



Human Mars Mission Power Architectures

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Human Mars Mission by the Numbers

No decisions have been made, current thinking

Multiple visits to a single landing site

100 km

Notional excursion from landing site

- Goal to extend as far as possible
- Robotic assets may rove further

~500

Days maximum stay for any given mission

- Driven by orbital mechanics



Twenty Six

Months between opportunities

Conjunction class missions

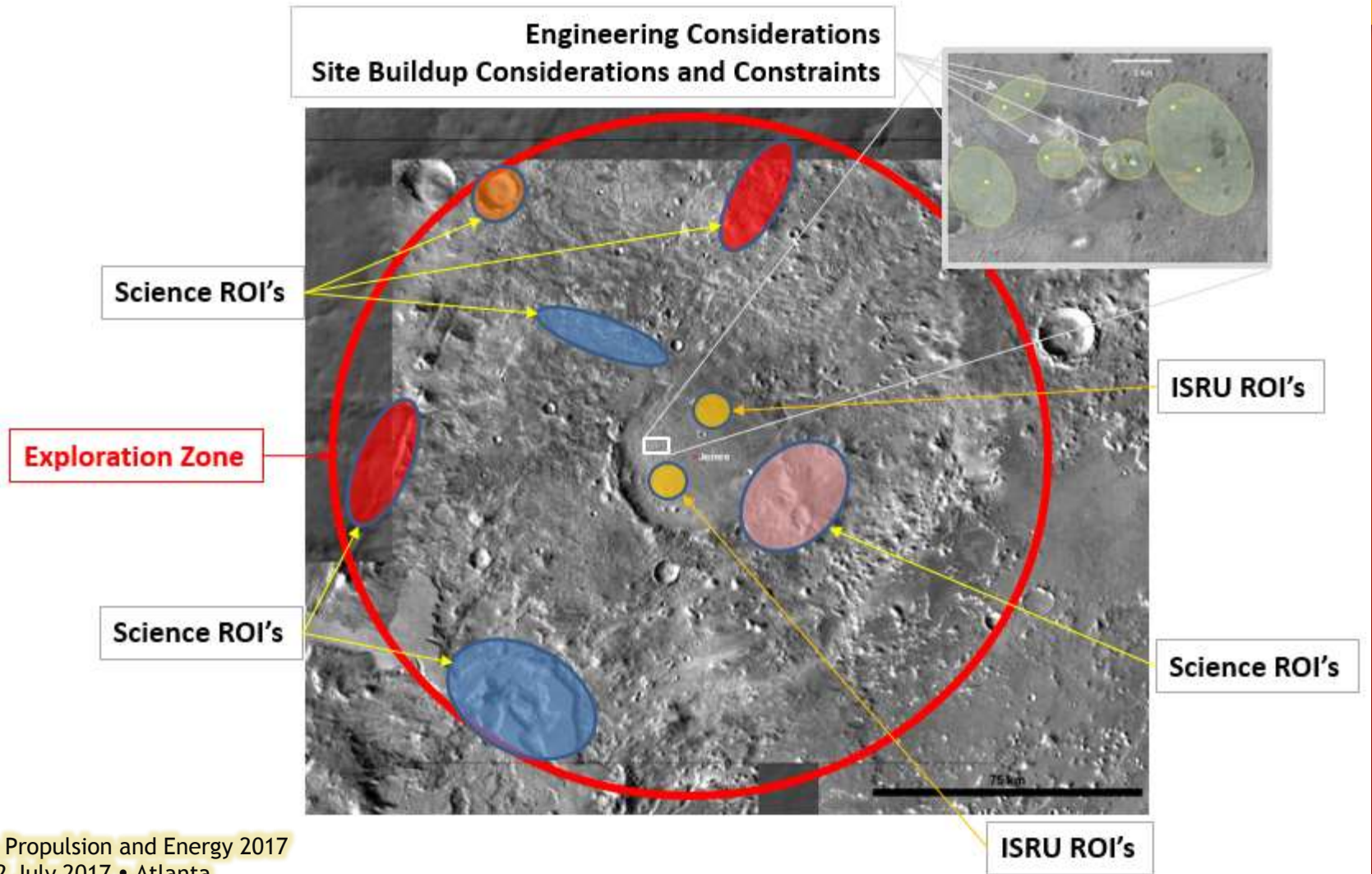


4+

Crew to the surface (and return) per expedition



The Exploration Zone (EZ), Mars Surface Field Station, and Surrounding Regions of Interest (ROI's)





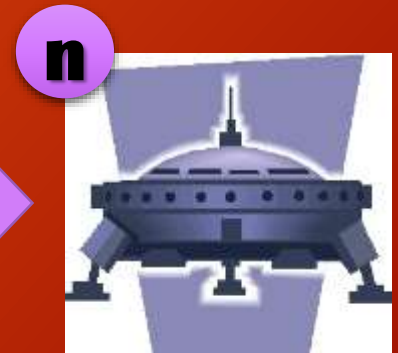
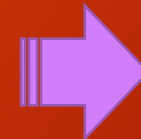
Here's What a Mars Campaign *Might* Look Like

