

A composite space image featuring Earth in the upper left, the Moon in the center, Mars in the lower center, Jupiter in the lower right, a comet streaking across the upper right, a satellite in the upper center, and a galaxy in the top right. The background is a dark starry field with a bright orange and yellow nebula-like glow.

Optical Performance Analyses for the 1-meter Optical Telescope for NASA GHAPS

Brian Catanzaro, Thomas Brooks, Bob Woodruff, Brian O'Connor, Will Johnson, Adam Burt

May 18, 2017





GHAPS

Gondola for High-Altitude Planetary Science

- **Definition**

- Planetary Science Observatory
- Stratospheric Balloon Platform
- Shared / Competed Resource available for Exchanging Instruments

- **History**

- Build Off of Experiences on BRRISON and BOPPS

- **GHAPS Goals**

- Support Science Outlined in Planetary Science Decadal Survey
 - NRC 2011
- Access to Wavelengths Inaccessible from
 - Ground-Based and Airborne Facilities
- Observe Science of Extended Periods of Time
- High Spatial Resolution at UV / Visible
- High Spectral Resolution at UV to IR

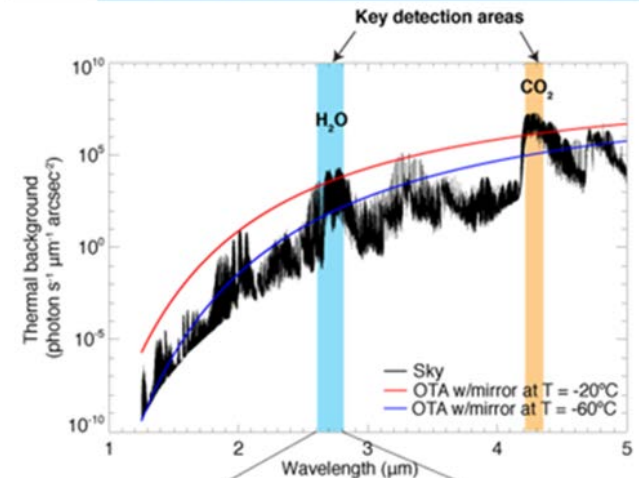


GHAPS Science Capability

- **GHAPS is a first generation platform optimized for multiple long duration flights and for planetary science**
- **IR observation design supports detection of water and carbon dioxide.**
- **Long duration flights enable temporal science not practically possible any other way**
 - Study Jupiter storms, Venus clouds and super rotation, methane or water cycles on Mars or Moon, volcanic tracking, atmospheric SO₂, Volatiles/organics (in comets, asteroids and Mercury), and more.
- **GHAPS is expected to evolve over time with science demands**



IR observation signal to noise



GHAPS will provide a re-usable platform for decadal class planetary science.



GHAPS Overview

GHAPS is a Class D, GLPR 7120.5.10A Silver Project

- Develop a Re-useable Balloon Platform to meet Planetary Science Goals and Objectives as outlined in Decadal Survey
- 1 meter Optical Telescope Assembly (OTA) with Sub-arc-second pointing capability
- Designed for a minimum of 5 flights from Balloon Program Office (BPO) launch locations
- Designed for mission durations up to 100 days
- Planetary Science Observations 300-5000 km
- Low cost refurbishment (1 yr) between flights
- The first flight is planned for Fort Sumner, New Mexico in the fall of **2020**
- The objective of the first flight is to demonstrate performance and conduct science observations
- A competitive process will be used to select investigators and the GHAPS Instrument Suite

GHAPS Gondola & Payload

Solar
Pointing

Balloon
System



Arc - Second
Pointer
(WASP)
(WFF)

Instruments

Balloon
System
Power
(BPO)



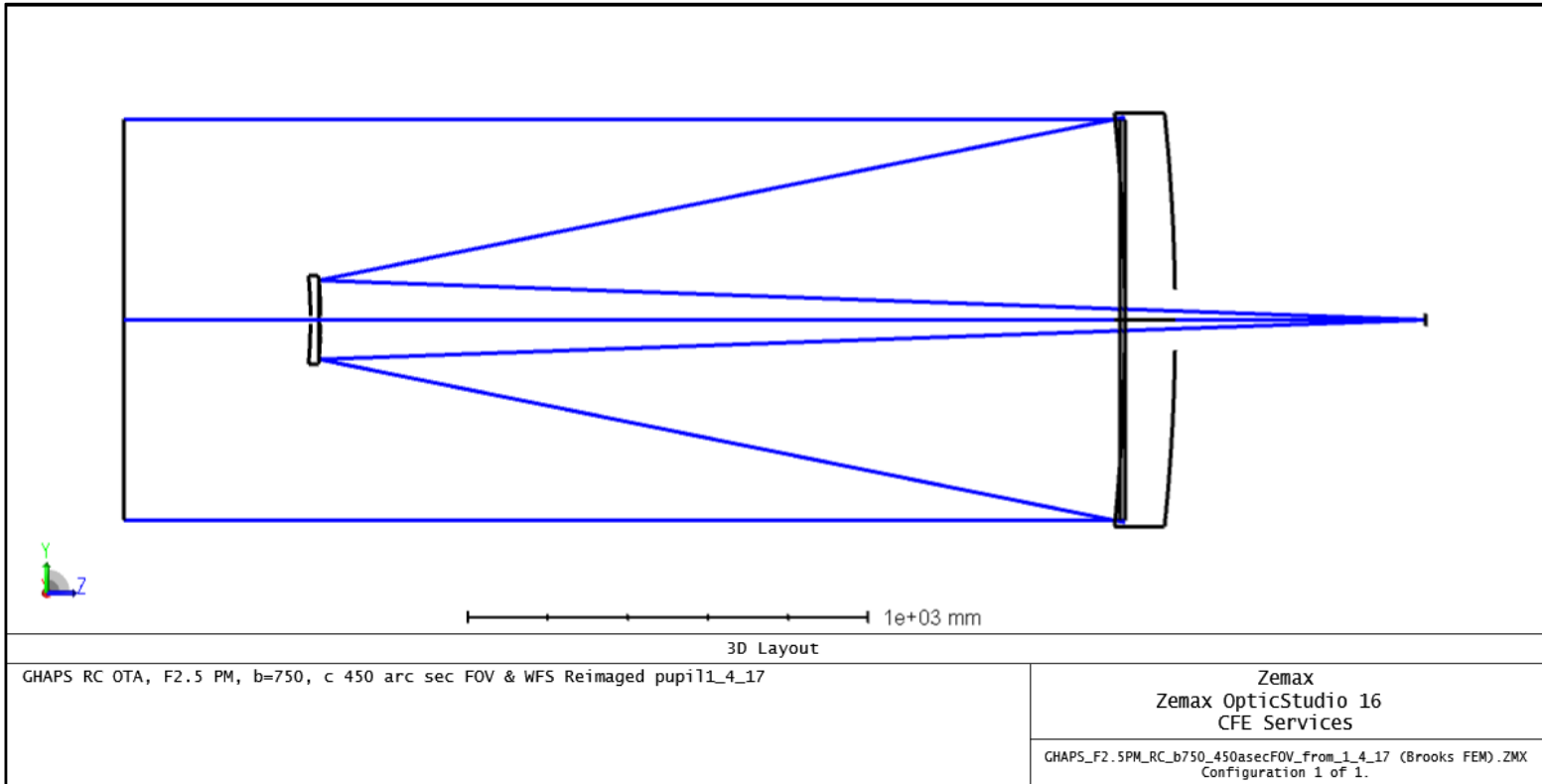
Summary of Optical Requirements

- **Aperture: 1-m**
- **Optical Quality**
 - Strehl > 80% @ 500 nm
 - FWHM < 0.12 arc-sec = (x1) Airy Radius @ 500 nm
 - WFE > 26.6 nm RMS @ 500 nm
- **FoV: > 1 arc-min Dia. @ Diffraction Limit**
 - Total FoV > 450 arc-sec [+7.5 arc-min]
- **Pointing**
 - Pointing Bias < 1 arc-sec @ 10 min
 - Jitter < 0.062 arc-sec
- **Wavebands**
 - UV / Vis (Supported by Resolution) = 300 nm to 1000 nm
 - IR (Supported by Low Emissivity) = 1 μ m to 5 μ m



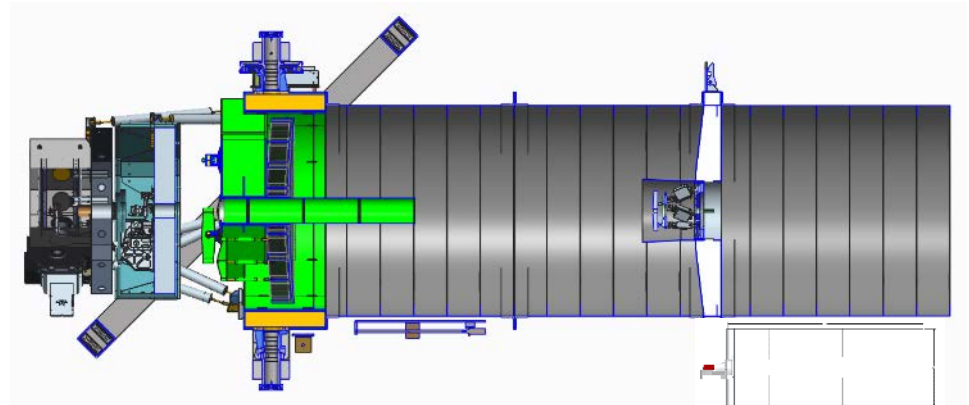
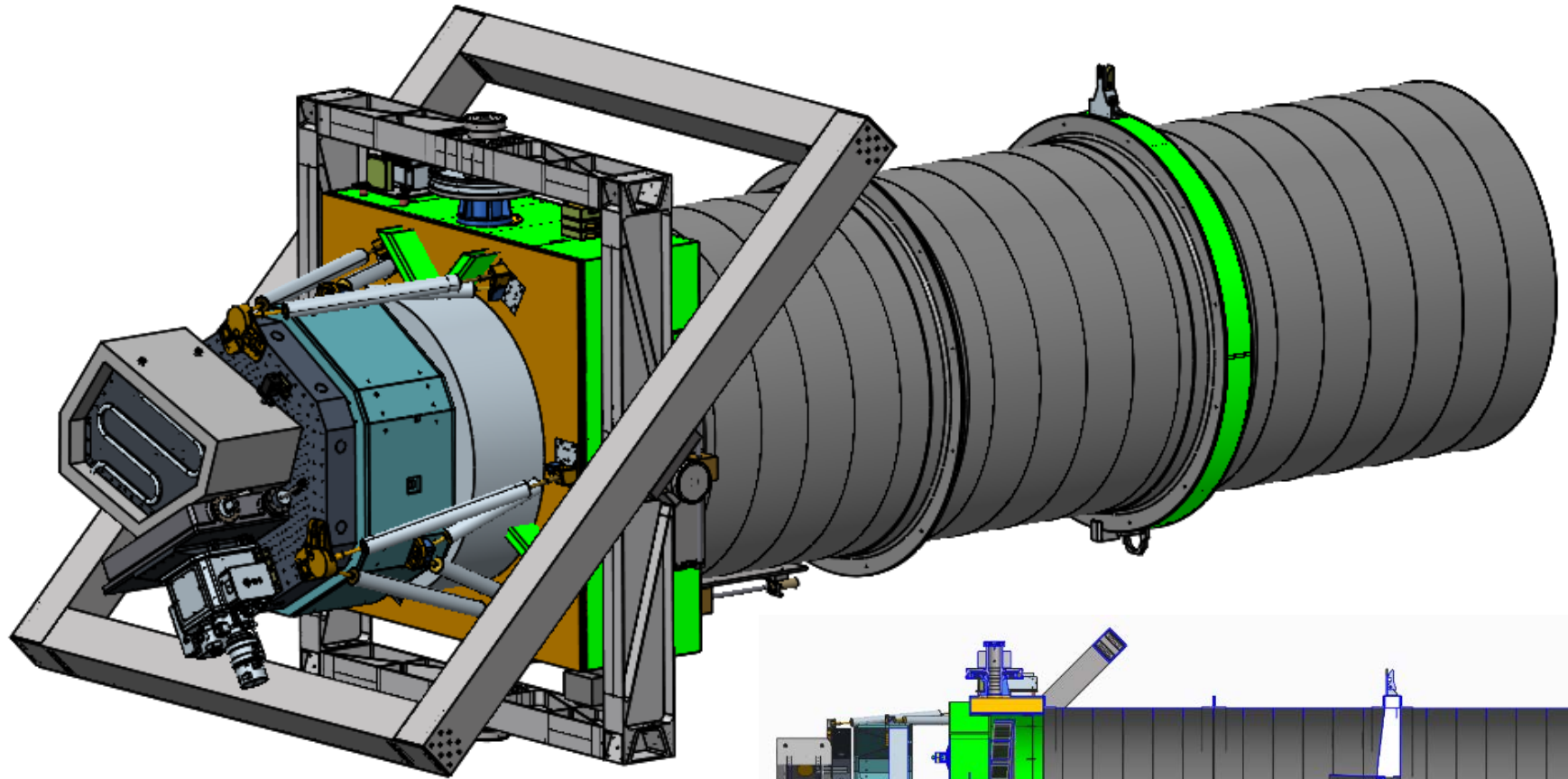
OTA Design

