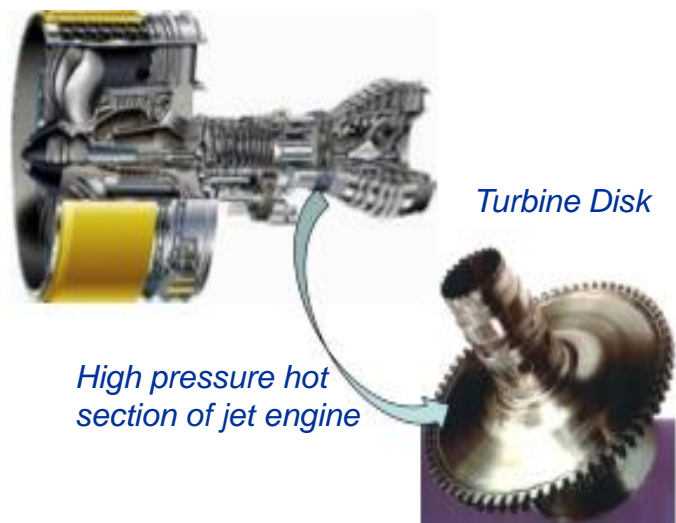




# Location-specific microstructure and the effect of heat treatment on electron-beam melted Ni-based superalloy LSHR



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**Acknowledgements:** ORNL's Manufacturing Demonstration Facility for EBM fabrication. Quintus Technologies for Pressurized Heat Treatment. Peter Bonacuse, Rick Rauser, Dereck Johnson, Rick Rogers, Joy Buehler support of NASA GRC for experimental and analytical support; Holly Sirk and Bethany Cook of Case Western Reserve University for intern support. NASA MSFC's AM facility for SLM trials.

**Funding:** NASA GRC Center Innovation Fund, NASA HEOMD Additive Manufacturing Structural Integrity Initiative, DOE Office of Energy Efficiency & Renewable Energy, Advanced Manufacturing Office, contract DE-AC05-00OR22725 with UT-Battelle, LLC.



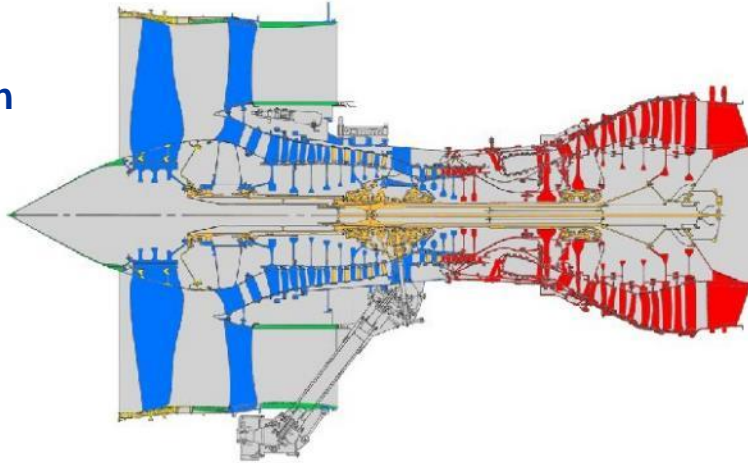
March 13, 2018



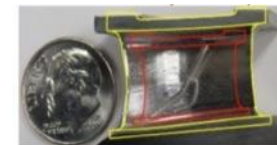
# The $\gamma'$ -strengthened Ni-based superalloys are used in demanding, high-temperature environments in gas turbine engines for aviation

50 % or more of the engine weight is from nickel alloys

Turbofan



Scale down of blades in gas turbines



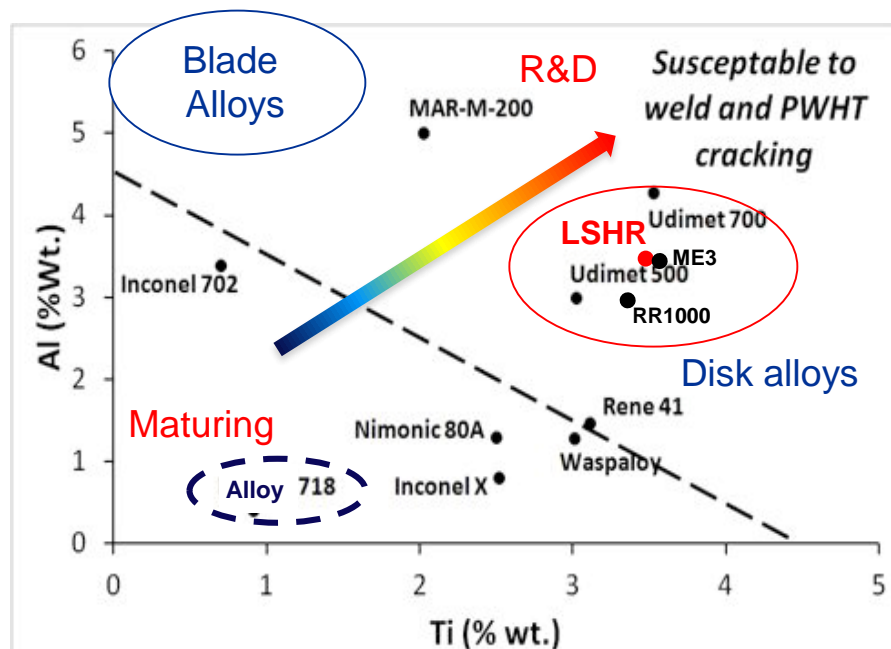
## Other turbomachinery applications:

- Next generation nuclear reactors
- Land based gas power generation
- Turbopumps for rocket engines
- Marine gas engines

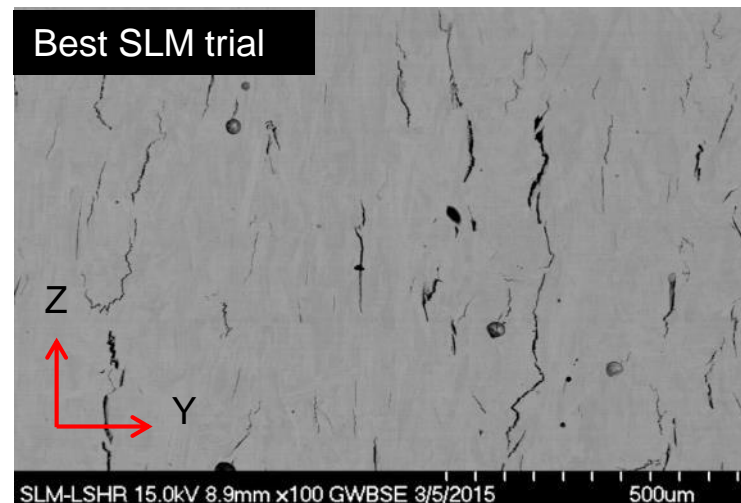
- *Investment casting cost effective for high production (100,000 parts / year)*
- *Target low production to avoid custom made tooling & intricate structures*
- *Target cooling efficiency, intricate net-shape with **precision powder-bed AM***

