



Ion Velocity in the Discharge Channel and Near-Field of the HERMeS Hall Thruster

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Outline

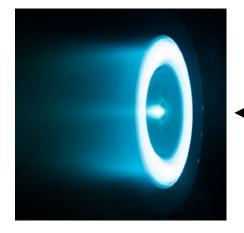


- Introduction
- Principles of LIF
- Experimental Setup
- Data analysis
- Results
 - Near the discharge channel
 - Downstream of pole covers
- Conclusion

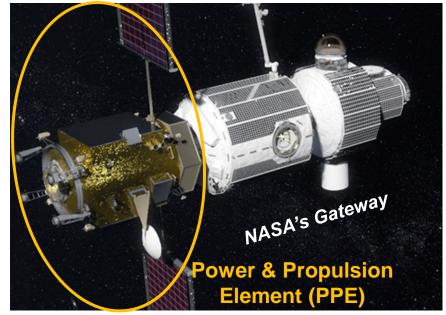
Introduction



- A NASA GRC and JPL team developed a 12.5-kW, magnetically-shielded Hall thruster, called Hall Effect Rocket with Magnetic Shielding (HERMeS)
- Transitioned to commercial production under Aerojet Rocketdyne's Advanced Electric Propulsion System (AEPS)
- Candidate to provide propulsion for the Power and Propulsion Element, the first element of NASA's Gateway
- Continuing risk reduction activities using HERMeS
- Developing a related plasma diagnostics package called Plasma Interaction Sensors for Correlation with Environment Simulations (PISCES)

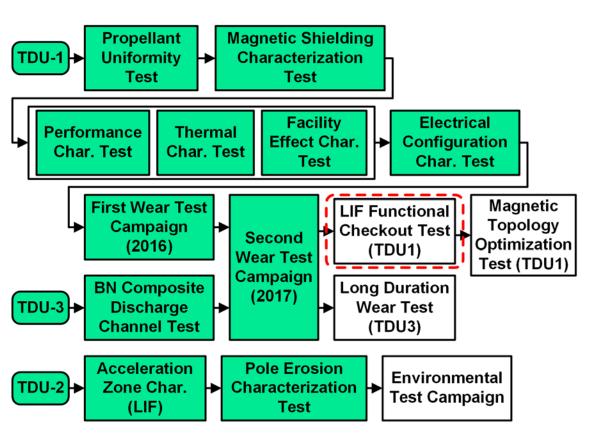


■ HERMeS in operation



HERMeS Test Campaign Status





- Other JPC papers on AEPS and HERMeS
 - Hall, AEPS hollow cathode testing (EP3, Mon 9:30a)
 - Benavides, Thrust vector probe (EP8, Mon 3:30p)
 - Mackey, Uncertainty in thrust stand (EP8, Mon 4:30p)
 - Frieman, TDU long duration wear test (EP10, Tue 9:30a)
 - Lobbia, TDU environmental testing (EP10, Tue 10:00a)
 - Lopez Ortega, Modeling pole erosion (EP10, 10:30a)
 - Lobbia, Accelerated backsputter test (EP10, 11:00a)
 - Kamhawi, Magnetic topology optimization (EP14, 3:30p)
 - Ahern, In-situ wear assessment (EP14, 4:00p)
 - Mikellides, Cathode spot-to-plume mode simulation (EP14, 4:30p)
 - Yeats, 13 kW EP system architecture (EP14, 5:30p)
 - Katz, Accel region electron transport sim (EP17, 9:30a)
 - Choi, 3D electron fluid model for plume (EP17, 10:00a)
 - Lopez Ortega, First principles transport model (EP20, 3:30p)

Why LIF?

