

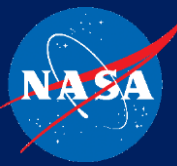
Temporal Evolution of the Plasma Sheath Surrounding Solar Cells in Low Earth Orbit



Image Credit: NASA

Emily M Willis and Maria Z. A. Pour

Outline



- Solar Array Operations Overview
- Standard ISS Floating Potential Observations
- Transient ISS Floating Potential Observations
- Particle in Cell Simulation
- Lumped Element Model
- Conclusion and Forward Work

Solar Array Operations Overview

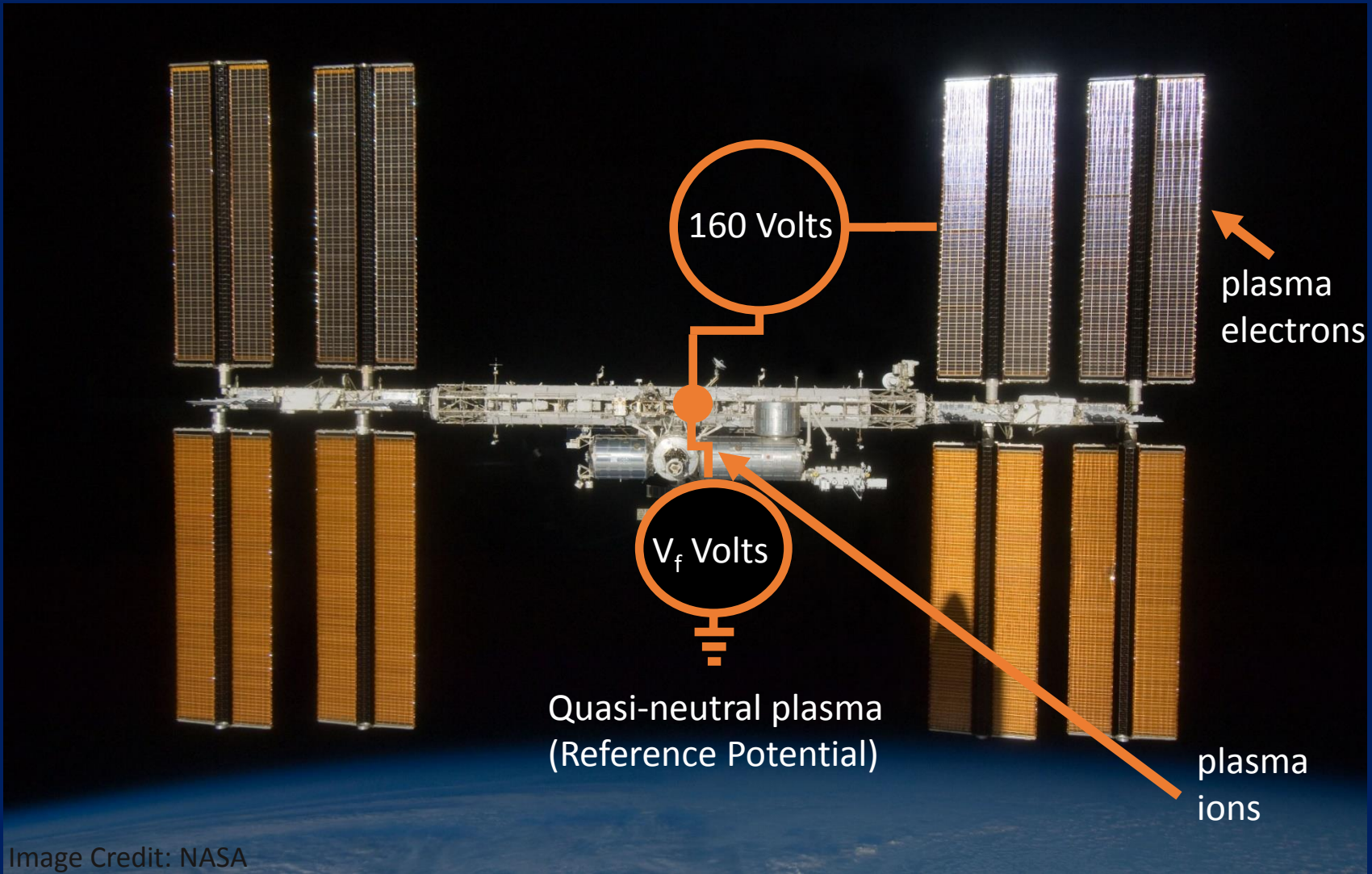
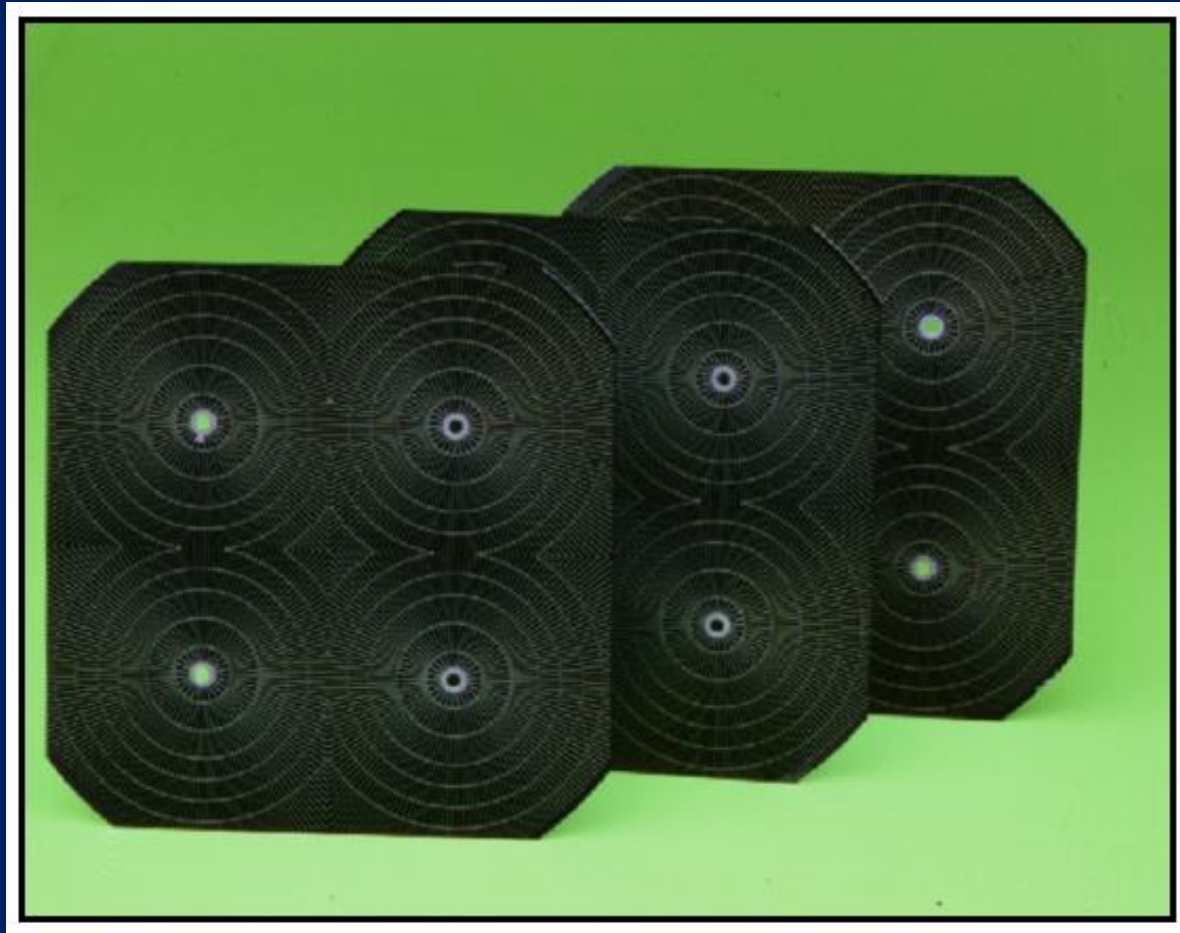


