31st Congress of the International Council of the Aeronautical Sciences

Belo Horizonte, Brazil; September 09-14, 2018



Key Performance Parameters for Future NATO Hybrid/Electric Military Air Vehicles

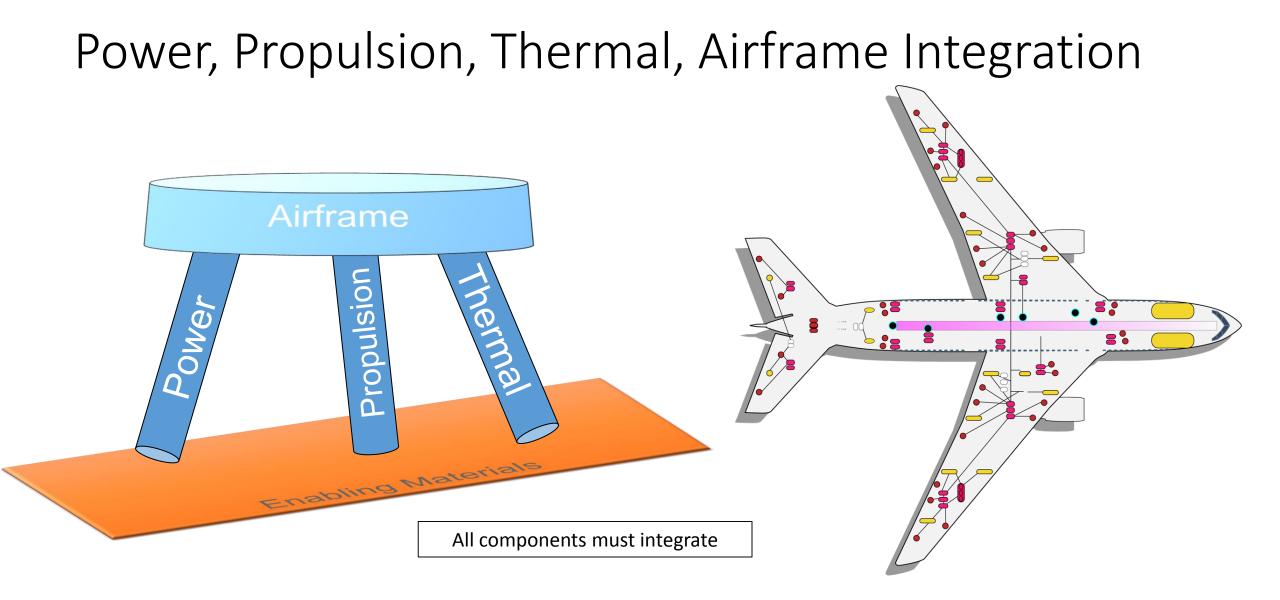
September, 2018

Dr. Rodger Dyson Hybrid Gas Electric Propulsion Technical Lead NASA Glenn Research Center



