

Initial Study of the

Fundamentals of Ice Crystal Icing Physics in the NASA Propulsion Systems Laboratory

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Outline

- Introduction & background
- NASA Fundamental Ice Crystal Icing Research Goals
 - Concepts using the NASA Propulsion System Laboratory (PSL)
- Experimental Description
- Results
 - Freeze-out characteristics of cloud
 - Changes in aero-thermal conditions at the test section
 - Accreted ice characteristics observed
- Summary



Introduction

- NASA investigating the fundamental physics of ice crystal icing (ICI)
- Challenging to study ice-accretion physics directly inside the engine
 - Trying to simulate local ICI environment without using an engine
- This paper presents an initial study of the fundamental physics of ICI using PSL
 - Test occurred in March 2016
 - Select results presented
 - Last year, presented preliminary work in preparation for this test
 - Complementary papers to follow

Advance Air Transport Technology
Project (AATT; 2015 +)
Advanced Aircraft Icing (AAI) Subproject

Technical Challenge:

Expand engine aero-thermodynamic modeling capability to predictively assess the onset of icing in current and N+2/N+3 aircraft during flight operation (FY21).

The simulation tools are well anchored in results from both fundamental physics studies and full engine tests.



NASA Fundamental ICI Research Goals

Ice

Identify and bound the conditions affecting icecrystal ice accretion at the (local) accretion site

Generate & characterize (i.e. measure) those conditions

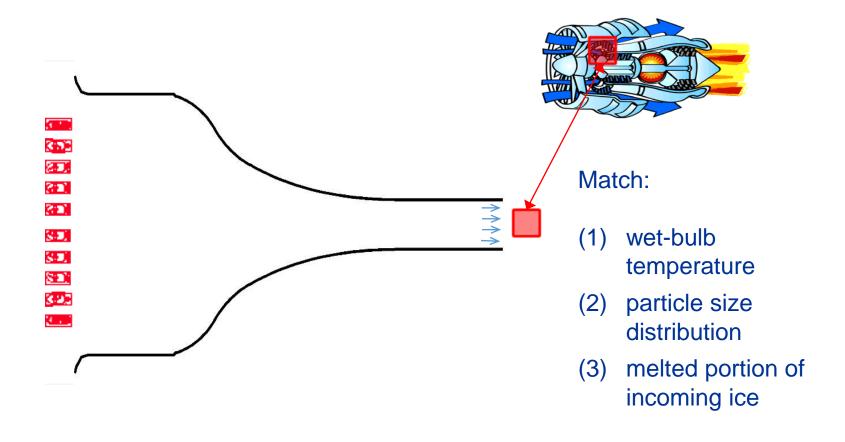
Gather data and develop models on ice-crystal icing factors

Graphic: NASA (www.ueet.nasa.gov) Crystal Icing **Engine Tests** Local Region Link actual conditions in an engine to fundamental work IWC, LWC Control volume on icing surface Graphic: NASA

Local region requires more information than full-scale test (e.g. melt ratio)



Concept Using PSL



Goal: Ability to generate a prescribed mixed-phase condition at the test section for fundamental ice-crystal icing research

