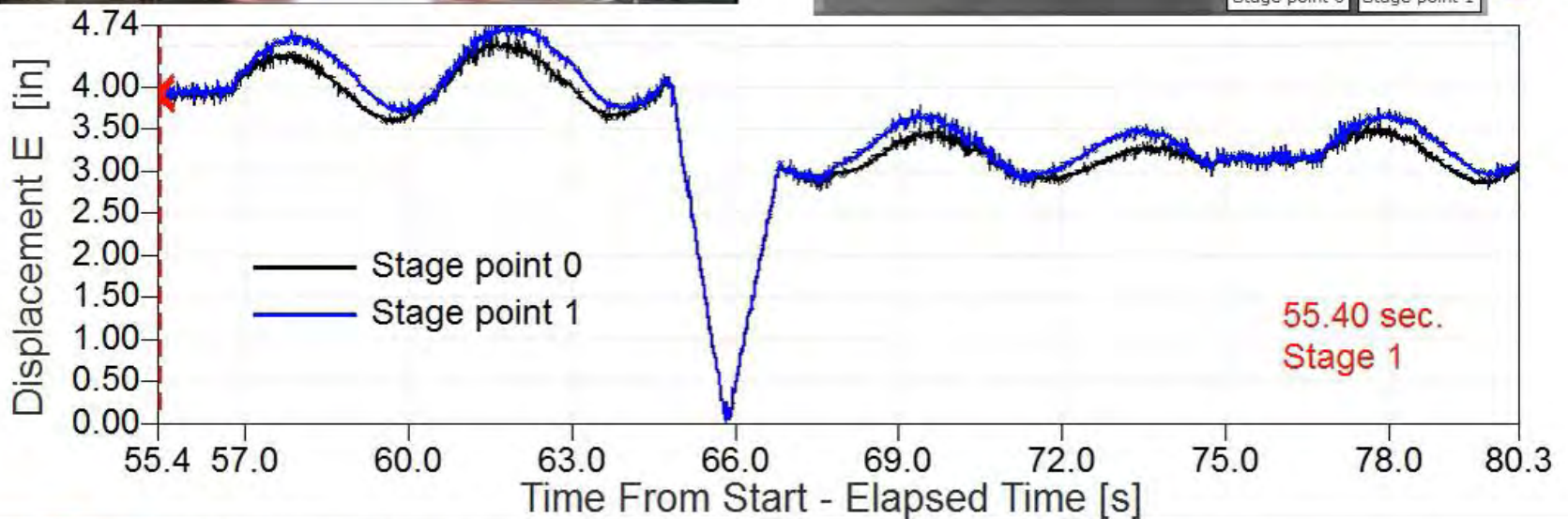
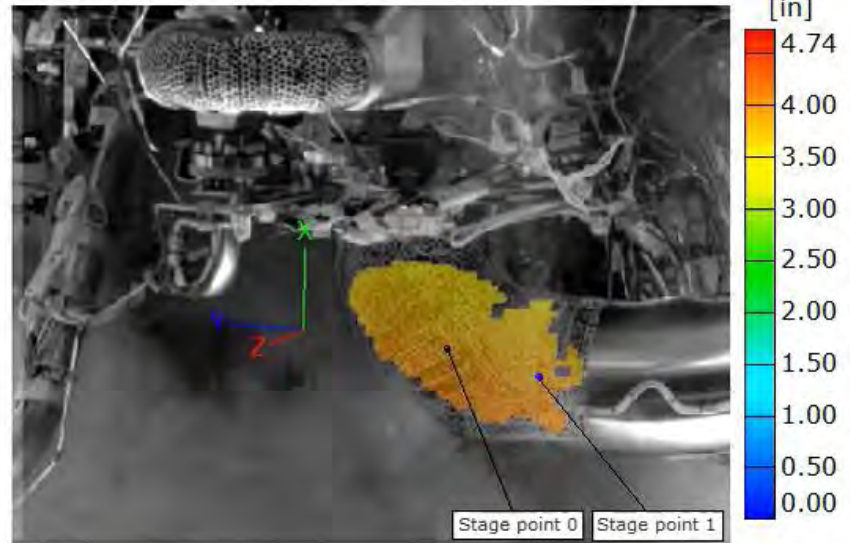
Engine Movement and Strains during Pre-test Ops



Ability to track engine during all chill and gimbal checkout operations



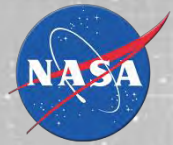
ARAMIS Full Surface Strain Measurement Proof of Concept Displacement during A1J028 Test



