



NASA Aeronautics Research Institute

# Combined Electric Aircraft and Airspace Management Design for Metro-Regional Public Transportation

NASA Aeronautics Research Mission Directorate (ARMD)  
2014 Seedling Technical Seminar  
February 19–27, 2014

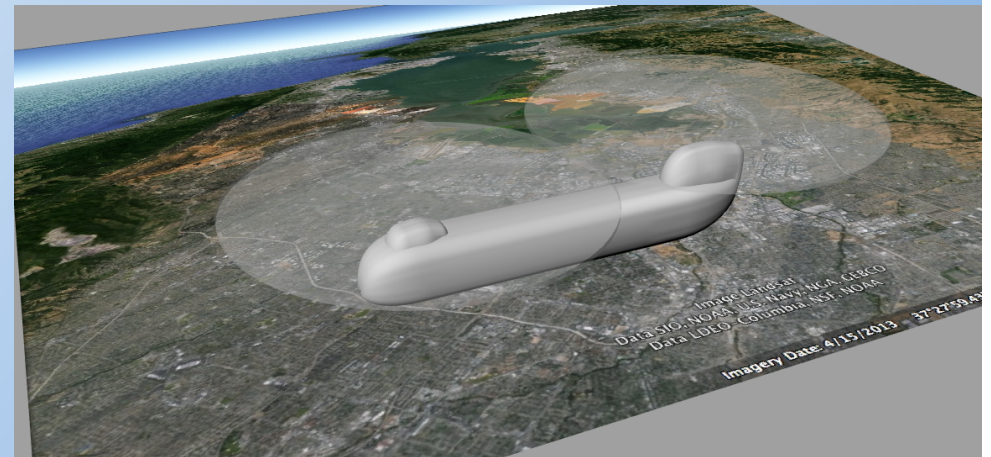


# Project Contributors



NASA Aeronautics Research Institute

- **Juan J. Alonso, Prof., Stanford University**
- **Heather Arneson, NASA**
- **John Melton, NASA**
- **Jeffrey Sinsay, US Army Aeroflightdynamics Directorate**
- **Brendan Tracey, Stanford**
- **Michael Vegh, Stanford**
- **Cedric Walker, NASA**
- **Larry Young, NASA, PI**





# Outline



NASA Aeronautics Research Institute

- **Innovation: Hopper Aerial Transport Network**
- **Technical approach**
- **Results**
  - Conceptual Design
  - Airspace Management Issues
  - Network Design and Simulation
  - Operations and Misc. Tech Issues
- **Future Work & Tech Transition**





# The Innovation

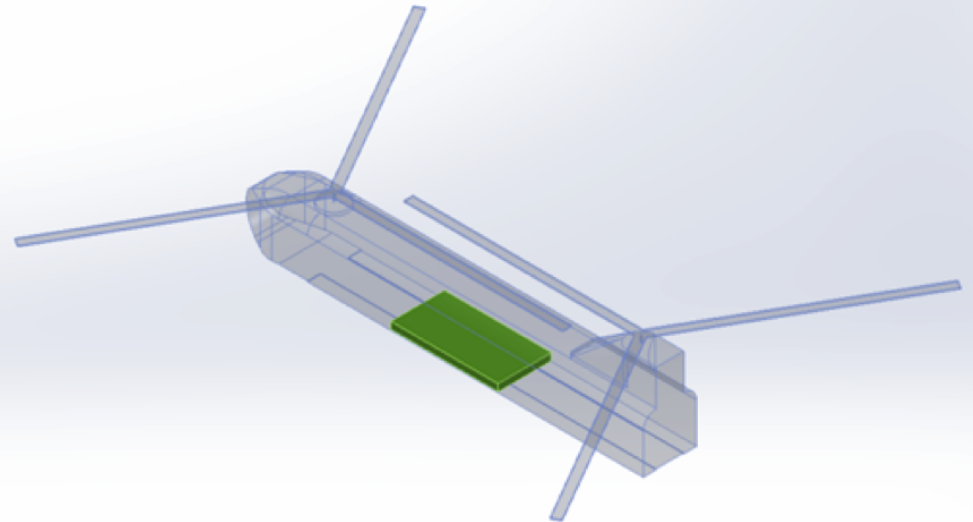


NASA Aeronautics Research Institute

- Investigate technical feasibility of electric aircraft to provide a solution to metro/regional public transportation