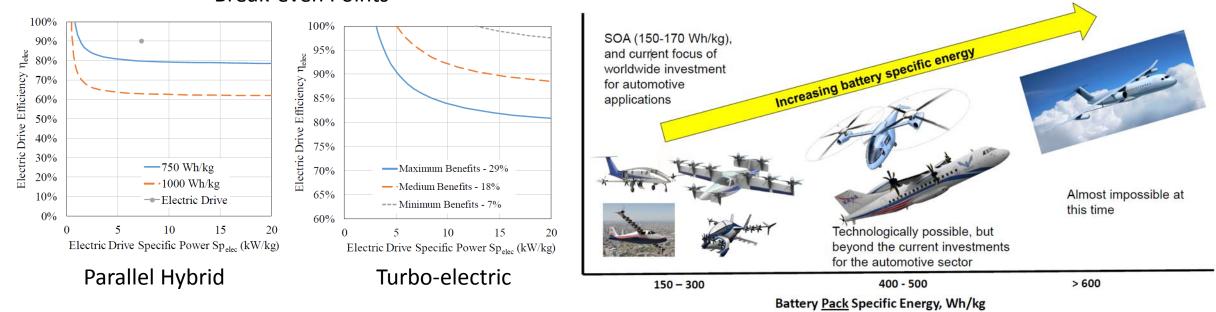
Rodger.W.Dyson@nasa.gov





Basic Technology Performance Requirements

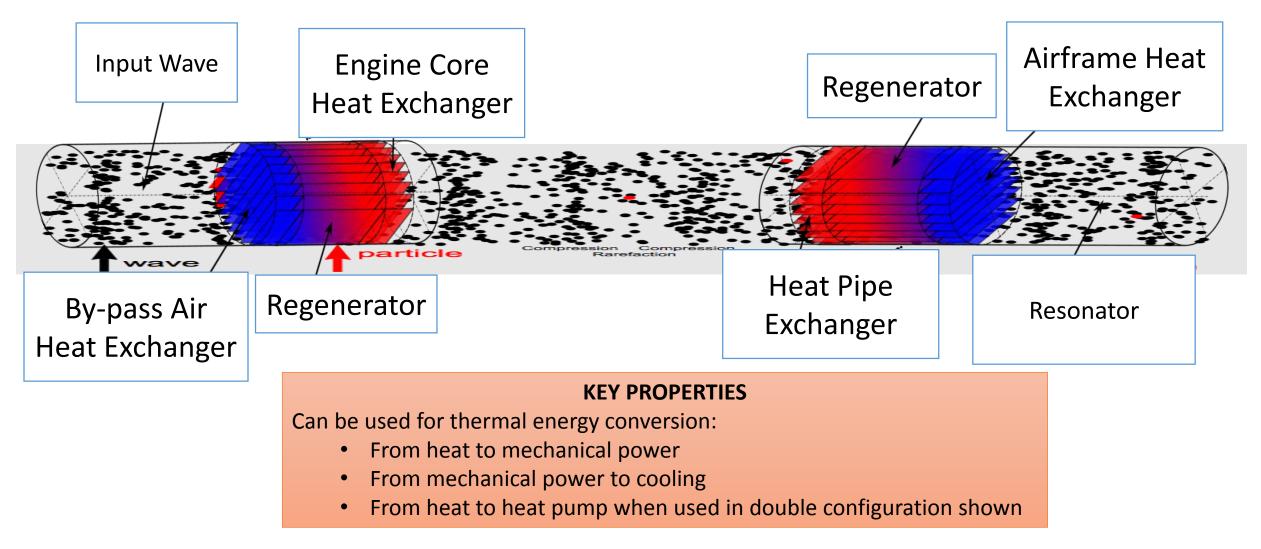
- Urban e-VTOL Air Taxi Battery Requirements:
 - 150 kWh, 300 Wh/kg,
 - 500 cycles by 2023, 1000 cycles by 2028, 2000 cycles by 2032
 - Peak charge time: add about 20% capacity in five minutes
- Transport Class Requirements:



Evolution of All-Electric and Hybrid Electric Aircraft

Break-even Points

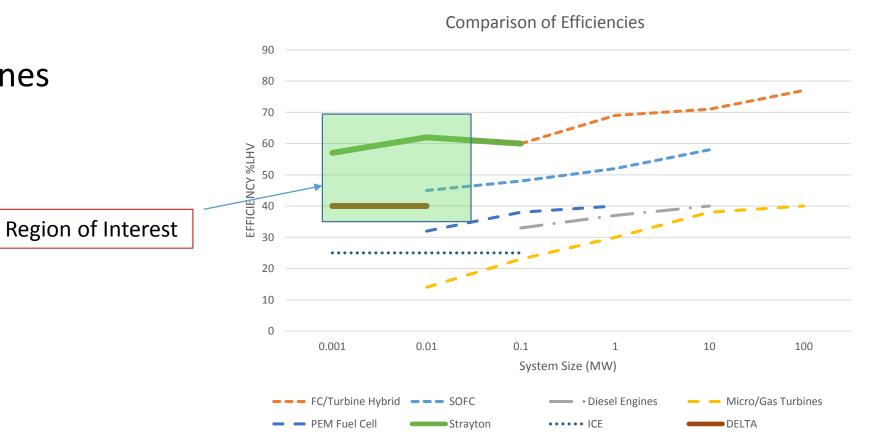
Basic Building Block for Electric Aircraft: Thermo-Acoustic Engine and Heat Pumping



