

# ADR Design and Operation for the X-Ray Imaging and Spectrometer Mission (XRISM)

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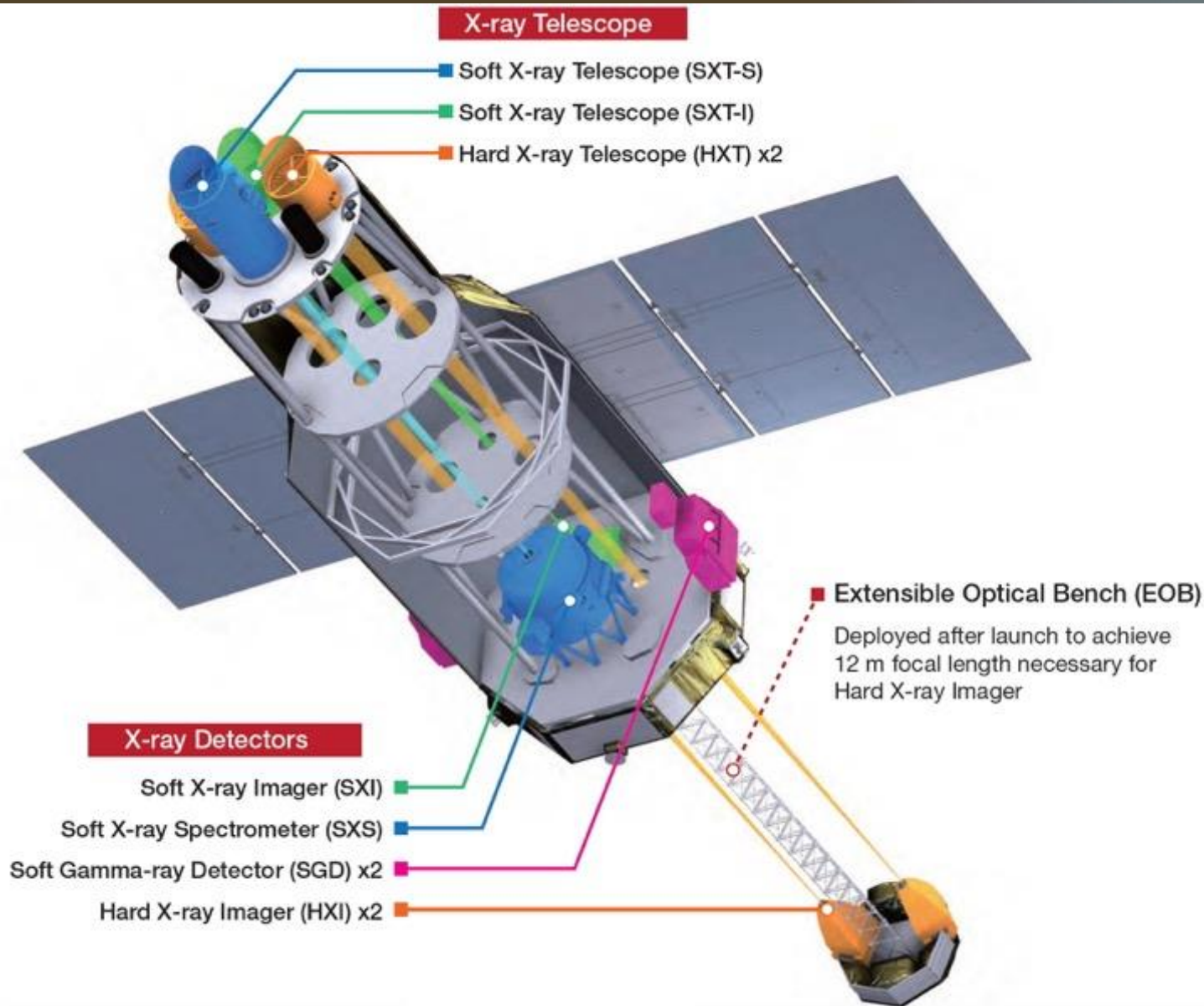
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NASA/Goddard Space Flight Center  
Greenbelt, MD 20771 USA



Launch: February 17, 2016

# Hitomi (formerly Astro-H)



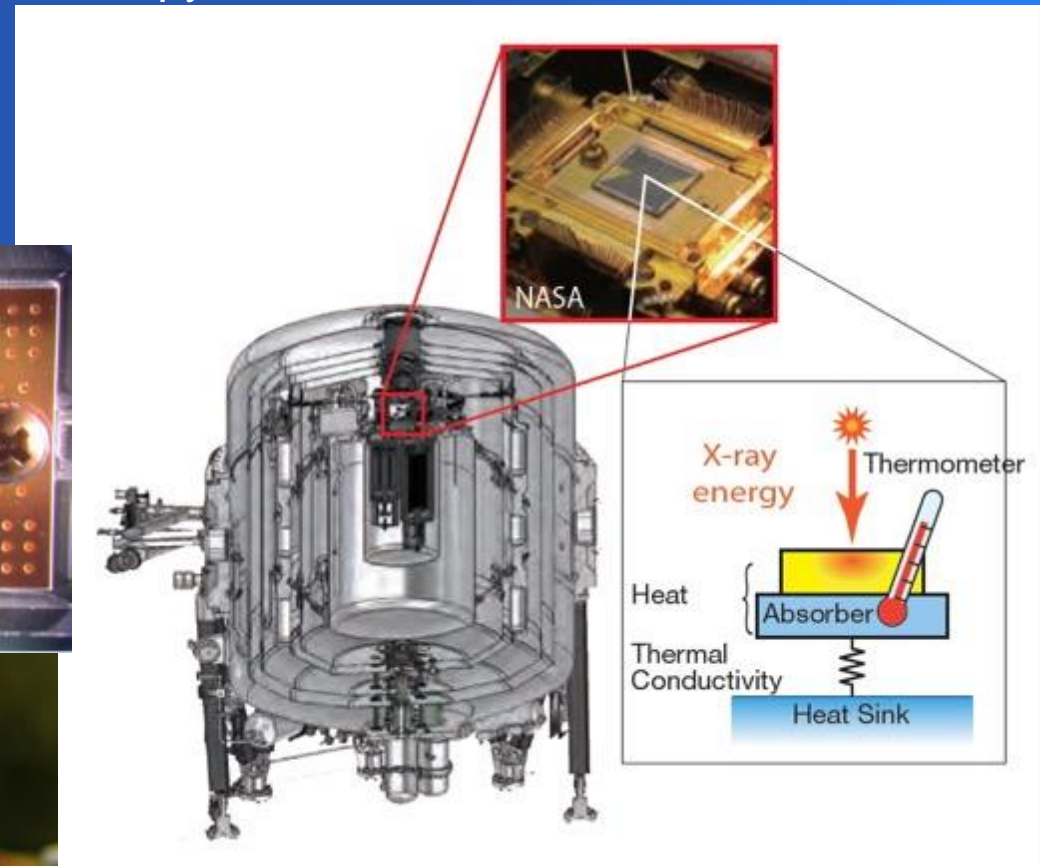
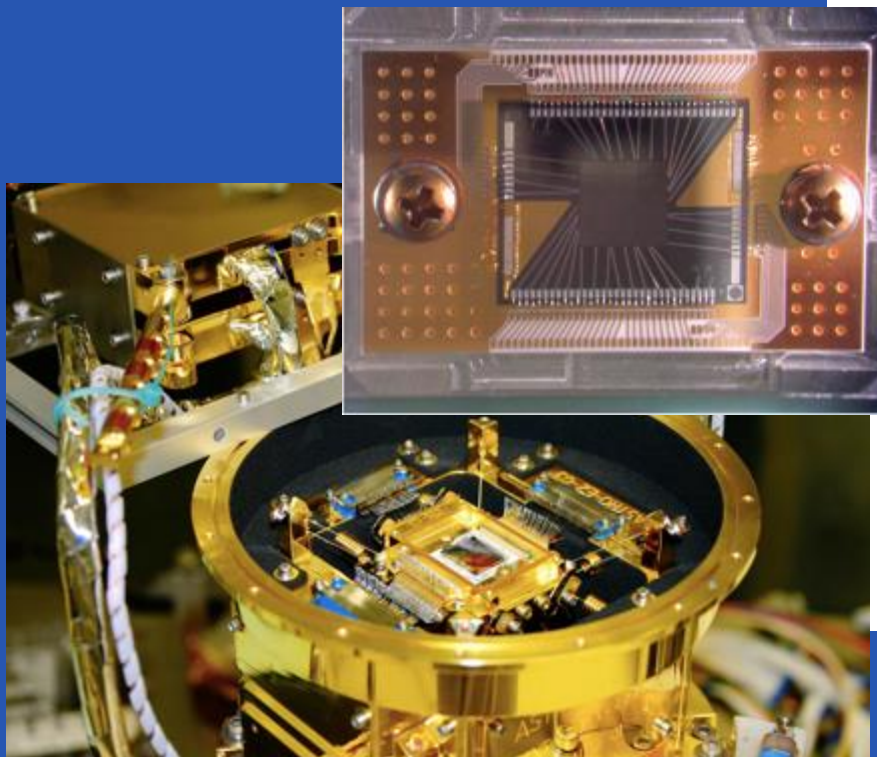
Study structure and evolution of the Universe

