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# SPACE LAUNCH SYSTEM

## SLS Model Based Design: A Navigation Perspective

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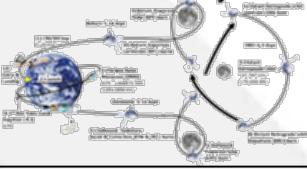
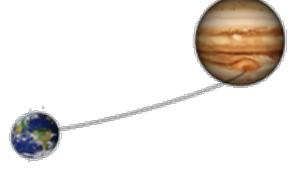
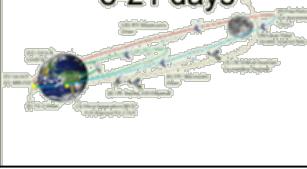
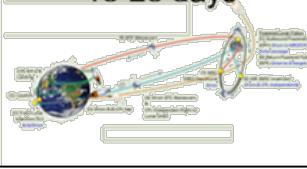


# Overview

- Introduction to Space Launch Systems (SLS)
- SLS Requirements and Design Math Models (DMMS)
- The SLS GN&C Model
- The SLS Inertial Navigation System (INS) Performance Model
- Marshall Advanced GPS Model for Analysis (MAGMA)
- Conclusions and Lessons Learned

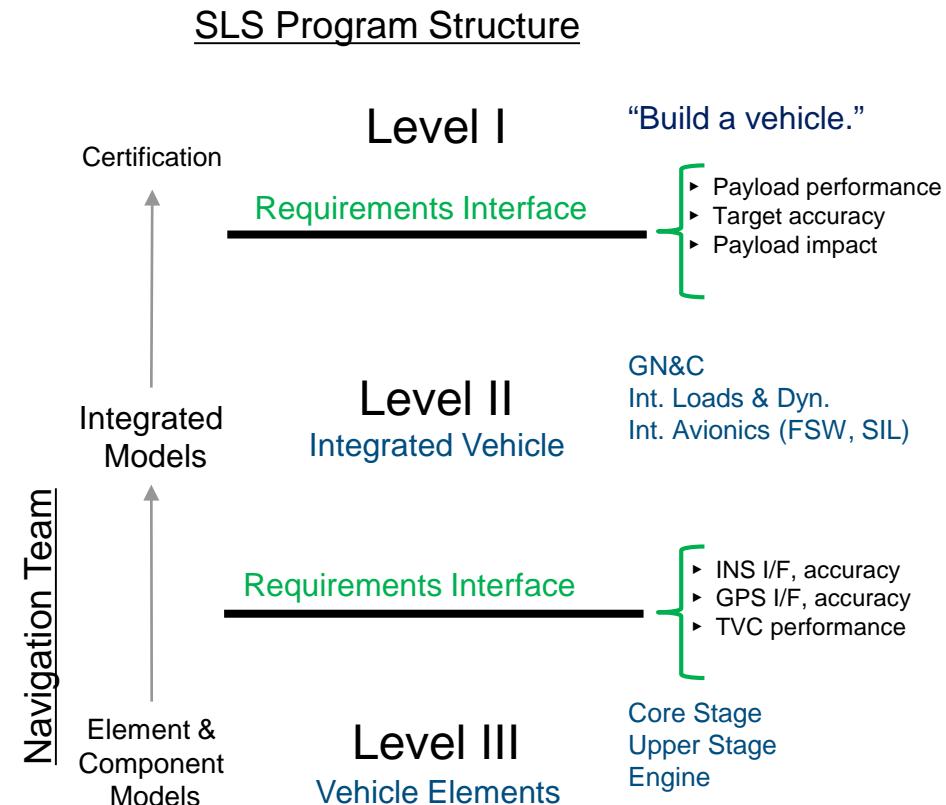
# Introduction to Space Launch System (SLS)

- NASA is developing a phased plan to deep space exploration enabled by SLS, an evolution of Launch vehicles.
  - Currently completing the design and building the Block 1 vehicle
  - In the process of Block 1B design

<p>SLS Block 1 Crew: 0</p> <p><u>EM-1</u></p> 	<p>SLS Block 1B Cargo</p> <p><u>EM-2</u></p>  Europa Clipper (subject to approval)	<p>SLS Block 1B Crew: 4 CMP Capability: 8-9T</p>  40kW Power/Prop Bus	<p>SLS Block 1B Crew: 4 CMP Capability: 10mT</p> <p><u>EM-3</u></p>  Habitation
<p>Distant Retrograde Orbit (DRO) 26-40 days</p> 	<p>Jupiter Direct</p> 	<p>Multi-TLI Lunar Free Return 8-21 days</p> 	<p>Near Rectilinear Halo Orbit (NRHO) 16-26 days</p> 