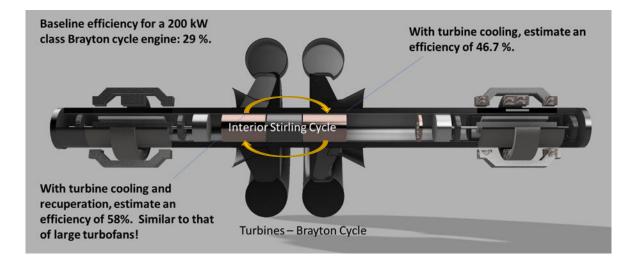
Power Options

- FC/Turbine Hybrid
- PEM/SOFC
- Micro/Gas turbines
- Diesel
- ICE
- DELTA
- Strayton

Technology	Net System AC Power (kW)	Net Fuel LHV to AC Electric Power Conversion Efficiency	Full Production Equipment Manufacturing Cost \$ per W	System Maintenance Cost \$/kwh	System Availability Percent	System Life (yrs)
Ideal	>100	>70	<0.9	0.02	>95	>20
SOFC-GT	>100	>70	<4	<1	>95	>5
Strayton	>100	>50	<0.05	< 0.02	>95	>20
Fuel Cell	>100	>50	<5	<1	>95	<5
μ-Turbine	<300	>20	<1	<0.1	>95	<5
ICE OTTO	>100	>25	<0.05	<0.03	>95	<10
Gas Turbine	>1000	>40	<0.5	<0.1	>95	<5

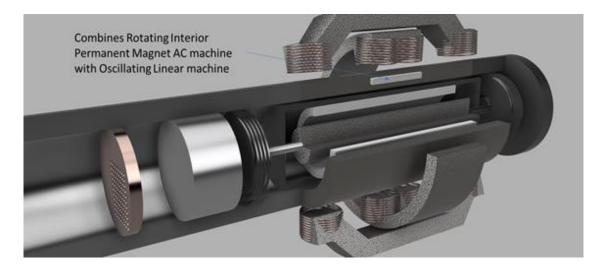


Stirling and Brayton (Strayton) Engine Genset



Key Features

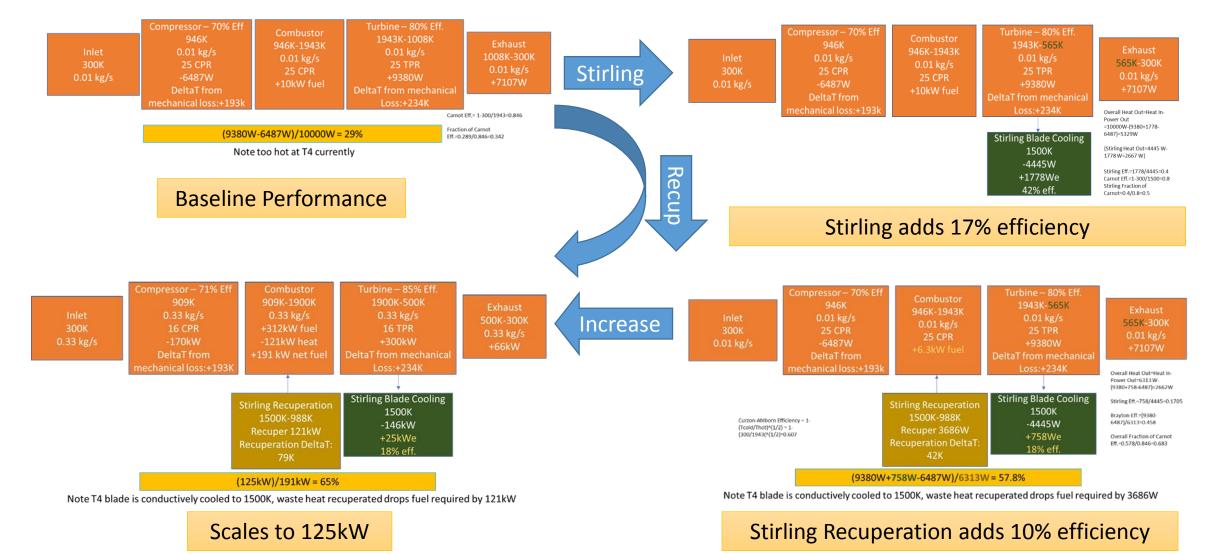
- Combines Stirling and Brayton cycles synergistically for <2MW high efficiency and specific power
- Provides both topping and bottoming cycles using a Brayton and Stirling cycle (both are top and bottom!)
- Achieves recuperation without a recuperator
- Naturally cools the turbine blades
- Power is extracted via rotating and oscillation
- Fuel Flexible with high turn-down ratio