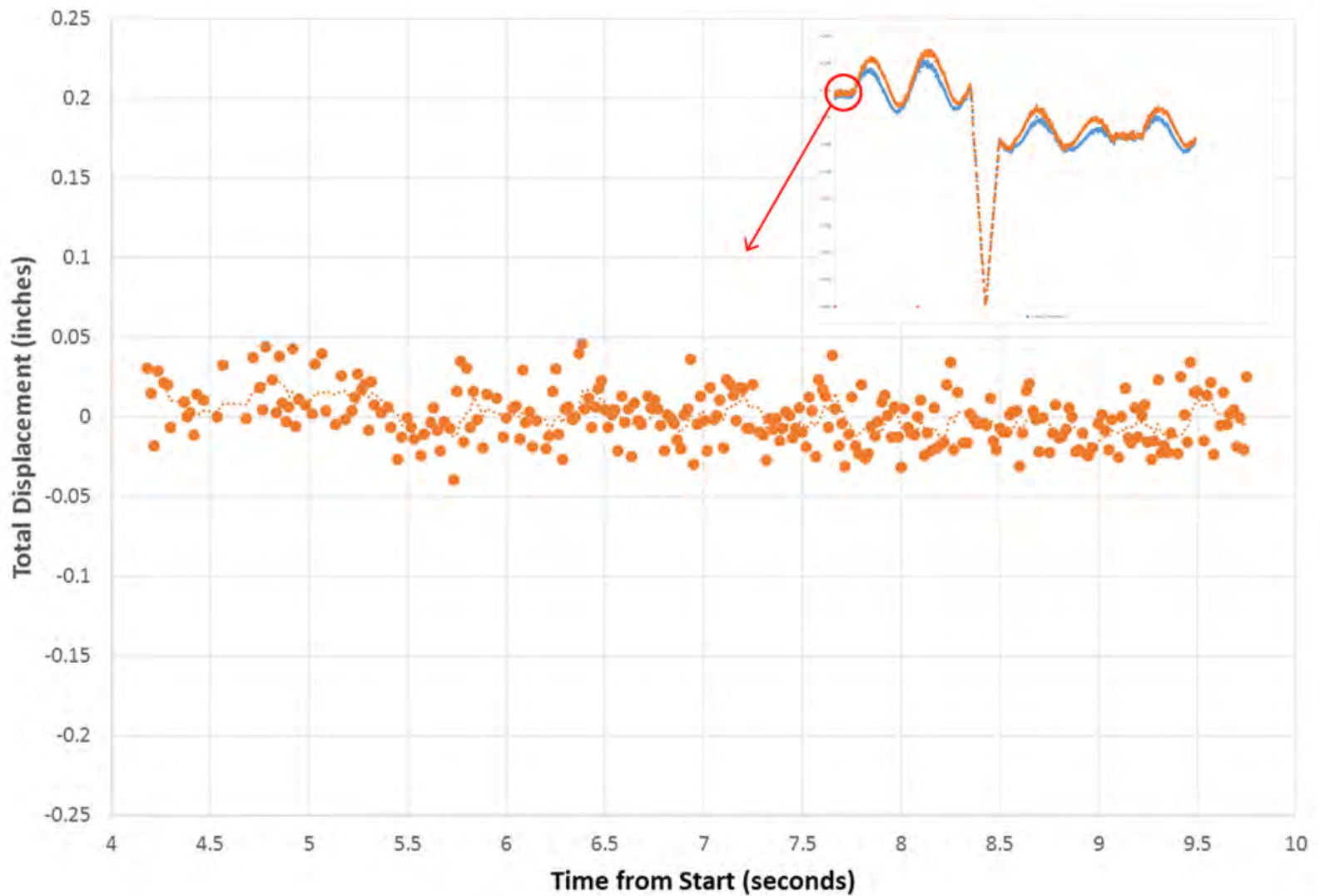
ARAMIS Trial on J-2X A1J028

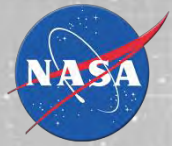
Paul Gradl
Gilbert Handley

Displacement E (Total X, Y, Z)



Error Associated With Measurements During Hotfire





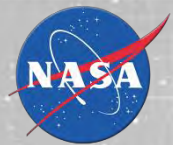
2.75" Hydra Testing Demo

Test Support: Paul Gradi/MSFC, Cory Medina/MSFC, John Tyson/Trillion, John "Yann" Psilopolous/Trillion



Demonstrated initial feasibility of using photogrammetry and digital image correlation for range testing of missile burst testing.





Feasibility of 6-dof Analysis of Missile Testing

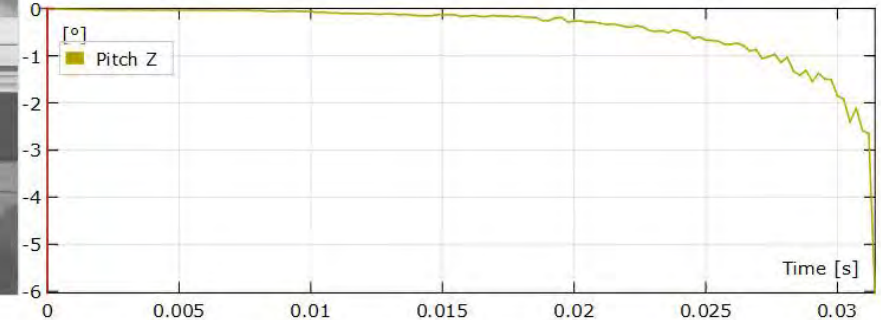
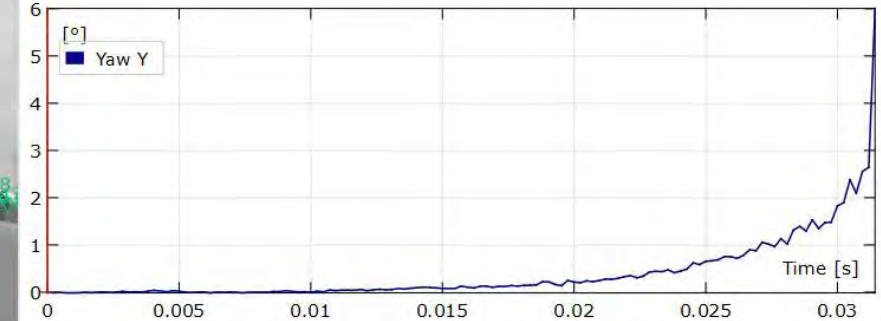
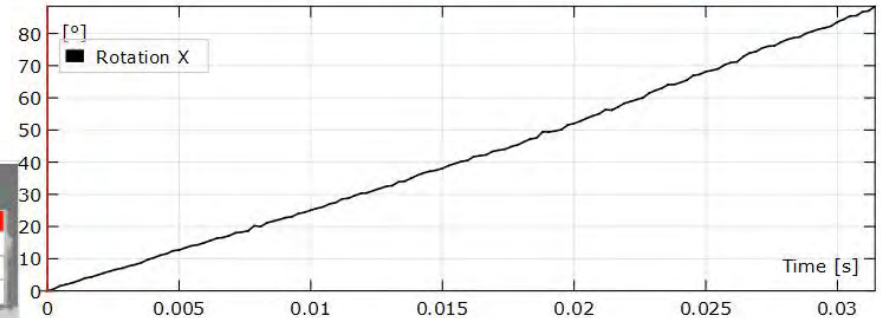
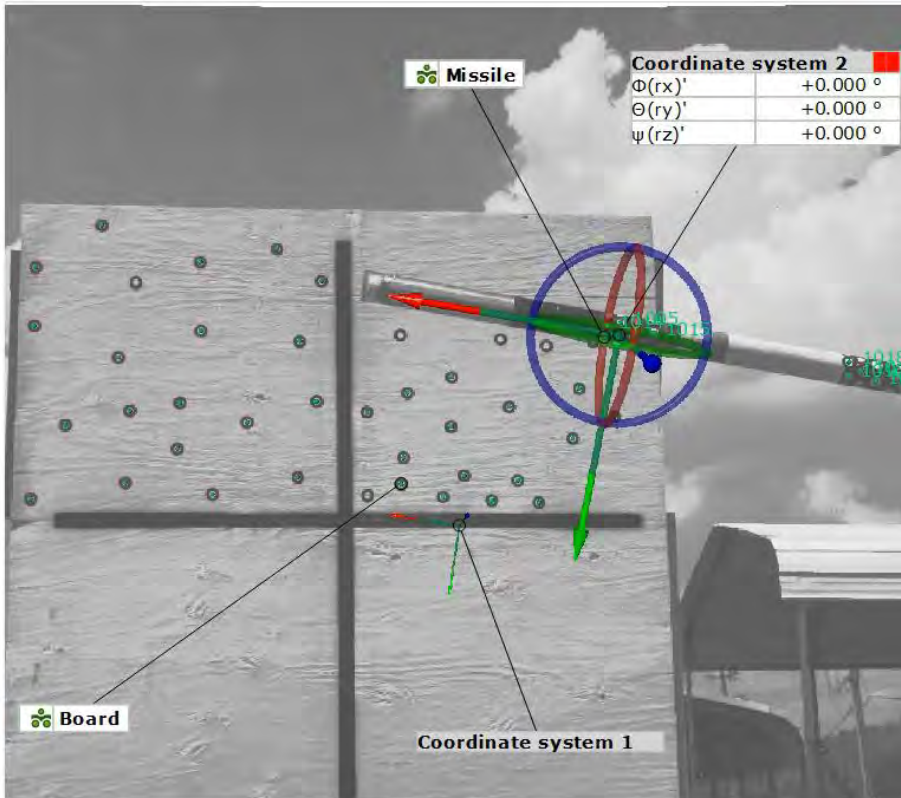