FIRST send cargo, including a surface power system

THEN send Ascent Vehicle and ISRU to fill empty prop tanks

WHEN tanks are full, crew lands and begins surface mission

SUBSEQUENT crews land at same site, use infrastructure



Power System + Cargo



Ascent Vehicle + Propellant Manufacturing



Habitat + Crew + Cargo

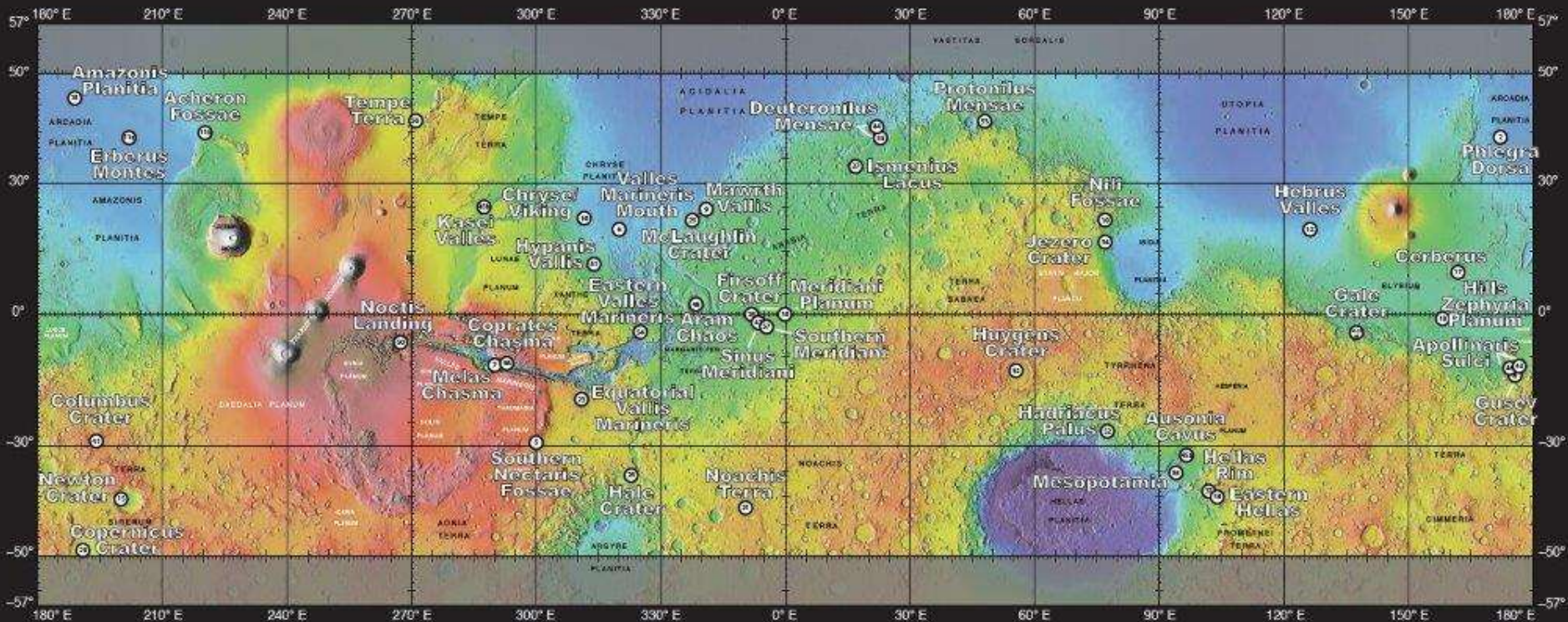


Additional Crew + Ascent Vehicles + Cargo



Preliminary Mars Surface Location Constraints for EZs

Potential Exploration Zones for Human Missions to the Surface of Mars



Exploration Zones proposed for humans to Mars. Numbers correspond to the abstract submission #. At the equator, circles are ~100km radius.

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Key To Successful Human Mars Surface Mission: Reliable Power

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What we did on the Moon won't work on Mars

	Apollo	Mars
Surface Stay	3 days	500+ days
Peak Load	~4 kW	~40 kW



- ❑ Apollo spacecraft were one-time use, each landing at a different site
 - NASA is looking at multiple missions to a single Mars landing site, with reusable surface elements
- ❑ Apollo crews ventured a few kilometers from landers
 - Mars crews may “road trip” 100+ km



Key To Successful Human Mars Surface Mission: Reliable Power

What we did with Mars rovers won't work for humans

- ❑ Sojourner, Spirit, and Opportunity
 - Fixed solar arrays and batteries
 - Arrays were sensitive to dust accumulation
- ❑ Curiosity
 - Radioisotope Thermoelectric Generator (RTG) and batteries

