# A Proposed Byzantine Fault-Tolerant Voting Architecture Using Time-Triggered Ethernet

Andrew Loveless, NASA Johnson Space Center Christian Fidi, Stefan Wernitznigg, TTTech

SAE 2017 AeroTech Congress & Exhibition Fort Worth, TX 26 – 28 September 2017





### **COTS in Manned Spacecraft**

- COTS technologies are attractive for use in human-rated spacecraft.
  - Reduces development and upgrade costs.
  - · Lowers the need for new design work.
  - Eliminates reliance on individual suppliers.
  - Leverages larger knowledge base.
  - Minimizes schedule risk.
- Problem? Hard to meet the high reliability and fault tolerance requirements.
  - E.g. 10<sup>-9</sup> failures/hour in ultra-dependable systems.
  - E.g. Crit-1, "fly-through" fault tolerance.
  - Studies for Orion showed purely COTS designs would result in poor reliability and undue expense.

Often custom proprietary solutions are needed.

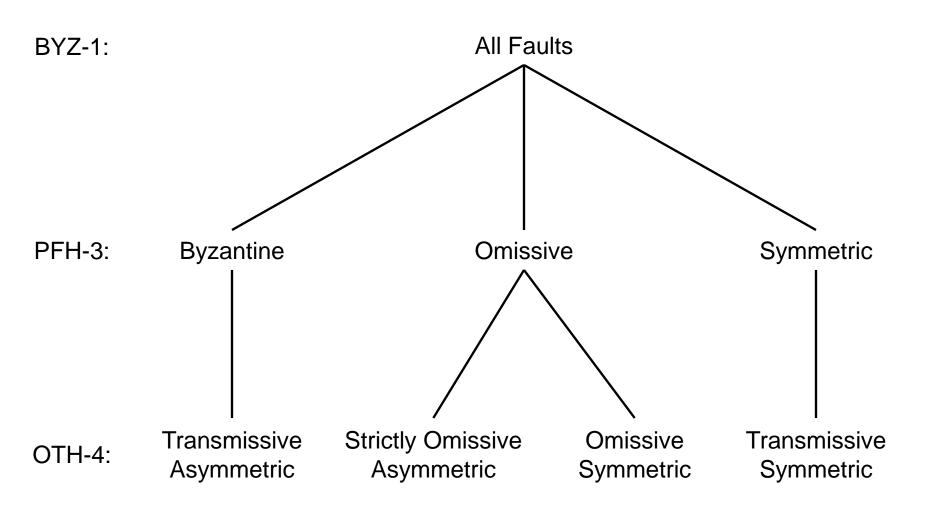


### **COTS in Manned Spacecraft (cont.)**

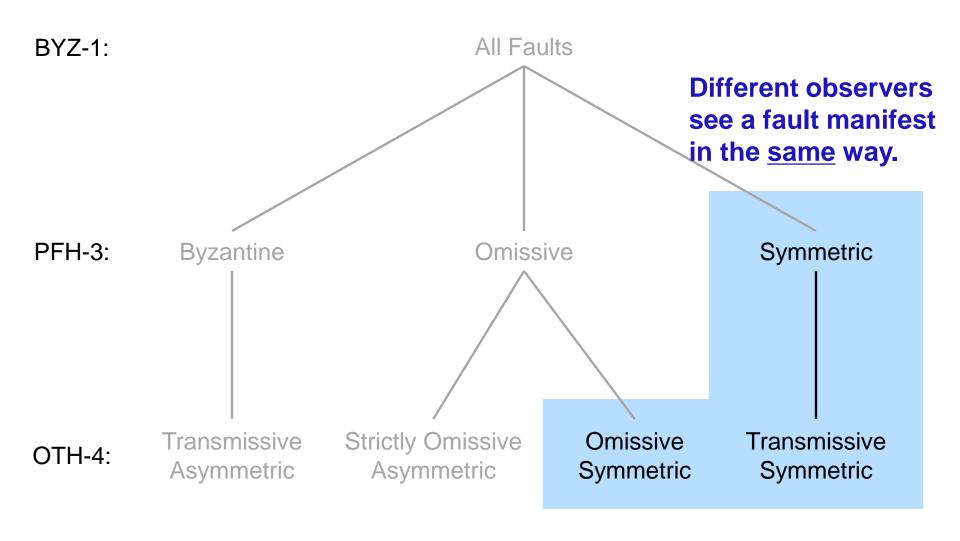
- But the inclusion of COTS technologies is becoming more feasible.
  - Greater availability of rad-tolerant components.
  - TMR (Maxwell SCS750), lock-step (ARM R5).
  - Ability to realize fault-containment regions.
  - Growing number of suppliers.
- NASA's strategy for future spacecraft has heavily prioritized using COTS parts.
  - · Includes launchers, landers, etc.
- Multiple projects have explored realizing safety-critical systems using COTS.
  - Scalable Processor-Independent Design for Extended Reliability (SPIDER).
  - Heavy Lift Vehicle (HLV) Architecture Study.
  - Evolvable Mars Campaign (lander).



