

Microstructure Evolution and Durability of Advanced Environmental Barrier Coating Systems for SiC/SiC Ceramic Matrix Composites

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NASA EBC and CMC System Development



- Highly loaded EBC-CMCs
- 2700-3000°F (1482-1650°C) turbine and CMC combustor coatings
- 2700°F (1482°C) EBC bond coat technology for supporting next generation
 - Recession: <5 mg/cm² per 1000 h
 - Coating and component strength requirements: 15-30 ksi, or 100- 207 MPa





NASA Advanced Environmental Barrier Coating Technology Development



- Development Objective: Develop advanced 2700°F+ capable bond coat and EBC systems with high strength
- Approaches:
- Fundamental studies of environmental barrier coating materials and coating systems, stability, temperature limits and failure mechanisms
- Focus on high performance high stability patented cluster HfO₂ and ZrO₂ -RE₂O₃-SiO₂/RE₂Si_{2-x}O_{7-2x} environmental barrier systems
 - Controlled silica content and transition element and rare earth dopants to improve EBC stability and toughness
 - Significantly reduce diffusion, grain growth, mechanical strength and toughness with multicomponent systems
 - Develop HfO₂-Si based + X (dopants)
 - Develop prime-reliant Rare Earth Si alloys and composites for integrated EBC-bond coat systems
- Develop advanced NASA high toughness, compositions and processing
 - Achieving high toughness and erosion resistance
 - Achieving high stability and recession resistance
 - Improving the resistance to CMAS and Volcano ash deposits

Outline



- Advanced EBC and Rare Earth Silicon based 2700°F+ capable bond coat developments
 - Development approaches
 - Oxidation resistance
 - Cyclic and thermomecahnical durability
- Microstructural and phase composition evolution of an environmental barrier coating (EBC) system
 - Consisting of a multicomponent rare earth silicate EBC, along with YbGdSi 2700°F (1482°C) capable bond coat
 - Tested in high heat flux tensile rupture, fluxture fatigue, and furnace cyclic tests up to 500 hours at 2700°F (1482°C)
 - Examine Microstructure changes after thermomechanical and furnace cyclic testing
 - EDS WDS Composition analysis Comparisons
- Summary and conclusion



Development Strategy for 2700°F EBC Systems (US Patent Zhu and Miller, US 7740960; US Patent Application: Zhu & Hurst, LEW 18949, 2012; PCT/US13/46946, 2013)

- Modify silicon with rare earth, zirconium, hafnium dopants to increase its melting point and developing slower growing protective scales and refractory silicates
 Control oxygen content and EBC-CMC interface oxygen partial pressure for improved protection and stability
- Composite bond coat systems with refractory oxides and silicates to reinforce silicon containing bond coat matrix
- Composite bond coat systems with refractory oxide or silicate matrix, with silicon containing bond coat inter-phases
- Develop rare earth metal -, zirconium -, hafnium silicon systems and the silicide containing systems for bond coats with engineered grain boundary phases

National Aeronautics and Space Administration Advanced High Temperature and 2700°F+ Bond Coat Development



- NASA advanced Top Coat Development approach:
 - Advanced compositions ensuring high strength, high stability, high toughness
 - Bond coat systems for prime reliant EBCs; capable of self-healing



National Aeronautics and Space Administration Furnace Cycle Test Results of Selected RESi and ZrSi +



- Dopant Bond Coats
- Testing in Air at 1500°C, 1 hr cycles
- Some initial multi-component systems showed excellent furnace cyclic durability at 1500°C
- FCT and steam tests also performed for RESiO-Hf systems
- Generally good correlation between FCT and oxidation resistance



The Environmental Barrier Coating System

- Alternating layered HfO₂-Rare Earth silicate EBC for fundamental stability studies
- 2700F capable Yb-YbO based bond coat
- Coated onto SiC/SiC CMC substrates using EB-PVD





Environmental Barrier Coating System

- YbGdSi(O) (+Hf) Bond Coat + multi-component EBC Top Coat on woven SiC/SiC CVI-MI SiC/SiC CMC
- Creep testing conducted with 15 ksi load and laser thermal gradient at 0.15% total creep strain, bond coat at up to 2700°F (1482°C)



National Aeronautics and Space Administration The Flexural Fatigue Tested Environmental Barrier Coating Systems



- Strength and Fatigue cycles in laser heat flux rigs in tension, compression and bending
- Fatigue tests at 3 Hz, 2600-2700°F, stress ratio 0.05, surface tension-tension cycles





National Aeronautics and Space Administration SEM – EDS Analysis of the Tensile Ruptured Tested Environmental Barrier Coating System - Continued





National Aeronautics and Space Administration The Flexural Fatigue Tested Environmental Barrier Coating



System

- Ytterbium containing bond coat help self-healing the composite fatigue cracking



National Aeronautics and Space Administration The Flexural Fatigue Tested Environmental Barrier Coating



System - Continued

- Ytterbium containing bond coat help self-healing the composite fatigue cracking



The Long-Term Furnace Cyclic Tested Environmental Barrier Coating System: Rare Earth doped HfO₂ and Rare Earth Silicates Showed Compatibility and Stability





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EDS (Si Drift Detector) and WDS Comparisons Showed Good Agreements in the Composition Analysis

