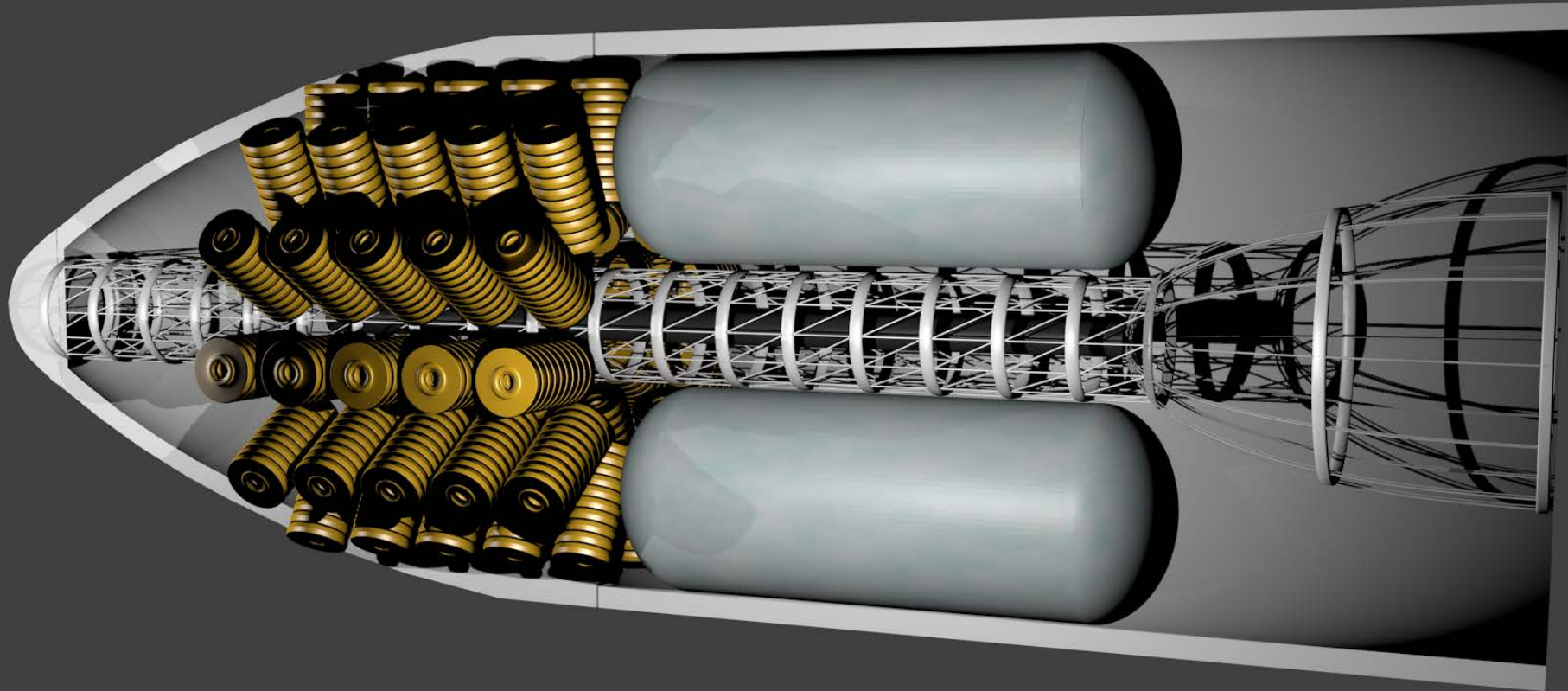


Pulsed Fission Fusion (PuFF) Propulsion System

Project PI: Robert B. Adams, Ph.D.





Team

Project PI: Robert B. Adams, Ph.D.



- Jason Cassibry, Ph.D. – Associate Professor, UAH
- Glen Doughty, NASA-MSFC/ER24
- Brian Taylor, NASA-MSFC/ER23
- Anson Koch, NASA-MSFC/ER23
- Patrick Giddens – Staff Engineer, UAH
- Bill Seidler, Ph.D. – Research Professor, UAH
- Rachel Wagner – Graduate Student, UAH
- Kevin Schillo – Graduate Student, UAH
- Nathan Schilling – Graduate Student, UAH
- Steve Howe, Ph.D. – Howe Industries

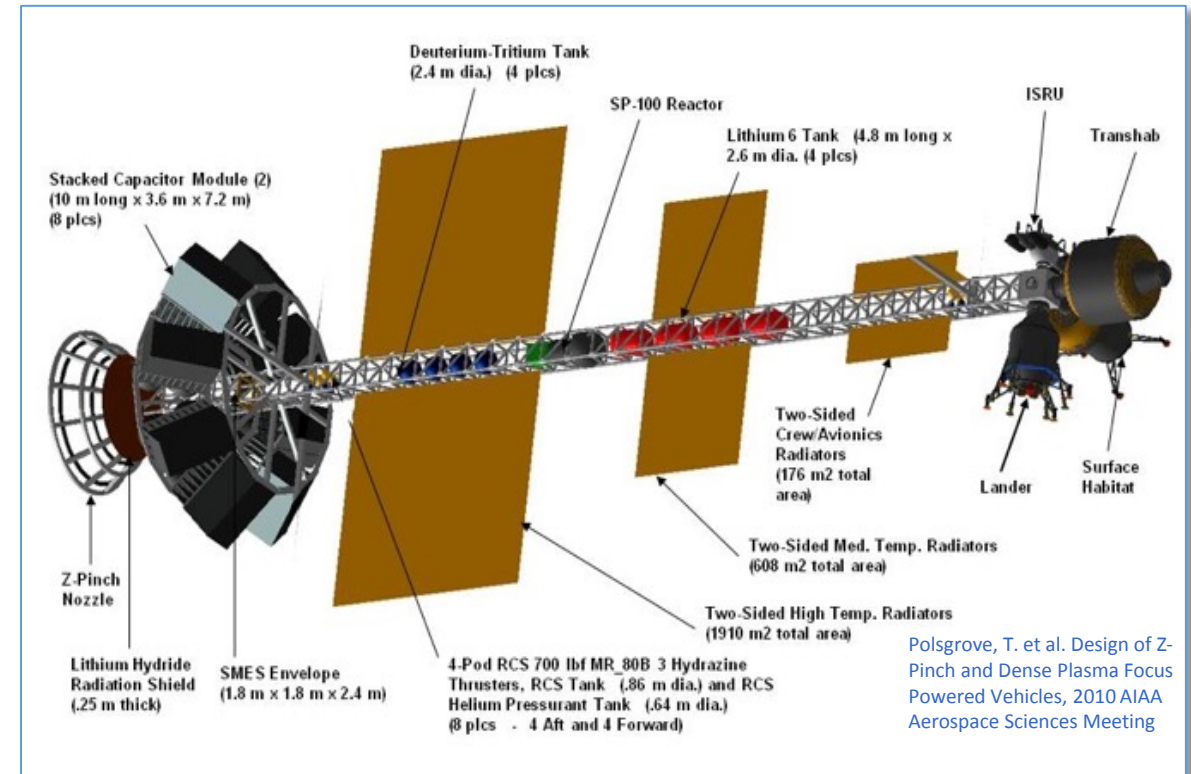
Pulsed Fission Fusion (PuFF) Propulsion Concept

Project PI: Robert B. Adams, Ph.D.