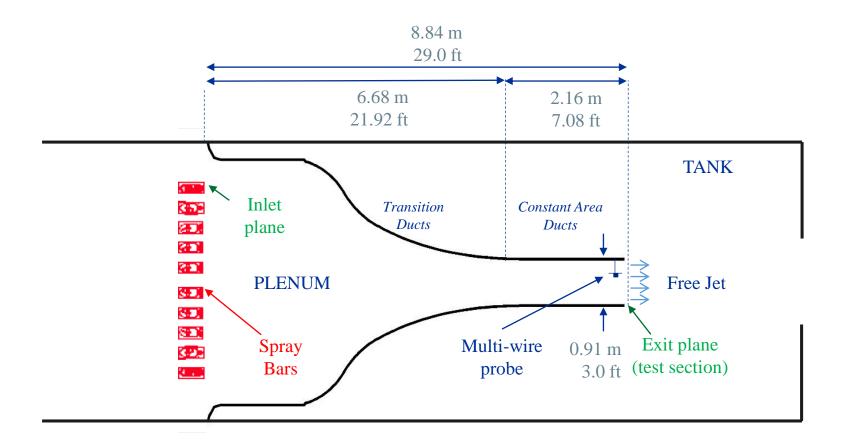


Fundamental Test #1

- Eight (8) days of testing occurred in March 2016
- Objectives
 - Examine spray bar and plenum parameters and how they affect the mixedphase at the exit of the free jet
 - Cloud characterization at the test section:
 - Melt ratio (fraction of freeze out)
 - Total water content
 - Temperature & humidity measurements at test section (cloud on vs. cloud off)
 - Particle size distributions
 - Uniformity
 - Observe ice accretion