Study matter in extreme environments

- Black holes
- Galaxies
- Heavy elements
- Cosmic rays

# The Resolve X-ray Spectrometer

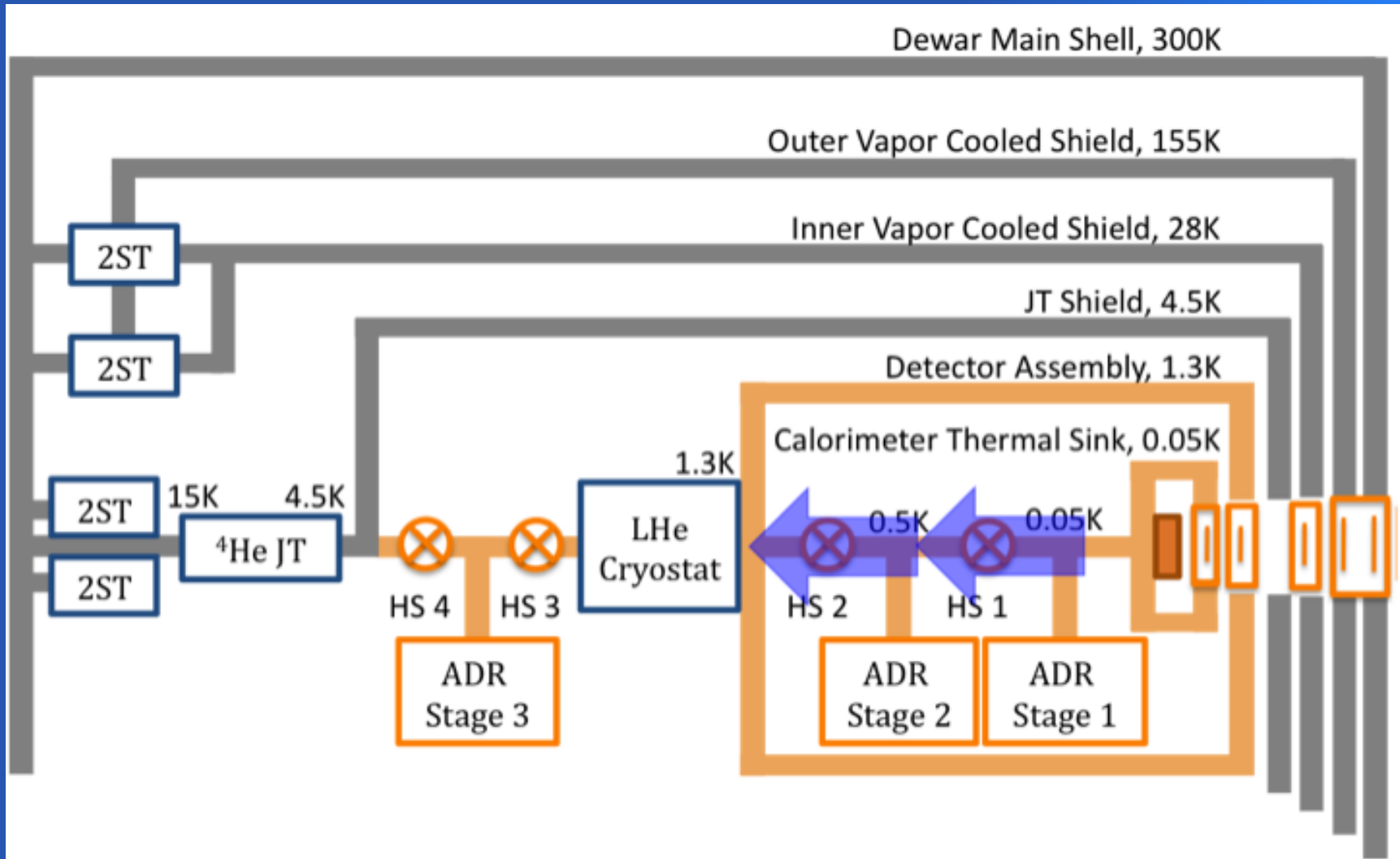
- 6x6 array of silicon microcalorimeters
  - High resolution imaging and spectroscopy of x-rays in the 0.2-13 keV band
- ADR for detector cooling
  - 50 mK operation



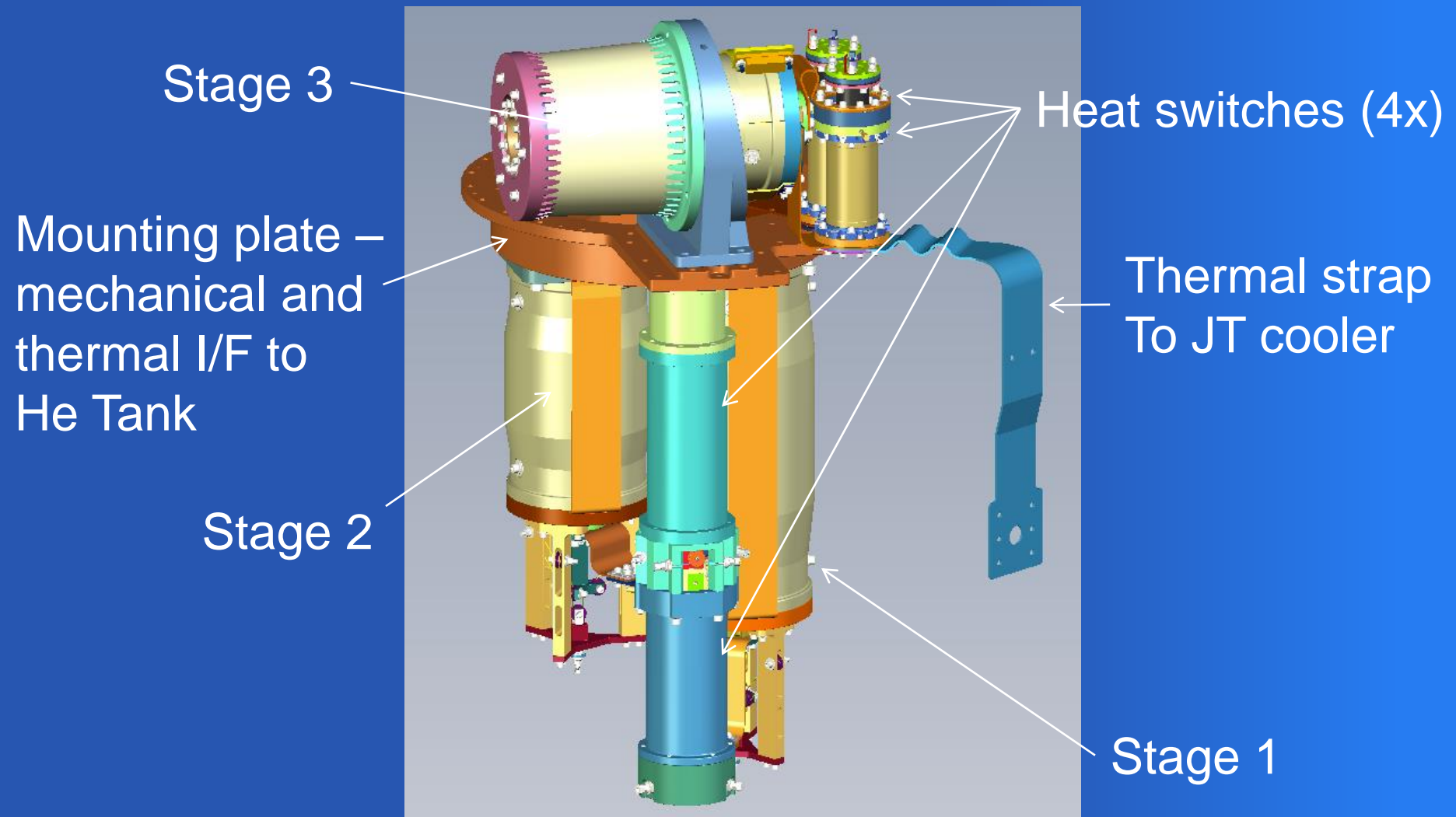
Low temperature  $\Leftrightarrow$  low heat capacity  
 $\Delta E \sim 4\text{-}5\text{ eV}$  for 0.2-13 keV x-rays