**Metals AM is rapidly advancing. To move beyond common alloys needs R&D.**  
**Goal:** Use LSHR as a gateway towards fabrication for other Ni-based  $\gamma'$ -superalloys

### Superalloy weldability map



### Cracking observed in our SLM trials



**Selective laser melting (SLM) → Thermal-induced cracking due to steep thermal gradients between melt pools**

**Prior Work:** Established **low porosity & crack-free LSHR** with Electron Beam Melting

- Heated-bed (> 900 °C) leads to less steep gradients between melt pools for EBM

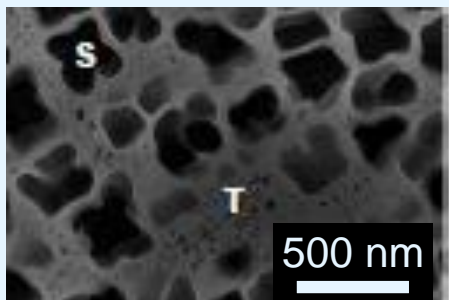
**Aim:** Understand post-processing heat treatment & location-specific microstructure

# Benchmark with state of the art

PM disk application targets strength, creep, and fatigue

Heat treated Low-Solvus High Refractory (LSHR) PM alloy

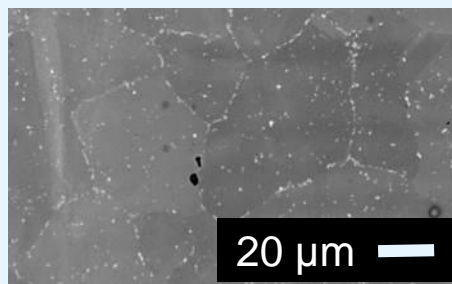
$\gamma'$



**Equiaxed grains**,  $d = 15\text{--}50\ \mu\text{m}$

Cr, Mo, Co-rich  $M_{23}C_6$  carbides  
• ornament GBs

Ti, Ta, Nb-rich MC and  $M_3B_2$   
• interior & GBs



$\gamma'$ -Ni<sub>3</sub>(Al, Ti) precipitates

Volume Fraction:  $\sim 55\%$

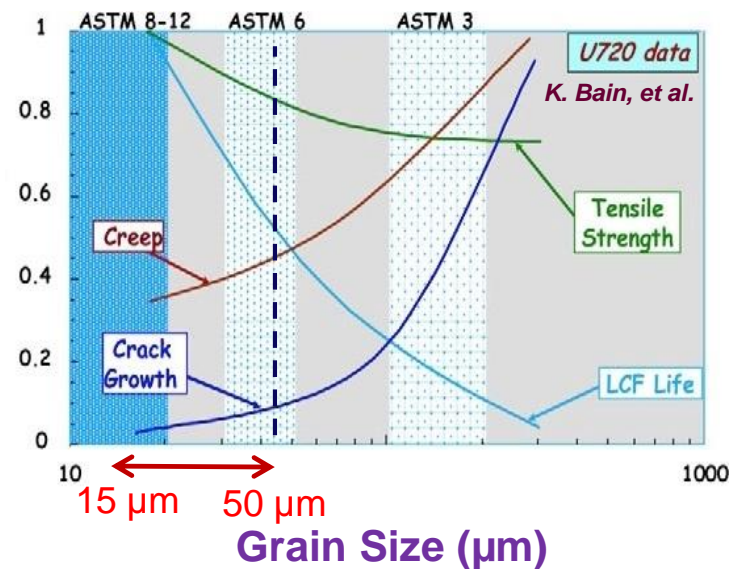
Secondary  $\gamma'$  :  $d = 200\text{--}350\ \text{nm}$

Tertiary  $\gamma'$  ,  $d = 20\text{--}40\ \text{nm}$

Patent US 6974508 B1, 2005, NASA/TM-2005-213645

## PM Mechanical Properties

### Normalized Life/Strength



What are the grain sizes for EBM LSHR?

### Nominal

wt.%	Cr	Co	Al	Ti	Nb	Ta	Mo	W	Zr	C	B	trace	Ni
<b>NASA LSHR</b>	12.5	20.5	3.5	3.5	1.5	1.5	2.7	4.3	0.05	0.05	0.03	Si, Fe, N, O, S	bal

10 wt% refractory elements



# Objectives

- Examine set-to-set variability with 4 build sets with the same layout
- Establish the effect of solutioning heat treatments (HT) on as-fabricated grain structure and microstructure –  $\gamma'$ -solvus = 1156 °C
  - (1) As-fabricated
  - (2) 2 hour subsolvus HT (-15 °C) + conventional 2-step aging
  - (3, 4, 5) 0.5 hour, 2 hours, 4 hours supersolvus HT (+18 °C) + 2 step aging
- Gain insight into location-specific microstructure to correlate later to mechanical properties (tensile, creep, LCF behavior)

## Outline

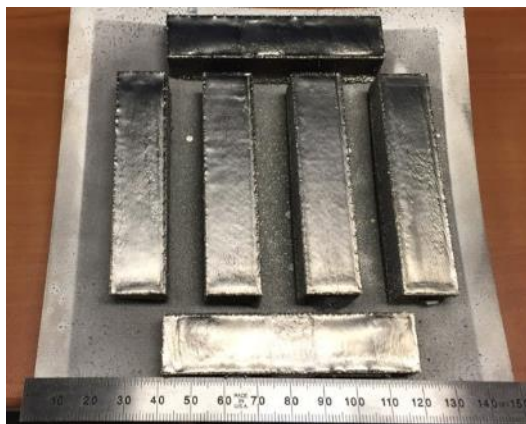
- About the EBM Builds
- Results:
  - Composition
  - As-fabricated grain structure → Deep Dive
  - Solidification observations
  - Minor phases
  - Heat-treatment response
- Concluding remarks and future work