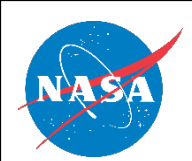
- **Two Mirror / Ritchey-Chretien**
 - F/14, D = 1 m, BFL = 0.75 m
- **Moveable Secondary Mirror**
 - Hexapod to Correct Aberrations on Float from Gravity / Thermal
 - Controlled by Wavefront Sensor





OTA Design





Structural, Thermal, Optical Performance

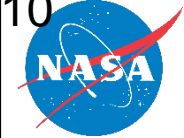
STOP ANALYSIS



STOP Analysis / Definition

- **Structural Thermal Optical Performance**
- **Optical Systems are Sensitive to Misalignment**
 - Displacements < 50 microns
 - Rotations < 5 arc-sec
- **Subtle Changes in Conditions can Impact Performance**
 - Stiffness
 - Elevation Changes in Gravity Field
 - Vibration from Instruments, Reaction Wheels, Pointing System
 - Thermo-Elastic Deformation
 - Thermal Soak / Variety of Materials, Differential CTE
 - Thermal Gradient / On-Float Environment, Solar, Earth-shine, ...

**How is All This Incorporated into
Design and Operations?**



Long, Rich Heritage in Integrated Modeling

Integrated Modeling Applied to the Terrestrial Planet Finder Mission

Andrew Kissil^a, Eug Kwack^a, Timothy Ho^a, Philip Dumont^a, Sandra Irish^b, Ichung Weng^c
^aJet Propulsion Laboratory, 4800 Oak Grove Drive, Pasadena, CA 91109, USA; ^bBall Aerospace & Technologies Corp., 1600 Commerce St., Boulder, CO, 80501, USA; ^cNASA ZP10, MSFC, Huntsville, AL, USA 35812

IMOS 2005

Integrated Structural and Optical Modeling of the Orbiting Stellar Interferometer

S. Shaklan, J. Yu, and H.C. Graber
 Jet Propulsion Laboratory
 California Institute of Technology
 4800 Oak Grove Drive
 M/S T1701
 Pasadena, CA 91109

Next generation lightweight mirror modeling software

William R. Arnold, Sr.^a, Matthew Fitzgerald^b, Rubin Jaca Rosa^b, H. Philip Stahl^c
^aDefense Acquisition, Inc., Jacobs ESSSA Group, Huntsville, AL, USA 35806-0001; ^bNASA Intern, MSFC, Huntsville, AL, USA 35812; ^cNASA ZP10, MSFC, Huntsville, AL, USA 35812

2013

IMOS 19

Development of a validated end-to-end model for space-based lidar systems

Mike Lieber[#], Carl Weimer, Michelle Stephens, and Ray Ball
 Ball Aerospace & Technologies Corp, 1600 Commerce St, Boulder, CO, 80501

EOSyM 2007

Advancements in Integrated Structural/Thermal/Optical (STO) Analysis of Optical Systems.

Gerhard Stoeckel, David Crompton, Gerard Perrin

Optical modeling activities for NASA's James Webb Space Telescope (JWST): VI. Secondary Mirror Figure and Tertiary Mirror Segment Motions

Lee D. Feinberg

2009

Automated Design Tools for Biophotonic Systems

Giacomo Vacca^a, Hannu Lehtimäki^b, Tapio Karras^c, Sean Murphy^d
^aKinetic River Corp., 661 S. Baywood Ave., San Jose, CA 95128; ^bHietalahdenranta 5 c A 6, FI-00120 Helsinki, Finland; ^cDesign Parameters, Inc., 1000 N. Milpitas Blvd., Suite #106, San Jose, CA 95129, USA; ^dSKMurphy Inc., 494 Chinaberry

BeamWise 2014

Integrated

telescope

R.W. Besuner¹, M.J. Sholl², M.D. Lieber³, M.L. Kaplan³,

¹Lawrence Berkeley National Laboratory

²University of California, Berkeley

³Ball Aerospace & Technologies Corporation

EOSy

Victor Genberg, Gregory Michels
 Sigmadyne, Inc. Rochester, NY

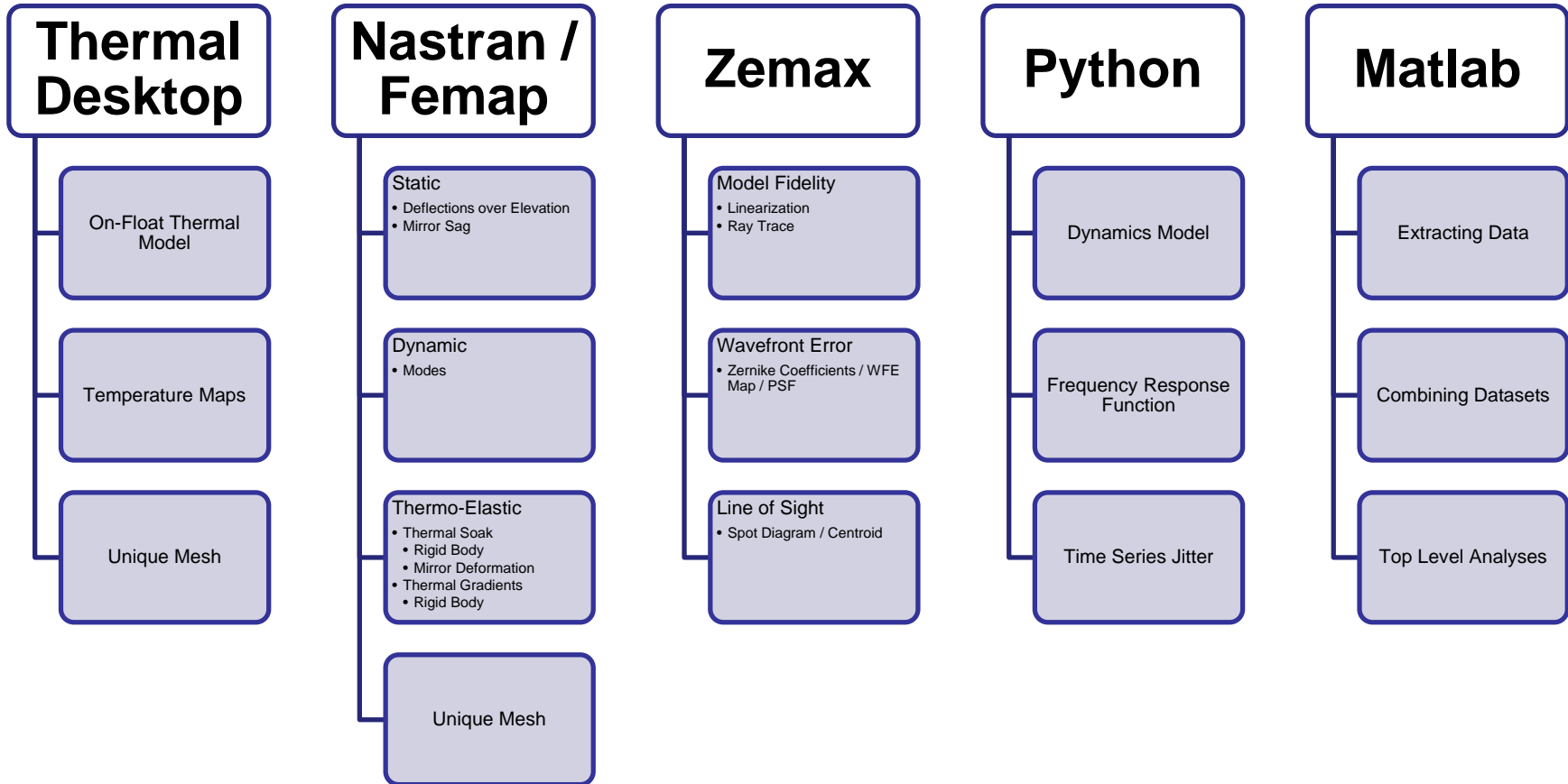
Keith Doyle

Optical Research Associates, Westborough, MA

SigFit 2002



STOP Analysis Tools for GHAPS





Cases Presented

- **Gravity**
 - Misalignment Due to Elevation
 - WFE / Mirror Deformation, Rigid Body Motion of Mirrors
 - LoS / Rigid Body Motion of Mirrors and Instruments
 - Polishing Conditions
 - AI/T Conditions
- **Temperature**
 - Soak
 - Rigid Body Motion
 - WFE / CTE non-uniformity in Primary Mirror Substrate
- **Dynamics**
 - Lessons from Integrated Model (FRF)
 - WASP Disturbances / LoS



Wavefront Error

- **Several “Offsets” or Re-Calibration Points**
 - Wavefront can be Nulled When Re-aligned with M2
 - Wavefront can be Polished Into Orientation when M1 is Built
- **Wavefront Error Linearity and Separability**
 - WFE is Small
 - Approximation of Linearity Checked for this Design
 - Aberrations can be (Almost) Arbitrarily Added (Subtracted)
 - Not Necessarily RSS'd
 - Misalignment, Thermal Soak, Gradient, Re-Calibration
 - WFE Conveniently Separated
 - Tip / Tilt: Line of Sight which is Calibrated
 - Focus / Coma: Aberrations Removed by M2 Alignment
 - Low Order (remaining) Zernikes (up to 36)
 - High Order Residuals (fine features)



Gravity

- **Balancing Stiffness**
 - Goal in Design
 - Mirrors will Move... They Must Move Together!
- **Impact on Operation over Elevation**
 - Does Mirror Deformation Affect WFE?
 - How Much Does Rigid Body Motion Affect
 - WFE? LoS?
 - How Often Would M2 Need to Be Adjusted
- **Impact on Primary Mirror Testing**
 - Why Polish M1 Facing Up? Sideways? At ?? Degrees?



