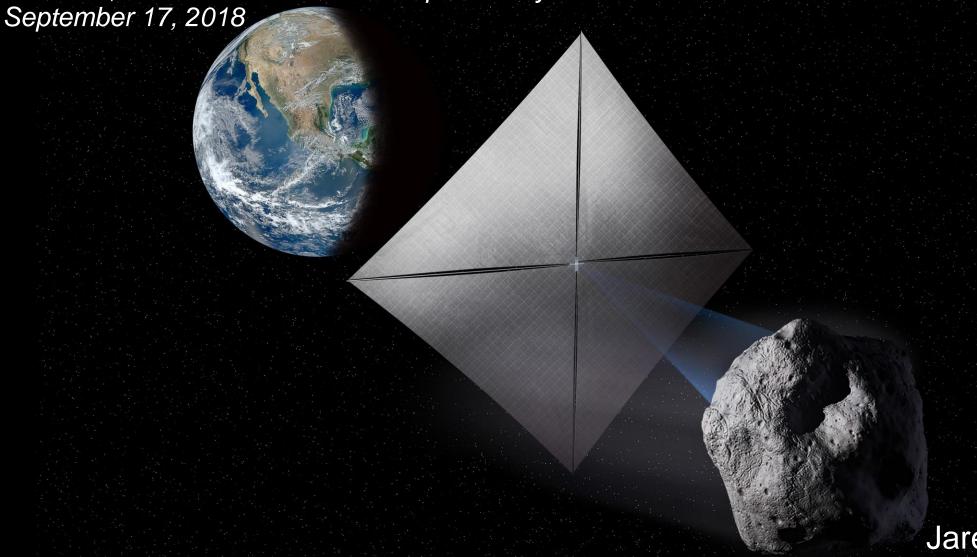


New Moon Explorer (NME) Robotic Mission Concept

2018 AIAA SPACE

EXPL-01, Advanced Power and Propulsion Systems



Jared A. Dervan

NASA Marshall Space Flight Center



Mission/Science Objectives

Science Objectives

- Observe Earth's 'new moon', the newly discovered near-Earth companion 2016HO3
- Obtain spin rate, pole position, shape, structure, mass, density, chemical composition, temperature, thermal inertia, regolith characteristics, and spectral type

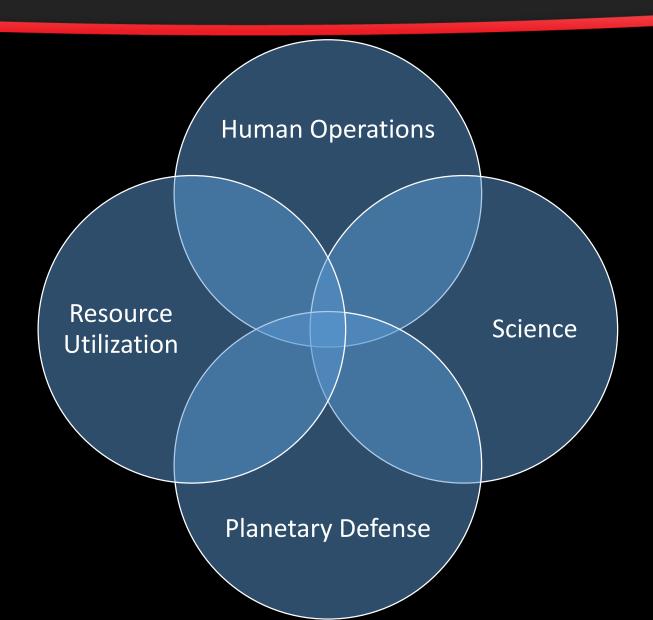
Technology Objectives

- Continue incremental development of solar sail technology
- Demonstrate use of thin-film power technologies

