

Considerations for Development of a Total Organic Carbon Analyzer for Exploration Missions

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Project Background

- TOCA is water monitoring technology critical to health and safety monitoring for regen water.
- There are gaps in the State-of-Art ISS TOCA versus exploration mission architectures/requirement, e.g., size, mass, consumables, sampling
- ECLSS community agrees an exploration-class TOCA is needed – development effort added to water monitoring roadmap.
- AES miniTOCA Project intends to advance the technology readiness of an exploration-forward TOCA system through:
 - Phase 0 - Technology Feasibility
 - Phase 1 – Ground Demonstration Prototype
 - Phase 2 – Flight Technology Demonstration (*if valued*)



TOCA History: How did we get here?

PCWQM – SS Freedom Process Control Water Quality Monitor

ISS TOCA1 – crit 3, for Russian and stored water analysis

- Shuttle DTO, 1999
- ISS operation: 2001-2002
- Project cancelled during post-Columbia return-to-flight

ISS TOCA2 – crit 1SR, required with U.S. Segment Regen ECLSS

- Development phase: 2005 – 2007
- Certification phase: 2008
- ISS operation: 2008-present

TOCA for exploration missions

- FY17 trade study started on mini-TOCA
- FY18 trade study, technology evaluation and testing



Mission Concept

The exploration mission concept is largely undefined.

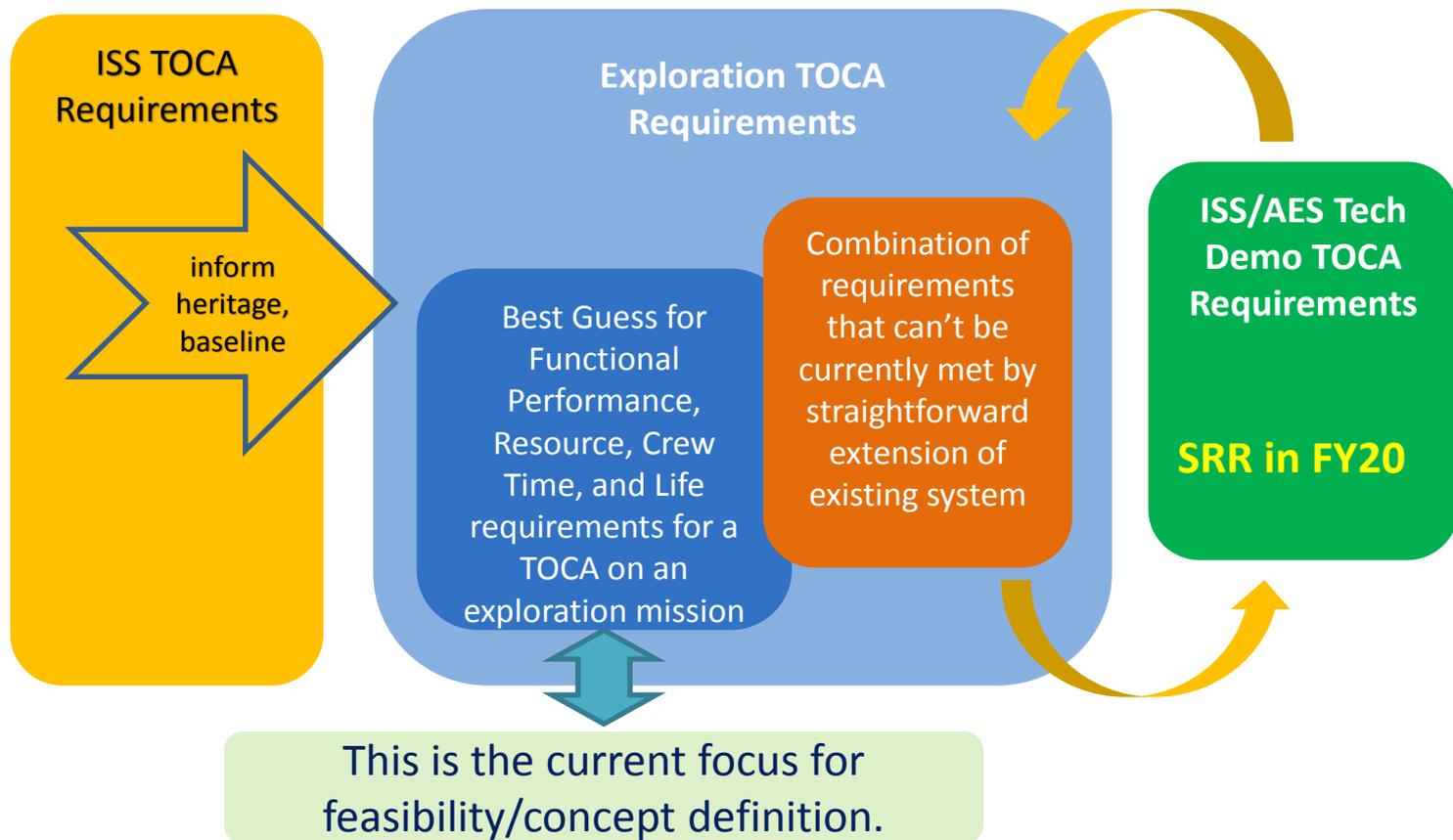
- Mars transit
- Lunar surface
- Orbital outpost

Commonality is that they are NOT low-earth orbit.

