



# Current EBC Development and Testing at NASA

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# Outline

- Background
- EBC Testing Facilities
- EBC Development
  - 1. Slurry EBCs w/ 2700F Bond Coat Capability
  - 2. Plasma-Sprayed Modified  $\text{Yb}_2\text{Si}_2\text{O}_7$  EBCs (w/ Si Bond Coat)

# Environmental Barrier Coating (EBC)



- An external coating to protect CMCs from rapid recession by H<sub>2</sub>O
- Enabling technology for CMCs

## Recession Model



$$\text{Volatility} \propto \frac{v^{1/2} \times P(\text{H}_2\text{O})^2}{(P_{\text{TOTAL}})^{1/2}}$$

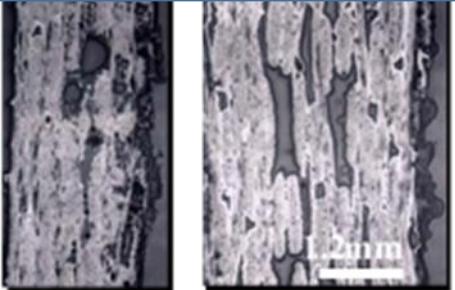
v : gas velocity

P(H<sub>2</sub>O) : water vapor pressure

P<sub>TOTAL</sub> : total pressure

E. J. Opila et al., Am. Ceram. Soc., 80[1], 197-205 (1997)

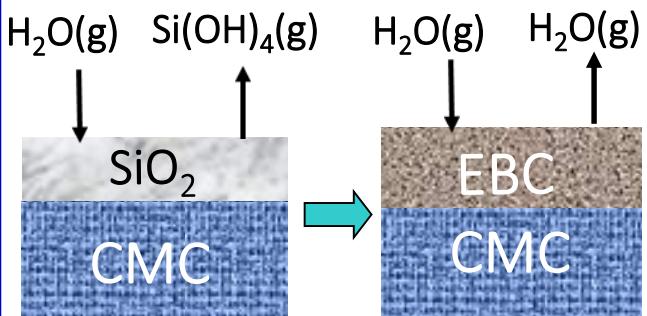
## Solar Turbine Engine Test



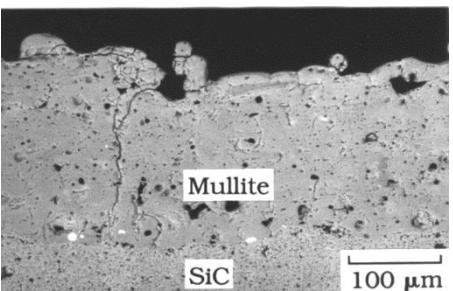
Inner Liner Outer Liner

M. van Roode, et al., J. Eng. Gas Turbines & Power, 129 [1], 21-30, 2007.

## EBC

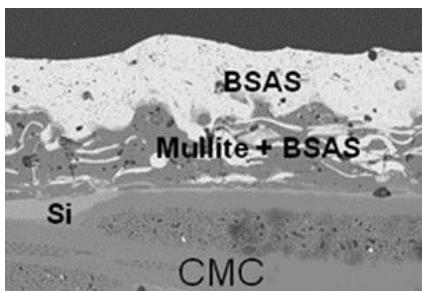


K. N. Lee, Surf. and Coat. Tech, 133-134 1-7 (2000).



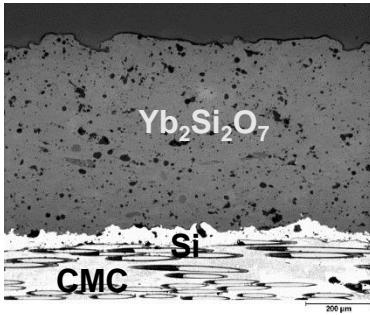
**Mullite Coating  
(NASA-1993)**

K. N. Lee, "J. Am. Ceram. Soc., 78(3) 705-710 (1995).

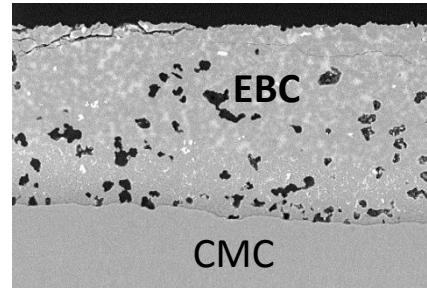


**Gen 1 EBC  
(EPM: NASA-GE-PW-1997)**

**Silicon bond coat (mp = 1416°C)**



**Gen 2 EBC  
(UEET: NASA-2003)**



**Next Gen EBC  
(NASA Developmental)**  
**Bond coat mp > 1500°C**

# Key EBC Failure Modes

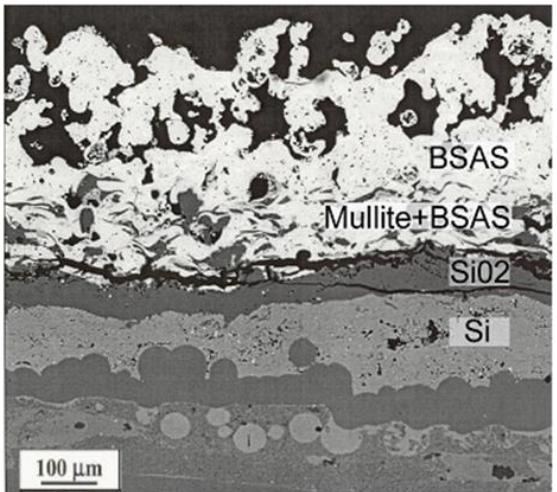
<ul style="list-style-type: none"> <li>• Recession by water vapor</li> </ul>	<p>Sand/Volcanic Ash Ingestion</p> <ul style="list-style-type: none"> <li>• CMAS: (Ca-Mg-Al-Silicates)</li> </ul>
<ul style="list-style-type: none"> <li>• Steam oxidation</li> </ul>	<ul style="list-style-type: none"> <li>• Erosion</li> </ul>
<p>Thermal Stress Mech Stress</p> <ul style="list-style-type: none"> <li>• Thermal fatigue</li> <li>• Thermo-mechanical fatigue</li> </ul>	<ul style="list-style-type: none"> <li>• Foreign/Domestic Object Damage</li> </ul>

**Synergy between failure modes likely leads to EBC failure**

# Steam Oxidation-Induced EBC Failure

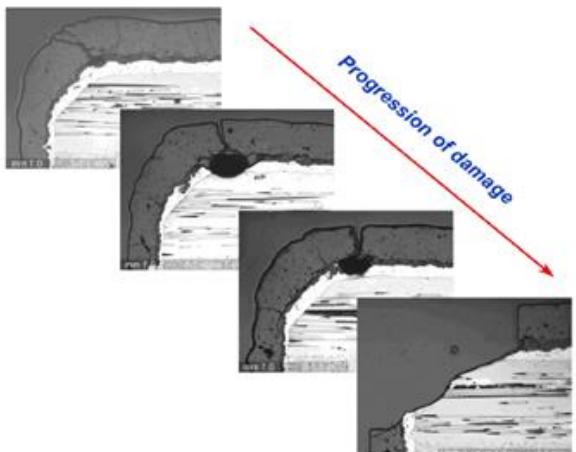


## 15,144-h Solar Combustor Liner Engine Test



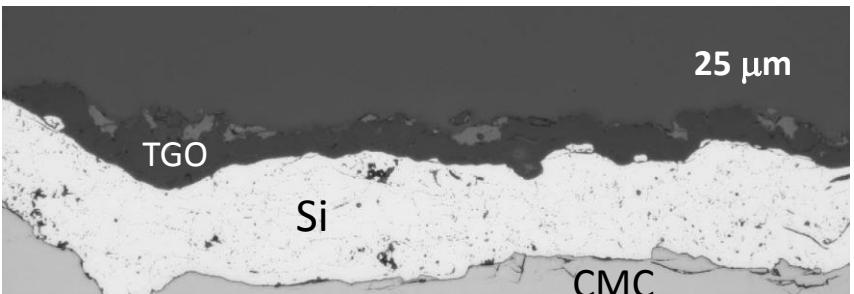
J. Kimmel et al., ASME paper GT2003-38920, ASME TURBO EXPO, Atlanta, GA, USA, June 16-19, 2003.