Key Features

- High-speed Brayton cycle and internal Stirling cycle use no-maintenance air and flexure bearings
- Power balancing between cycles via direct control
- No contact rotating bearings and power transfer
- High speed rotation enable short conductive blades
- No hot moving Stirling cycle parts
- Leverages recent HEMM work for flexure stiffness
- Pedigree from previous DOE/Reliance Electric 2008

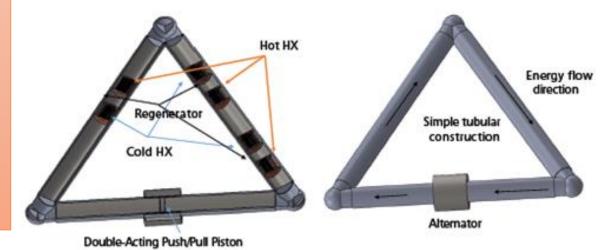
Thermodynamic Efficiency Step-by-Step



Double-Acting Extremely Light-Weight Thermo-Acoustic Generator (DELTA)

Key Features

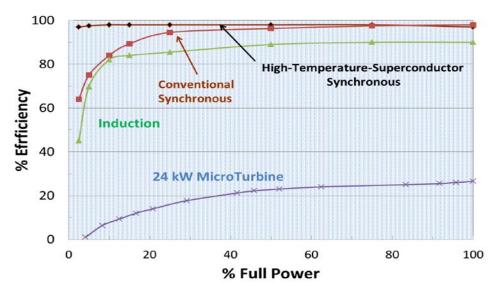
- Utilizes multistage high frequency thermo-acoustics
- Uses a double-acting piston and engine reactive power to minimize required spring
- Fuel flexible including cryogenic
- Shape flexible for embedding in unusual locations
- Silent operation
- Higher efficiency and comparable specific power with ICE

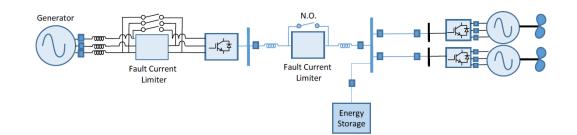


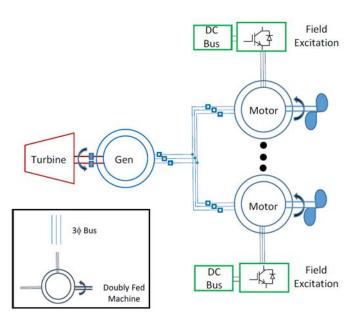
Provides silent power for APU and UAV applications

Propulsion Options

- Fully Superconducting
- Partially Superconducting
- PM Synchronous
- Induction
- Double-fed



