



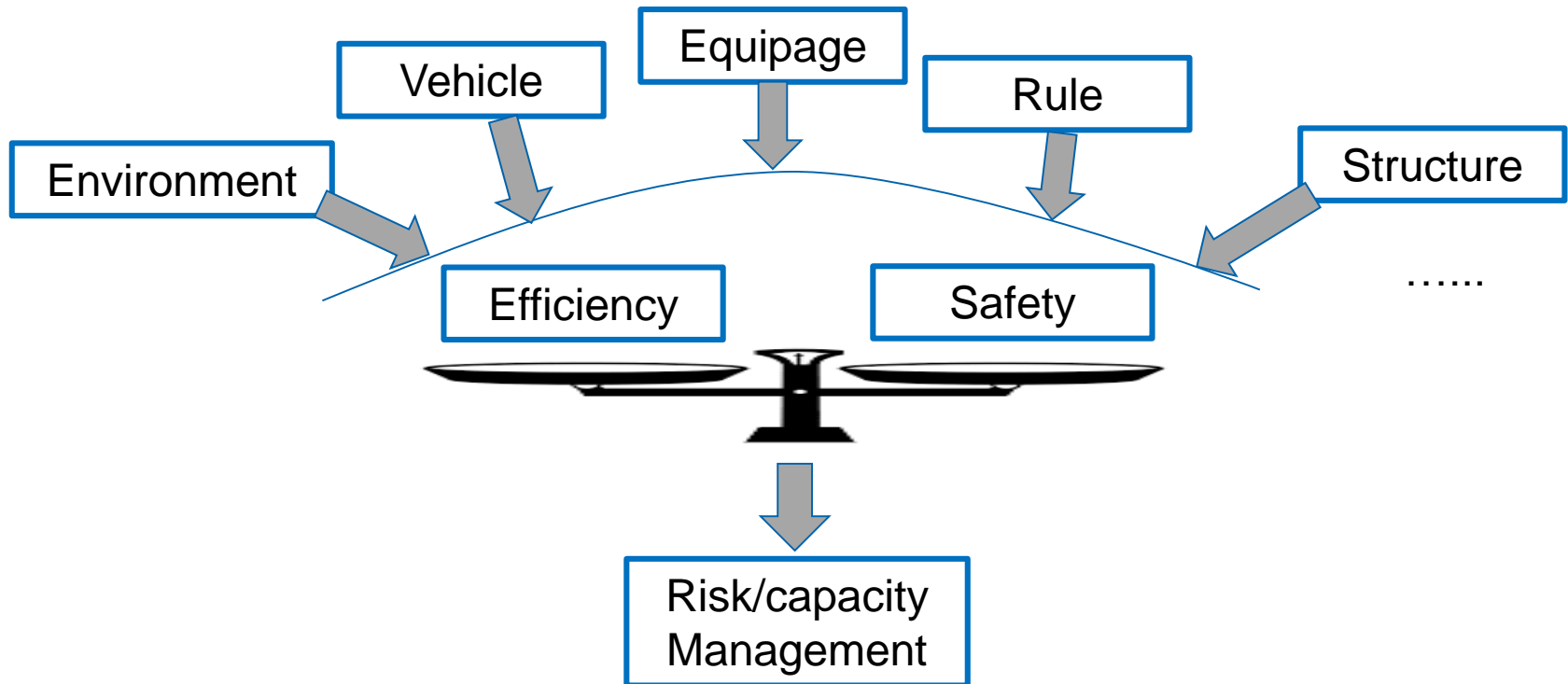
Initial Study of An Effective Fast-time Simulation Platform for Unmanned Aircraft System Traffic Management

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Motivation



Objective: Initial study and justification of developing an effective fast-time simulation platform

Outline

- Overview of existing simulations
- Requirements of UTM simulations
- Experiments using UTM simulations
- Summary

Simulation Categories

- Operations (multiple aircraft)
 - Manned aircraft: CTAS, FACET, ACES
 - Small UAV: Jenie^[JGCD2016], Cook^[AIAA2016]
- Encounter (~two aircraft)
 - MIT Lincoln Lab
 - Mueller^[MST2016]
- Vehicle centric (single aircraft)
 - Reflection^[NASA-TP2006]
 - Others