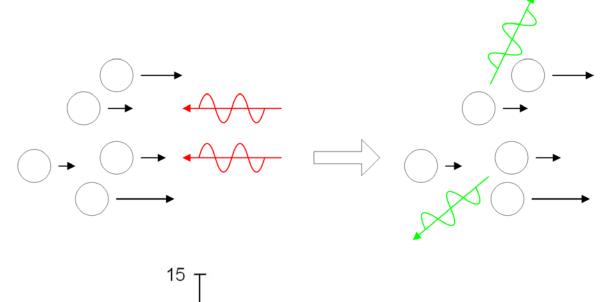


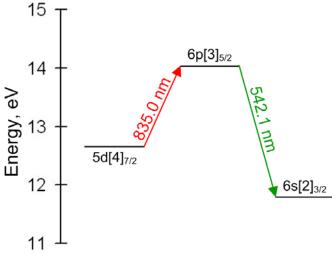
- HERMeS/AEPS project need plasma data from inside the discharge channel for model validation
 - Injected probes (ex: HARP) are too perturbative (Jorns, AIAA-2015-4006)
- LIF can get ion velocity without perturbing plasma, which can be related back to electron mobility
- Concurrently conducting LIF studies at JPL (Chaplin, IEPC 2017-229) and GRC
 - Functional checkout test and get reference TDU data in GRC VF6
 - EDU test in GRC VF5 at lowest achievable background pressure
 - Time resolved LIF at JPL Owens chamber
- Goals
 - Complete data set for model validation
 - Confirmation that EDU and TDU have the same discharge characteristics

How does LIF work?



- Moving atoms absorb light at shifted frequency (Doppler effect)
- Collect emitted fluorescence while varying laser frequency to measure velocity distribution function (VDF)
- XE II 835.0 nm is easy to access with commercial diode laser
 - Metastable
 - Representative of bulk ion VDF
 - Fluoresce in green, 542.1 nm



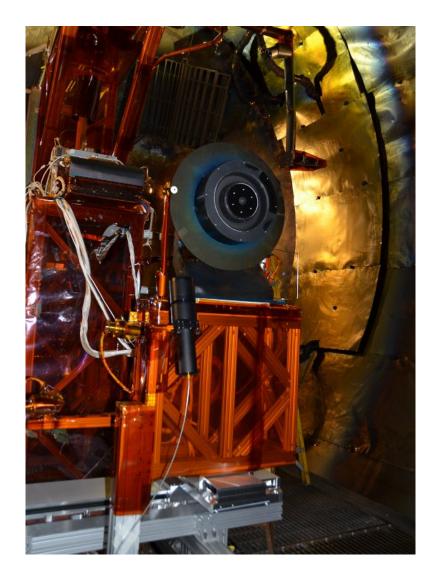


Experimental Setup – Test Article



HERMeS TDU1

- Throttle range from 0.6 to 12.5 kW, 2000 to 3000 sec
- Magnetic shielding topology
- Centrally mounted cathode, 7% cathode flow fraction
- Cathode tied to thruster body
- Test was in VF6, ~1.2e-5 Torr near thruster
- This presentation will focus on:
 - **300**, 400, 500, and 600 V conditions
 - Nominal magnetic field



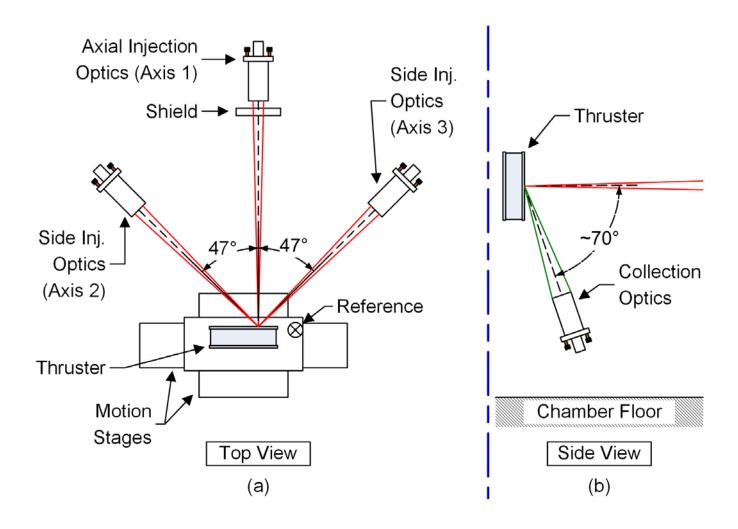
Experimental Setup – Air Side Injection Optics