Hughes Helibus concept circa 1967 (Courtesy AHS International)





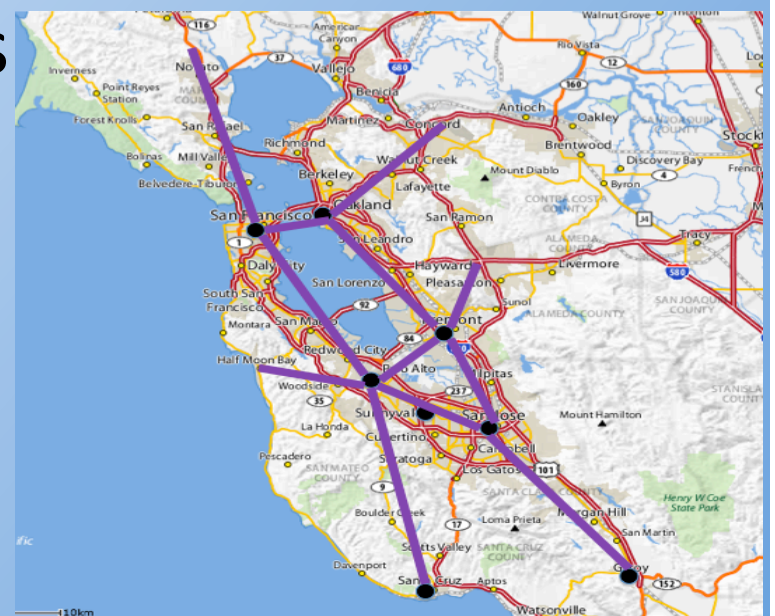
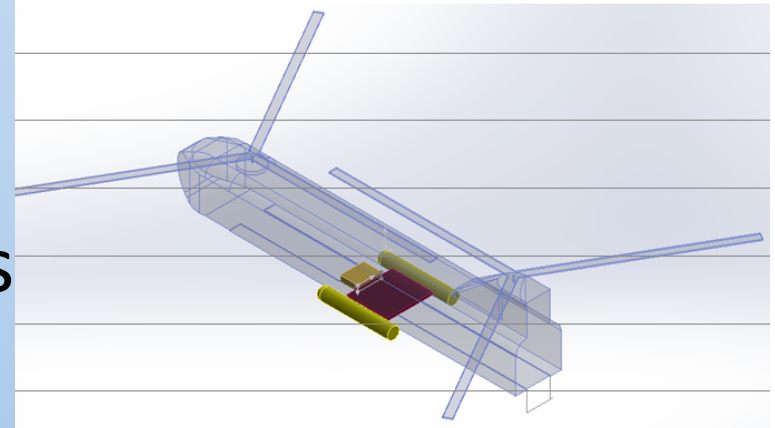