Image Credit: NASA

V_f is the floating potential of the ISS

Solar Cells



8 cm x 8 cm

plasma electrons



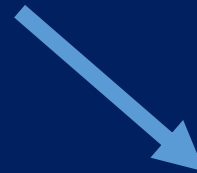
coverglass

coverglass

Silicon (160 Volts)

Silicon (160 Volts)

Kapton



coverglass

coverglass

Silicon ($160 - V_f$ Volts)

Silicon ($160 - V_f$ Volts)

Kapton

Normal

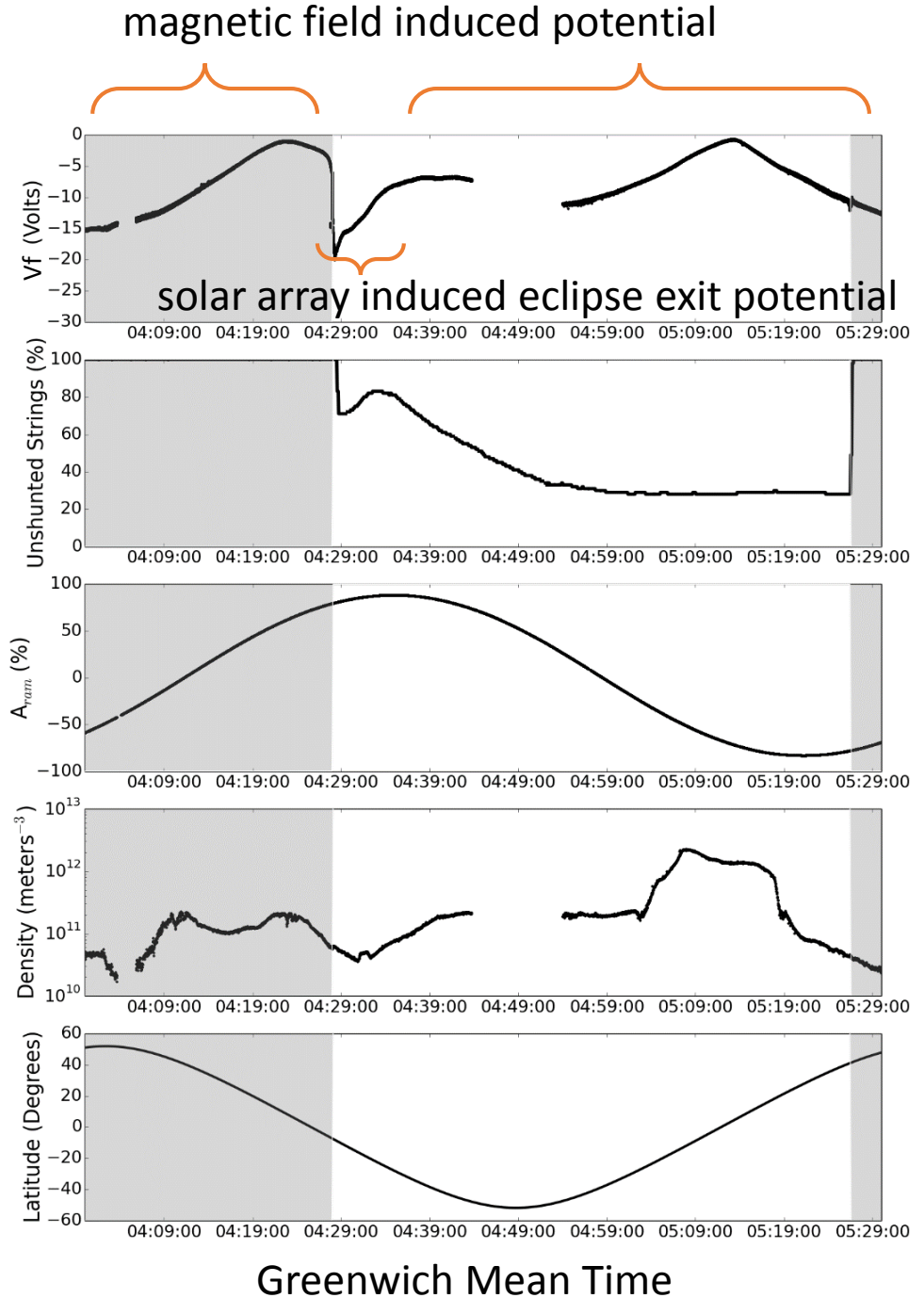
a) Floating Potential →