- Premium on low launch mass
- Infrequent resupply capability
 - drives high reliability, low maintenance, long life



Deriving Driving Requirements for an Exploration-Class TOCA



Exploration TOCA Design Goals – Functional Performance

| Title | ISS Requirement | New TOCA Goal | Source/Rationale |
|-----------------------------|-----------------|--|--|
| Accuracy | +/- 25% | +/- 25% | ISS precedent |
| Precision | +/- 25% | +/- 10% | Provide reliable trending of data. |
| ★ Range [see next slide] | 1 – 25 mg/L TOC | 1 - 10 mg/L TOC Challenge: 0.25 - 10mg/L TOC | Exceeds detection of potability limit (5 mg/L) |



TOCA History:

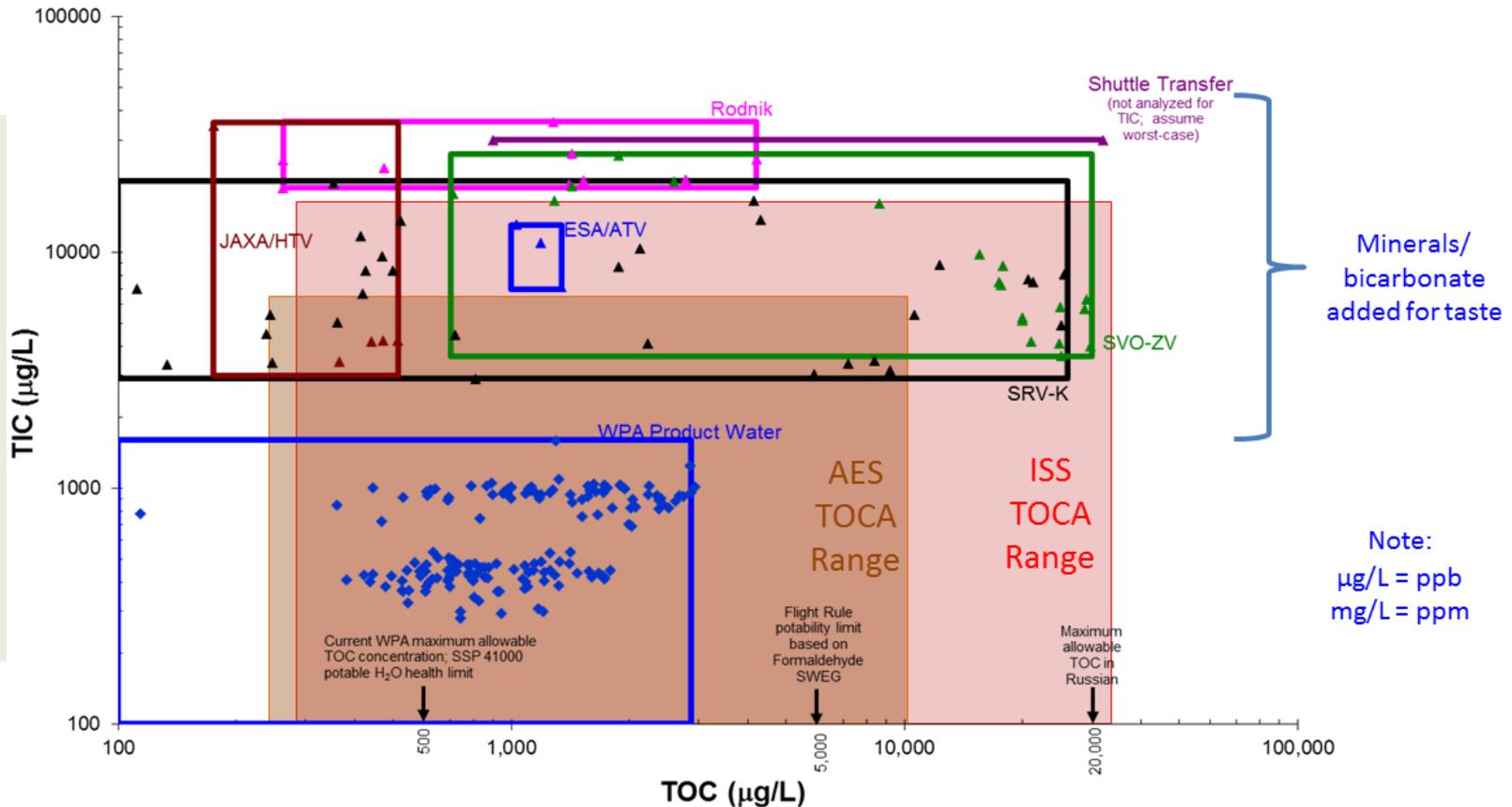
Total inorganic /organic carbon (TIC/TOC) measurements from multiple water sources

