

***Trustworthy Autonomy
Development & Flight Demonstration
Multi-Monitor Run Time Assurance Research Update***

Mark Skoog
Armstrong Flight Research Center

Research Timeline

1980

Automated Maneuvering Attack System (AMAS)



Automation Research

2000

Automated Collision Avoidance

Air



Ground



Integrated

Dedicated Safety Work for Fighters

AFTI/F-16

Advanced Fighter Technology Integration



AFTI & ACAT/F-16

Automated Collision Avoidance Technology

Quad-Rotor

Transition

Ground Collision Avoidance

GA

Small UAS



Platform Diversity

SUAV/iGCAS/SR22

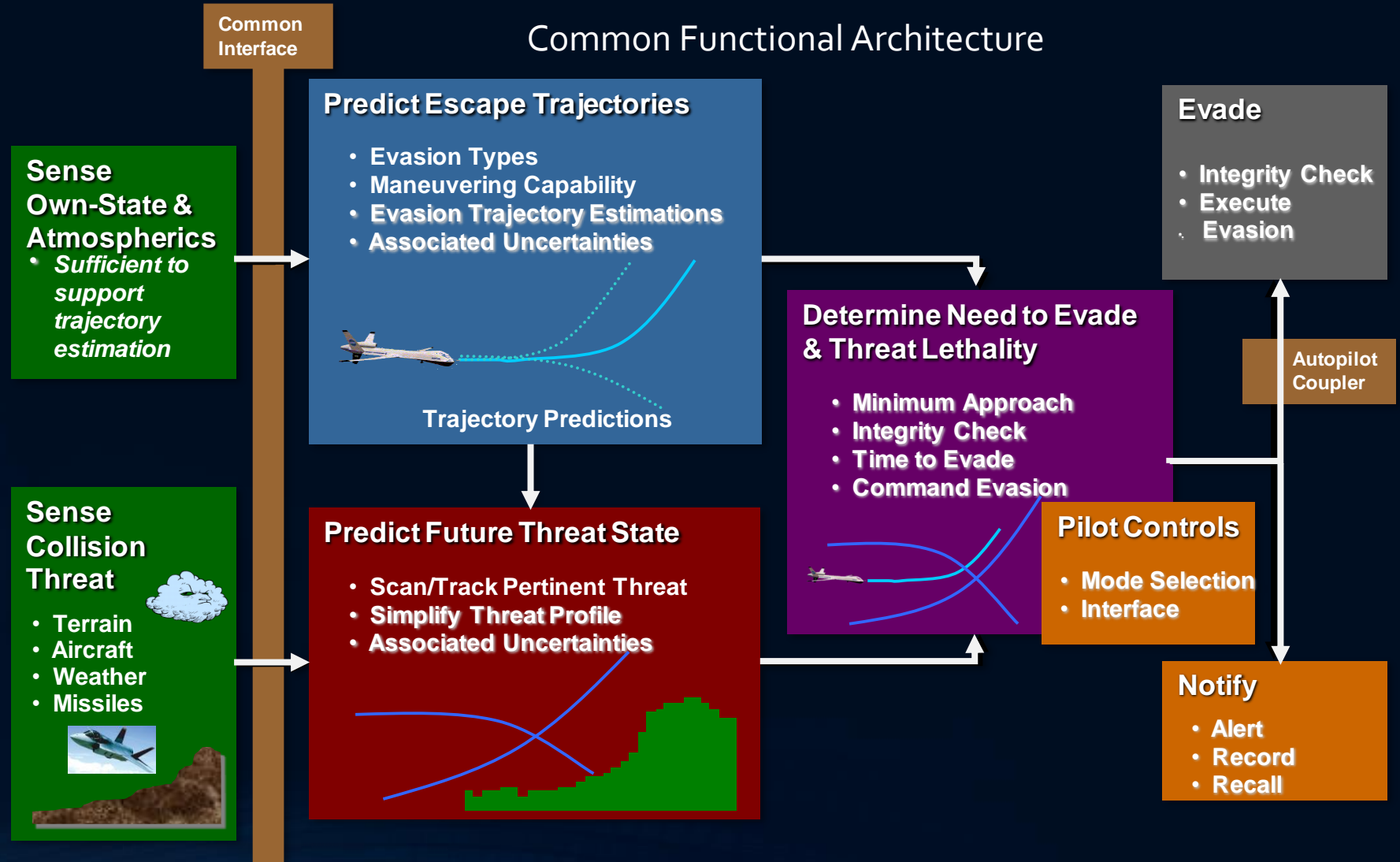
Improved Collision Avoidance System



Ground Collision Avoidance System (GCAS)



Common Functional Architecture



Avoid Collisions



***Do Not
Impede
the Pilot***



sUAV

Automatic Air Collision Avoidance System (Auto ACAS)