Comparison

| Simulation | UTM required |
|---|--------------------------------------|
| Maximum number of vehicles per scenario | >100 |
| Fidelity of vehicle models | >medium |
| Vehicle's controller modeled? | ✓ |
| Wind effect | Along-track + cross-track + vertical |
| Limited flight duration? | × |
| Capability of Monte Carlo simulations? | ✓ |
| Collision avoidance algorithm included? | ✓ |

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Small UAV Trajectory Model

Dynamics:

$$\begin{bmatrix} \dot{p}_n \\ \ddot{p}_n \\ \dot{p}_e \\ \ddot{p}_e \\ \ddot{h} \\ \ddot{\phi} \\ \ddot{\theta} \\ \ddot{\psi} \end{bmatrix} = \begin{bmatrix} \ddot{p}_n + \omega_n \\ -(\cos \phi \sin \theta \cos \psi + \sin \phi \sin \psi) F_z / m \\ \ddot{p}_e + \omega_e \\ (-\cos \phi \sin \theta \sin \psi + \sin \phi \cos \psi) F_z / m \\ g - \cos \phi \cos \theta F_z / m \\ M_\phi / J_x \\ M_\theta / J_y \\ M_\psi / J_z \end{bmatrix}$$

Controller: [proportional-derivative (PD)]

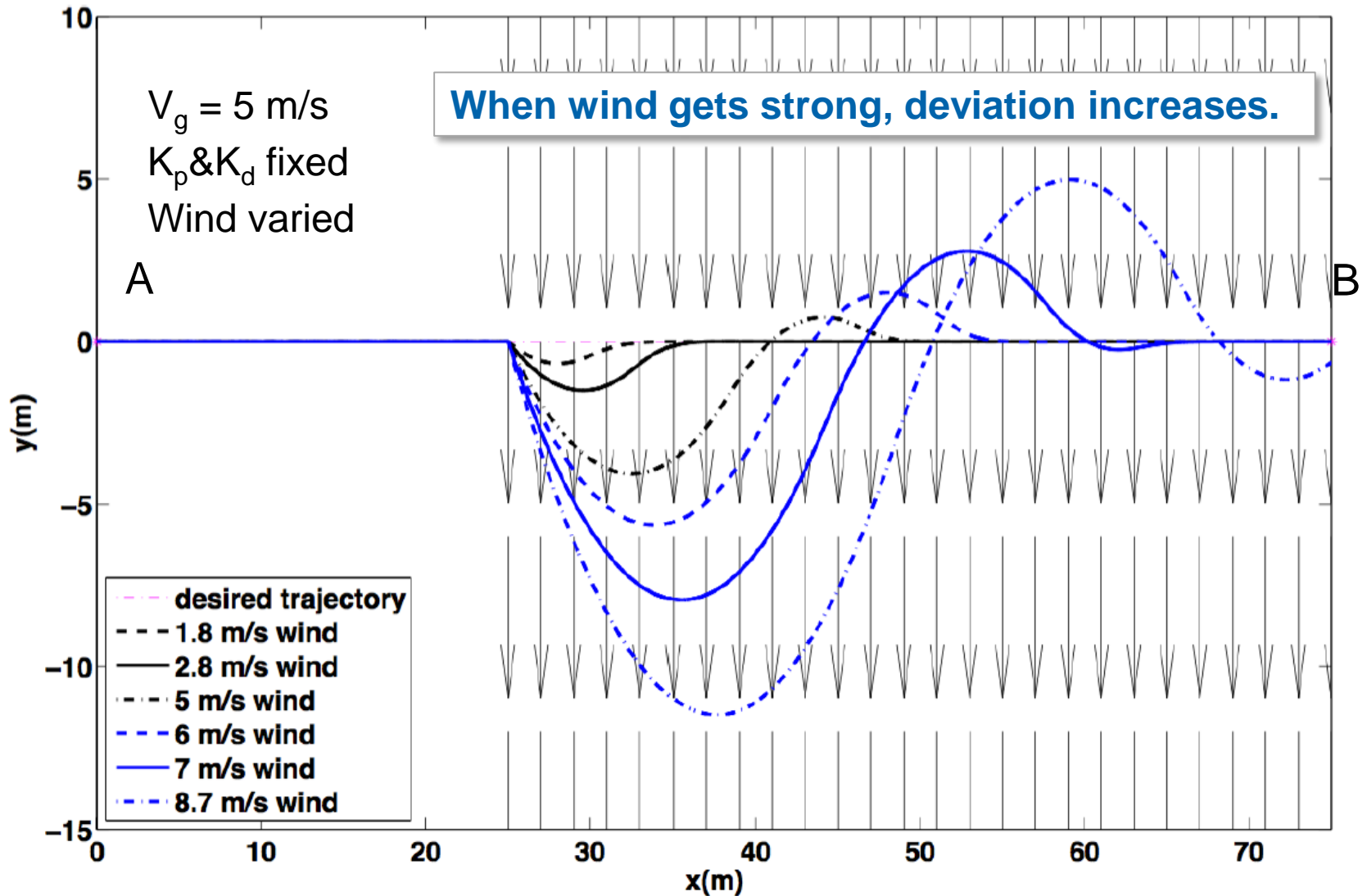
$$\begin{bmatrix} \ddot{p}_e \\ \ddot{p}_n \end{bmatrix} = \begin{bmatrix} k_p(p_{e,d} - p_e) + k_d(\dot{p}_{e,d} - \dot{p}_e) \\ k_p(p_{n,d} - p_n) + k_d(\dot{p}_{n,d} - \dot{p}_n) \end{bmatrix}$$