#### Fault Classifications

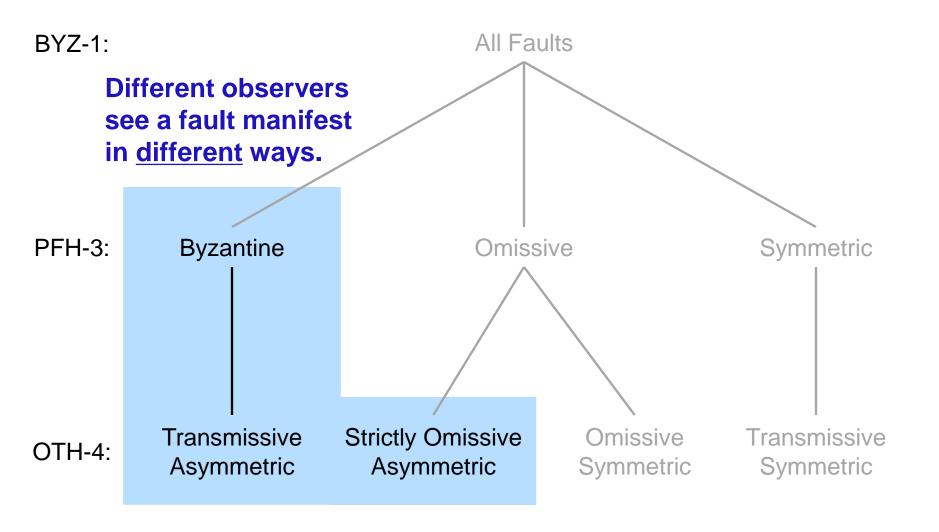


### Fault Classifications (cont.)



SAE INTERNATIONAL Paper # 2017-01-2111 5/23

### Fault Classifications (cont.)



SAE INTERNATIONAL Paper # 2017-01-2111 6/23

### Fault Classifications (cont.)

- Manned spacecraft must tolerate Byzantine faults.
  - Especially for dynamic mission phases with short time to effect.
  - Higher number of "all-or-none" events (e.g. deploy parachutes).
  - Failure could result in loss of life.



#### Byzantine faults are often Fanot considered in satellites.

- Possibility is considered low enough to not warrant additional complexity.
- Impacts of faults are less severe (e.g. not taking a picture).



nissive

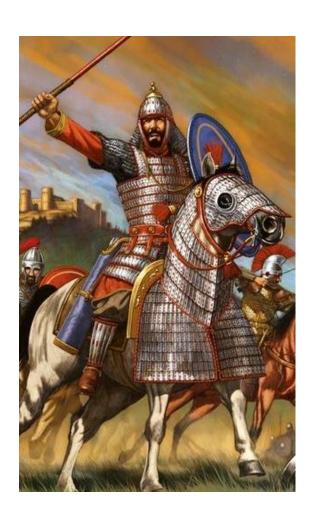
### **Byzantine Faults**

#### Byzantine faults can disrupt consensus among redundant processors.

- E.g. on internal state information.
- E.g. on sensor data.
- E.g. on diagnosis of system faults.

#### Occur at rates much > 10<sup>-9</sup> failures/hour.

- Slightly-off-specification (SOS) hardware.
- Stuck transmitter different receivers can interpret a marginal signal differently.
- Time base corruption messages received slightly too early or too late.
- Several architectural approaches for Byzantine-resilient systems.
  - Hierarchical e.g. SAFEbus, Orion VMCs.
  - Full exchange e.g. Draper FTMP, SPIDER.