PuFF enables robotic missions to the outer solar system and near interstellar space, and greatly enhances crewed missions in the inner solar system

- The PuFF engine system provides a propulsive impulse operating on the principle of a pulsed two stage nuclear reaction combining Fission and Fusion processes triggered by the compression of a nuclear fuel target (containing small quantities of uranium/tritium) using an intense electrical pulse
- Resultant charged particles, emitted by the impulse, are deflected by magnetic nozzle, also serving as a energy capture device to energize the primary power system capacitors for subsequent pulse
- Concept focused on a single reusable vehicle design enabling a wide range of mission architectures. For example, Mars mission performance sufficient to carry Space Habitat, CEV, Lander, Surface Habitat & ISRU facility (120 mT payload).



	PuFF
Isp	30,000 sec
Thrust	29,000 N
In-Space System Weight (Mars Mission)	240 Metric Tons

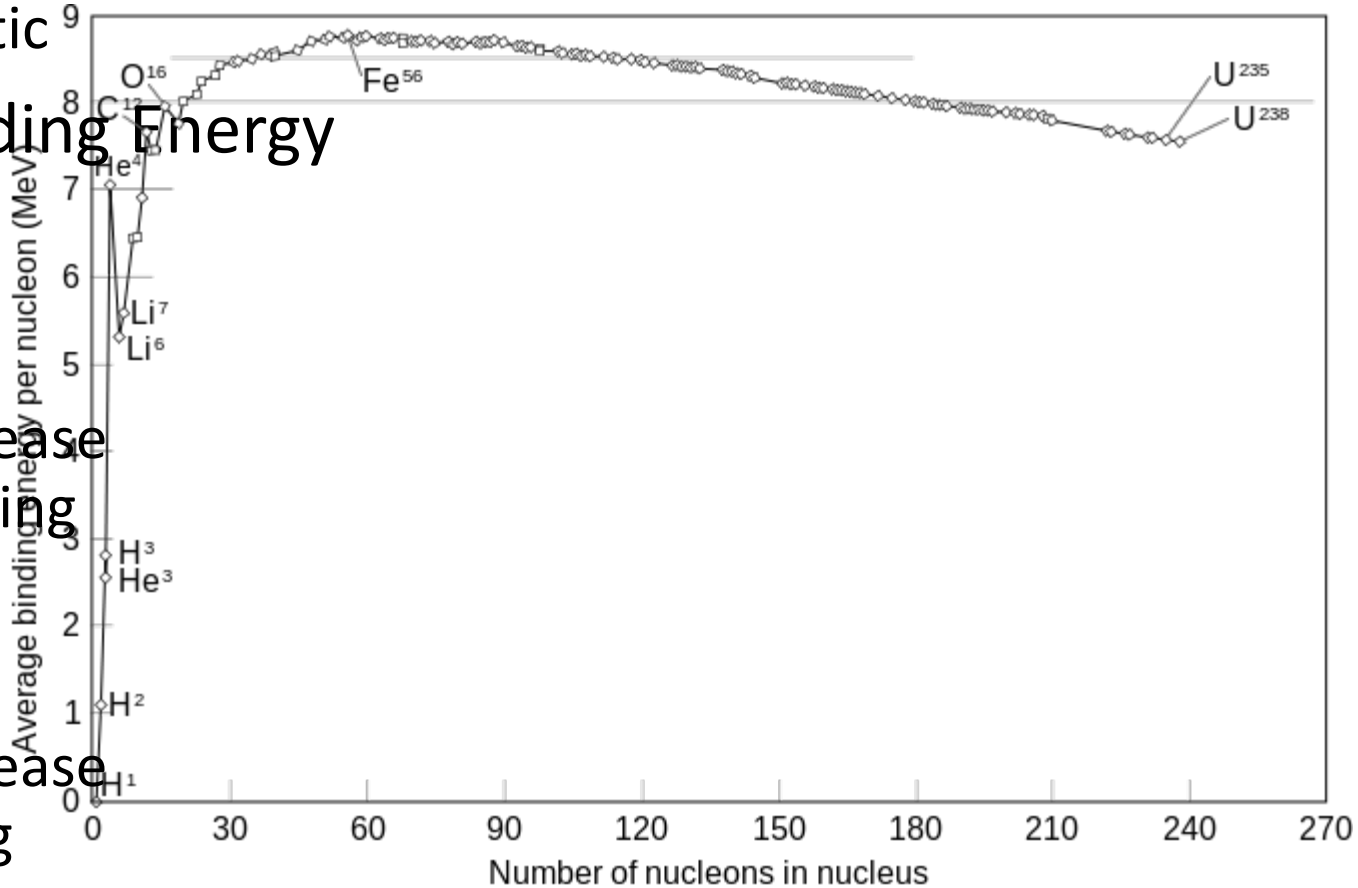
PuFF enables missions throughout the solar system and beyond with a single, in-space, reusable engine design.

Fission and Fusion Energy Release

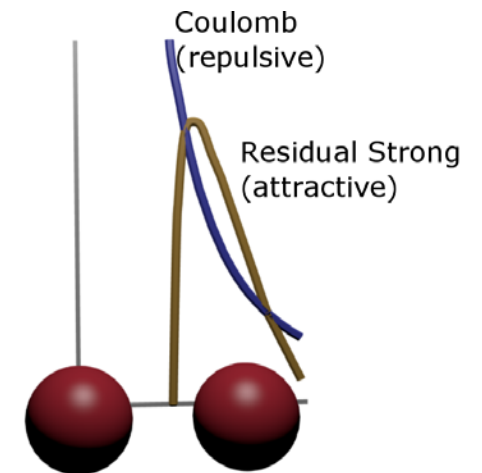


Project PI: Robert B. Adams, Ph.D.

- Mass Defect = Mass of free nucleons – mass of assembled nucleus
 - Nuclear force (residual strong force) stronger than electrostatic
- Nuclear Binding Energy
 - $\frac{E}{A} = \frac{\Delta m}{A} c^2$
- Fusion
 - Energy release by combining nuclei
- Fission
 - Energy release by splitting nuclei