## 5,366-h Rig Test



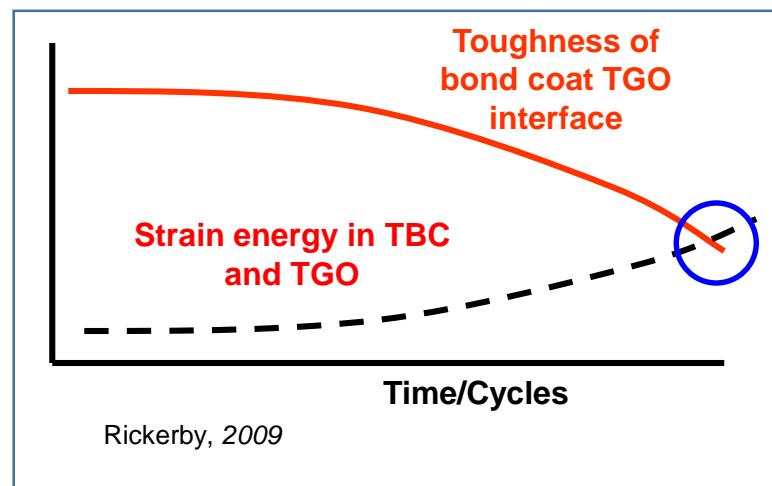
## NASA Steam Cycle Test (Si/Yb<sub>2</sub>Si<sub>2</sub>O<sub>7</sub>)

2400F, 90% H<sub>2</sub>O-O<sub>2</sub>, 550 hr/550 cycles



### Failure mode is similar to TBC:

Failure driven by the stored energy in the ceramic and TGO and the decrease in the toughness of the bond coat/TGO interface





- EBC Testing Facilities

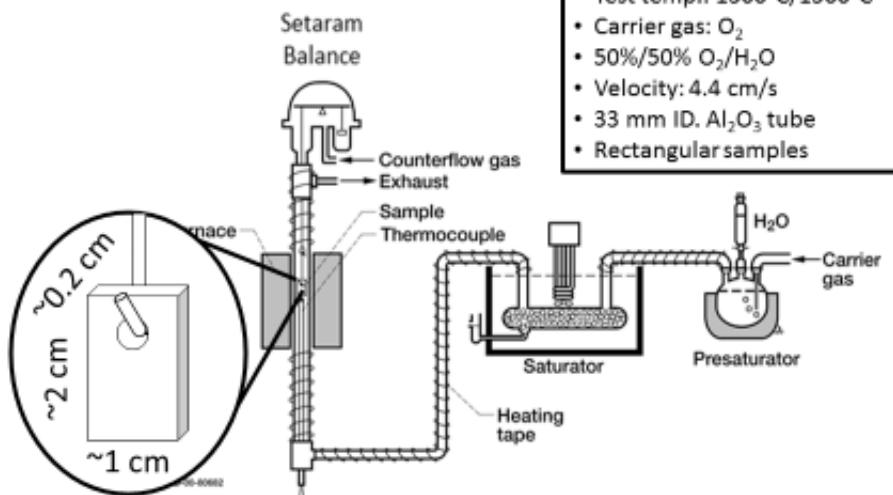
# NASA EBC Test Rigs



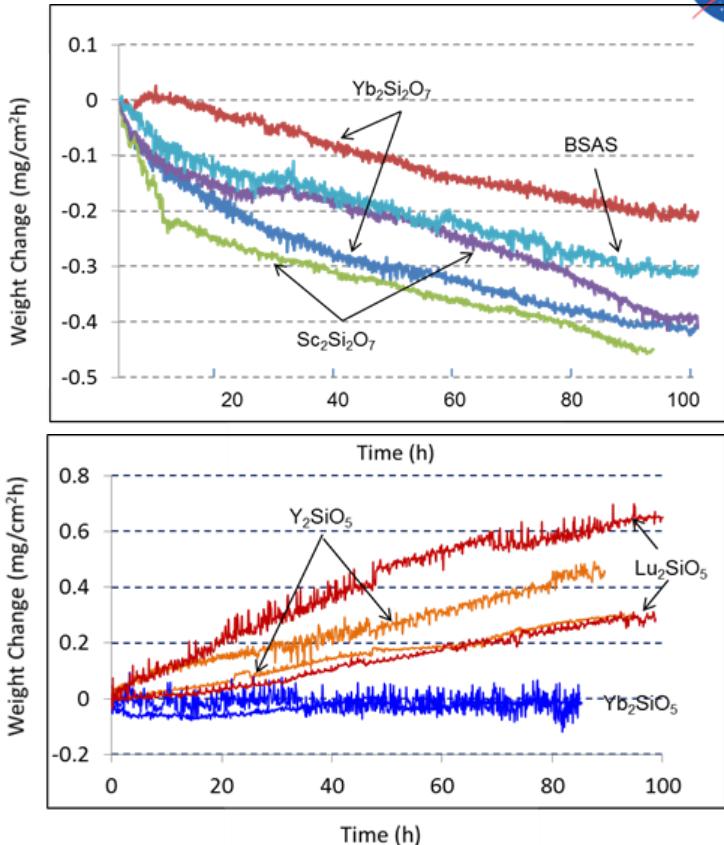
Rig	Capability	Failure modes to be tested
Mass Spectrometer	$P(H_2O) = N/A$ $v = N/A$ $P_{total} = N/A$	Recession (High pressure measurement of reaction products and Low pressure measurement of activities)
Steam TGA	$P(H_2O) = \text{up to } \sim 0.5 \text{ atm}$ $v \sim 10 \text{ cm/s}$ $P_{total} = 1 \text{ atm}$	Recession (Initial screening of candidate materials).
Mach 0.3 Burner rig	$P(H_2O) = \sim 0.1 \text{ atm}$ $v = 230 \text{ m/s}$ $P_{total} = 1 \text{ atm}$	CMAS, Erosion, FOD
Steam cycling rig	$P(H_2O) = \text{up to } \sim 1 \text{ atm}$ $v = \text{a few cm/s}$ $P_{total} = 1 \text{ atm}$	Steam oxidation
High heat flux laser rig	$P(H_2O) = \text{ambient air}$ $v = \text{zero}$ $P_{total} = 1 \text{ atm}$	Thermal fatigue in temp gradient Thermo-mechanical fatigue in temp gradient
Natural gas burner rig	$P(H_2O) \sim 0.5 \text{ atm}$ $v \sim 250 \text{ m/s}$ $P_{total} = 1 \text{ atm}$	Recession Thermal fatigue in temp gradient (Coupons, Tensile bars, components)
CE-5 combustion rig	$P(H_2O) \sim 3 \text{ atm}$ $v \sim >30 \text{ m/s}$ $P_{total} \sim 30 \text{ atm}$	Steam oxidation w/ temperature gradient Recession (Coupons, Tensile bars, components)