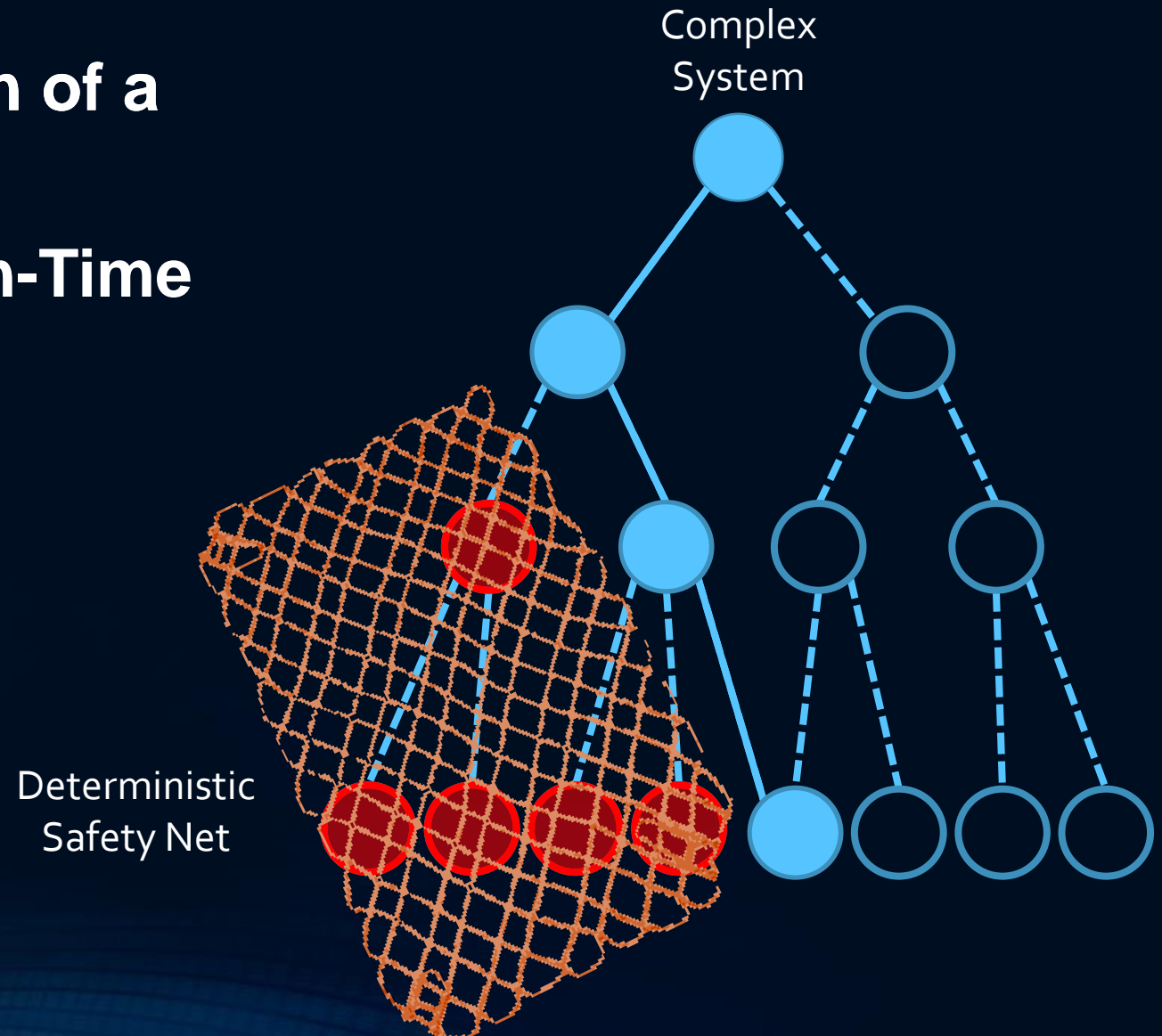


Automatic Integrated Collision Avoidance System (Auto ICAS) - Air & Ground Multi-Ship

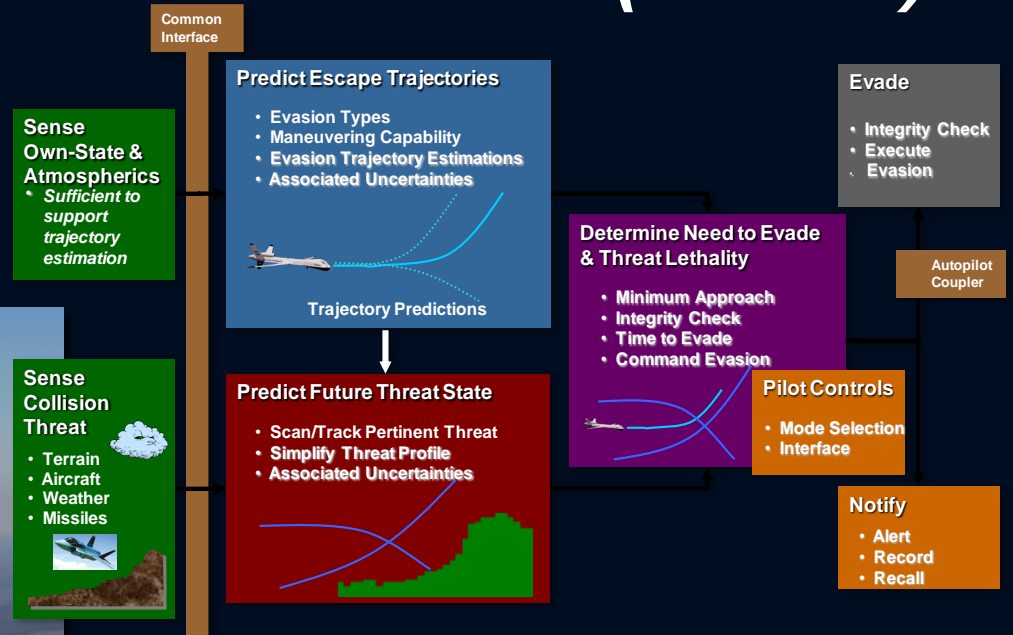
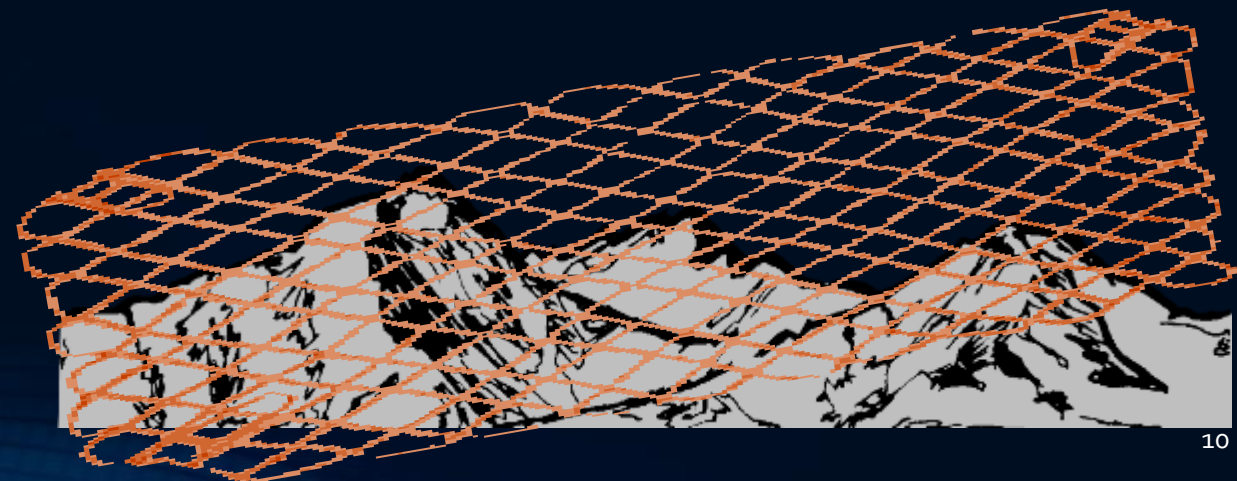


