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SPACE LAUNCH SYSTEM

SLS Model Based Design: A Navigation Perspective

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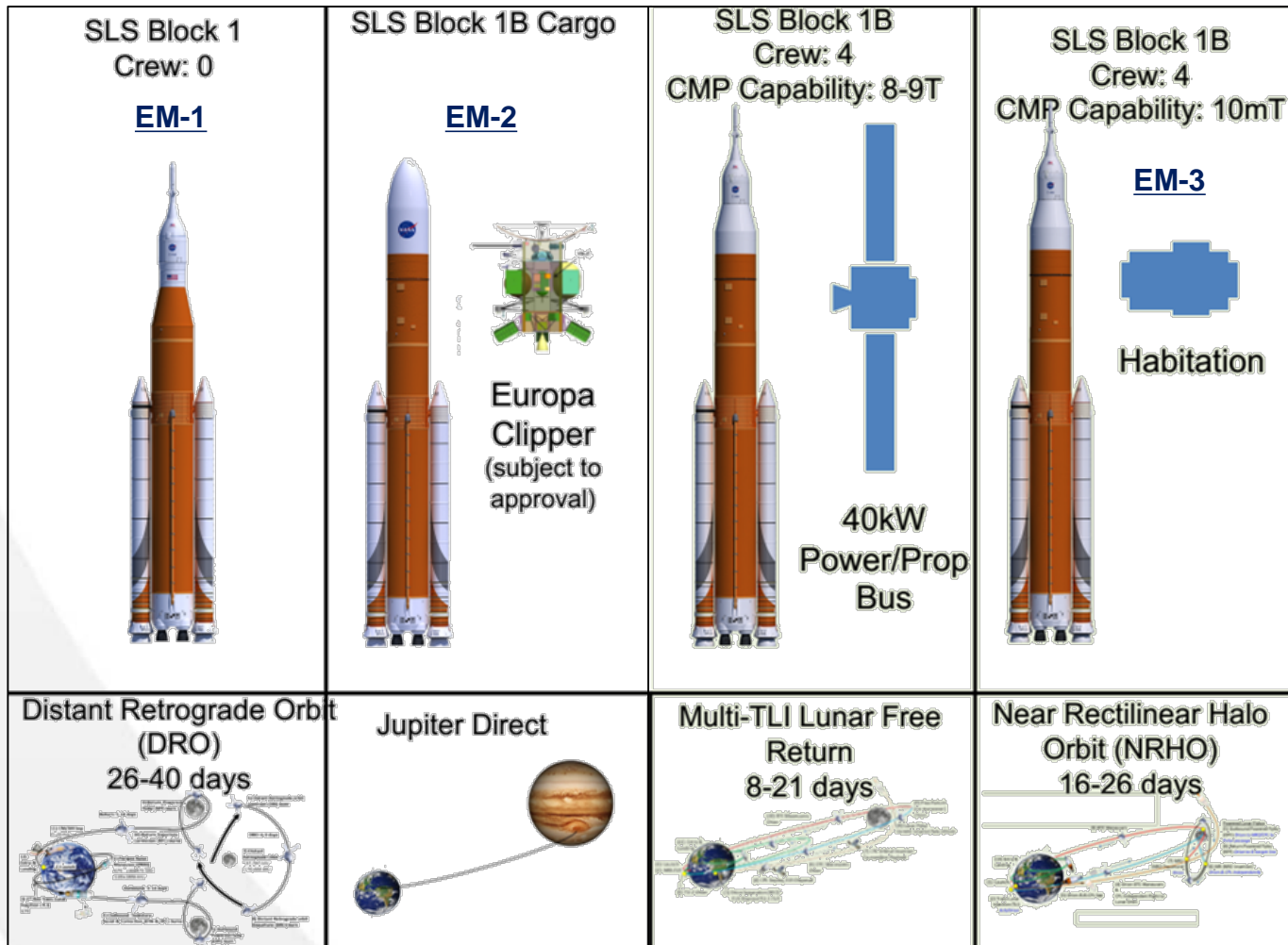
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Overview