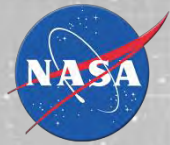
Report - PONTOS Testing

Test 62 - TM1 Hydra Missile Testing

Test Data provided:

Paul Gradl and Cory Medina





Conclusions and Future Work

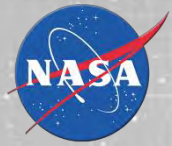
NASA MSFC has advanced a series of dynamic digital image correlation techniques for use during hotfire engine testing

- Subscale and full scale testing and analysis has demonstrated feasibility to accurately determine local and global displacements and surface strains

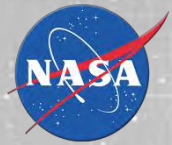
NASA will continue to advance this technology for rocket engine testing, subscale testing, component testing and bench top testing

- Replace traditional measurement systems
- Integrate with modern analysis tools
- Combine advanced techniques such as IR thermography and digital image correlation
- Continue to research and advance techniques for elevated temperature applications

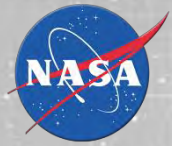
Share lessons learned with industry and government through technical papers and presentations



The possibilities of dynamic data collection are endless...

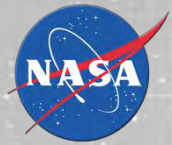


Dynamic responses
require an input to
excite the system...



Images were collected using a projected pattern instead of painting a speckle pattern on her belly...
High Speed cameras were post triggered after movements felt.

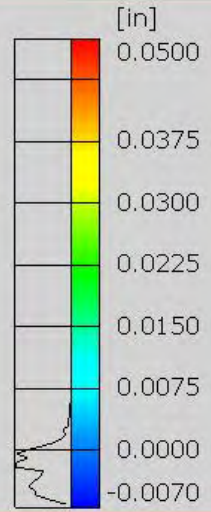
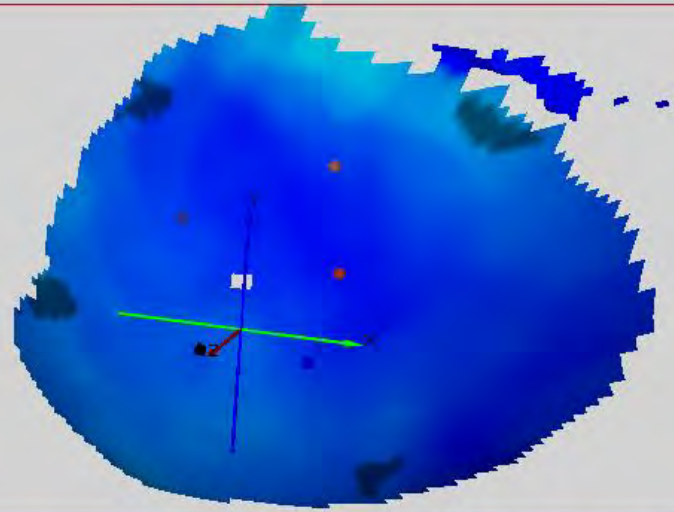
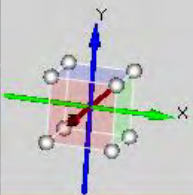




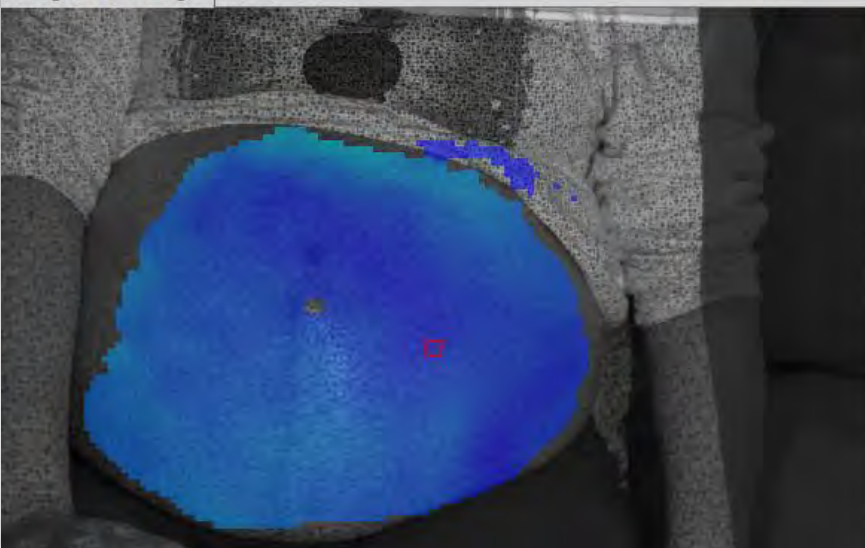
To ensure that kicks and movement data was real a background test was conducted with no baby movement (to correct for breathing and body motion)

3D View

Baby G ARAMIS_Cine 5 - No Kick.dap	
Visualization	Displacement Z
Stage from to	0 -> 4