The Challenge of Autonomy

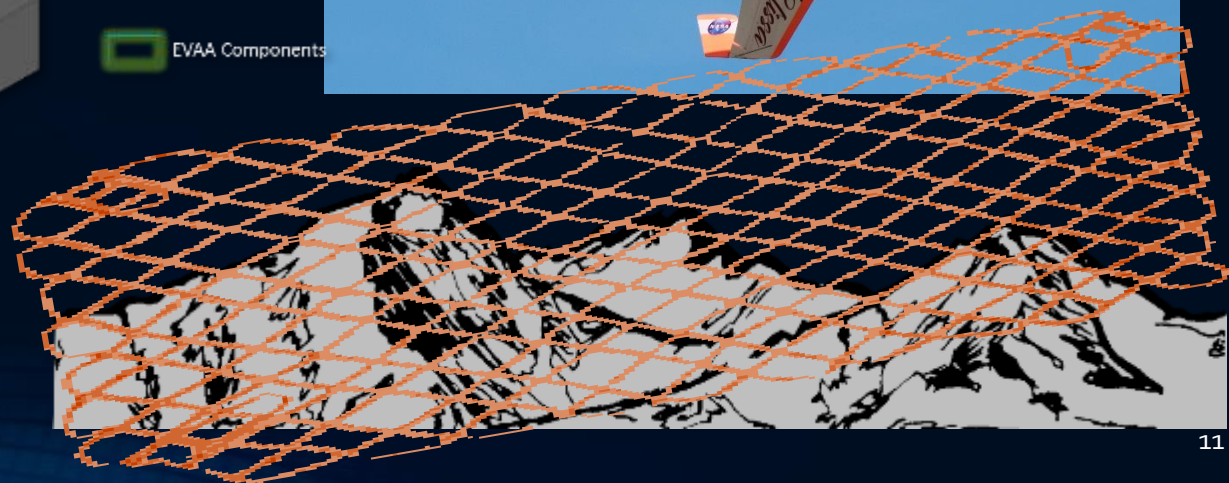
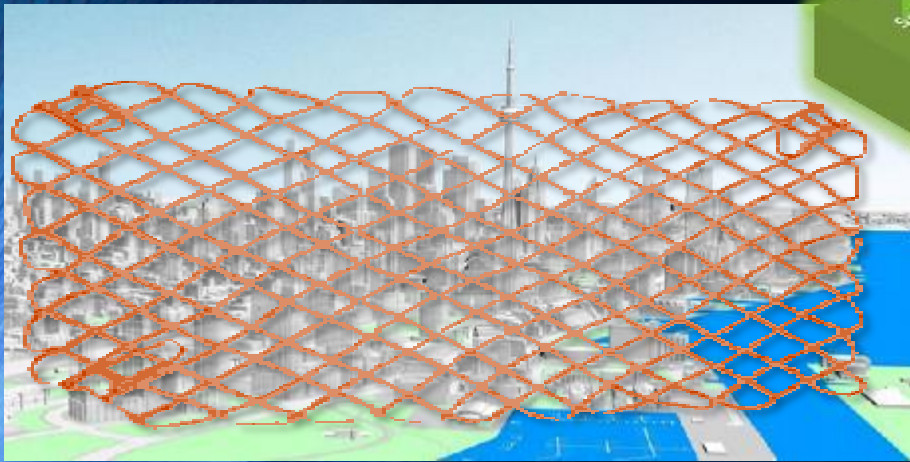
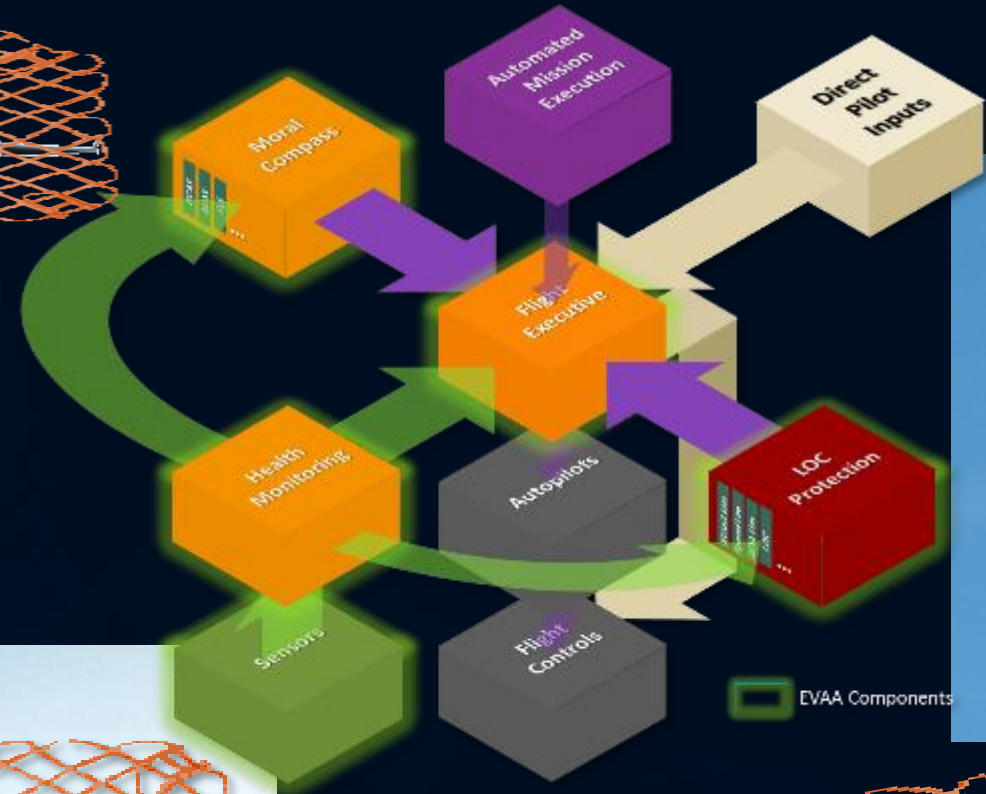
- **Verification & Certification of a Complex System**
- **A Possible Solution – Run-Time Assurance (RTA)**



Ground Collision Avoidance System (GCAS)