Computational Process wo/Zemax

Collect Nodes from Each Load Case

- Individually for Each Mirror
- Individually for Each Load Case

Fit Rigid Body Movement

- Translation
- Rotation

Compute Zernike Coefficients

- Use Linearization and Sensitivities

Compute Aberrated WFE

Compute RMS WFE



Computational Process w/Zemax

Collect Nodes from Each Load Case

- Individually for Each Mirror
- Individually for Each Load Case

Fit Rigid Body Movement

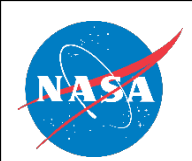
- Translation
- Rotation

Displace / Rotate Model

- Use Matlab + Zemax ZOS-API

Retrieve Zernikes

Compute RMS WFE



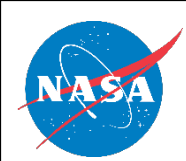
Linearization / Matlab, Zemax

FEA / Nastran, FEMAP

Rigid Body Analysis / Nastran, Matlab

Full WFE Analysis / Matlab, Zemax

DETAILS



**How Sensitive is the Optical Performance to Motion?
Taylor Series / Partial Derivatives / Jacobian**

LINEAR SENSITIVITY MODEL



Mathematical Model

- **Same Concept as a Taylor Series**

- Perturb the Model with Each DoF
- Record the Variation in the WFE
- Ignore Cross Terms

$$W = \sum_{i=1}^{36} C_i \times Z_i(x, y)$$

← Zernike Expansion

$$C_i(\delta x, \delta y, \delta z; \delta \theta_x, \delta \theta_y, \delta \theta_z) \approx \delta x \times \frac{\partial C_i}{\partial x} + \delta y \times \frac{\partial C_i}{\partial y} + \dots$$

← Taylor Series

$$\delta u \times \delta v \times \frac{\partial^2 C_i}{\partial u \partial v} \ll \delta u \times \frac{\partial C_i}{\partial u}$$

← Ignore Higher Order Terms

Jacobian Matrix Model:

C is a Dependent Variable Vector (F)

DoF's are Independent Variable Vectors (x)

$$J = \begin{bmatrix} \frac{\partial F_1}{\partial x_1} & \dots & \frac{\partial F_1}{\partial x_n} \\ \vdots & \ddots & \vdots \\ \frac{\partial F_m}{\partial x_1} & \dots & \frac{\partial F_m}{\partial x_n} \end{bmatrix}.$$



Extract Values from Zemax Model

Use Matlab to Interrogate Optical Model

Perturb a Degree of Freedom (DoF)

Collect Zernike Coefficients

Map Coefficients to WFE

Record RMS vs. DoF



Degrees of Freedom for Sensitivities

M2 Despace

- Move M2 along the Optical Axis

M2 Translation

- Move Along X Axis
- Move Along Y Axis

M2 Rotation

- Rotate About X Axis
- Rotate About Y Axis

M1 Translation

- Move Along X Axis
- Move Along Y Axis

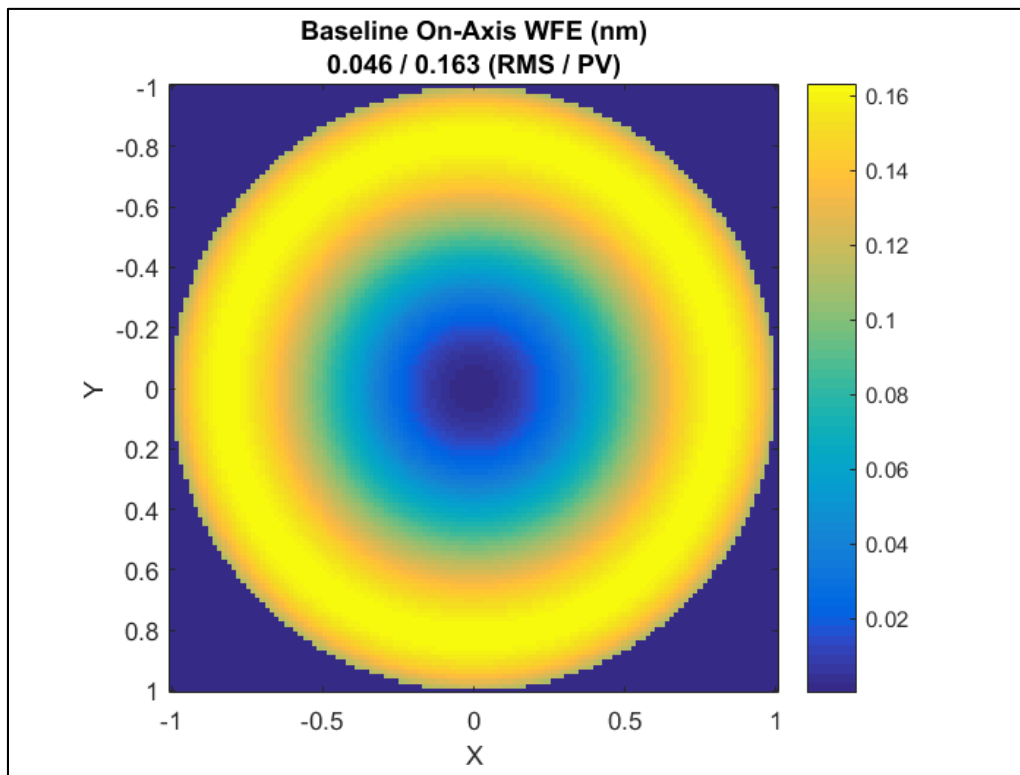
M1 Rotation

- Rotate About X Axis
- Rotate About Y Axis

1. All Results are in Coordinate System of Optical Model
2. All Results Appear Linear over the Range of Interest
3. Cross Terms were Not Evaluated
4. Zernikes are *Not* Orthogonal Over Annulus...Ignored this for Now



On-Axis Performance

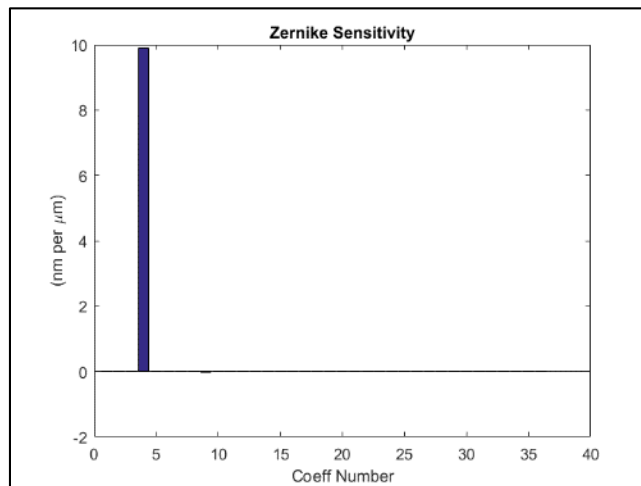
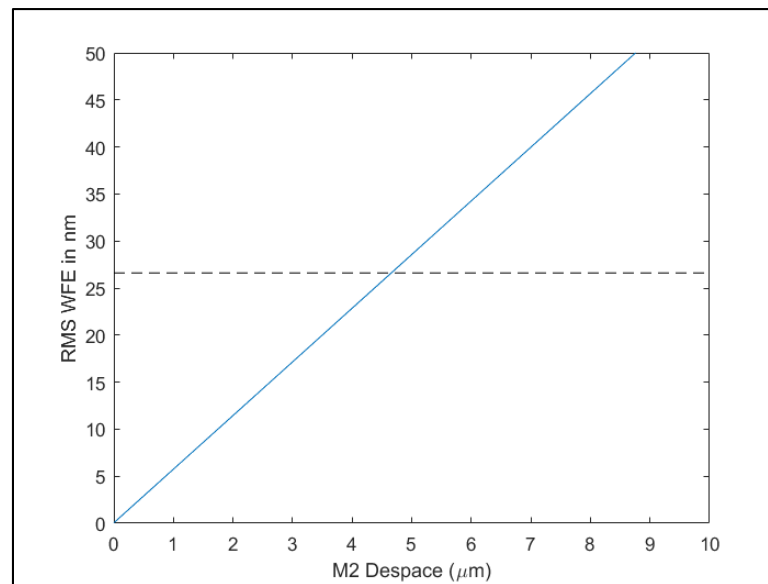
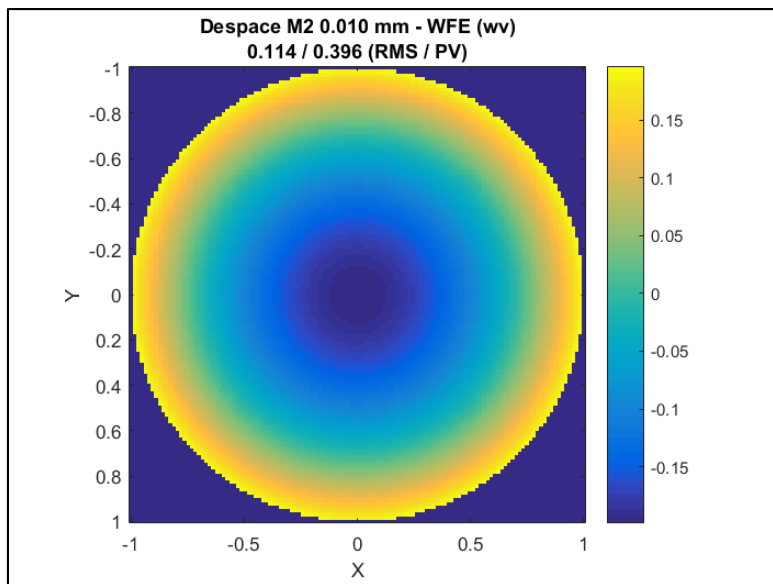


- **Collect Zernikes**
- **Map WFE Over the “Unit Circle”**
 - Unit Circle: Normalized to Radius = 1 at Edge of Pupil
 - Ignore Central Obscuration for Now...



