



Total Temperature Measurements Using a Rearward Facing Probe in Supercooled Liquid Droplet and Ice Crystal Clouds

Juan H. Agui, and Peter M. Struk.

NASA GRC

And

Tadas P. Bartkus

Ohio Aerospace Institute

10th AIAA Atmospheric and Space Environments Conference

Atlanta, GA

25–29 June 2018



Outline

- » Background
- » Probe & Facility
- » Thermal model
- » Test Campaign
- » Results
- » Conclusions



Background and Motivation

- **Engine Icing**
 - Performance loss: rollback, surge, flameout, and even internal engine damage
 - Partial melting and refreeze of ice inside engine core (Mason et al., 2006)
 - Ingestion of ice crystals and aggregates, mixed-phase droplets, or supercooled liquid droplets
 - Need to better understand the conditions and properties that lead to engine icing.
- **Simulation and analysis (physical and computational, and modeling)**
 - Test facilities (PSL, NRC, ...)
 - Thermal and computational models and analysis
- **Probes**
 - Multiple probes (aerothermal probes and ice cloud characterization probes and techniques)
 - Total temperature
 - Traditional total temperature probes (vented forward facing)
 - Heated total temperature probes (De-Ice total temperature probe, Goodrich)
 - Rearward facing (developmental)

Background

Total temperature (thermal and inertial):

$$T_0 = T + \frac{V^2}{2C_p}$$

$$\frac{T_0}{T} = 1 + \frac{\gamma - 1}{2} M^2$$

Total temperature relevance –

- Thermal interaction between the icing cloud and air flow
- impinging particles contribute to kinetic heating effect (Gent et al., 2000)

Measurement considerations–

- Temperature sensor accuracy
- Incomplete recovery of total temperature
 - Thermal surfaces (sources and sinks)
 - Flow effects (viscous losses)
 - Debris contamination, including icing and ice ingestion

Background

Recovery factor and correction

$$Y = \frac{T_r - T_s}{T_0 - T_s} \qquad \eta = \frac{T_0 - T_r}{T_0} \quad , \quad \eta = f(M)$$

(T_r – recovery temperature ~ measured temperature)

For ice cloud interaction at $M = \text{const.}$,

$$\frac{T_{0,1} - T_{r,1}}{T_{0,1}} = \frac{T_{0,2} - T_{r,2}}{T_{0,2}} \qquad T_{0,2} - T_{0,1} \left(\frac{T_{0,2}}{T_{0,1}} \right) = T_{r,2} - T_{r,1} \left(\frac{T_{0,2}}{T_{0,1}} \right)$$

1- before ice cloud
2- during ice cloud

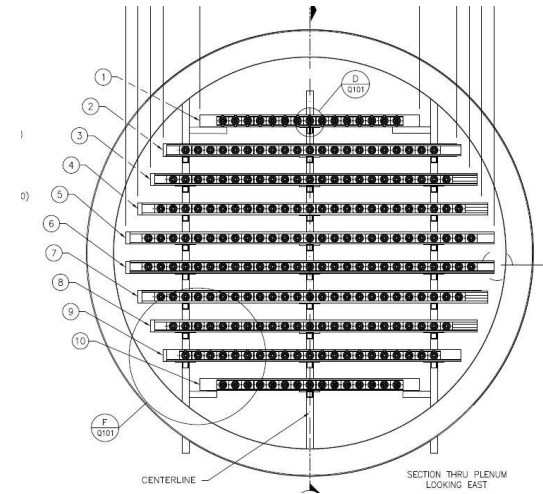
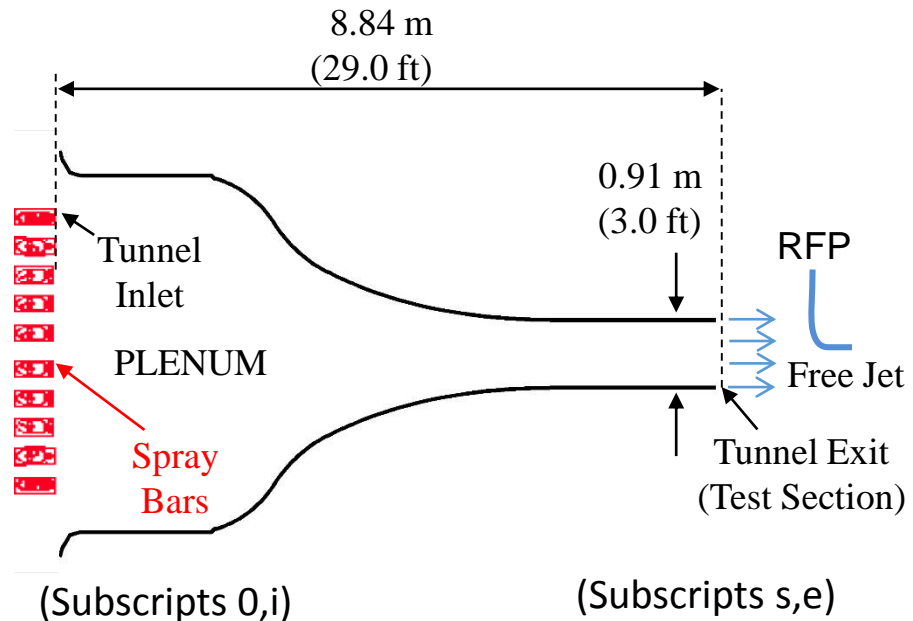
For small temperature changes around freezing,