# LSHR EBM Builds at ORNL MDF

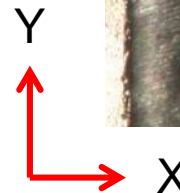
## EBM fabrication at MDF Arcam S12 (modified A2-like)

- Stainless steel substrate
- EBM powder cut: 45-106  $\mu\text{m}$
- Pre-heats/ Re-heats: **900-1100 °C**
- Continuous snaking scan strategy
- **90  $\mu\text{m}$  layer thicknesses**
- $< 5 \times 10^{-3}$  Torr of He

## (1) “Short” In-plane Rectangular Blanks

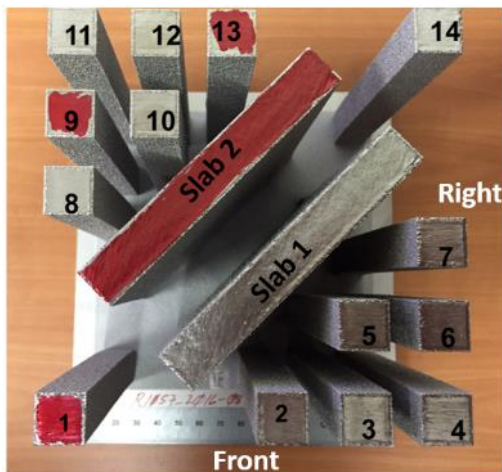


16 mm x 79 mm x 18 mm



CTE differences led to warping of SS plate with several mm deflection

## 4 build sets with this layout



## (2) Four identical build sets with Z-direction Rectangular Blanks & Slabs for In-Plane Blanks

- 14 Z-bars: vertical rectangular blanks in the build direction
- Each slab allows up to 6 in-plane blanks (12 total per set)

CT scans Defect free	Set 1	Set 2	Set 3	Set 4
Z-bars	14/14	14/14	14/14	14/14
In-plane (Slab)	6/12	12/12	12/12	11/12

Z-bar: 15 mm x 15 mm x 95 mm, Slab: 15 mm x 95 mm x 95 mm

Z is the build direction, Y is perpendicular to scan, while X is parallel



# Composition: Excellent Agreement within Build & Build-to-Build

Major Element	Build-to-Build	Upper-to-Lower	Upper: 15 mm beneath top surface					Lower: 15 mm above build plate					Powder (wt.%)	Delta (wt.%)	Major Element
			Set 1	Set 2	Set 3	Set 4	Average	Set 1	Set 2	Set 3	Set 4	Average			
Al	✓	✓	3.50	3.51	3.51	3.52	3.51	3.50	3.50	3.50	3.52	3.51	3.50	✓	Al
Co	✓	✓	20.43	20.43	20.44	20.40	20.42	20.44	20.41	20.42	20.39	20.41	20.56	0.15 ↓	Co
Cr	✓	✓	12.11	12.14	12.20	12.12	12.14	12.16	12.14	12.20	12.07	12.14	12.35	0.21 ↓	Cr
Fe	✓	✓	0.015	0.016	0.016	0.015	0.015	0.015	0.016	0.018	0.016	0.016	0.013	✓	Fe
Mo	✓	✓	2.76	2.75	2.74	2.76	2.75	2.74	2.75	2.73	2.78	2.75	2.74	✓	Mo
Nb	✓	✓	1.53	1.52	1.51	1.52	1.52	1.52	1.52	1.51	1.53	1.52	1.51	✓	Nb
Ni	✓	✓	49.9	50.0	49.9	49.9	49.9	50.0	49.9	50.0	49.9	49.9	49.6	bal	Ni
Ta	✓	✓	1.47	1.47	1.48	1.52	1.48	1.47	1.49	1.48	1.53	1.49	1.59	0.10 ↓	Ta
Ti	✓	✓	3.54	3.53	3.51	3.53	3.52	3.52	3.53	3.51	3.55	3.52	3.48	✓	Ti
W	✓	✓	4.48	4.48	4.45	4.49	4.47	4.46	4.48	4.45	4.50	4.47	4.49	✓	W
Minor Element (Trace)													Minor Element (Trace)		
B	✓	✓	0.050	0.045	0.050	0.050	0.049	0.050	0.050	0.050	0.050	0.050	0.020	0.030 ↑	B
C	✓	✓	0.038	0.038	0.038	0.039	0.038	0.038	0.039	0.039	0.038	0.038	0.040	✓	C
O (ppm)	~	✓	45	89	87	85	76	86	56	101	57	75	110	35 ↓	O (ppm)
Si	~	~	0.030	0.035	0.040	0.040	0.036	0.035	0.050	0.060	0.040	0.046	0.021	0.022 ↑	Si
Zr	✓	✓	0.085	0.085	0.085	0.083	0.084	0.086	0.084	0.088	0.084	0.085	0.040	0.044 ↑	Zr

Agreement   
 Some variability   
 Depletion   
 Enrichment

- Powder chemistry agrees with EBM builds for Al, Fe, Mo, Nb, Ti, W, and C
- Builds are slightly depleted in Co, Cr, Ta but still within specification for LSHR
  - 33% reduction in oxygen content in builds likely due to loss of absorbed oxygen from surfaces of powder particles



# As-fabricated builds have very elongated needle-shaped grains in the build direction

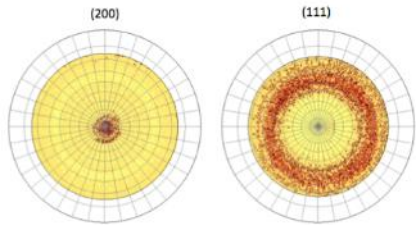
## Short In-Plane Blank

**Zone 1:** Top volume with few passes of e-beam has finer  $\gamma'$ -precipitates (diameter= 90-120 nm)

**Zone 2:** Interior volume with multiple reheats and sustained heat ( $\gamma'$ -cube length= 240-360 nm)

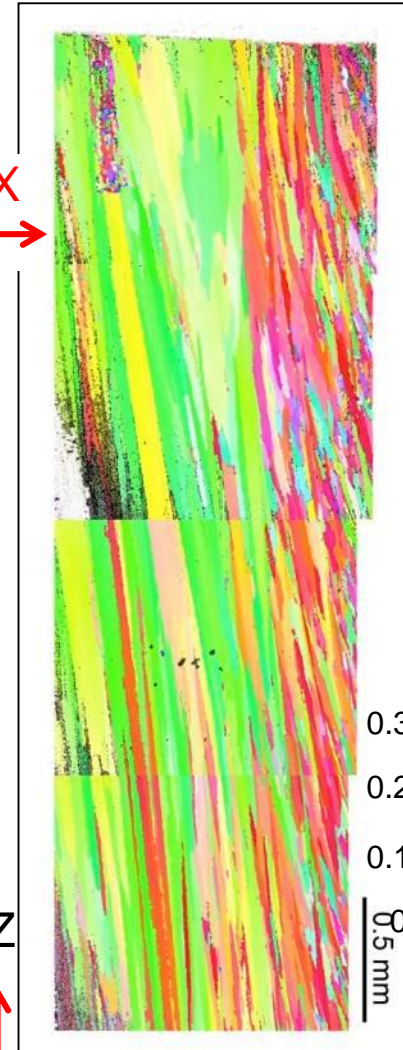
**Zone 3:** Strained volume adjacent to build plate with interdiffusion from plate