- Introduction to Space Launch Systems (SLS)
- SLS Requirements and Design Math Models (DMMS)
- The SLS GN&C Model
- The SLS Inertial Navigation System (INS) Performance Model
- Marshall Advanced GPS Model for Analysis (MAGMA)
- Conclusions and Lessons Learned

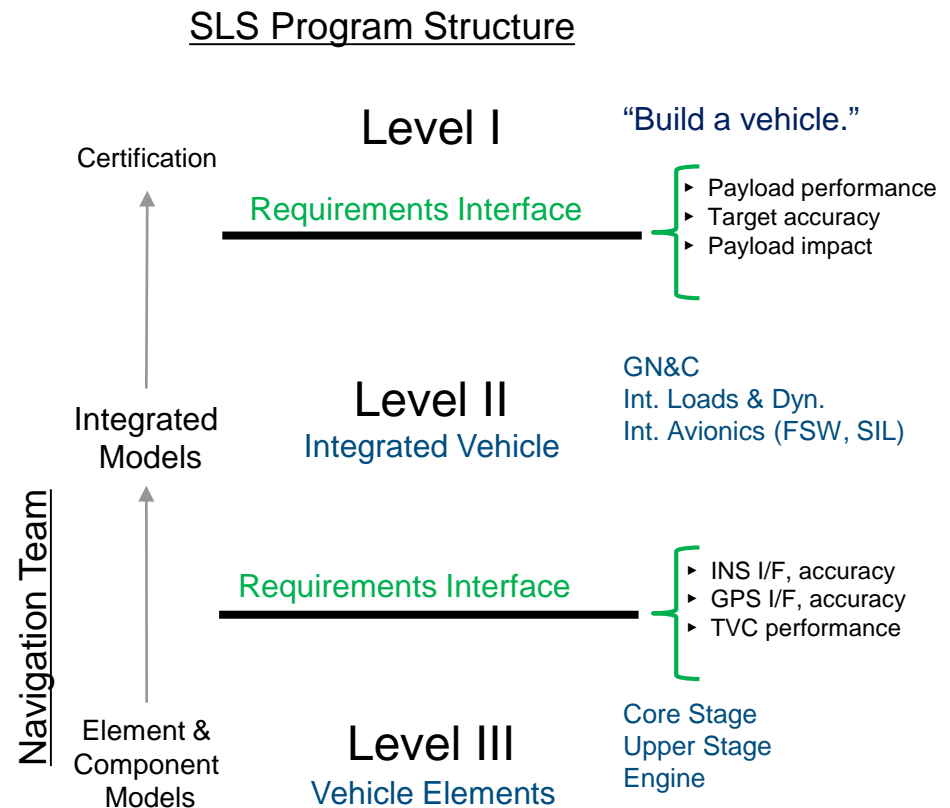
Introduction to Space Launch System (SLS)

- NASA is developing a phased plan to deep space exploration enabled by SLS, an evolution of Launch vehicles.
 - Currently completing the design and building the Block 1 vehicle
 - In the process of Block 1B design