# Resolve Cryogenic System

2-stage ADR uses liquid helium as a heat sink



# SXS ADR Layout





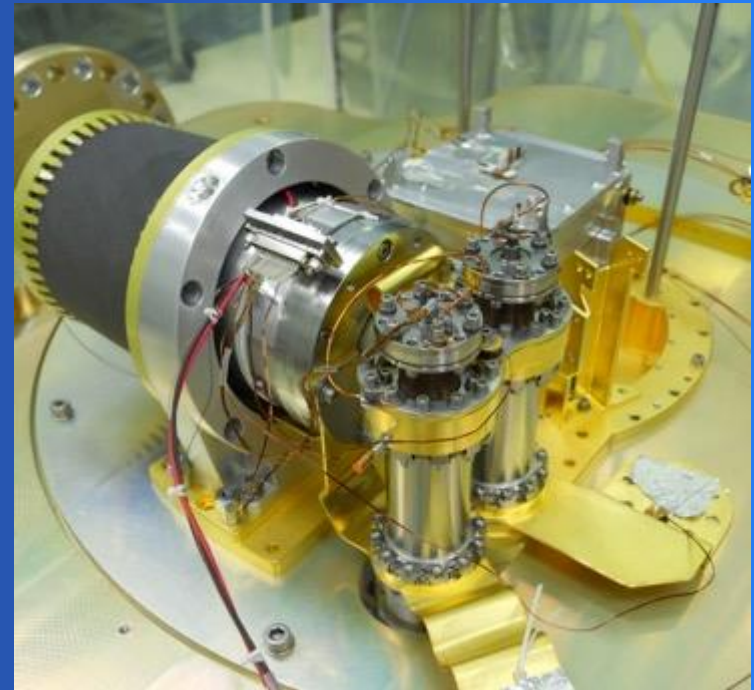
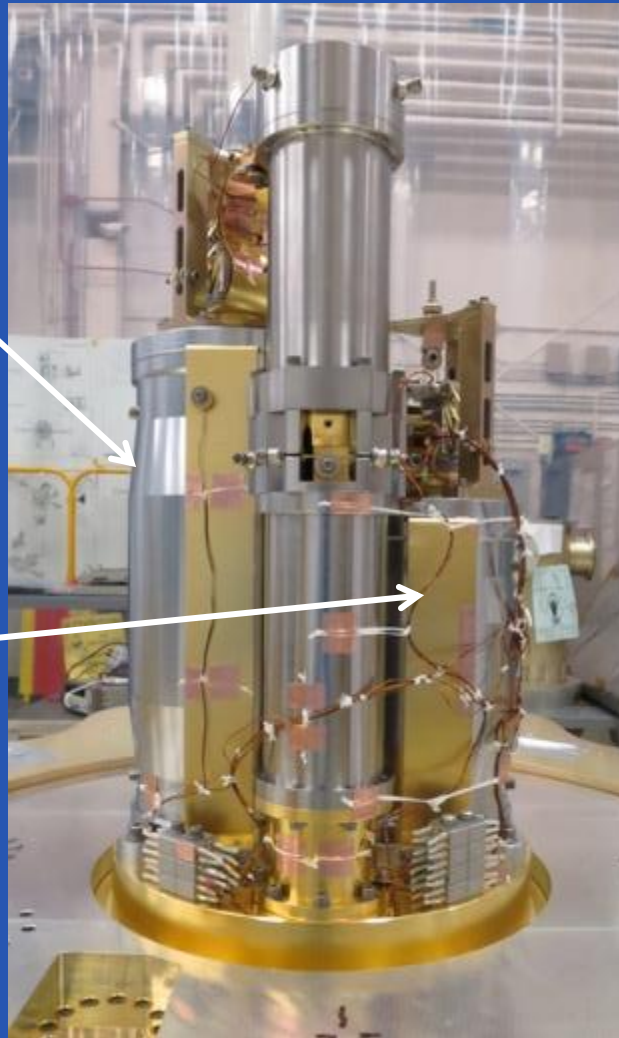
# ADR Assemblies

Stage 1:

- 270 g CPA
- 2 T, 2 amp magnet

Stage 2:

- 150 g GLF
- 3 T, 2 amp magnet

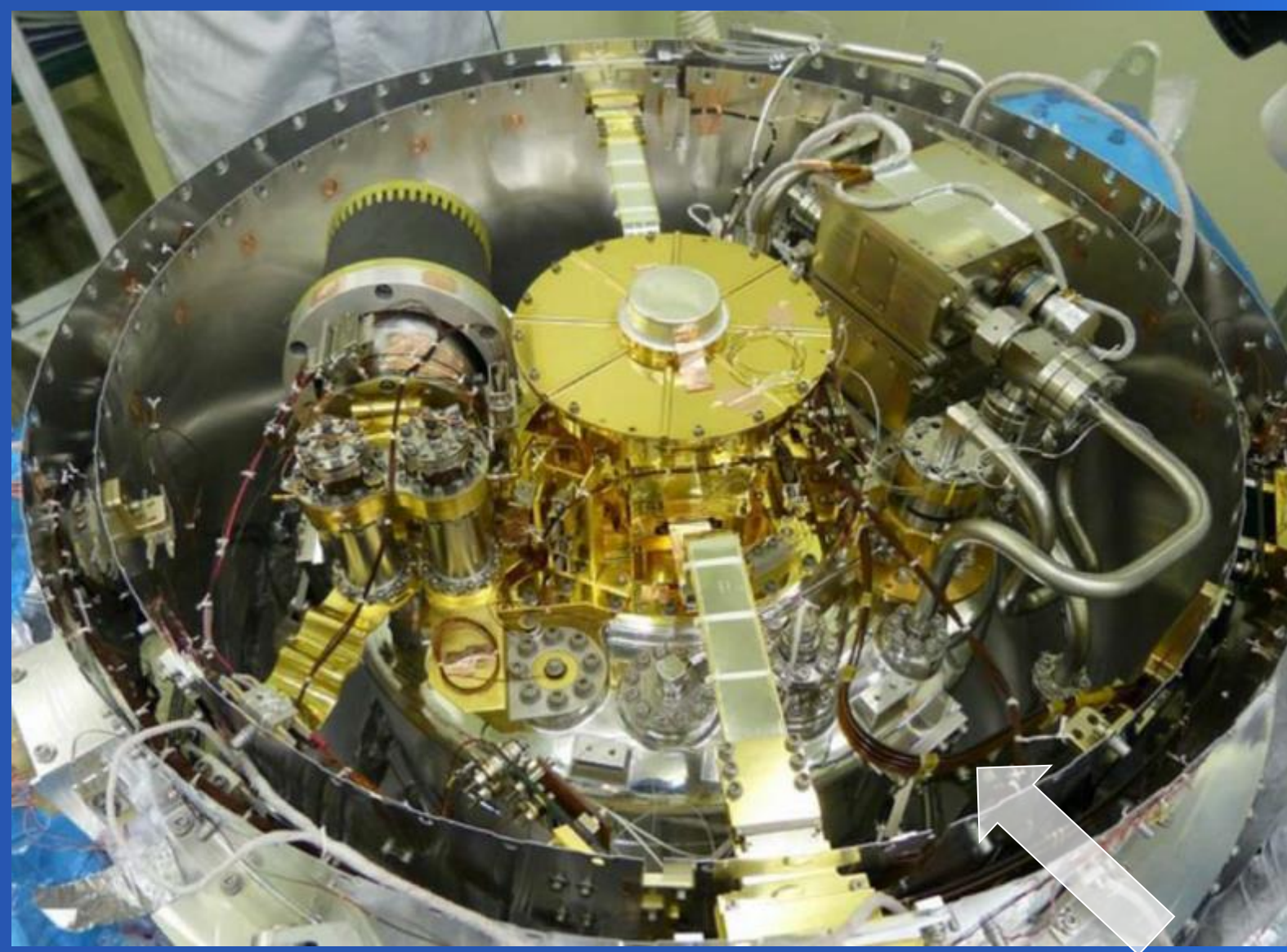


Stage 3:

- 150 g GLF
- 3 T, 2 amp magnet

Heat switches are active gas-gap

# Flight ADR, Detector and Dewar (April '14)



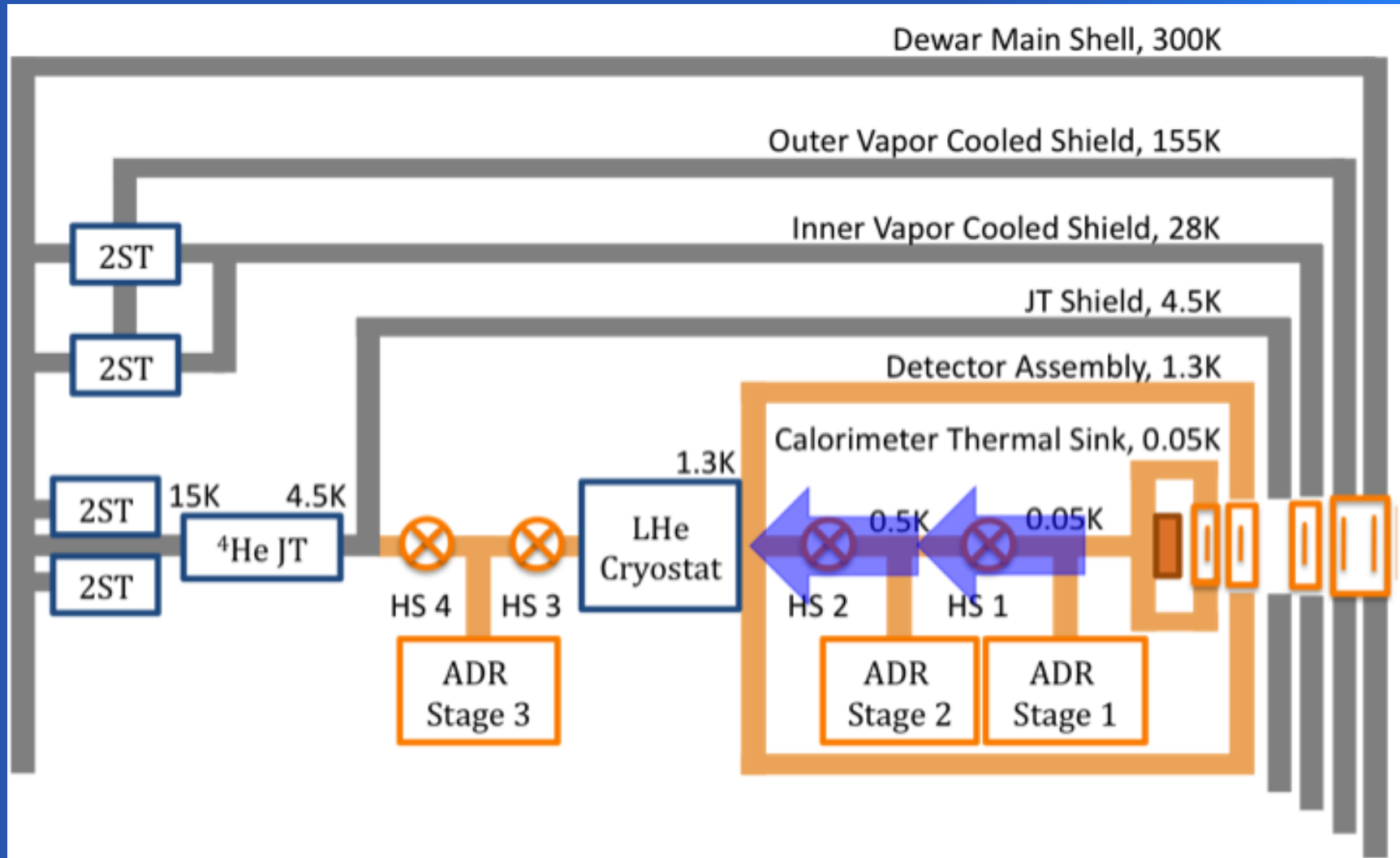
HTS leads:

YBCO  
tape

Supported  
by carbon  
fiber  
composite  
carrier

# ADR Operation with Liquid Helium

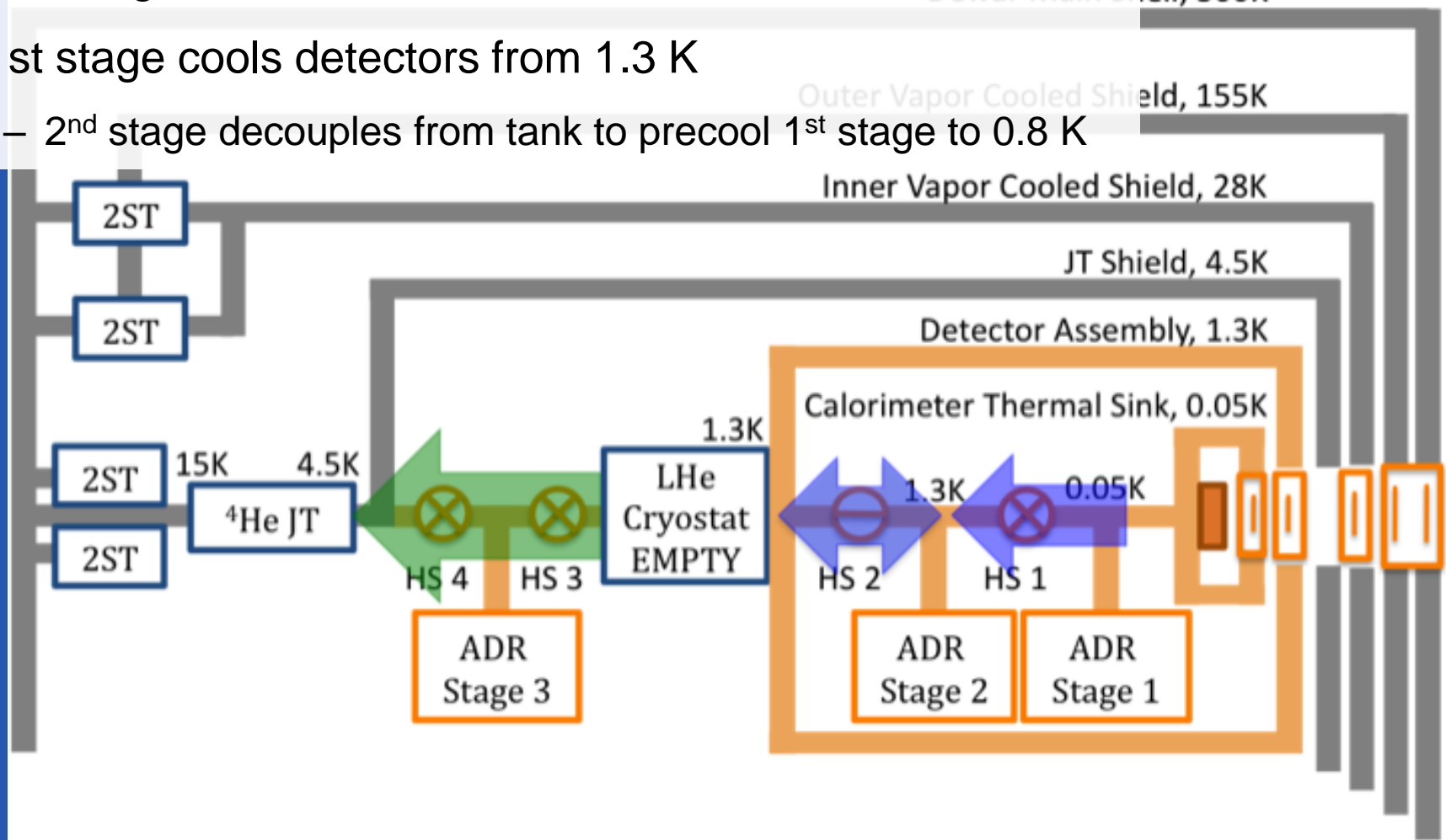
2-stage ADR uses liquid helium as a heat sink





# ADR Operation in Cryogen-Free Mode

- 3rd stage transfers heat from tank to JT cooler
- 2nd stage stabilizes helium tank at  $\sim 1.5$  K
- 1st stage cools detectors from 1.3 K
  - 2<sup>nd</sup> stage decouples from tank to precool 1<sup>st</sup> stage to 0.8 K