### Interior – (001) Texture

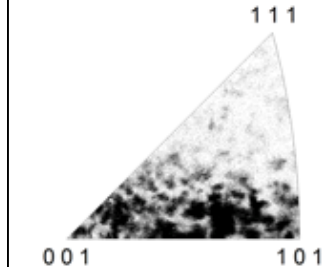


Interior Near Edge

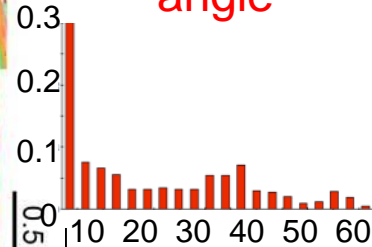
3X  
→



Inverse pole figure



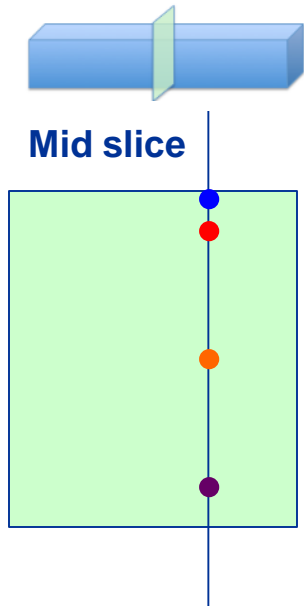
Misorientation angle





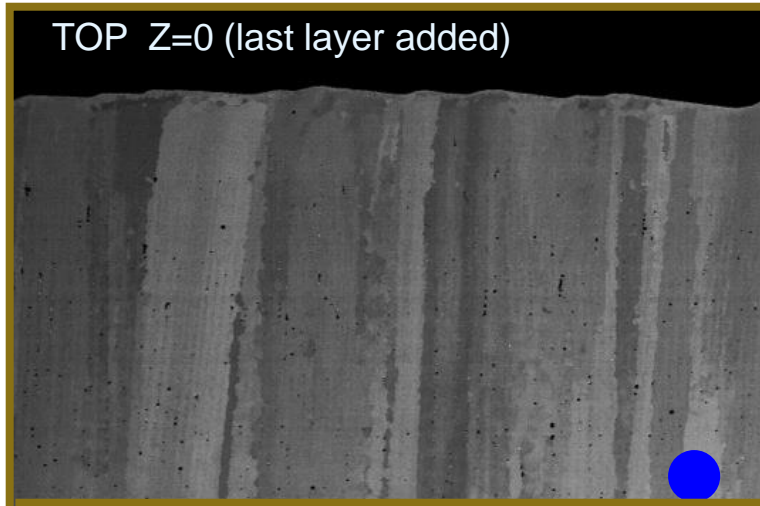
# Hot time, multiple passes, and substrate strain effects

**Short In-Plane Blank** *Grain boundaries are stabilized by  $\gamma'$ -precipitates*

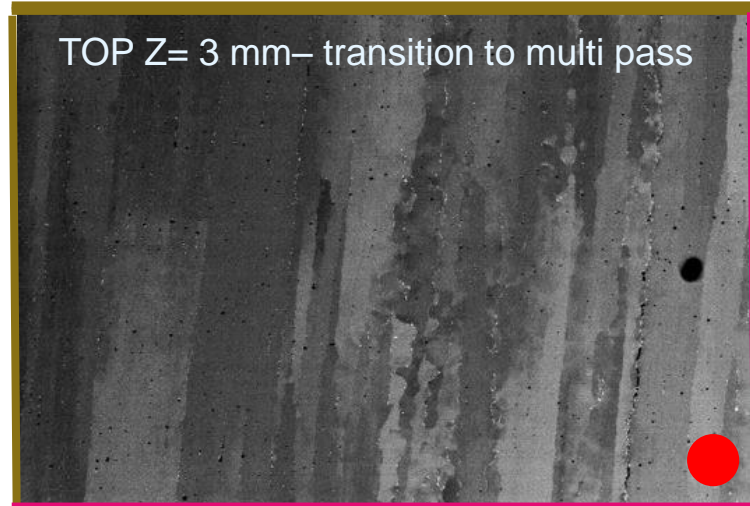


Mid slice

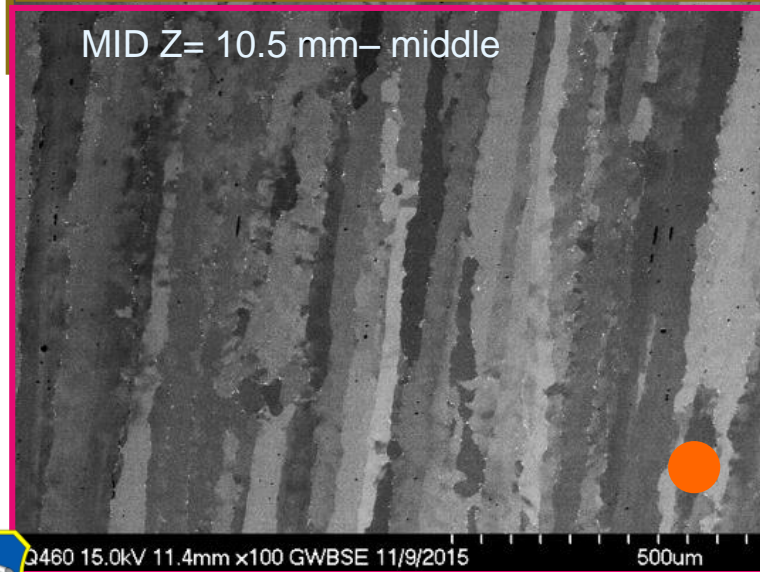
- Top (zone1)
- Top (zone2)
- Mid (zone2)
- Bot (zone3)