Multi-Monitor RTA (MM-RTA) with Risk-Based Decision Making



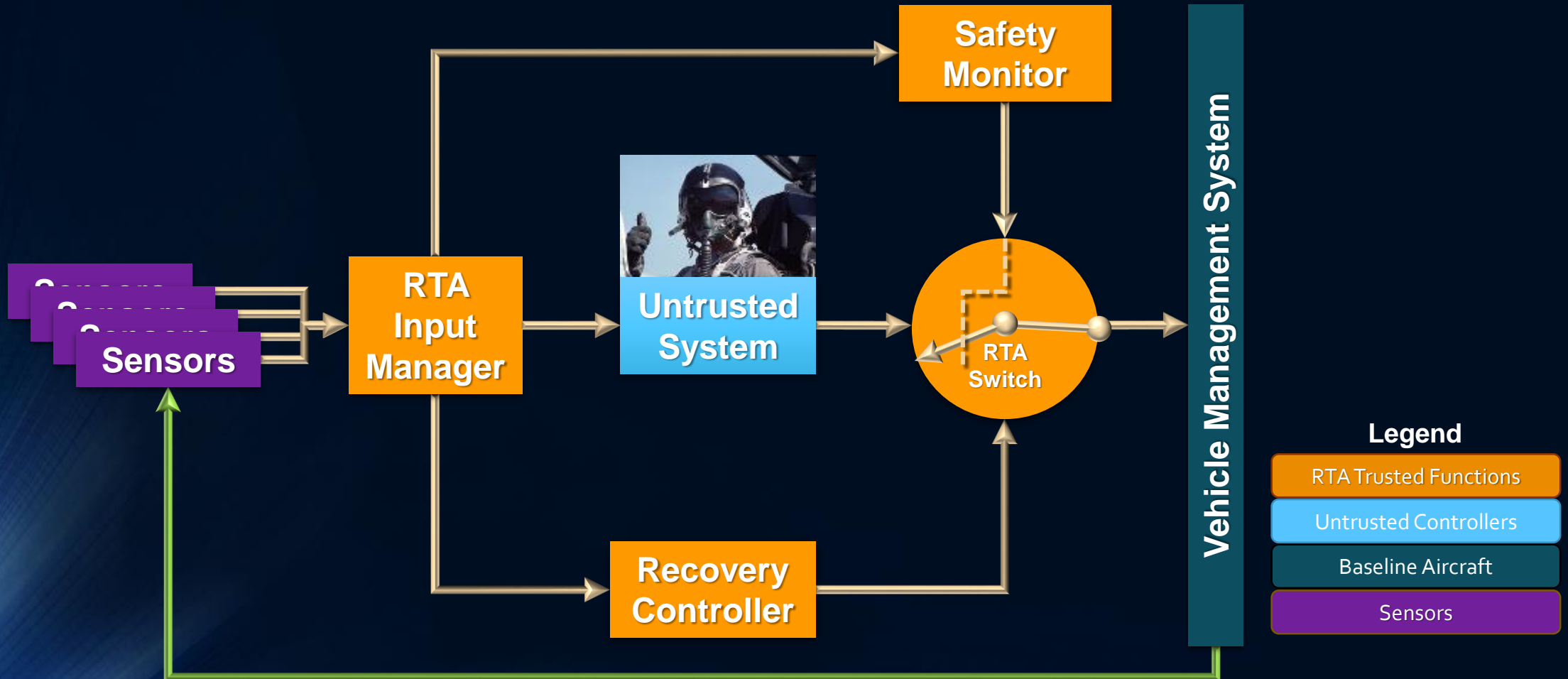
Informing the Standards Community

Research findings vetted with ASTM International through Working Group 53403 (WK53403)

- WK53403 Goal: Develop a standard practice that safely bounds the flight behavior of autonomous UAS
- Involvement originated from AFRC collaboration with FAA regarding Auto GCAS and integrity management work on early autonomy concepts
- Published Industry Standard Practice in Oct 2017

