



Cooling the Origins Space Telescope

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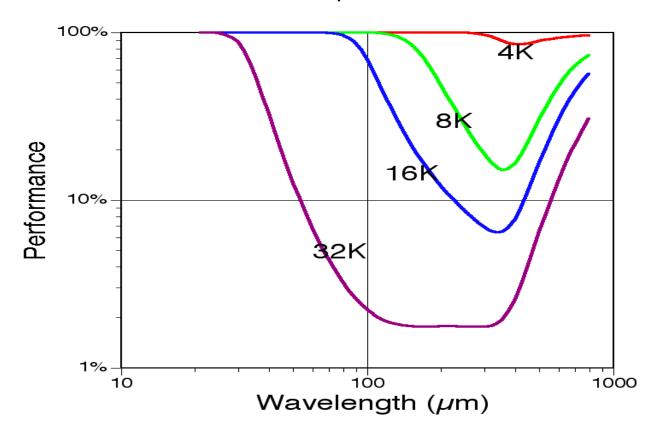
- NASA Headquarters Astrophysics Division commissioned 4 studies for one possible flagship mission to launch in the 2030's
 - Lynx (X-ray Surveyor)
 - LUVOIR (Large UV, Optical, and near InfraRed Observatory
 - HabEx (Habitable Exoplanet mission)
 - Origins Space Telescope (OST) (Far IR Surveyor)





Why Do We need a 4.0 K Telescope?

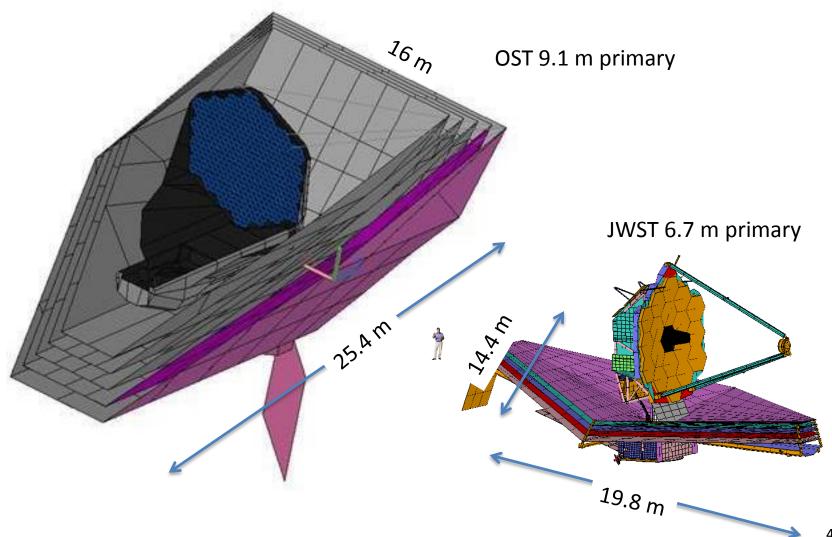
- OST will cover the wavelength range from 6 μm to 600 μm
 - The goal is to be background limited limited by the cosmos rather than self emission from the telescope







How Big is This Thing?







One Question

- How can we do a 9 m diameter 4 K telescope when a 6.7 m 40 K telescope (JWST) was so difficult?
 - Cryocoolers are now more mature
 - Cryocoolers at low temperature have a huge advantage over radiative cooling at low T
 - OST has a longer wavelength so the optics are less challenging
 - Low temperature has advantages
 - Low thermal contraction with changing temperatures for one
 - The design is driven by cryo/thermal considerations
 - Cryogenics leads to solutions!!!