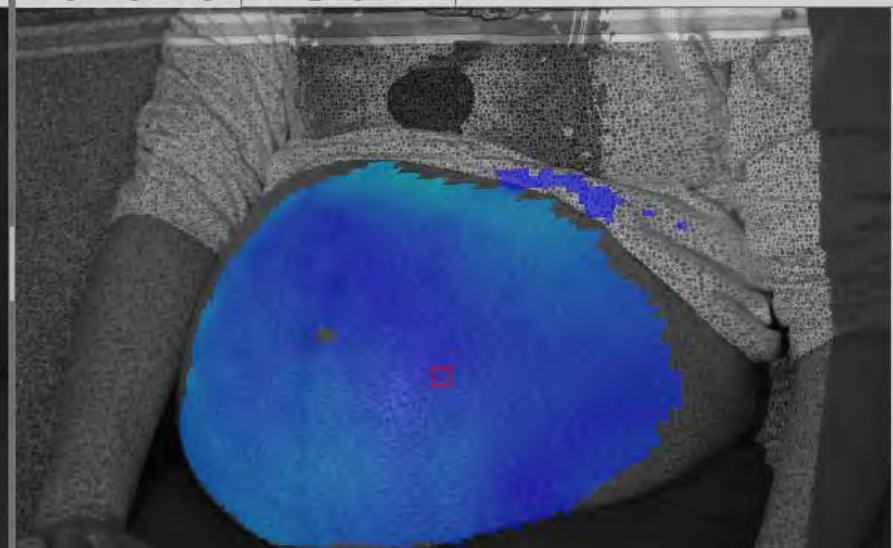


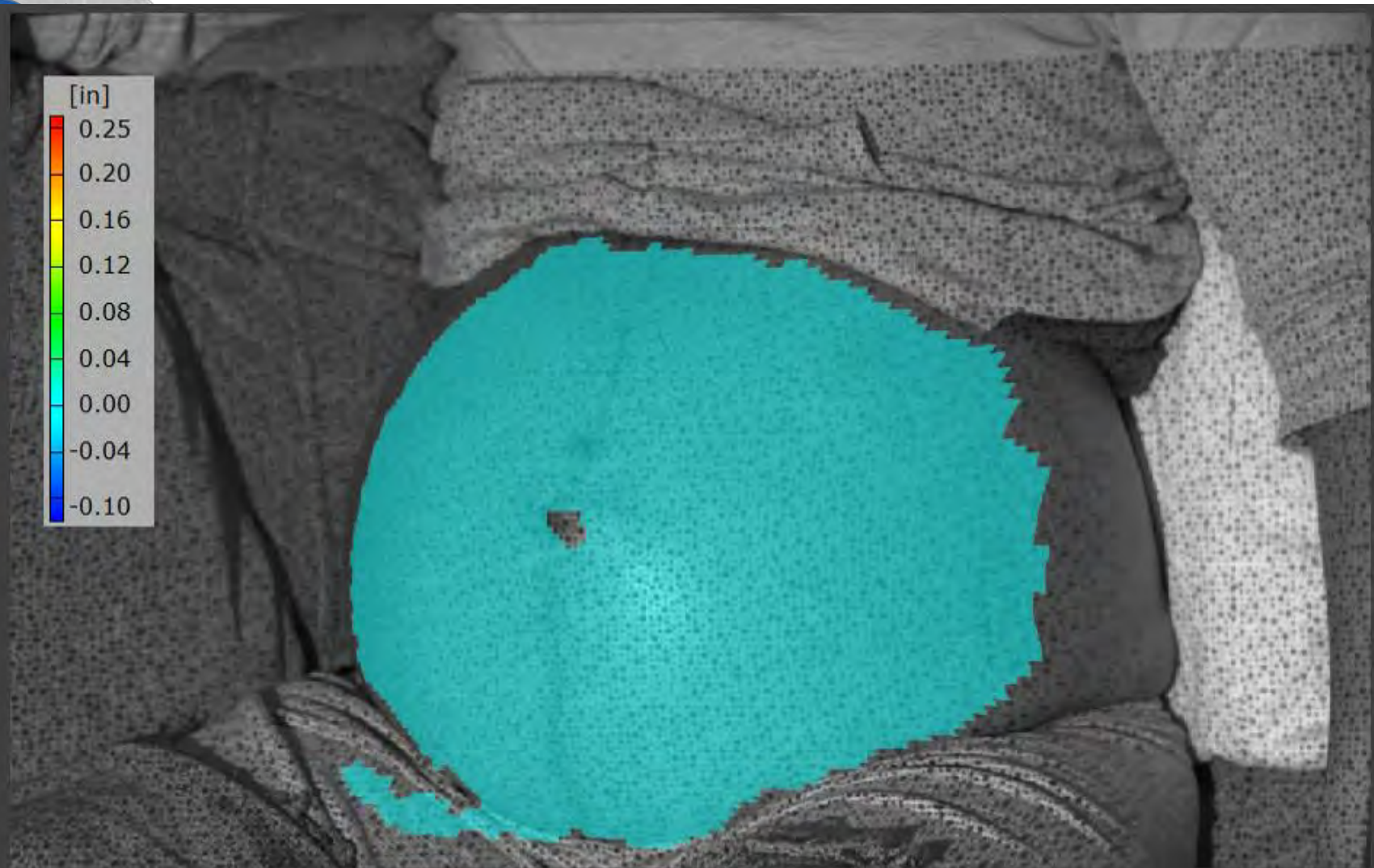
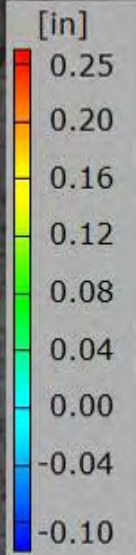
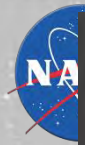
Stage 4 Left Image