$$E=mc^2$$

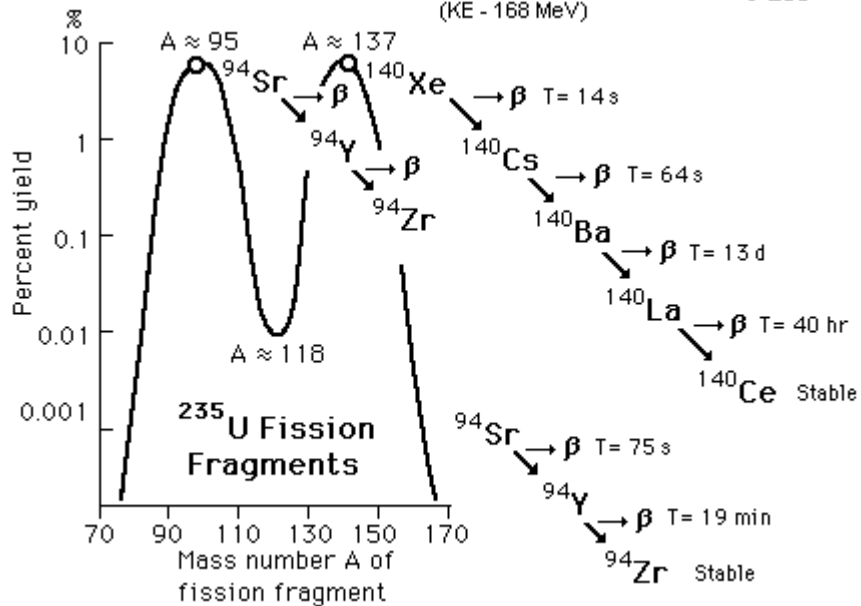
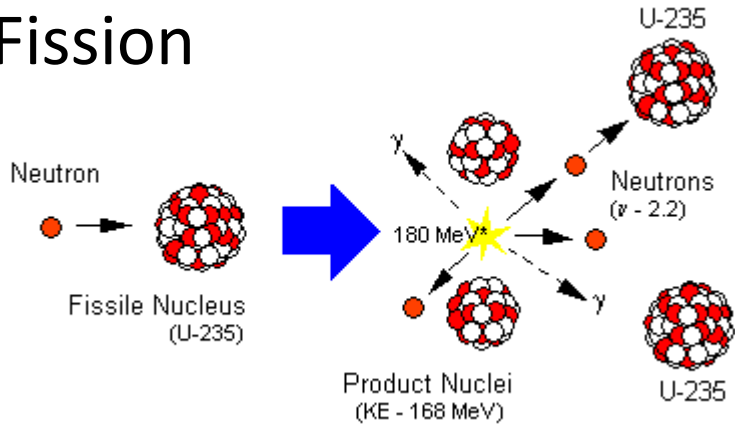


Fission/Fusion Reaction Space

Project PI: Robert B. Adams, Ph.D.



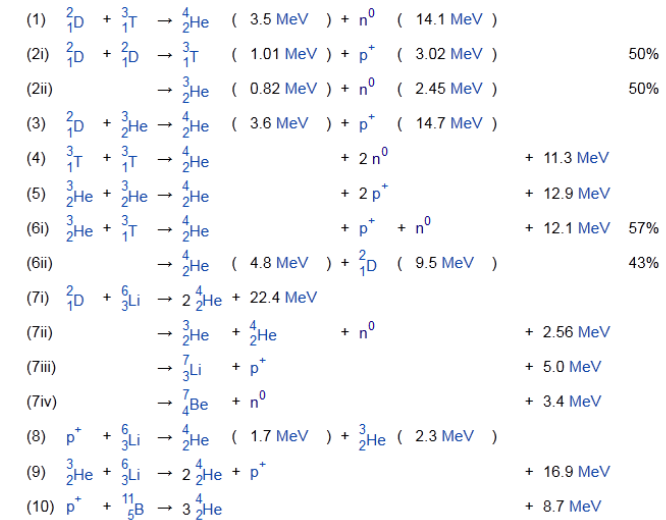
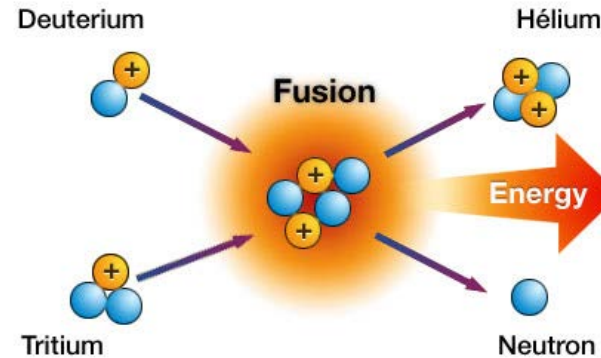
Fission



http://www.propagation.gatech.edu/ECE6390/project/Fall2010/Projects/group10/MANTIS_2010_SatCom/MANTIS_2010_SatCom/PowerSys/default.html

<http://www.mwit.ac.th/~physicslab/hbase/nucene/fisfrag.html#c1>

Fusion



<http://fusionforenergy.europa.eu/understandingfusion/>

http://en.wikipedia.org/wiki/Nuclear_fusion

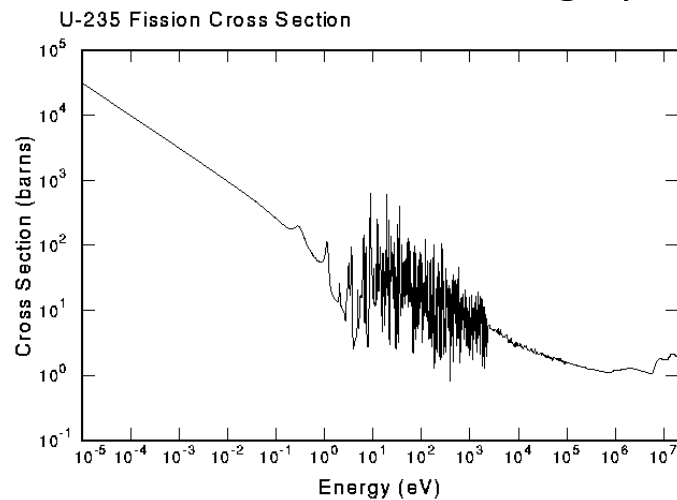
Fission/Fusion Ignition Requirements

Project PI: Robert B. Adams, Ph.D.



- Fission

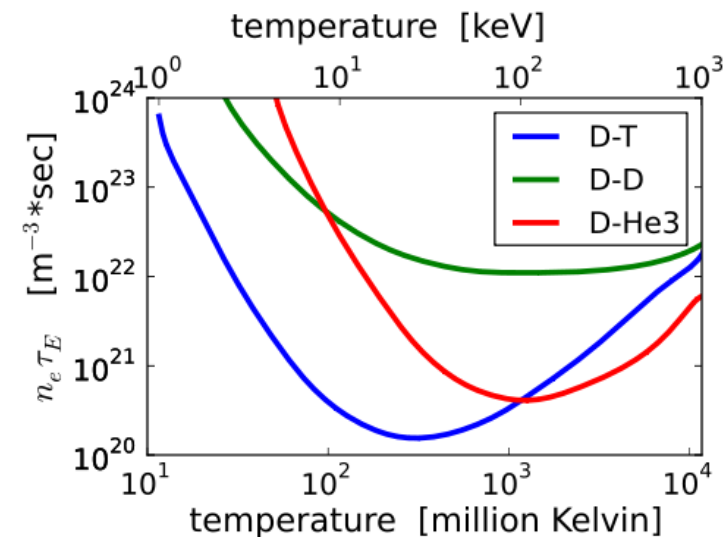
- Criticality is a function of
 - fission cross section
 - Number density
 - And geometry
- Neutrons must balance
 - Lost outside reactor
 - Absorbed through photon



<http://t2.lanl.gov/nis/tour/sch002.html>

- Fusion

- Breakeven is a function of
 - Fusion cross section
 - temperature distribution
 - density
- Lawson Criterion



http://en.wikipedia.org/wiki/Lawson_criterion

Pulsed Fission Fusion (PuFF) Propulsion Operation



Project PI: Robert B. Adams, Ph.D.