b) Active Array Strings →

c) Array Orientation →

d) Plasma Density →

e) ISS Latitude →



Transients

a) Floating Potential →

b) Active Array Strings →

c) Array Orientation →

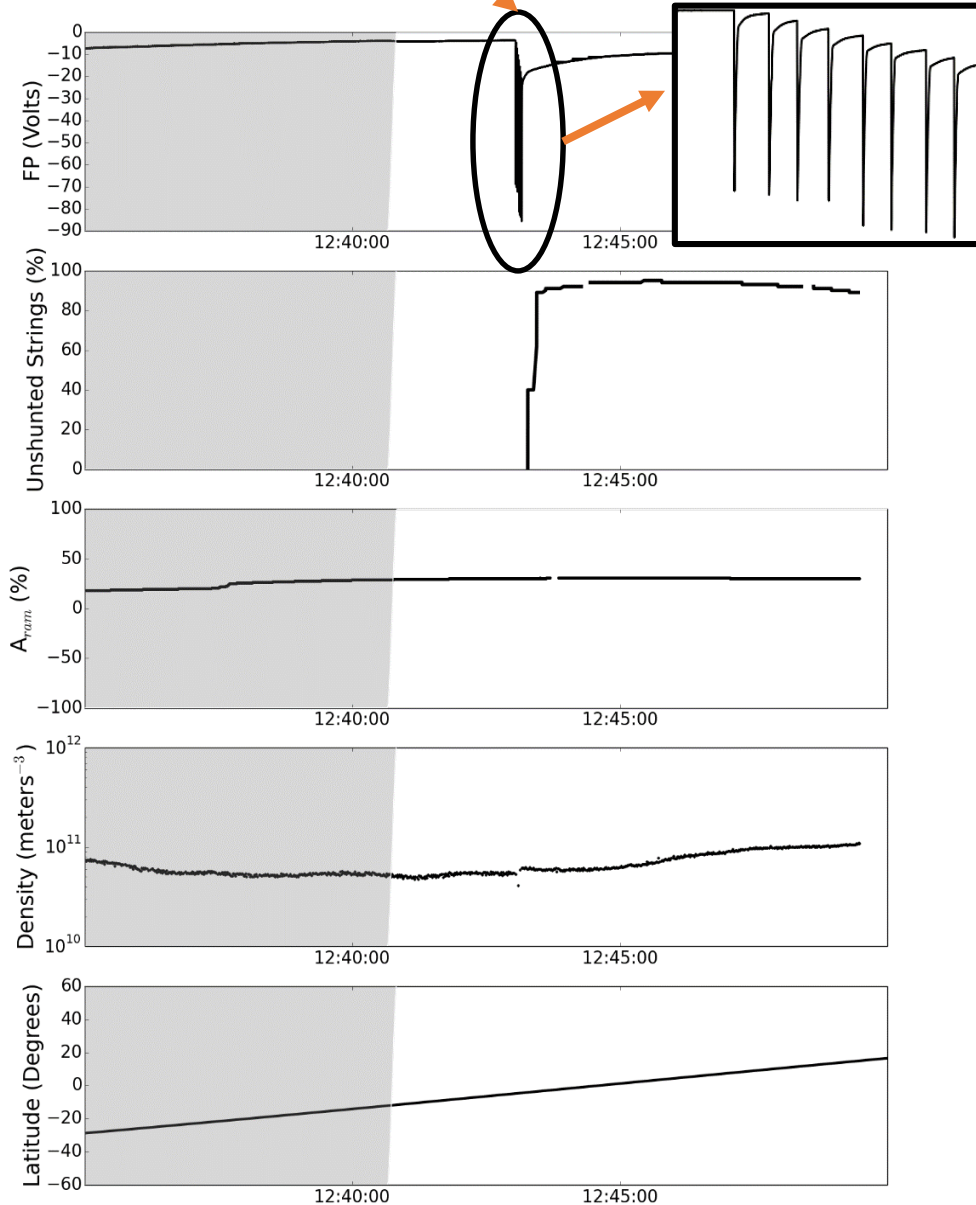
d) Plasma Density →

e) ISS Latitude →

Each of eight arrays

unshunted in full sunlight

close-up



Greenwich Mean Time

Transients