$$\Delta T_0 \approx \Delta T_r$$

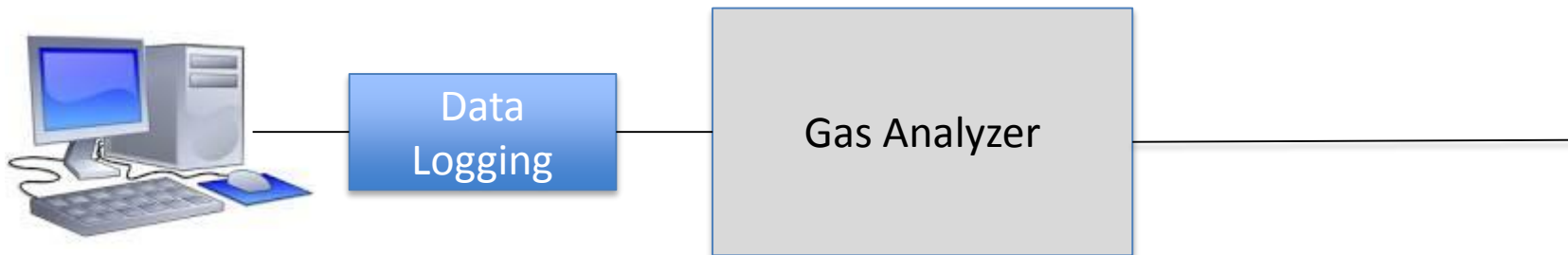
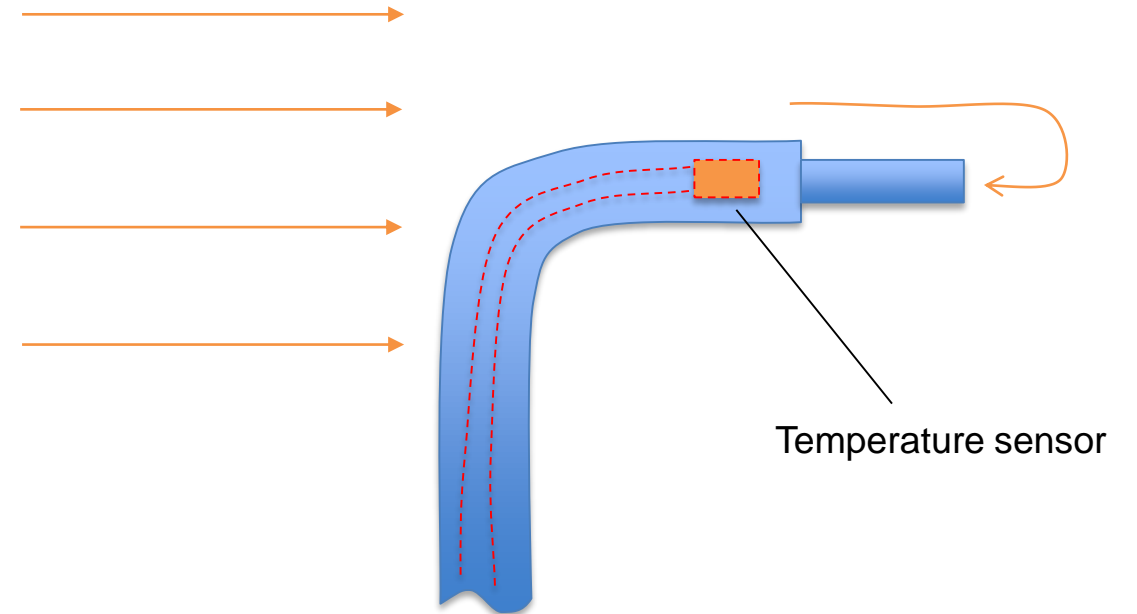
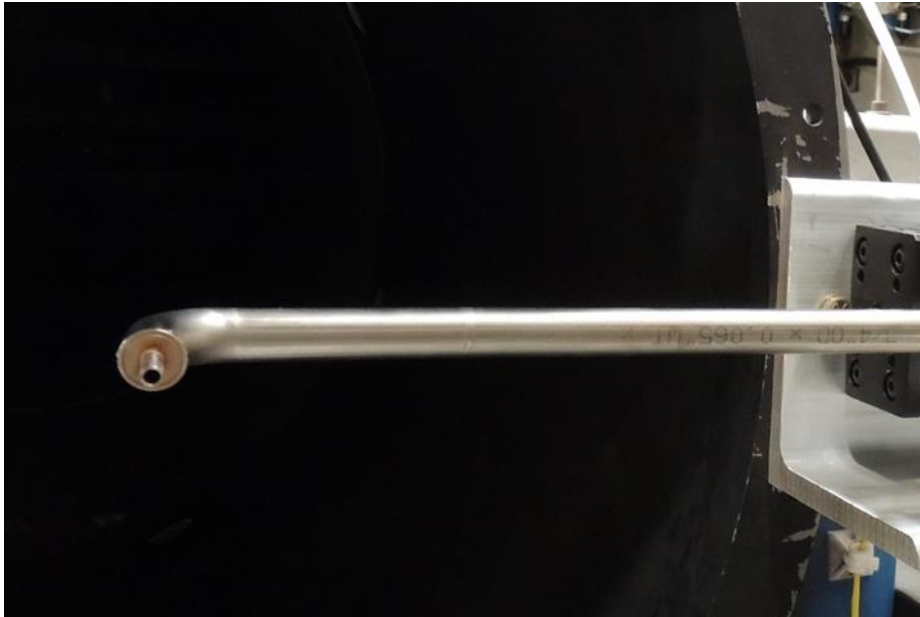
Propulsion Systems Laboratory (PSL)

Tunnel Capability

- Freeze out liquid cloud
- 12 parameters can be varied
 - P , V , T_{air} , T_{water} , RH, MVD, TWC, Water Type, Nozzle Pattern...