- Combinations of rigs to investigate synergy between failure modes

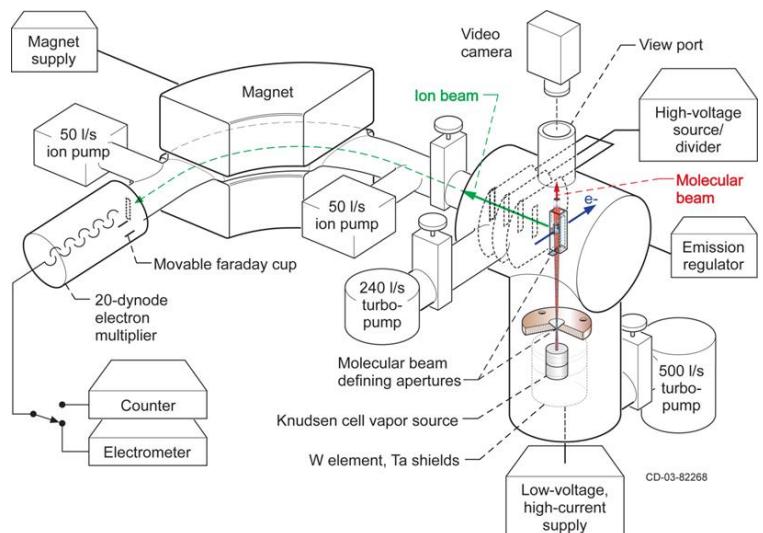
# Steam TGA



K. N. Lee, D. S. Fox, and N. P. Bansal, *J. Euro. Ceram. Soc.* **25**, 1705-1715 (2005).



# Mass Spectrometer

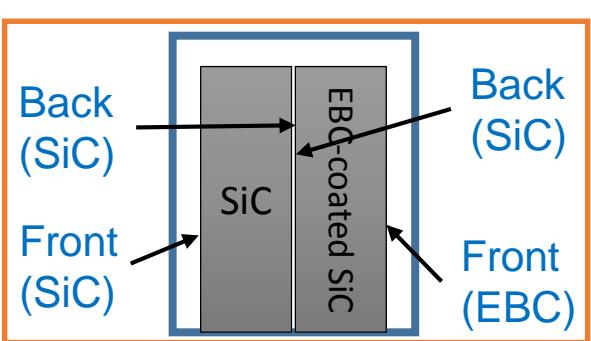
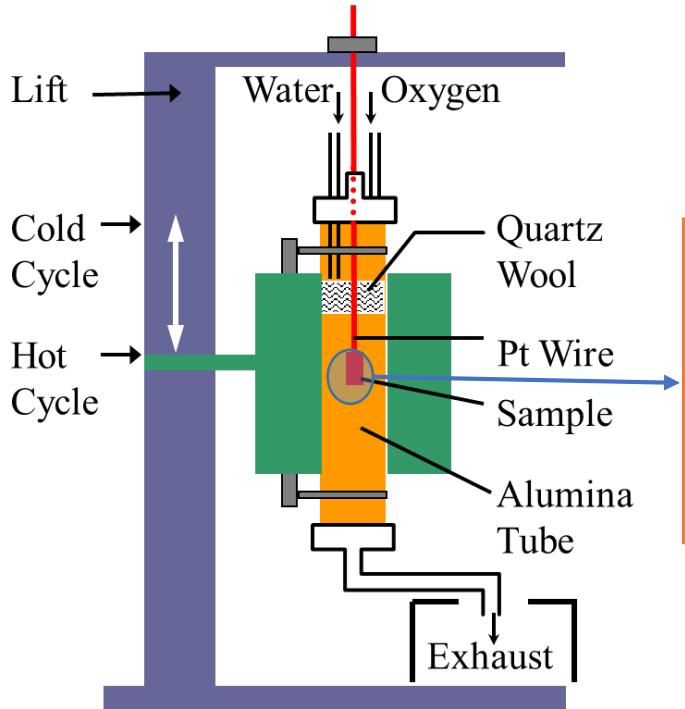


G. Costa and N.S. Jacobson, *J. Eur. Ceram. Soc.* 2015

	RE = Y	RE = Yb
a(SiO <sub>2</sub> ) <sub>RE<sub>2</sub>Si<sub>2</sub>O<sub>7</sub></sub>	0.281	0.194
a(SiO <sub>2</sub> ) <sub>RE<sub>2</sub>SiO<sub>5</sub></sub>	0.000804	0.00298
a(SiO <sub>2</sub> ) <sub>RE<sub>2</sub>Si<sub>2</sub>O<sub>7</sub></sub> / a(SiO <sub>2</sub> ) <sub>RE<sub>2</sub>SiO<sub>5</sub></sub>	350	65

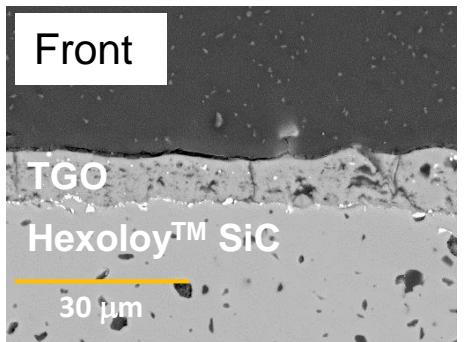
# Steam Cycle Rig

(1 hr at Temp and 20 min at T<100C)

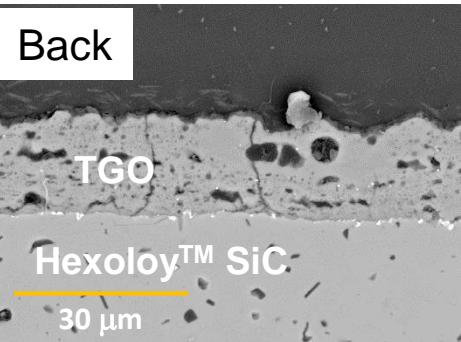


<b>Temp</b>	Up to ~2700F
<b>Velocity</b>	~10 cm/s
<b>Water vapor</b>	Up to ~0.9 atm
<b>Pressure</b>	1 atm

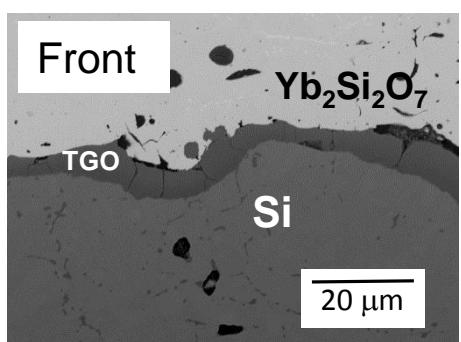
**2400F (1316C) in 90% H<sub>2</sub>O-Bal O<sub>2</sub>, 100h/100 cycles**



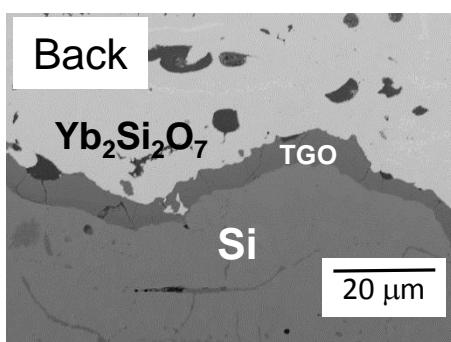
TGO ~ 8.8 μm



TGO ~ 15.6 μm



TGO ~ 4~5 μm



Silica scale is twice thicker on the backside  
- Lower silica volatility on backside due to restricted gas flow