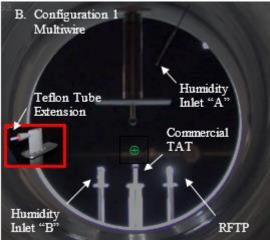
PSL Configuration

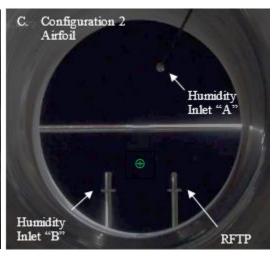


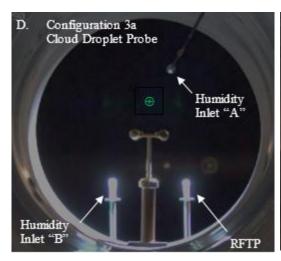


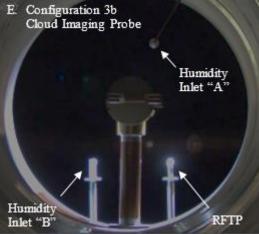
Test Configurations

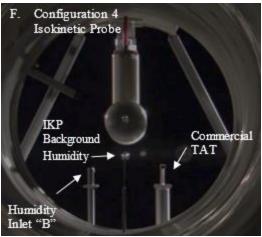










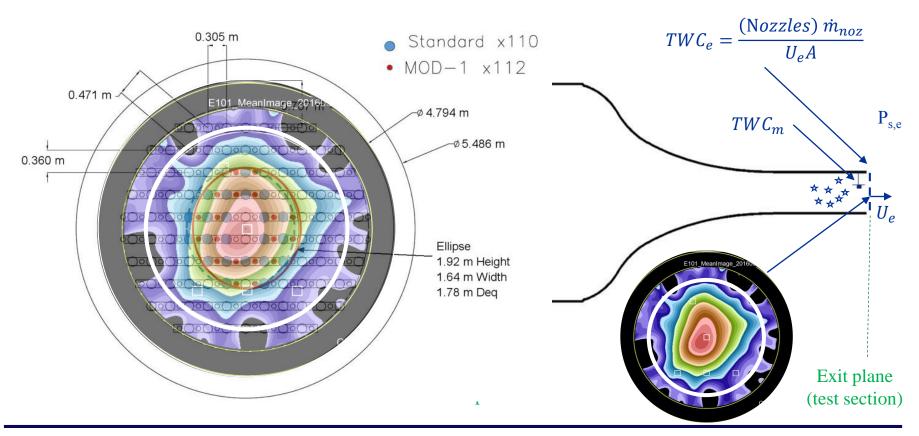




Plenum Relative Humidity Sweep Approach

Parameters

Nomenclature

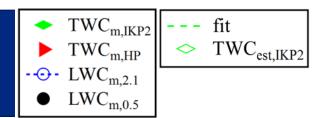


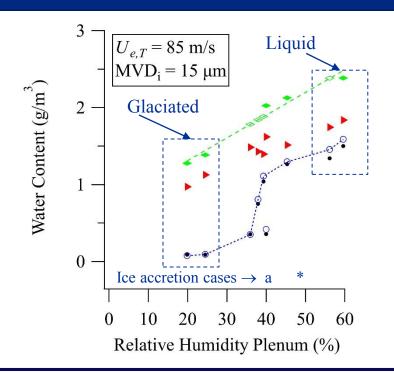


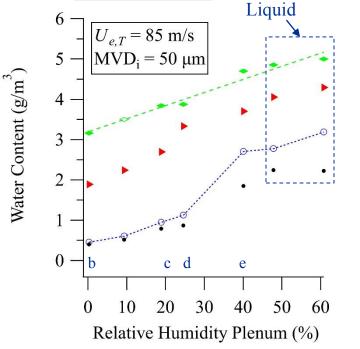
Plenum Relative Humidity Sweep Results

Water Content Measurement Results

 $P_{0.i.T}$ = 44.8 kPa (6.5 psia) and 42.8 kPa (6.21 psia) $T_{0,i,T} = 7.2^{\circ}C$ $TWC_{e,T} = 6.5 \text{ g/m}^3 \text{ (* Estimated)}$





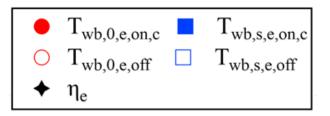


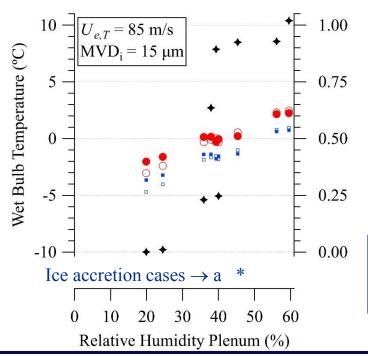


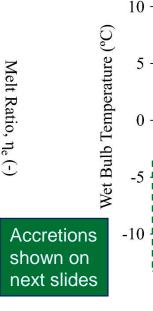
Plenum Relative Humidity Sweep Results

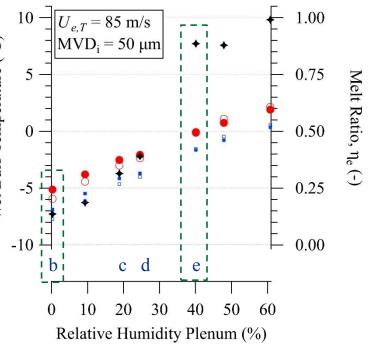
Test Section T_{wb} (static) and Melt Ratio

 $P_{0.i.T}$ = 44.8 kPa (6.5 psia) and 42.8 kPa (6.21 psia) $T_{0,i,T} = 7.2^{\circ}C$ $TWC_{e,T} = 6.5 \text{ g/m}^3 \text{ (* Estimated)}$









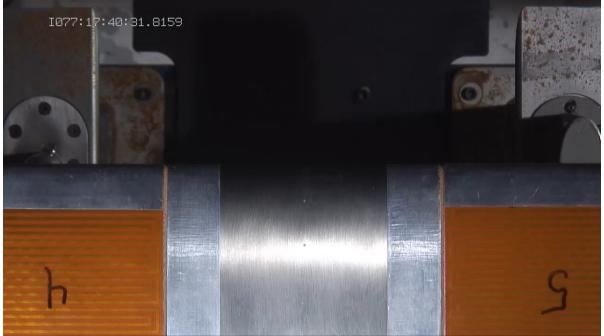


Ice Accretion Examples

Accretion "b" Low melt ratio

8x actual speed (10 minute spray time)