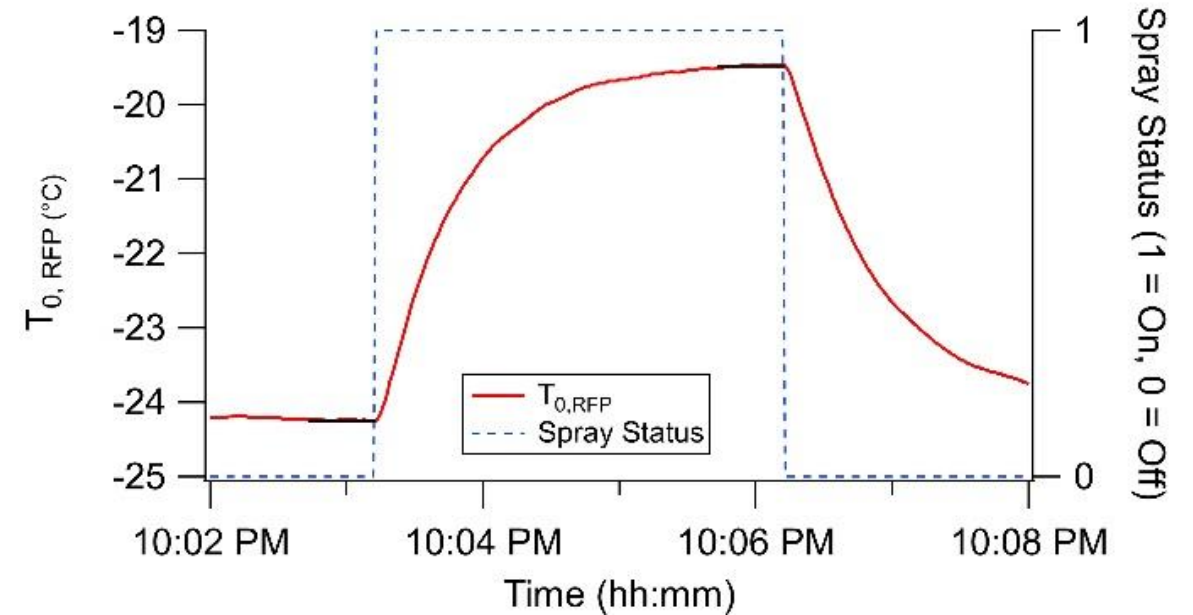
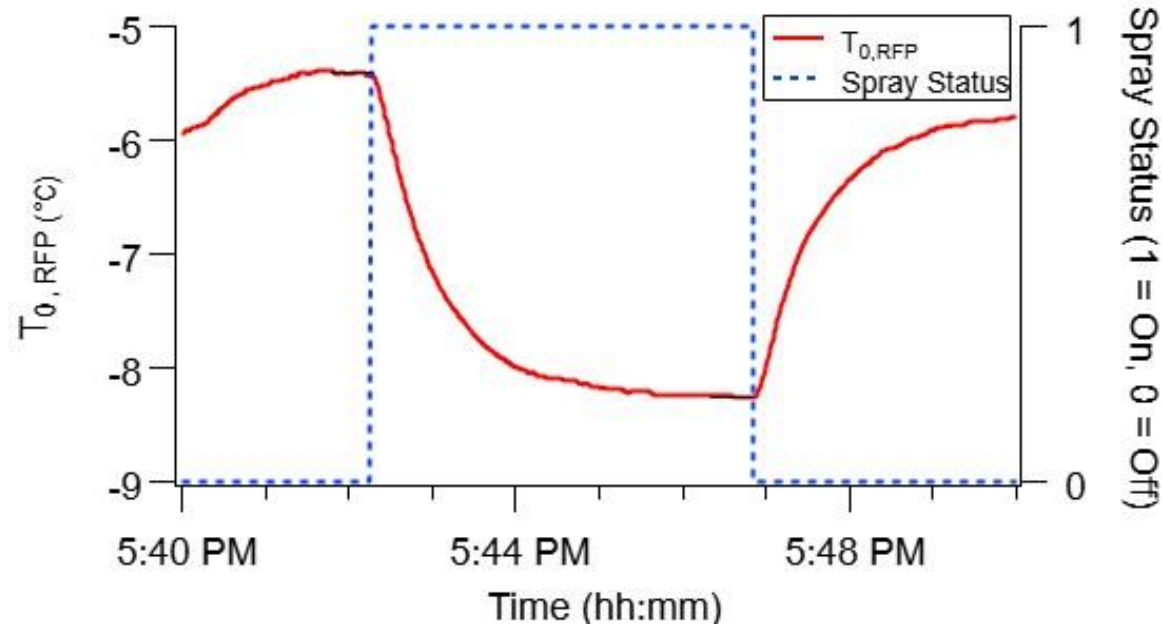


Rearward Facing Probe (RFP)