### **A Typical Approach**

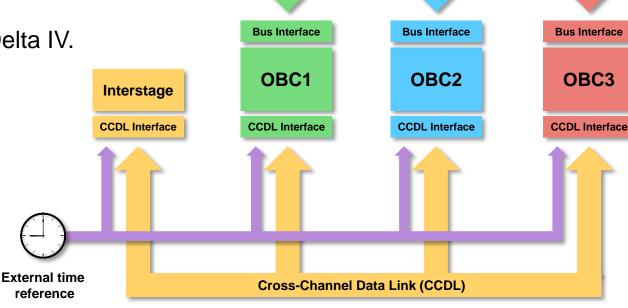
"Channelized bus" approach is common in launchers.

- Each OBC can only access devices on its local bus.
- Uses full exchanges.
- Usually designed to be 1FT.

#### Examples:

X-38 CRV, Ares I, Delta IV.

#### Shortcomings?



**Bus Channel A** 

PDU<sub>1</sub>

RIU1

COM<sub>1</sub>

PDU2

Bus

Channel

COM<sub>2</sub>

PDU3

RIU2

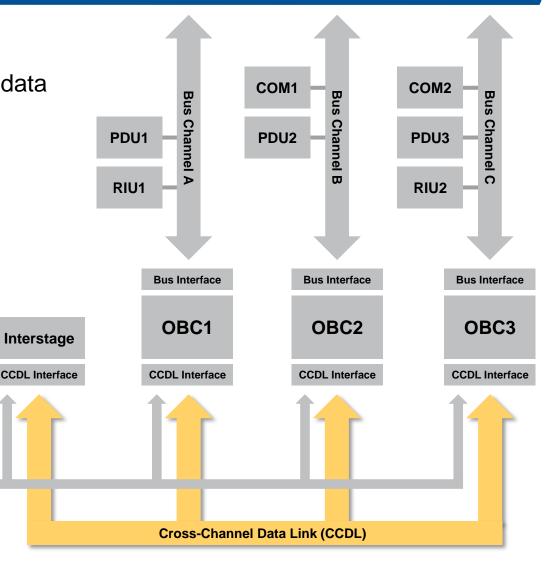
**Bus Channel** 

**External time** 

reference

#### Shortcomings?

1. Requires separate CCDL for data exchange between OBCs.

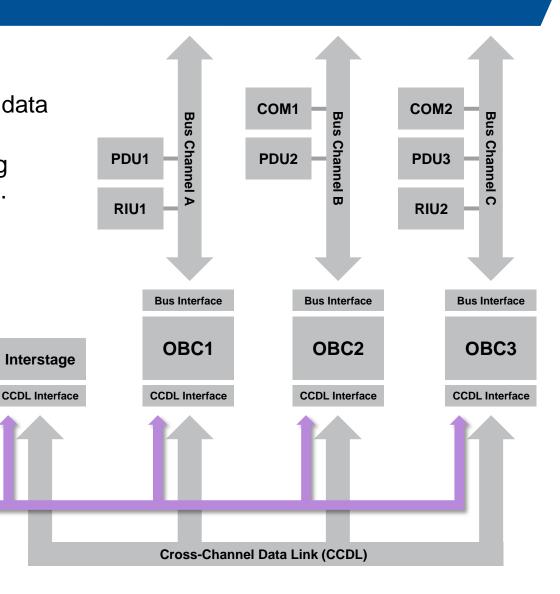


**External time** 

reference

#### Shortcomings?

- 1. Requires separate CCDL for data exchange between OBCs.
- 2. Often requires external timing hardware for synchronization.