Water Sources:

- WPA¹
- Shuttle²
- SRV-K²
- SVO-ZV²
- JAXA/HTV²
- ESA/ATV²
- Rodnik²

¹based on ISS TOCA results

²based on ground testing



Exploration TOCA Design Goals – Functional Performance

- Water Sample Composition:

| Title | ISS Requirement | New TOCA Goal | Source/Rationale |
|----------------|------------------|-----------------|---|
| ★ TIC content | up to 15mg/L TIC | up to 5mg/L TIC | No minerals added in regen water. Equilibrium of CO ₂ in air @ 2mmHg to water= \sim 4.5ppm _{CO₂} . |
| ★ Conductivity | N/A | <10 μ S/cm | Water processor specification Not an HSIR or medical requirement |
| ★ pH | N/A | pH 4.5 – 9 | HSIR, MPCV70024, section 3.2.2.1 Note: TOCA may eliminate acidification if sample pH is <8 with no buffering capacity. |
| Free gas | 5% | 0.1% | NASA-STD-3001 (although HSIR states 5%) |

Note: Water composition may interfere with the analysis using certain technologies.



Exploration TOCA Design Goals – Resource Allocation

| | Title | ISS Requirement | New TOCA Goal | Source/Rationale |
|---|------------------------------|-------------------------------|------------------------------|---|
| ★ | Volume | 4160 in ³ (actual) | < 1200 in ³ | Notional reduction. Balancing achievable with return on development investment. |
| | Device Weight | 80 lbs (actual) | < 25 lbs | Notional reduction. Balancing achievable with return on development investment. |
| ★ | System Weight per 5-year ops | N/A | <35lbs | Includes device plus all consumables and unreclaimed water consumption |
| | Power Consumption | < 175 W avg. < 225 W peak | < 175 W avg. < 225 W peak | ISS precedent. AES vehicles will have reduced power availability than ISS. |
| | Sample Size | < 150mL | < 150mL | Any water returned to the water balance does not count against the requirement. |
| | Supply Gases | N2 | N2 or O2 are acceptable | Compressed N2 and O2 can be utilized if favorable to overall design trades. Supplied H2 is not available. |



Exploration TOCA Design Goals – Crew Time Allocation

| Title | ISS Requirement | New TOCA Goal | Source/Rationale |
|---|----------------------|---|--|
| Analysis Time | 190 mins | 12 hours GOAL: less than 190 mins | No hard requirement. GOAL = less than ISS TOCA |
|  Analysis Frequency / yr. | N/A | 60 analyses / year | Allows > weekly analyses per ops concept |
| Crew Time for Analysis | < 15 mins / analysis | < 15 mins / analysis | No more crew time than ISS TOCA. Goal for inline, automated sampling.  |
| Crew Time for Maintenance | < 8 hrs / year | < 8 hrs / year | Includes consumable replacements and calibration. |

 Automated sampling should be traded with size, complexity. The current assumption is that automation reduces size due to large size of crewed interfaces.

Forward Work: Combine total resource and crew time into equivalent system mass calculation for future evaluation.



Exploration TOCA Design Goals – Life

| | Title | ISS Requirement | New TOCA Goal | Source/Rationale |
|---|----------------------|--------------------------------|-------------------------------|--|
| ★ | TOCA Lifetime | 5 years with maintenance | 10 years with maintenance | Match the entire life of TBD habitat to eliminate resupply costs. (Device life = ground assembly/certification + mission life) |
| | TOCA Cycle Life | 1200 analyses with maintenance | 600 analyses with maintenance | 60 samples/year x 10 years |
| ★ | Component Shelf Life | 1 year | > 3 years | assumes a minimum resupply frequency for maintenance components every 2 years. |



Feasibility Studies

Potential sources for exploration TOCA technology include:

- Current ISS TOCA
- Original Crit3 ISS TOCA
- Commercial TOC analyzers (i.e. modify and fly)
- SBIR spaceflight TOC analyzer prototypes
- Future new development options



Why not use the ISS TOCA?

