

# Creep, fatigue and environmental interactions and their effect on crack growth in superalloys

J. Telesman, T.P. Gabb, L.J. Ghosn, T. Smith

NASA Glenn Research Center - Cleveland, OH

TMS 2017 Annual Conference San Diego March 1, 2017

Research funded by NASA's Aeronautics Research Mission Directorate - Advanced Air Transportation Technology Project – Jim Walker Program Manager



#### **Presentation Overview**

- Complex interactions of creep/fatigue/environment control dwell fatigue crack growth (DFCG) in superalloys.
- Crack tip stress relaxation during dwells significantly changes the crack driving force and influence DFCG.
- Linear Elastic Fracture Mechanics, Kmax, parameter unsuitable for correlating DFCG behavior due to extensive visco-plastic deformation.
- Magnitude of remaining crack tip axial stresses controls DFCG resistance due to the brittle-intergranular nature of the crack growth process.
- Proposed a new empirical parameter, Ksrf, which incorporates viscoplastic evolution of the magnitude of remaining crack tip stresses.
- Previous work performed at 704°C, extend the work to 760°C.

## **Experimental**

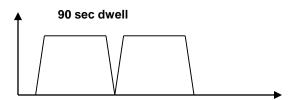


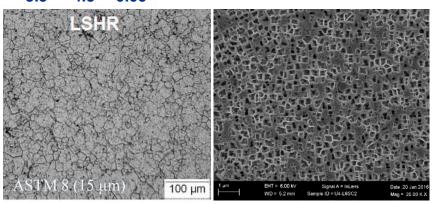
## Material: Low Solvus High Refractory (LSHR) P/M nickel-base disk alloy

Wt. % Al B C Co Cr Mo Ni Nb Ta Ti W Zr LSHR 3.5 .03 .045 20.4 12.3 2.7 Bal. 1.5 1.5 3.5 4.3 0.05

#### **Four Supersolvus Heat Treatments Evaluated**

. our supersorrus ricut ricutinents Etarautea			
	<b>Cooling Rate</b>	Aging	Thermal
Condition	(°C/min)	Treatment	Exposure
		855°C/4 h	
FC+2SA	202°C/min	+775°C/8h	None
		855°C/4 h	
SC+2SA	72°C/min	+775°C/8h	None
		855°C/4 h	
FC+2SA+440	202°C/min	+775°C/8h	815°C-440 h
		855°C/4 h	
SC+2SA+440	72°C/min	+775°C/8h	815°C-440 h





#### Testing performed at 704 °C and 760°C

#### **Baseline FCG Testing:**

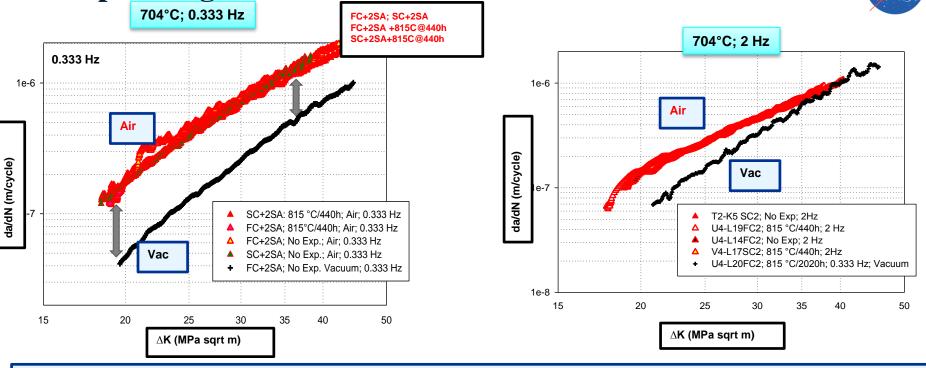
- Cyclic FCG in Air and Vacuum; 0.333 to 30 Hz
- Dwell FCG in Air and Vacuum; 90 sec hold at  $\sigma_{max}$
- Specimen Geometry: Surface Flaw (KB bar)

#### **Baseline Stress Relaxation Testing:**

- Strained to 1% total strain
- Stress relaxation measured for 100 h.
- Specimen: Cylindrical (4.05 mm diam.)