	Rovers	Humans
Keep-Alive	<25 W	>25 <u>kW</u>
Peak Load	<650 W	>35 <u>kW</u>

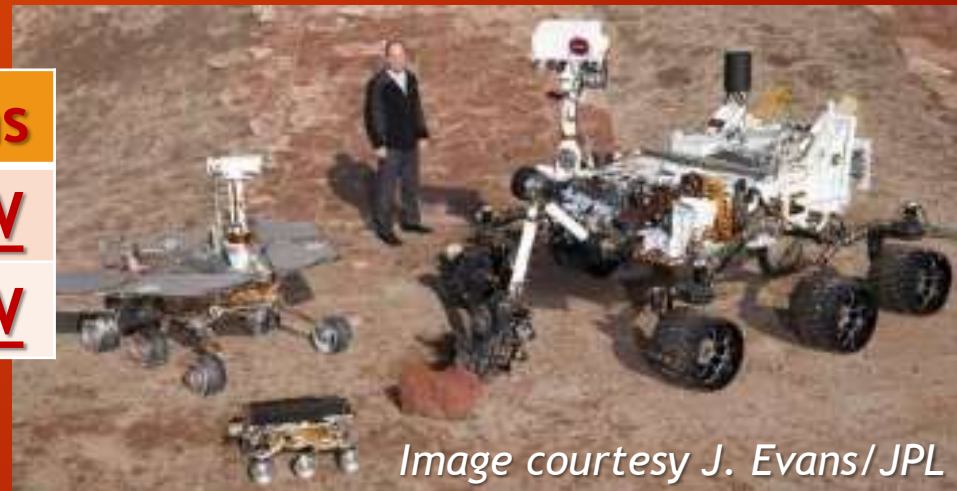


Image courtesy J. Evans/JPL



Rovers Can Hibernate When Power Isn't Readily Available

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But humans have to breathe, eat, stay warm (and possibly make return propellant)

Spirit Selfie on Sol 586

*Image courtesy of
Cornell University*



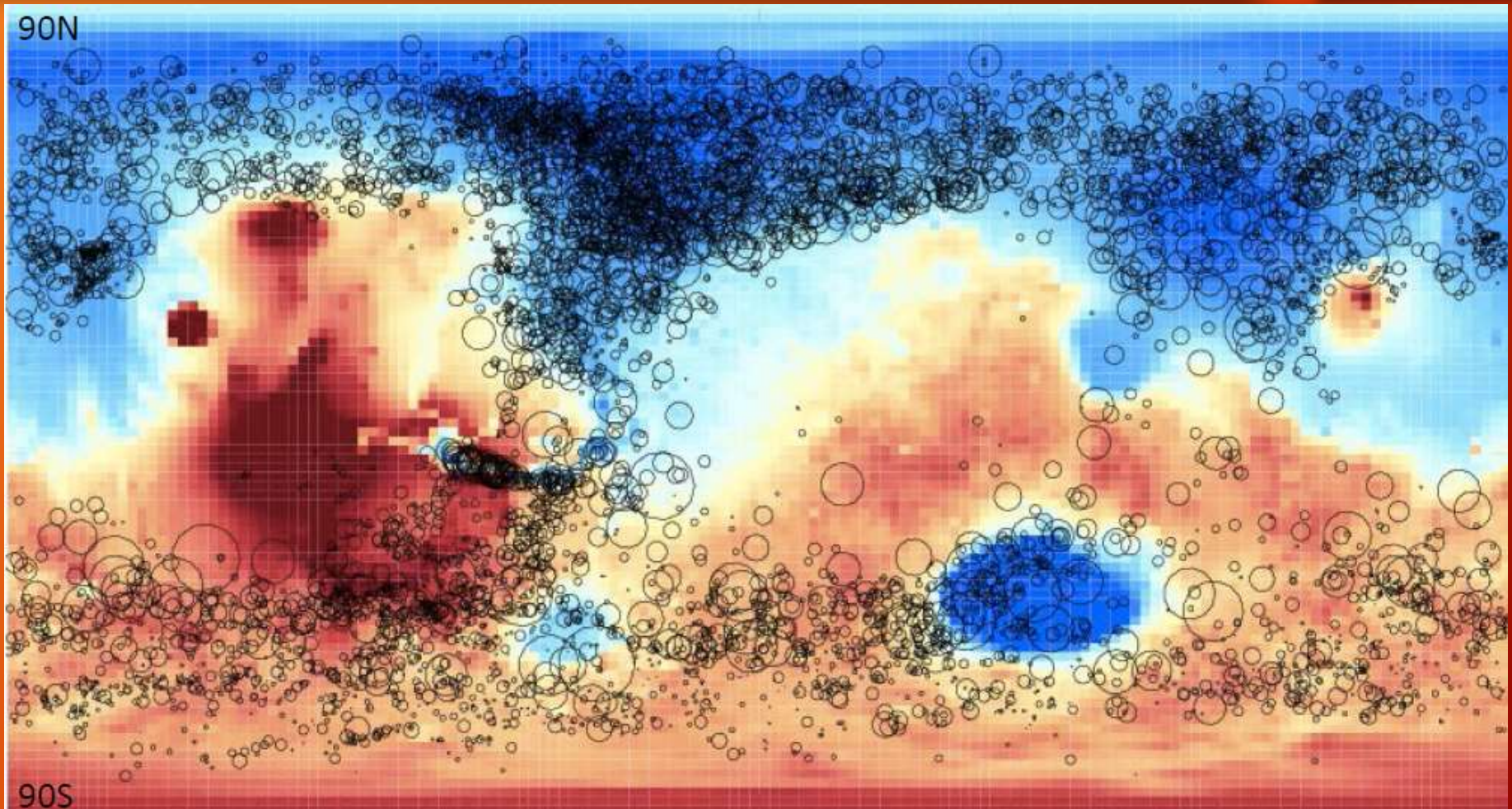
Dust Accumulation on Spirit's solar arrays reduced available power