Strategic Objectives

Address synergies across multiple NASA and industry needs

Synergies

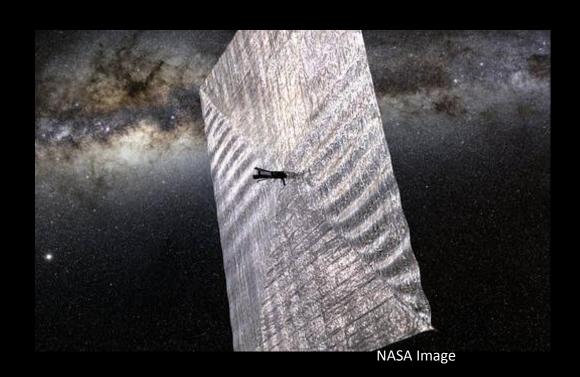


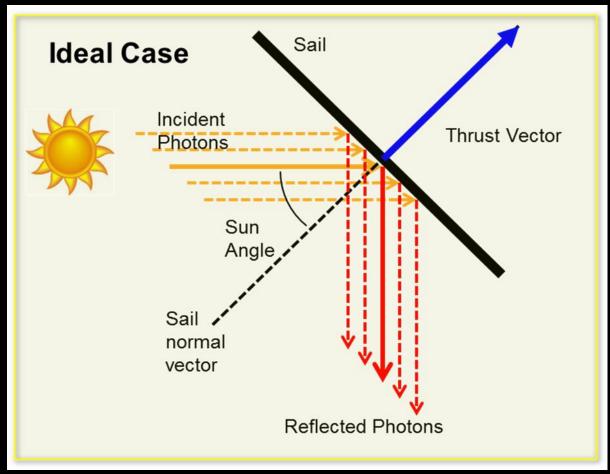
New Moon Explorer sits at the intersection of numerous NASA and commercial objectives



Solar Sails Derive Propulsion By Reflecting Photons

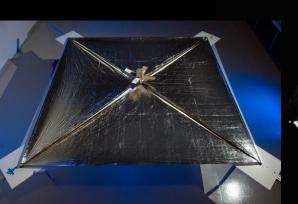
Solar sails use photon "pressure" or force on thin, lightweight, reflective sheets to produce thrust.

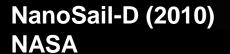






Solar Sail Missions Flown (as of September 7, 2018)





Earth Orbit Deployment Only

3U CubeSat 10 m²



IKAROS (2010) JAXA

Interplanetary Full Flight

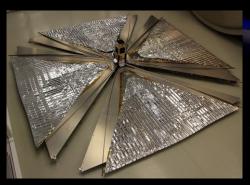
315 kg Smallsat 196 m²



LightSail-1 (2015)
The Planetary Society

Earth Orbit Deployment Only

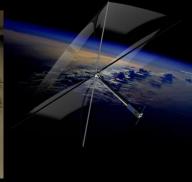
3U CubeSat 32 m²



CanX-7 (2016) Canada

Earth Orbit Deployment Only

3U CubeSat <10 m²



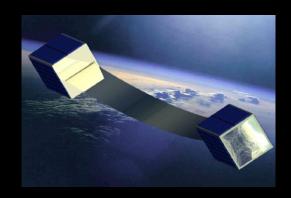
InflateSail (2017) EU/Univ. of Surrey

Earth Orbit Deployment Only

3U CubeSat 10 m²



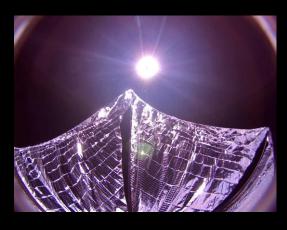
Planned Solar Sail Missions (as of September 7, 2018)