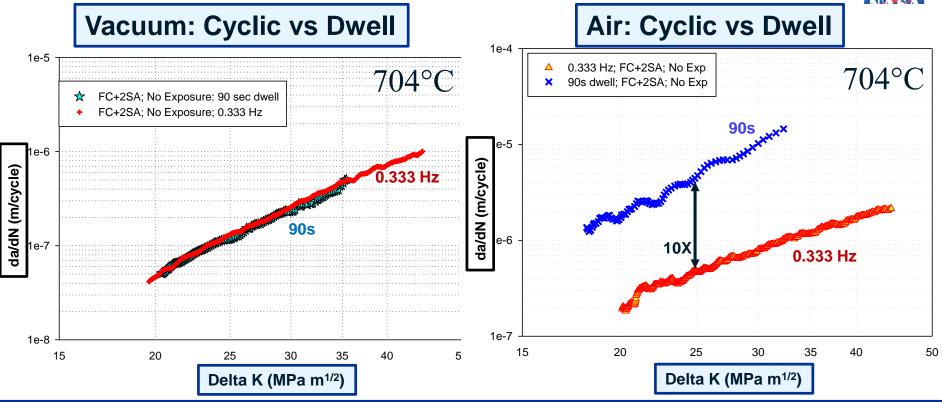
Separating Environmental Effects from Stress Relaxation





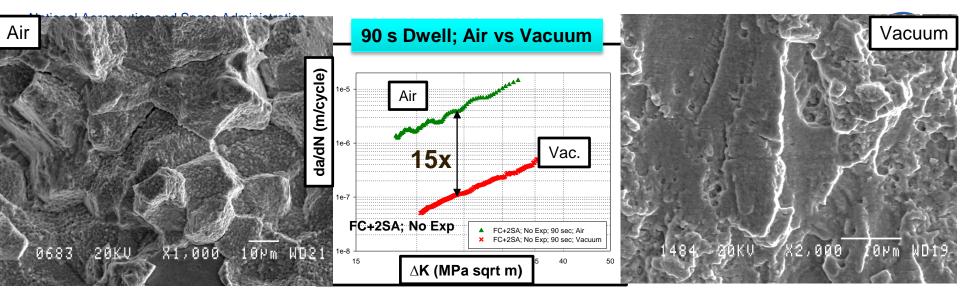
- All four conditions show faster cyclic FCGR in air than in vacuum environmental debit.
- All conditions exhibited similar FCG resistance behavior but environmental effect is smaller at higher frequency.
- Assume these conditions posses similar *intrinsic* environmental resistance.
- Any differences in their dwell FCG resistance are then due to stress relaxation effects.



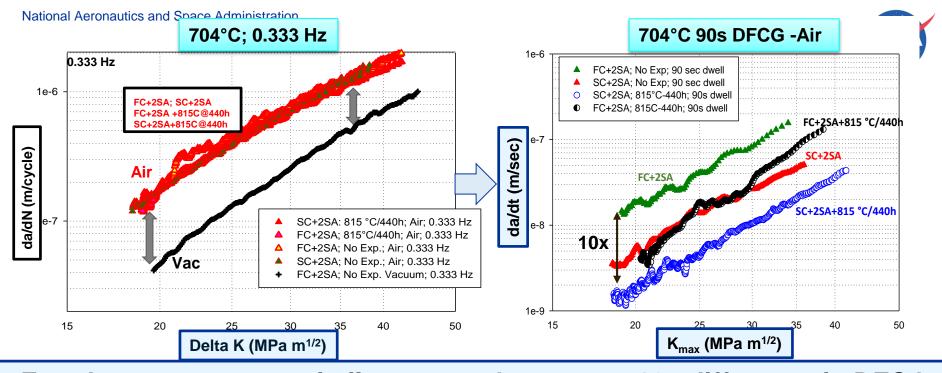


### At 704°C:

- 90 sec dwell FCG rates in vacuum same as cyclic FCG in vacuum No Dwell <u>Debit</u>
- Creep crack growth does not contribute towards dwell crack growth
- An order of magnitude increase in DFCG in air due to environmental damage