# “Hopper” Metro/Regional Aerial Transportation System



NASA Aeronautics Research Institute

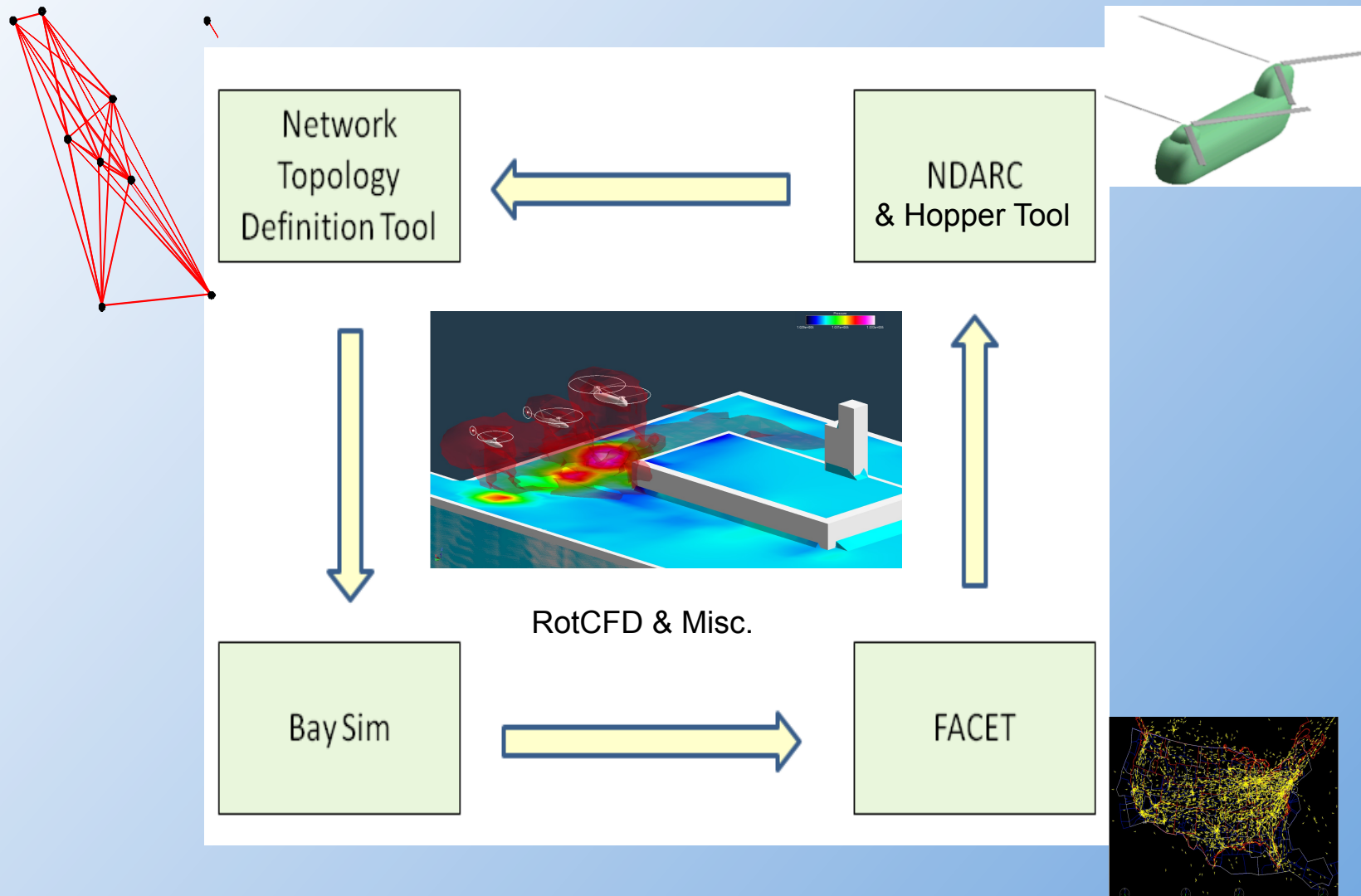
- Rotorcraft with electric-propulsion
- Extremely short haul vehicles (<100nm)
- Network of vertiport stations in San Francisco Bay Area
- Different network topologies studied
- An evolutionary build-up of networks studied