Despace: Moving M2 Along Z

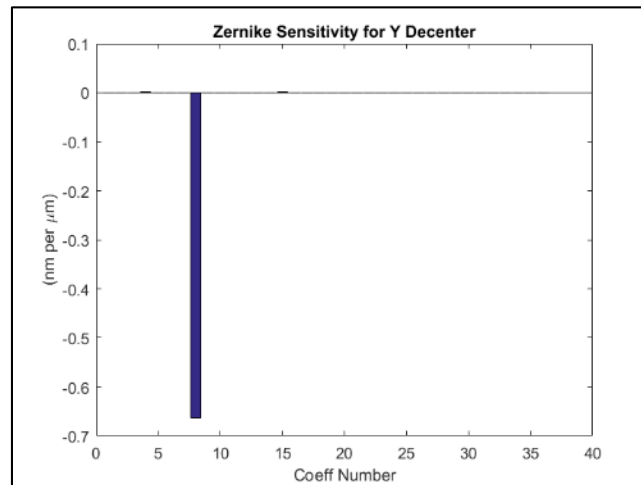
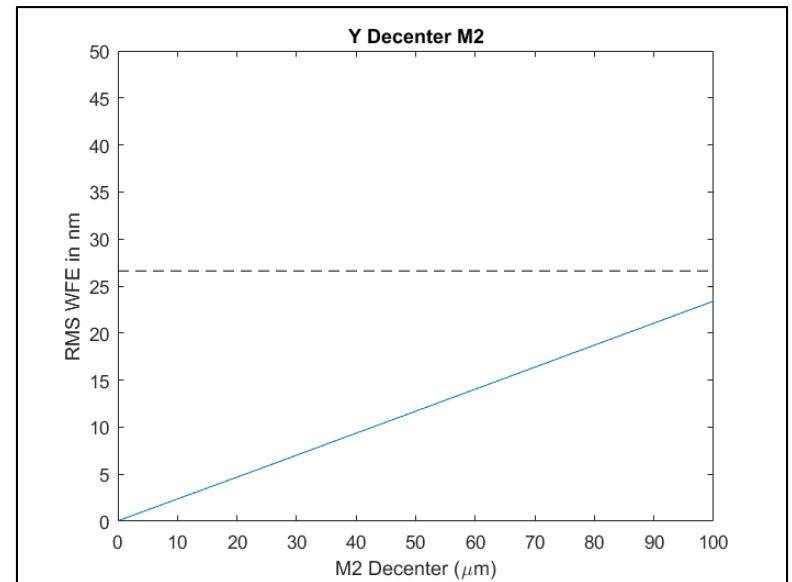
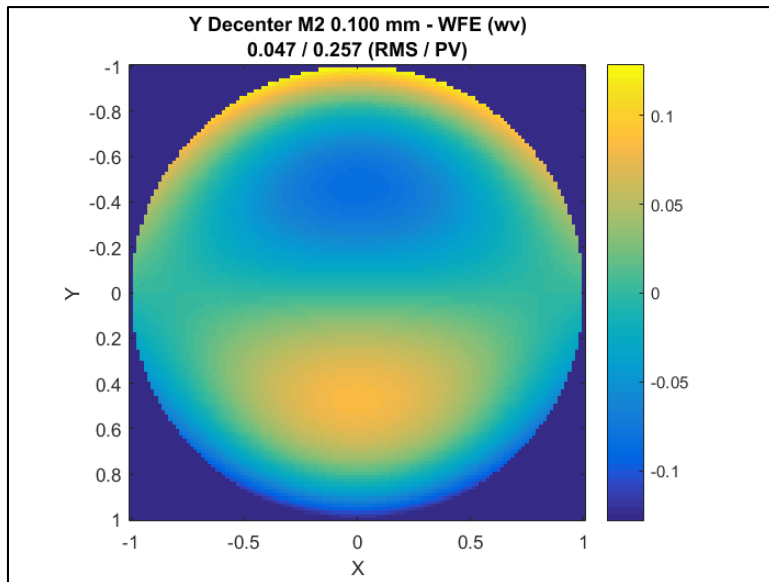
- Range: 0 to 10 μm





Decenter: Moving M2 Along Y

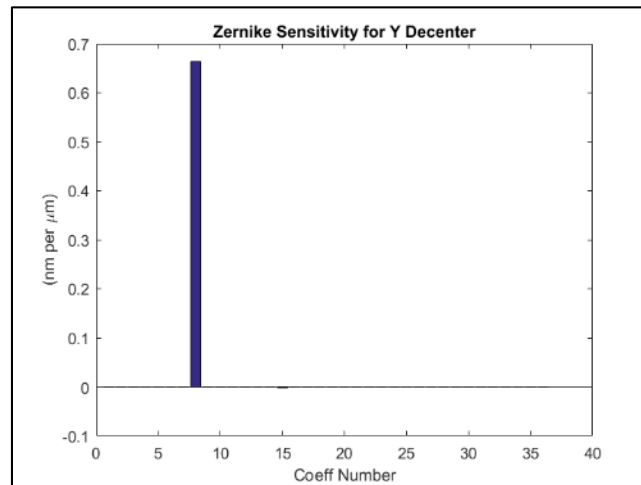
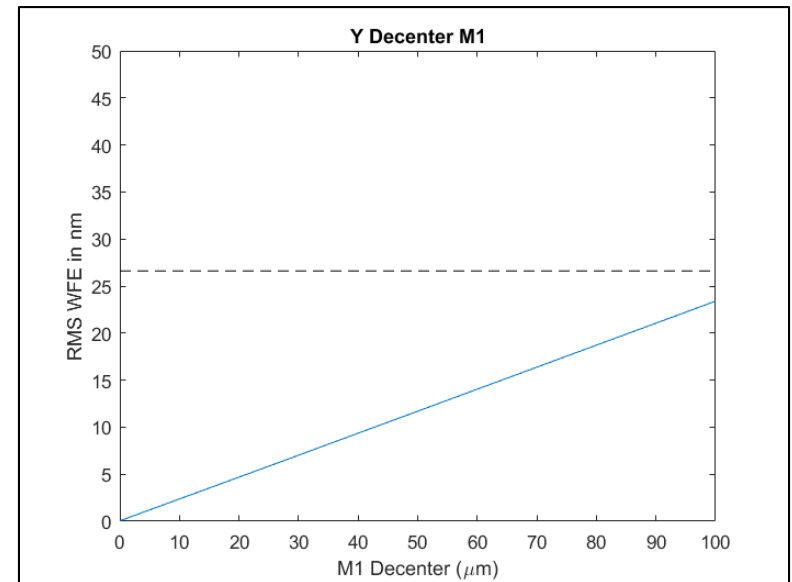
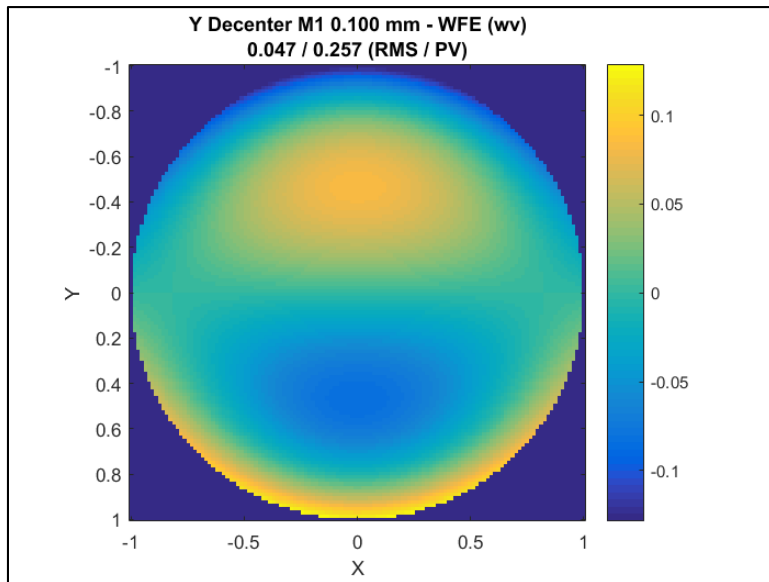
- Range: 0 to 100 μm





Decenter: Moving M1 Along Y

- Range: 0 to 100 μm

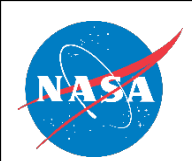




Line of Sight

- **Add Feature to Zemax Model**
 - Coordinate Break to Represent Instrument Deck
- **Rigid Body Motion of the Instrument Deck**
 - Translates Around Center of Deck
 - Rotates Around Center of Deck
- **Follow Spot After Instrument Deck**
 - Spot Location vs. Elevation

$$\vec{\theta}(\theta_{elev}) \equiv \begin{pmatrix} \theta_x(\theta_{elev}) \\ \theta_x(\theta_{elev}) \end{pmatrix} \approx \begin{pmatrix} x(\theta_{elev}) \\ y(\theta_{elev}) \end{pmatrix} \times \frac{1}{f}$$