# SLS SE&I Model Based Design

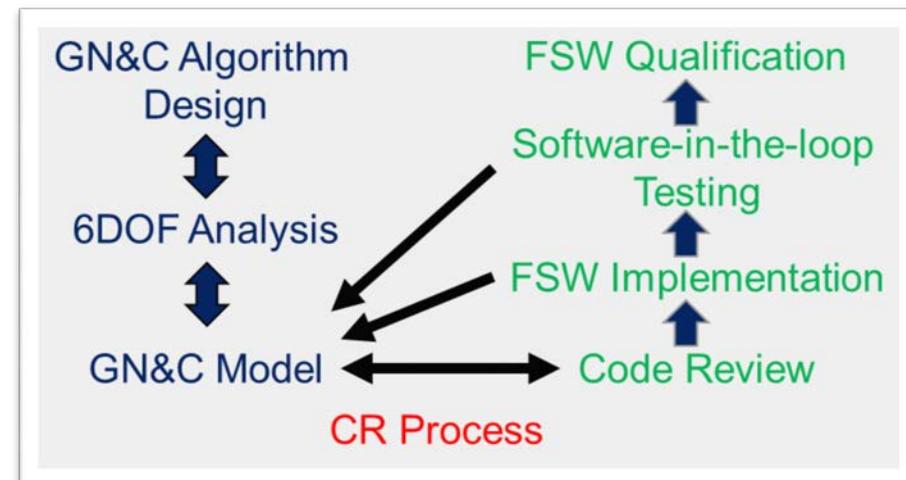
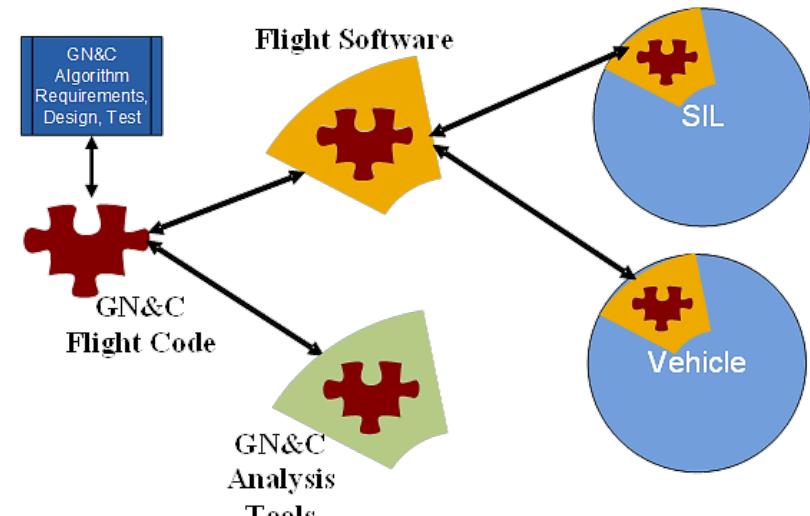
- Reduced Program structure
- Emphasis on heritage hardware
- Relatively sparse requirements set over previous design projects
- DMMs convey the design
  - Controlled at program level
  - Maturity/limitations/use tightly tracked
  - Component models are verified against vendor design and validated against flight hardware (or equiv.)
  - Physics models (e.g. 6DOF sim) verified against other simulations and validated with test data.
  - Model parameters of high sensitivity can be elevated to requirements
- SLS Navigation Supports Level II and Level III
- Example
  - Level II DMMs: GN&C Model, MAVERIC (6DOF Sim)
  - Level III DMMs: INS Performance, GPS



# GN&C Model

- Began as pilot program 2010
- Common GN&C code across SLS Disciplines & Functions
- Efficient GNC/FSW Process
- DMM Contents
  - Executable Algorithms
  - Parameter Definition
  - Technical Memorandum
  - Interface assumptions
  - Unit test cases
- **GN&C/Navigation Model**
  - Inertial Measurement Processing
  - State derived quantities
  - RINU Initialization
  - RINU FDIR Parameters
  - GCA Convergence check
  - RINU Frame check
  - GPS Measurement Processing
  - SDINS algorithms
  - Navigation EKF