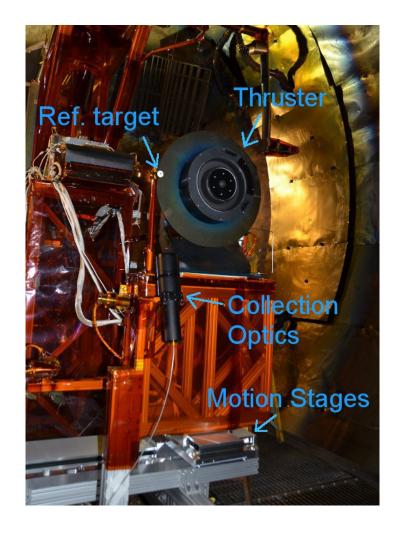




Experimental Setup – Vacuum Side Optics







Experimental Setup – Tower Cooling and In-Situ Alignment









Experimental Setup – Air Side Collection



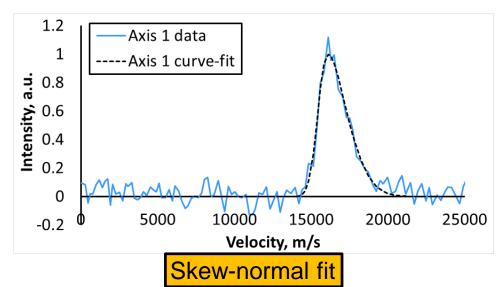
- Collected fluorescence > monochromator > photomultiplier > trans-impedance amplifier > lock-in amplifier > computer data
- Stationary reference signal > lock-in amplifier > computer data
- Computer
 - Control thruster motion stages
 - Control optics alignment motors
 - Read wavemeter
 - Read laser power monitor
 - Read lock-in amplifier outputs

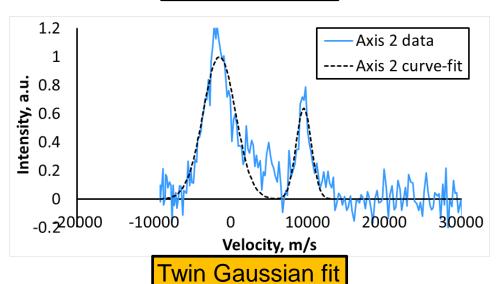


Data Analysis



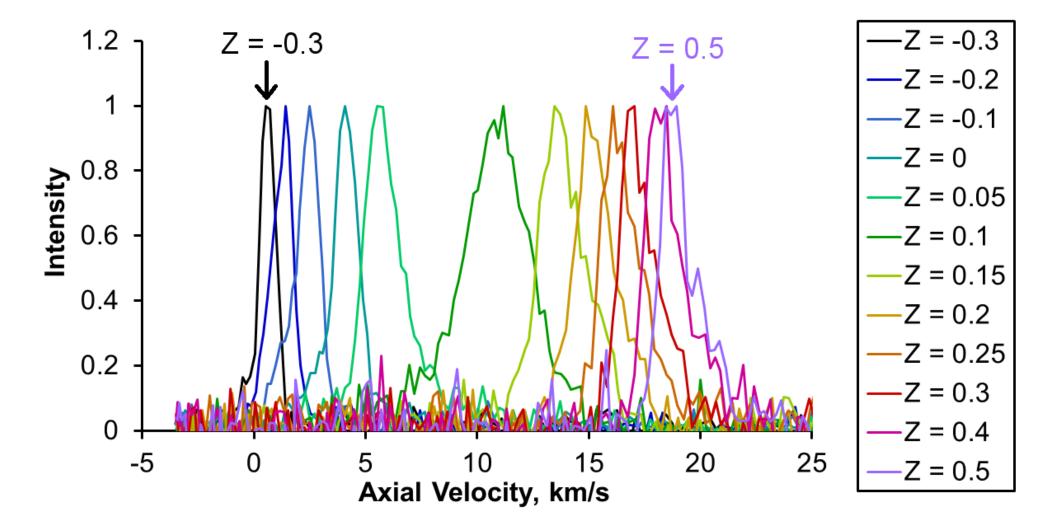
- Saturation study was performed, broadening no more than 10% on narrowest VDFs
- Hyperfine structure and natural broadening small compared to the VDFs
- Zeeman effect uncorrected and will be treated in future analysis
- Data analysis steps:
 - Convert wavemeter and OG signal to velocity
 - Correct intensity by laser power variation
 - Apply curve-fits (Gaussian, skew-normal, twin Gaussian)
- Spatial uncertainty: 0.5 mm
- Velocity uncertainty: ±100 m/s typical (±600 m/s for noisiest scans)