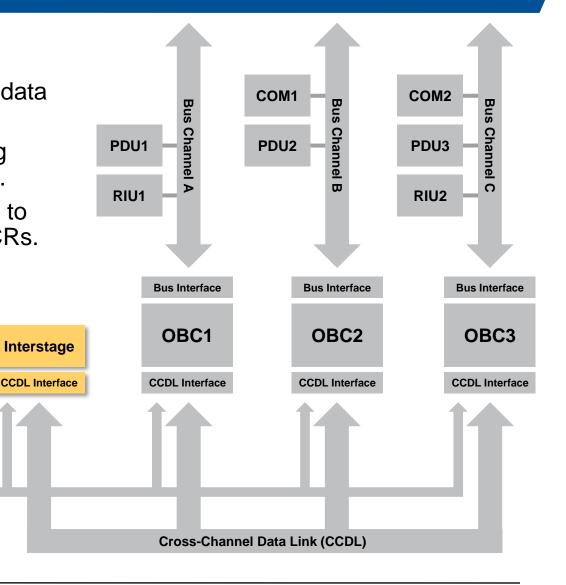


External time

reference

#### Shortcomings?

- 1. Requires separate CCDL for data exchange between OBCs.
- 2. Often requires external timing hardware for synchronization.
- 3. Requires separate interstage to meet minimum number of FCRs.

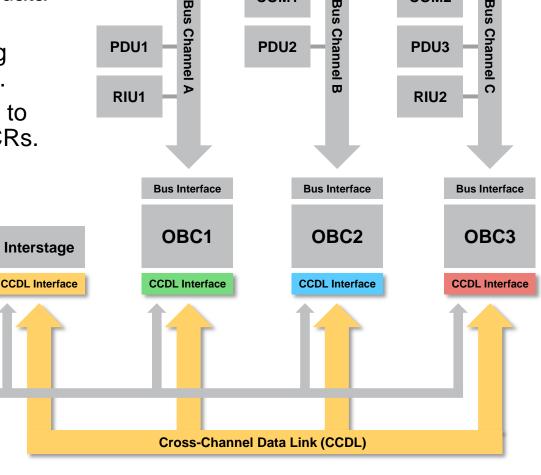


External time

reference

#### **Shortcomings?**

- 1. Requires separate CCDL for data exchange between OBCs.
- 2. Often requires external timing hardware for synchronization.
- 3. Requires separate interstage to meet minimum number of FCRs.
- 4. Requires two rounds of data exchange between OBCs.



COM<sub>1</sub>

PDU<sub>2</sub>

**Bus Channel** 

COM<sub>2</sub>

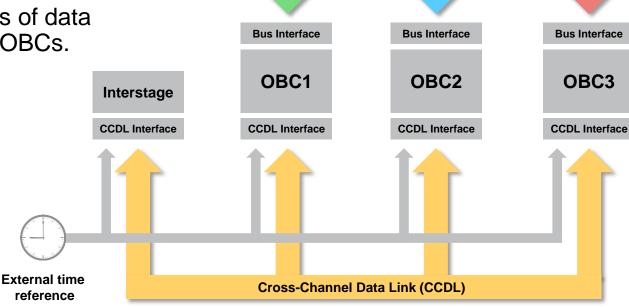
PDU<sub>3</sub>

PDU<sub>1</sub>

#### Shortcomings?

- Requires separate CCDL for data exchange between OBCs.
- 2. Often requires external timing hardware for synchronization.
- 3. Requires separate interstage to meet minimum number of FCRs.
- 4. Requires two rounds of data exchange between OBCs.

5. Bandwidth limited.



**Bus Channel A** 

PDU<sub>1</sub>

RIU1

COM<sub>1</sub>

PDU<sub>2</sub>

Bus

Channel

COM<sub>2</sub>

PDU<sub>3</sub>

RIU2

**Bus Channel** 

### **An Approach Using TTE**

#### ■ 1FT "switched voter" using TTE.

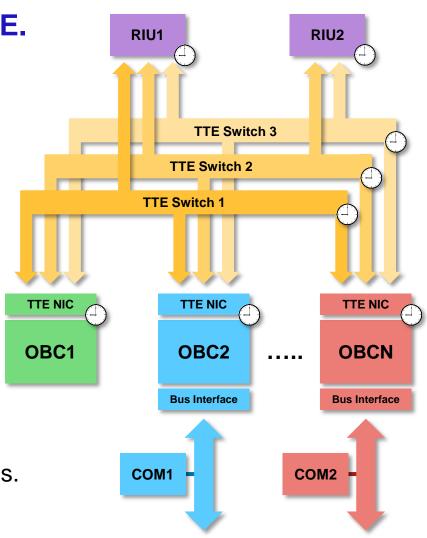
- Requires only 3 full processors.
- Requires 2-3 redundant switches.
- Devices can connect to OBCs directly or via TTE network.
- Assumes minimum number of SMs and CMs are present for sync.