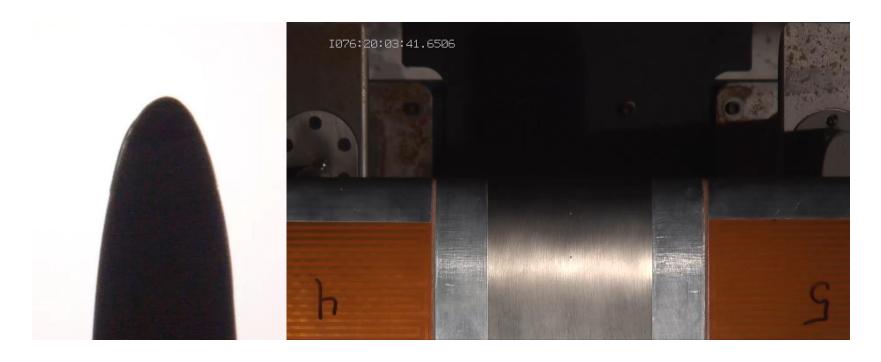






Ice Accretion Examples

Accretion "e" High melt ratio 8x actual speed (10 minute spray time)

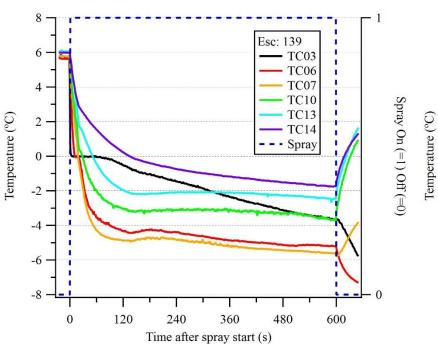


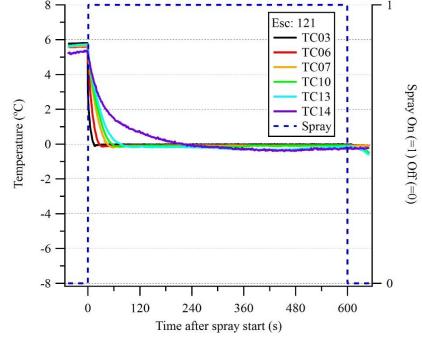


Surface TC meaurements

Accretion "b" Low melt ratio

Accretion "e" High melt ratio







Summary

- NASA conducting research on fundamentals of ICI:
 - Identify and bound the conditions at the (local) accretion site
 - Generate & characterize conditions
 - Develop models & gather data on ice-crystal icing factors
- Generate environment outside of an engine to facilitate study
 - Using PSL as test bed
- Presented data from an 8-day test effort in March 2016, examining:
 - Freeze-out characteristics of cloud
 - Changes in aero-thermal conditions at the test section
 - Ice characteristics observed
- These result offer modelers a dataset to help develop and validate icecrystal, mixed-phase accretion models.



Acknowledgement

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 - NASA's, Advanced Aircraft Icing (AAI) Subproject
 - Mr. Tony Nerone, Project Manager
- Special thanks to:
 - Staff of the NASA PSL
 - Mr. Chris Lynch for his excellent imaging work.



Questions



Backup Slides



Measurements

- **Temperature**
 - Rearward Facing Temperature Probe
 - Commercial TAT Sensor
- Humidity
 - Spectra sensor WVSS-II
- Pressure / velocity / Mach
- Uniformity
 - Traverse RTFP
 - Temperature
 - Humidity
 - Condensed phase water
 - PSL Tomography

- Total water content
 - Isokinetic Probe version
- Liquid water content
 - SEA Multi-Element Probe
- Particle size distributions
 - Cloud Droplet Probe (CDP)
 - Cloud Imaging Probe (CIP)
 - High Speed Imager (HSI)
 - Phase Doppler Interferometer (PDI)
- Video cameras recorded ice accretion



Plenum Relative Humidity Sweep Results

Test Section Changes is T and ω when cloud activiated

 $P_{0.i.T}$ = 44.8 kPa (6.5 psia) and 42.8 kPa (6.21 psia) $T_{0,i,T} = 7.2^{\circ}C$

 $TWC_{e,T} = 6.5 \text{ g/m}^3 \text{ (* Estimated)}$