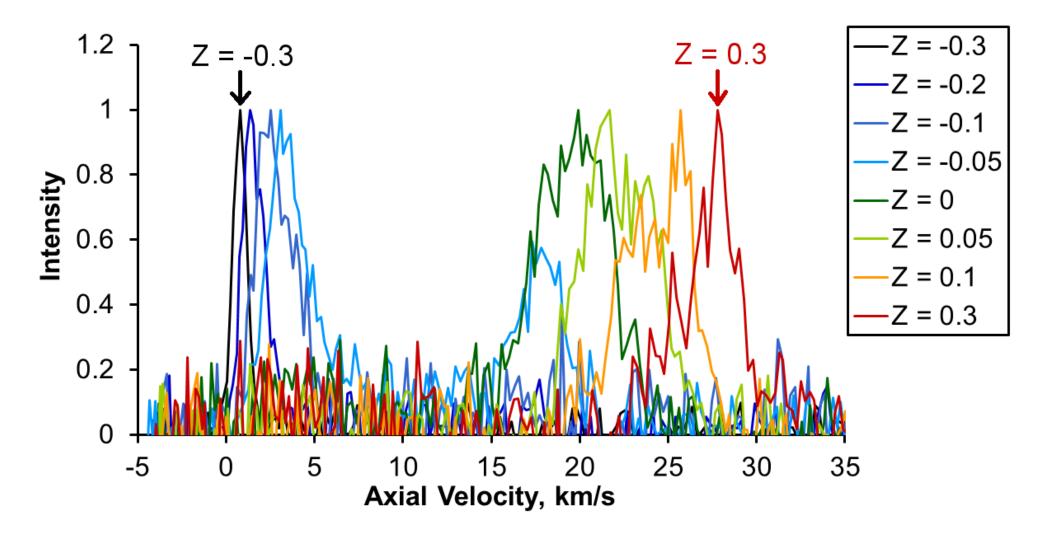
Channel Centerline VDFs: 300 V, 6.3 kW





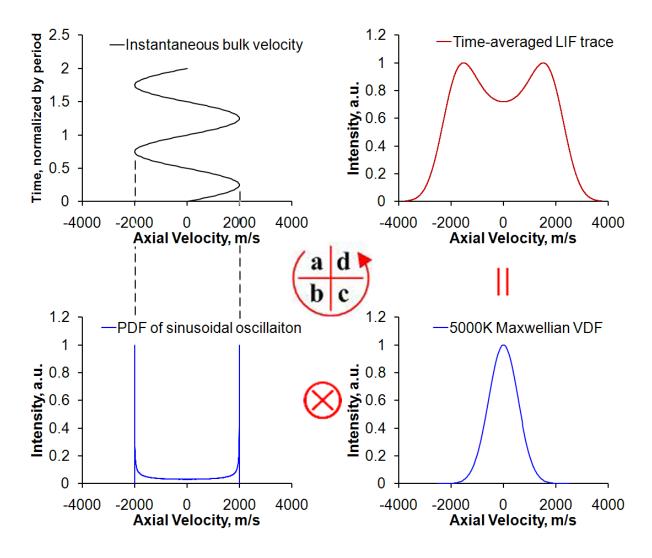
Channel Centerline VDFs: 600 V, 12.5 kW





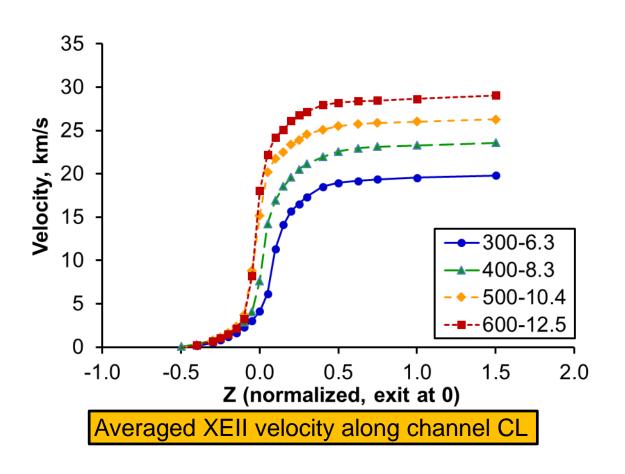
Why sinusoidal spatial oscillation appears as twin peak structure in time-averaged LIF