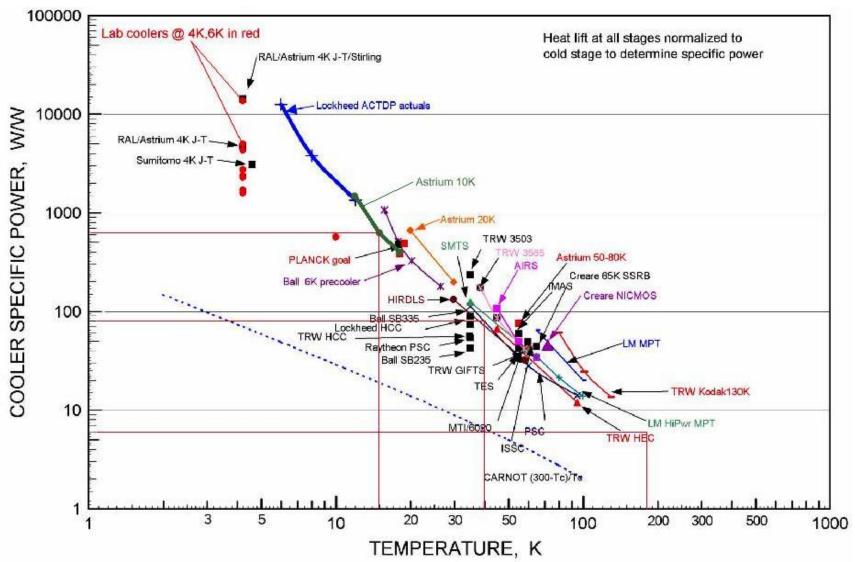
State of the Art for cooling

- ACTDP and follow up matured coolers from 3 different companies
- Approximately 10,000 W of input power per W of cooling power
- Space Cryocooler Reliability is Extremely High
 - From Ron Ross's ongoing survey



Cooling Power Vs. T









Staged Cooling

- Utilize staged cooling to go from 300 to 0.05 K
 - Start with radiative cooling of the sunshield
 - Use 3 stages of cryocooler cooling
 - Finish with a subKelvin cooler to provide 50 mK





Nominal Heat Flows

- Rough calculation of the heat absorbed by the 4 K cryocooler stage is broken down as follows:
 - Telescope: radiation: 55 mW, conduction from structure
 20 mW, conduction from harnesses 30 mW
 - Instrument dissipation: Maximum 100 mW mainly from low temperature preamplifiers
- Will use eight 50 mW cryocoolers in parallel which provides redundancy, 100% margin on the expected cooling load, and keeps size close to current technology





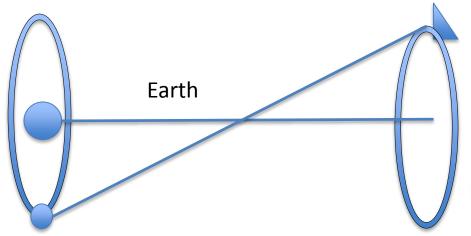
Sunshield Principle

- Radiate horizontally, block radiation perpendicular
- Practical Considerations
 - Solar pressure imbalance
 - Deployment
 - The Sun, Earth, and Moon do not stay in one place





Keep Out Angles

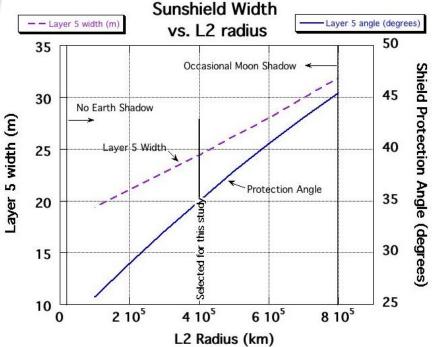


OST

L2 Orbit

Moon

Field of Regard adds to shield size OST Pitch= $+5^{\circ}$ /- 45° , Roll = $\pm 5^{\circ}$, Yaw = 360°







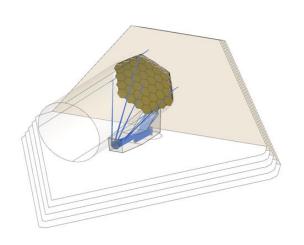


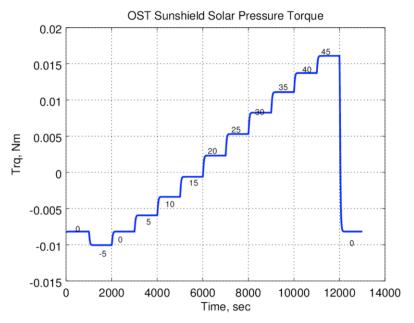
- Ideally the center of solar pressure (\sim 9 μ Pa) and center of mass are in the same place
- Any offset must be overcome with momentum wheels and propulsion
 - Aside from mass, frequent propulsive maneuvers disrupt observing time
- [Two Cases]