$$\begin{bmatrix} \phi_d \\ \theta_d \end{bmatrix} = \frac{m}{F_z} \begin{bmatrix} -\sin \psi & -\cos \psi \\ \cos \psi & -\sin \psi \end{bmatrix}^{-1} \begin{bmatrix} \ddot{p}_e \\ \ddot{p}_n \end{bmatrix}$$

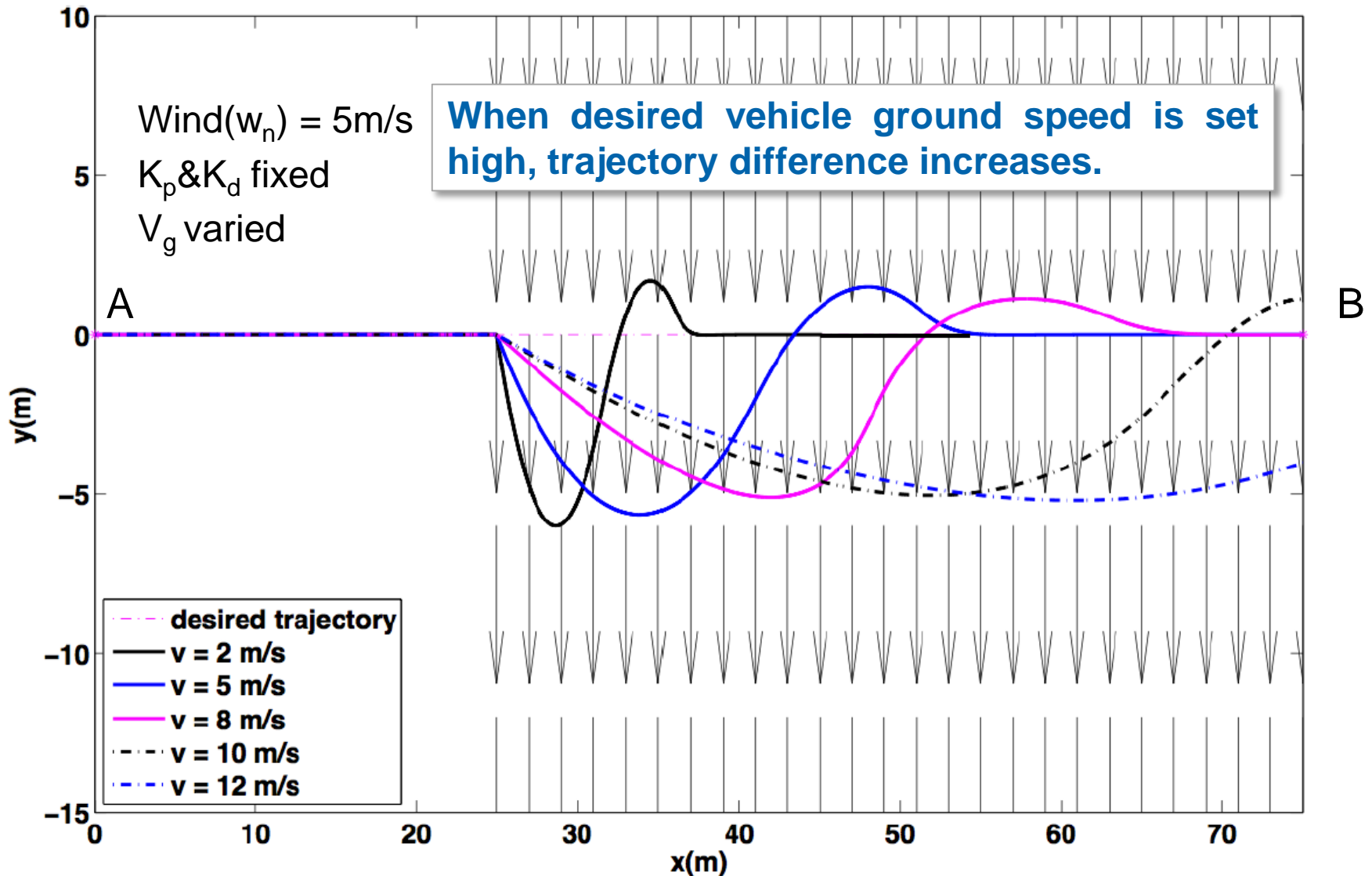
$$\begin{bmatrix} M_\phi \\ M_\theta \end{bmatrix} = \begin{bmatrix} k_{p,\phi}(\phi_d - \phi) + k_{d,\phi}(\dot{\phi}_d - \dot{\phi}) \\ k_{p,\theta}(\theta_d - \theta) + k_{d,\theta}(\dot{\theta}_d - \dot{\theta}) \end{bmatrix} l$$