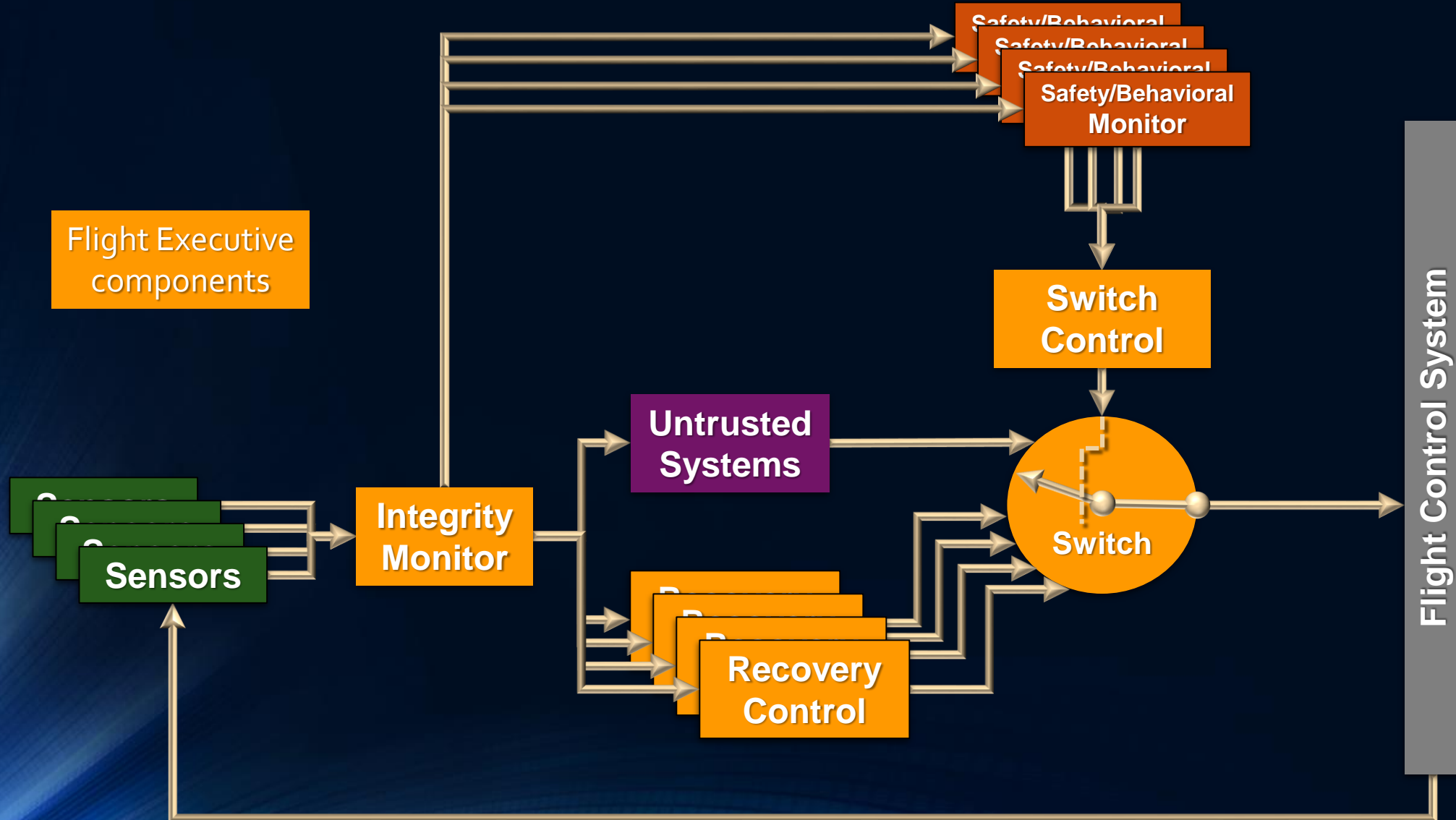


Traditional RTA Framework



MM-RTA Framework

This Work is Unique to AFRC



Phase 1 EVAA Development

Objective

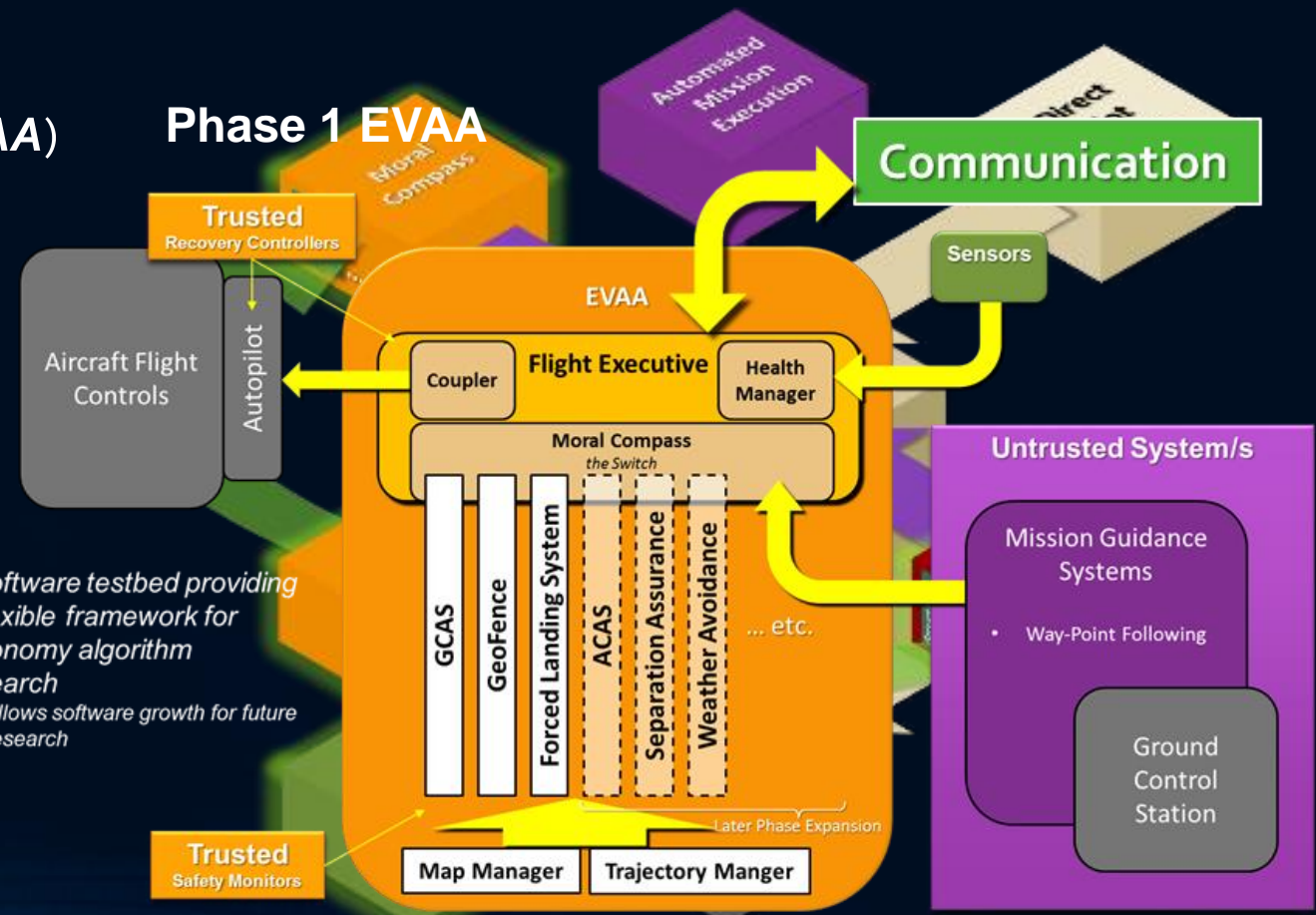
- Develop research findings to inform standards development for certifiable autonomy
- Evaluate the dynamic interaction of an MM-RTA with no integration between monitors

Expandable Variable-Autonomy Architecture (EVAA)

- Stretching the paradigm of autonomy
 - Deterministic Rulesets Bounding Autonomous Behavior
 - Functionally Partitioned Monitors
 - Risk-Based Decision Making
- A process enabling certification
 - Software Architecture/Framework
 - Test Approach
- Scalable autonomy
 - Pilot-in-the-Loop to “Fully Autonomous”

Low Altitude Small UAS Test Ranges (LASUTR)

- A tool for certification
- High-risk integrated research



MM-RTA: Key EVAA Accomplishments

• Aircraft/Testbed Modifications

- Research Processor Integrated Jan 17
- Sound & Lighting System Installed May 17

• Research System

- Functional Requirements Completed Nov 16
- Design Completed Feb 17
- Coding Completed Mar 17
- Patent for GCAS Monitor Issued May 17

• V&V

- Hardware in the Loop Sim Completed Mar 17
- Integrated V&V Completed May 17

• Flight Test

- Aircraft Characterization Test Completed Mar 17
- EVAA Flight Test Began May 17