CU Aerospace (2018) Univ. Illinois / NASA

Earth Orbit Full Flight

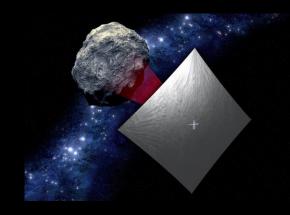
3U CubeSat 20 m²



LightSail-2 (2018) The Planetary Society

Earth Orbit Full Flight

3U CubeSat 32 m²



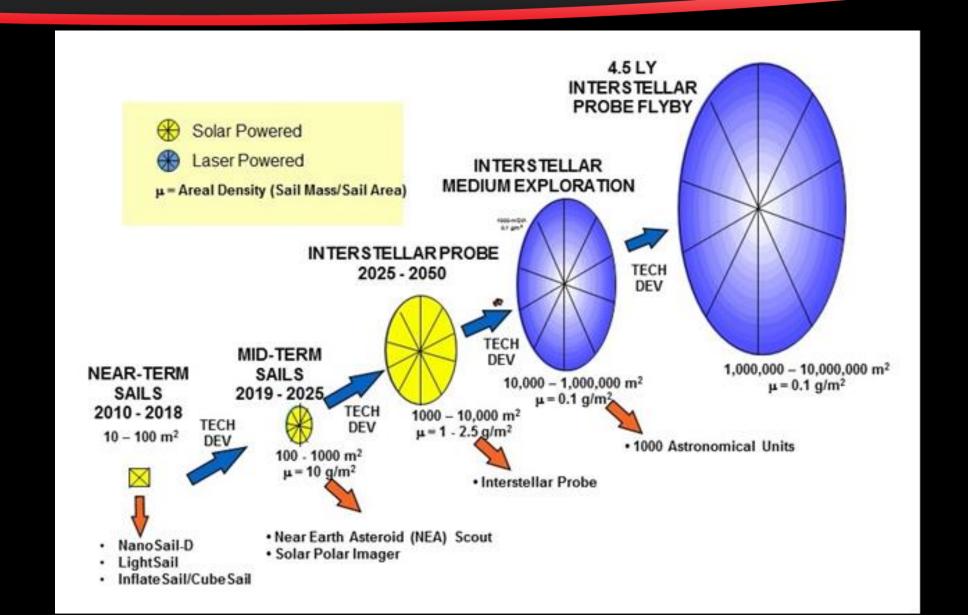
Near Earth Asteroid Scout (2020) NASA

Interplanetary Full Flight

6U CubeSat 86 m²



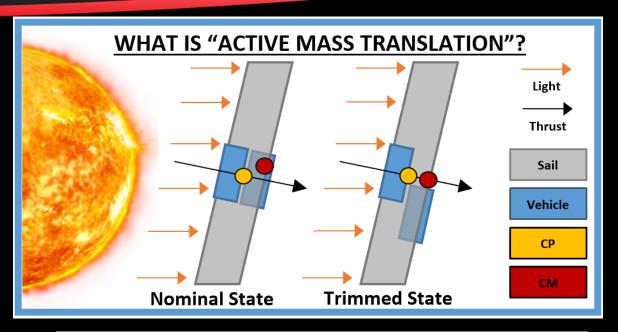
Notional Roadmap To The Future of Solar Sails

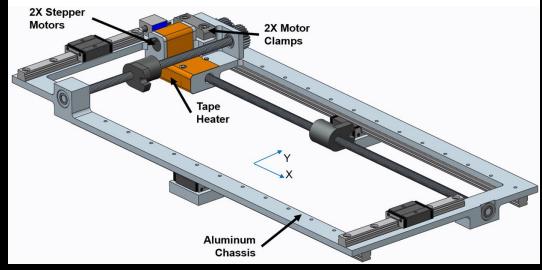




Momentum Management System

- Solar Radiation Pressure imparts a persistent torque on the spacecraft for the duration of the mission
- Use of expendable propellant to maintain desired Solar Sail attitude and/or desaturate reaction wheels would be mission limiting, particularly in small form factors
- A momentum management system is needed to accompany a solar sail concept
- NEA Scout utilizes Active Mass Translation (right) while IKAROS utilized Liquid Crystal Devices







Thin-Film Power Generation

- Leverages technology development from Lightweight Solar Array and an Tenna (LISA-T)
- Thin-film photovoltaics coated with polyimide and solvent bonded on Toughened CP1
- Cells electrically interconnected via microwelded ribbons and embedded traces
- Placed on independent substrate and deployed (can be integral to Solar Sail)
- Phased array antenna can be similarly embedded resulting in integrated propellantless propulsion, power generation, and telecommunications capability





Thin-Film Power Generation (contd.)