Silica scale was the same on both sides  
- Gas velocity does not affect oxidation rates

# Natural Gas Burner Rig



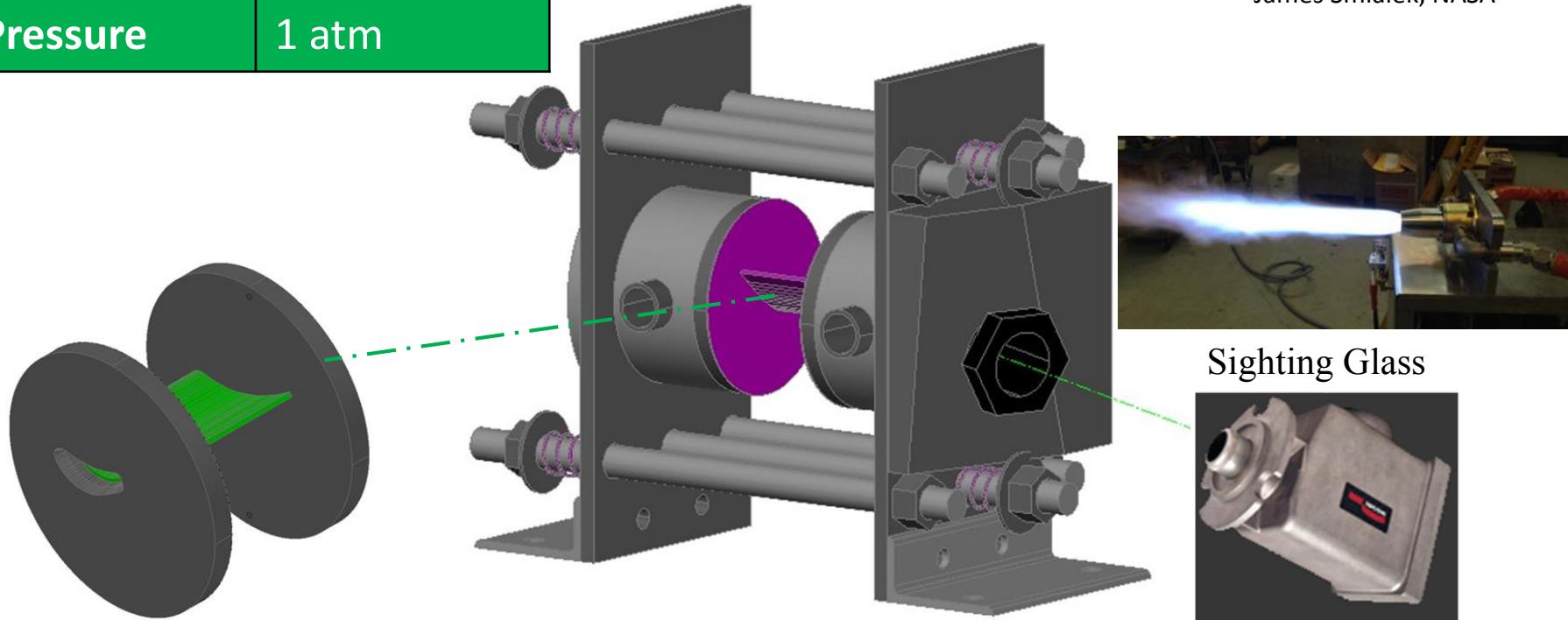
- Thermal cycling test under temp gradient
- Recession test
- Coupon and subcomponent test
- Feb in progress

<b>Temp</b>	Up to ~2700F
<b>Velocity</b>	~250 m/s
<b>Water vapor</b>	~0.5 atm
<b>Pressure</b>	1 atm

	P(total), atm.	%H <sub>2</sub> O	v, m/s	vapor flux	mass loss	
steam tube	1	1	175	1.00	1	
NG-02	1	0.5	250	0.30	193	*
HPBR	15	0.1	30	0.24	155	*
HPBR	6	0.1	300	0.19	124	*
HPBR	6	0.1	185	0.15	97	*
HPBR	6	0.1	30	0.06	39	*
M0.3	1	0.1	100	0.01	5	*
CE-5	30	0.05	30	0.17	110	*

\* per 1" dia. Sample

James Smialek, NASA



# CE-5 Combustion Rig



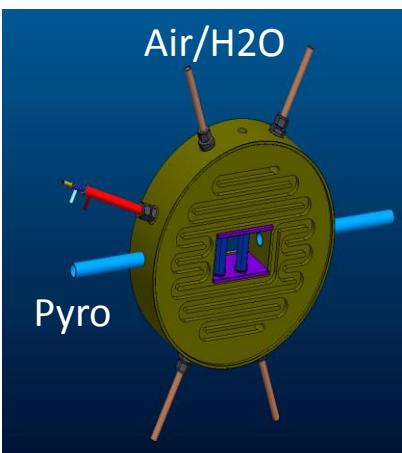
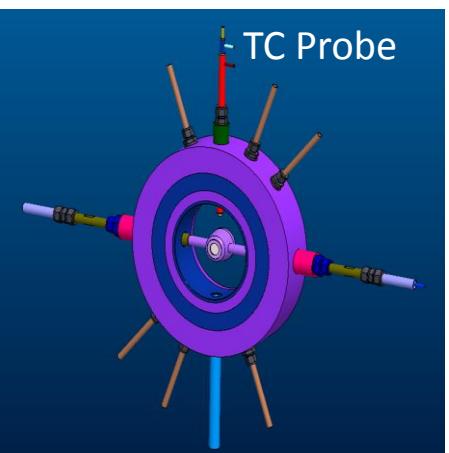
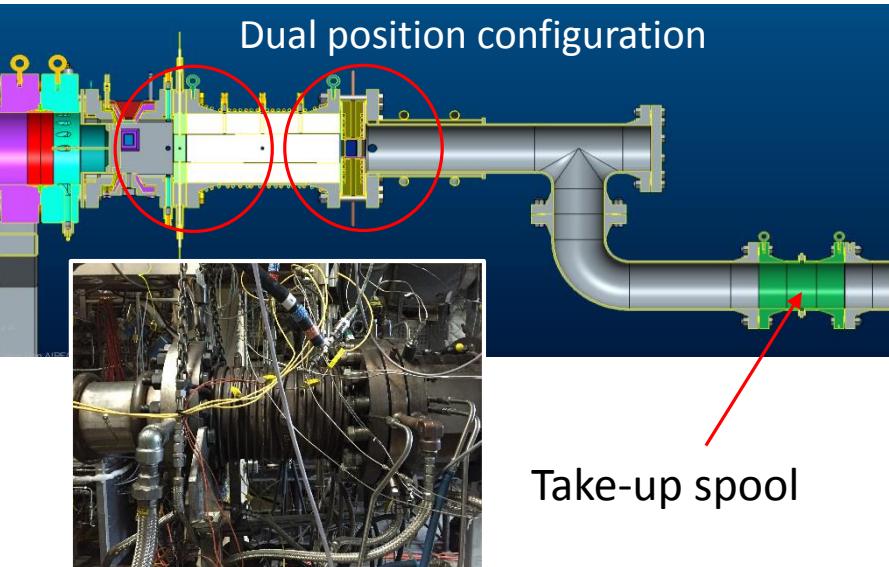
- Coupon & Vane holder Designs

- Button Sample Holder
  - (1) 1" dia button
  - Backside cooling
  - Fab in progress
- Vane pack sample holder
  - (2) 3" x 3" vanes
  - Backside cooling
  - Fab in progress