Stage 4 Right Image

NASA_Stage Points

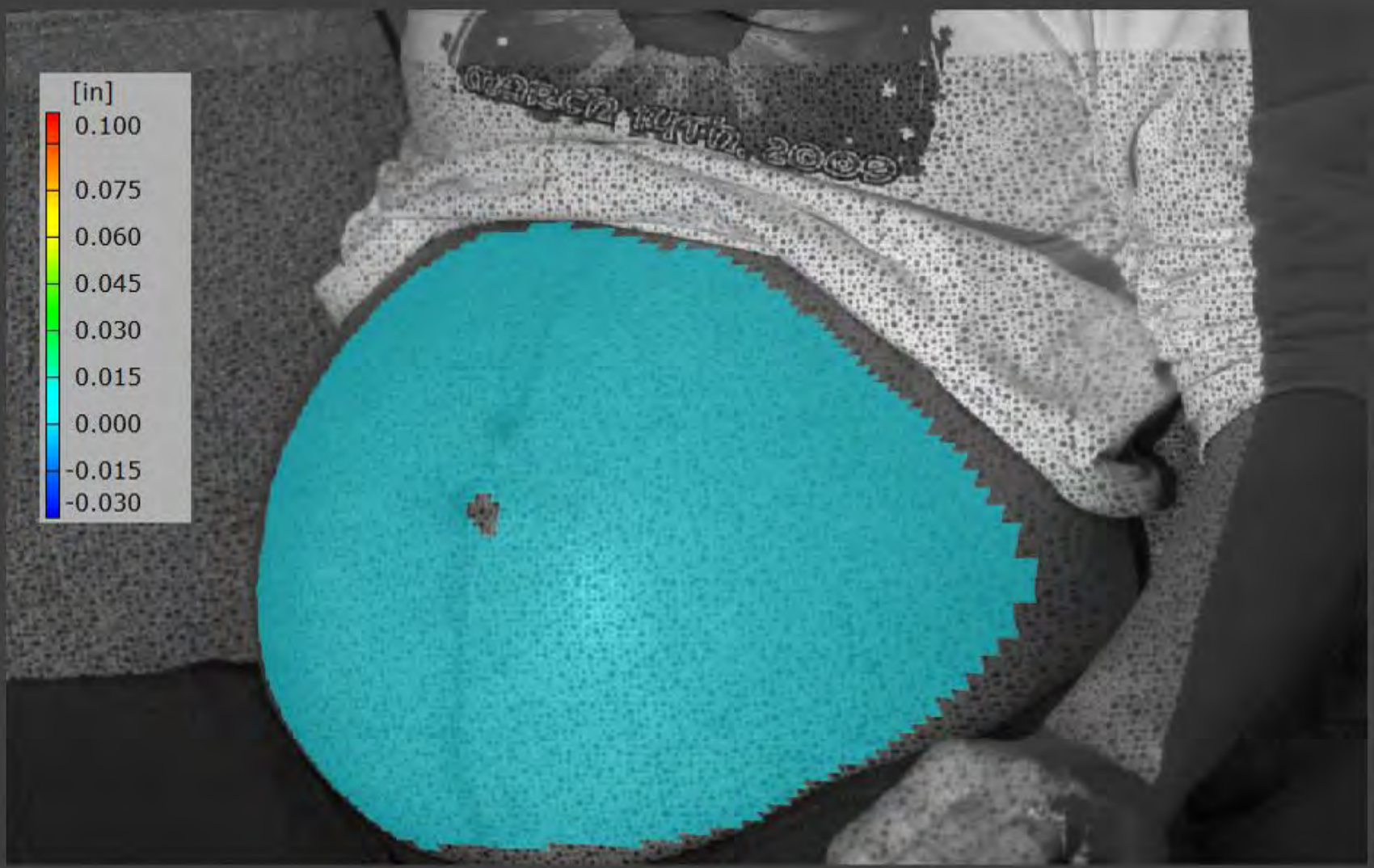
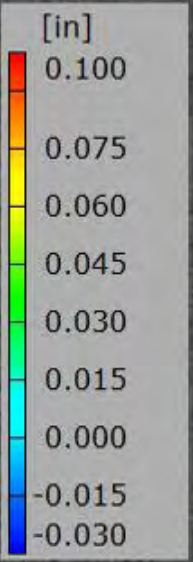
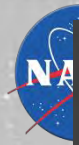




Time 0.00 seconds

 [Click to Play Video](#)

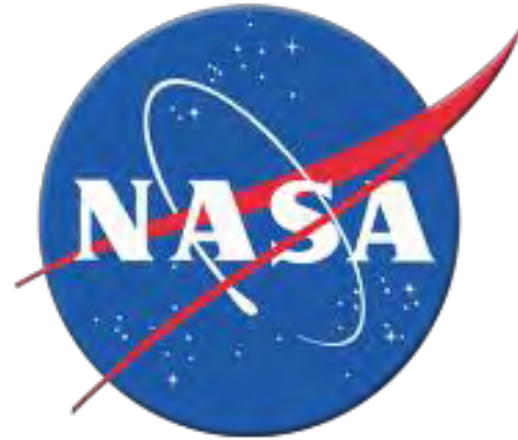
Displacement in Z Axis
Baby Gradl Movement - Shift to Right Side



↓ [Click to Play Video](#)

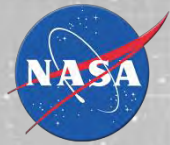
Time 0.00 seconds

**Displacement in Z Axis
Baby Gradl Movement - Baby Kicking**



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256.544.2455
Paul.R.Gradl@nasa.gov

BACKUP



References

Gradl, P.R. "Dynamic Measurements for Rocket Engine Hotfire Testing." Trillion Quality Systems Optical Metrology Workshop, Huntsville, AL on June 20, 2013

Gradl, P.R. "Application of Optical Measurement Techniques during Fabrication and Testing of Liquid Rocket Nozzles." Paper presented at 62nd JANNAF Propulsion Meeting/8th Liquid Propulsion Subcommittee, June 1, 2015. Nashville, TN

Gradl, P.R. "Digital Image Correlation Techniques Applied to Large Scale Rocket Engine Testing." To be presented at AIAA Joint Propulsion Conference, Salt Lake City, UT. July 2016.

Black, R., Gradl, P.R. "Resurrecting the F-1 – 3D Scanning to Digitally Capture the Saturn V Main Engine". Capture 3D Measurement Innovation 2012, August 23, 2012. Costa Mesa, CA.

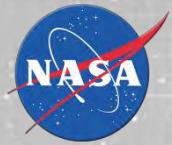
Gradl, P.R. "Application of Optical Measurement Techniques during Stages of Pregnancy". To be published in Summer 2016, TBD.

Kynard, M., Gradl, P.R. Town Hall Panel. "Where's My Apollo Vision for the Future?". Structured Light and D.I.C. presented to forum at AIAA Propulsion and Energy 2014, Cleveland, OH.

Cannon, J., Gradl, P.R. Status of Liquid Engines Optical Measurement Techniques presented to Integrated High Payoff Rocket Propulsion Technology (IHPRT). Presented September 2012, April 2013, March 2014.

Gradl, P. "Rapid Fabrication Techniques for Liquid Rocket Channel Wall Nozzles." AIAA-2016-4771, Paper presented at 52nd AIAA/SAE/ASEE Joint Propulsion Conference, July 27, 2016. Salt Lake City, UT.