Omni – GN&C simplicity and non-pointed



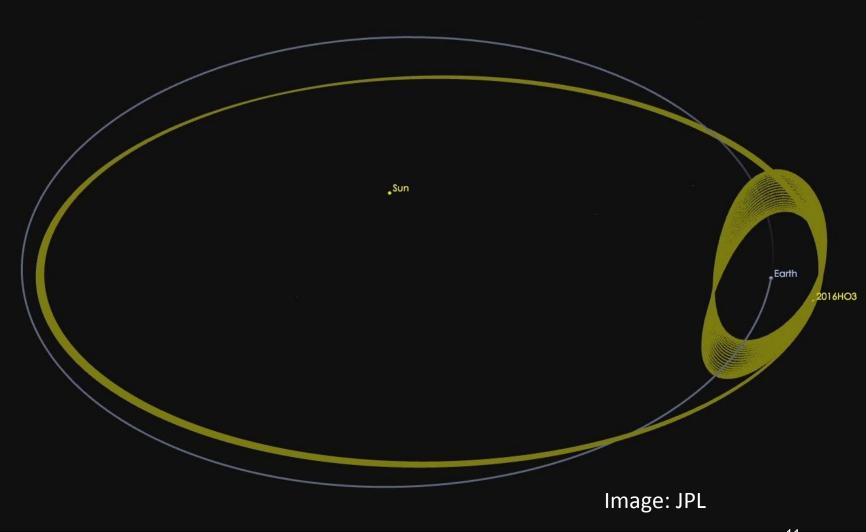
Planar – pointed, high performance





Target Overview

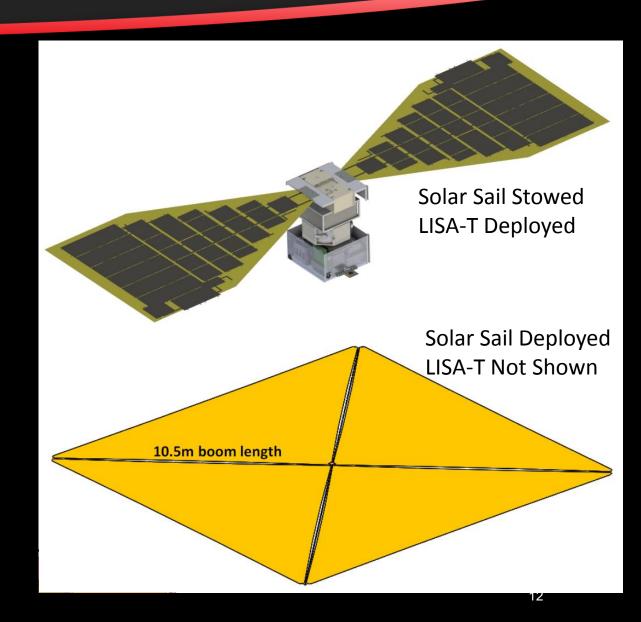
- 2016HO3 is a Near-Earth companion representing the closest, most stable quasi-satellite to Earth
- Discovered by Pan-STARRS on April 27, 2016
- 40-100 meters in diameter
- Earth MOID 0.0348 AU (13.6 LD)
- Fast rotator with an estimated rotational period of 0.467 hours