$$k_{p,\phi} = 4.5, k_{d,\phi} = 0.5, k_{p,\theta} = 4.5, k_{d,\theta} = 0.5, k_p = 7.5, k_d = 4.2$$

Impact of Wind Speed



Impact of Desired Vehicle Ground Speed



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Monte Carlo Method

- UTM requires parameter and uncertainty/error studies
- UTM uncertainties/errors are high-dimensional
- Monte Carlo method is independent of the problem dimension
- The rate of convergence of order is : $O(1/\sqrt{n})$
- Error percentage can be computed by:

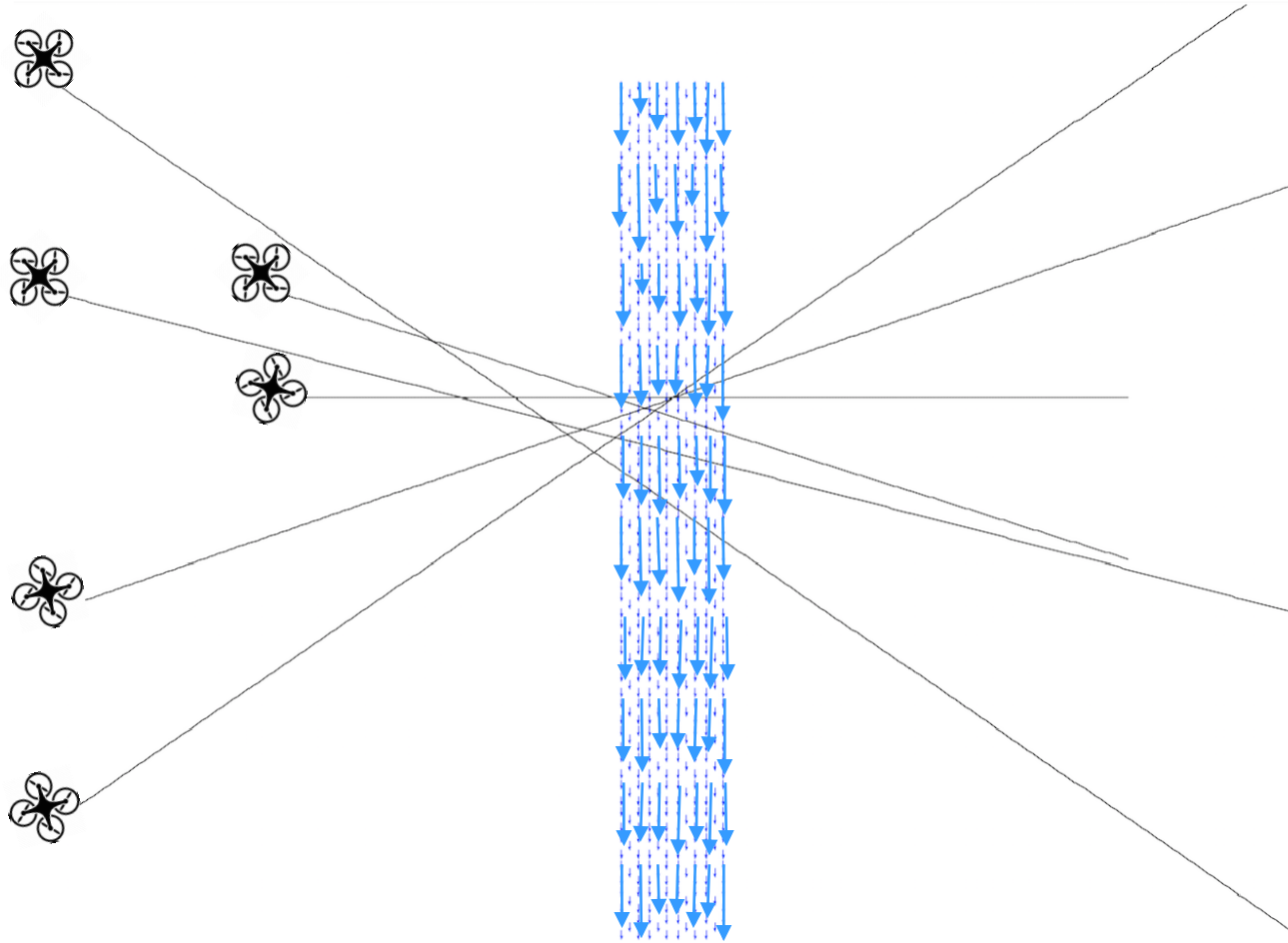
$$E = \frac{100z_c S_x}{\bar{x}\sqrt{n}}$$