PuFF utilizes a multistage nuclear process

• **Pre-reaction**

- Lithium shell/cone is injected to bridge the power system anode to target holder.
- 2 mega-amps (at 2 mega-volts) travels along the liquid lithium cone to target.
- Lorentz force ($j \times B$) produced by the current/magnetic field compresses a hybrid target of uranium/Deuterium-Tritium (D-T), reaching criticality for the Uranium.

• **First Stage (Fission)**

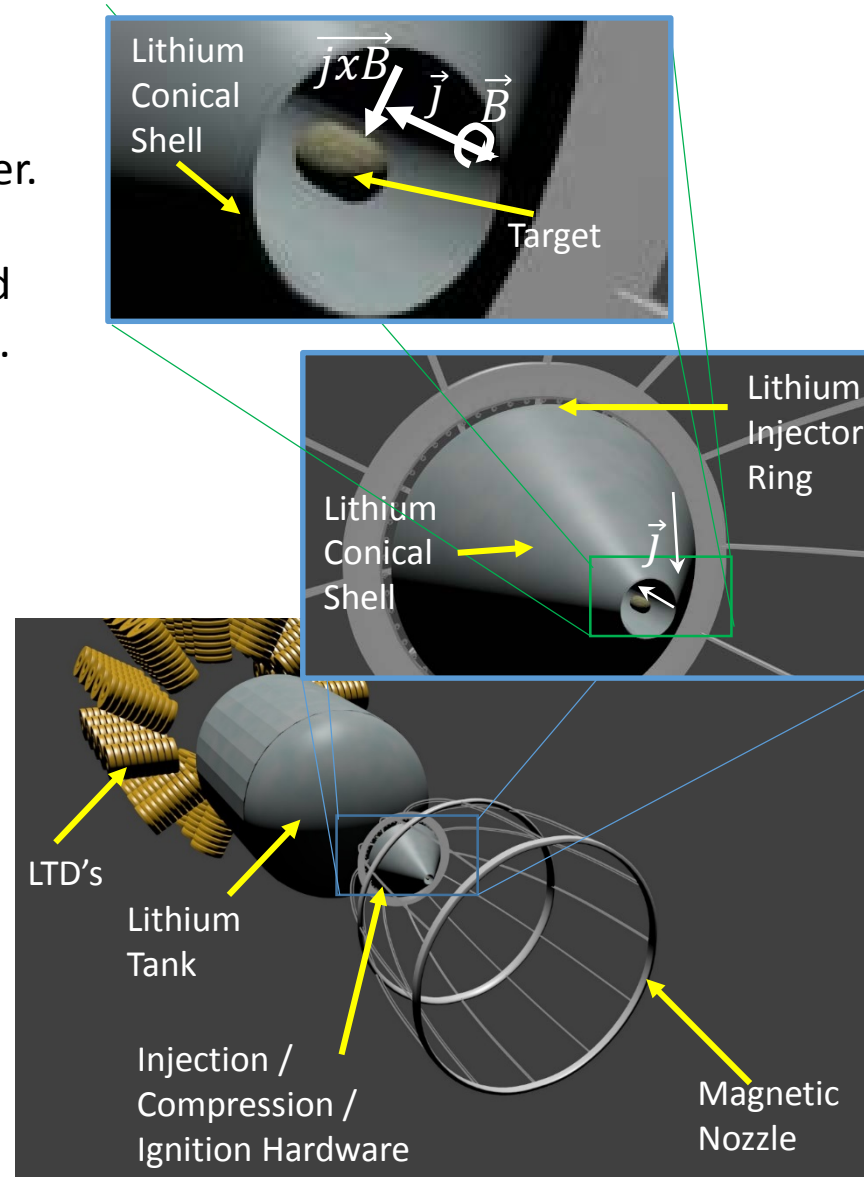
- Uranium criticality produces spontaneous fission reaction (heating)
- Fission heats the D-T fuel creating fusion conditions (interaction cross-section)

• **Second Stage (Fission - Fusion Cascade)**

- Fusion produces additional neutron which in turn ignites more fission
- Additional fission reactions generate more heat, boosting fusion rate
- Fission to D-T fusion cycle cascades until burnout.

• **Expansion**

- Plasma produced during impulse expands outward against magnetic nozzle
- Magnetic nozzle directs particles generating thrust & captures energy necessary to initiate the next pulse
- Single target impulse event requires several microseconds; repeat up to 100 Hz