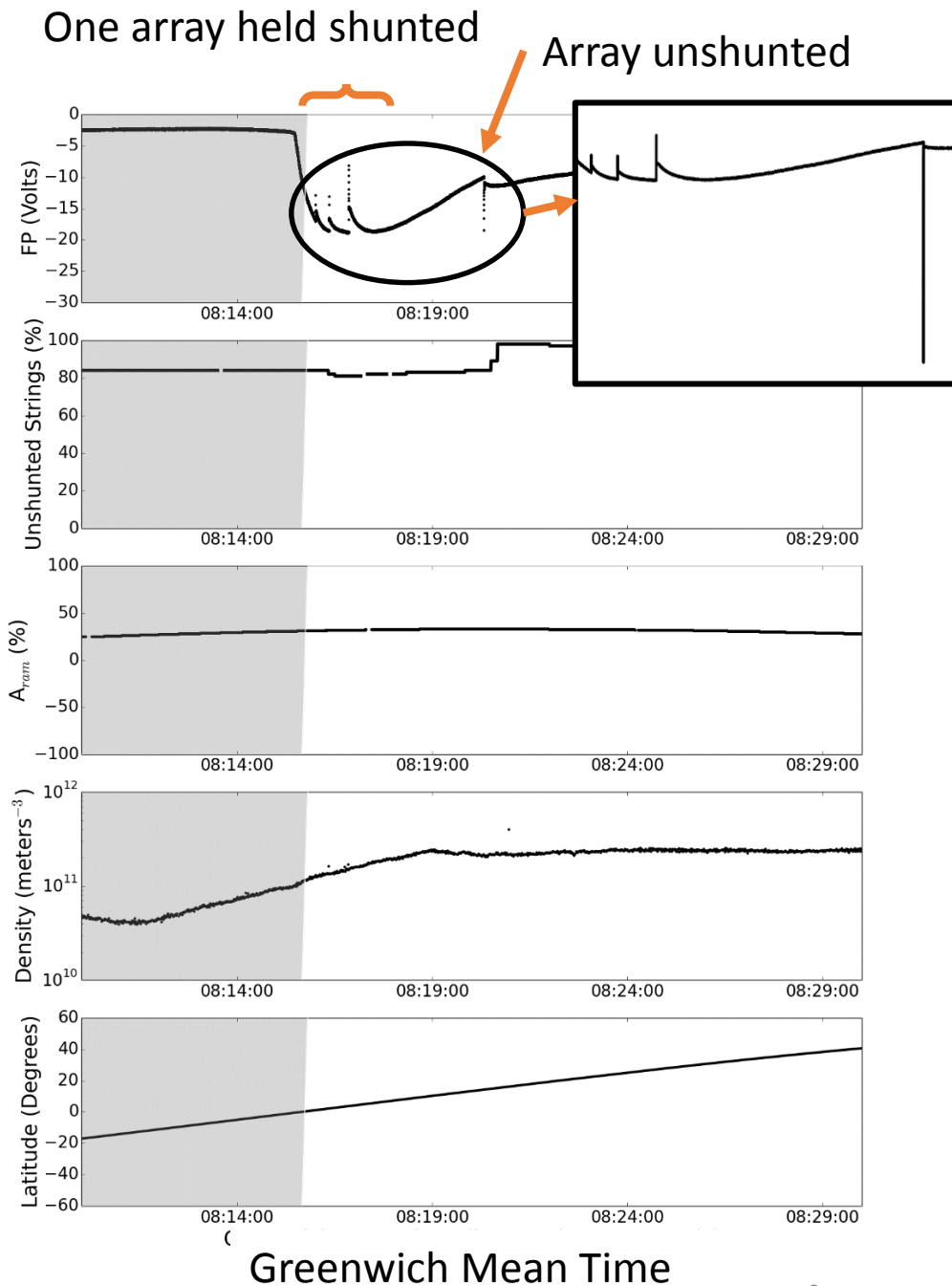
a) Floating Potential →

b) Active Array Strings →

c) Array Orientation →

d) Plasma Density →

e) ISS Latitude →



Transients

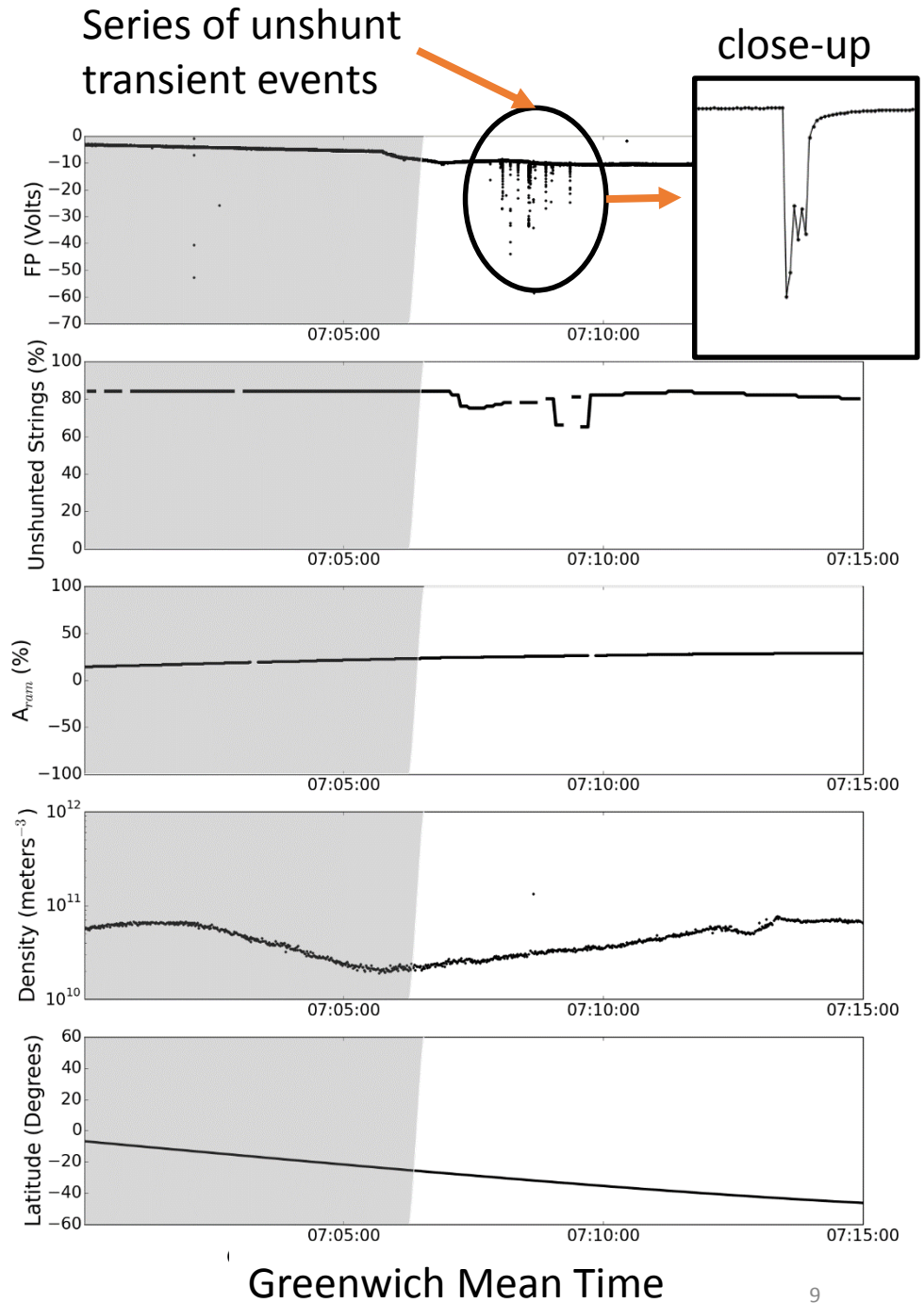
a) Floating Potential →

b) Active Array Strings →

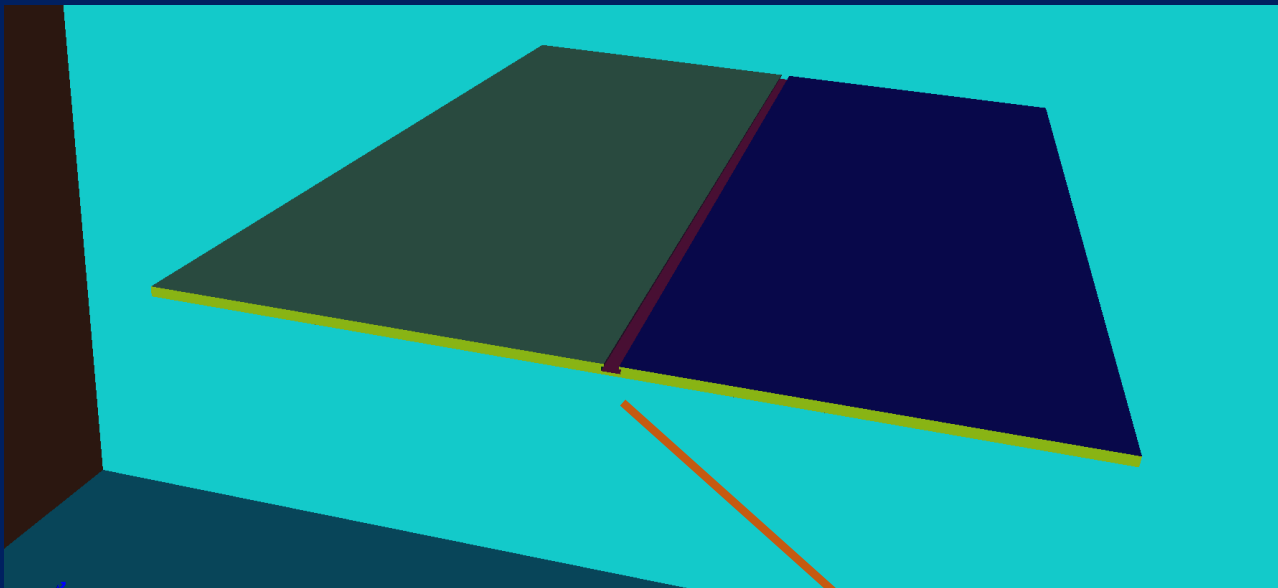
c) Array Orientation →

d) Plasma Density →

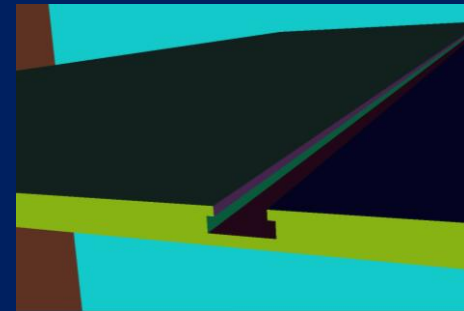
e) ISS Latitude →