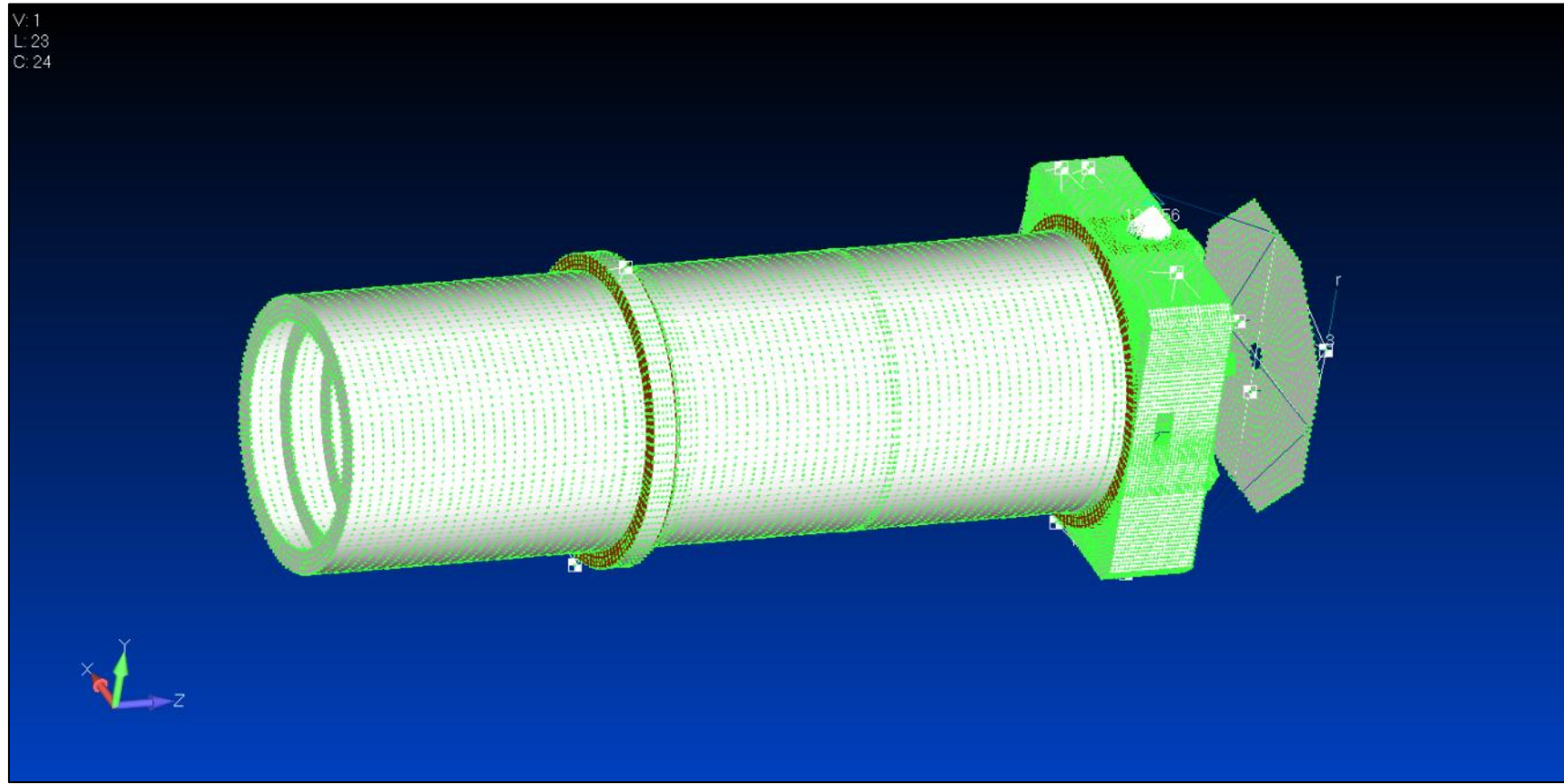


Nastran Linear Model

STRUCTURAL MODEL

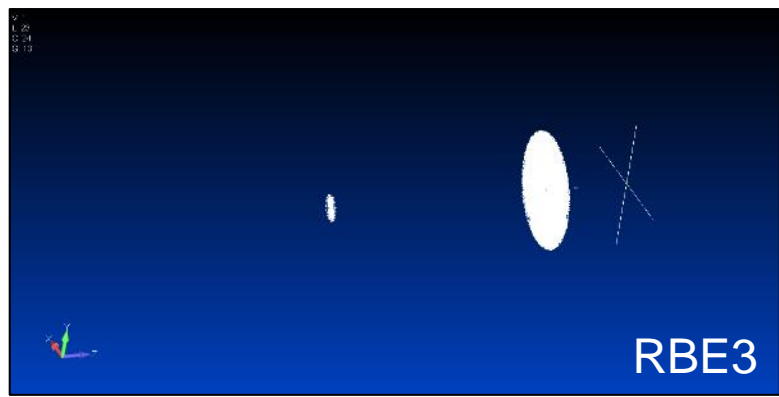
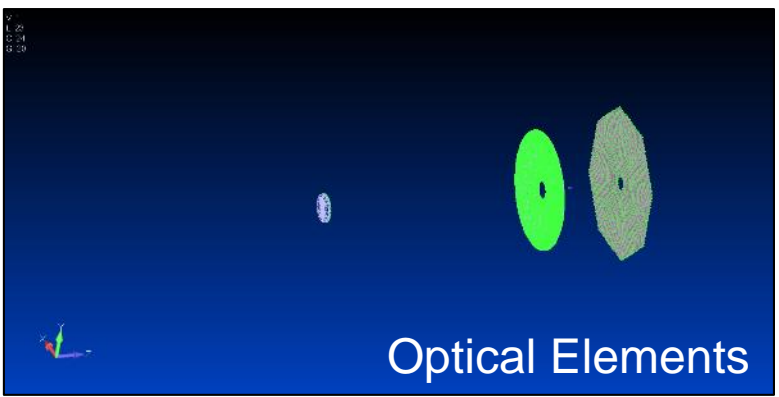
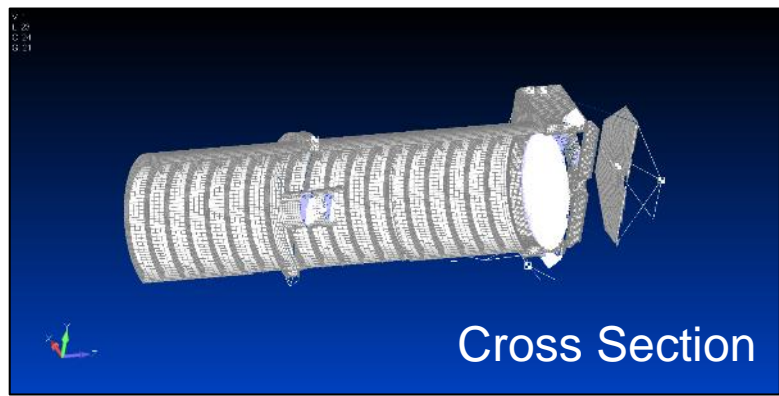
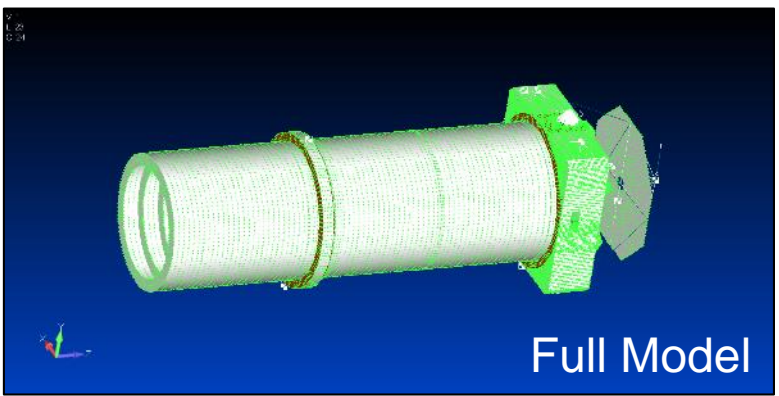


Model Viewed in Femap





FEM Views





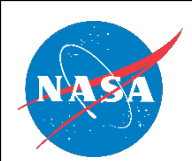
Combining Load Sets

- **FEA without Load = 0-G (in space!?)**
 - Performance over Elevation = Difference Between Various Load Cases
- **Load Cases**
 - Horizon ($\theta_{\text{Elev}} = 0$), Zenith ($\theta_{\text{Elev}} = 90$)
 - Various Elevations: $\theta_{\text{Elev}} = \{15, 30, 37, 45, 65\}$
- **Interpretation**
 - Assume Telescope is Aligned at Horizon
 - Assume Model is Linear
 - Deformation is Relative to the Horizon (1G-Y) Load Case

$$\{u_{deformed}\} = \{u_{node}\} + \{\delta u_{load}\}$$

$$\{u_{zenith}\} = \{u_{1G-Z}\} - \{u_{1G-Y}\} \quad \text{Relative to Horizon Load Case}$$

$$\{u(\theta_{elev})\} = \left\{ u \left(\vec{F}(\theta_{elev}) \right) \right\} - \left\{ u \left(\vec{F}(\theta_{elev} = 0) \right) \right\}$$