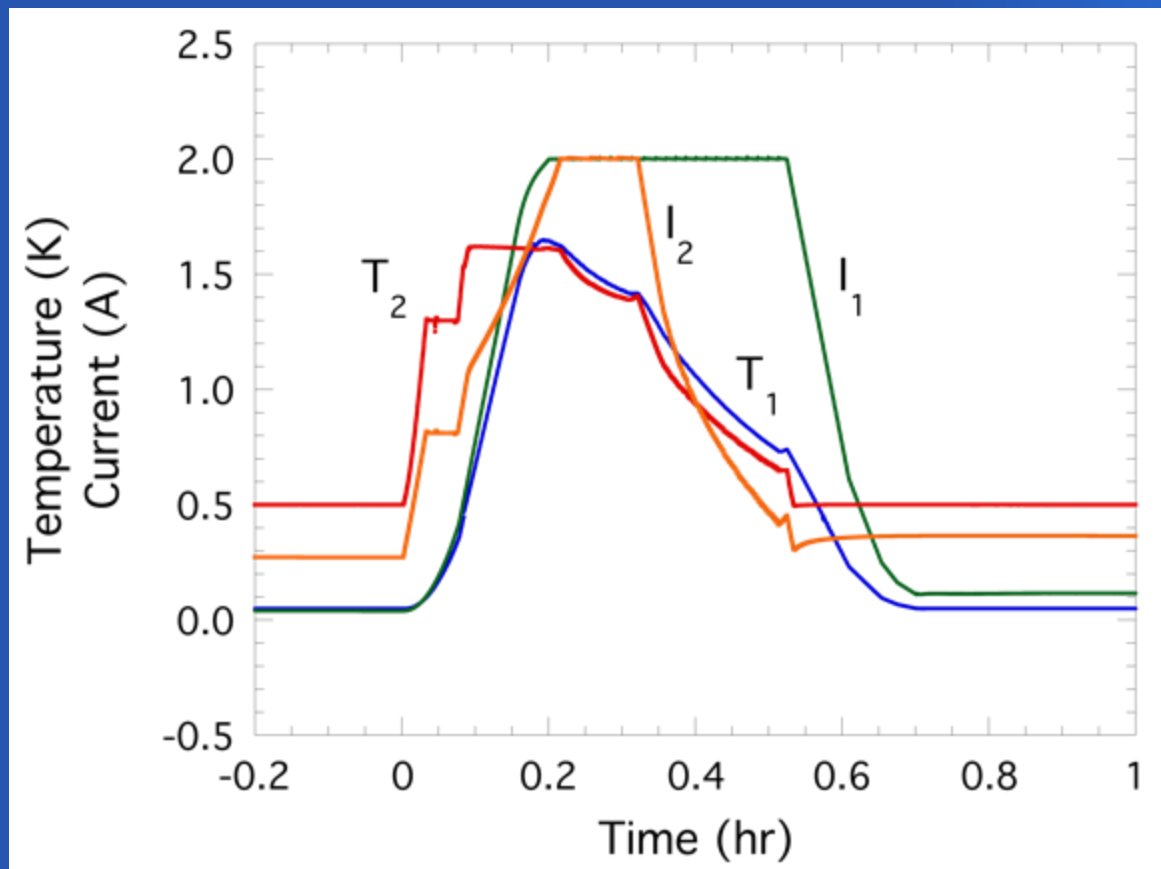


# SXS/Resolve Design and Operating Strategy

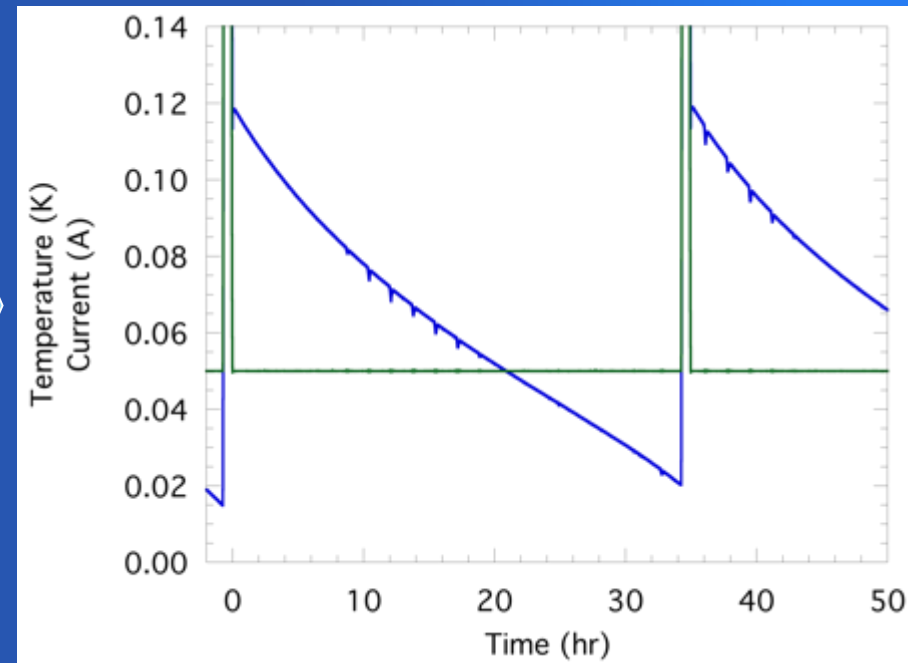
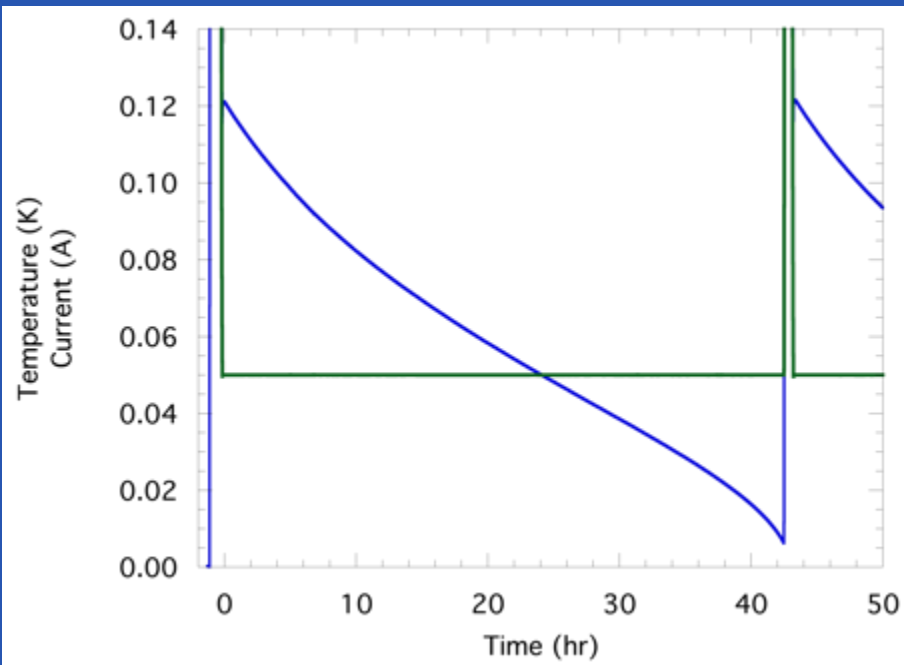
- Launch with minimum of 33 L of liquid helium ( $<1.3$  K)
  - Cryocoolers act as guards to intercept parasitic heat loads
  - Nominal lifetime  $>3$  years
- 2-stage ADR single-shot cools the detectors, rejecting heat to the liquid helium
- When helium is depleted, ADR operation changes to 3-stage operation

# ADR Recycling

- Recycling sequence was structured to minimize recycle time and to minimize  $I^2R$  heating of the IVCS
  - JFET amplifiers are thermally coupled to IVCS



# Stage 1 – Ground vs On Orbit



Recycle time ~45 minutes

Heat load 0.80  $\mu$ W

Hold time\* ~42 hours

~45 minutes

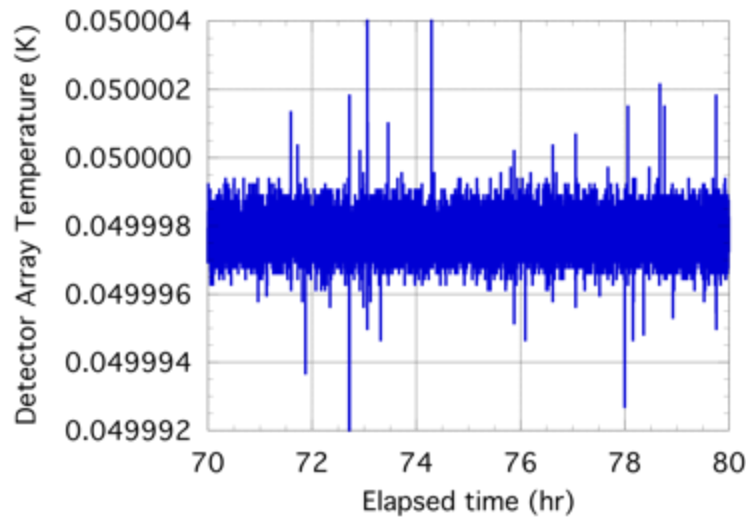
0.86  $\mu$ W

~39 hours

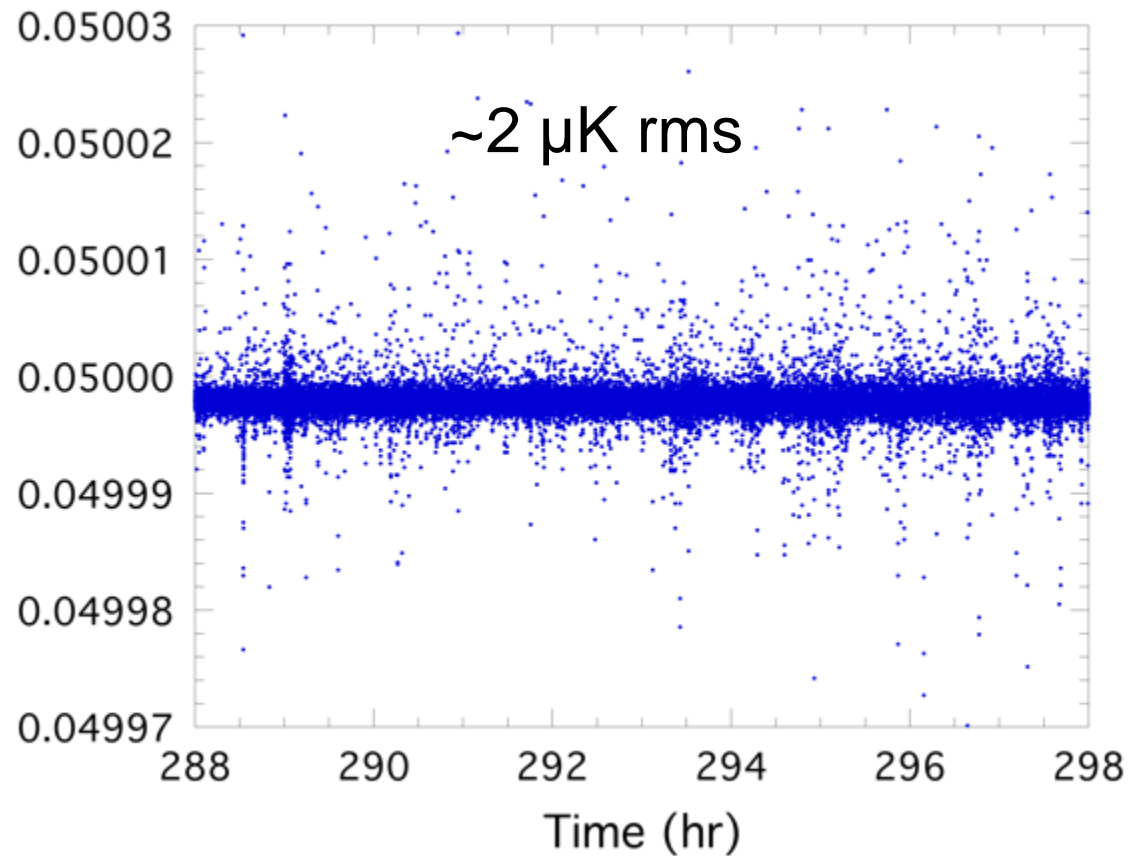
\* Dependent on tank temperature; values are for ~1.20 K