SPIS Setup – Steady State

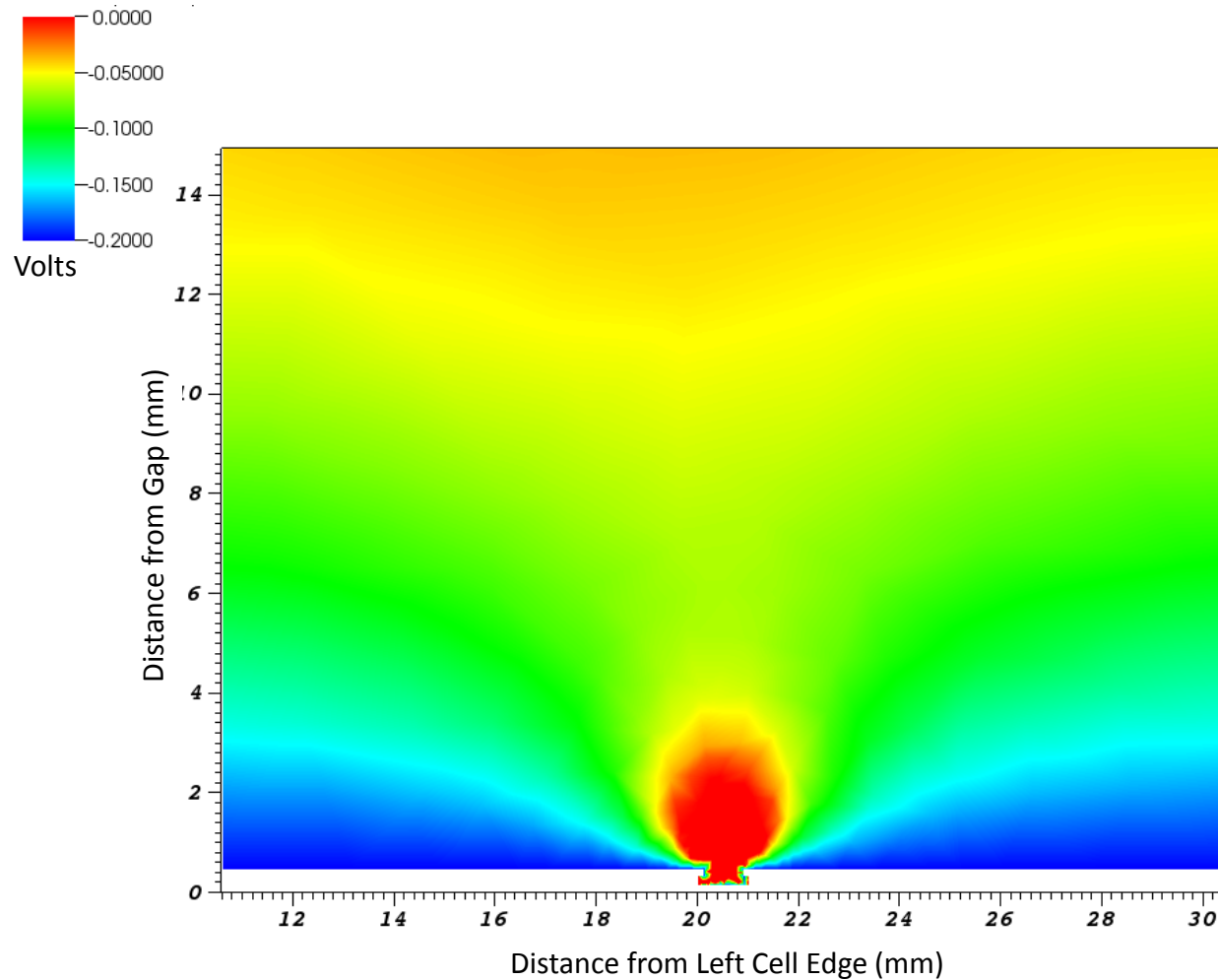


Coverglass surface set to -0.2 Volts.
Solar Cell set to 150 Volts
Plasma Density $1E11 \text{ m}^{-3}$
Plasma Temperature 0.1eV
Cell area $2 \text{ cm} \times 4 \text{ cm}$
Cell spacing 0.8mm
Coverglass overhang 0.2mm



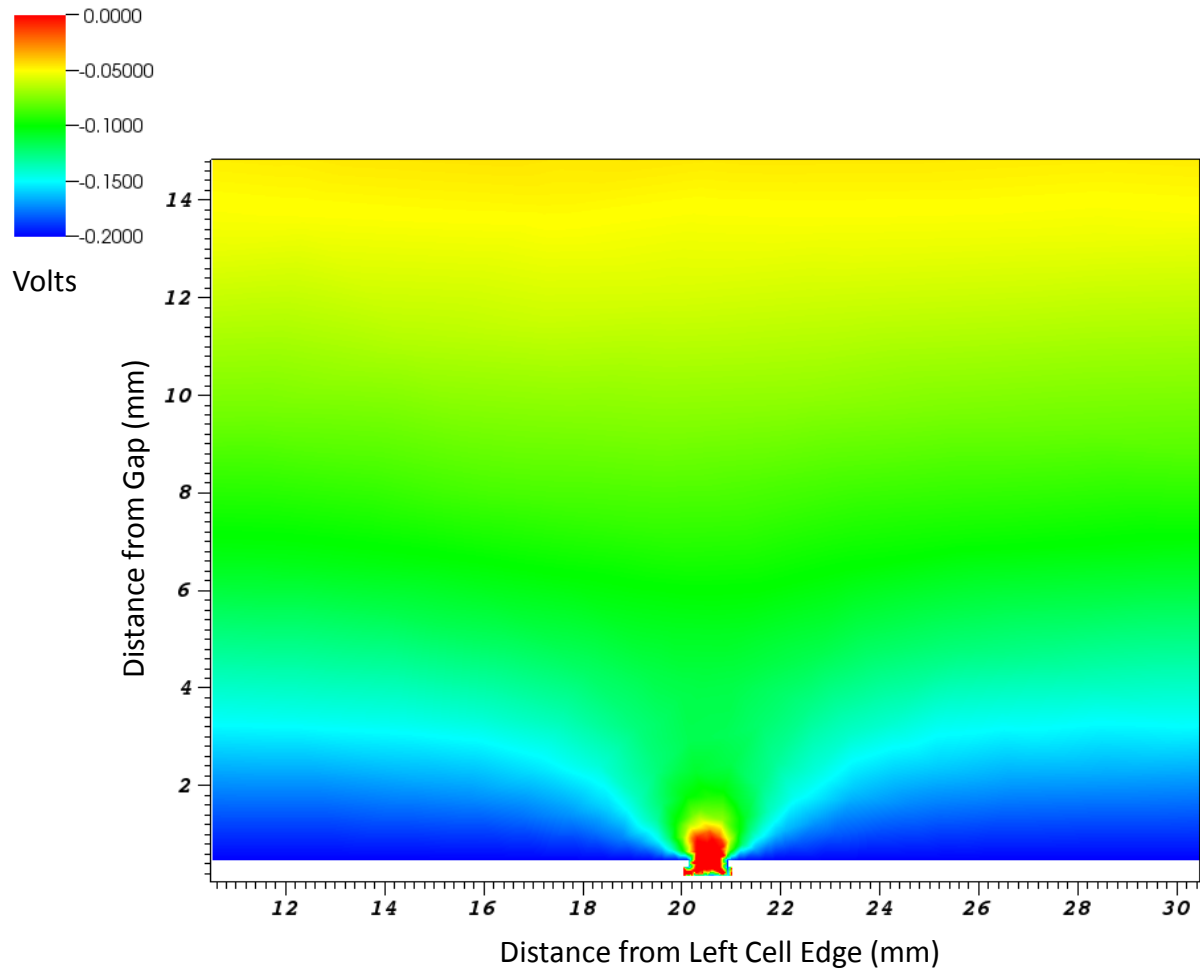
SPIS Results for 150 V Cell at Steady-state