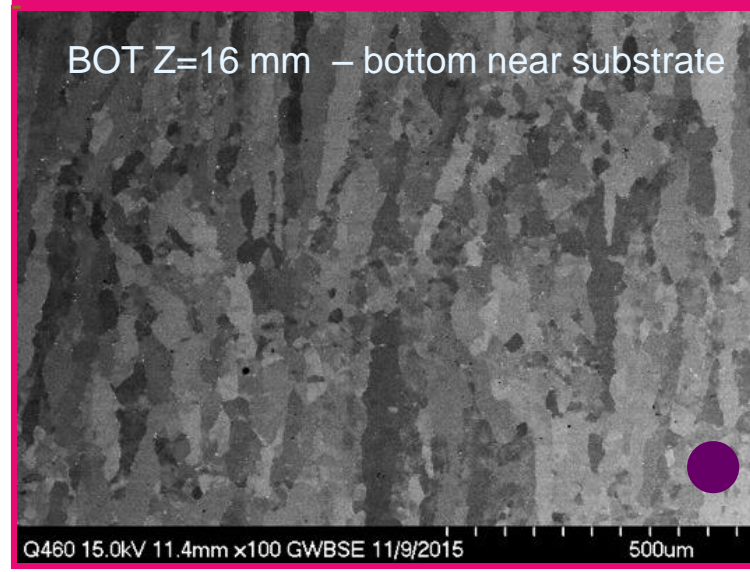
TOP Z=0 (last layer added)



TOP Z= 3 mm– transition to multi pass



MID Z= 10.5 mm– middle



BOT Z=16 mm – bottom near substrate

Z  
↑

Q460 15.0kV 11.4mm x100 GWBSE 11/9/2015

500um

Q460 15.0kV 11.4mm x100 GWBSE 11/9/2015

500um



*Grain boundaries stabilized by minor phases*

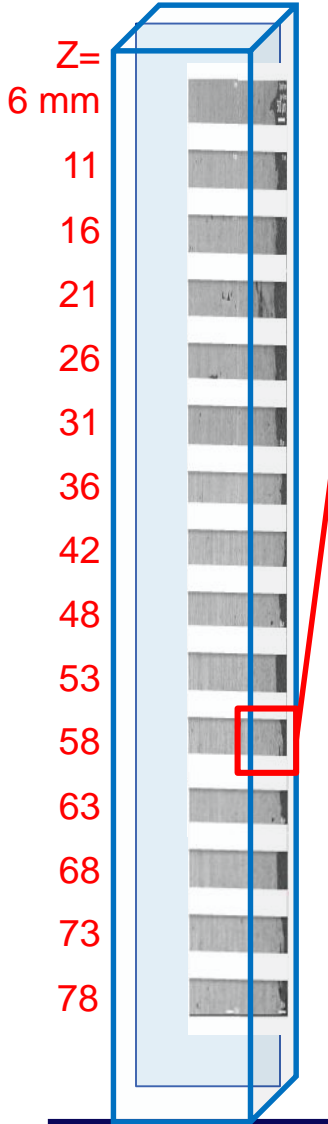
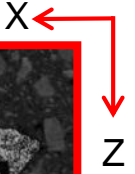


# As-fabricated: Map grain structure in build direction (Zone 2)

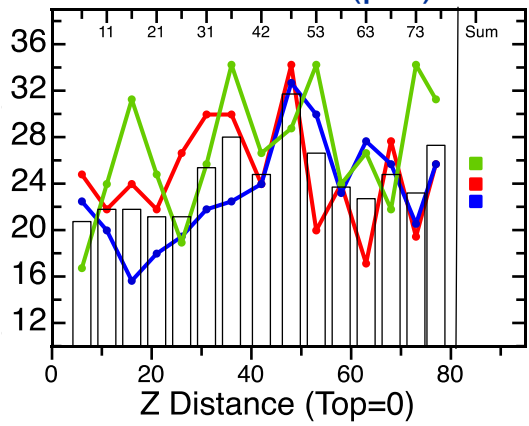
## Vertical Z-bar Blank

15 x 15 x 95 mm<sup>3</sup>

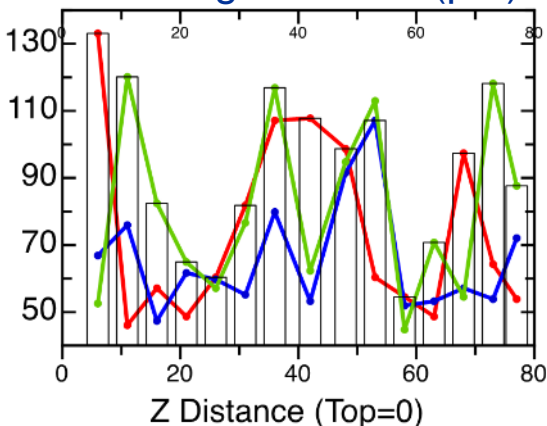
Transverse Grain Widths: Measure linear intercept across in three areas for each Z  
 Zone 2: each layer experiences multiple passes of E-beam



Grain width (μm)



ALA grain width (μm)

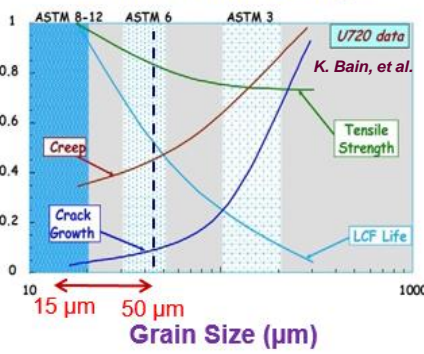


—●— 2 mm    —●— 4 mm    —●— 6 mm from edge    □ Combined 3 areas

**2X-5X larger ALA than the overall average width**

- Overall average AF grain width:  $24 \pm 2 \mu\text{m}$
- For constant Z, grain widths show up to 2X scatter between 2 mm, 4 mm, 6 mm areas yet similar trends
- Arch shape with Z for combined data: Highest values observed for the middle of the bar