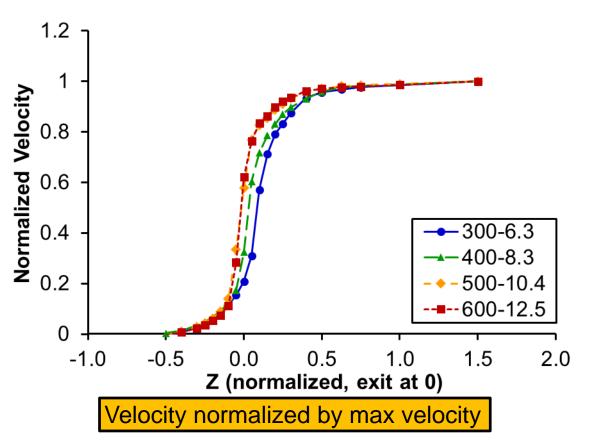




Channel Centerline Velocity Profiles

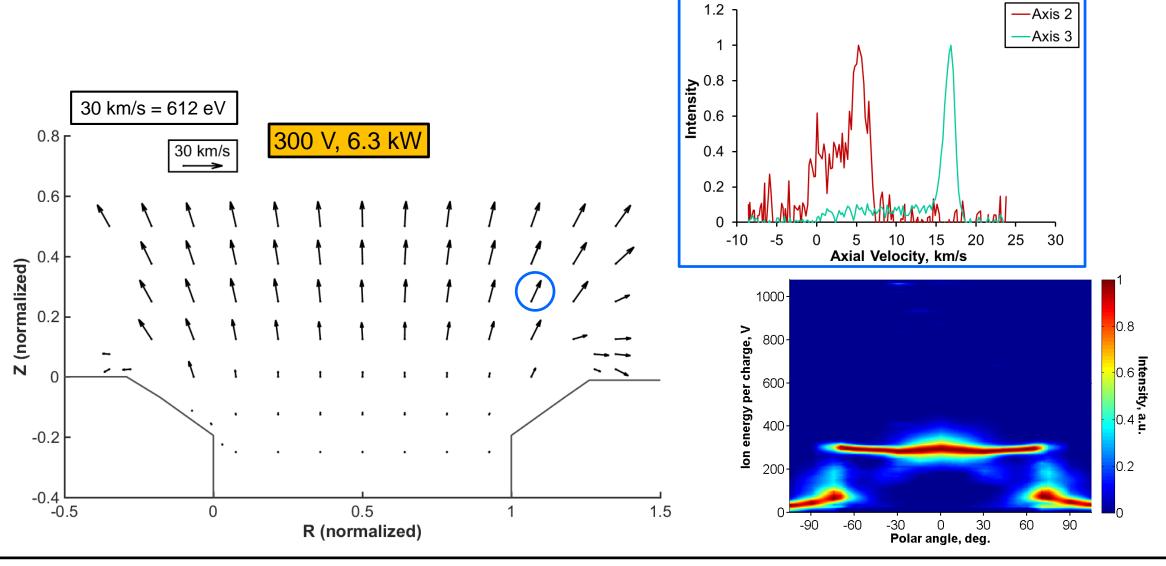






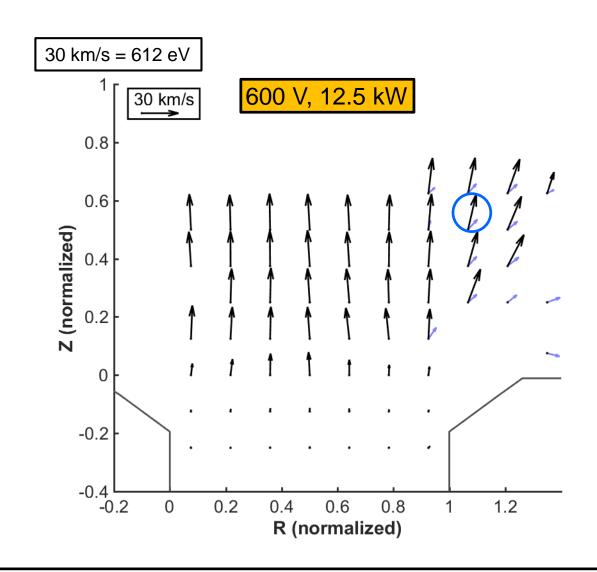
Discharge Channel Ion Velocity Vector: 300 V, 6.3 kW