barrier to electron
collection of
of -0.07V at 6mm
from the gap.

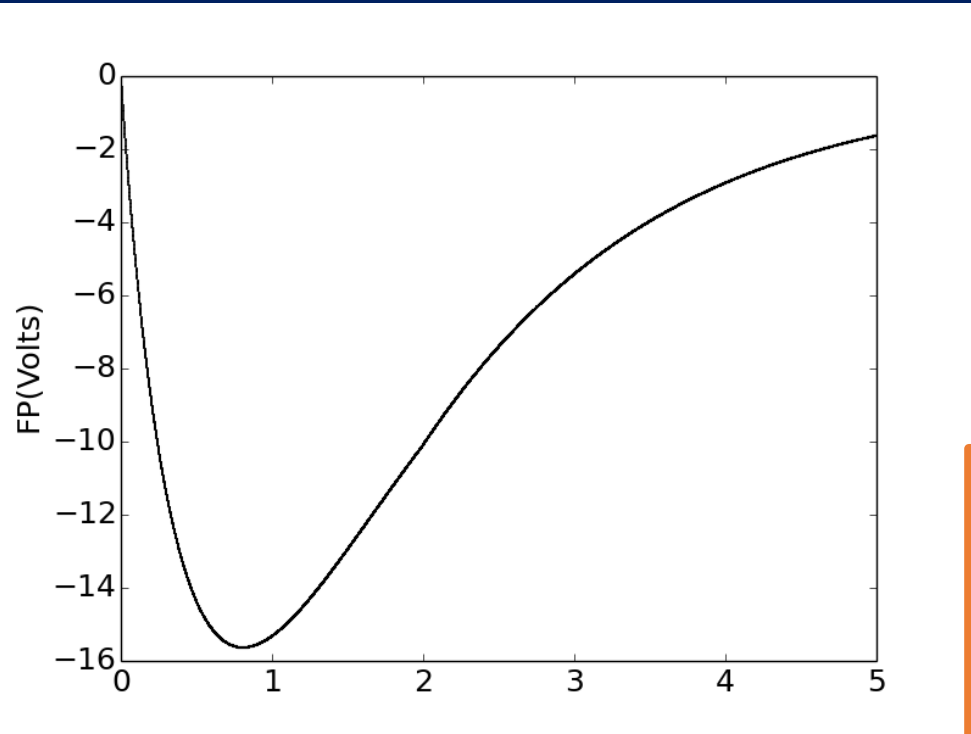


SPIS Results for 50 V Cell at Steady-state

barrier to electron
collection of
of -0.12V at 3mm
from the gap.