## ■ TTE network used for data distribution and sync.

- Eliminates need for separate CCDL.
- Eliminates need for timing hardware.
- Bandwidth up to 1 Gbit/s.

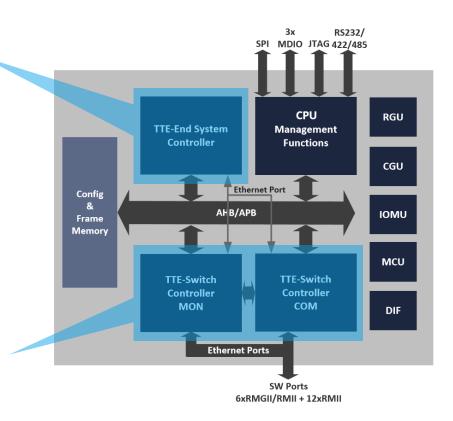
#### Switches act as interstages.

- Messages reflected to/from the switches.
- Eliminates need for fourth processor.



### **Failure Assumptions**

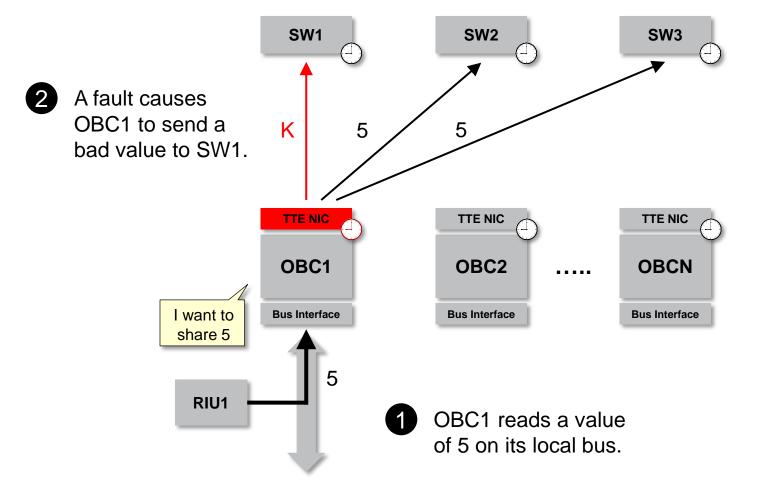
- End systems may be subject to Byzantine failures.
  - May send arbitrary messages.
  - May transmit at any point in time.
  - May send different messages to different switches.
- Switches are restricted to inconsistent omission failures.
  - May not create (nor modify to produce) a new "valid" message.
  - May drop or fail to receive an arbitrary number of messages.
  - May relay messages asymmetrically some receivers may not get data.
  - Acts as a "trusted sender".



Fault propagation from switches theoretically requires dual-correlated simultaneous faults.

 $\rightarrow$  10<sup>-6</sup> ×10<sup>-6</sup> = ~10<sup>-12</sup> failures/hour

#### **Agreement on Local Data**



### Agreement on Local Data (cont.)

**SW1** SW<sub>2</sub> SW3 K 5 5 K 5 5 Each switch K relays the data to all OBCs. TTE NIC TTE NIC TTE NIC OBC<sub>1</sub> OBC<sub>2</sub> **OBCN Bus Interface Bus Interface Bus Interface** 