Spatial Distribution of Dust Storms

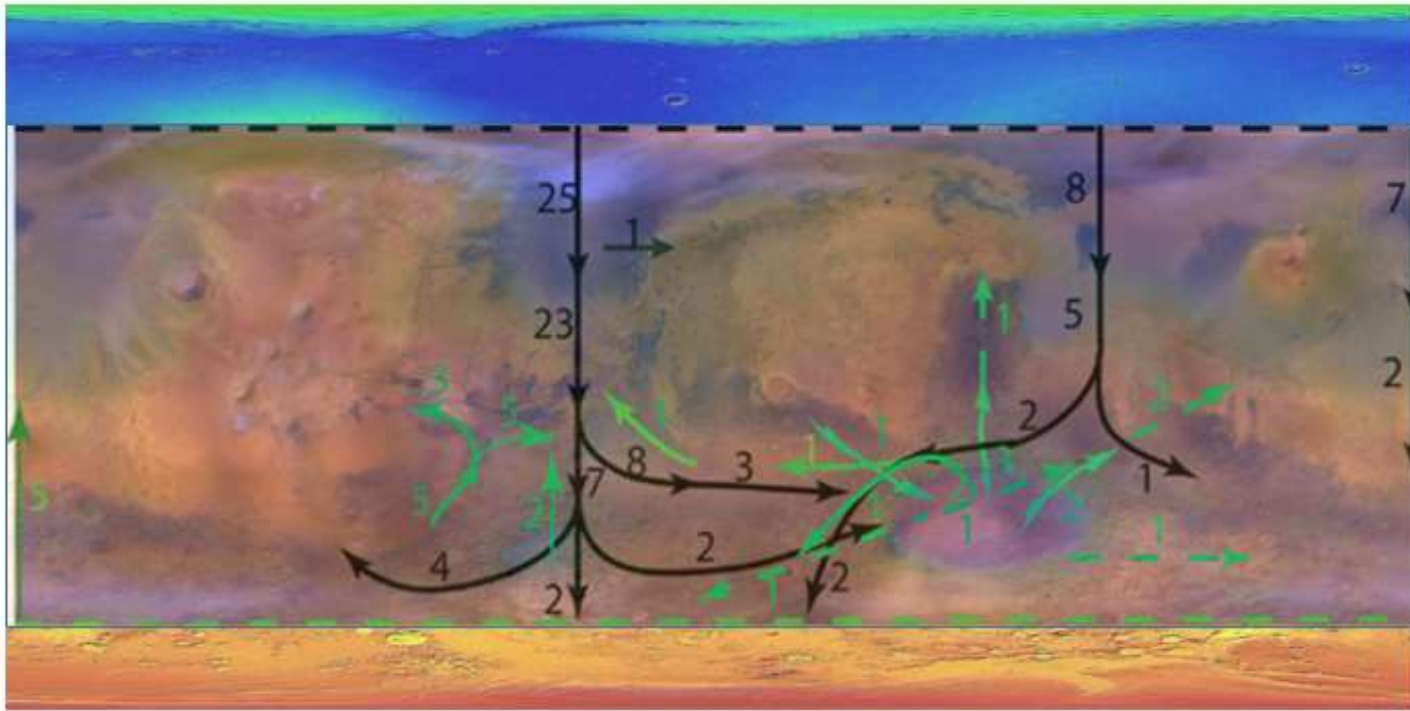
Derived from 4 Mars Years of MARCI Mars Daily Global Maps (MDGMs)