Size of existing TOCA cannot be reduced dramatically unless the electrochemical oxidizer is redesigned due to flow rates required

Advantages

- Proven environmental compatibility, range, accuracy, reliability, safety.
- Detection range: 0 – 25ppm TOC;
- Accuracy: +/- 25%; Reliability: >4.5 years of operation on ISS
- Maintenance: 1st maintenance and calibration occurred after 3.5 years (238 samples)

Disadvantages

- 80 lbs, 22 x 16 x 12 inches
- Not packaged for “in-line” potable water monitoring.
- Requires resupply and consumable replacement: manual waste water bag replacement every 6 samples; acidic buffer container replacement every 7 month or 46 samples.



ISS TOCA2
Developed 2005-2008
1st unit ops 2009-2013
2nd unit ops 2013-current



TOCA1 technology is also a possibility...

Designed by Sievers with Wyle/NASA

H x W x D, in. 8.9 x 19.3 x 16.3

Weight, lb. 54.1

Max. power, W 69.3 avg.; 93.2 peak

Criticality 3

Range: 0-25ppm TOC

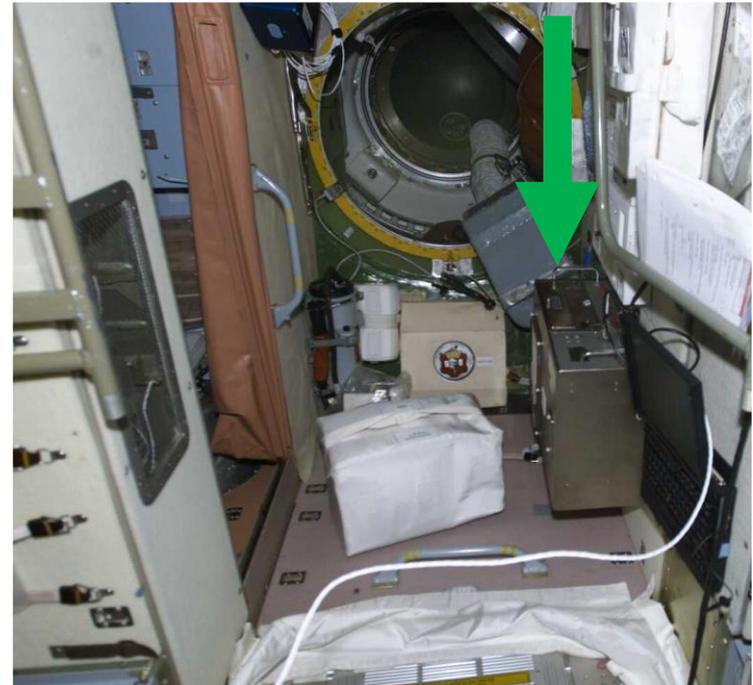
Mission duration spec.:

RME, days 90+ required/365 design

ISS, mos. 12

ISS, analyses 50 required/85 design

TOCA1 in use on ISS October 2001



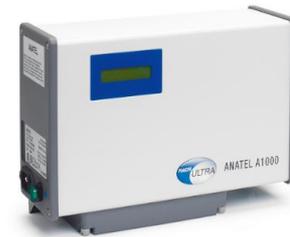
1. **Size is still too large**
2. **1 year life due to persulfate**
3. **Tox 2 hazardous chemicals**



Are COTS TOC Analyzers an Option?

COTS TOC analyzers generally have at least one of the problems below

- Hazardous chemicals are used for acidification and oxidation.
- Reagentless analyzers employ mercury UV lamps.
- Reagentless analyzers are only compatible with purified water sources.
- Infrared CO₂ detection requires gas-liquid separation typically employing gravity-dependent sparging.
- Conductivity-based TOC measurement requires an ultra pure water source.
- Additional CO₂ separation membrane is required for conductivity detection to eliminate of non-carbon conductivity interferences.
- Non-spaceflight reliability and safety.



Hach
A1000



Millipore
A10 TOC
monitor



OI /Xylem Aurora
UV Combustion or
UV Persulfate /
NDIR



Sievers
M9



Can we utilize previous SBIR development?

- SBIR awards have produced two prototype TOC analyzers that were developed with goals for small size, no hazardous reagents, and microgravity compatibility.
- Prototypes were delivered to NASA in 2007 and 2011.
- NASA priorities at that time did not warrant continued funding to Phase III.

- Lessons learned and design solutions from the previous SBIR prototypes may be useful and are currently being investigated.



Assessment of ISS and Current Commercial Technologies: Summary

- Proposed AES TOCA requirements **cannot** be met directly by
 - ISS TOCA2
 - ISS TOCA1
 - Commercial TOC analyzers
 - SBIR prototype deliverables

- SBIR components will require additional evaluation and testing.

- Custom commercial work is costly and appears likely to lead to a “one off” device for an ISS demo, not a development of knowledge and capabilities for a sustainable flight project.