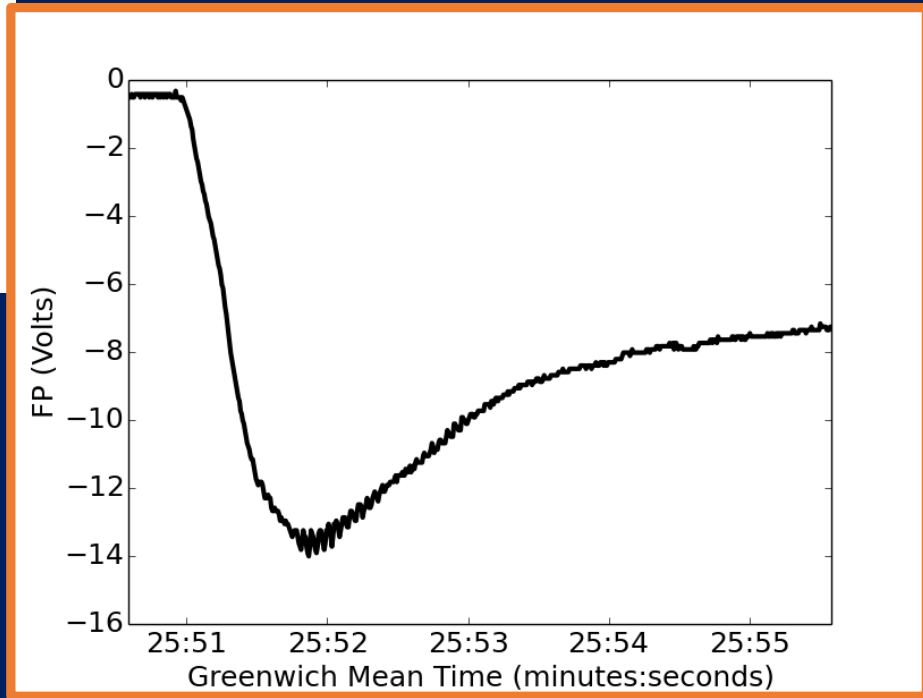


Existing Current Balance Model for Rapid Charging¹⁻²

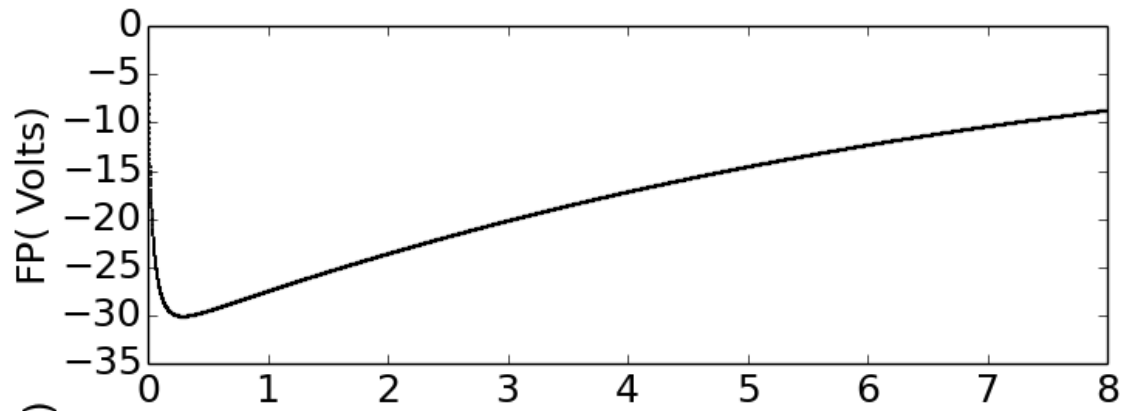


← Model Output

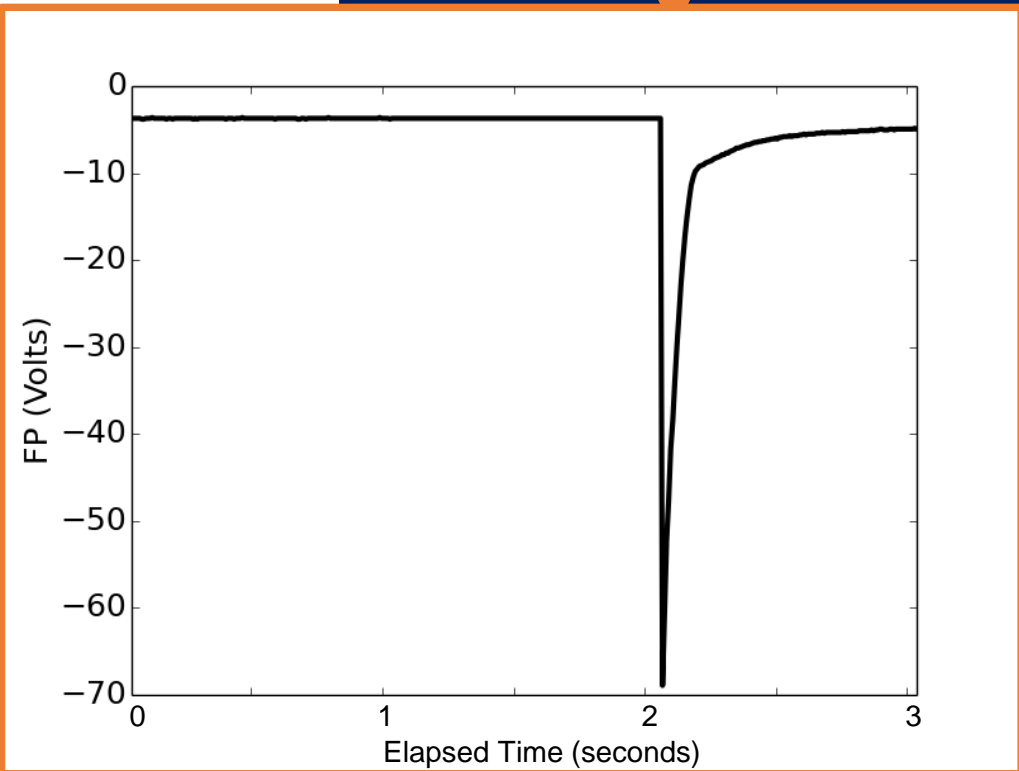
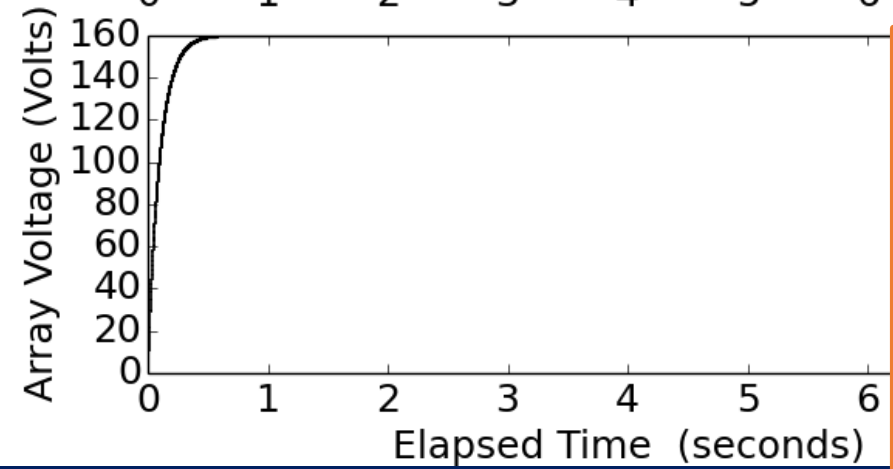
FPMU Data