Rearward Facing Probe

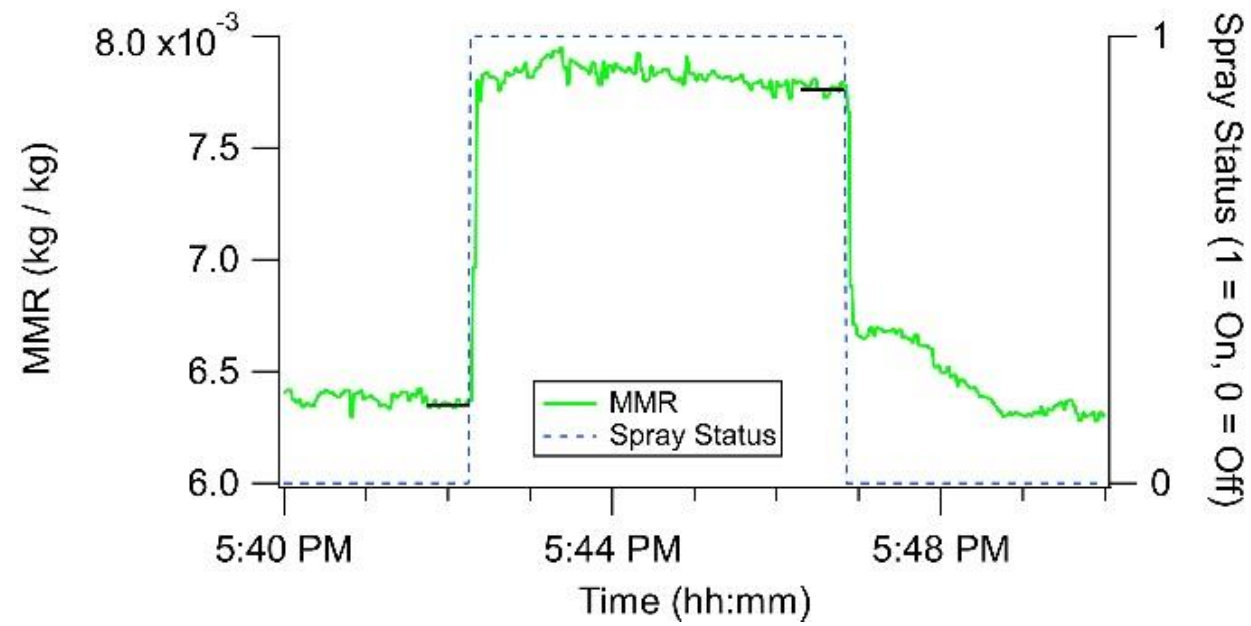
Total Temperature signals



Rearward Facing Probe

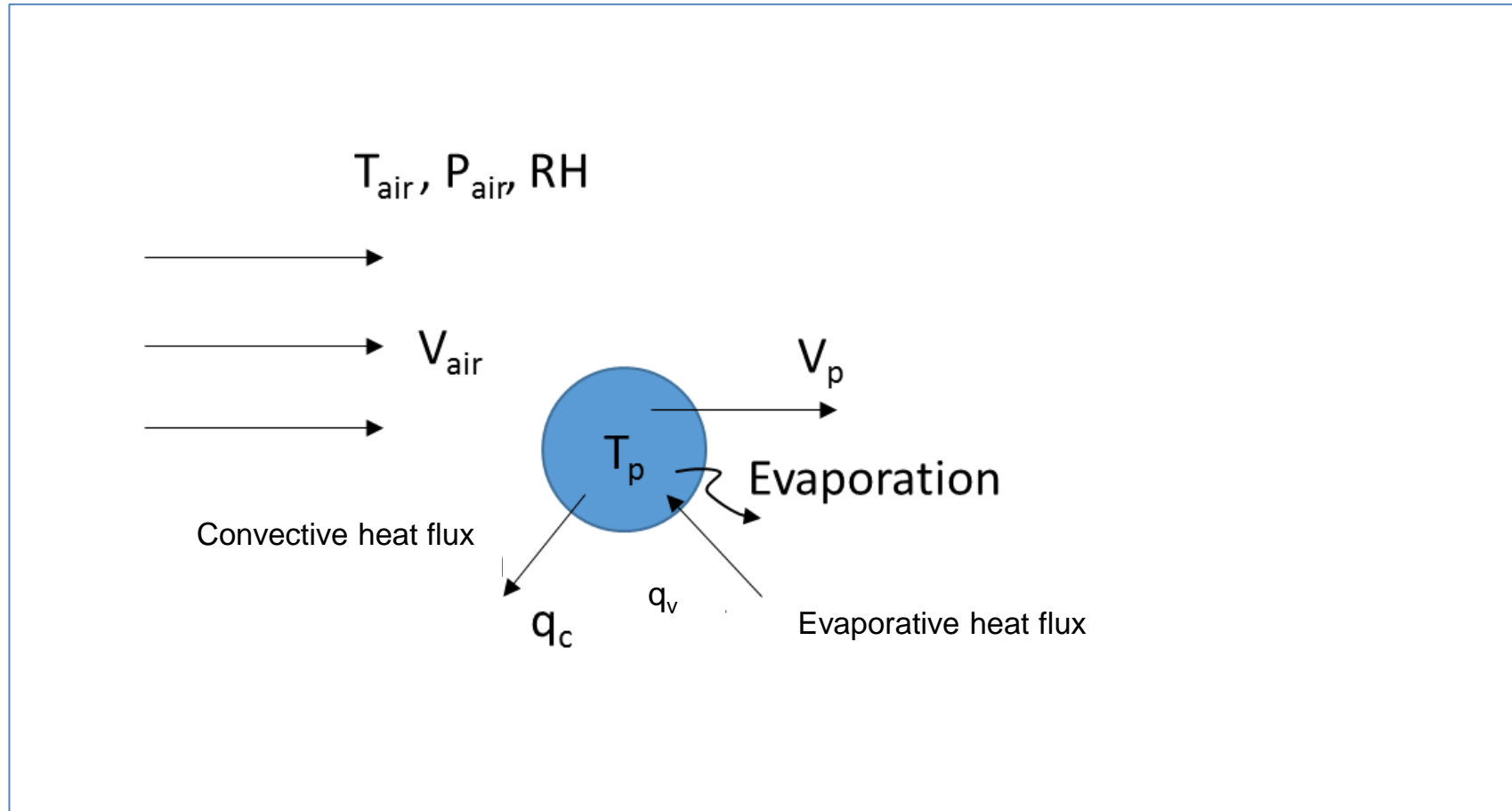
Humidity signal

Sampled flow → Gas/Humidity Analyzer



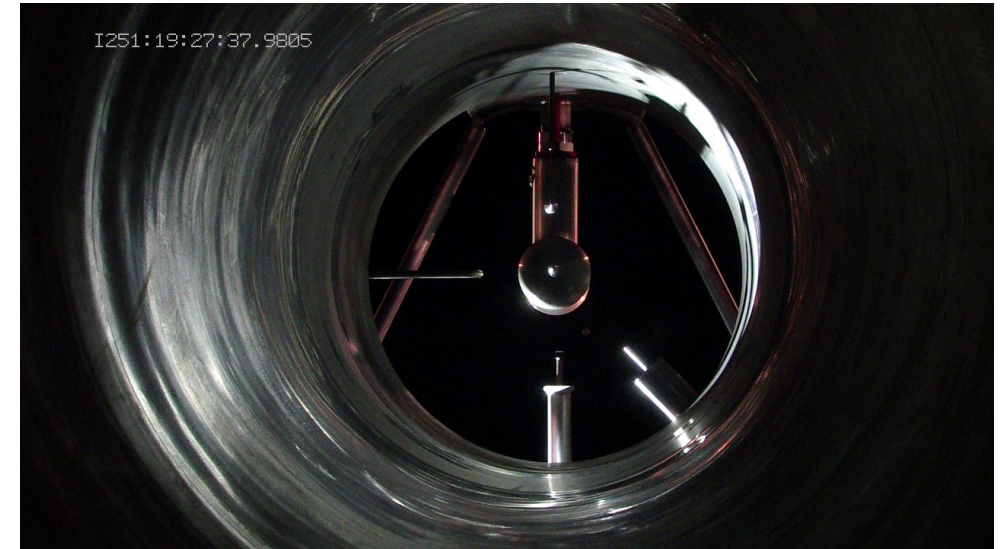
Thermal Model

Bartkus et al. (2015, 2016, 2017)