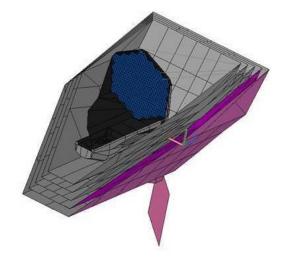


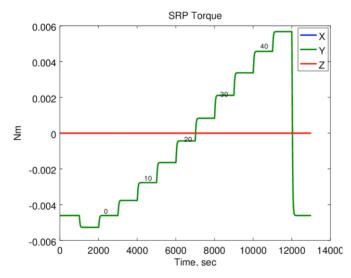


ORIGINS Space Telescope Solar Torque - Two Cases











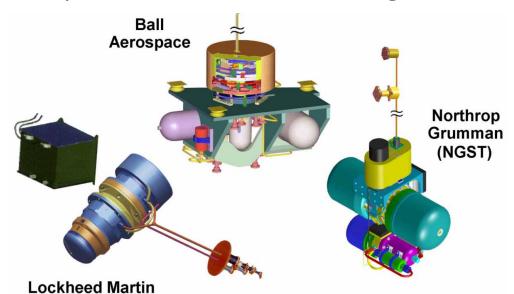
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Cryocooler Considerations

- Staged heat extraction
- Vibration issues
 - Not so much for image stabilization as for microphonics on detectors
- Packaging and distributing cooling

50 mW cooling at 4 K plus 20 K and 70 K cooling for 500 W

of input power







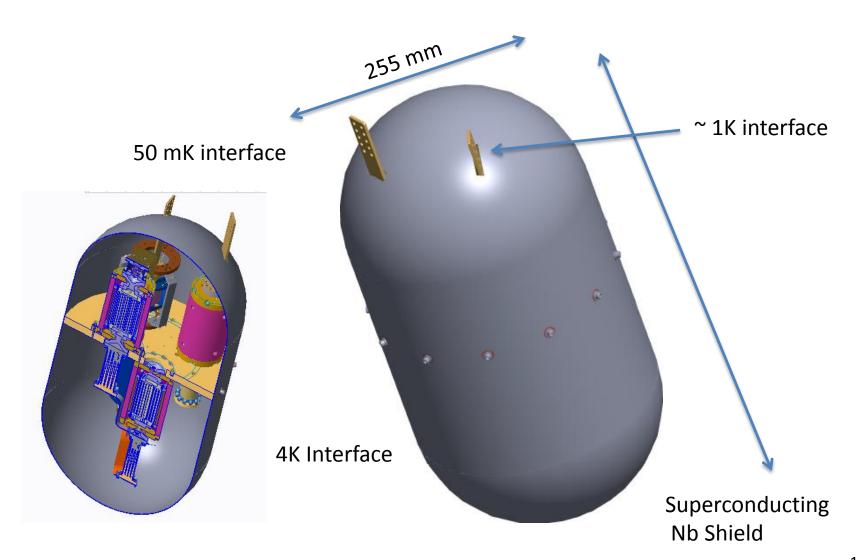
Sub-Kelvin Cooling

- Instruments whose detectors require cooling to less than 1 K will be sized to accommodate a provisional sub-Kelvin cooler.
 Such a cooler is currently at TRL4 and will be at TRL6 by the end of the current SAT (end of 2019)
 - Up to 5 μW continuous cooling at 50 mK (max duty cycle)
 - Up to TBD mW cooling at ~1K
 - Heat rejection to 4 K
 - 6 mW (at max duty cycle), 3 mW at min duty cycle
 - Magnetic shielding to provide $< 1\mu T$ at the boundary of the cooler
 - Mass ~ 21 kg
 - Volume (see graphic)





ORIGINS Space Telescope Multi-Stage ADR Sub-Kelvin Cooling





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Summary and Conclusions

- The Origins Space Telescope is being studied as a flagship class astrophysics mission for the 2030's
 - 9.1 m diameter primary operating at 4 K
 - Cooling is achieved by proper staging of radiative coolers (sunshields), mechanical cryocoolers
 - 4 K and lower instruments
 - 3 instruments require subKelvin (50 mK) temperatures which could be provided by new continuous adiabatic demagnetization refrigerator