- Flexible Configurations

- a) Either holder downstream as piggy-back to injector testing
- b) Coupon upstream + Vane downstream as stand alone testing
- c) Investigating “dog bone”, CMC panel, and combustor liner configurations

Temp	Up to ~3000F
Velocity	>30 m/s
Water vapor	~3 atm
Pressure	~30 atm





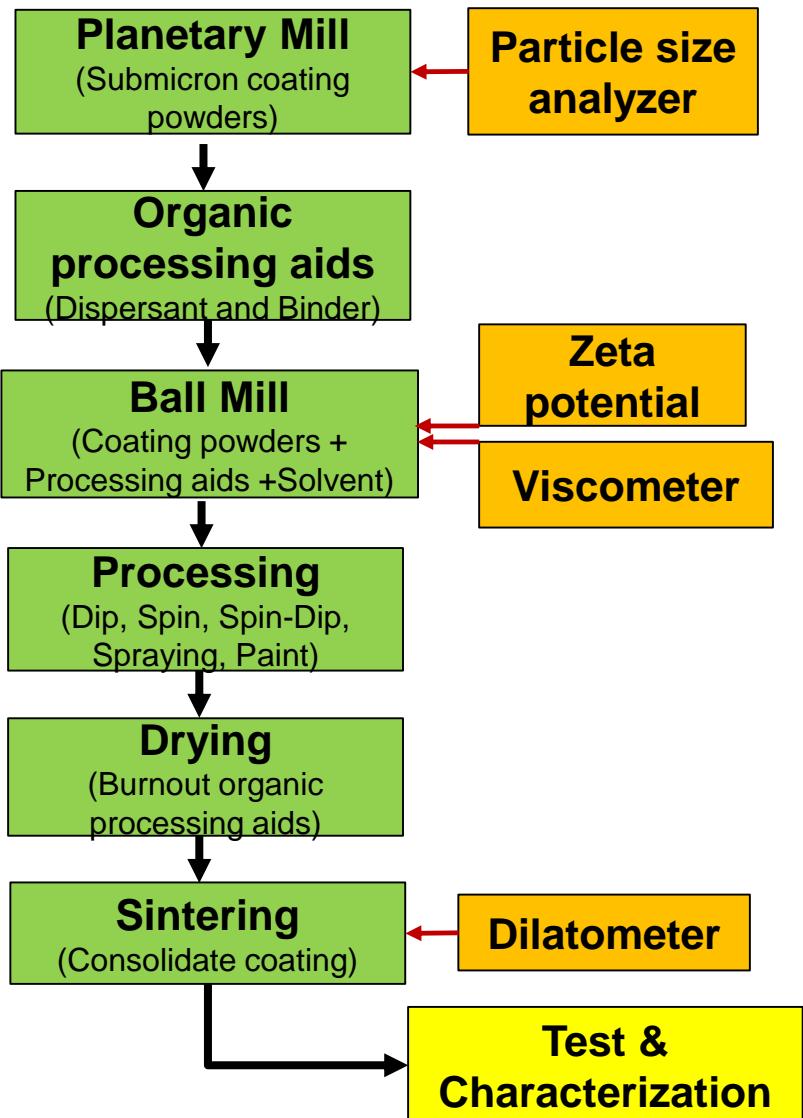
- EBC Development
  - 1. Slurry EBCs w/ 2700F Bond Coat Capability
    - EBC on CMC Coupons
    - EBC on SiC Heating Element