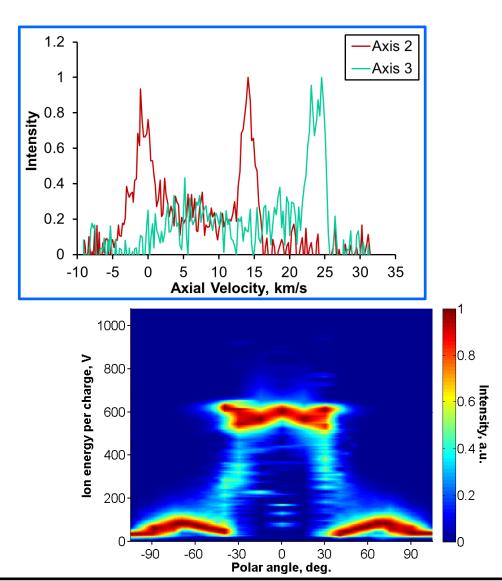




Discharge Channel Ion Velocity Vector: 600 V, 12.5 kW

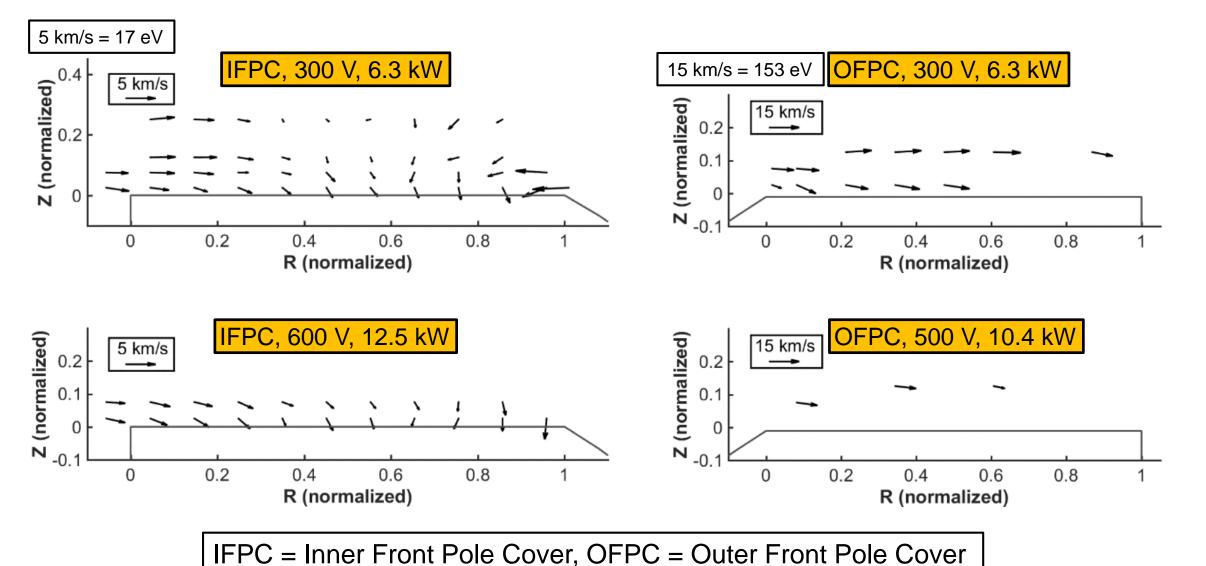






Pole Cover Ion Velocity Vector





Preliminary Results for Energy of Ions Bombarding Pole Covers



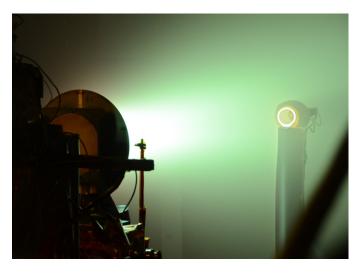
Operating condition	Average ion energy, IFPC, eV	FWHM energy, IFPC eV*	Average ion energy, OFPC, eV	FWHM energy, OFPC, eV*
300-6.3	0 to 20	25 to 72	81 to 119	33 to 91
400-8.3	3 to 7	19 to 74	77 to 99	97 to 145
500-10.4	2 to 5	26 to 46	75 to 77	102 to 155
600-12.5	2 to 15	20 to 48	Low signal	

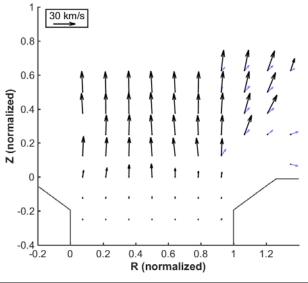
^{*}Full-width-at-half-maximum value of the ion energy distribution. FWHM energy near IFPC were artificially broadened by Zeeman effect.

Conclusion



- New LIF capability for characterizing highpower EP devices at GRC
 - Compatible with engineering hardware
- Completed functional checkout and collected TDU data
- Presence of low-energy population near discharge channel, likely to be CEX ions
 - Energy and direction of high-energy and lowenergy ions in excellent agreement with far-field RPA data
- lons near IFPC have low average energy while ions near OFPC have high average energy; pole ions have large spread in energy





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