Gradl, P. "Digital Image Correlation Techniques Applied to Large Scale Rocket Engine Testing." AIAA-2016-4977 Paper presented at 52nd AIAA/SAE/ASEE Joint Propulsion Conference, July 26, 2016. Salt Lake City, UT.

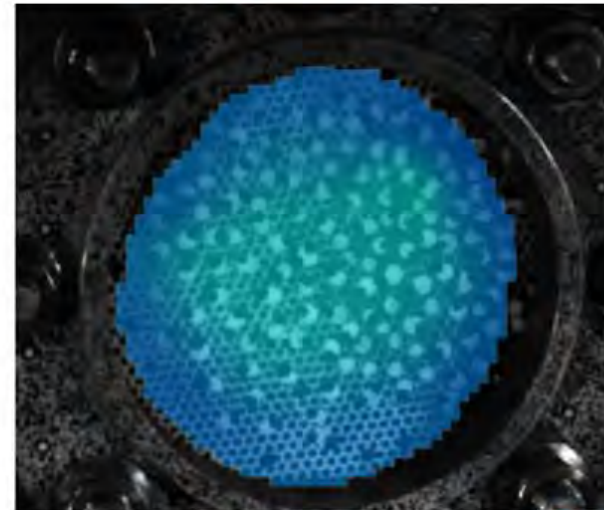
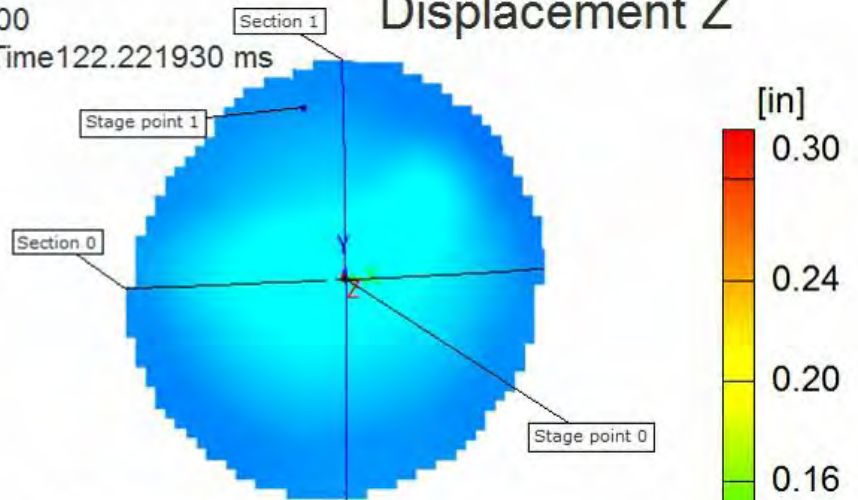
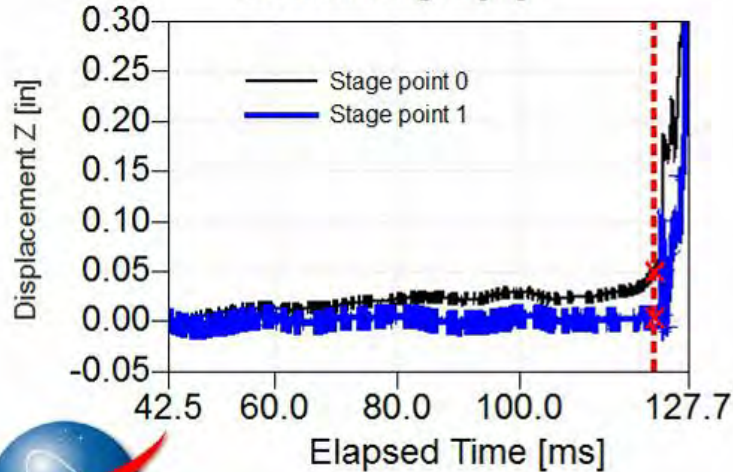
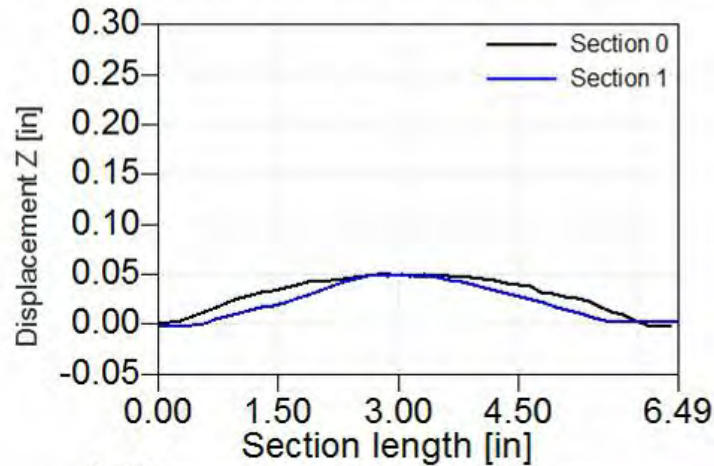