• Reporting

- Update to FAA & ASTM May 17



Flight Controls



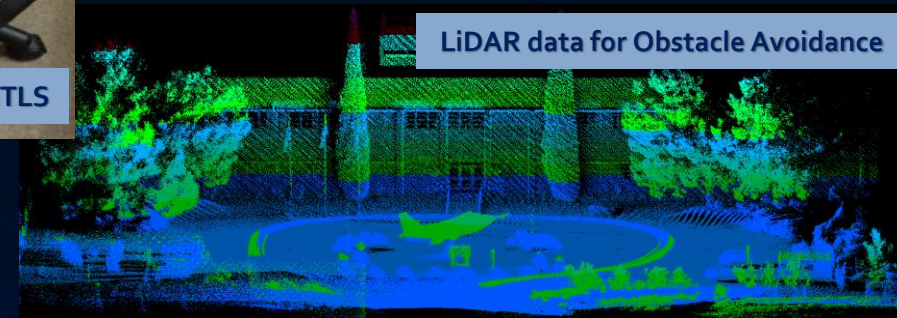
EVAA Processor



MM-RTA Flight Test Began



Development Environment HITLS

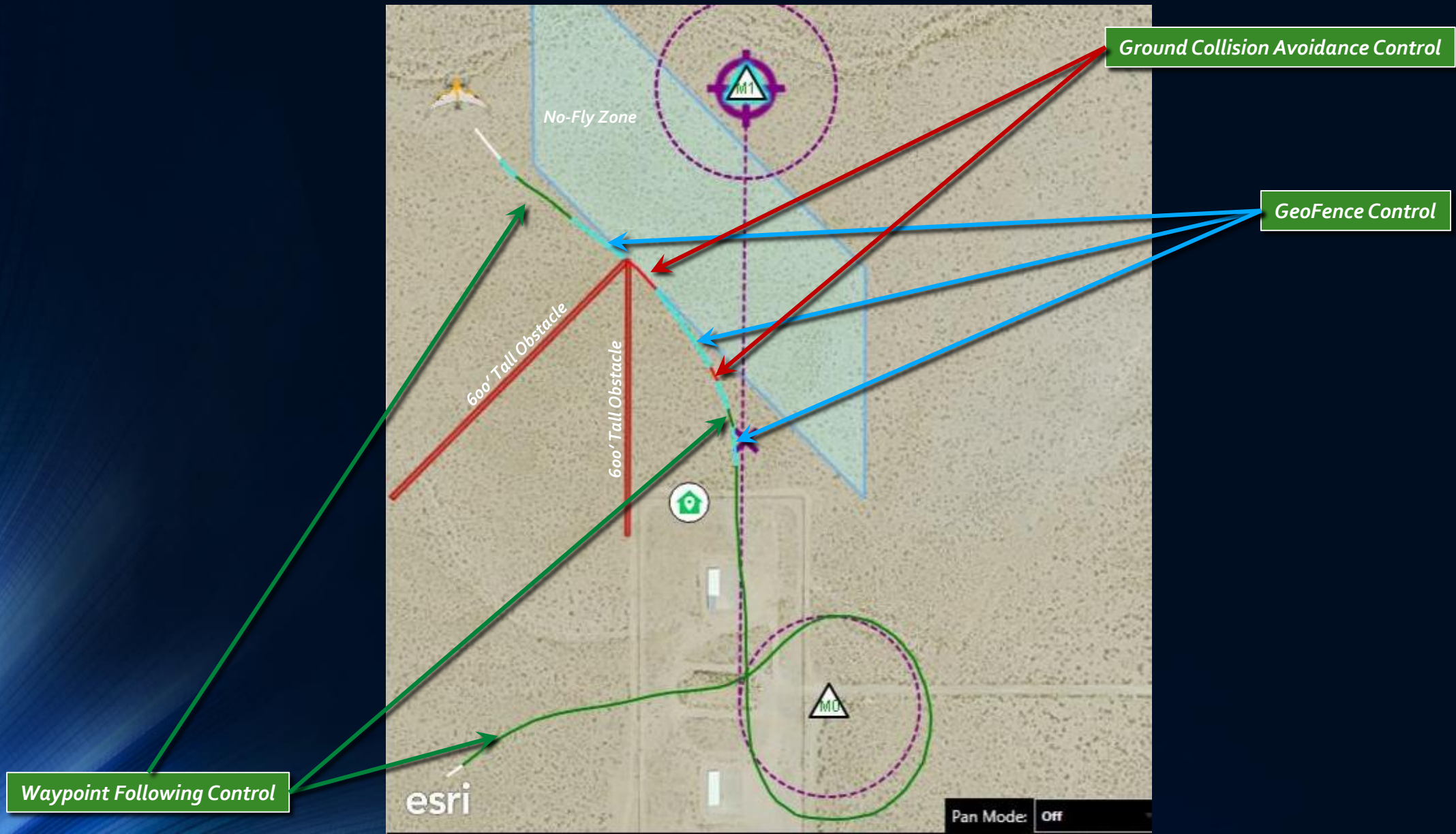


LiDAR data for Obstacle Avoidance



Flight Test Accomplishments

EVAA Command Delegation with Conflicting Multi-Monitor Resolution

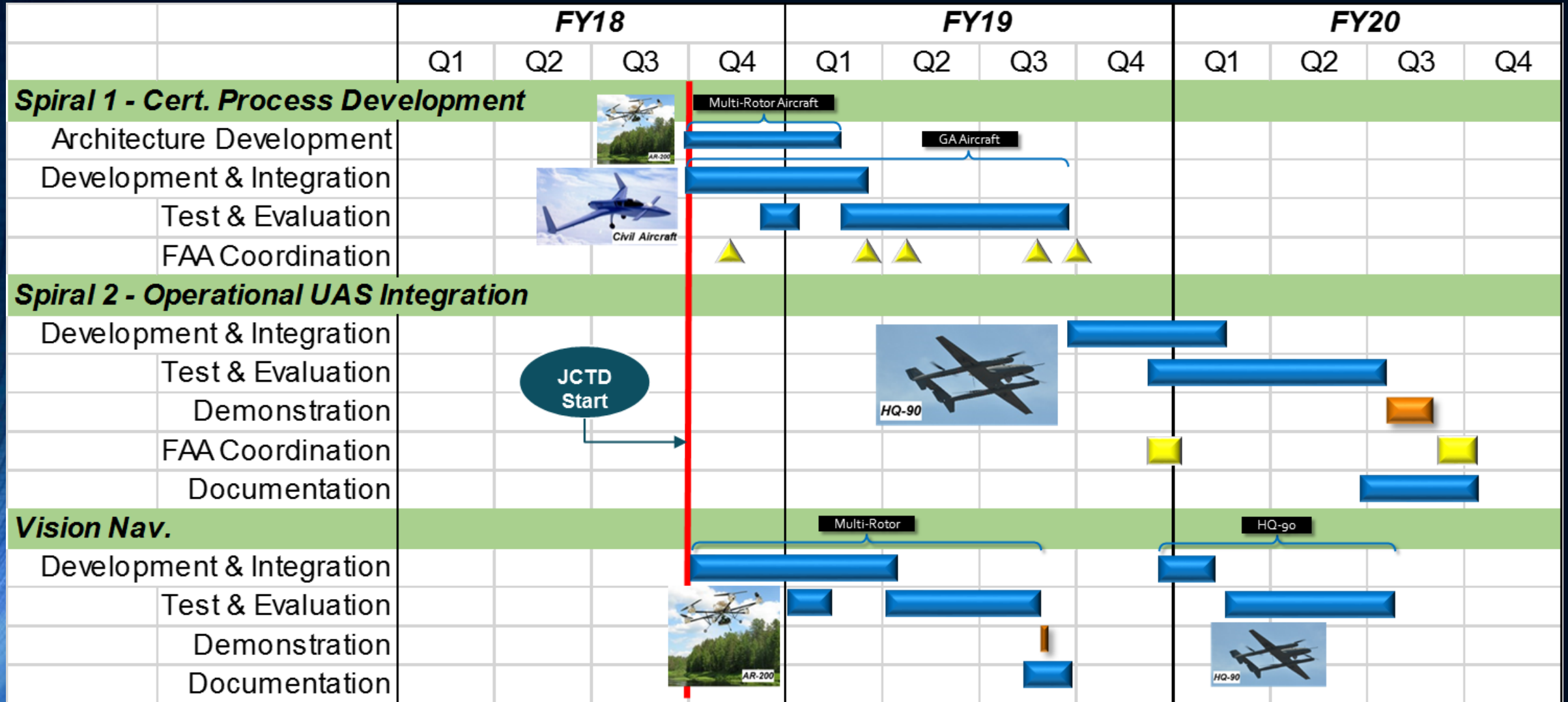


EVA A Phase 2

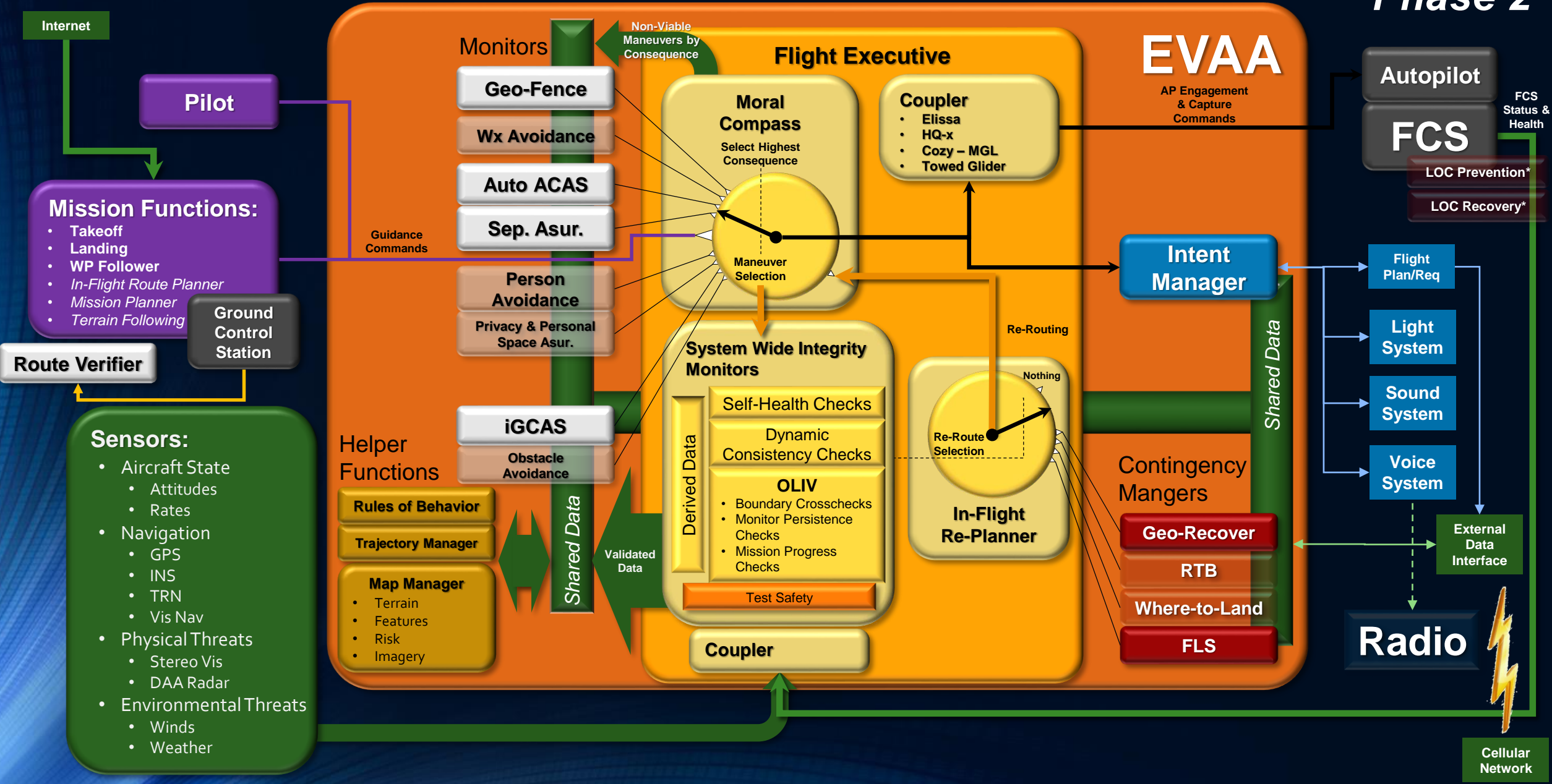
EVA A Phase 2

EVAA Phase 2 Development

OSD's JCTD Resilient Autonomy Project



Expandable Variable-Autonomy Architecture (EVAA) Phase 2



OSD Resilient Autonomy

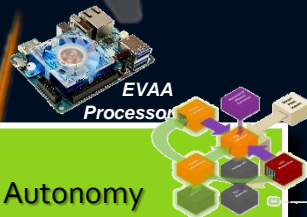
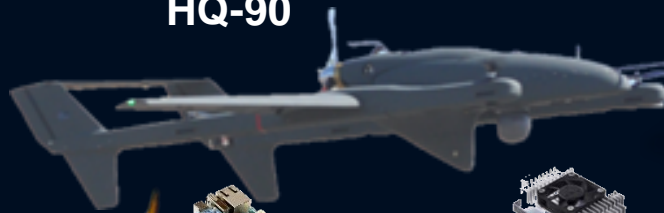
Big Data

Cellular or Other Link



Non-Safety Critical Link

HQ-90



EVAA

- Certifiable Autonomy
- Safe Pilotless BLOS Ops
- Risk-Based Decision Logic
- Easily Tailored to any Vehicle & Mission