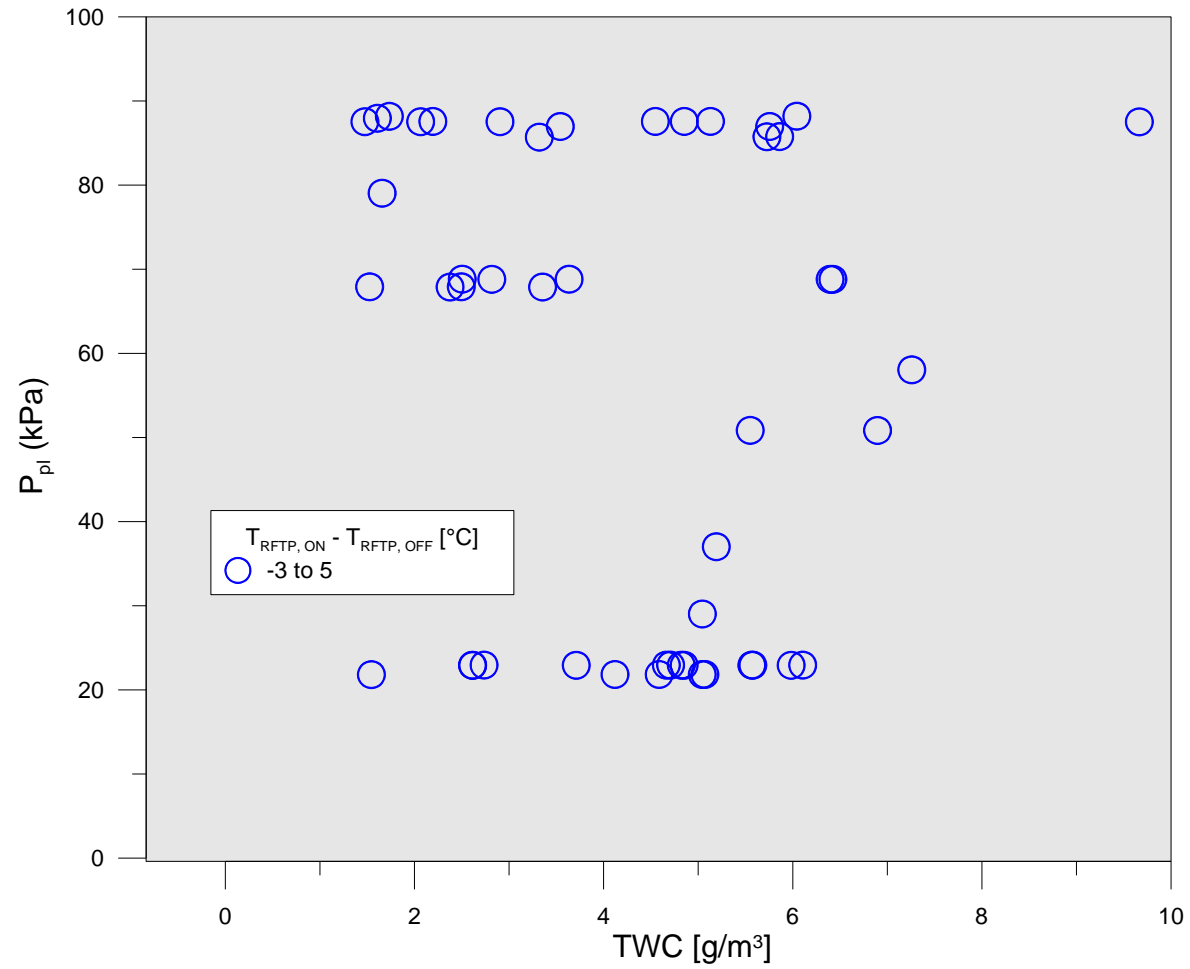


2017 Cloud Calibration Test Campaign

- **Test objectives**
 - Expand facility and measurement capabilities
 - Validate models
- **223 Test runs (conducted over 13 days)**
- **12 parameters can be varied:**
 - P, V, T_{air}, T_{water}, RH, MVD, TWC, Water Type, Nozzle Pattern...
- **Data reduction**
 - Discard any unsteady or fluctuating signals or signals that did not reach equilibrium during cloud spraying.
 - average variables before and during spray
 - Determine delta Temperatures and humidity
- **Selection of variable sweeps (e.g. Total Water Content)**