SLS SE&I Model Based Design

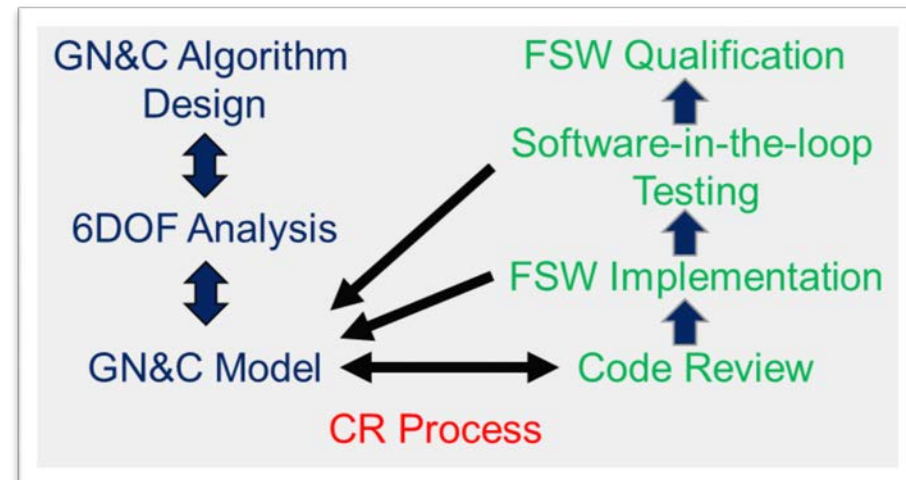
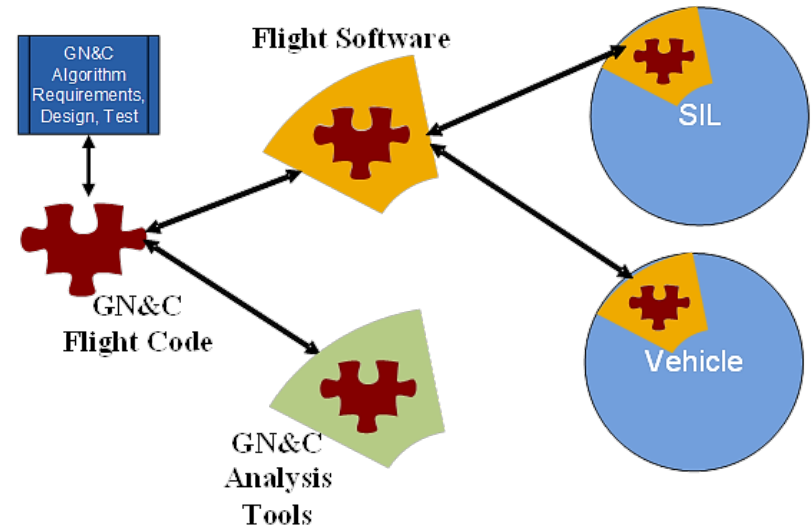
- Reduced Program structure
- Emphasis on heritage hardware
- Relatively sparse requirements set over previous design projects
- DMMs convey the design
 - Controlled at program level
 - Maturity/limitations/use tightly tracked
 - Component models are verified against vendor design and validated against flight hardware (or equiv.)
 - Physics models (e.g. 6DOF sim) verified against other simulations and validated with test data.
 - Model parameters of high sensitivity can be elevated to requirements
- SLS Navigation Supports Level II and Level III
- Example
 - Level II DMMs: GN&C Model, MAVERIC (6DOF Sim)
 - Level III DMMs: INS Performance, GPS



GN&C Model

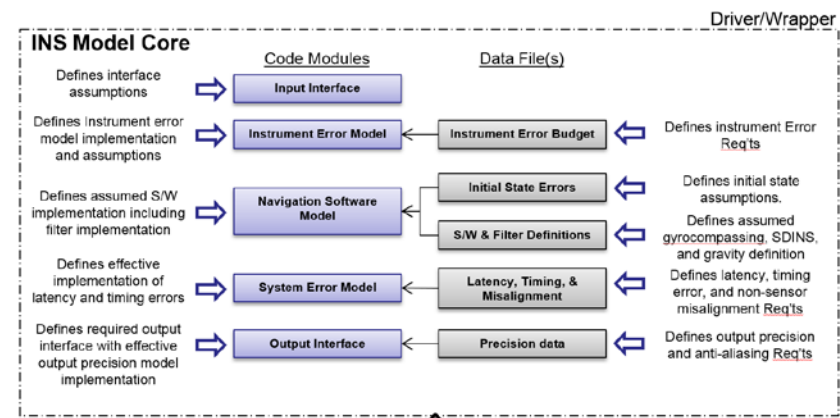
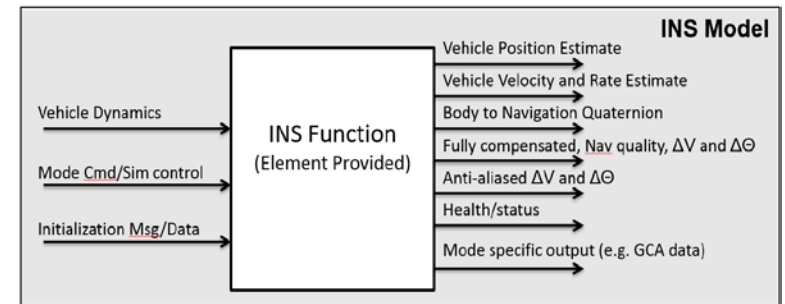
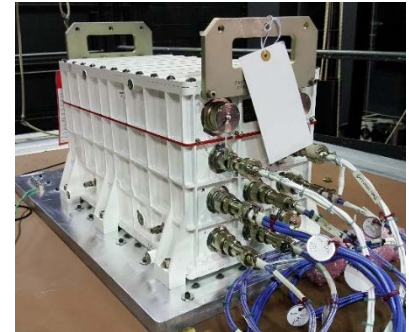
- Began as pilot program 2010
- Common GN&C code across SLS Disciplines & Functions
- Efficient GNC/FSW Process
- DMM Contents
 - Executable Algorithms
 - Parameter Definition
 - Technical Memorandum
 - Interface assumptions
 - Unit test cases
- GN&C/Navigation Model
 - Inertial Measurement Processing
 - State derived quantities
 - RINU Initialization
 - RINU FDIR Parameters
 - GCA Convergence check
 - RINU Frame check
 - GPS Measurement Processing
 - SDINS algorithms
 - Navigation EKF