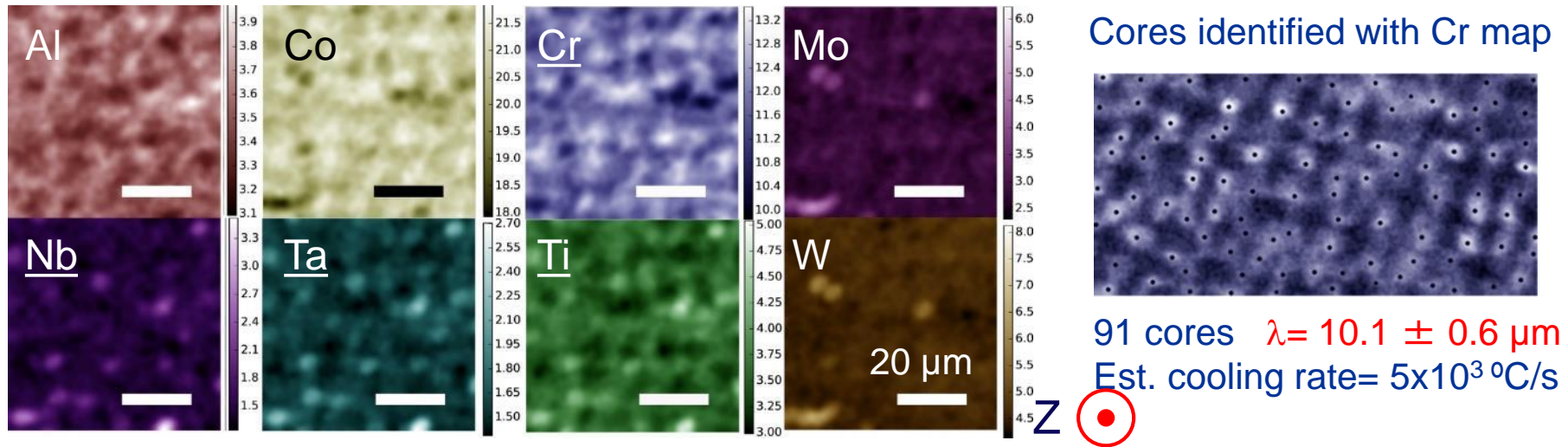
Normalized Life/Strength



Substrate

# Solidification-induced segregation of fine dendrites

## Short In-Plane Blanks: EPMA maps of transverse interior area

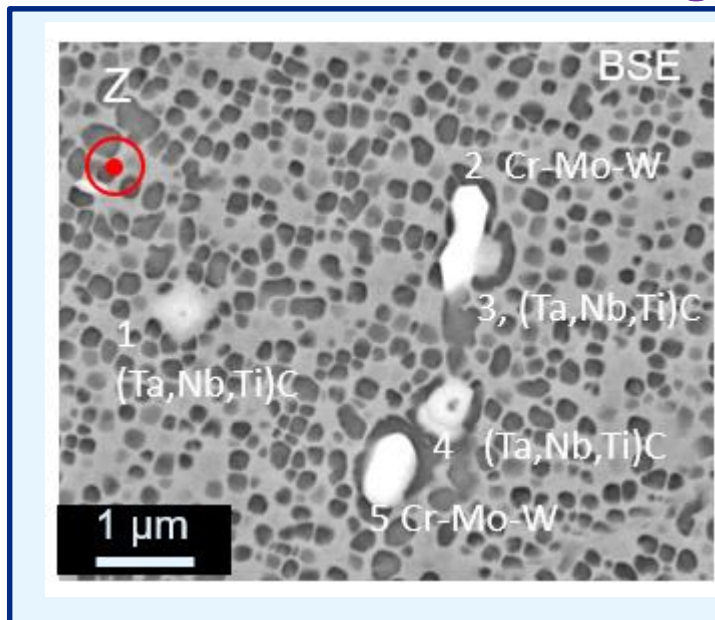


- Finely spaced with primary dendrite arm spacing ( $\lambda$ ) of  $10.1 \pm 0.6 \mu\text{m}$ , which is 20X smaller than what may be observed for a casting
- Dendrite cores: Cr and Co rich, to smaller extent Mo, W
- Interdendritic regions: Al, Nb, Ta, Ti rich
- The compositions of the Regions of interest (ROI) are used to predict ROI  $\gamma'$ -solvus temperature and  $\gamma'$ -volume fraction

JMatPro8-Nidat7 $\gamma'$	$\Delta T_{\text{sol}}$	Vf (%)
Core-blanks	-13 $^\circ\text{C}$	53.2
InterD-blanks	+14 $^\circ\text{C}$	58.7

# Microstructure similar to conventionally processed LSHR

## Heated-bed leads to coarsening of minor phases at GBs



### As-fabricated:

$(Cr,Mo)_{23}C_6$  are predicted to be unstable at 937 °C

### DTA liquation temperatures for as-forged PM LSHR

Stable to high HTs:

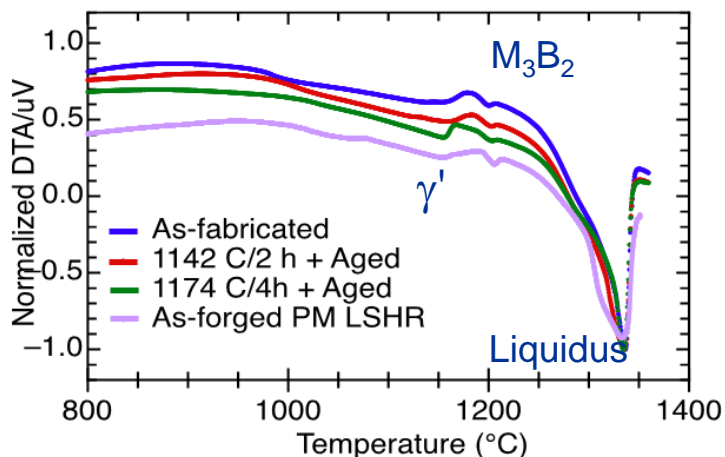
$T_m[(Cr,Mo)_3B_2] = 1202$  °C

$T_m[(Ti,Ta,Nb)C] = 1310$  °C

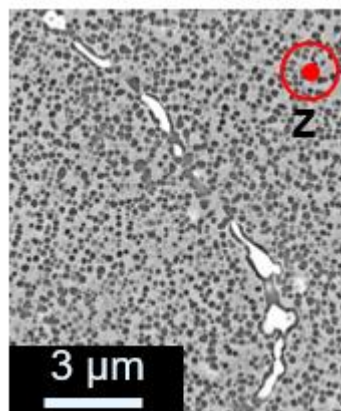
TP Gabb, CK Sudbrack et al.  
NASA/TM-2012-217604 (2012)