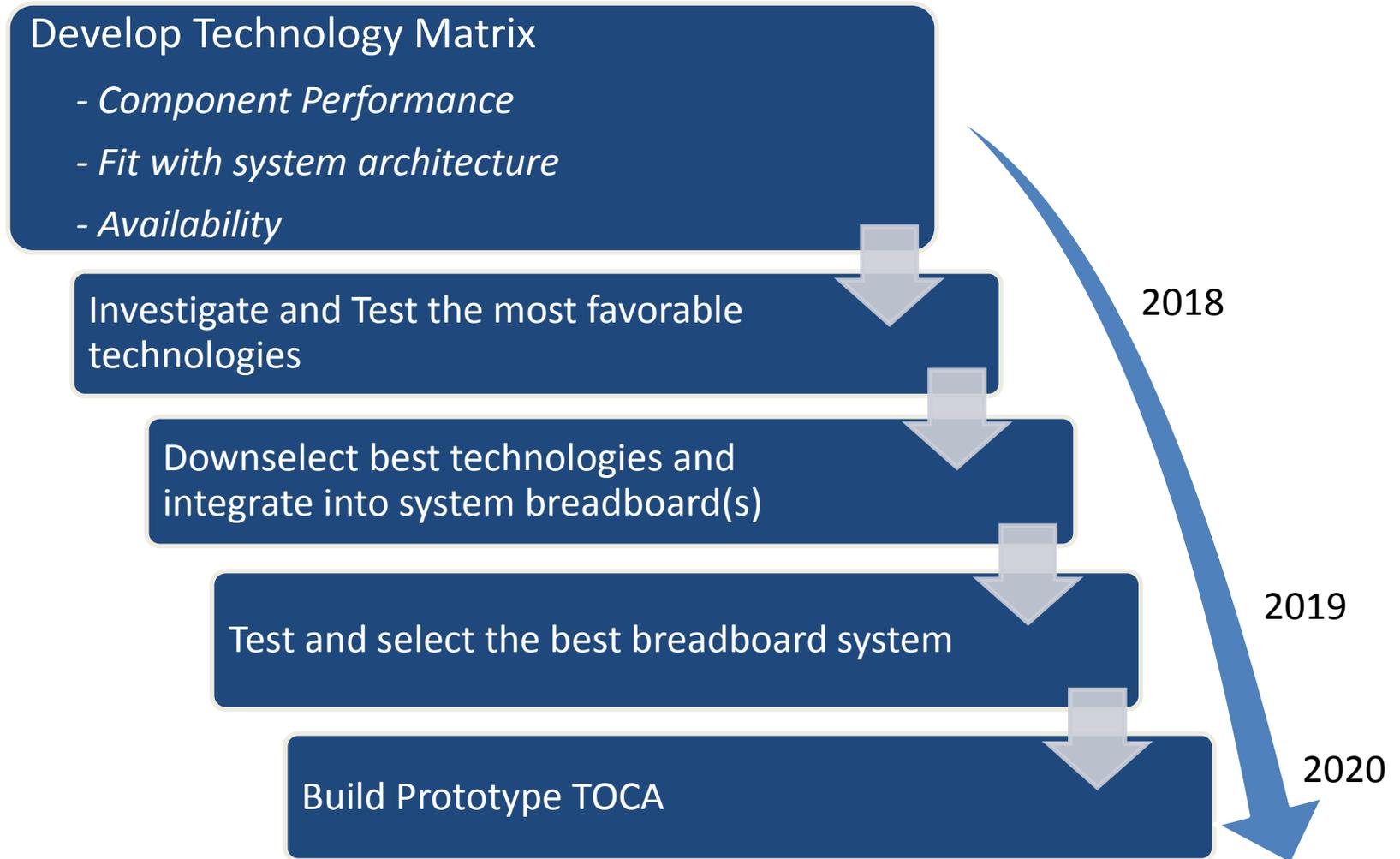
- Many subsystem/component technologies within the list above are attractive.

- The current focus is to evaluate available technologies for selection based on performance and cost.



Project Plan

Technology Maturation



Technology Matrix Highlights

| Acidification | Oxidation | CO2 Detection |
|--------------------------------------|-------------------------------------|---|
| Electrolytic anion removal | Mercury UV | Membrane conductivity (aq) |
| Electrolytic protonation | Excimer UV (172nm) | Raman spectroscopy (aq) |
| Electrochemical generation from salt | LED UV / catalyst | Laser Spectroscopy (aq) (absorbance or acoustic) |
| Chemical reagent | Combustion (catalytic 450-850C) | NDIR (gas) |
| | Boron-doped diamond electrochemical | Laser Spectroscopy (gas) |
| | Ozone | Methanizer/Flame Ionization (gas) |
| | Chemical reagent | UV/Vis spectroscopy (aq) |
| | | Pulsed discharge detector (gas) |
| | | Thermal conductivity (gas) |