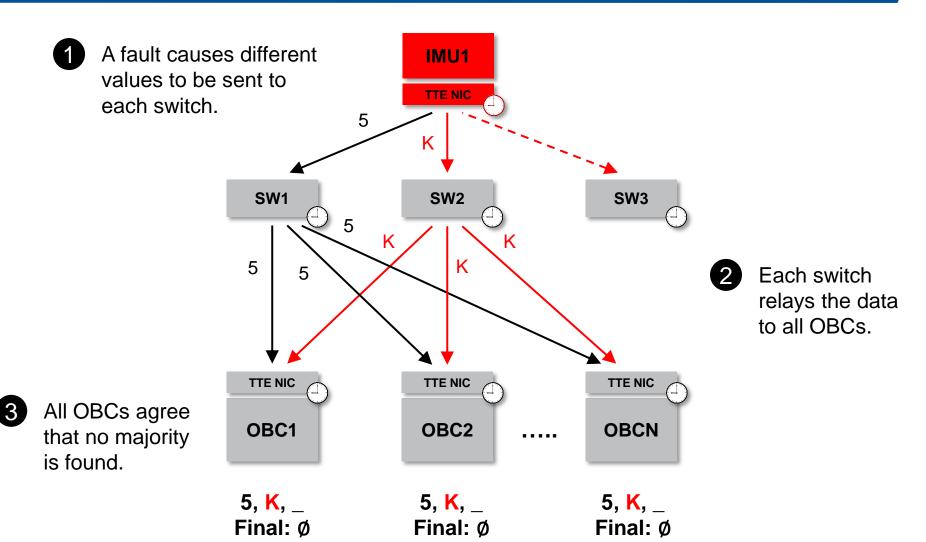
4 Each OBC votes the values sent from the switches.

Absent data is not included in the vote.

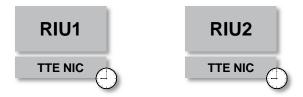
K, 5, 5 Final: 5

K, 5, 5 Final: 5 **K**, 5, 5 Final: 5 ! Vote could be implemented in TTE NIC or in software on the OBCs.

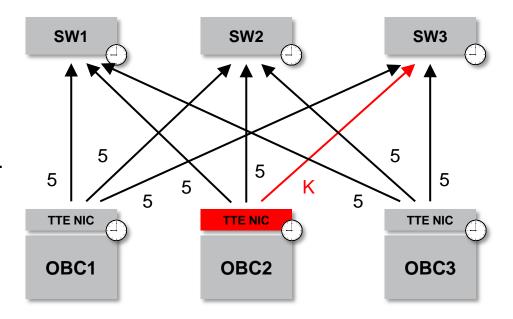
#### **Agreement on External Data**



### Commanding

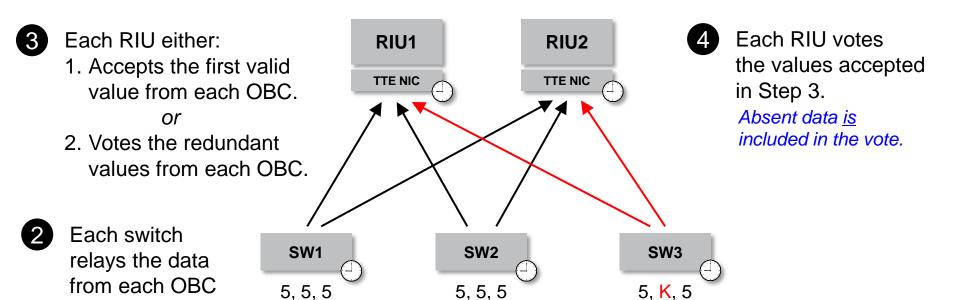


1 A fault causes OBC2 to send a bad value to SW3.



#### **Commanding (cont.)**

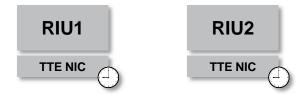
to all RIUs.





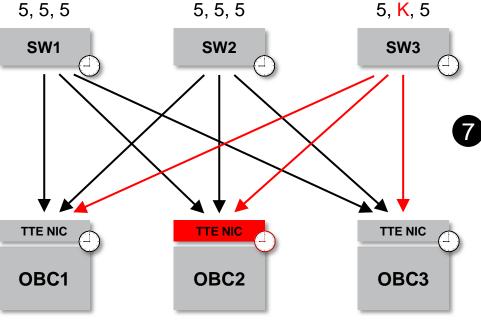
### **Commanding (cont.)**

#### Happening Simultaneously ...



- 5 Each switch reflects the original data back to all OBCs.
- 6 Each OBC votes the redundant values from each OBC.

Absent data <u>is not</u> included in the vote.



Each OBC votes the results from Step 6 to diagnose faulty OBCs.

Absent data <u>is</u> included in the vote.

### **Questions?**