## Larger carbides re-resolution during high temperature solutioning

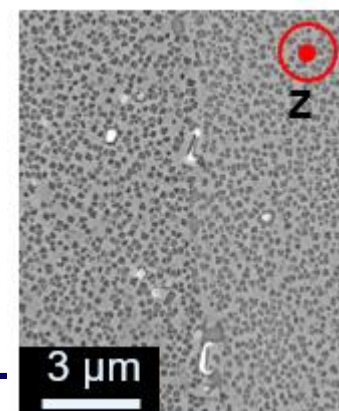
### Differential Thermal Analysis: PM vs. EBM



### As-fabricated



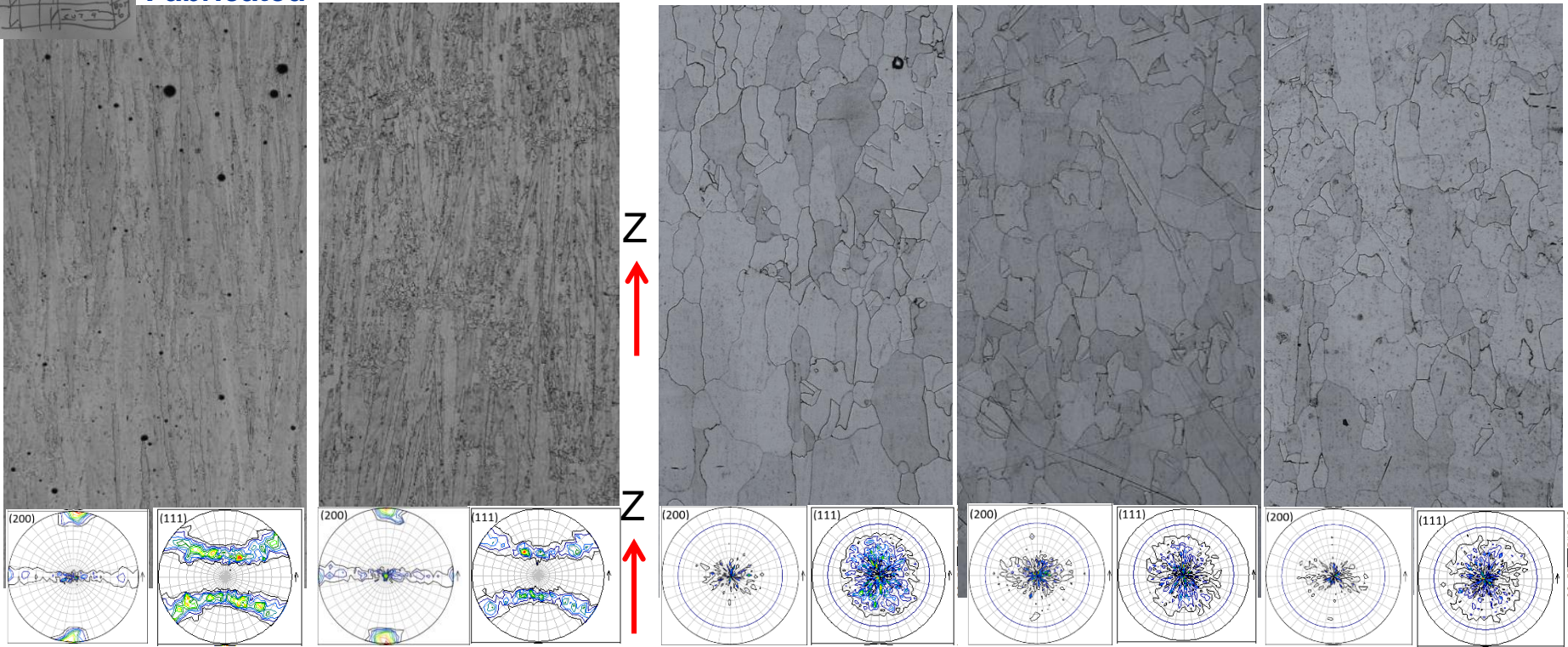
### 2 h Supersolvus HT





# Recrystallization & modest grain growth during supersolvus HT

As-Fabricated      Subsolvus HT (-15°C) for 2h + Aging      Supersolvus HT (+18°C) for 0.5 h + Aging      Supersolvus HT (+18°C) for 2 h + Aging      Supersolvus HT (+18°C) for 4 h + Aging



*Etch micrographs and XRD pole figures similar near edge*

500 μm

## Z-bar – Transverse Grain Traces

	Avg. Feret Diameter	AF	Sub 2 h	Sup 0.5 h	Sup 2 h	Sup 4 h
Upper		21 μm	38 μm	67 μm	79 μm	88 μm
Lower		35 μm	37 μm	85 μm	93 μm	107 μm