- 1500°C, 500 hr, 1 hr cycles, in air



Elt.	Line	Intensity (c/s)	Error 2-sig	Conc	Units
0	Ka	40.77	2.065	14.484	wt.%
Si	Ka	0.00	1.554	0.004	wt.%
Y	La	4.34	1.641	1.054	wt.%
Gd	La	6.01	1.312	6.290	wt.%
Yb	La	10.20	1.349	19.181	wt.%
Hf	La	25.25	1.730	58.987	wt.%
Total				100.000	wt.%



WDS 10 points to obtain the average composition of the L region

IMG52-L	PRO	DBE DATA-a	average 10	pts		
Elt.	Line	Spec-	Conc,	Units	Conc	Units
		Xtal	wt			
0	Ka	4-LDE1	13.24	wt.%	62.4	at%
Si	Ka	2-PETJ	0.036	wt.%	0.098	at%
Y	La	2-PETJ	2.093	wt.%	1.78	at%
Gd	La	5-LIFH	4.9	wt.%	2.35	at%
Yb	La	5-LIFH	15.59	wt.%	6.8	at%
Hf	Ma	2-PETJ	62.796	wt.%	26.57	at%
Total			98.66	wt.%	100	at%



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EDS (Si Drifting Detector) and WDS Comparisons Showed Good Agreements in the Composition Analysis - Continued - 1500°C, 500 hr, 1 hr cycles, in air



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-	NASA SEI	15.0kV	×1,500	10µm WD11mm

WDS 10 points to obtain the average composition of the N region

Elt.	Line	Intensity (c/s)	Error 2-sig	Conc	Units
0	Ка	62.15	2.382	19.238	wt.%
Si	Ka	82.63	2.975	7.027	wt.%
Y	La	14.54	1.749	5.614	wt.%
Gd	La	12.38	1.364	19.333	wt.%
Yb	La	17.14	1.480	48.788	wt.%
Hf	La	0.00	0.831	0.000	wt.%
Total				100.000	wt.%

IMG52-	N	PROBE DATA-average 10pts					
Elt.	Line	Spec- Xtal	Conc	Units	Conc	Units	
0	Ka	4-LDE1	21.67	wt.%	60.25	at%	
Si	Ka	2-PETJ	13.25	wt.%	20.997	at%	
Y	La	2-PETJ	6.2	wt.%	3.11	at%	
Gd	La	5-LIFH	15.37	wt.%	4.35	at%	
Yb	La	5-LIFH	42.019	wt.%	10.81	at%	
Hf	Ма	2-PETJ	0.893	wt.%	0.223	at%	
Total			99.46	wt.%	100	at%	



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EDS (Si Drifting Detector) and WDS Comparisons Showed Good Agreements in the Composition Analysis - Continued - 1500°C, 500 hr, 1 hr cycles, in air

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P		J r(egion	
65				
	NASA SEI	15.0kV	×4,000	1µm WD11mm

WDS 4 points to obtain the average composition of the N region

IMG45-J	PROBE DATA-average 4pts						
Elt.	Line	Spec- Xtal	Conc	Units	Conc	Units	
В	Ka	1-LDE2	30.46	wt.%	43.34	at%	
С	Ka	1-LDE2	0.015	wt.%	0.018	at%	
N	Ka	1-LDE2	24.47	wt.%	27.01	at%	
0	Ka	4-LDE1	13.74	wt.%	13.72	at%	
Si	Ka	2-PETJ	27.77	wt.%	15.59	at%	
Y	La	2-PETJ	0.01	wt.%	0.0019	at%	
Gd	La	5-LIFH	0.009	wt.%	0.0009	at%	
Yb	La	5-LIFH	0.0015	wt.%	0.0001	at%	
Hf	Ма	2-PETJ	3.33	wt.%	0.33	at%	
Total			99.8	wt.%	100	at%	

⊑п.	Line	mensity	EIIOI	CONC	Units
		(c/s)	2-sig		
В	Ka	19.68	1.340	0.300	wt.%
С	Ka	2.70	0.848	0.264	wt.%
Ν	Ka	31.68	1.669	21.438	wt.%
0	Ka	1.33	0.646	44.318	wt.%
Si	Ka	295.83	4.950	25.893	wt.%
Y	La	0.00	0.824	0.000	wt.%
Gd	La	0.14	0.417	0.415	wt.%
Yb	La	0.40	0.383	4.435	wt.%
Hf	La	0.41	0.393	2.938	wt.%
Total				100.000	wt.%





Summary and Conclusions

- Environmental barrier coatings with YbGd-HfSi(O) based bond coat, and HfO₂ multicomponent Rare Earth silicate top coat were tested on SiC/SiC ceramic matrix composites for initial durability studies
- The coatings showed excellent oxidation resistance in O₂ and air testing environments, adhesion and good protection for SiC/SiC CMCs
- The initial silicon content range of the Rare Earth-Silicon coatings was down-selected, multicomponent systems designed and demonstrated for further improved stability
- The rare earth silicon based coatings showed 2700°F or 1500°C operating temperature viability and durability on SiC/SiC ceramic matrix composites, the EBC-CMC system microstructure and phase changes were investigated
- The rare earths, hafnium and silica showed wide range solubility, and composition ranges of EBC materials are being optimized for coating stability and performance
- The extensive studies of the EDS and WDS composition analyses of the EBC system showed good agreements
 - WDS may be more sensitive to light elements;
 - Field emission gun SEM Silicon drift detector EDS has spatial resolution advantages



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