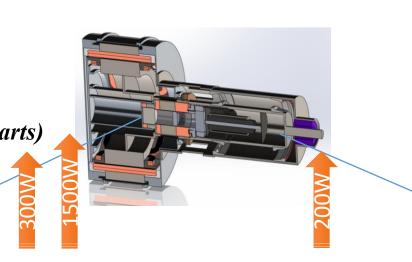




HEMM w/Embedded Cryocooler

HEMM is designed to operate as

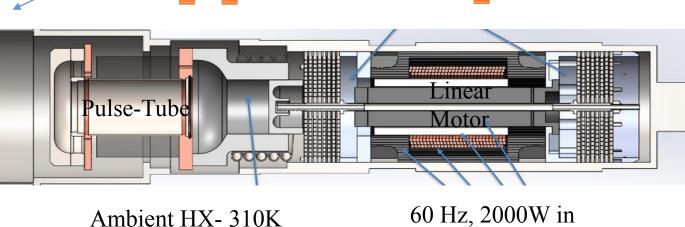
- a *1.4 MW* motor
- with *direct drive*
- High torque/low speed
- >98% efficient
- >16 kw/kg (active E-M parts)



Cryocooler Key Features:

- Cool superconducting rotor
- Fit inside rotating motor
- Integrates cooler and linear machine
- Operate rotating or stationary
- No cold moving parts