Technology Architectures: ED / UV / MC

Acidification

Electrolytic acidification

Oxidation

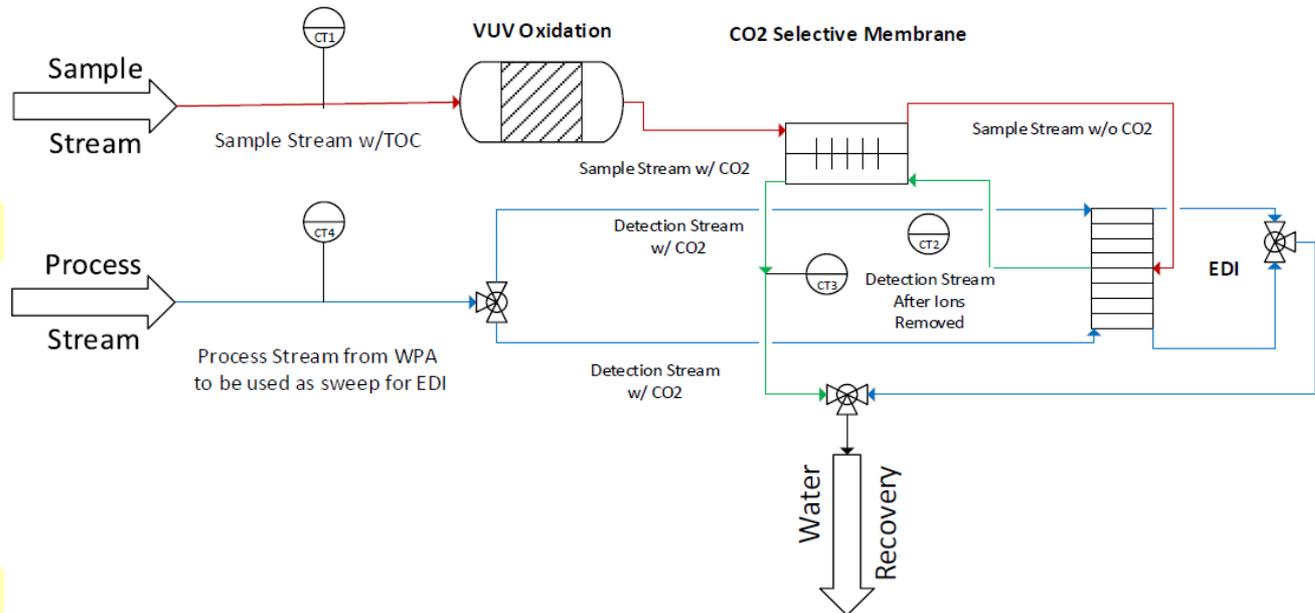
Excimer VUV photo-oxidation

CO₂ Detection

Memb sep / conductivity

Advantages:

- Architecture based on proven techniques performed by Sievers (now Suez) – UV oxidation with membrane/conductivity detection.
- Electrodeionization cell is added to eliminate consumable and life limited acidic reagent. Development needed.
- Excimer UV is proposed instead of mercury-based UV. Excimer lamp was utilized in the TOCA1 along with persulfate.



Technology Architectures: UV/Raman

Acidification

Not required

Oxidation

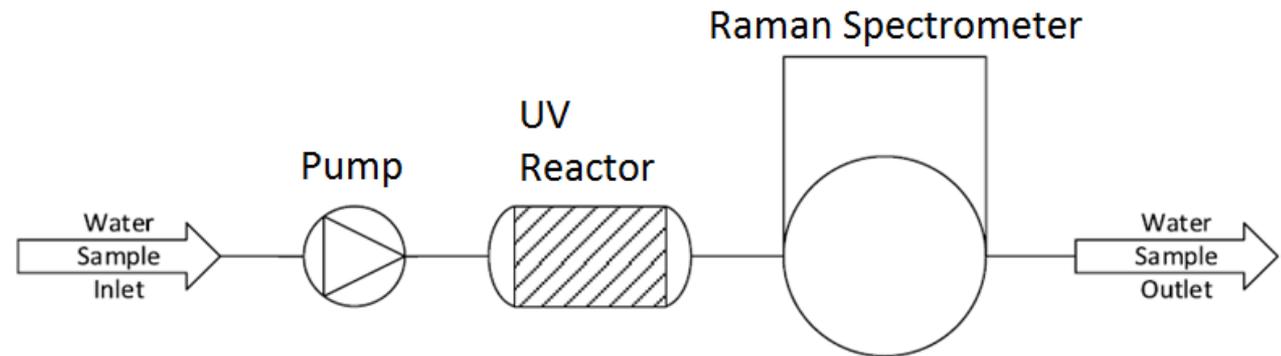
Excimer VUV
photo-oxidation

CO₂
Detection

Raman
spectroscopy

Advantages:

- Raman has capability for direct measurement of CO₂ and carbonate species in the aqueous phase which could eliminate need for acidification and membrane separation
- Raman needs development to reach low-level sensitivity.