- Monte Carlo is widely used in finance and engineering

Outline

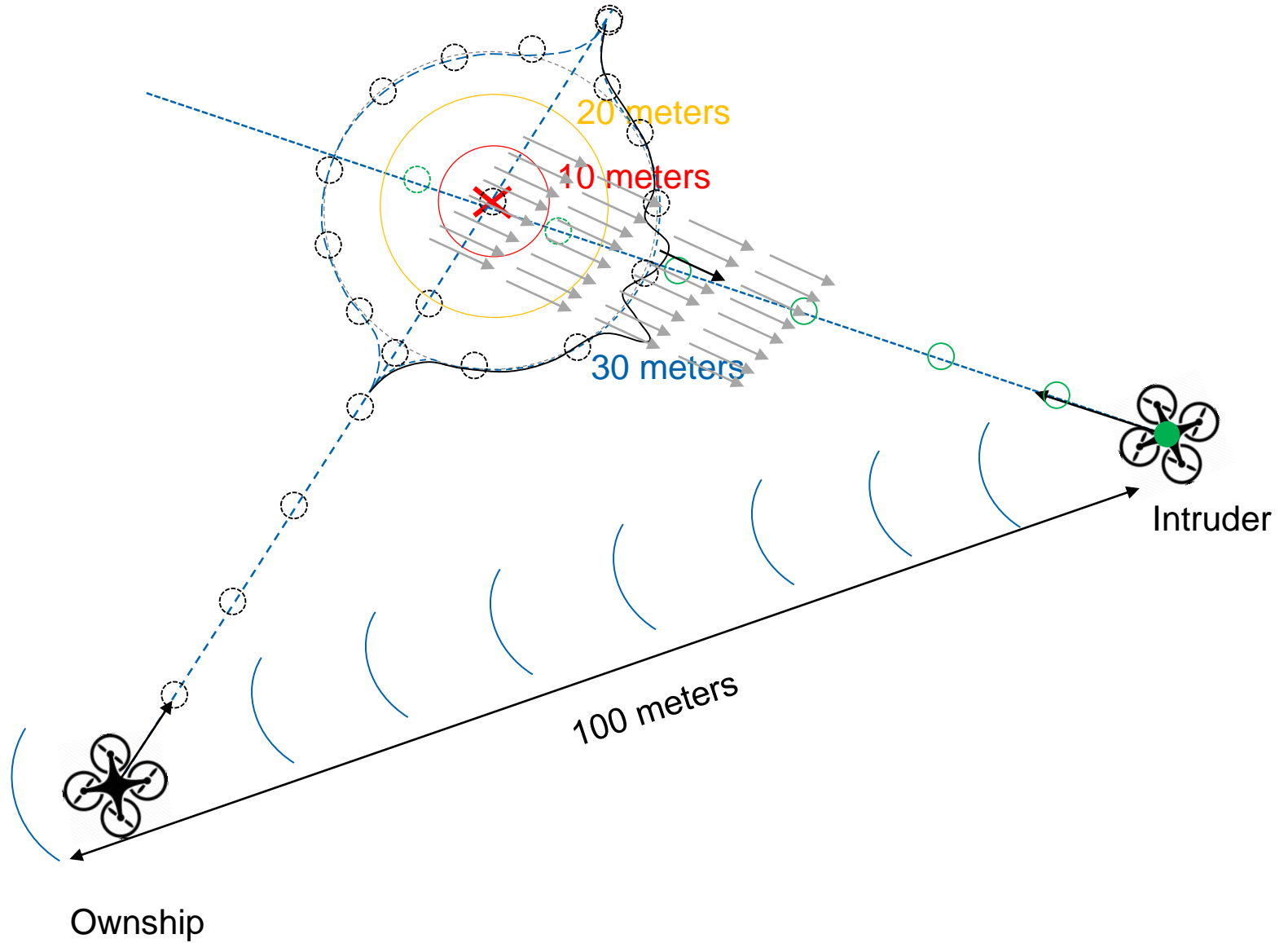
- Overview of existing simulations
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- **Experiments using UTM simulations**
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Scenario