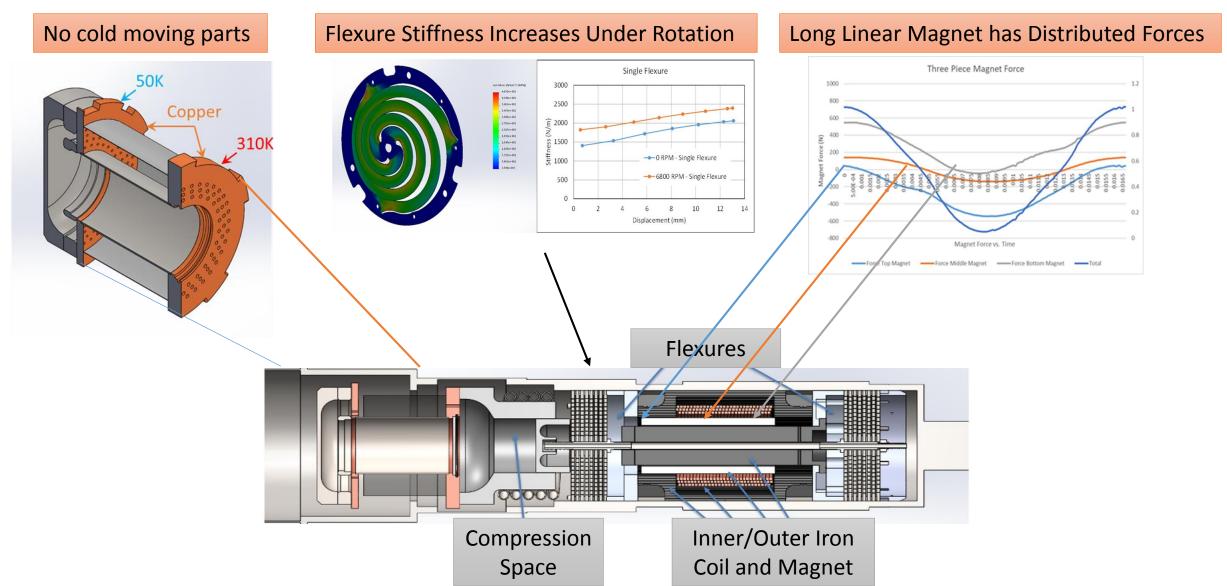
Cold Tip-
55W/50K
55W



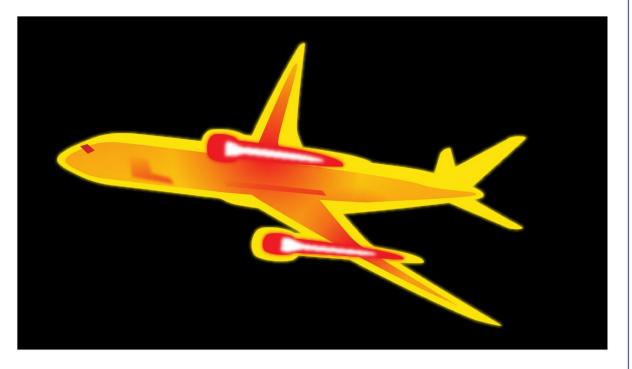
Top Level Parameter	Value		
Frequency	60 Hz		
Gas	Helium		
Pressure	6.2 MPa		
Heat Lifted @ 50K	55W		
Heat Rejected	2000W		
ElectricalIn	2000W		
Mechanical PVin	1661W		
Coil Current Density	4 A/mm2		
Piston Amplitude	1.3 cm		

Superconducting inside the motor and provides Strayton risk reduction

Linear Machine and Pulse-Tube Cooler

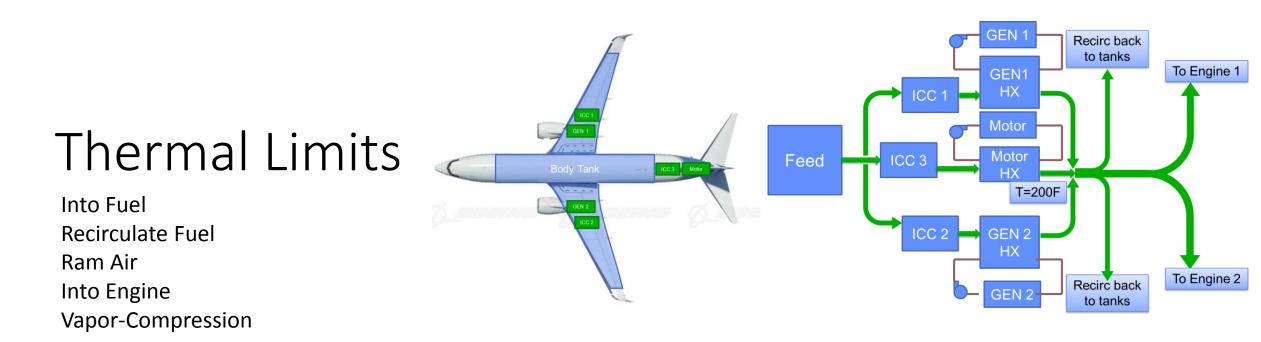


Thermal Options

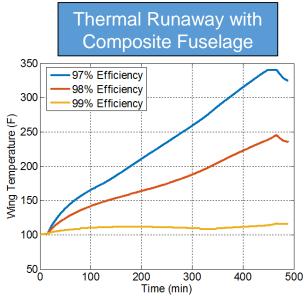


Current proposed solutions (and limits) include: Ram air HX

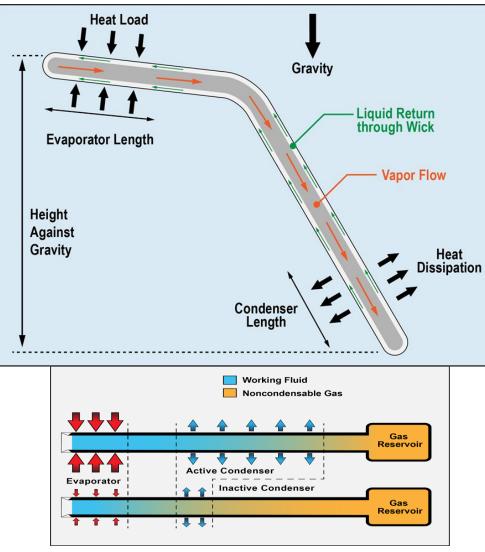
- adds weight and aircraft drag
- Convective skin cooling HX
 - adds weight, drag, and inefficient
- Dumping heat into fuel
 - limited thermal capacity
- Dumping heat into lubricating oil
 - limited thermal capacity
- $\hfill\square$ Active cooling
 - adds weight and consumes engine power
- Phase change cooling
 - adds weight and limited thermal capacity
- Heat pipe, pumped multiphase, vapor compression
 - adds weight and consumes engine power