# Slurry EBC Flow Diagram

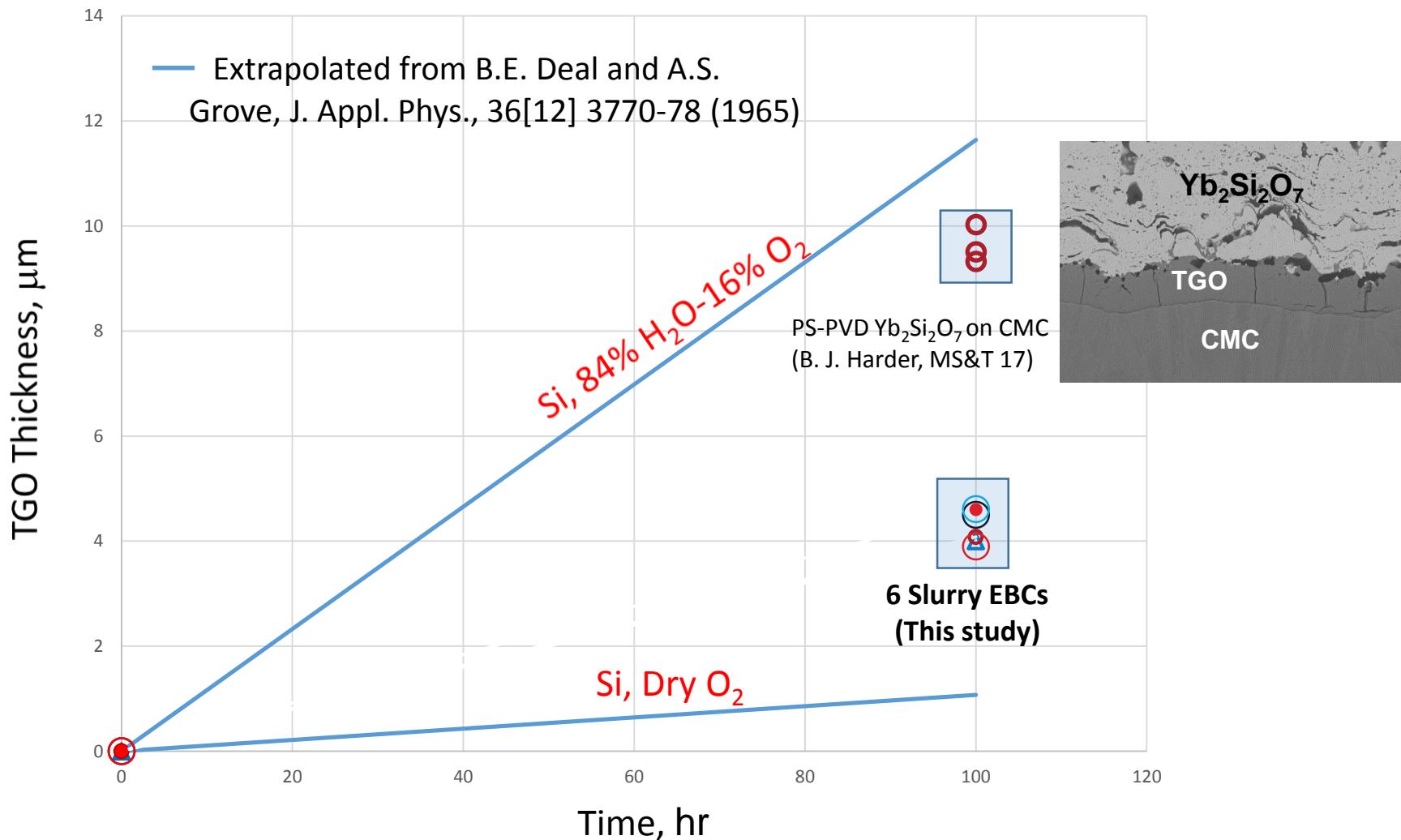


## Processing

## Characterization



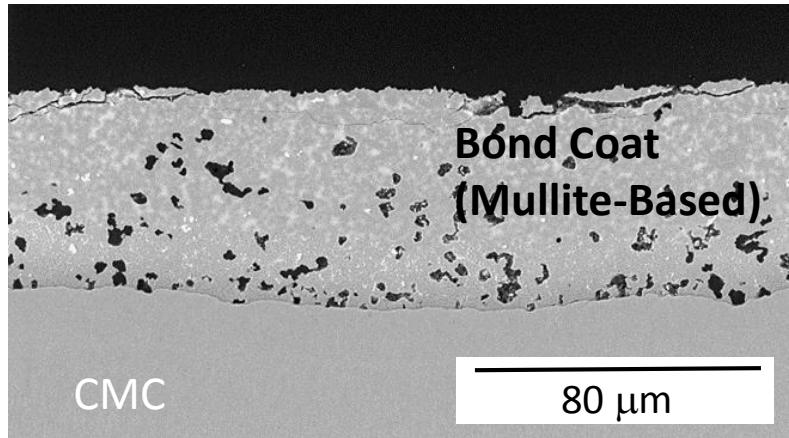
# Steam Oxidation of Slurry EBCs on CMC at 2600F (1 hr at 2600F (1427°C) / 20 min at T<100C, 90% H<sub>2</sub>O)



- TGO (~4 μm at 100h) is thinner than PS-PVD baseline Yb<sub>2</sub>Si<sub>2</sub>O<sub>7</sub> by ~2.5x
- Optimization and long-term testing (future work)

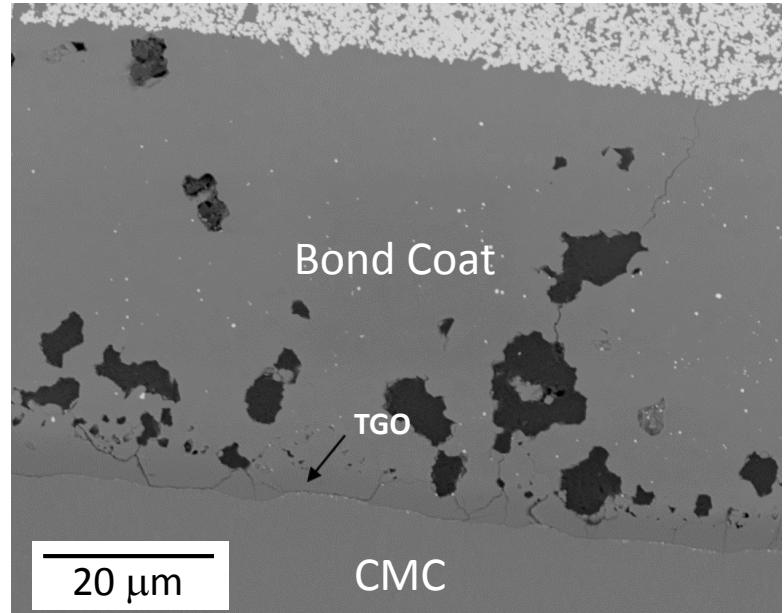
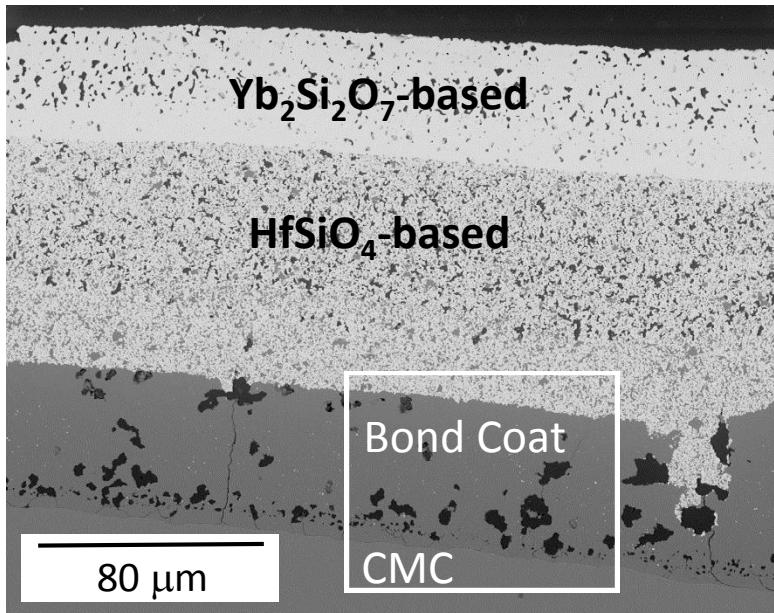
# Slurry EBC

## As-Sintered Bond Coat

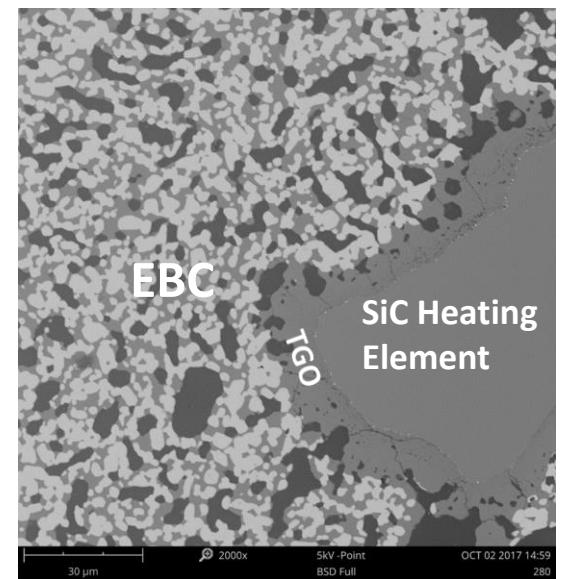
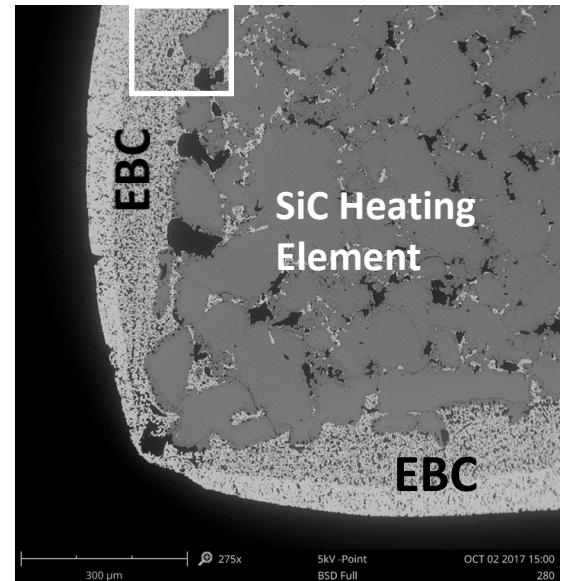
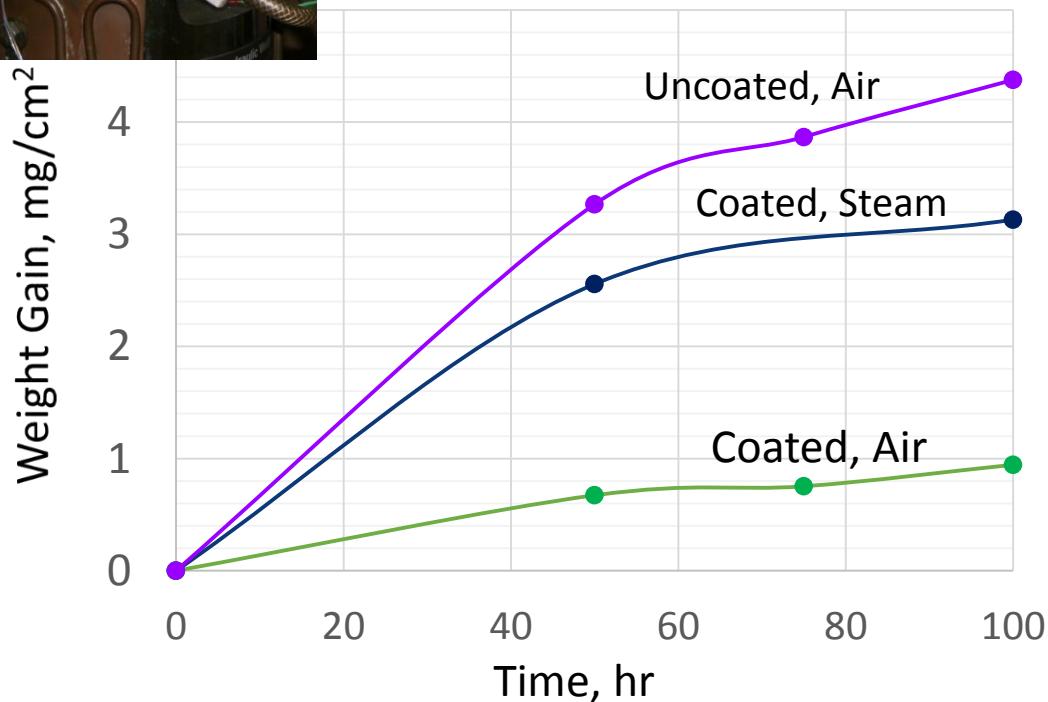