# Temperature Stability



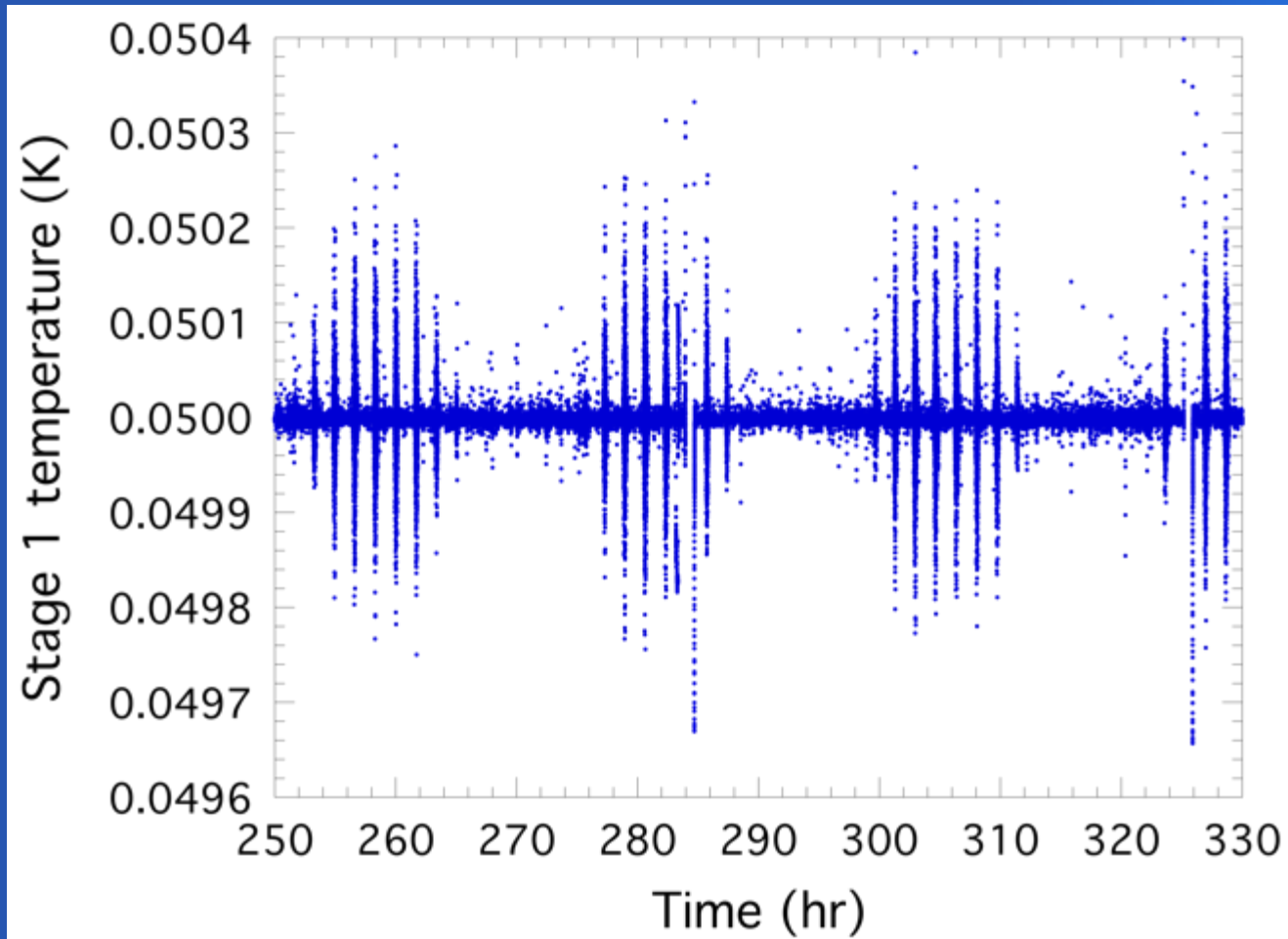
Ground tests:  $\sim 0.4 \mu\text{K rms}$



On orbit:  $\sim 2 \mu\text{K rms}$

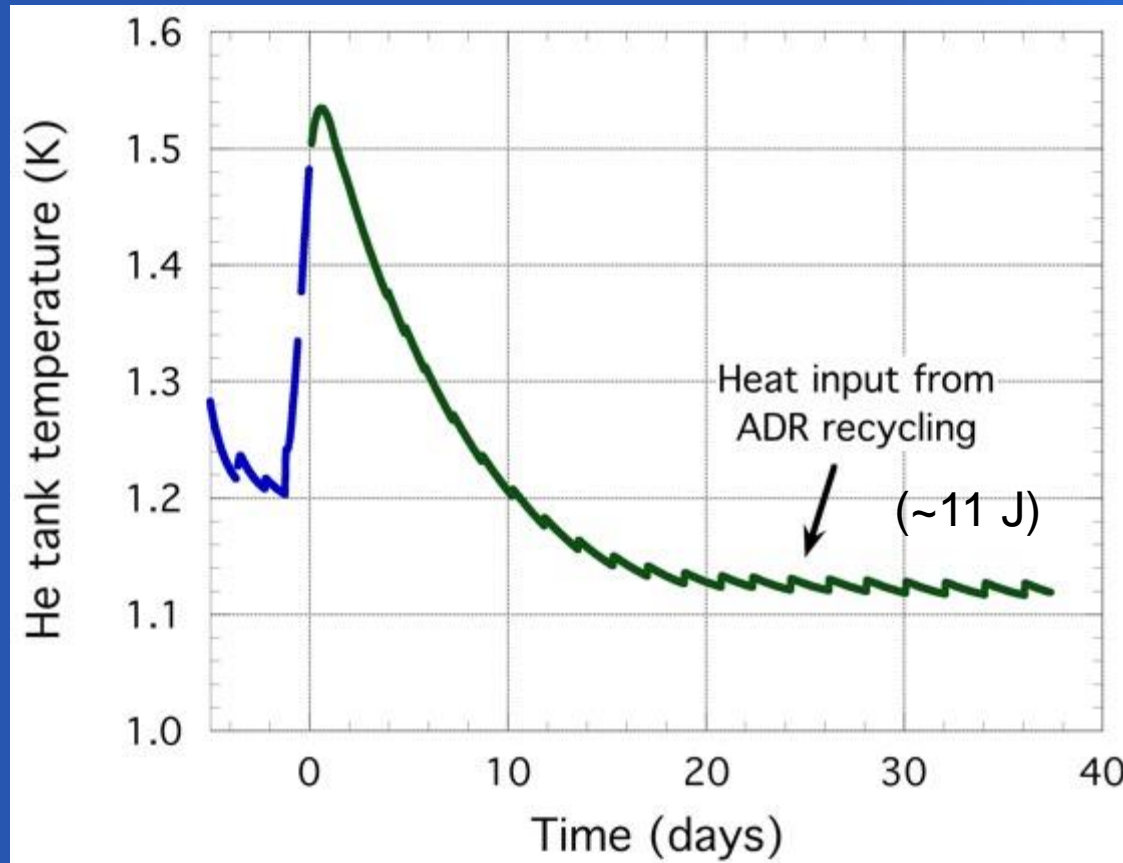
# Effect of High Energy Particles

- Effect concentrated in SAA
- Response appears to be primarily within thermometers