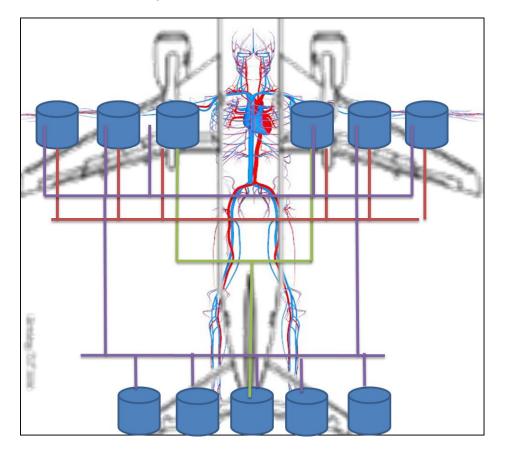
	1% H	ot Day	Standard Day		
	Total Penalty	Total Penalty	Total Penalty	Total Penalty	
	(zero exit	(non-zero exit	(zero exit	(non-zero	
	Velocity)	velocity)	Velocity)	exit velocity)	
900NM	4.98%	3.31%	2.76%	2.36%	
3500NM	5.00%	3.62%	3.01%	2.57%	



Variable Conductance Heat Pipe



Solid-state Heat Transfer Switching

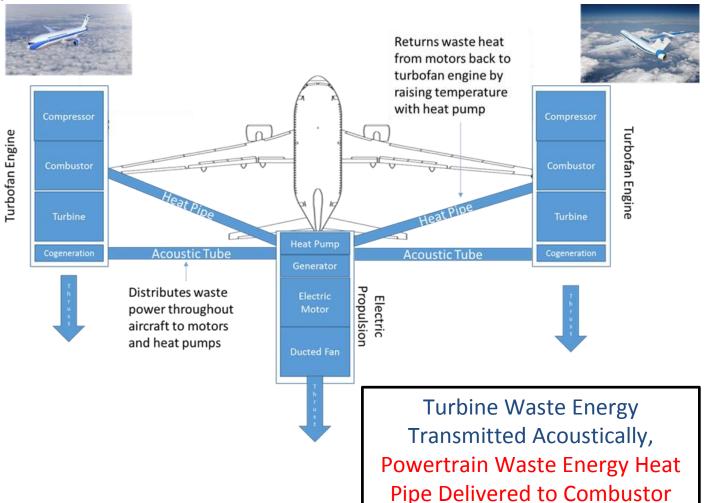


Acoustic and Heat Pipe Tubes Embedded in Airframe

TREES – Thermal Recovery Energy Efficient System Complete Cycle

Solid-state (no moving part) energy recycle and control

 Localized skin heating for active lift/drag management, de-icing, powertrain cooling, cabin management, and military cloaking



Integrated Benefit

Strayton Engine

- High efficiency/specific power
- Integrated thermal conversion

DELTA

- High efficiency/specific power
- Quiet

TREES

- Distributes waste heat with waste heat
- Saves fuel
- Improves aerodynamics
- Naturally de-ices

HEMM Motor

- High efficiency/specific power
- Integrated thermal conversion

Advanced Integration is Required at Component and System Level

Conclusion

- Maximum benefit with NATO electric aircraft is achieved by integrating at both the component level and system level.
- Thermal Energy Conversion technologies provide the fundamental building block for this integration.
- **HEMM** motor provides flight-weight high efficiency at high power
- Strayton engine provides flight-weight high efficiency at medium power
- **DELTA** engine provides flight-weight high efficiency at low power
- **TREES** enables the tight integration of all these technologies at the vehicle level.