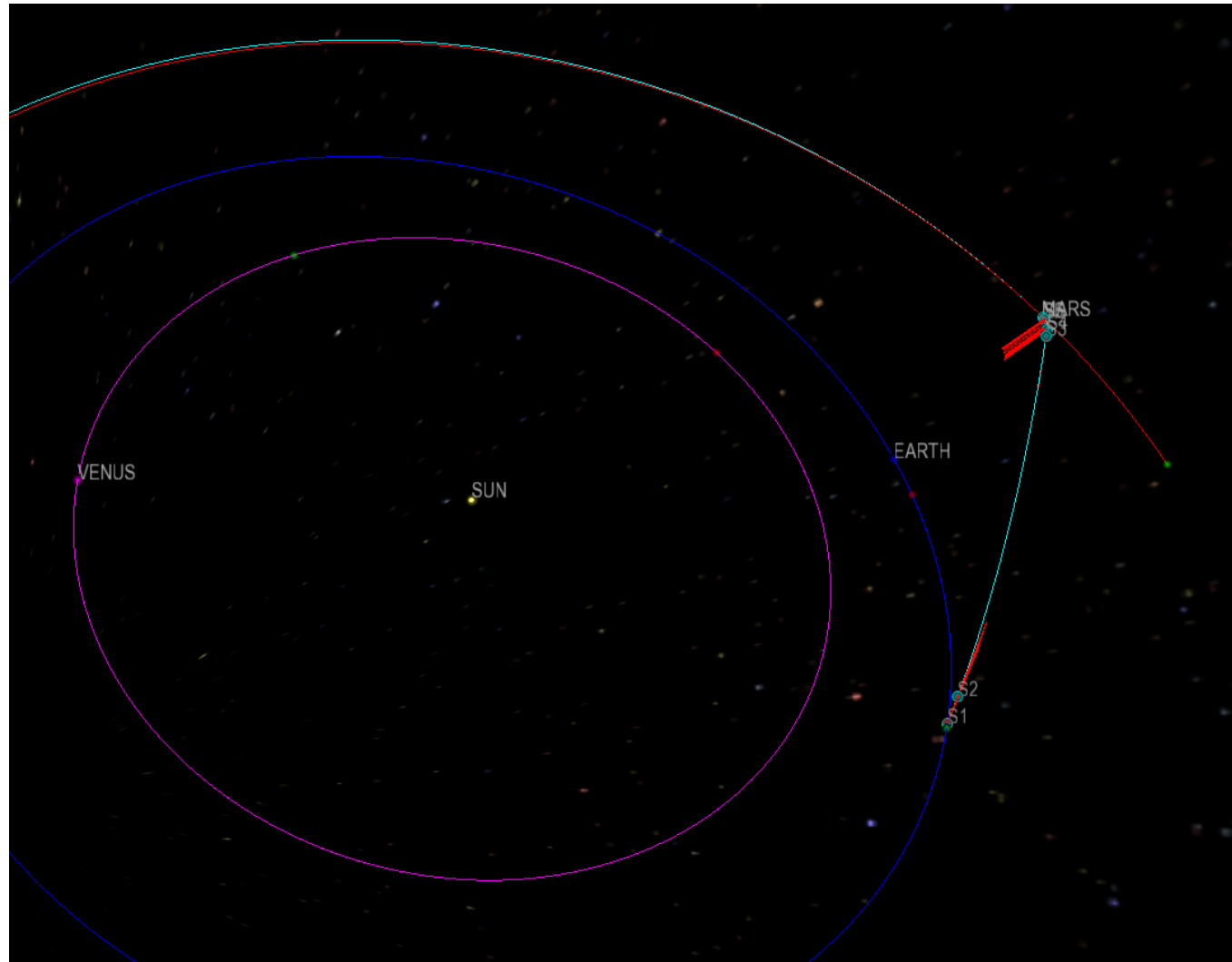


Mars High Speed Mission

Project PI: Robert B. Adams, Ph.D.



- Earth to Mars in 37 days
 - 0.6 Earth escape
 - 2.6 day TMI
 - 31.4 day coast
 - 0.8 day Mars deceleration
 - 2.1 day Mars capture
- Payload
 - 25 mT crew compartment



Interstellar Precursor Mission Analysis

Project PI: Robert B. Adams, Ph.D.

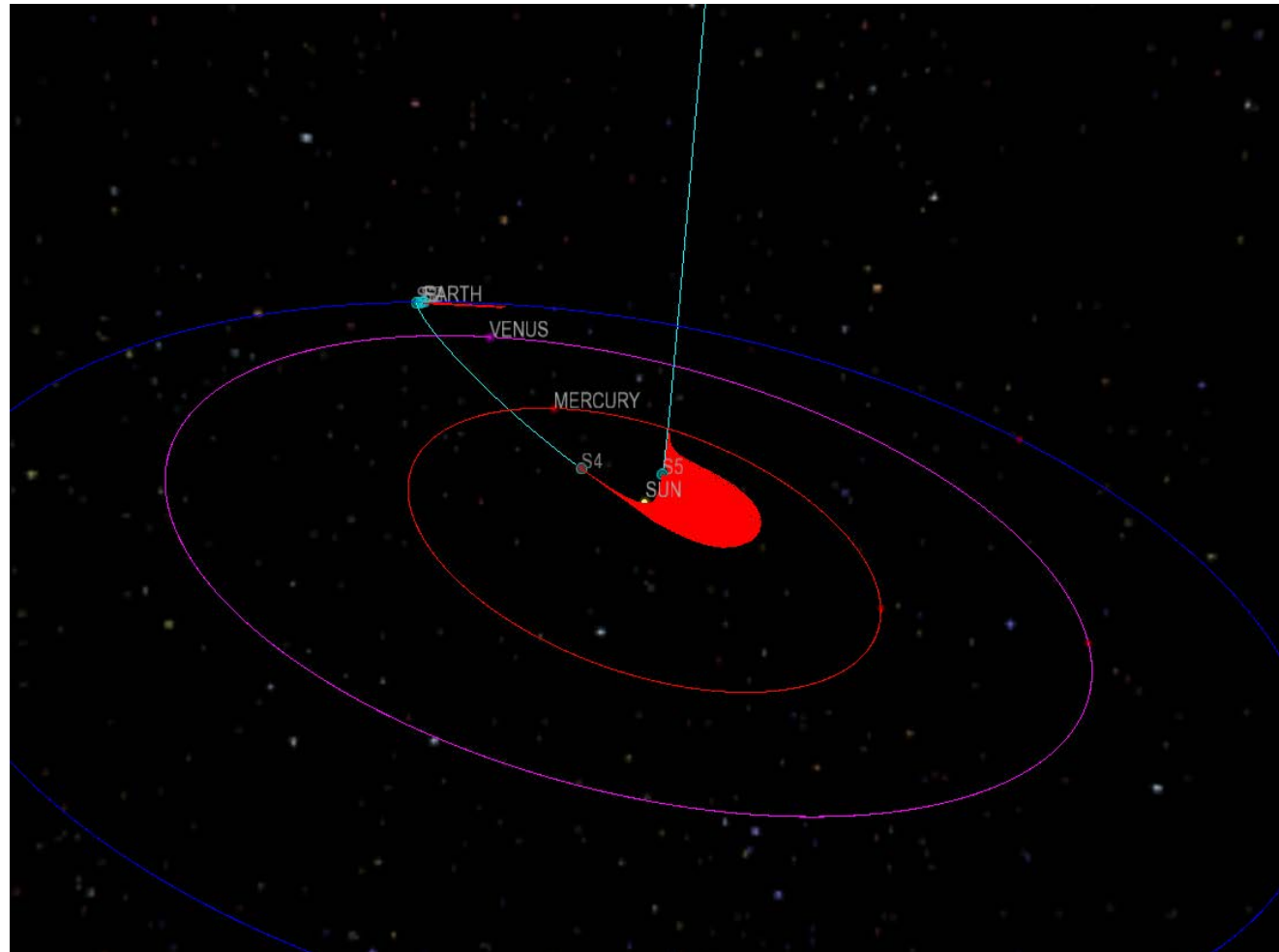


- Interstellar Space

- Termination shock in 5 years (pass Voyager I)
- 275 AU in 10 years
- Solar gravitational lens in 20 years
- 1000 AU in 36 years

- Burn profile

- 0.4 days Earth escape
- 1.4 days deorbit
- 48 day inbound coast
- 2.5 day solar burnout