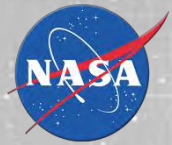
High Speed Fragmentation Testing

Test 91 April 3, 2013
300 SS 0.005" Half H2O

Stage 4300
Elapsed Time 122.221930 ms

Displacement Z

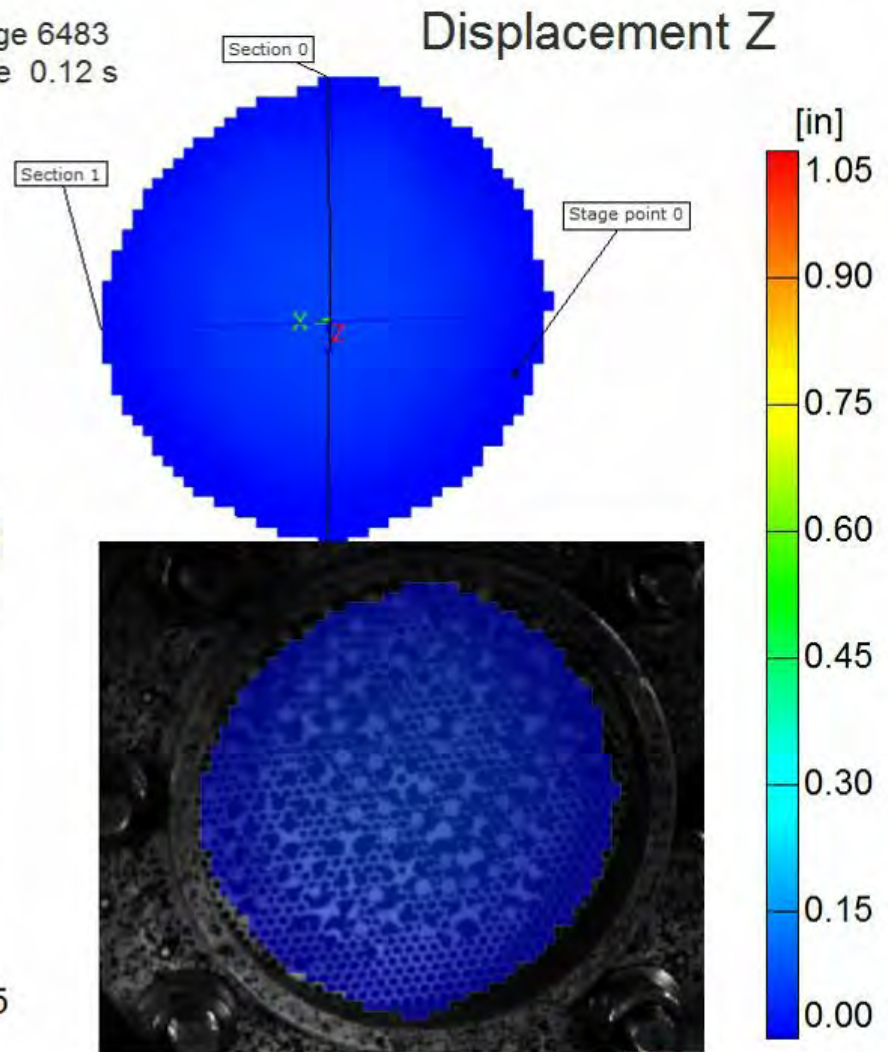
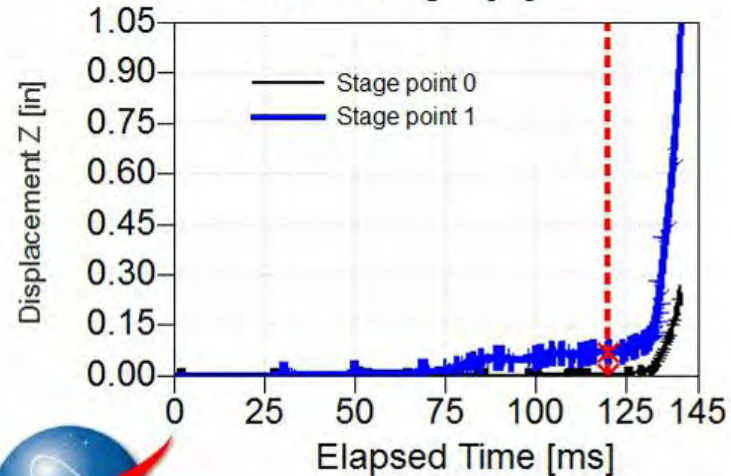
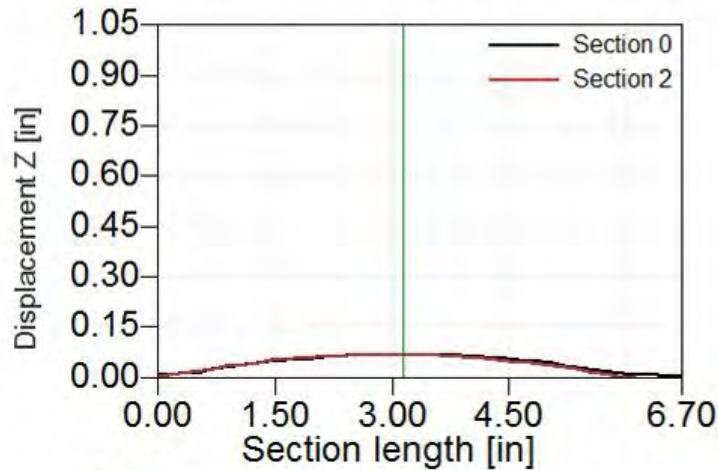


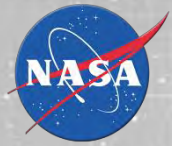


High Speed Fragmentation Testing (cont)

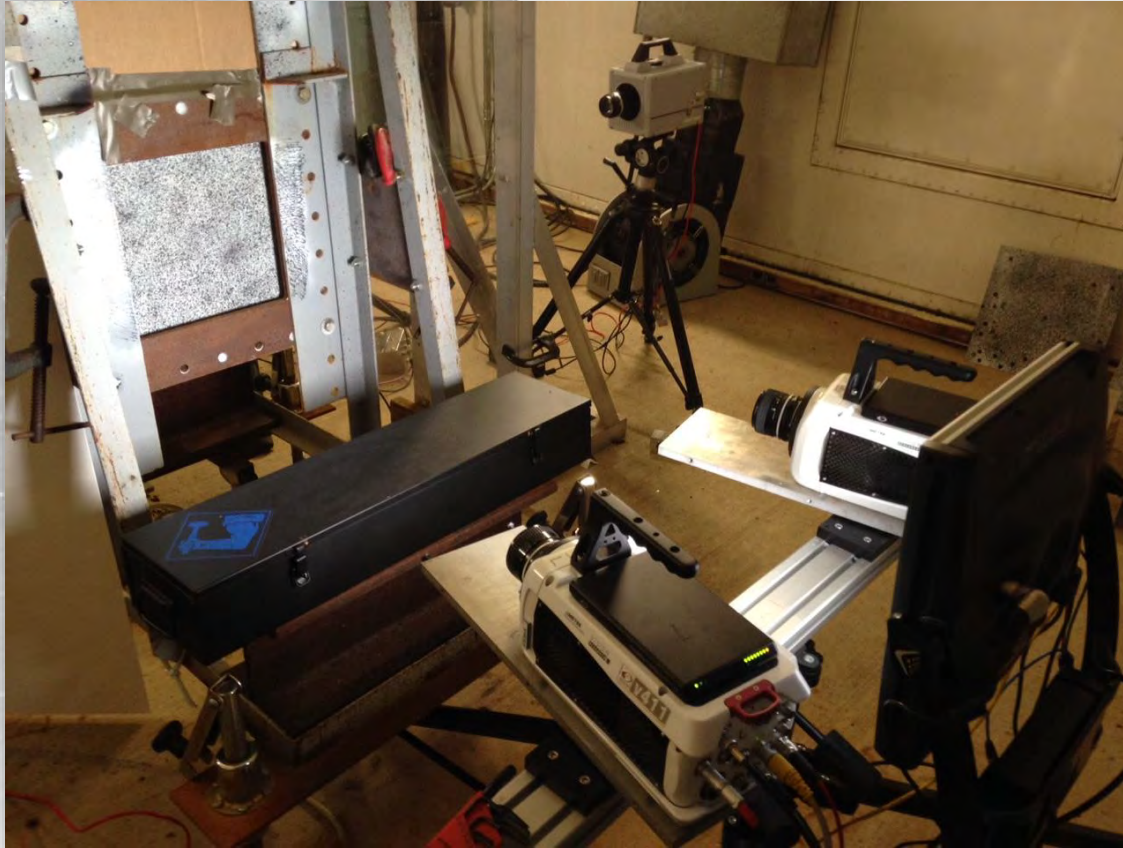
Test 93 April 3, 2013
300 SS 0.003" Full H2O