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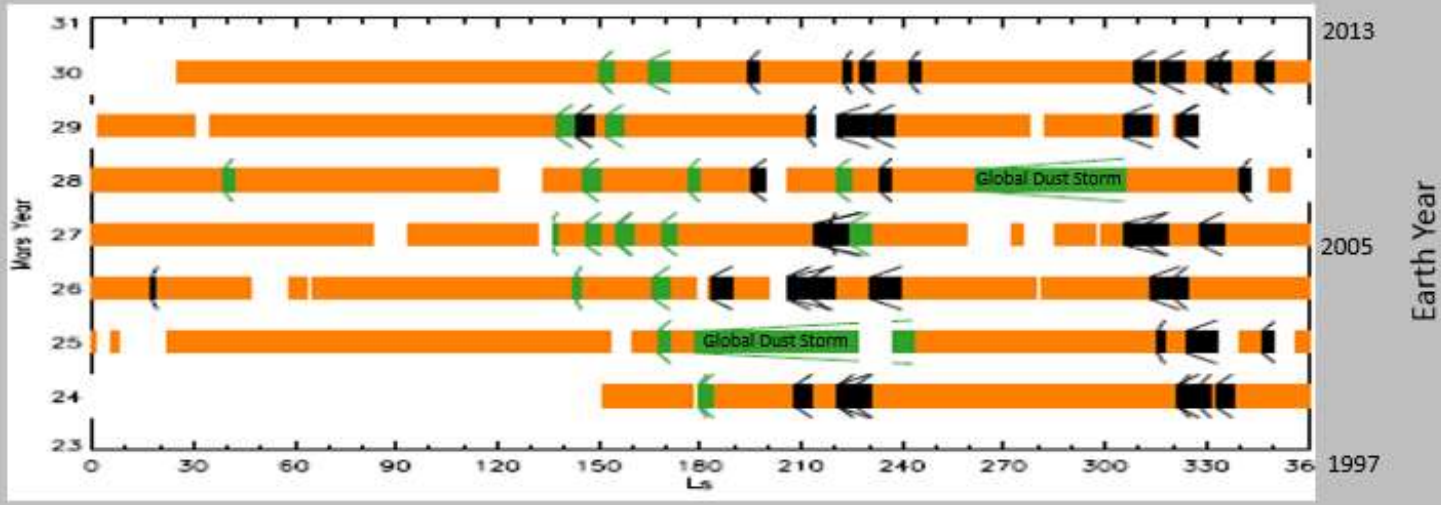
Dust Storm Ground Tracks

Numbers indicate total number of dust storms following this track between Mars years 23 and 31



Dust Storm Occurrence by Mars Year

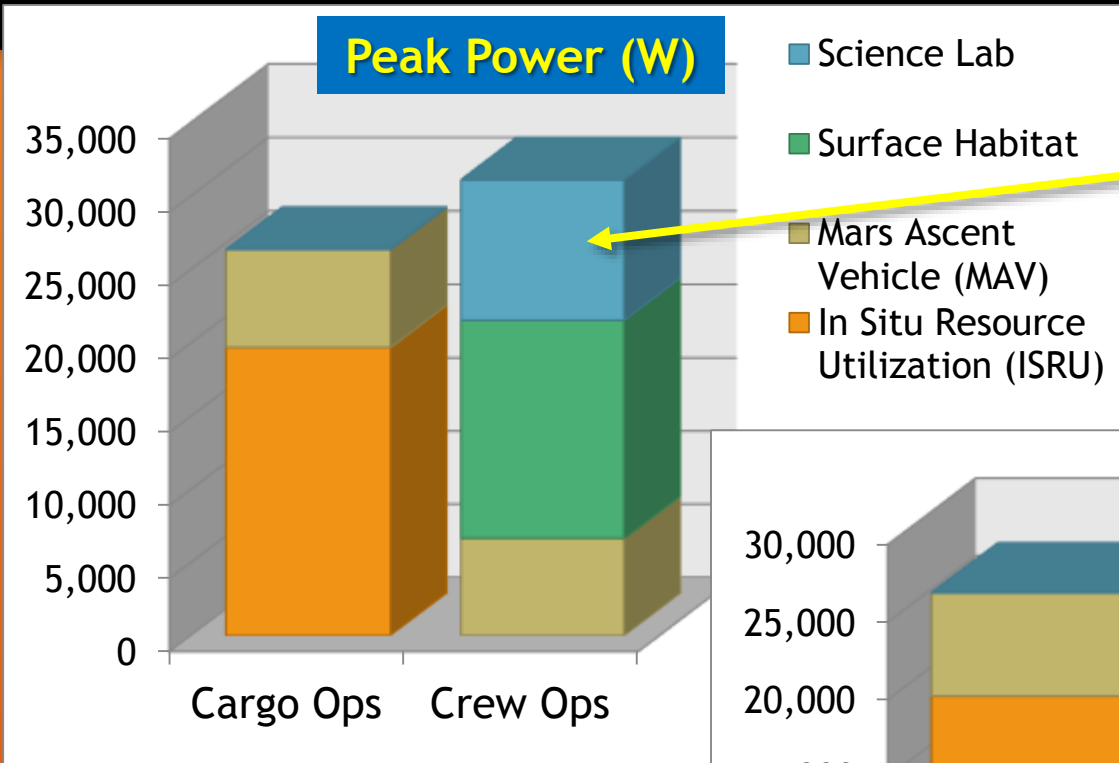
Green = southern hemisphere origin
Black = northern hemisphere origin