Technology Architectures: Combustion / NDIR or Combustion / TLS

Acidification

Not required

Oxidation

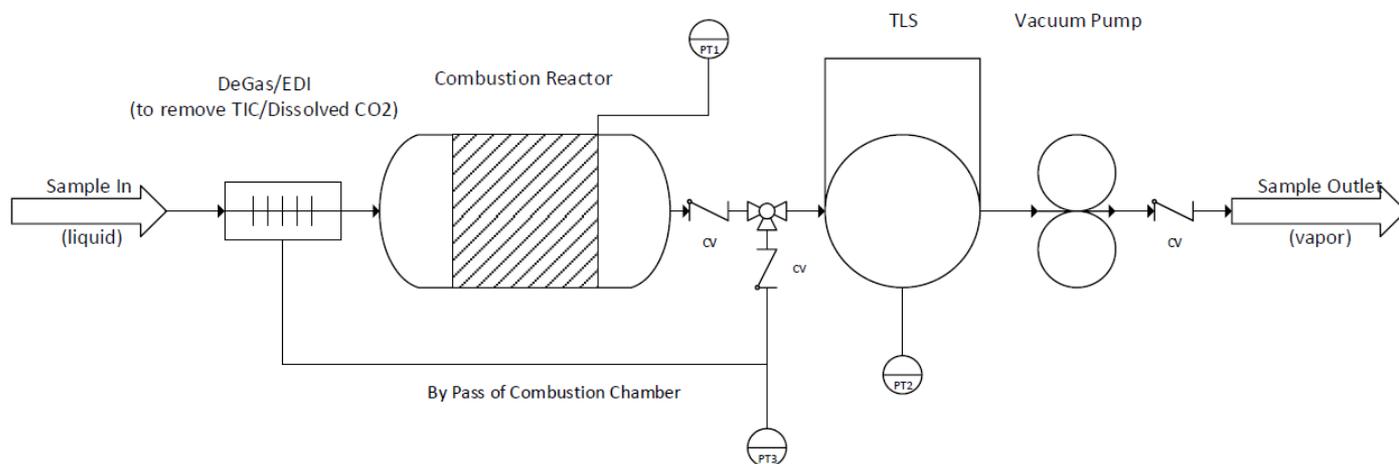
High
Temperature
Combustion

CO₂
Detection

NDIR or TLS

Advantages:

- Combustion/NDIR is a common TOC analysis technique. See Shimadzu, Teledyne, OI Analytical, etc...
- Combustion allows small sample sizes and complete oxidation.
- Tunable Laser Spectrometer is less sensitive to water vapor interference than NDIR.
- pH control is not required if mineral carbonates precipitate and gaseous CO₂ is measured and subtracted.