Tests





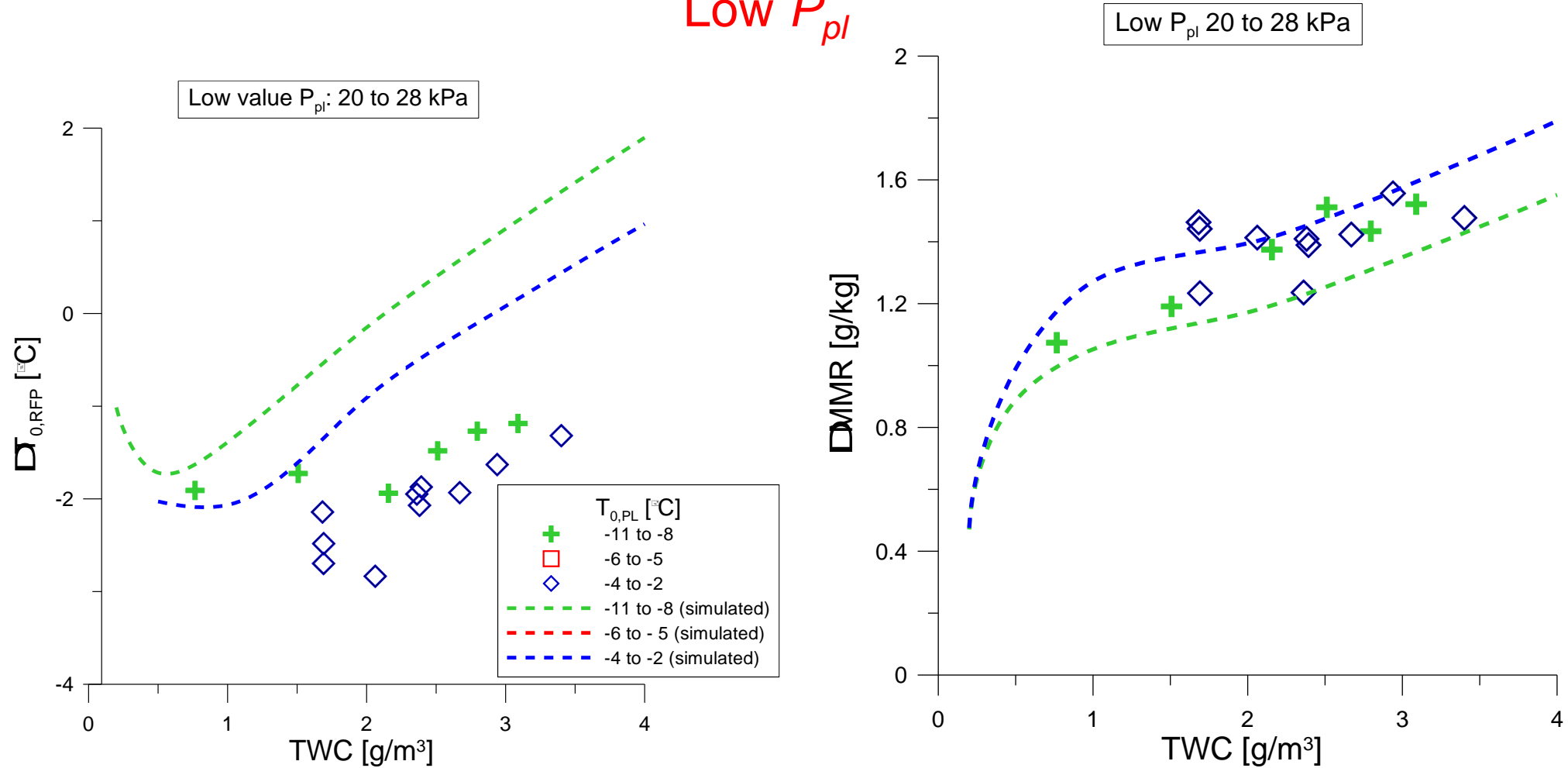
TWC sweeps

Plenum Pressure (P_{pl}) [kPa (Pisa)]	Plenum Temp. (T_{pl}) [°C]	Parameter in plots	Particle MVD [μm]	Mach	Tw [°C (°F)]	City/DI water	RH %
low: 20 to 28 (2.9 to 3)	low, mid, high*	Temp	15 - 20	0.44	7.2 (45)	City	45
mid: 62 to 70 (9 to 10.2)	low, mid, high*	Temp	15 - 20	0.22	82 (180)	DI	45
high: 90 to 97 (13 to 14)	low, mid, high*	Temp	15 - 20	.13 - .22	82 (180)	DI	45

Results

Negative Changes in Total temperature

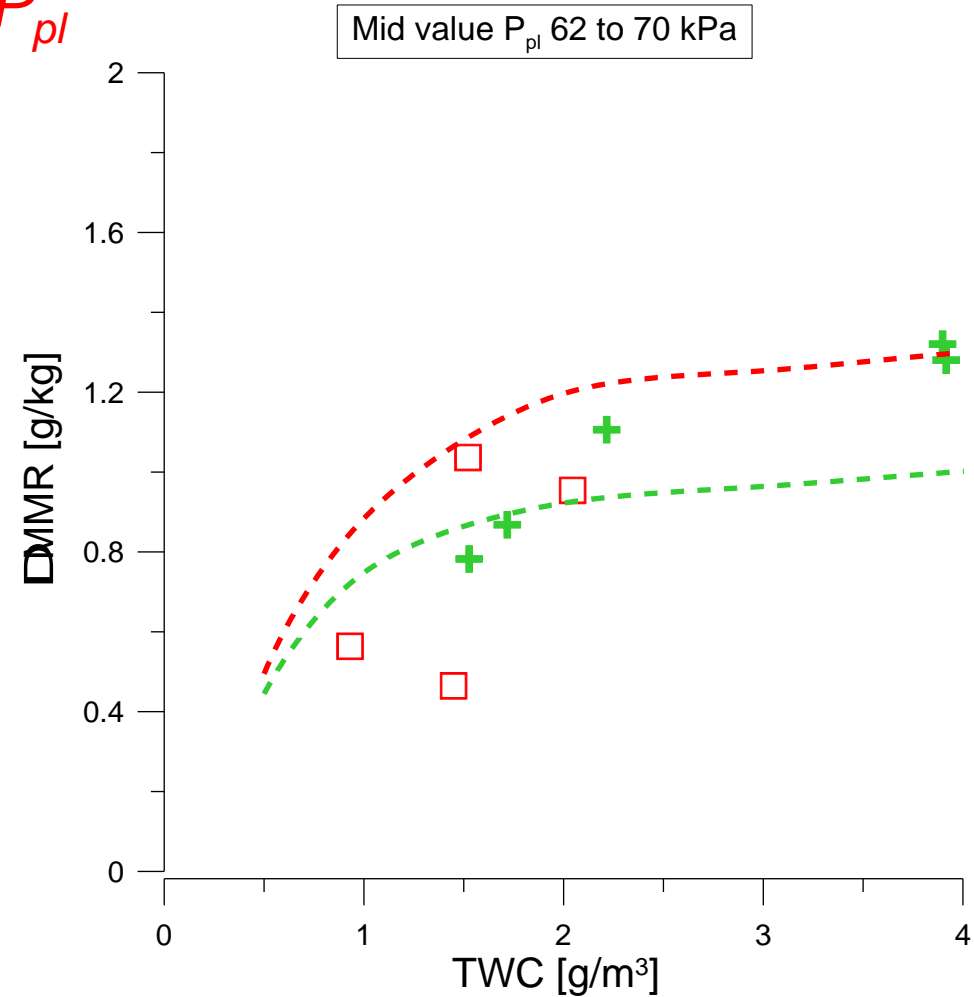
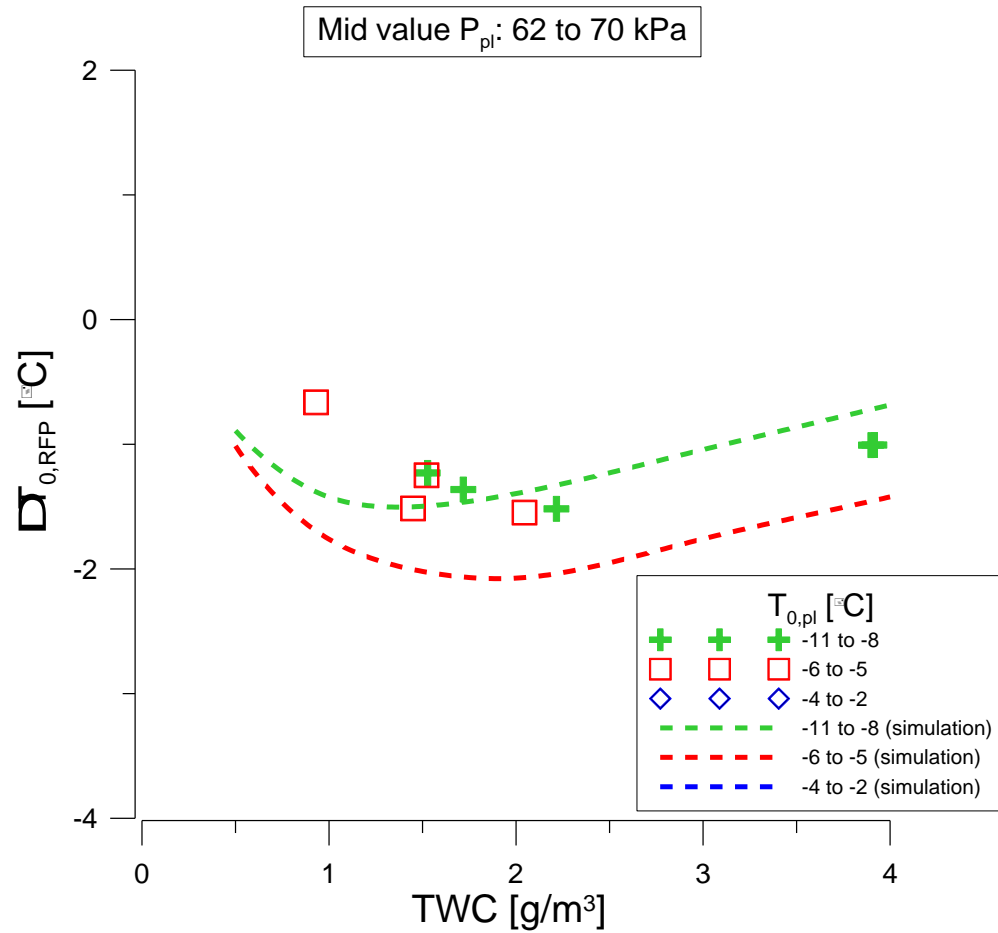
Low P_{pl}



