Computational Schlieren with Comparisons to Experiment

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Motivation and Previous Work

- Motivation
 - -interest in additional means of comparing experiment and CFD results -specifically, we would like to compare RBOS results with the schlieren based on CFD solutions, CFD schlieren

 - -working towards more precise comparisons, more than side-by-side
- Previous Work

 - –Yates (1993), Images Constructed from Computed Flowfields -Svakhine et al. (2005), line-of-sight rays, shadowgraphs, schlieren -Brownlee et al. (2011), refracting rays, rendering at interactive rates, GPU -Settles & Hargather (survey of some relatively recent work, 2017)





Our CFD Schlieren Application: raycast

- Raycast produces images by casting rays through a CFD domain, integrating a user-defined scalar field along the ray
- Raycast uses straight, line-of-sight rays (*i.e.*, no refraction)
- Projection can be orthographic or perspective, arbitrary direction
- The user provides a scalar field equation, based on solver fields
 - -the equation is parsed at runtime (*i.e.*, not hard coded)
 - -operators such as grad, sqrt and mag are available
- We use the Embree library from Intel in order to build a spatial data structure that significantly accelerates ray queries





Ray-Integration Formulae

- Let the x and y coordinate axes correspond to the horizontal and vertical axes, respectively, relative to the viewer
- The incremental change in the angular deflection:

$$\delta \varepsilon_x = \frac{1}{n} \, \frac{\partial n}{\partial x}$$

-where *n* is the index of refraction

• The index of refraction, in terms of density (Gladstone-Dale):

$$n = \kappa \rho + 1$$

• Rewriting the deflections in terms of the gradient of density:

$$\delta \varepsilon_x = \frac{\kappa}{\kappa \rho + 1} \, \frac{\partial \rho}{\partial x}$$



 $\delta \varepsilon_y = \frac{1}{n} \, \frac{\partial n}{\partial y} \tag{1}$

$$\delta \varepsilon_y = \frac{\kappa}{\kappa \rho + 1} \, \frac{\partial \rho}{\partial y} \tag{3}$$



(2)

LAVA / Structured-AMR Schlieren



- an Immersed Boundary Method. AIAA 2016.

 Solver: Launch, Ascent and Vehicle Aerodynamics (LAVA) framework • C. Brehm et al. Open Rotor Computational Aeroacoustic Analysis with

LAVA / Overset-Curvilinear Schlieren

- Structured Overset Grids. AIAA 2017.

 Solver: Launch, Ascent and Vehicle Aerodynamics (LAVA) framework • J. Housman et al., Jet Noise Prediction using Hybrid RANS/LES with

Overflow Schlieren

Space Launch System (SLS) booster separation simulation

Mesh: adaptive, overset curvilinear mesh, ~255M vertices

booster separation simulation ilinear mesh, ~255M vertices

Overflow Schlieren, SLS Super-Sized

NASA Ames hyperwall 2: (16
Approximately 0.25B pixels

• NASA Ames hyperwall 2: (16 x 1600) x (8 x 1200) = 25600 x 9600

AirBOS vs. Cart3D Schlieren

AirBOS Image

• Image courtesy of M. Aftosmis.

RBOS vs. LAVA / Polyhedral Schlieren

RBOS

Supersonic nozzle-plume study, Mach 1.6.

RBOS vs. LAVA / Polyhedral Schlieren

RBOS

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 Supersonic nozzle-plume study, Mach 2.0.

Alternate Views: e.g., Looking Up From Below

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RBOS / CFD Schlieren Comparison by Overlay

- Rather than view side-by-side, we want to overlay the experiment and CFD results in order to create a more precise comparison
- Requires image registration
- The model is deflected slightly by wind conditions –we have photogrammetry data measuring the deflection
- The model vibrates slightly, RBOS images produced by averaging
- The camera used for RBOS may move slightly, but we do not have measurements of that
- We chose 3 corresponding regions in the RBOS and CFD schlieren, match by cross correlation, and solve for affine transform

Registering RBOS with CFD Schlieren

Registration implemented using OpenCV library.

• Green: CFD model outline, blue: target regions, red: match in RBOS

Displaying Registered Results: Color Channels

- Where the RBOS and CFD schlieren differ, we want to see whether the RBOS intensity is greater than the CFD schlieren, or vice versa
- We use the red channel for RBOS
- We use the green channel for CFD schlieren
- We assign the average of the red and green channels to blue
 - -also considered minimum, maximum
 - -areas where RBOS and CFD schlieren match: gray scale
 - -areas where CFD schlieren is higher intensity: green/cyan
 - -areas where RBOS is higher intensity: red/magenta
- Primary interest is in seeing how well the shocks match

RBOS vs. CFD Schlieren by Color Channels

M1.6

RBOS vs. CFD Schlieren: M1.6, V

- Implement imaging based on refractive rays -would be slower, but much of the infrastructure would remain the same
- Further image comparison support, more automated registration
- More mesh types
 - –FUN3D unstructured is the format most requested for us
- Work with native domain decomposition of solver
 - -for high-resolution simulations, more efficient than re-composing into single large mesh object

-preparing for very large runs on Summit, 1B+ vertices, unstructured

Future Work

- We have implemented a new ray-caster renderer: raycast
 - -capable of integrating through custom, user-defined fields
- -so far mostly used to produce schlieren based on density gradient magnitude Works with a variety of CFD solution types -overset curvilinear grids
 - -polyhedral unstructured
- -structured adaptive mesh refinement (AMR), unstructured AMR Comparisons with RBOS look very good
- We are continuing to investigate more precise means of comparing **BOS and CFD schlieren results**

Conclusion