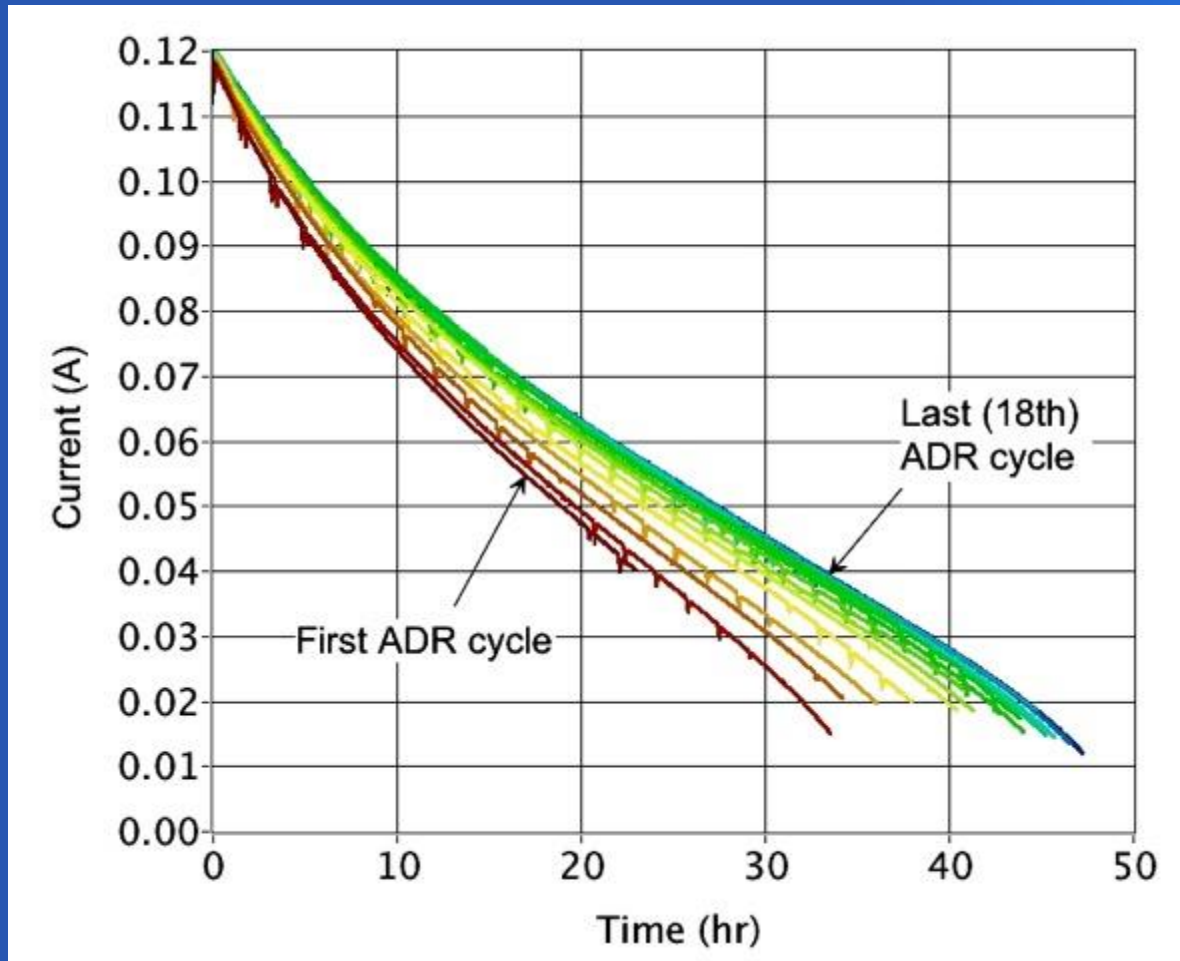
# He Tank Temperature

- Cooling rate is consistent with ground tests of porous plug phase separator
- Steady-state temperature was  $\sim 1.12$  K



# ADR Performance

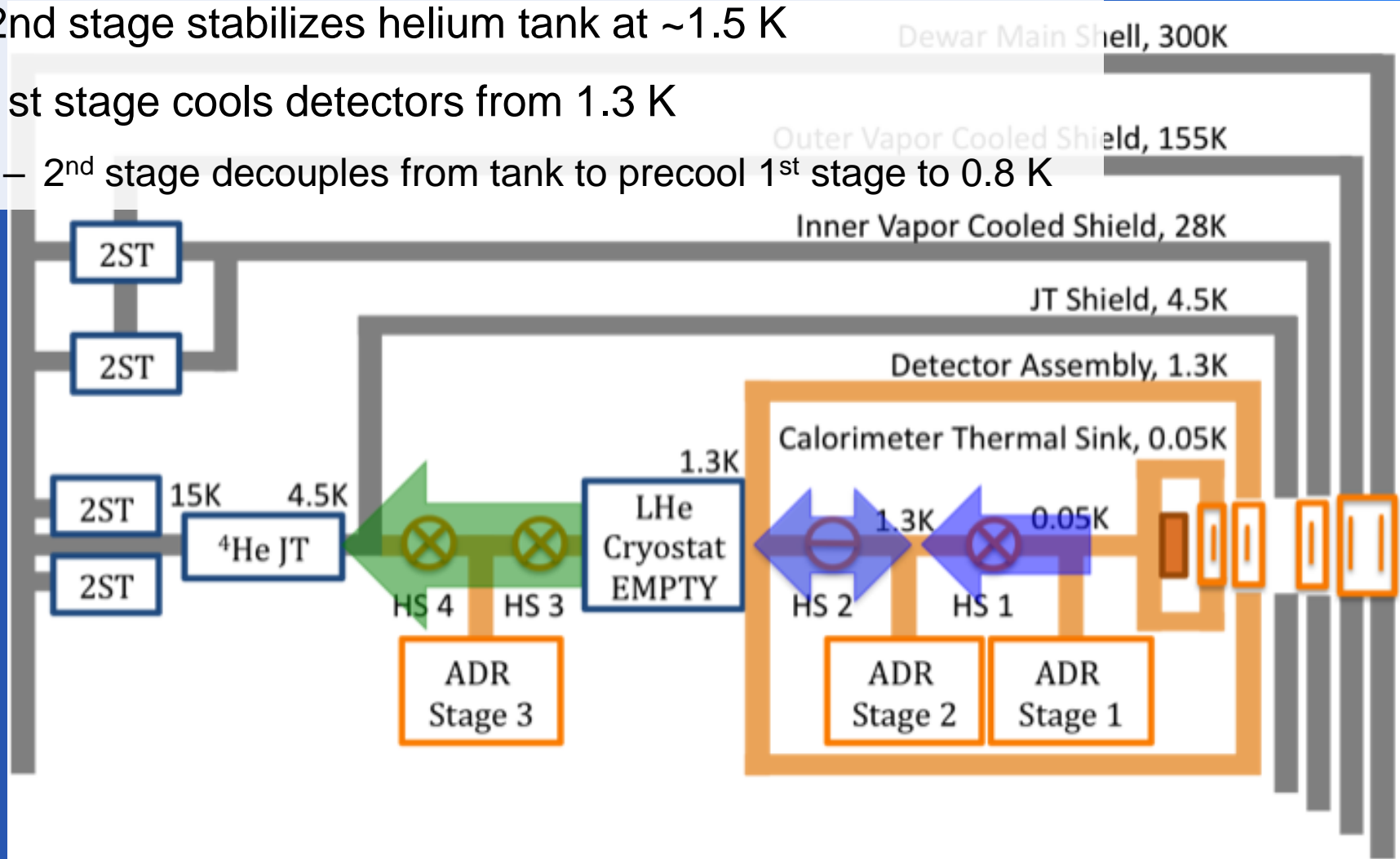
- ADR hold time steadily increased as He tank cooled
- At 1.12 K, hold time was ~48 hours





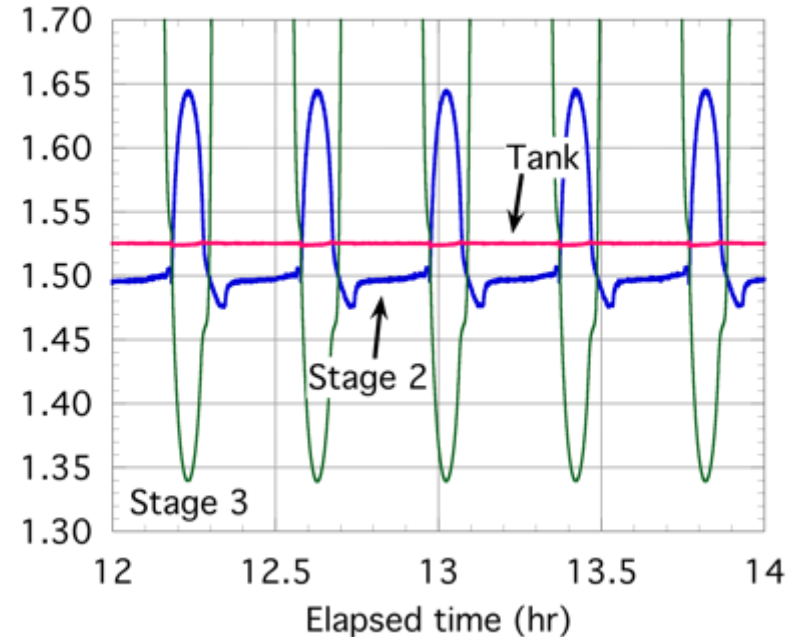
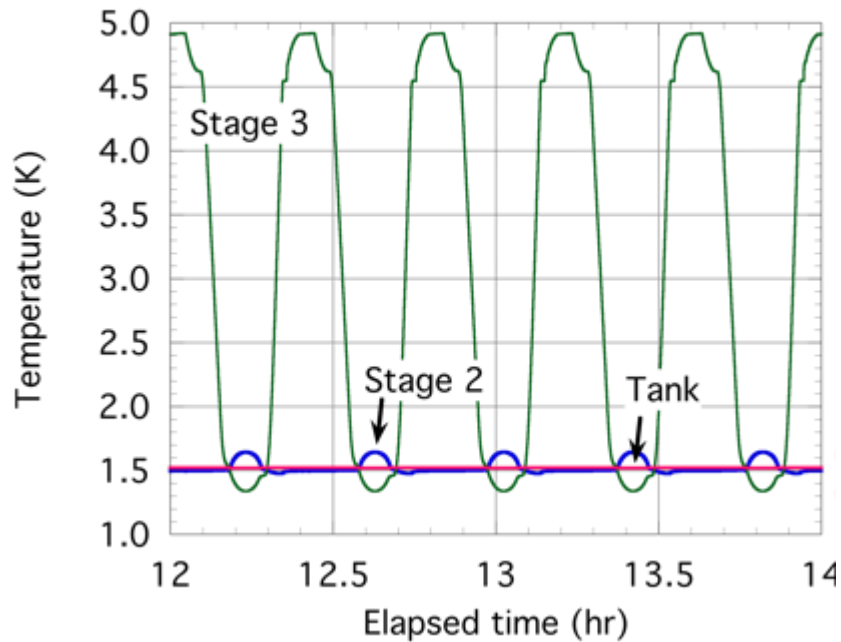
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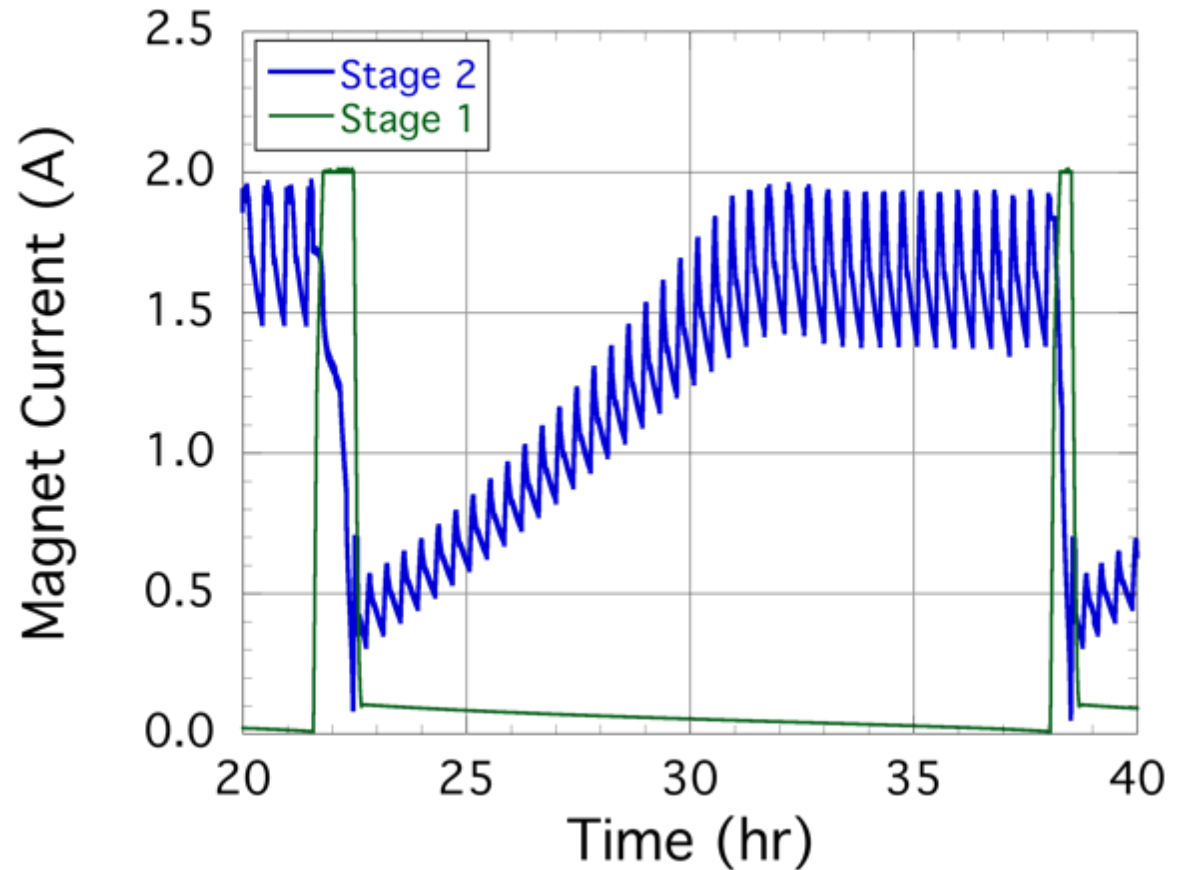
# Continuous ADR Cooling