# Technical Approach

NASA Aeronautics Research Institute





NASA Aeronautics Research Institute

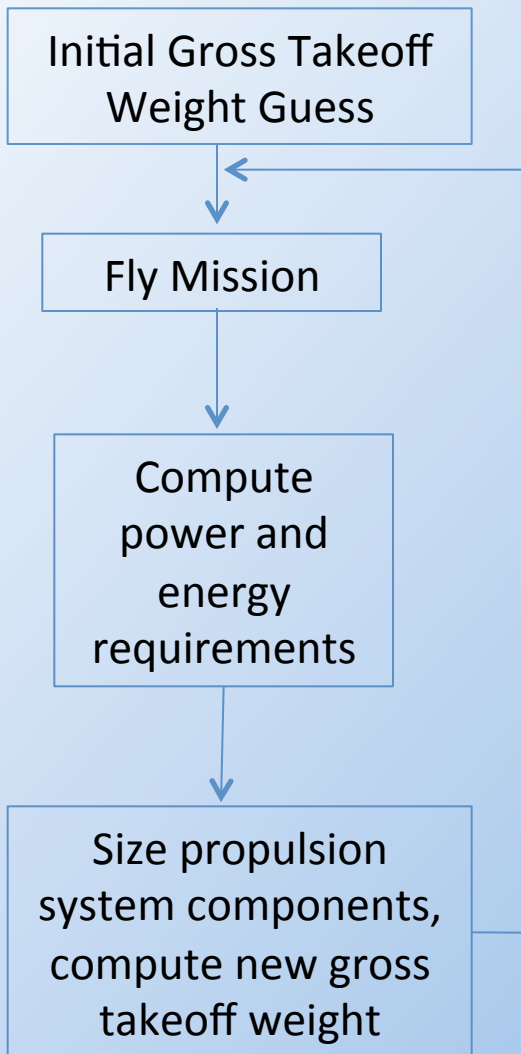
# Conceptual Design: Hopper/ Electric-Propulsion



# “Hopper” Tool Architecture



NASA Aeronautics Research Institute



Iterate until convergence

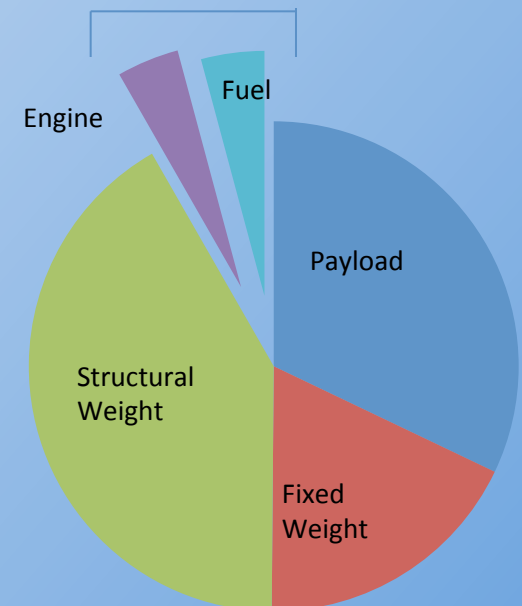
## Power Requirements

$$P_{hover} = \frac{W * \sqrt{\frac{DL}{2\rho}}}{\{FOM_{AC}\} * \{\eta_{motor}\}} \quad P_{cruise} = \frac{W * \{V\}}{\{\frac{L}{D_e}\} * \{\eta_{motor}\}}$$

{bold}=code inputs

Redesigned this section of the helicopter and observed effect on gross takeoff weight