Technology Architectures: Combustion(Polyarc[®]) / FID

Acidification

Not required

Oxidation

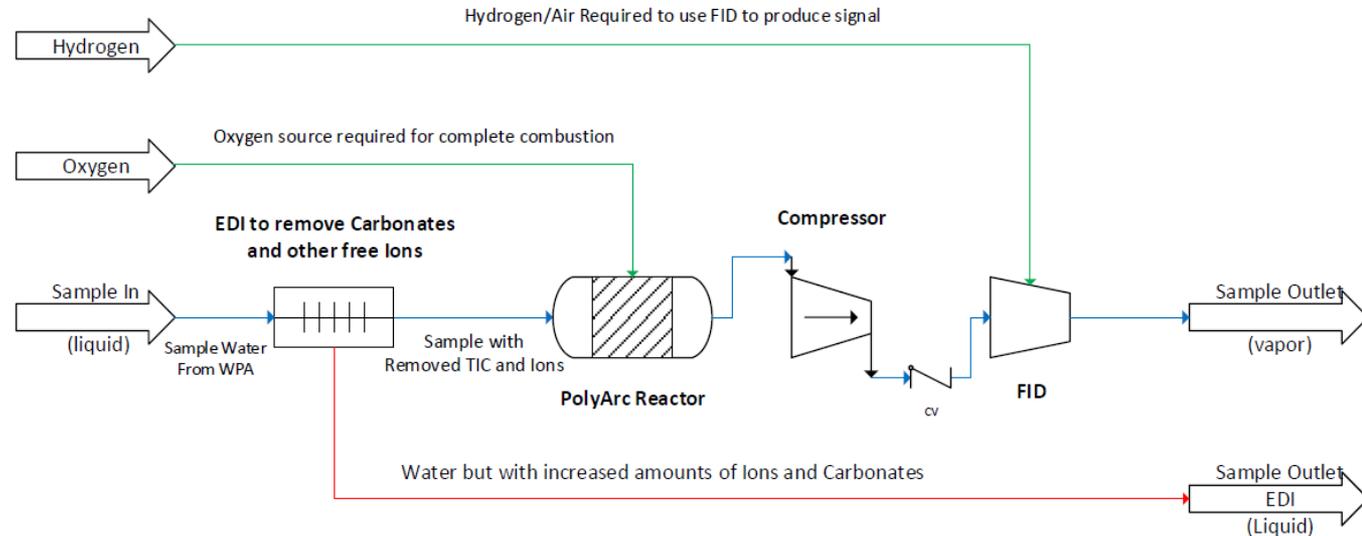
High
Temperature
Combustion

CO₂
Detection

CH₄ - Flame
Ionization
Detector

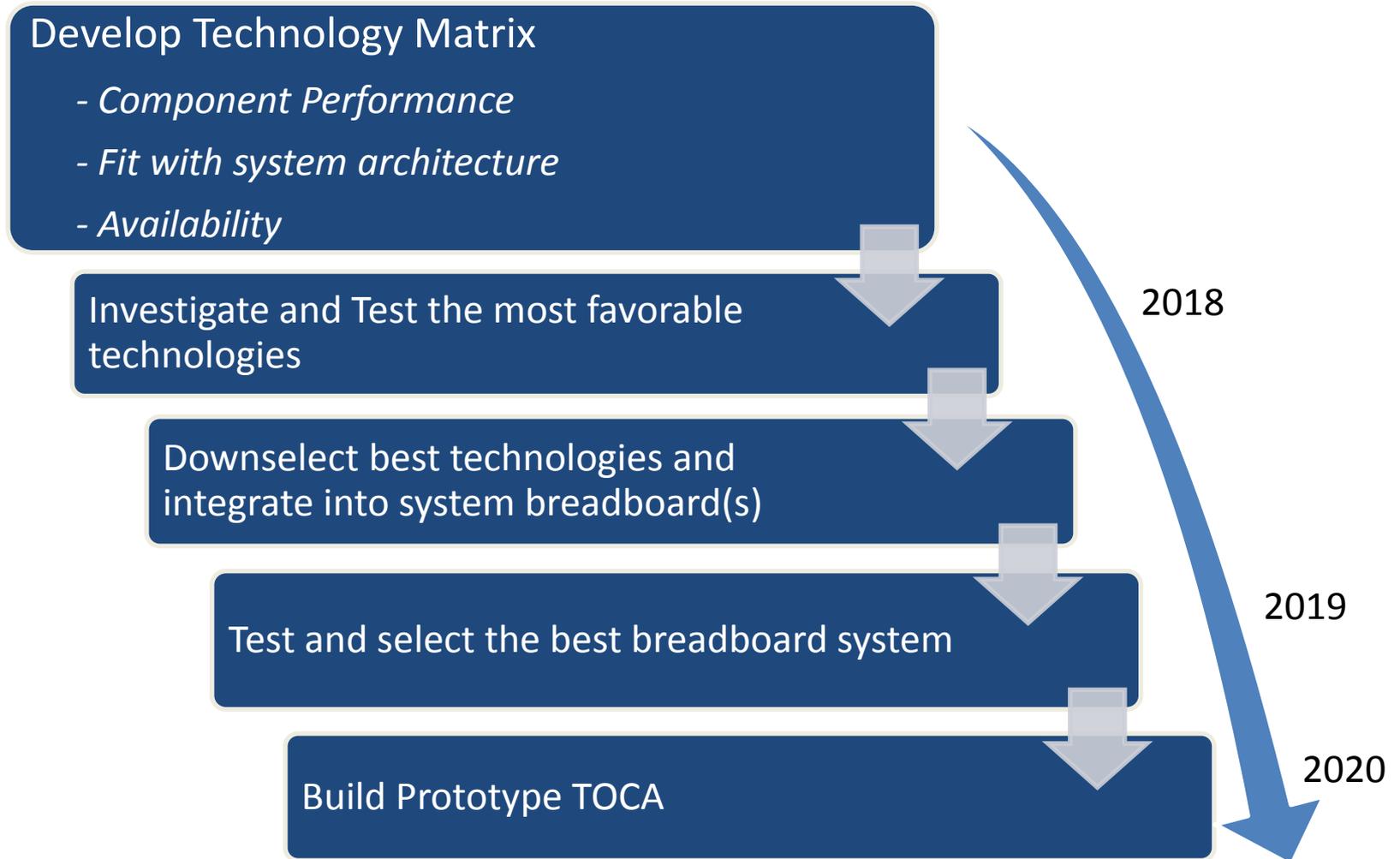
Advantages:

- Polyarc is marketed for liquid injection gas chromatography analysis and has not been applied to TOC analysis.
- Polyarc performs catalyzed combustion and methanization in one step at a lower temperature than TOC combustion chambers.
- FID is selective to methane. Insensitive to water vapor or other interferences.



Project Plan

Technology Maturation



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