- Sintered at T>2700°F (1482°C)
- Good chemical compatibility between layers
- TGO ~4 μm
- Phase and chemical analysis in progress

**2600F (1427C) in 90% H<sub>2</sub>O, 100h/100 cycles**



# Cyclic Oxidation of Slurry EBC on SiC Heating Element (1 hr at 2600F (1427C) / 20 min at T<100C, 90% H<sub>2</sub>O)

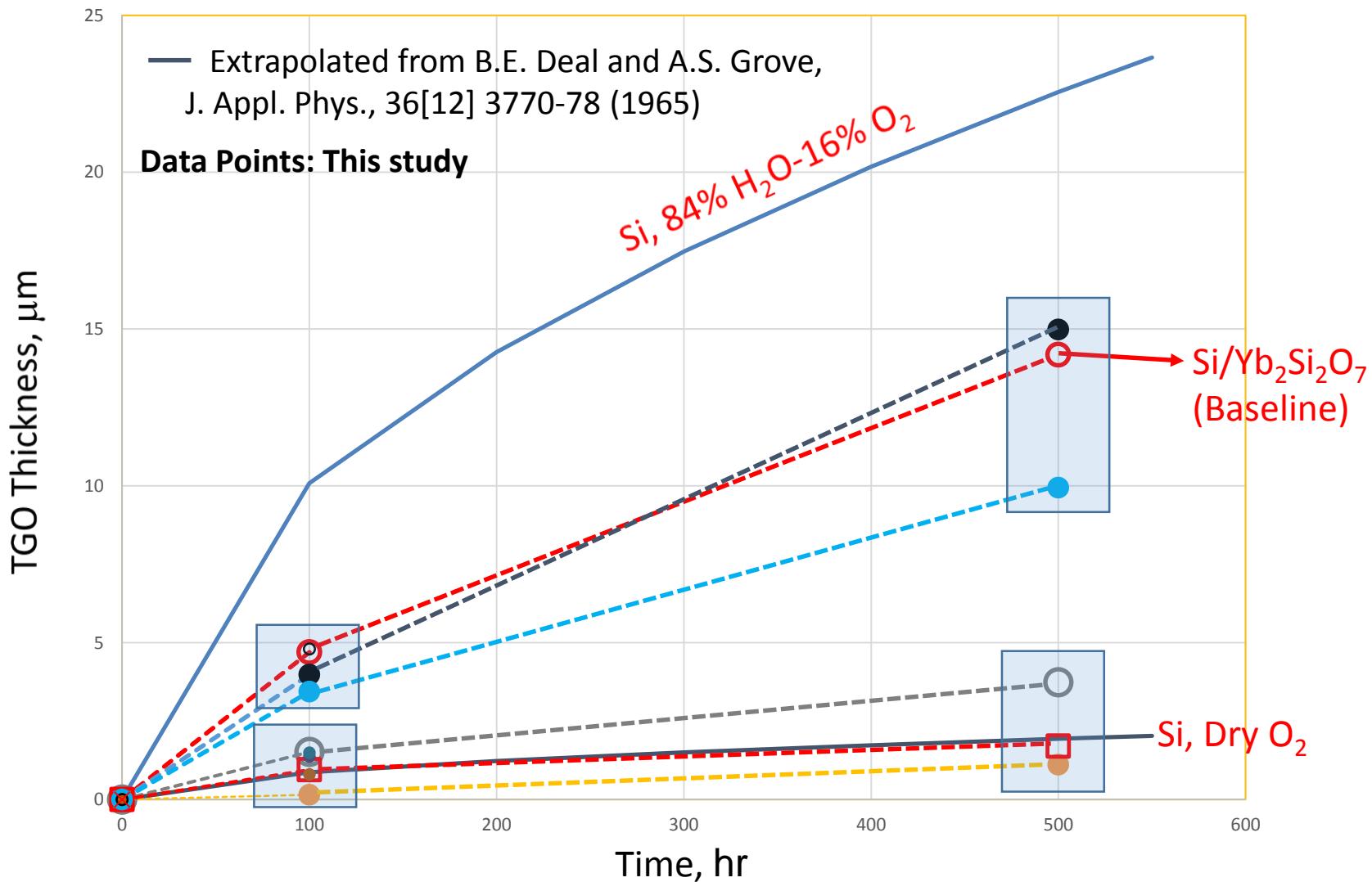


- High oxidation rate of uncoated SiC due to additives and high porosity
- EBC is effective in reducing oxidation rate in air and steam



- EBC Development
  - 2. Plasma-Sprayed Modified  $\text{Yb}_2\text{Si}_2\text{O}_7$  EBCs  
(w/ Si Bond Coat)

# Steam Oxidation of Si/Modified $\text{Yb}_2\text{Si}_2\text{O}_7$ at 2400F (1 hr at 2400F (1316C) / 20 min at T<100C, 90% H<sub>2</sub>O)