Results

Negative Changes in Total temperature

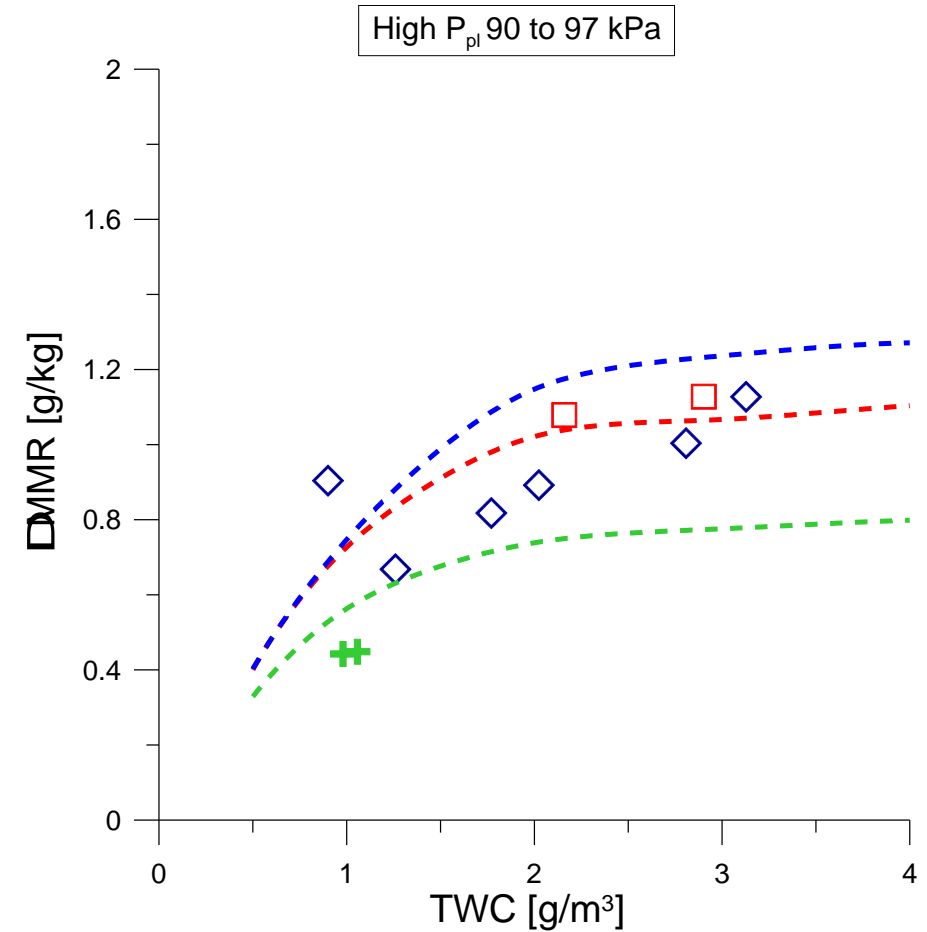
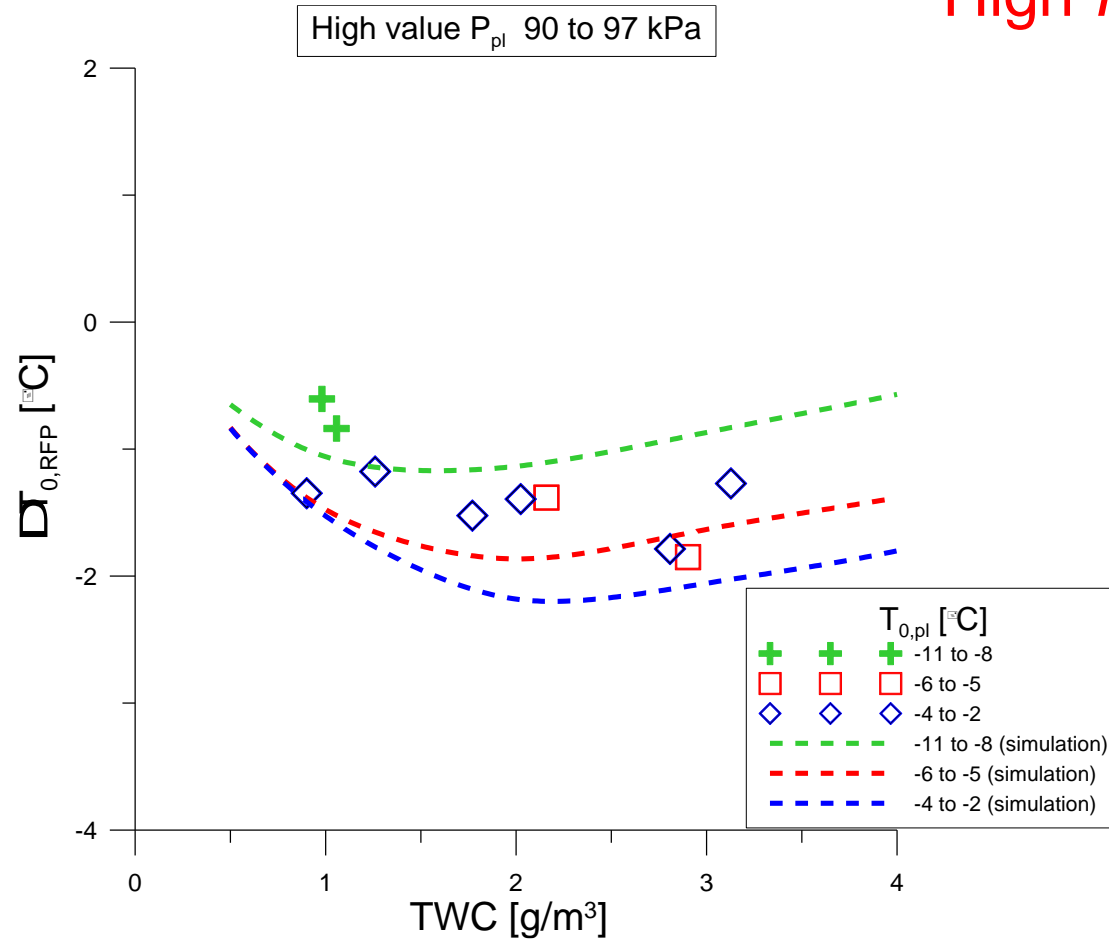
Mid P_{pl}