ADS-B & DAA Radar



Automatic Well-Clear & Air Collision-Avoidance

Automatic Airspace Boundaries & Safe-Ditch Contingency Management

Visual-Nav System – transient operation in GPS denied or degraded environment



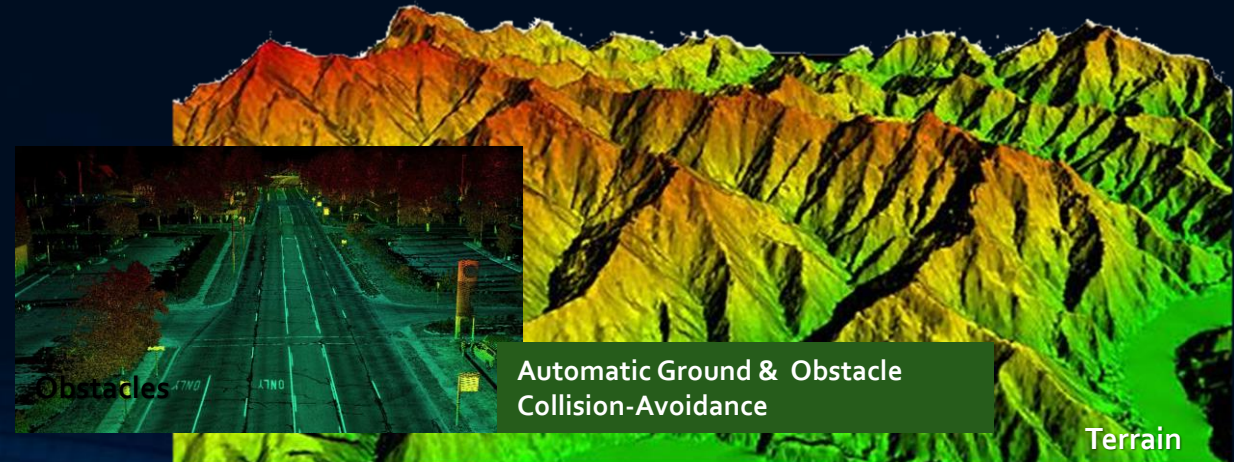
Cooperative & Non-Cooperative Targets



Non-Safety Critical Link

GCS Features

- Mission Plan Verification
- Situational Awareness Displays



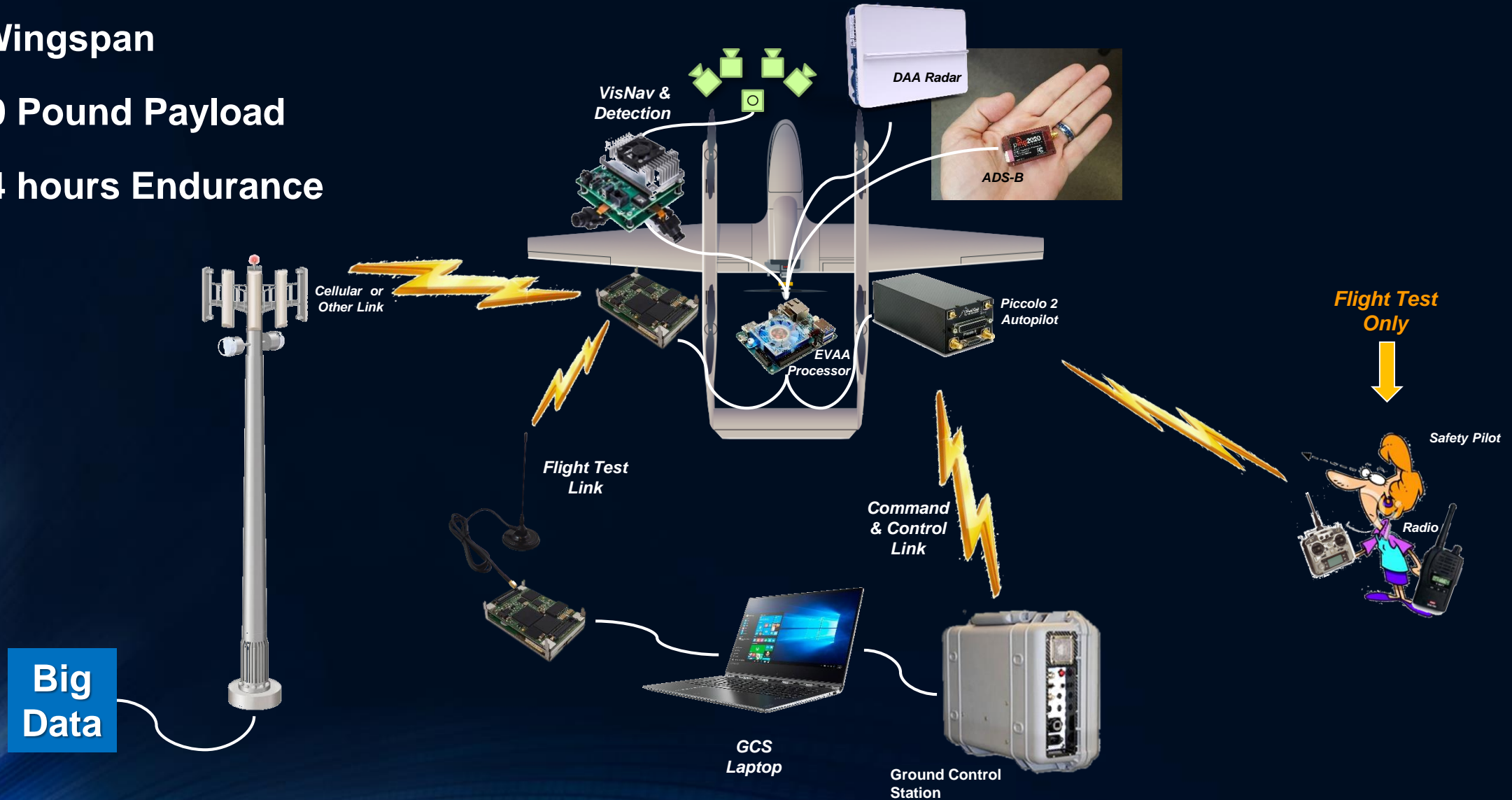
Obstacles

Automatic Ground & Obstacle Collision-Avoidance

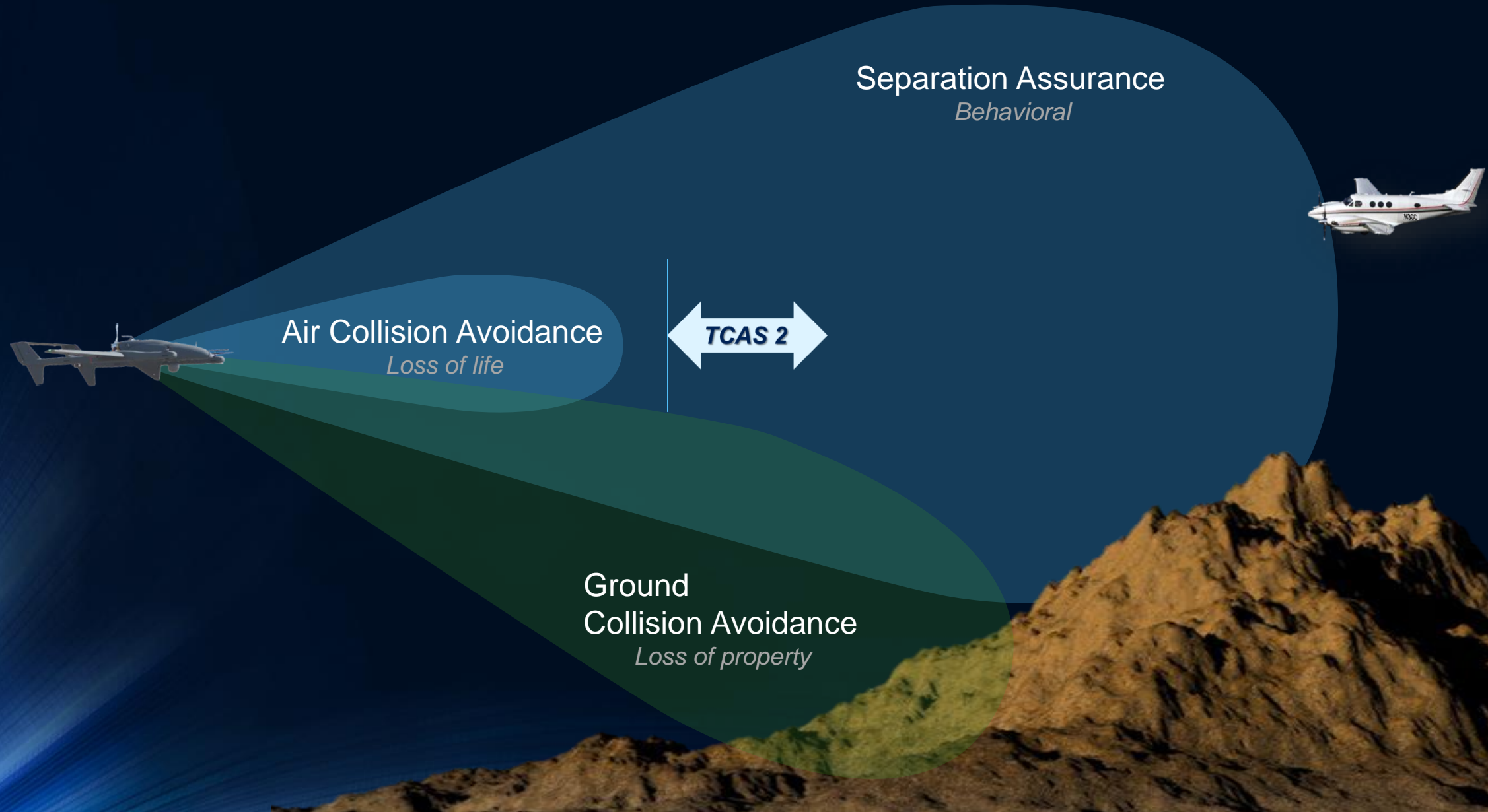
Terrain

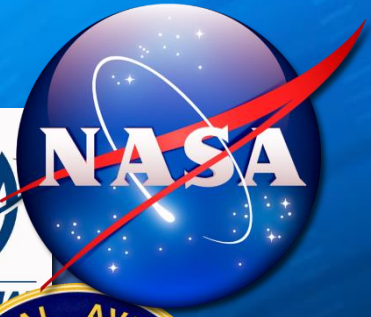
HQ-90 Testbed

- 103 Lbs. Max Gross Takeoff Weight
- 14' 8" Wingspan
- 20 to 30 Pound Payload
- 12 to 24 hours Endurance



DAA Approach





Questions