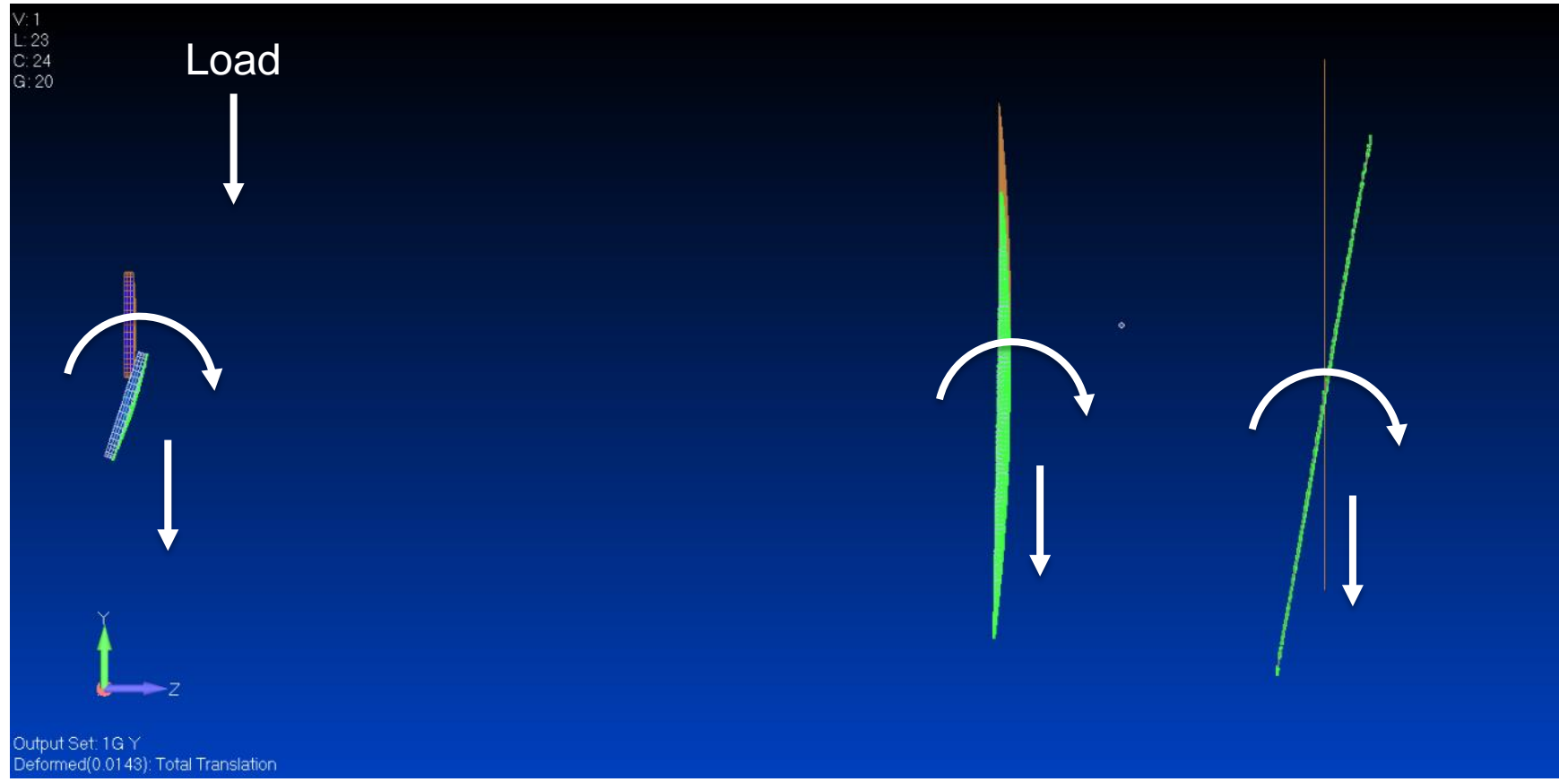
Rigid Body Motion

STRUCTURAL MODEL



Pointing At Horizon

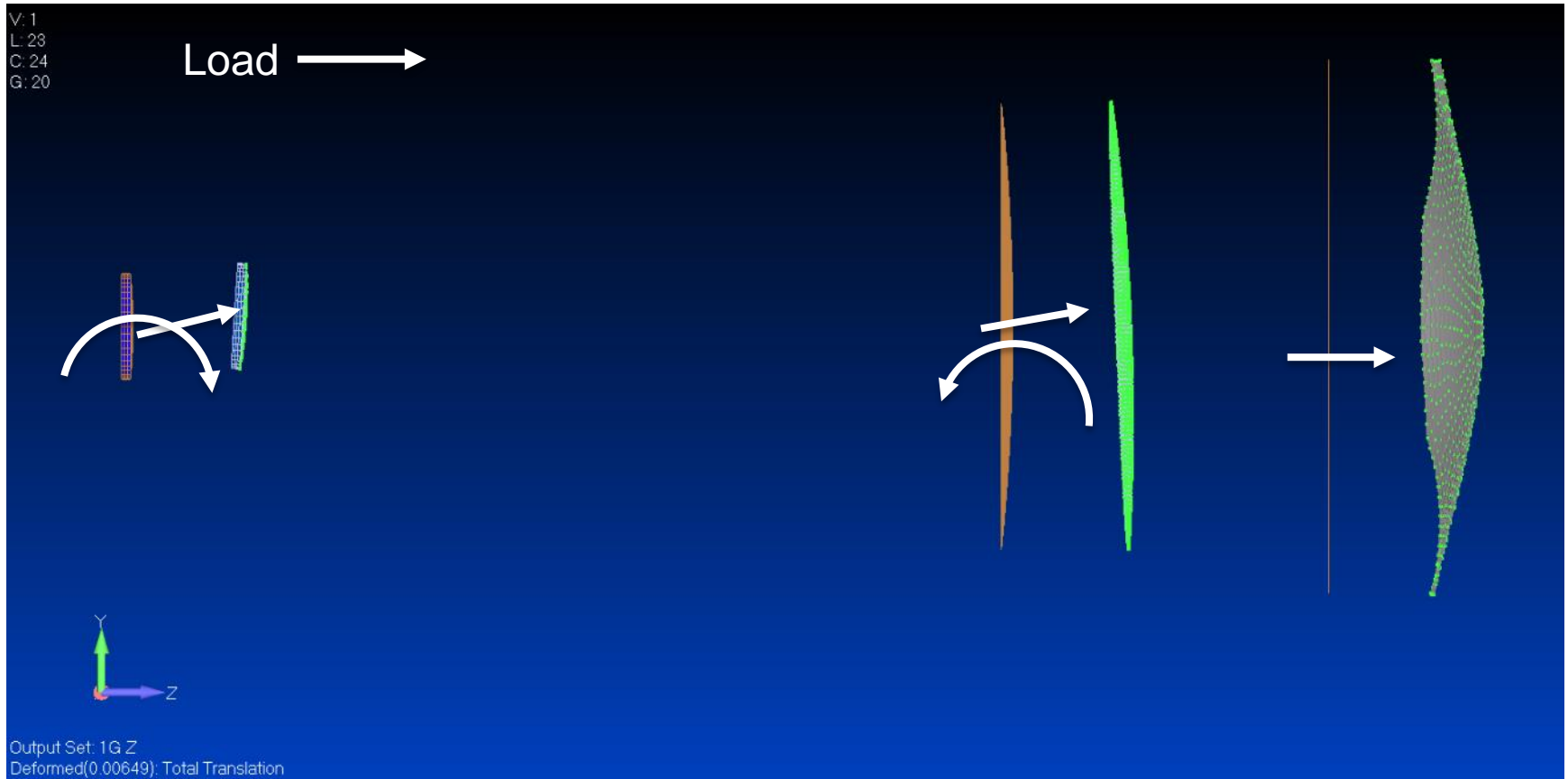
- Deformation from 0-G





Pointing At Zenith

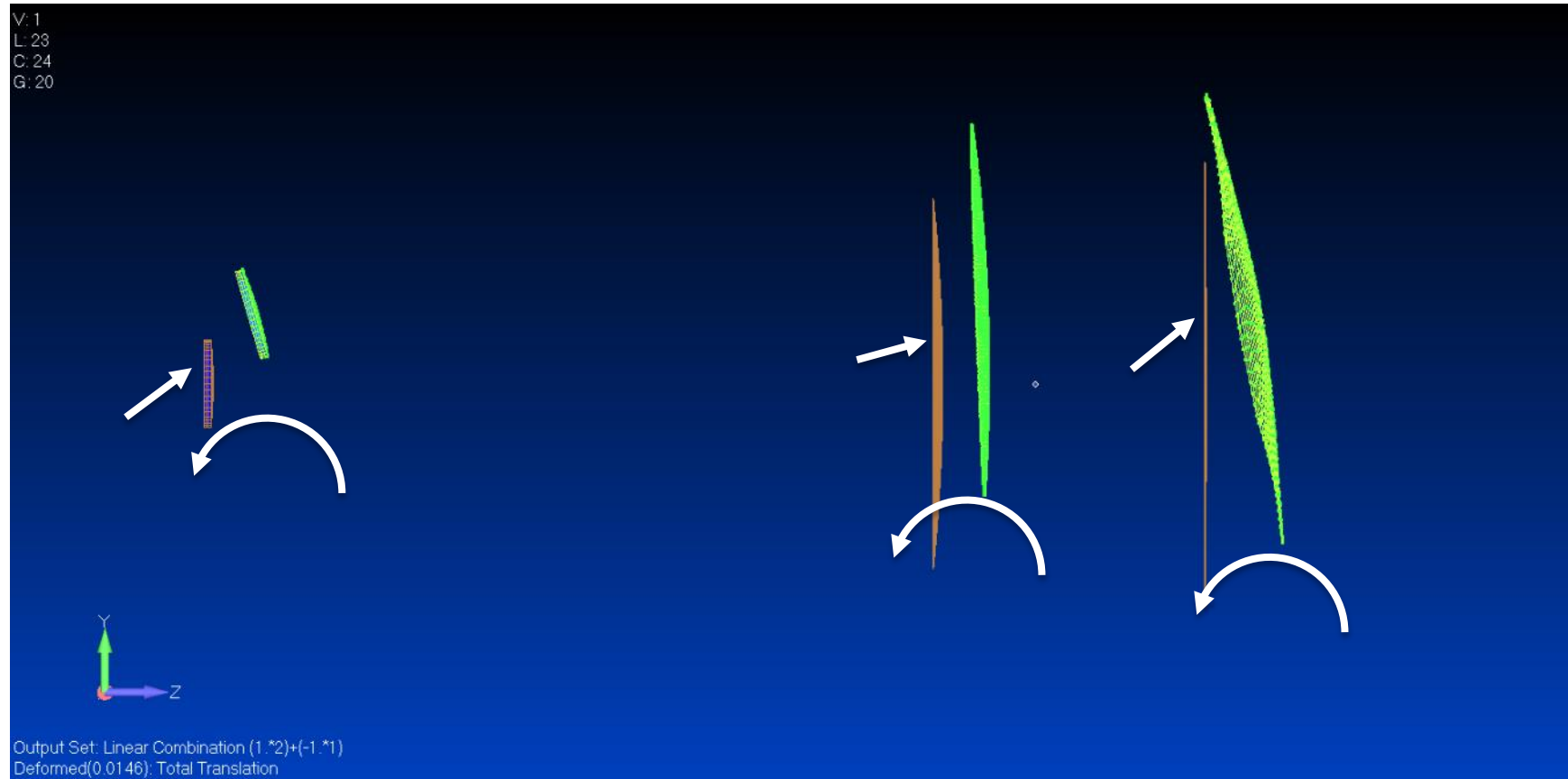
- Deformation from 0-G





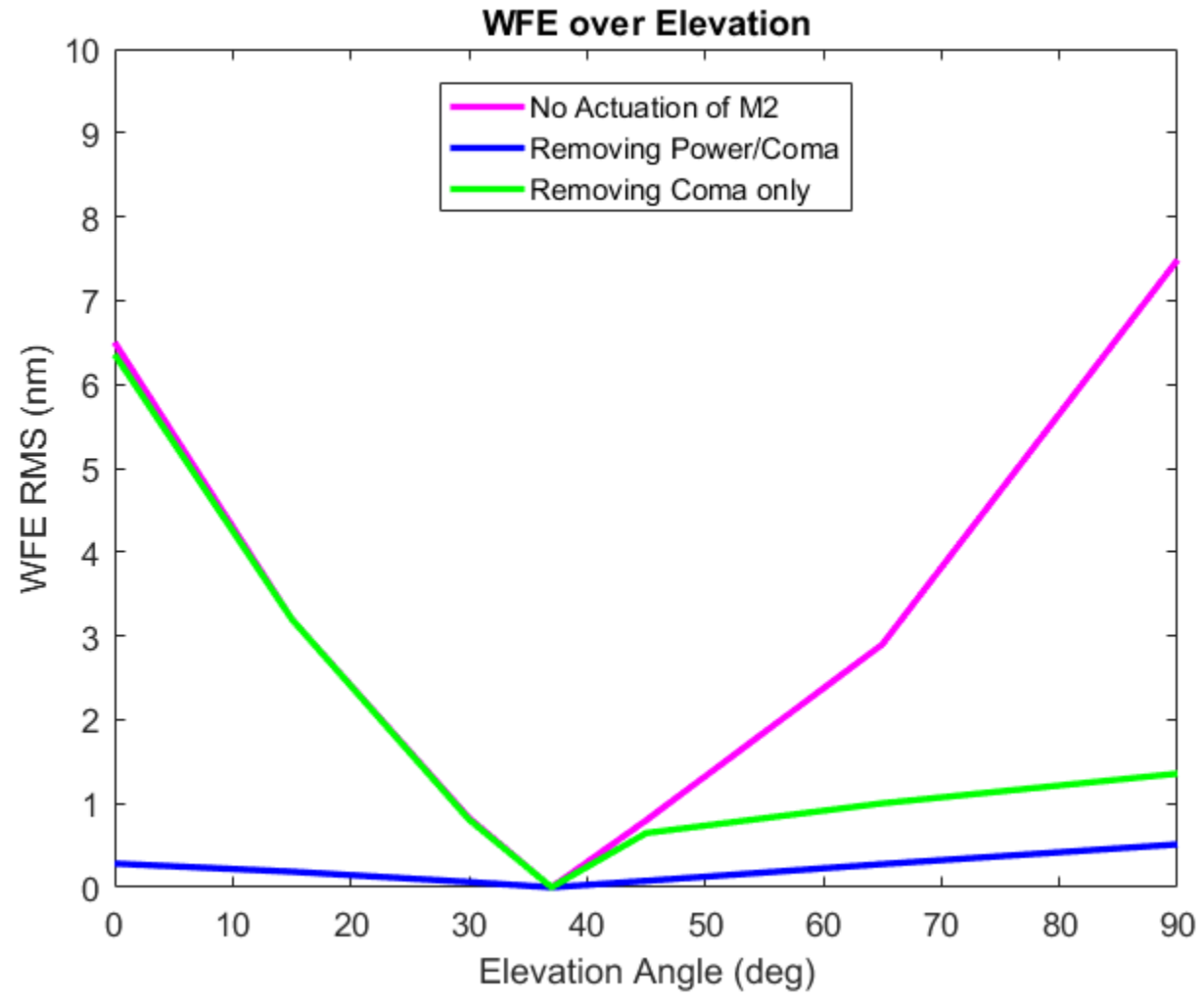
What Does the Difference Look Like?