- Six quadrotors with $V_g = 5$ m/s
- A rectangular north wind field with uncertainty

Setup



Experiment #1: Impact of Wind

| Wind speed (m/s) | | Avoidance maneuver | Loss of separation (probability) | | | Extra flight distance (m) | | | Extra flight time (s) | | |
|------------------|------|--------------------|----------------------------------|------|----------|---------------------------|------|----------|-----------------------|------|----------|
| mean | Std. | | mean | Std. | Error(%) | mean | Std. | Error(%) | mean | Std. | Error(%) |
| 0 | 0 | Right turn | 0 | 0 | 0 | 165.5 | 0.0 | 0.0 | 31.0 | 0.0 | 0.0 |
| 3 | 1 | Right turn | 0 | 0 | 0.0 | 168.8 | 3.6 | 0.12 | 31.0 | 0.03 | 0.01 |
| 5 | 2 | Right turn | 0.01 | 0.08 | 97.2 | 212.1 | 42.9 | 1.7 | 32.4 | 1.8 | 0.46 |

$$E = \frac{100z_c S_x}{\bar{x}\sqrt{n}}$$

Experiment #2: Impact of Avoidance Maneuver

| Wind speed (m/s) | | Avoidance maneuver | Loss of separation (probability) | | | Extra flight distance (m) | | | Extra flight time (s) | | |
|------------------|------|--------------------|----------------------------------|------|----------|---------------------------|------|----------|-----------------------|------|----------|
| mean | Std. | | mean | Std. | Error(%) | mean | Std. | Error(%) | mean | Std. | Error(%) |
| 3 | 1 | Right turn | 0 | 0 | 0 | 168.8 | 3.6 | 0.17 | 31.0 | 0.03 | 0.01 |
| 3 | 1 | Left turn | 0.847 | 0.36 | 3.46 | 71.0 | 23.3 | 2.7 | 9.5 | 3.4 | 3.0 |
| 3 | 1 | Hover | 0.04 | 0.20 | 38.9 | 5.95 | 4.1 | 5.6 | 20.9 | 4.4 | 1.72 |

Summary

- Reviewed some existing simulations
- Identified UTM required attributes
- Conducted trajectory sensitivity analysis
- Conducted preliminary experiments using Monte Carlo

Future Work

- Implement the platform on the Cloud
- Incorporate and generalize more vehicle dynamic and control systems
- Implement and generalize more collision avoidance algorithms
- Implement onboard sensor and communication device models
- Environmental data (wind, temperature, etc.)
- Geographic Information System (GIS) data (terrain, population, etc)