- Stages 2 and 3 operate together as a continuous ADR to cool the helium tank



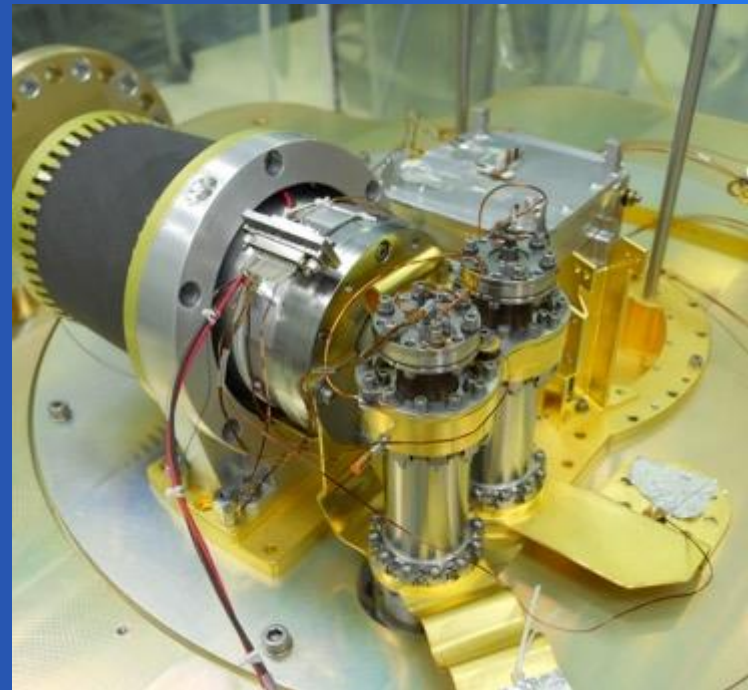
# Full Cycle with He tank at 1.525 K

- S1 is automatically recycled when current falls below 10 mA
- 45 minute recycle
- 15.0 hour hold
- >94% observing efficiency



# Changes for Resolve

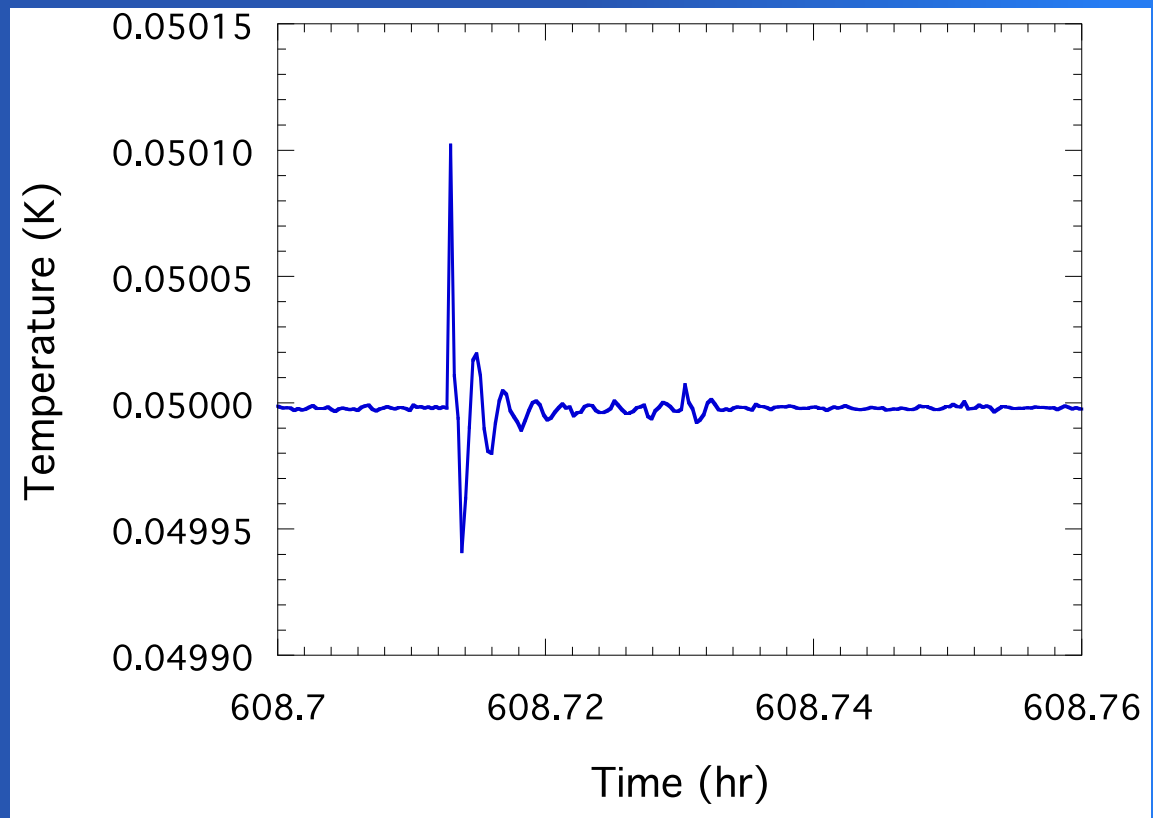
- Increase magnetic shielding of ADR Stage 3
  - Interactions between the detector array and the stage required limiting its current to 1.75 A, which reduced the ADR's cooling power
  - 1.5 mm increase in magnetic shield thickness (~200 grams) reduced fringing fields by an order of magnitude
  - Cryogen-free operation is (hopefully) improved by maintaining the helium tank at lower temperature





# Changes for Resolve

- Reduce ADR controller response to cosmic ray events
  - Suppress response to a reading outside a narrow band at 50 mK



# Summary

- XRISM received final approval for funding by JAXA in mid-2018
  - Launch is currently scheduled for early 2022
- The Resolve instrument is essentially a built-to-print copy of SXS
  - Limited number of changes based on experience with ground and on-orbit performance
  - Challenge: SXS was not fully demonstrated on orbit
    - Gate valve was not opened to expose the detectors to full energy range of x-rays
    - Long-term performance (cryocoolers, progressively smaller LHe volume)
    - Environmental effects: micrometeoroid damage, residual helium