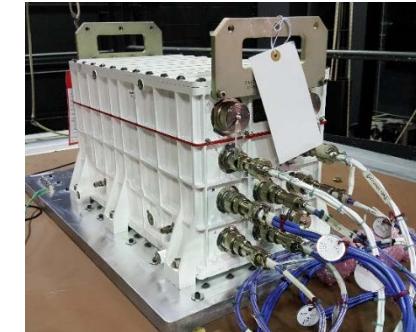
(Block 1B only)



# INS Performance Model

- **RINU: Redundant Inertial Navigation Unit**
- **Level II Requirements Definition**

- Interface and frequency response
- Performance constrained with reference trajectory
- Reduction in requirements with explicit modeling



- **Level III Model Description**

- Detailed instrument error modeling
- Algorithms which affect performance
- Detailed interface model

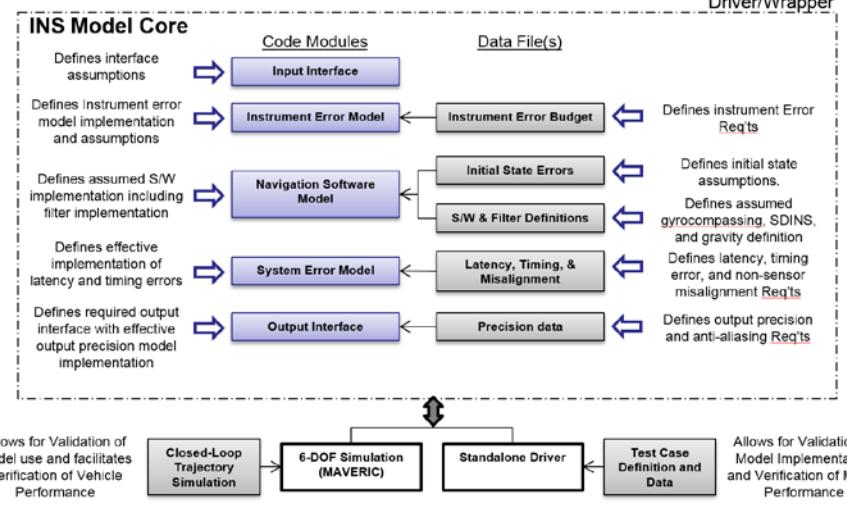
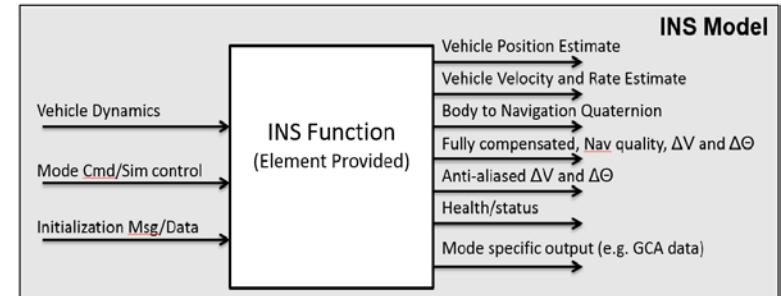
- **Verified against vendor documentation, FQT data, and analysis**