(Block 1B only)



INS Performance Model

- **RINU: Redundant Inertial Navigation Unit**
- **Level II Requirements Definition**
 - Interface and frequency response
 - Performance constrained with reference trajectory
 - Reduction in requirements with explicit modeling
- **Level III Model Description**
 - Detailed instrument error modeling
 - Algorithms which affect performance
 - Detailed interface model
- **Verified against vendor documentation, FQT data, and analysis**
- **Validated against test data**
 - GCA 6DOF Test
 - Frequency response test
 - Vendor ATP/QTP data
- **Analyses Performed**
 - Navigation performance
 - Gyrocompassing alignment
 - Coning/Sculling
 - Integration into vehicle 6DOF



Allows for Validation of Model use and facilitates Verification of Vehicle Performance

Closed-Loop Trajectory Simulation

6-DOF Simulation (MAVERIC)

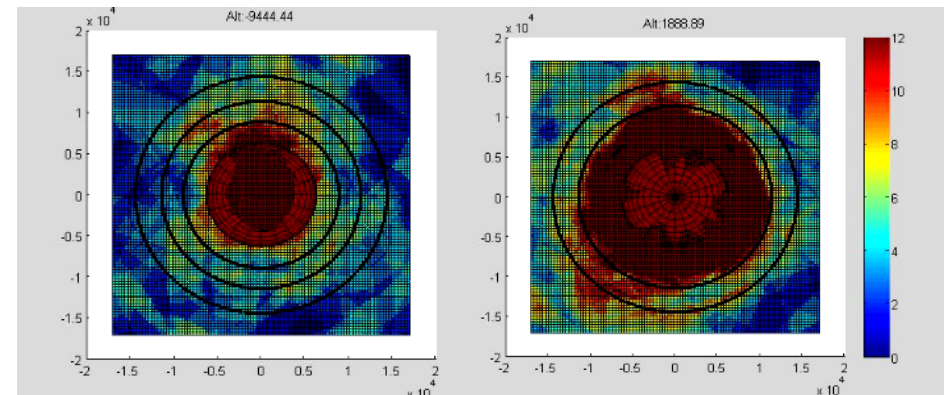
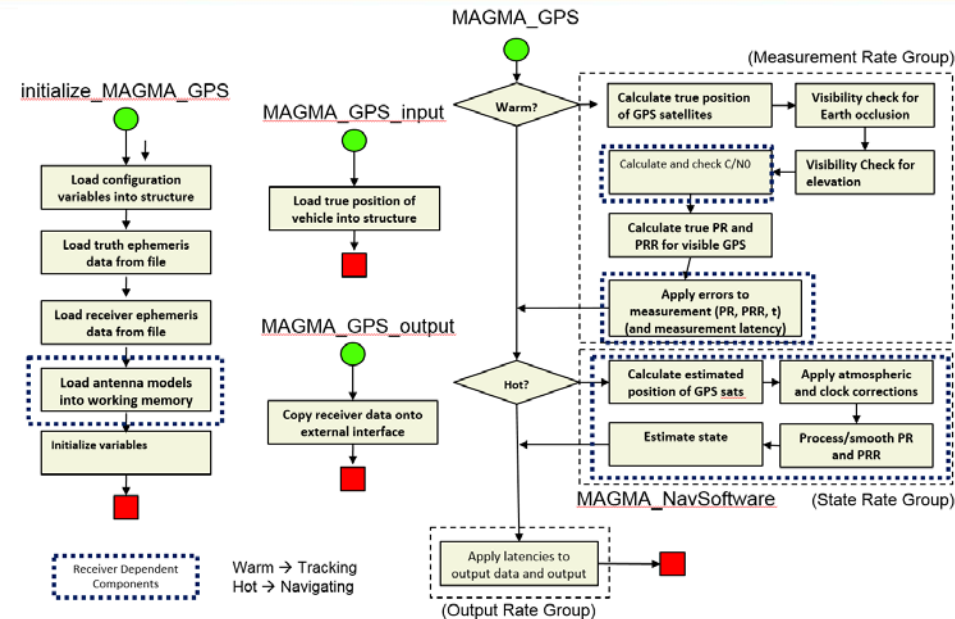
Standalone Driver