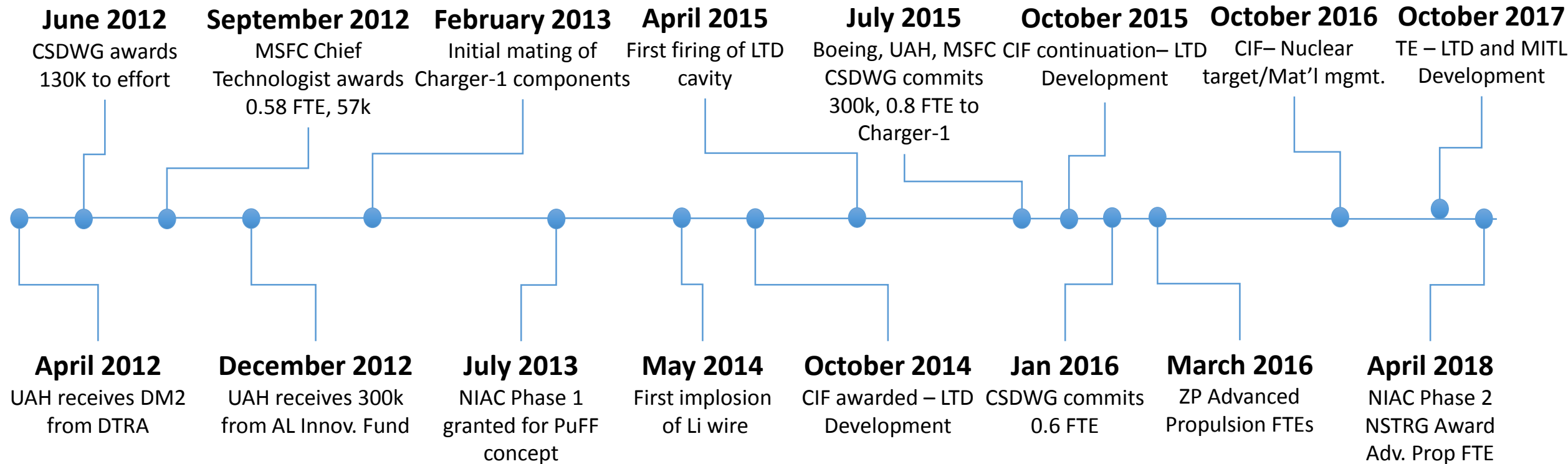
Research and Development Support for Charger – 1, PuFF, LTD



Project PI: Robert B. Adams, Ph.D.

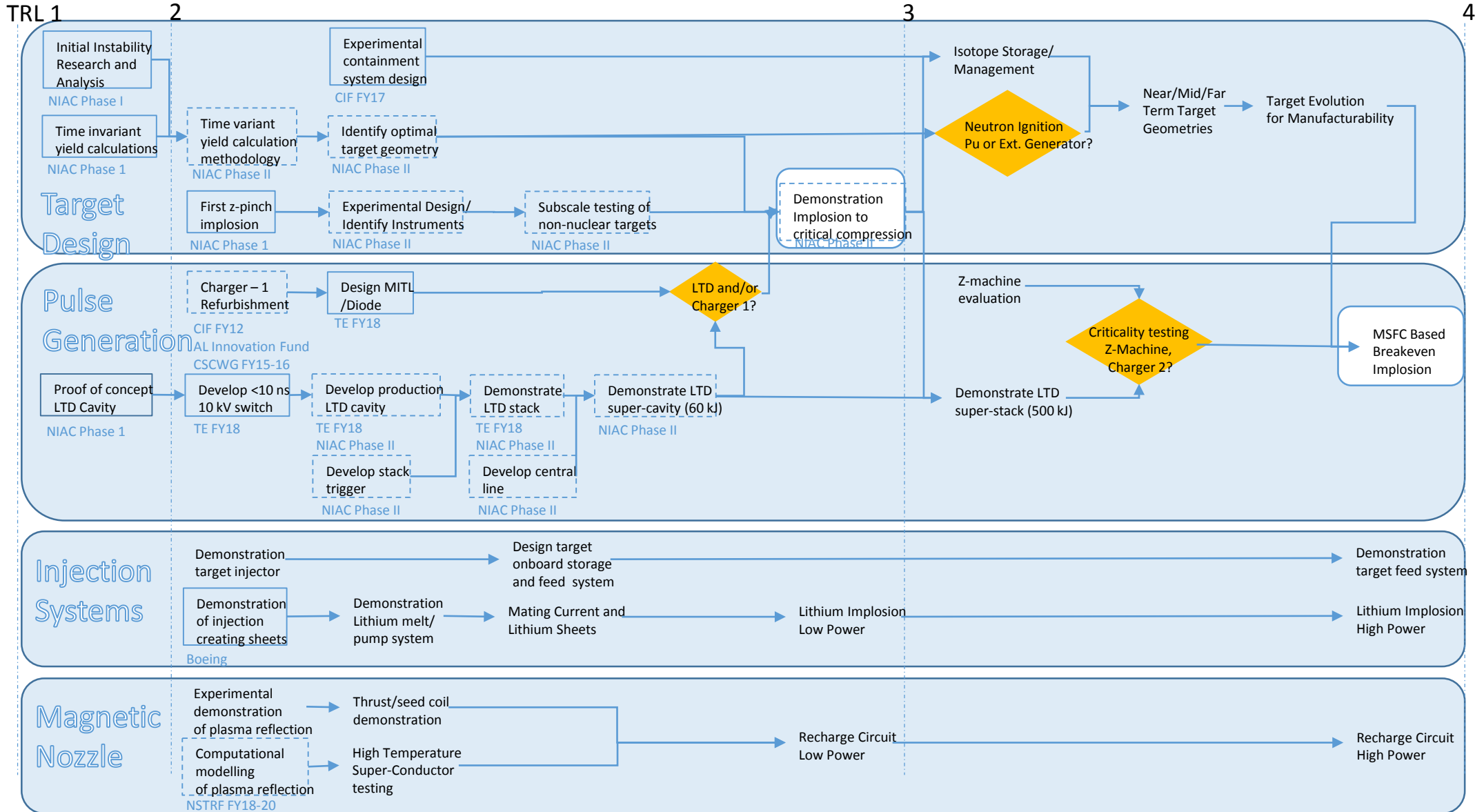
Historical timeline for basic R&D activities related to PUFF

- Charger-1, PuFF and LTD development provides a unique high power research and development capability for MSFC
- Limited funding from varied sources and partners – focused on small hardware evaluations
- Striving to maintain forward momentum