- **Validated against test data**

- GCA 6DOF Test
- Frequency response test
- Vendor ATP/QTP data

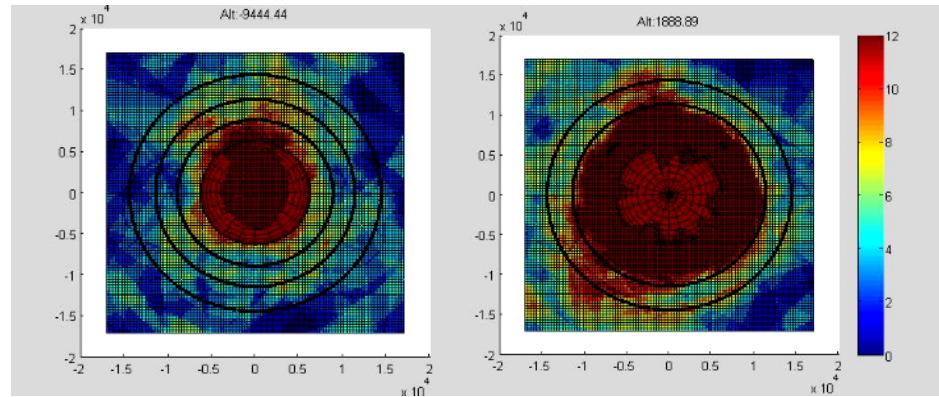
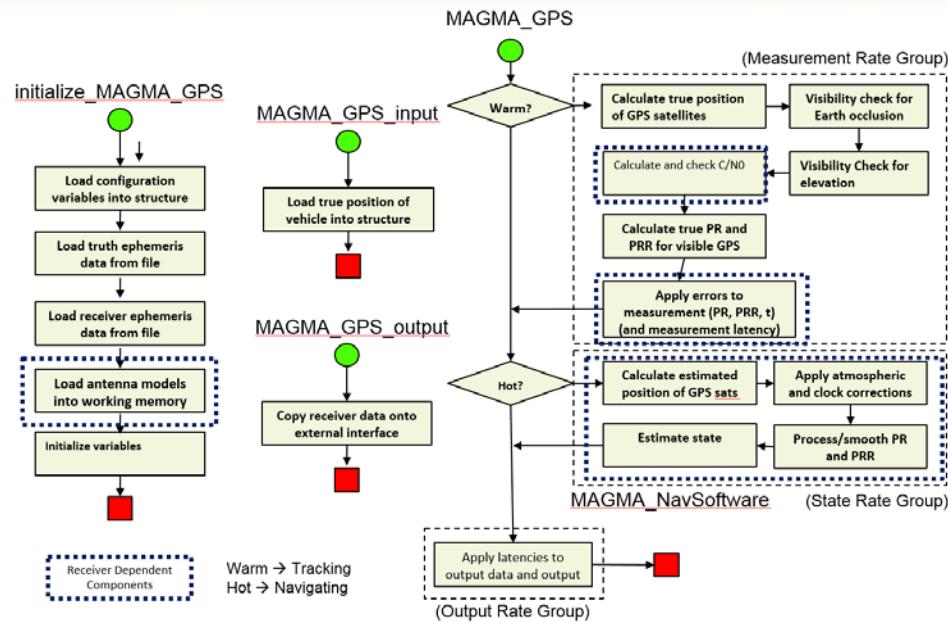
- **Analyses Performed**

- Navigation performance
- Gyrocompassing alignment
- Coning/Sculling
- Integration into vehicle 6DOF



# MAGMA GPS Model

- Marshall Advanced GPS Model for Analysis
- Framework developed to support
  - Requirements development,
  - Early Navigation System design
  - Seed Level III DMM development
- Level II Requirements
  - Interface definition
  - Measurement accuracy
- Functional Components
  - Detailed truth model
  - SV and Receiver Antenna modeling
  - Receiver hardware modeling
  - Receiver software modeling
- Models measurement availability, accuracy, and latency

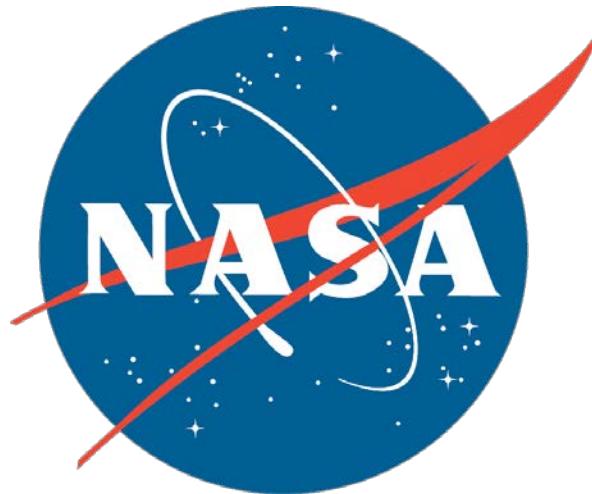


Simulation of GPS availability within 2D plane

# Conclusion

- **Implementation of MBD on SLS has significantly increased efficiency**
  - Reduced requirements burden
  - Provide explicit communication of component and integrated system design
- **Provides a mechanism for**
  - Detailed modeling and design insight
  - Identification of key vehicle sensitivities
  - Gaining additional insight through testing and validation process
  - Enforcing rigor in modeling through validation
- **DMM V&V process forces high fidelity emulation of hardware**
- **Lessons Learned:**
  - Model form and function should consider user and developer
  - GN&C Model
    - Software requirements drive the software test program
    - Approach conflicted with established FSW processes and culture
  - Component models,
    - Good data requirements and supplier integration are key to enabling process
    - V&V plans should be defined early to support data requirements definition and to identify gaps which require additional testing.
    - Sensitivity analyses should be used to identify key performance drivers
    - Commonality between HWIL models and Performance/Analysis models reduces cross-validation effort in verification

# Thank you!



# Any questions?