Test Case Definition and Data

Allows for Validation of Model Implementation and Verification of Model Performance

MAGMA GPS Model

- **Marshall Advanced GPS Model for Analysis**
- **Framework developed to support**
 - Requirements development,
 - Early Navigation System design
 - Seed Level III DMM development
- **Level II Requirements**
 - Interface definition
 - Measurement accuracy
- **Functional Components**
 - Detailed truth model
 - SV and Receiver Antenna modeling
 - Receiver hardware modeling
 - Receiver software modeling
- **Models measurement availability, accuracy, and latency**

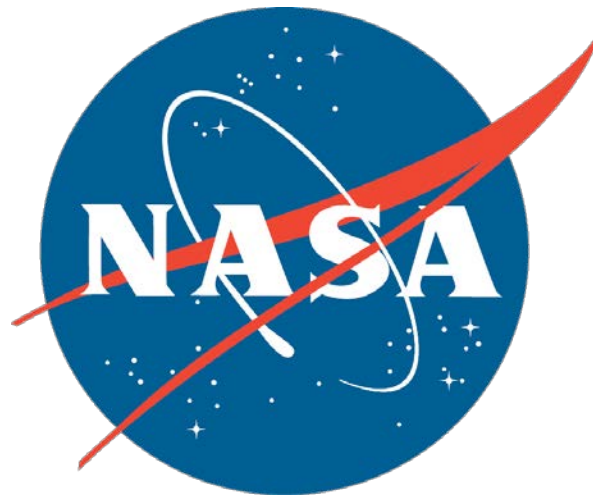


Simulation of GPS availability within 2D plane

Conclusion

- **Implementation of MBD on SLS has significantly increased efficiency**
 - Reduced requirements burden
 - Provide explicit communication of component and integrated system design
- **Provides a mechanism for**
 - Detailed modeling and design insight
 - Identification of key vehicle sensitivities
 - Gaining additional insight through testing and validation process
 - Enforcing rigor in modeling through validation
- **DMM V&V process forces high fidelity emulation of hardware**
- **Lessons Learned:**
 - Model form and function should consider user and developer
 - GN&C Model
 - Software requirements drive the software test program
 - Approach conflicted with established FSW processes and culture
 - Component models,
 - Good data requirements and supplier integration are key to enabling process
 - V&V plans should be defined early to support data requirements definition and to identify gaps which require additional testing.
 - Sensitivity analyses should be used to identify key performance drivers
 - Commonality between HWIL models and Performance/Analysis models reduces cross-validation effort in verification

Thank you!



Any questions?