Stage 6483
Time 0.12 s

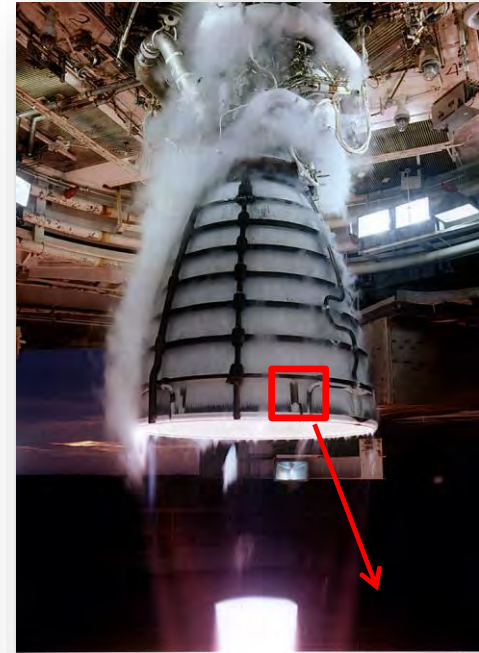


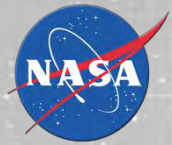


Space Launch System (SLS) Debris Impact Testing



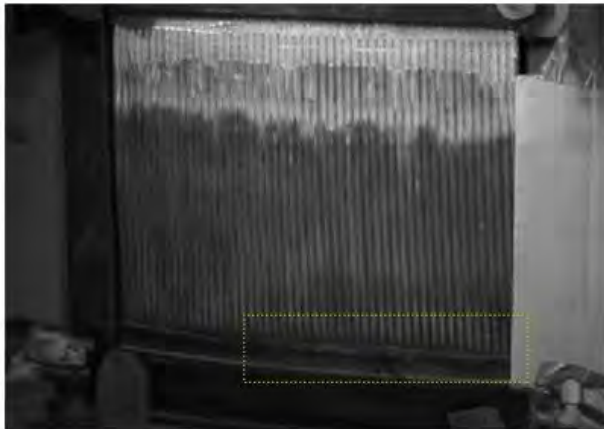
Test provided by: Paul Gradl and Cory Medina
Chip Kopicz, Perry Gray, Bart Suggs



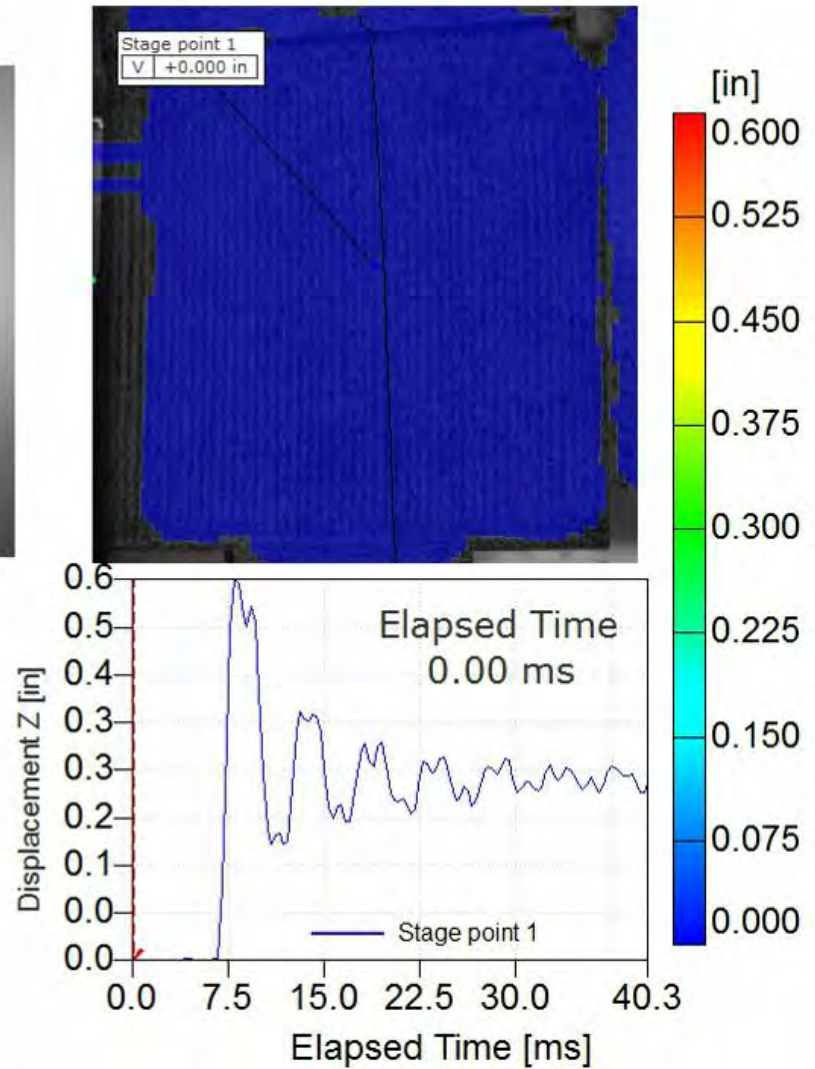


SLS RS25 Nozzle Pressurized Panel

SLS RS25 Pressurized Panel Testing 13 Nov 2015; 6# Foam



Displacement Z



Paul Gradl Cory Medina Chip Kopicz