# Chemical banding produces bimodal $\gamma'$ -distributions for subsolvus HT

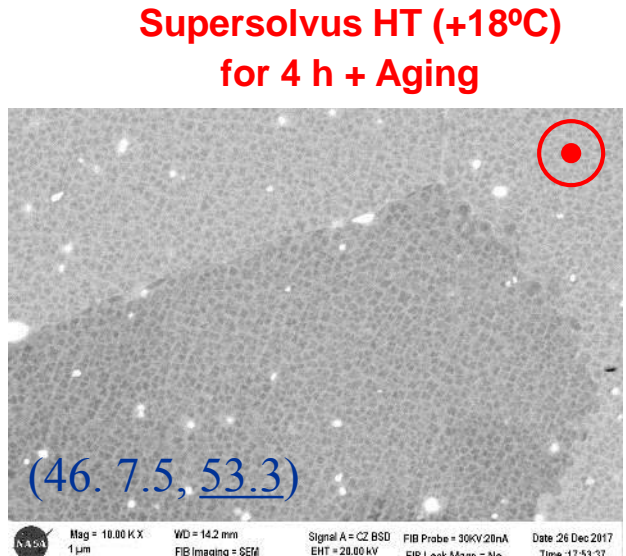
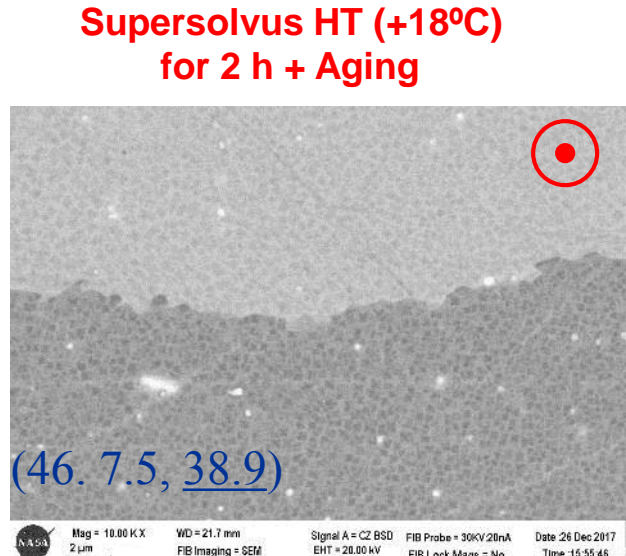
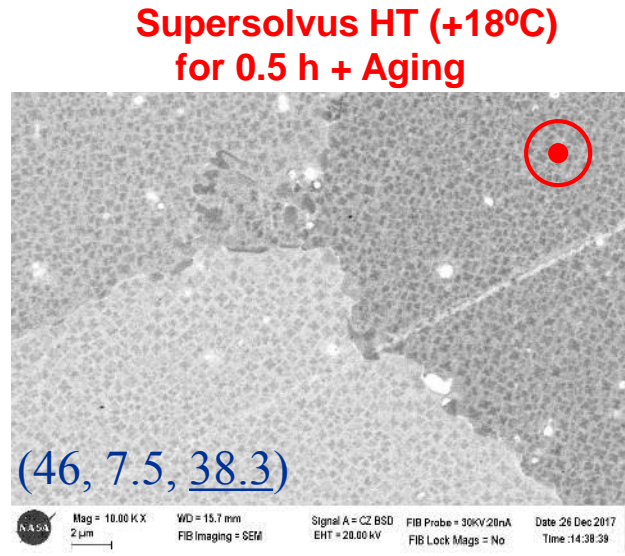
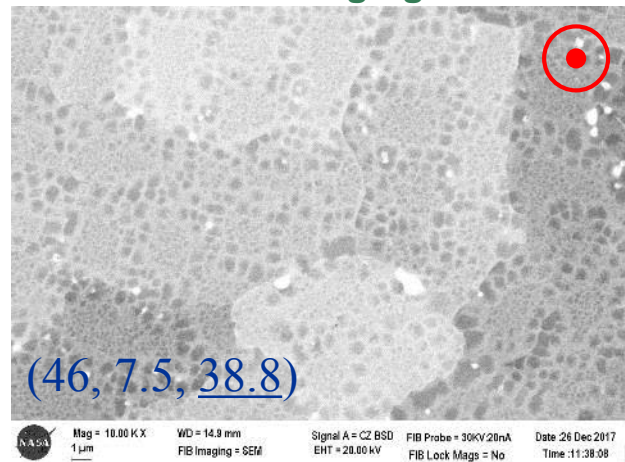
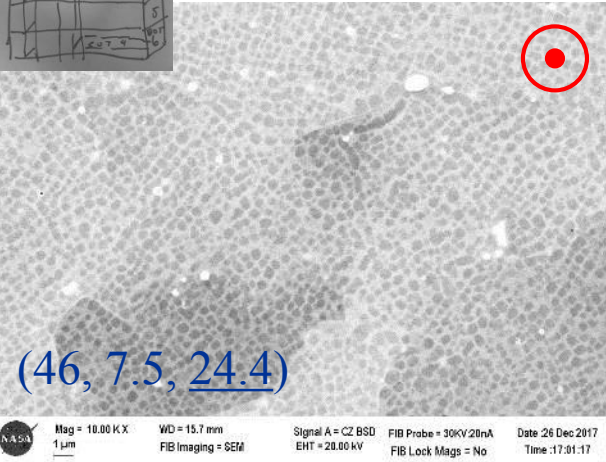


**As-Fabricated**

**Subsolvus HT (-15°C) for 2h + Aging**

- **Subsolvus: Precipitates at the GBs stabilize grains**
- **Supersolvus: Exhibits modest grain growth with increased hold time**

JMatPro8-Nidat7	$\Delta T_{sol}$	Vf (%)
Core-blanks	-13 °C	53.2
InterD-blanks	+14 °C	58.7





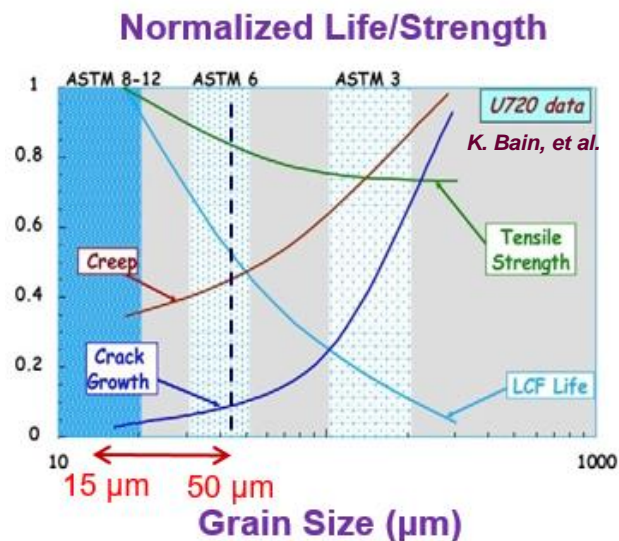
# Concluding remarks

- EBM is promising for  $\gamma/\gamma'$  LSHR, which suggests good potential for EBM of other high  $\gamma'$ -volume fraction Ni-based superalloys
  - Excellent compositional consistency from build-to-build & throughout sample volume
  - Low porosity as-fabricated that is manageable with heat treatment and crack-free material is allowed by optimized processing even for large build elements (slabs)
- Coarser grain structure may necessitate to pivot application use
- EBM LSHR showed similarities to conventional PM LSHR:
  - Extraordinary fast cooling rates that led to fine dendritic structure with no secondary arm growth yet microsegregation
  - Similar microstructural response with carbides, borides and  $\gamma'$ -precipitates
- Subsolvus heat treatment led to undesirable features: bimodal precipitate size distribution and high density of fine grain colonies
- Supersolvus heat treat produced near equiaxed grains with reasonable heat treatment control despite large grain sizes



# Future Work

- Test mechanical behavior for comparisons to conventionally processed LSHR
  - In-plane vs. Z-direction
  - Tensile, creep, and fatigue at elevated temperatures



**Thank you for your attention! Questions?**