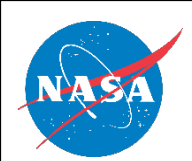
- Deformation at Zenith Relative to Horizon





WFE over Elevation (Aligned at 37 deg)





Primary Mirror Deformation

STRUCTURAL MODEL

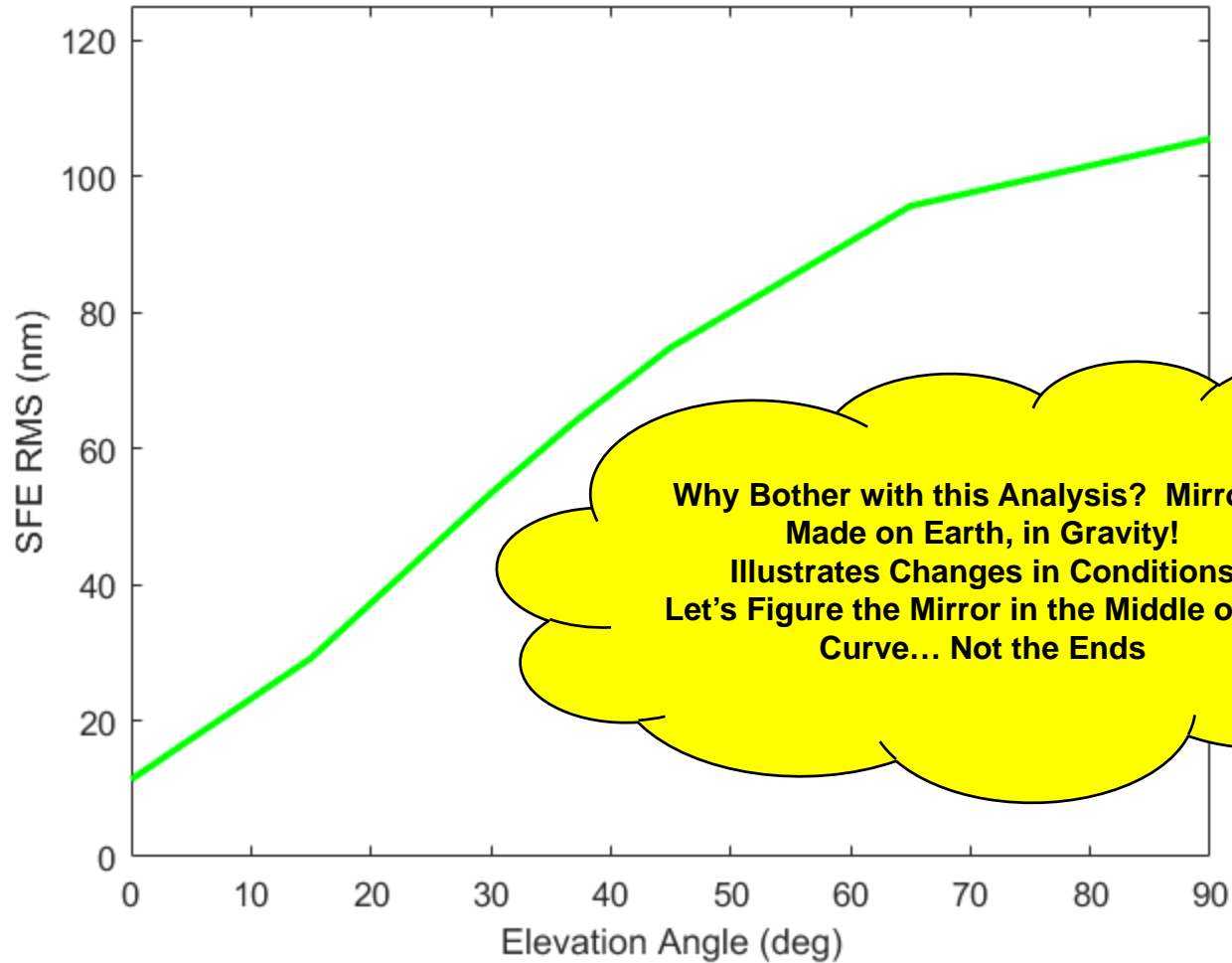


Mirrors are Not Infinitely Stiff

- **Solid Mirrors Deflect when Elevation Angle Changes**
 - So Do Lightweight Mirrors
- **Key Questions**
 - Does the Change Over All Elevations Meet Budget?
 - Can Mirror be Figured (Polished) with a “Bias” to Ensure Budget is Met at the Extremes?



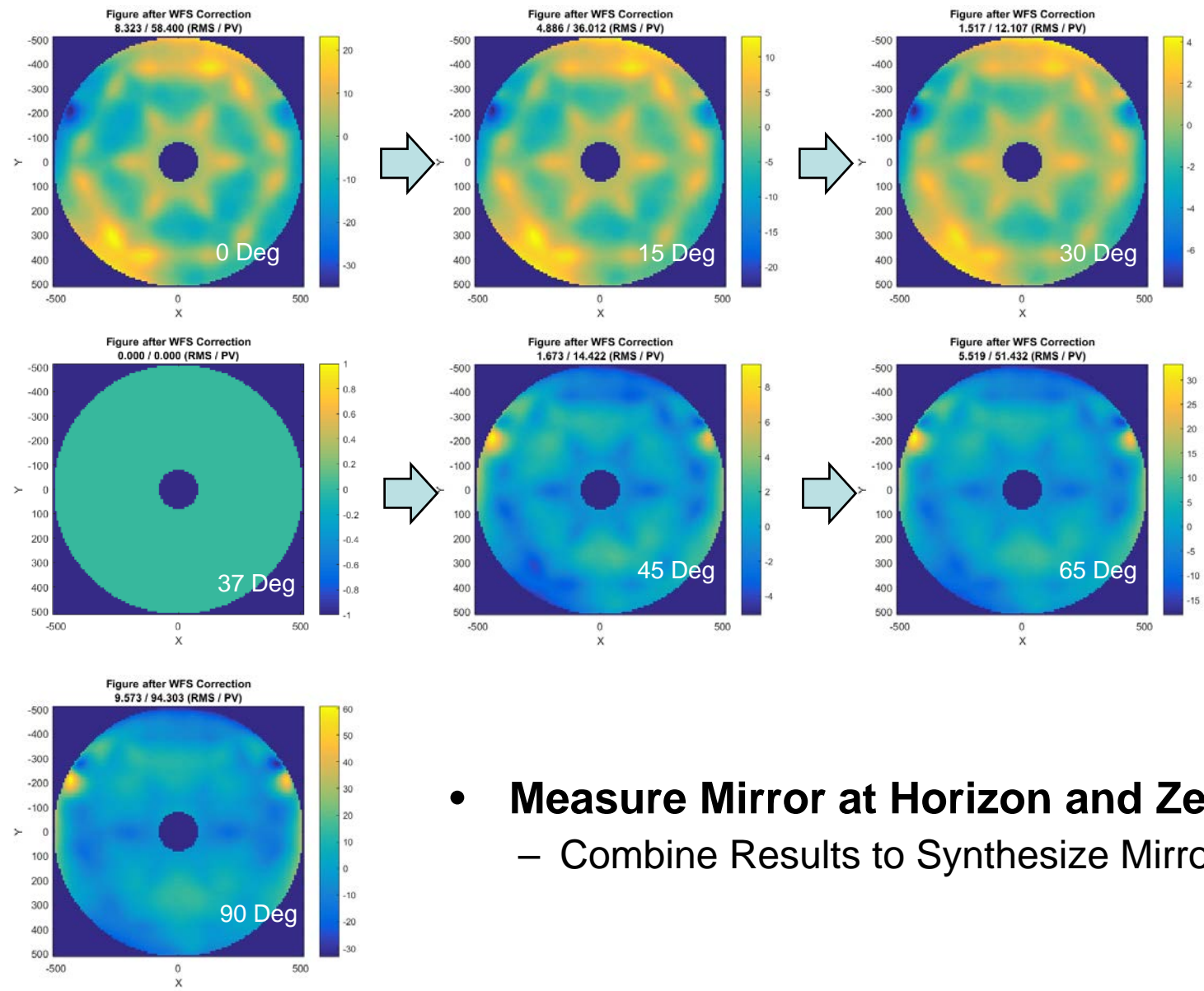
Surface Figure Error if Made in 0-G



**Why Bother with this Analysis? Mirror is Made on Earth, in Gravity!
Illustrates Changes in Conditions
Let's Figure the Mirror in the Middle of the Curve... Not the Ends**



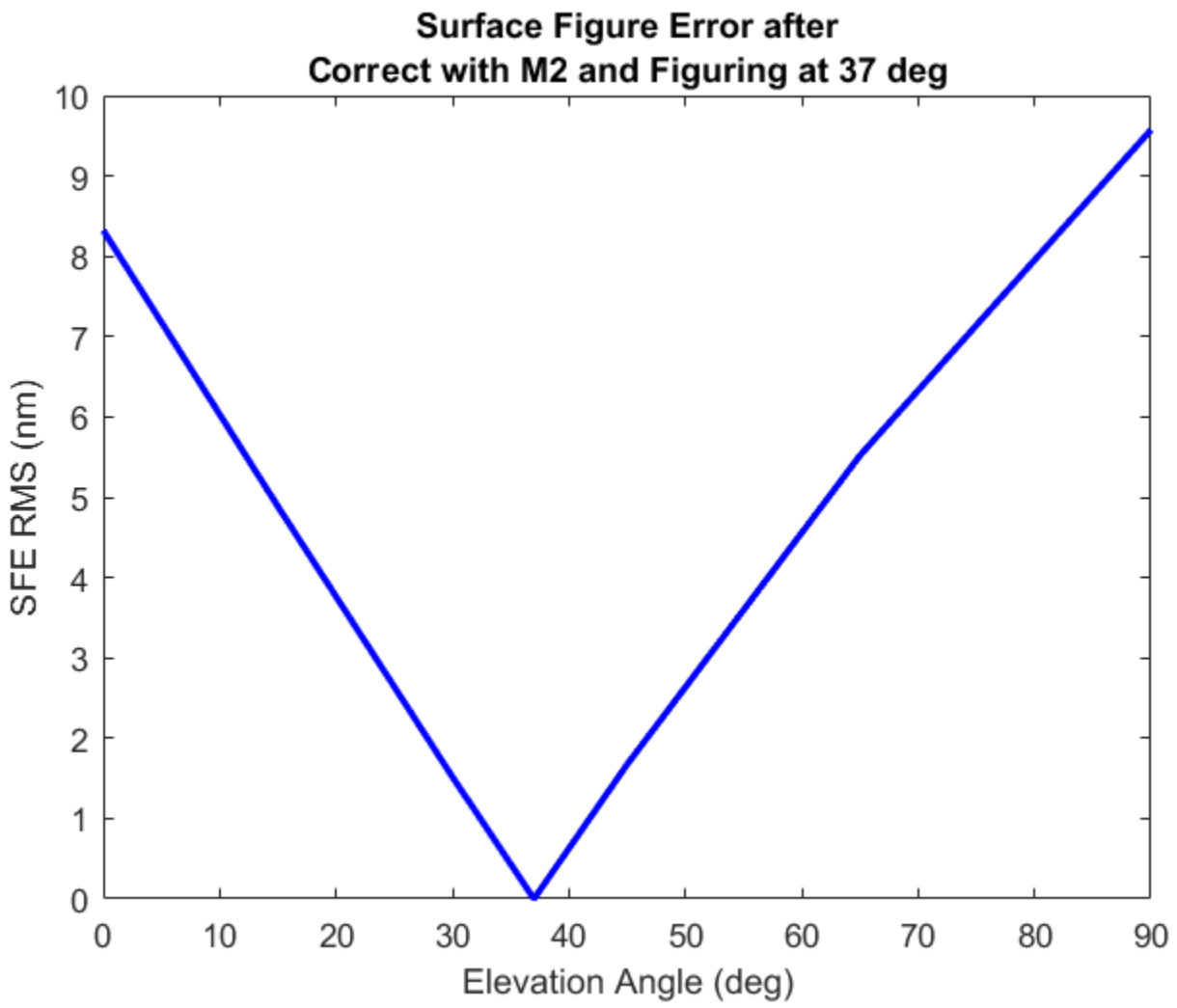
Mirror Figured at 37 Deg

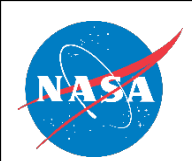


- **Measure Mirror at Horizon and Zenith**
 - Combine Results to Synthesize Mirror at 37 Deg