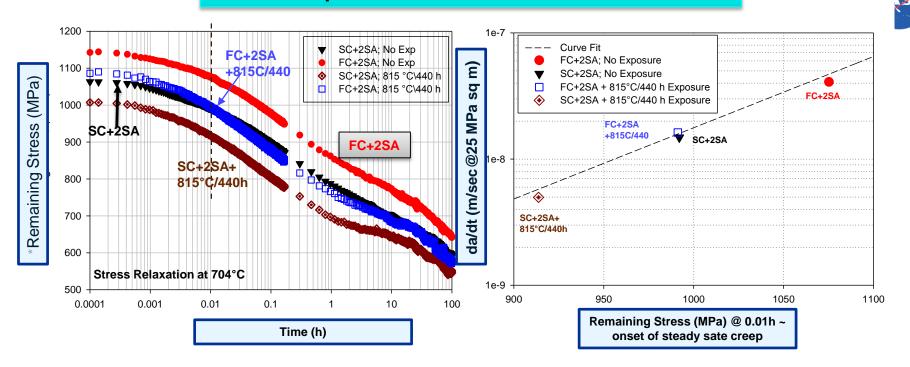


- Brittle-intergranular failure mode was operative in air for 90 sec dwell.
- Only transgranular failure mode operative in vacuum. No evidence of grain boundary sliding or microvoid coalescence found (classical creep crack growth did not directly contribute to DFCG).
- Grain boundaries are strong! Cracks avoid growth along grain boundaries when environmental embrittlement or creep mechanisms are <u>not</u> operative.



- Four heat treatments: similar env. resistance  $\rightarrow$  10x difference in DFCG.
- "Creepier heat treatments" i.e. slower cooling rates and thermal exposures improve DFCG resistance.
- Environmental resistance similar DFCG differences due to stress relaxation.
- LEFM Kmax parameter unsuitable for correlating visco-plastic influenced DFCG response.

#### **Relationship Between Stress Relaxation and DFCG**

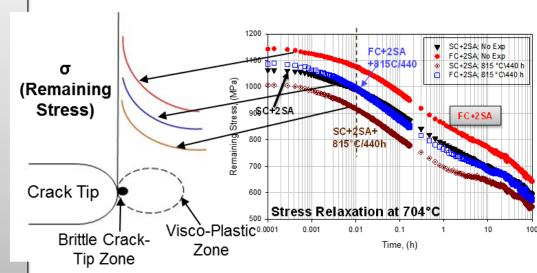


- Stress relaxation stresses decrease with slower cooling rates and ↑thermal exposure.
- Remaining stresses closely correlate with dwell fatigue crack growth Yet... Classical creep propagation mechanisms DO NOT contribute to crack growth
- Why is magnitude of remaining stresses important? What governs the relationship?

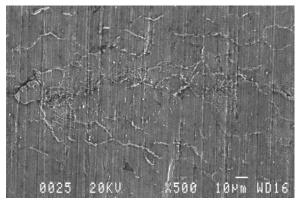
## **DFCG Failure Mechanism**

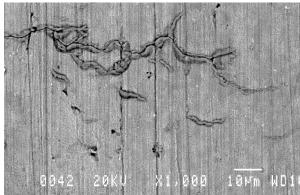
NASA

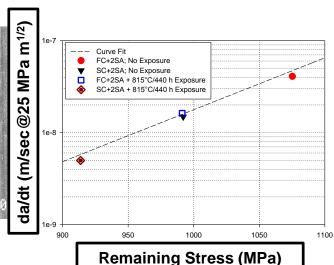
- Cracks grow through brittle-intergranular process controlled by crack tip tensile stress
- Magnitude of crack tip tensile stress controls DFCG propagation rates
- Stress relaxation behavior sets the magnitude of crack tip tensile stresses
- Strong, yet indirect relationship between stress relaxation and DFCG behavior