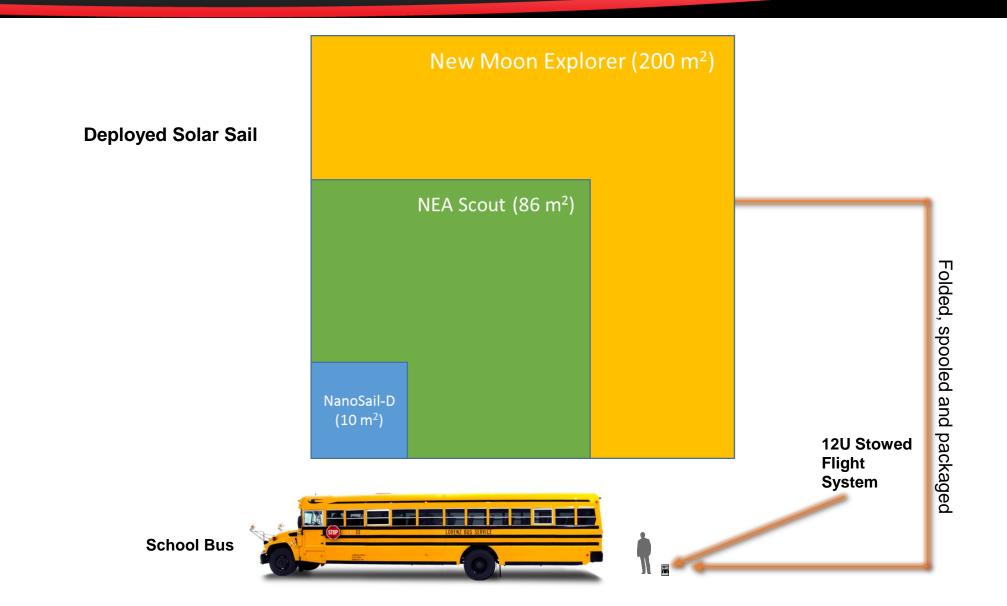
Spacecraft Features

- Low-cost 12U form factor
- Solar Sail propelled
 - 200 m² toughened CP1 quadrant configuration
 - 4x 10.5-m Slit-tube composite booms laminate designed using Roccor Solar Sail Tool (SST)
 - Active Mass Translator MMS
- Planar, bi-pedal 'LISA-T' for power generation and telecommunications
- Deep space CubeSat avionics as utilized on MarCO (launched 2018) and NEA Scout and IceCube missions (launch 2020)
- Cold gas for momentum desaturations and impulsive events
- Leverages developmental lessons learned from the NEA Scout mission





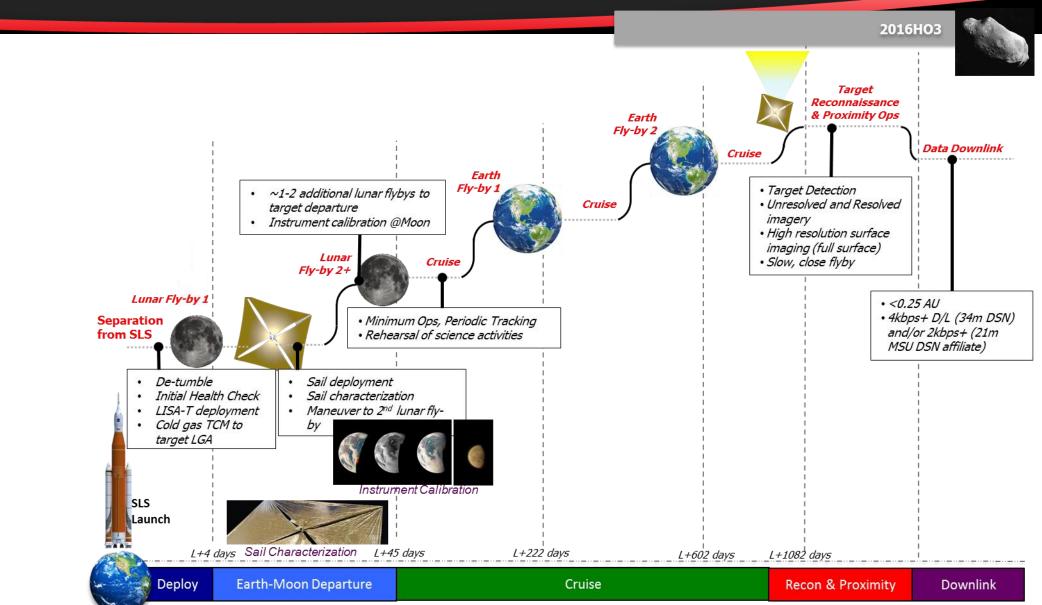
Deployed Solar Sail Approximate Scale





Concept of Operations

Earth





Co-Author Acknowledgements

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- Bejamin Malphrus (Morehead State University)
- Michael Combs (Morehead State University)



BACKUP



NASA's Near Earth Asteroid Scout

The Near Earth Asteroid Scout Will:

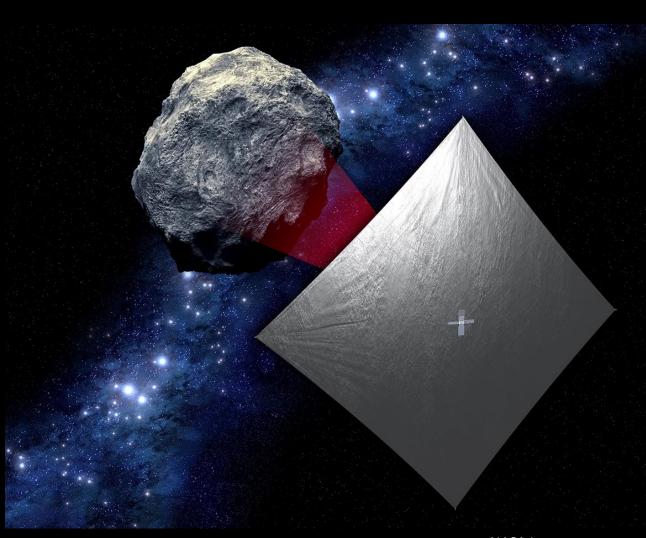
- Image/characterize a NEA during a slow flyby
- Demonstrate a low cost asteroid reconnaissance capability

Key Spacecraft & Mission Parameters

- 6U cubesat (20 cm X 10 cm X 30 cm)
- ~86 m² solar sail propulsion system
- Manifested for launch on the Space Launch System (EM-1/2019)
- Up to 2.5 year mission duration
- 1 AU maximum distance from Earth

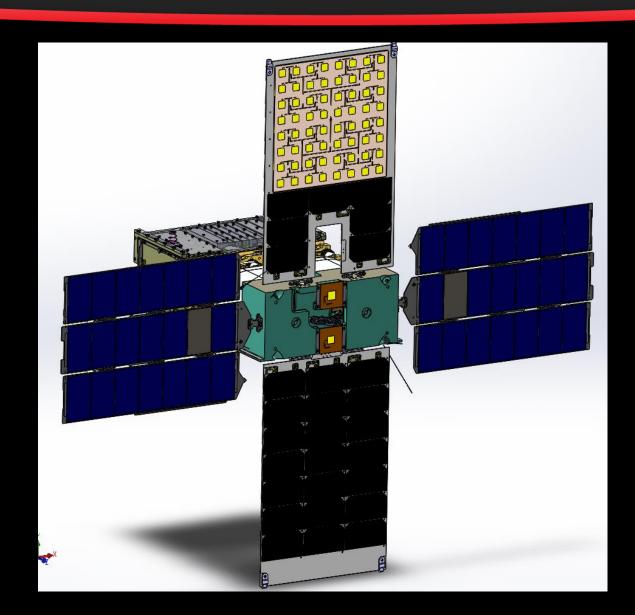
Solar Sail Propulsion System Characteristics

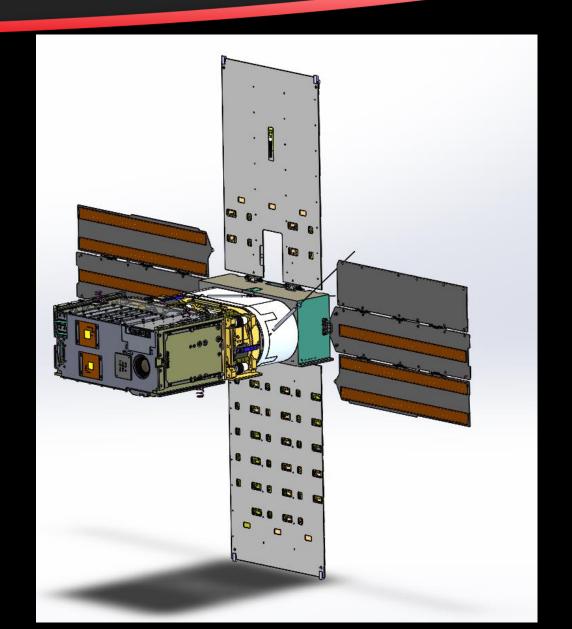
- ~ 7.3 m Trac booms
- 2.5µ aluminized CP-1 substrate
- > 90% reflectivity





NEA Scout Flight System







NEA Scout Hardware Overview











NEA Scout Full Scale Successful Deployment