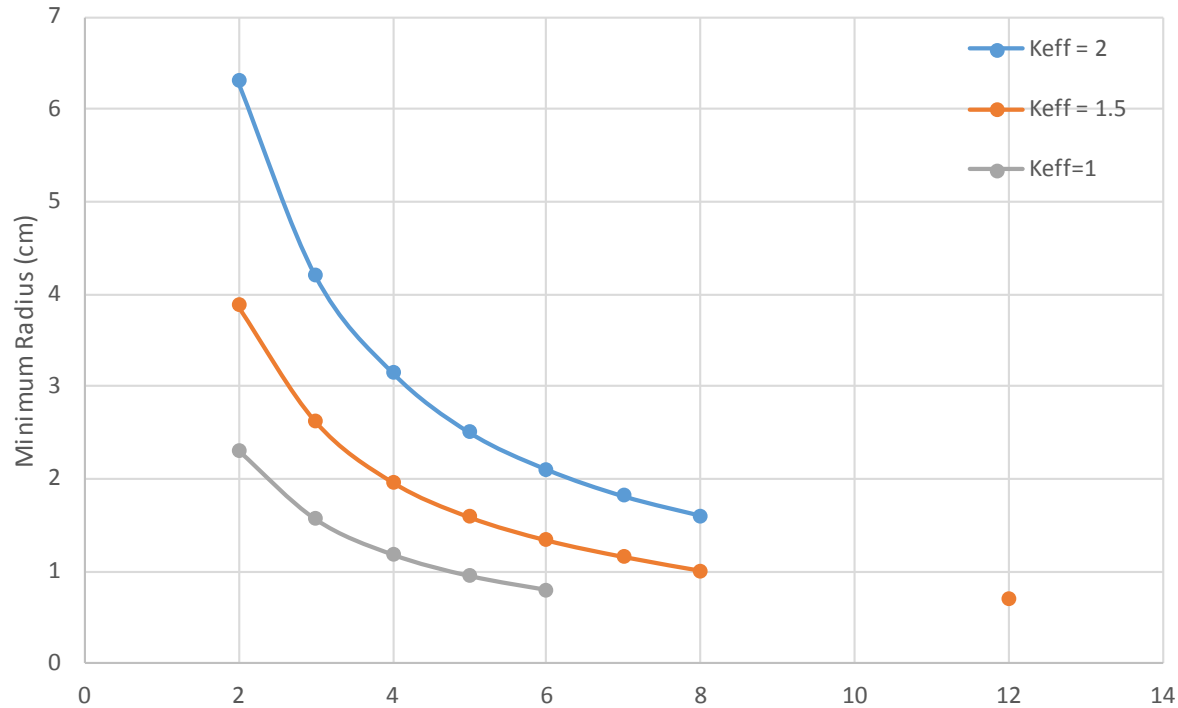
Integrated Development Program – TRL 1-4

Project PI: Robert B. Adams, Ph.D.



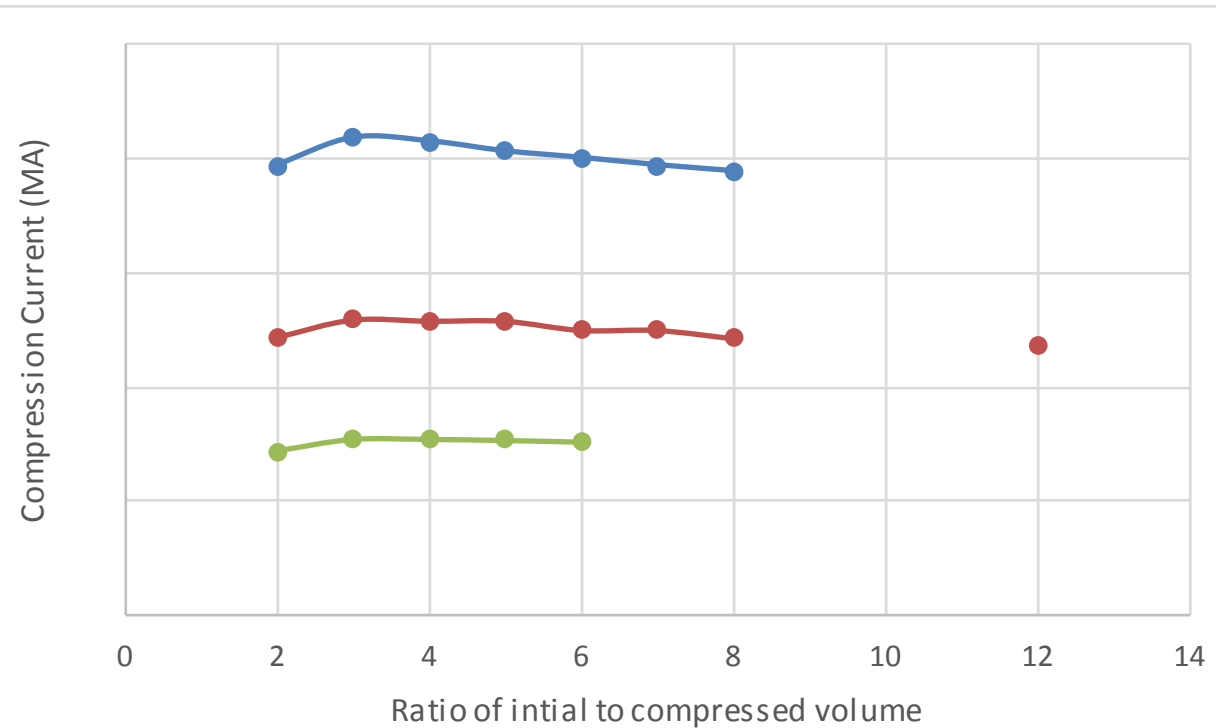
Target Design and Modelling

Project PI: Robert B. Adams, Ph.D.