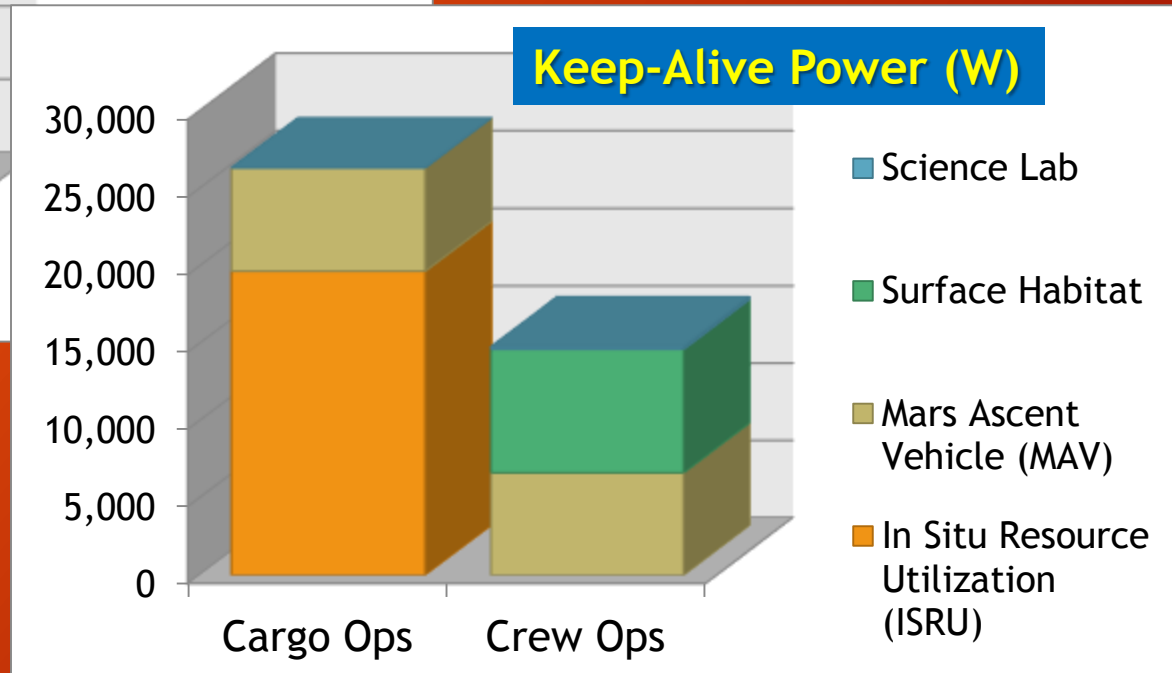


Surface Mission Power Example

4 Crew, 22.8 MT Ascent Propellant in 420 Days



Optional Science Lab with all systems running; Assume power can be phased to stay below cargo ops total peak



ISRU makes ascent propellant

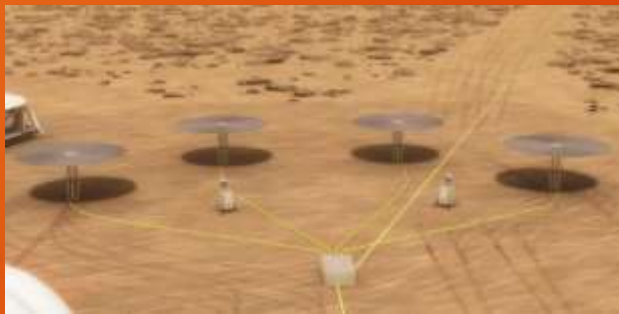


So How Do We Power a Mars Surface Mission?

Two Primary Options Being Evaluated

1

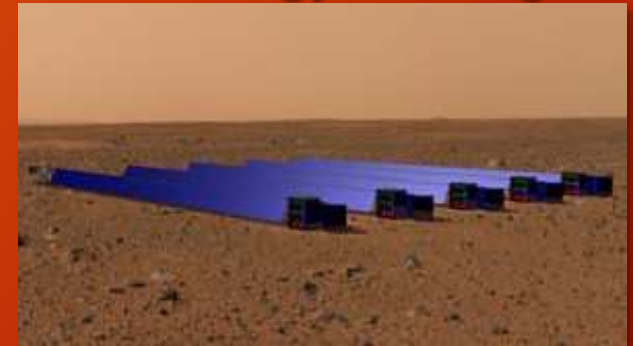
Kilopower Fission System



Modules up to 10 kWe ganged together

2

Deployable Solar Arrays with Energy Storage



Possibly with RTG emergency backup

Other options (such as wind turbines) don't trade as well for mass, volume, reliability



Mars is Like a Bowl of Spaghetti

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You can't make one decision without getting tangled up in other decisions

Example: If first decision is to pick a landing site, that dictates:



- ✓ Whether we can rely on solar power for surface operations
- ✓ Whether Mars water is readily available to make reactants for fuel cell energy storage
- ✓ Dust mitigation needs for seasonal dust storms



Every technology decision has implications to other technology decisions



Fission vs. Solar

		 Fission Power	 Solar Power
Readiness		Lower	Higher
Mass		Lower for most sites	Higher
Reliability	At night	High	Need energy storage
	Dust storm	High	Risky
	At equator	High	Good
	At poles	High	Must be large
Deployment Time		Lower	Depends on size
Cost		High development, handling overhead	Lower
Portability		Higher (compact)	Lower