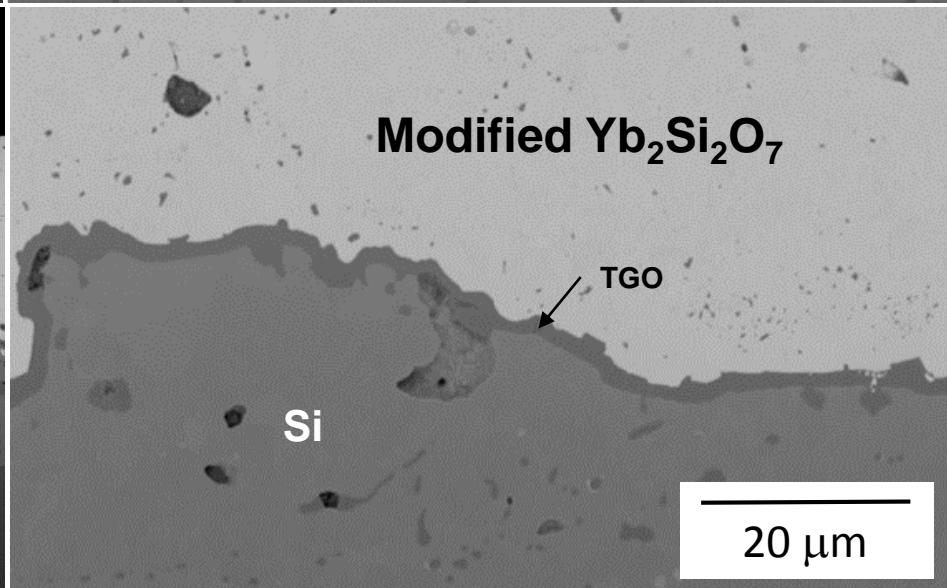
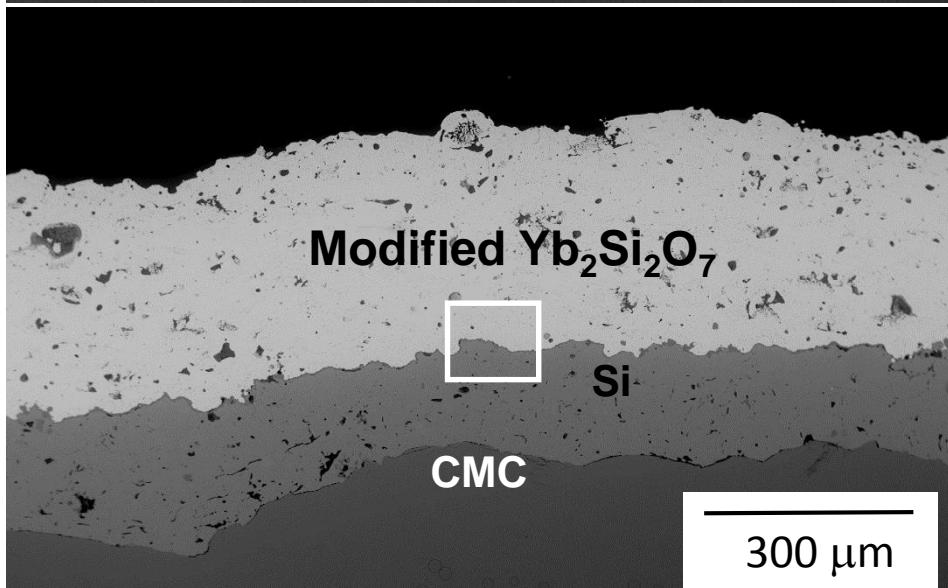
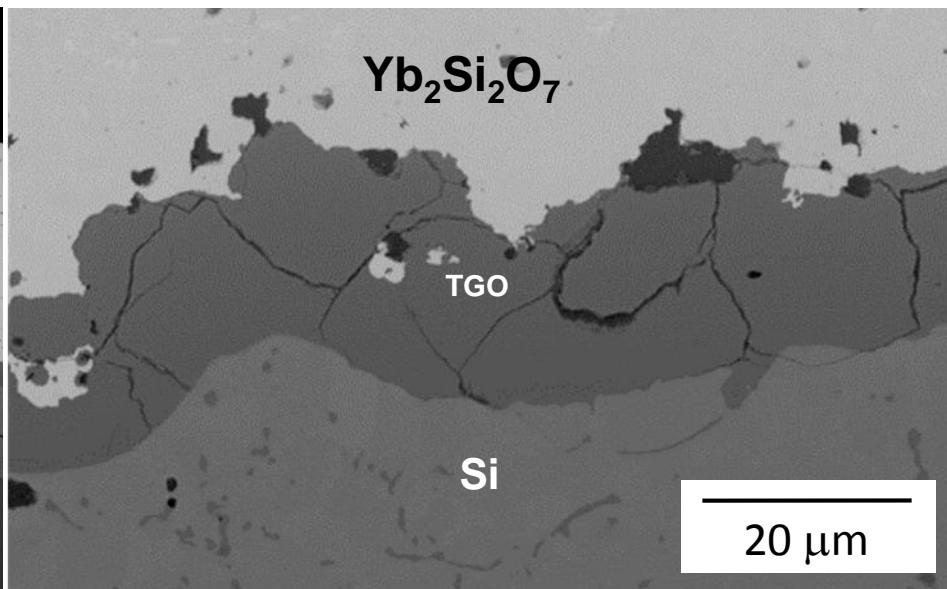
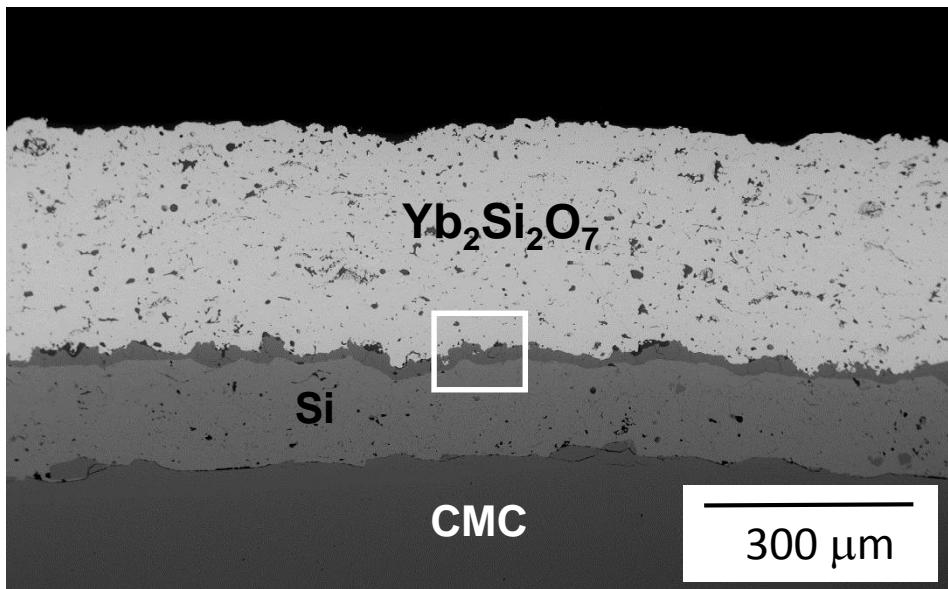


- Some modified  $\text{Yb}_2\text{Si}_2\text{O}_7$  EBCs significantly reduce steam oxidation rate of Si BC
- Optimization and long-term testing (future work)



# Baseline vs. Modified $\text{Yb}_2\text{Si}_2\text{O}_7$

2400F (1316C) in 90% H<sub>2</sub>O, 500h/500 cycles





# Summary

- H<sub>2</sub>O is the predominant oxidant in EBC steam oxidation
- Oxidation-induced failure mechanism appears to be similar to TGO-driven TBC failure mechanism
- Potential for slurry-based 2700F (1427C) bond coat demonstrated
  - Phase/chemical analysis in progress
  - Optimization, long-term test, and CMAS study (future work)
- Modified APS Yb<sub>2</sub>Si<sub>2</sub>O<sub>7</sub> EBCs reduce TGO growth rates on Si bond coat at 2400F (1316C) by two orders of magnitude
  - Very effective in reducing oxidation rates
  - Phase/chemical analysis and CMAS study in progress
  - Optimization and long-term test (future work)



## Acknowledgements

- Dagny Sacksteder (Summer Intern)
  - Help with slurry fabrication and SiC heating elements cycle test
- Bryan Harder (NASA)
  - Helpful discussion on EBC steam oxidation