#### Embrittled crack tip region – Interrupted 90s dwell tests



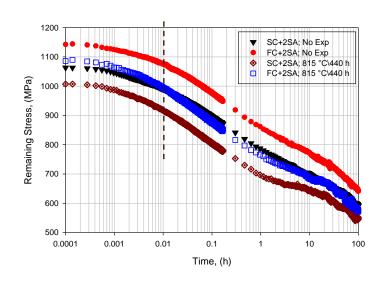




## New Empirical Parameter for Modeling Dwell Crack Growth in Air



Approach: Use stress relaxation results to simulate and normalize the differences in the crack tip tensile stresses under visco-plastic conditions



$$Ksrf = Kmax/SRF$$

 $K_{srf}$  - modified stress intensity factor normalized by SRF

 $K_{max}$  – Applied LEFM stress intensity factor during dwells

$$SRF = (\sigma_0/\sigma_m)^4$$

SRF= stress relaxation factor

 $\sigma_0$  = remaining stress at the onset of steady state creep (highest remaining stress condition)

 $\sigma_m$ = remaining stress for other conditions – onset of steady state creep

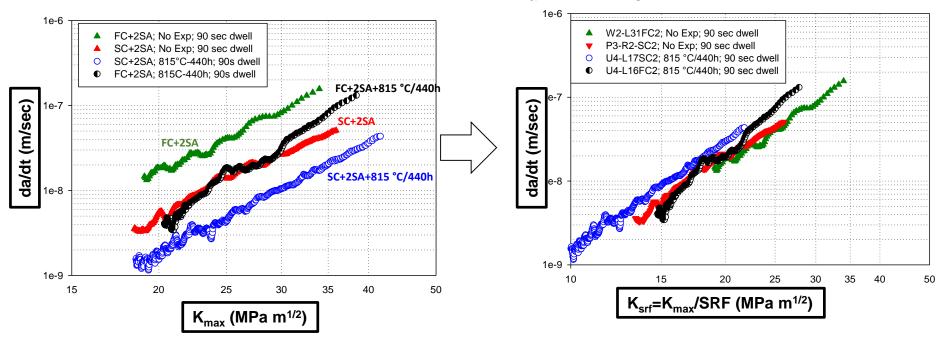
$$\dot{\varepsilon} = A\sigma^{n1}t^m + B\sigma^{n2}$$

n2=4 (steady state creep component per the relaxation fit)

## New Empirical Parameter for Modeling Dwell Crack Growth in Air



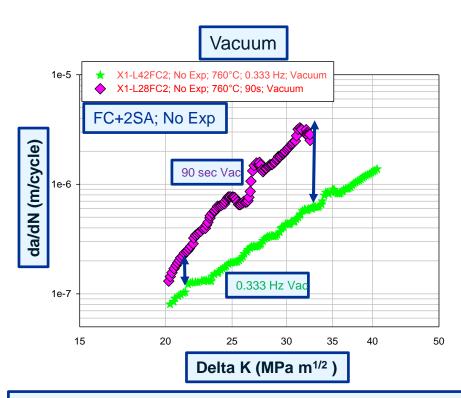
## Plots of da/dt vs $K_{max}$ and $K_{srf}$ : 704°C



• New Ksrf parameter able to compensate for a 10x spread in DFCG rates using standard LEFM parameter.