Surface Figure Error – Meets Budget Allocation





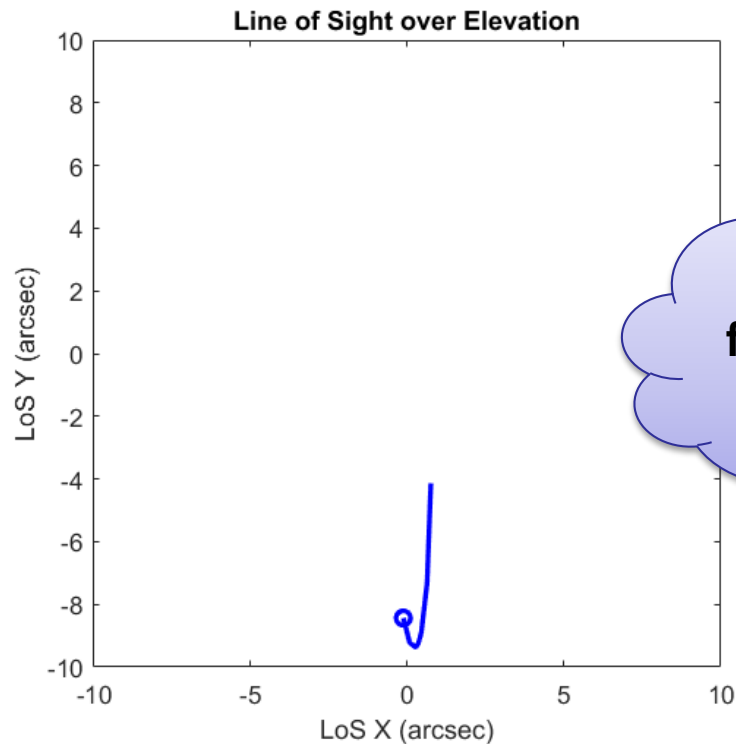
Line of Sight

STRUCTURAL MODEL

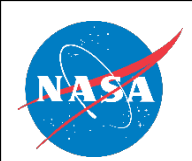


Line of Sight / Changes with Elevation

- **Raw Motions are Large**
 - 5 to 50 μm and 0 to 200 μrad
- **Relative Motions are Small**
 - Balancing Stiffness Keeps M1, M2 Aligned... the Sag Together
- **Compute LoS from Spot Diagram**
- **Easily Withing Range of Guidance System**



**Trajectory
from 0 Deg to
90 Deg**



Thermo-Elastic Results from Thermal Soak
STRUCTURAL MODEL



Temperature

- **Impact of Soak**

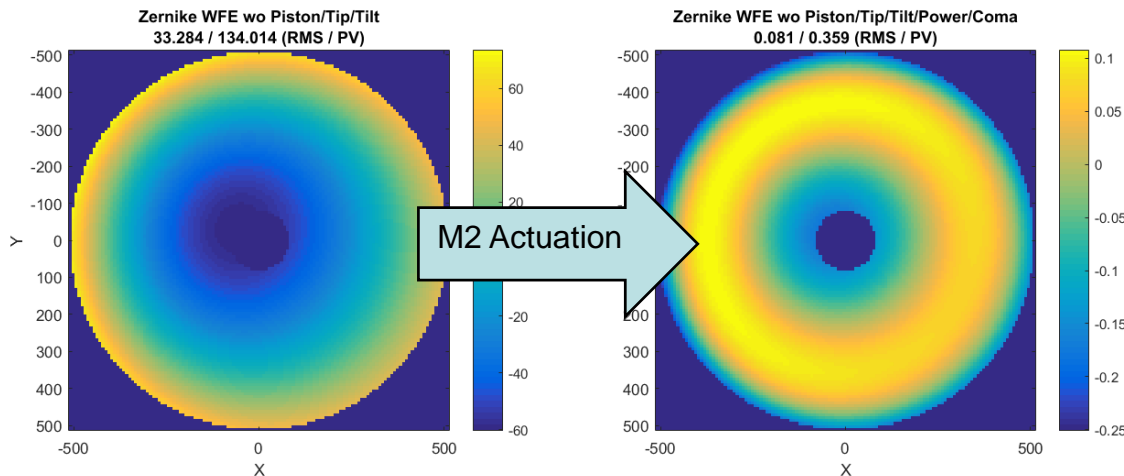
- Does Soak Deform M1 in a Manner that is Not Correctable with Alignment?
- What is the Required Capture Range for a Wavefront Sensor from Ground to Float?

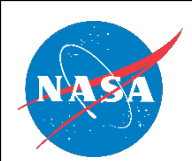
- **Same Process**

- Apply Thermal Soak Conditions
- Recover Mirror / Instrument Motion
- Apply to Optical Model
- Recover WFE

WFE < 34 nm RMS
Easily within Capture Range

Residual after Correction
WFE_{residual} < 1 nm RMS



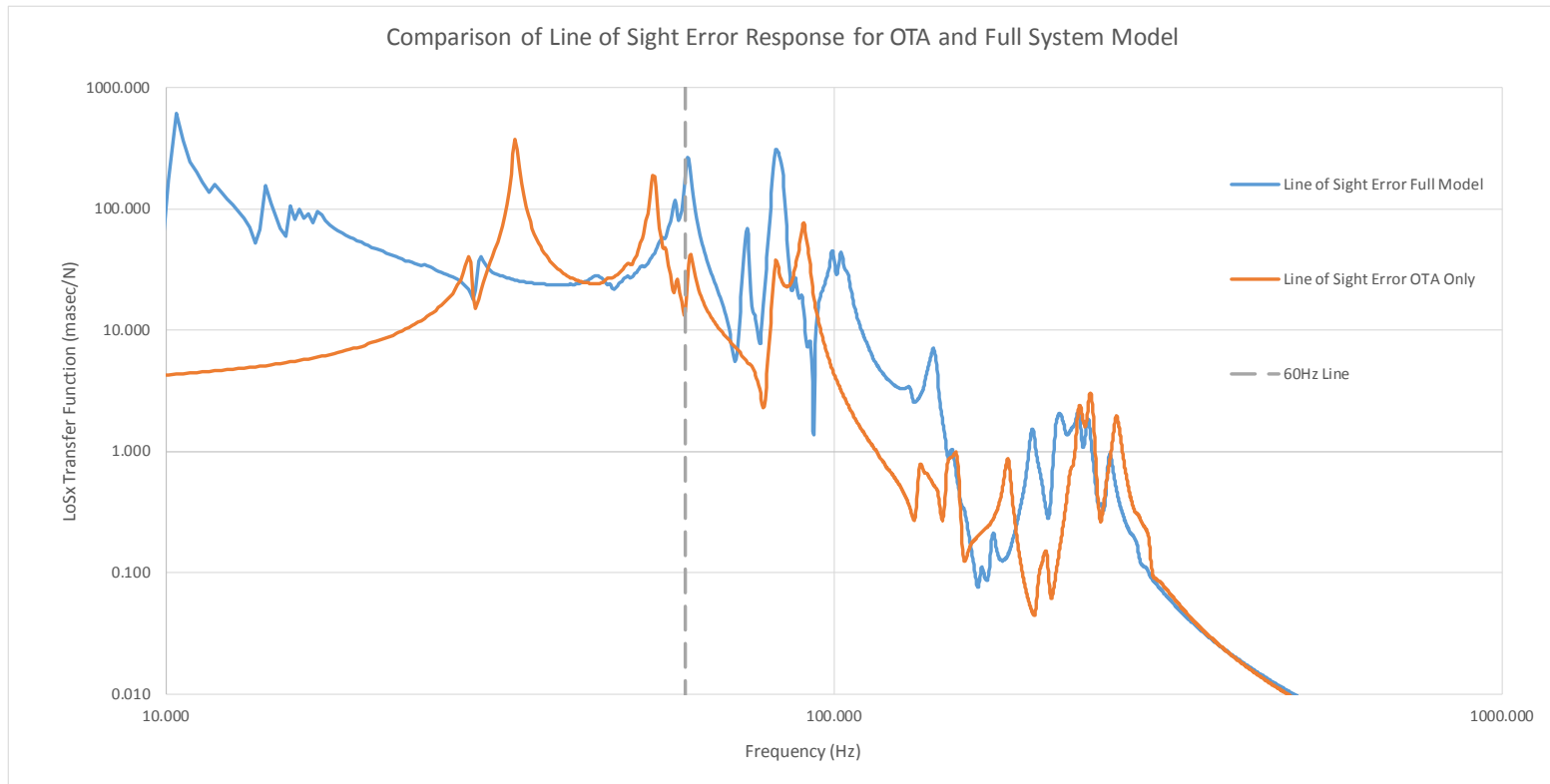


Modes to Jitter

DYNAMICS MODEL



Integrated vs. OTA Model



- **LoS Transfer Function = LoS from Disturbance vs. Frequency**
- **Used to Evaluate Cryo-Cooler Disturbance (60 Hz)**
- **Note Shift in Response of OTA vs. Integrated Model**

Lesson Learned: When Possible, Use Integrated Model



Line of Sight / WASP Disturbance

- **What is the Jitter (Fast Motion of PSF) when Observing Science?**
 - Take WASP Disturbance
 - Apply to Integrated Model
 - Capture M1, M2, Instrument Motion
 - Convert to Line of Sight
- **Jitter < 5 milli-arc-sec**



Summary

- **Described Rigorous STOP Analysis Process**
 - Integrated Inputs from (x4) Models into System Performance
- **Supports WFE and LoS Budgets**
 - See B. Woodruff's Poster