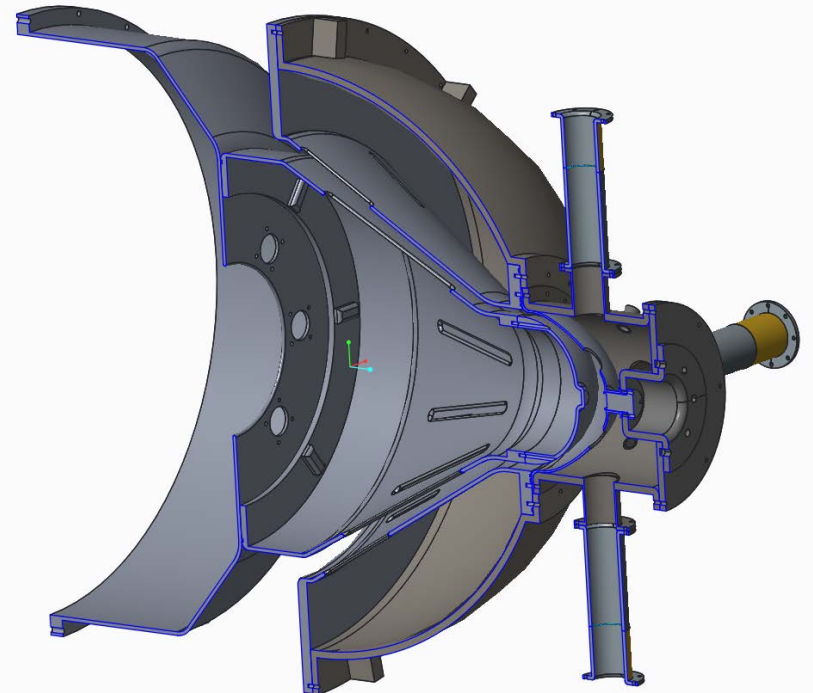
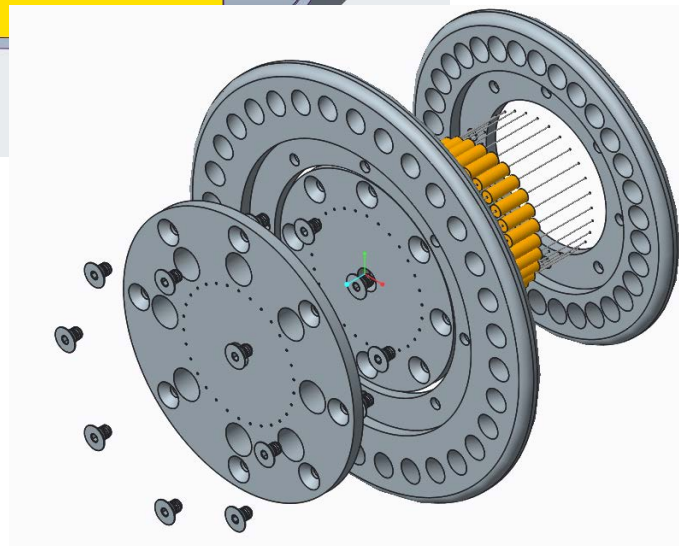
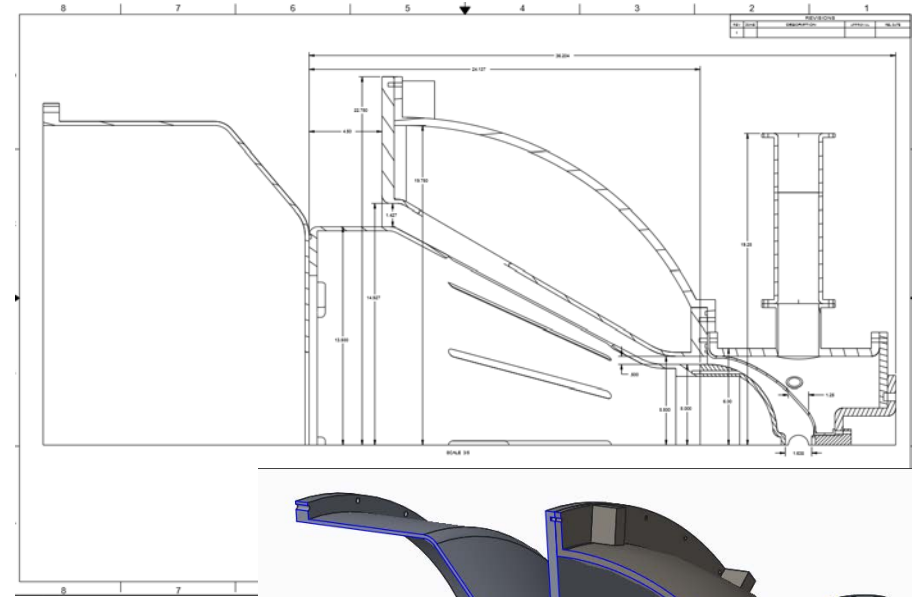
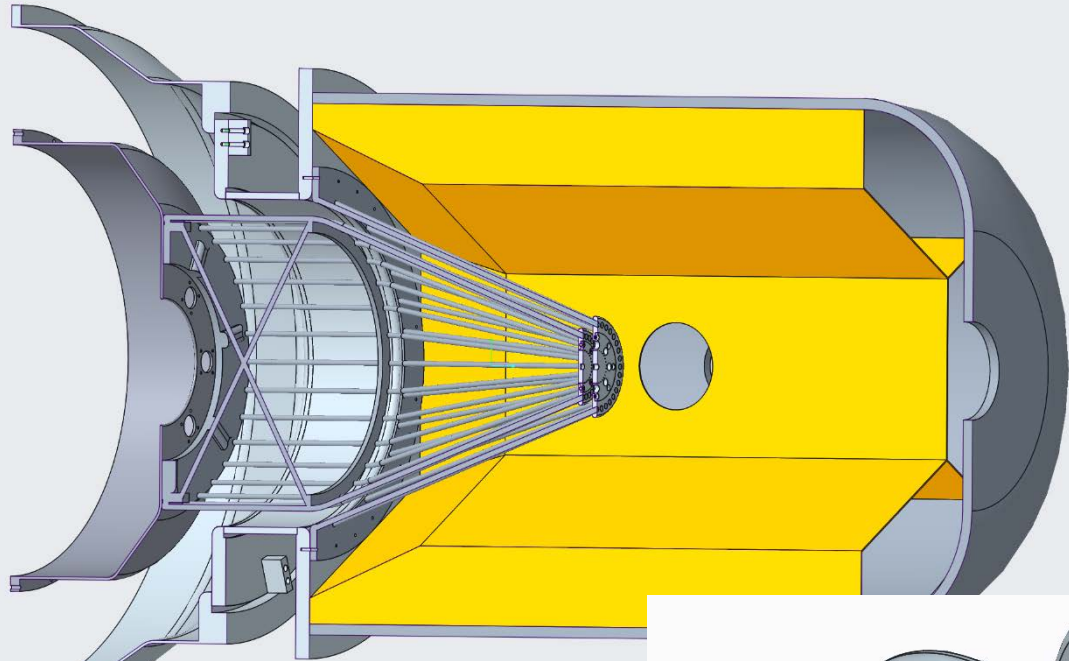
- Time invariant model

- MCNP criticality runs
- EOS to determine pressure, current reqt's



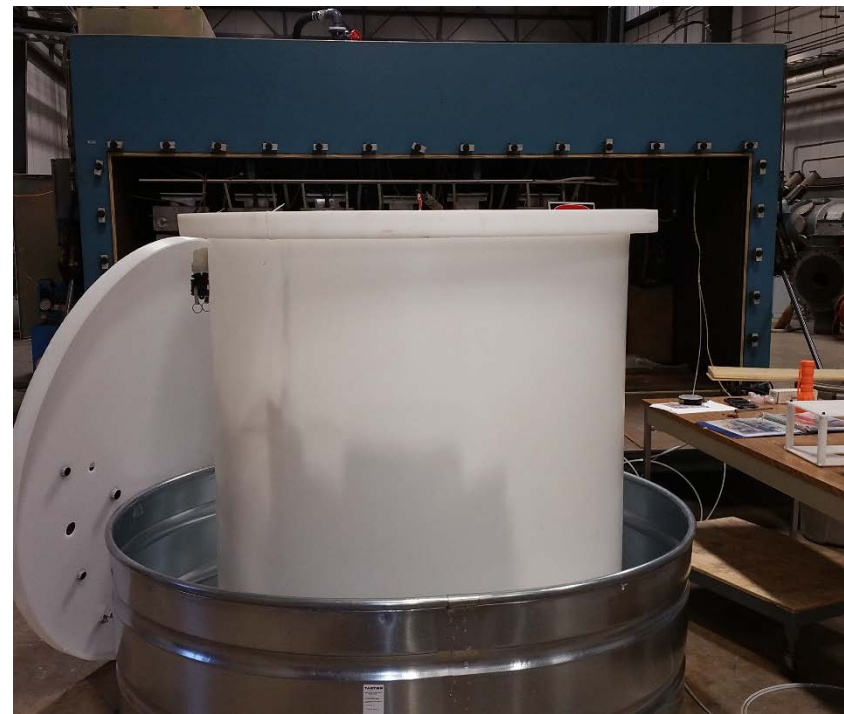
Charger 1 Refurbishment – MITL Development

Project PI: Robert B. Adams, Ph.D.



Charger 1 Refurbishment – Mini-Marx Testing

Project PI: Robert B. Adams, Ph.D.



Linear Transformer Driver – Cavity and Stack Development

Project PI: Robert B. Adams, Ph.D.