#### DFCG in Vacuum and Air – Behavior at 760°C



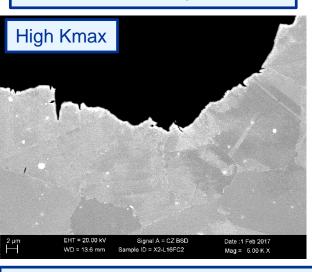
- Vacuum and Air X1-L14FC2; No Exp; 760°C; 90s; Air FC+2SA; No Exp X1-L42FC2; No Exp; 760°C; 0.333 Hz; Vacuum X1-L28FC2; No Exp; 760°C; 90s; Vacuum 90 sec Air da/dN (m/cycle) 100x 0.333 Hz Va 1e-7 15 50 Delta K (MPa m<sup>1/2</sup>)
- In contrast to 704°C, creep crack growth occurs in vacuum at 760°C
- More significant at high  $\Delta K \rightarrow$  increase in crack tip plasticity
- DFCG in air 100x faster than in vacuum.
- Environmental degradation is predominant.... Ksrf approach may still be applicable.

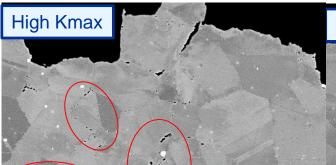
# Comparison of Microstructural Damage Mechanisms

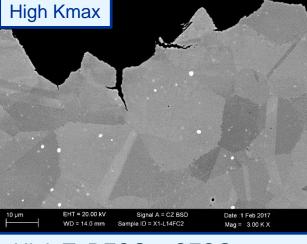
704°C – 90 sec; Vacuum

760°C – 90 sec; Vacuum

760°C - 90 sec; Air







- •Lower T.; DFCG=Cyc FCG;
- •No evidence of creep damage
- Transgranular failure mode

•High Temp; DFCG > CFCG

Sample ID = X1-L28FC2

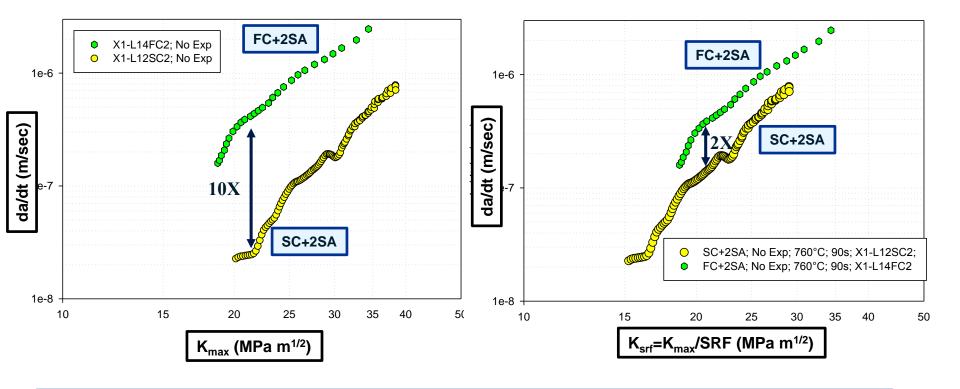
Mag = 3.00 K X

- Slow DFCG
- •Grain boundary creep cavitation

- •High T; DFCG>>CFCG
- Fast DFCG
- •Environmentally induced intergr. failure mode dominant
- Inadequate time for creep cavitation



# 760°C DFCG; K<sub>max</sub> vs K<sub>srf</sub>



- Identical methodology used to calculate K<sub>srf</sub> as at 704°C
- K<sub>srf</sub> correlated DFCG within 2X for FC+2SA and SC+2SA (no exposures)
- Other two conditions experienced likely specimen mixup at the vendor...No agreement between stress relaxation repeats, sorting out the issues...



## **Conclusions**

- A new empirical parameter, Ksrf, proposed to correlate DFCG in superalloys.
- The new parameter modifies LEFM\_Kmax parameter by accounting for differences in visco-plastic evolution of the magnitude of remaining crack tip axial stresses.
- Magnitude of remaining crack tip axial stresses controls DFCG resistance due to the brittle-intergranular nature of the crack growth process.
- The parameter works well at 704°C and looks promising even at 760°C.
- Creep crack growth mechanisms are active at 760°C but are still considerably lower than the environmentally induced DFCG debit.



## **Acknowledgment**

Special acknowledgment to Mr. Andrew Ring for his excellence and diligence in performing a multitude of complex mechanical testing.