Model output agrees well
with FPMU data

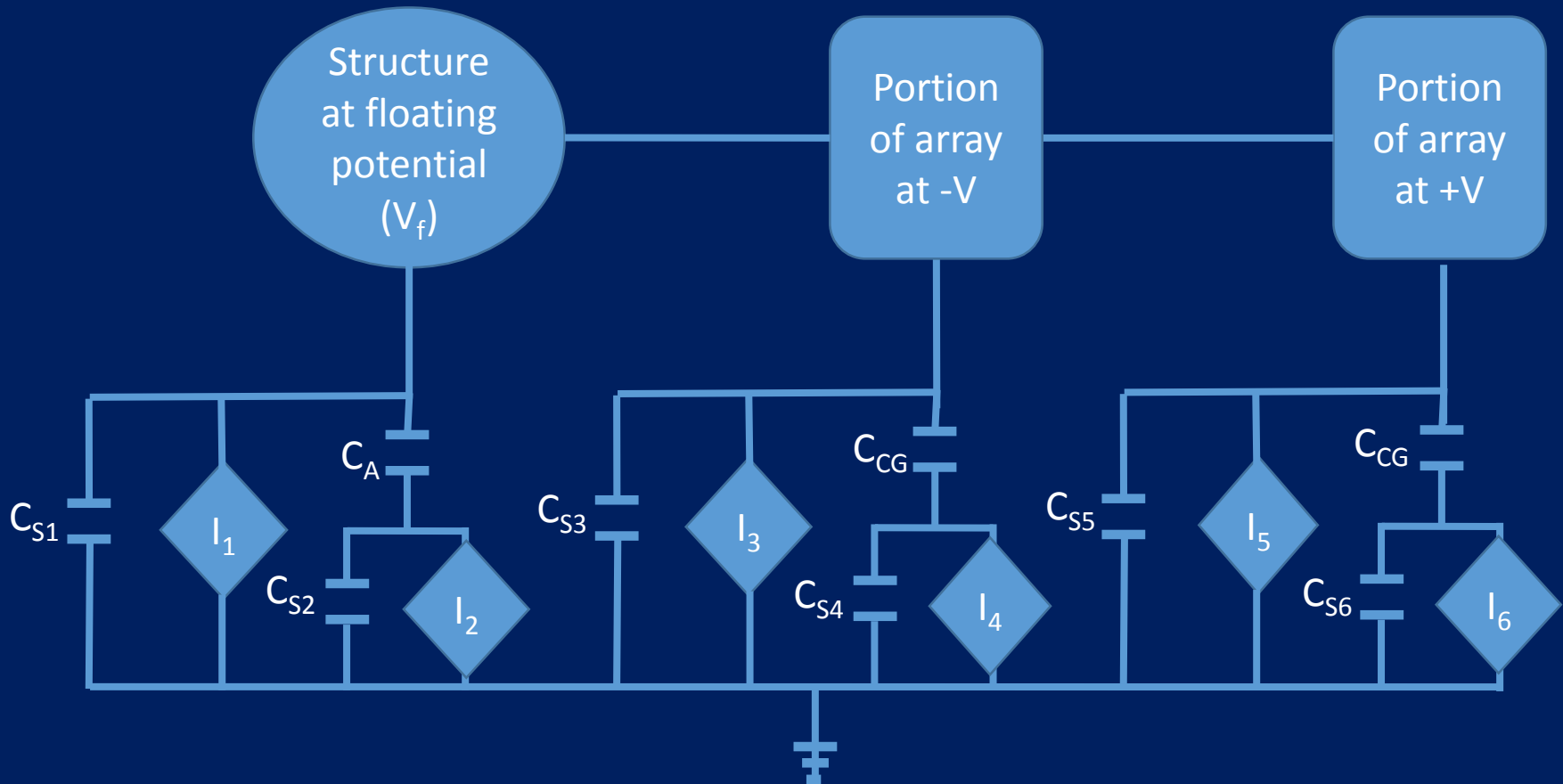


← Model Output
FPMU Data

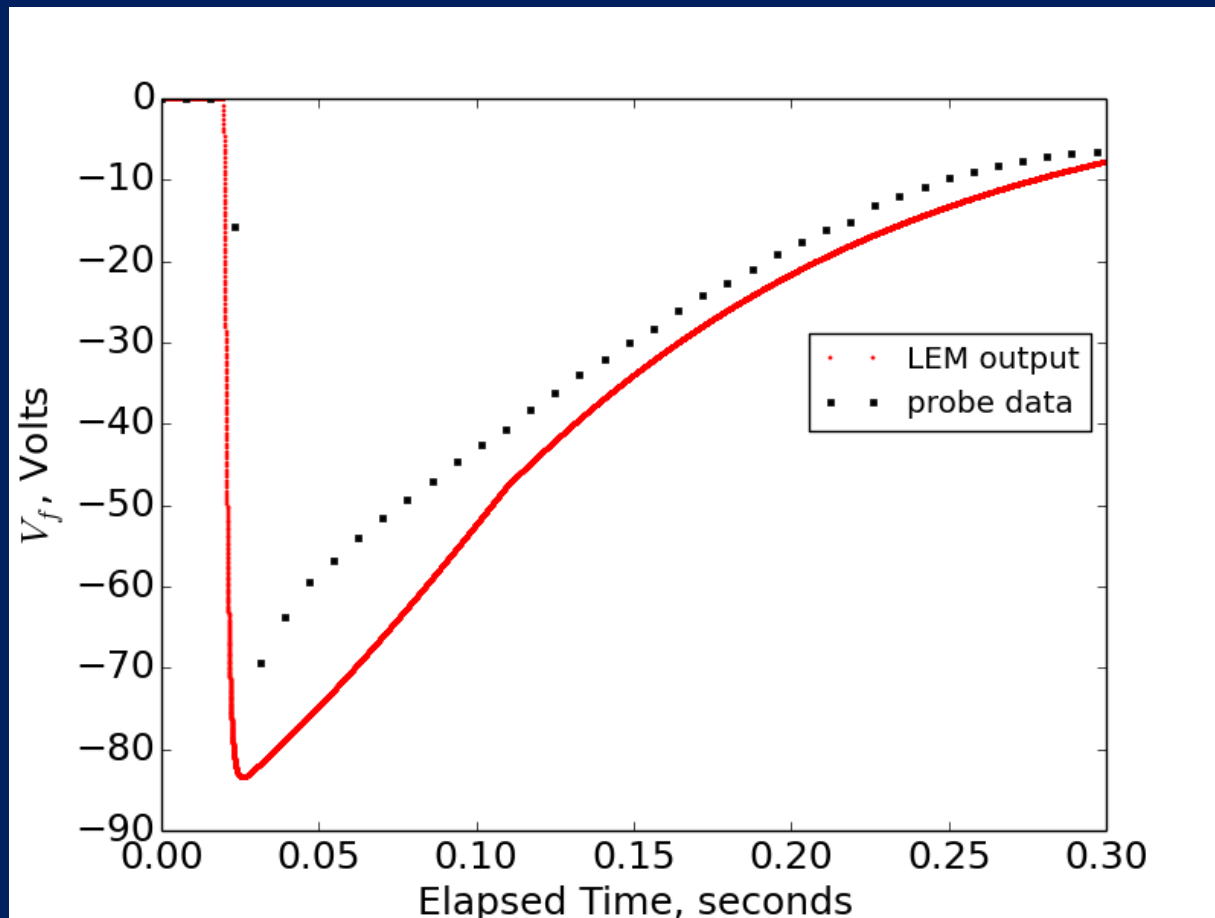


Model output does not reproduce FPMU data

New Lumped Element Model Design



Initial LEM Results



Model output agrees well with FPMU data

- Floating potential transients attributed to solar array operations have been observed in ISS FPMU data.
- These transients are not reproduced by existing current balance models, therefore a more accurate model of the current collection is needed.
- This research is investigating the time dependent development of the barrier potential. The questions to answer include:
 - How long does it take for the barrier to develop?
 - How does the current collection develop in time along with the choking effect?
- These questions will be answered by:
 - Particle In Cell simulation of a unit ISS solar cell using parameters consistent with LEO and ISS operations.
 - Evaluation of the results of the PIC simulation to determine if it is possible that the electron collection to the solar cells can account for the transient observations.

References

- [1] J. Huang, Z. Yi, H. Zhao, L. Meng, and Y. Liu, “Model for rapid-charging events for the International Space Station, ”Journal of Spacecraft and Rockets, vol. 51, no. 1, pp. 11-15, 2014.
- [2] J. Huang, Z. Yi, H. Zhao, L. Meng, and Y. Liu, “Mechanism for rapid charging events on International Space Station,“ Journal of Spacecraft and Rockets, vol. 51, no. 3, pp. 917-921, 2014.