Energy Storage For Night Use + Portable Equipment

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Two Primary Options Being Evaluated

Goal: high energy density, long shelf life (4+ years), long service life (10-12 years), low maintenance, no Earth-origin resupply/consumables

1

Rechargeable Batteries



2

Fuel Cells



Reactants ideally from
Martian resources

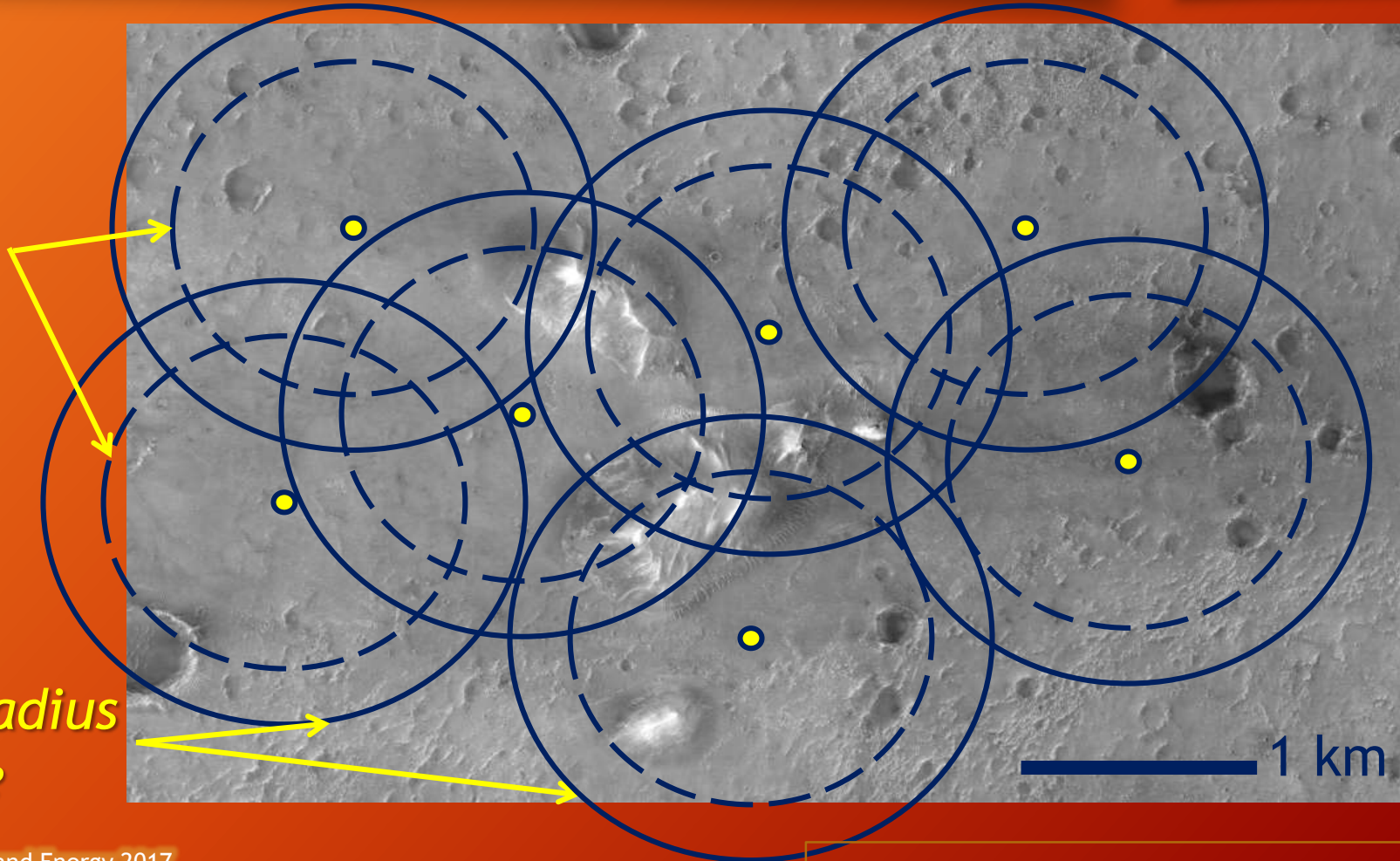


Managing Multiple Landers

Close—but not too close—to each other

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700 m
descent
plume
hazard



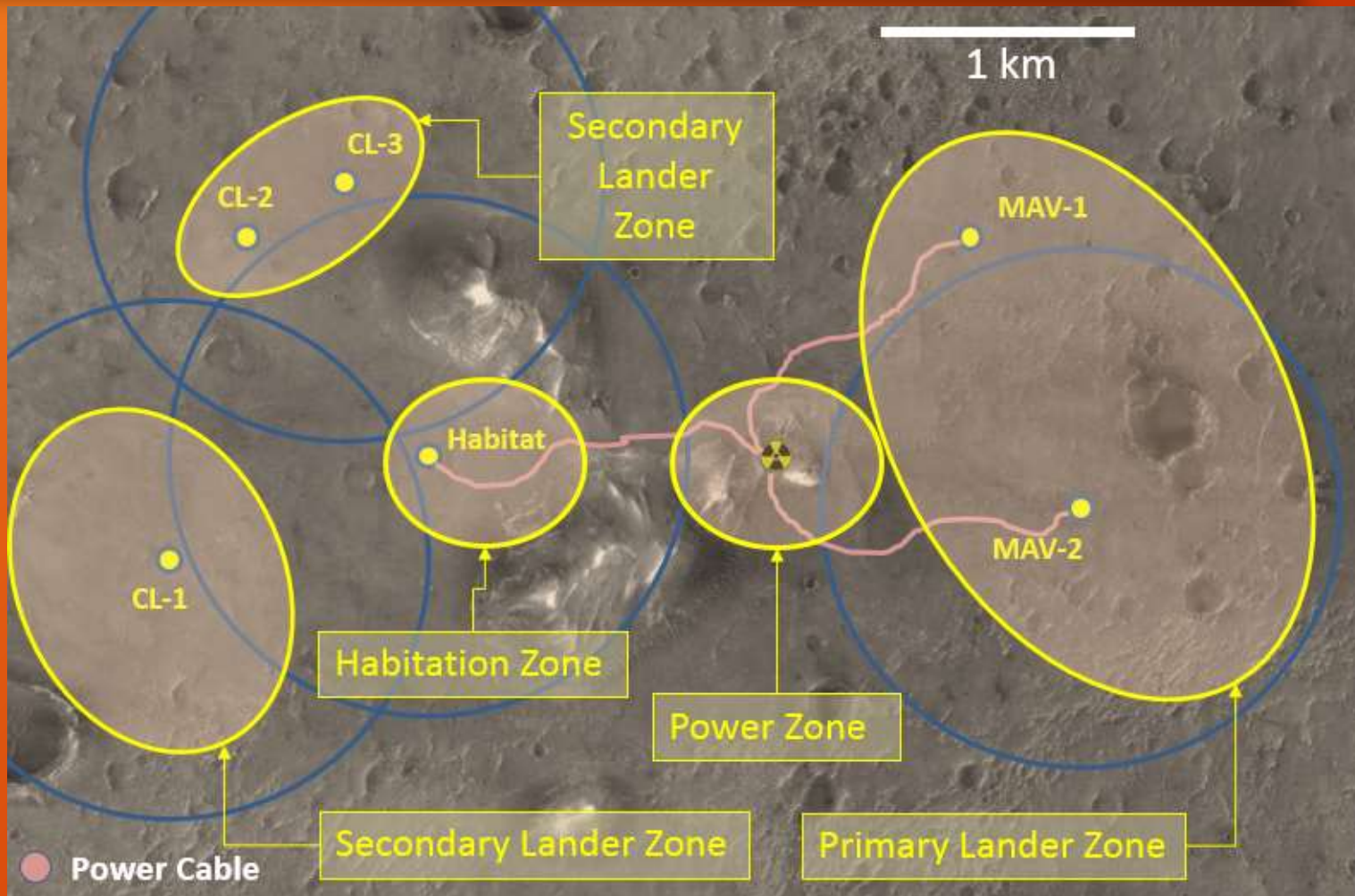
1000 m radius
safe zone

• 100 m dia landing site



Example of Field Station Layout with Specific Utilization Zones Identified

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Multiple Landers Complicates Power Management & Distribution

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We'll need to power multiple elements, up to 1 km apart (*some before crew arrive*)

Autonomously Deployed Cable



Robotic Assembly



Multi-Element Power Management



Options include a single power lander, power generated at every lander, and/or distributed “charging stations”



Mars Power R&D Opportunities

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- **Power Management and Distribution (PMAD)**
 - Tough, long life, high voltage power cable compatible with Martian temperatures/low pressure
 - Autonomous cable deployment (up to 1 km)
 - Robotically actuated, dust-resistant connectors

- **Energy Storage**
 - High energy density
 - Long shelf life (4+ year)
 - Long service life (10-12 years)
 - Low maintenance
 - No Earth-origin resupply/consumables



Mars Power R&D Opportunities

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☐ Solar Power

- Dust mitigation for solar arrays
- Lightweight solar array structures
- Autonomous solar array deployment mechanisms
- High efficiency solar cells

☐ Fission Power

- Compact fission reactor
- Stirling development
- Heat pipe/core bonding
- Dust mitigation for radiator panels



Key Take-Aways

Human Mars Mission Power

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Modular	30-40 kWe
Infrastructure Build-Up	High Peak & Keep-Alive Power
Reliable	Portable
500-Day Missions >12-Year Life	100+ km Excursions

Exciting Mars research & development opportunities await!

AIAA Propulsion and Energy 2017
10-12 July 2017 • Atlanta



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Questions?

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