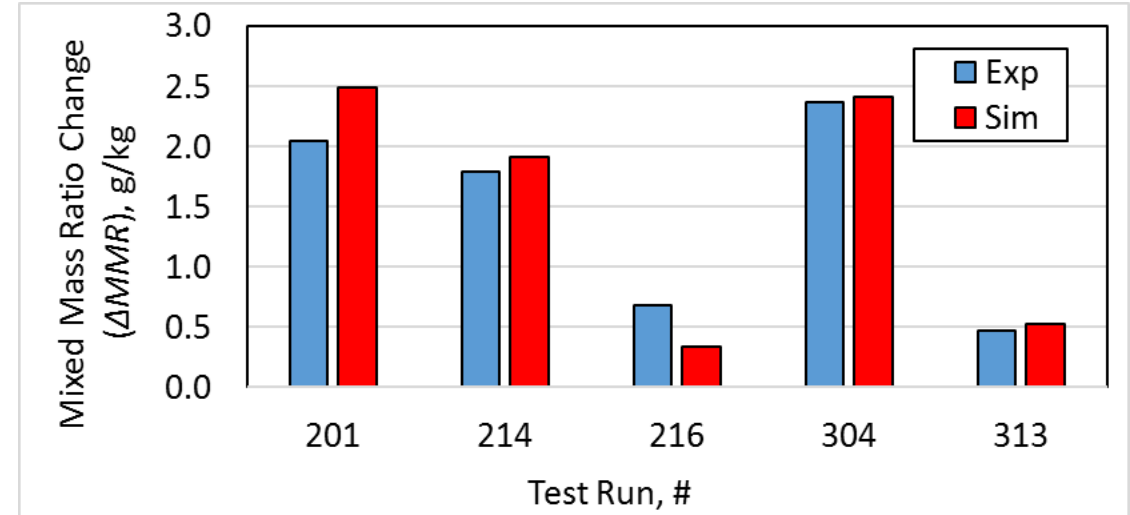
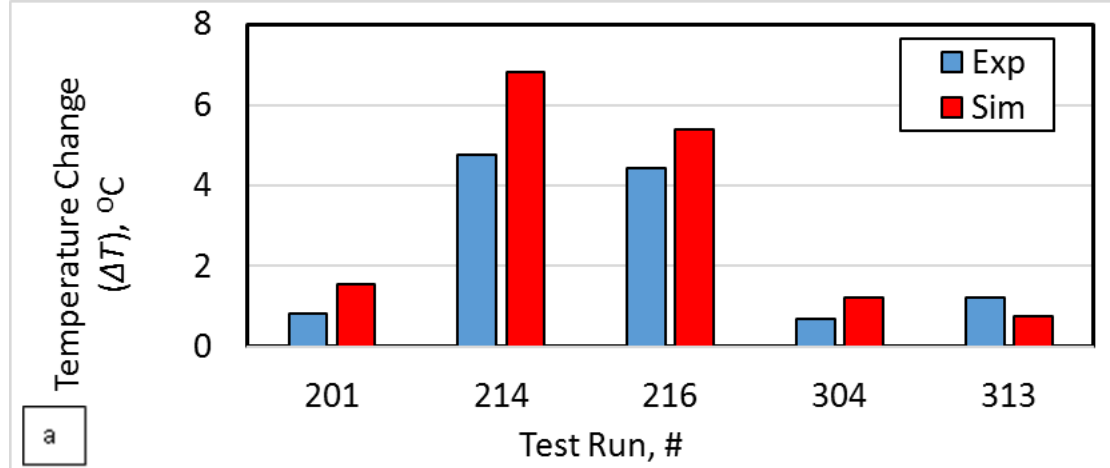
Results

Negative Changes in Total temperature

High P_{pl}



Cases with Positive increase in temperature



Test Run	T _{PL} (total)	P _{PL} (total)	RH _{PL} (Total)	Exit Air Velocity	Target TWC	Approx Initial MVD	Water Type	Initial Water Temp
#	[°C]	[kPa]	[%]	[m/s]	[g/m ³]	[μm]	[City/DI]	[°C]
201	-3.1	22.5	45	144	6.52	33	City	8
214	-23.7	21.5	45	101	9.26	33	City	8
216	-35.7	23.9	45	128	4.70	41	City	8
304	-3.2	22.5	45	142	6.39	45	City	8
313	-15.7	86.6	45	115	6.45	24	City	8

Conclusions

- A Rearward Facing Probe is being developed in-house to measure local total temperature and humidity during atmospheric icing flow conditions.
- The thermal model showed that the large temperature differential between the injected droplet and the atmospheric flow produced competing evaporative and convective heat transfer effects.

Results:

- Small total temperature drops in the range of 0.6 to 2.8 °C and up to 1.5 g/kg of water vapor rise through the interaction.
- The largest changes in total temperature and humidity generally occurred at plenum conditions of low pressure and high temperature, and under glaciated cloud conditions.
- The least effects in total temperature were found at large *TWC* and low temperatures.
- Under certain high *TWC* conditions and glaciated , the interaction with the cloud produced a warming of the airflow.
- The thermal model in terms of evaporative and convecting heat transfer mechanisms helped in interpreting these trends.