## Turboshaft Weight Breakdown



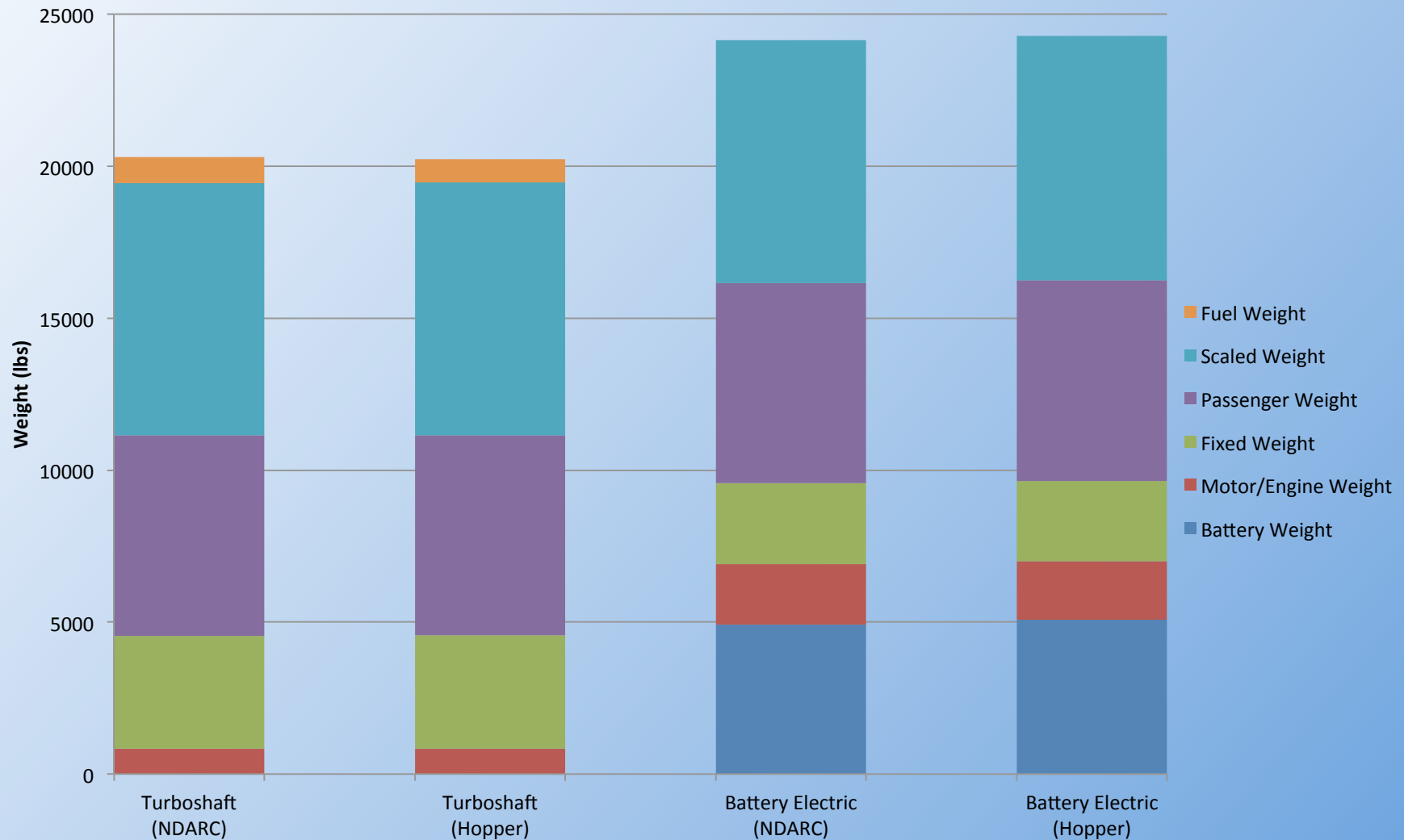




# NDARC (Phase I) and Stanford Tool (Phase II) Validation Comparison



NASA Aeronautics Research Institute

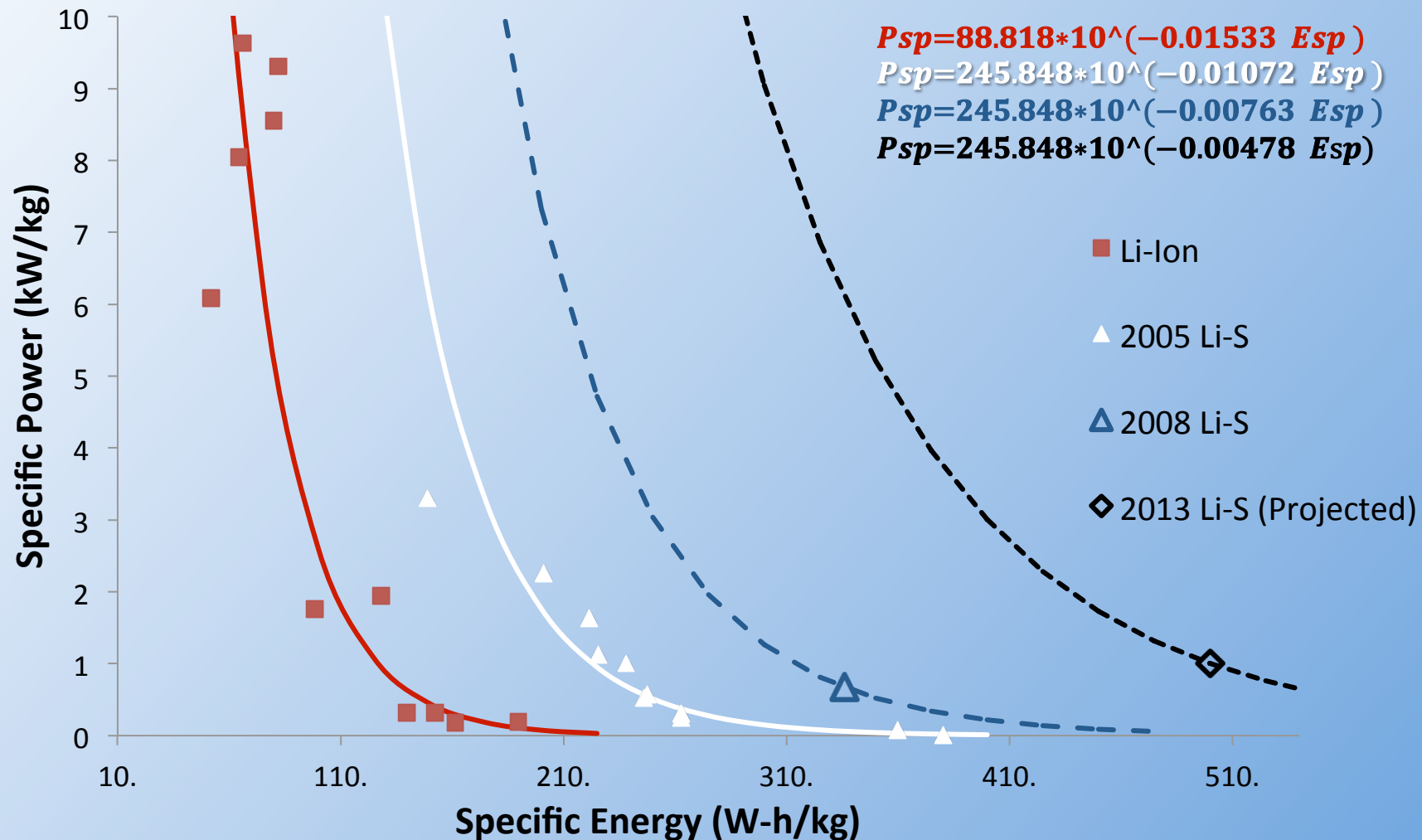




# Ragone Plot: Battery Specific Power vs. Energy



NASA Aeronautics Research Institute





# Sizing Requirements



NASA Aeronautics Research Institute

| Mission Requirements     |                                  |
|--------------------------|----------------------------------|
| Cruise velocity          | 130 kt.                          |
| Cruise altitude          | 3000 ft.                         |
| Disk Loading             | 4 psf                            |
| L/De                     | 4.2                              |
| Aircraft Figure of Merit | 0.5                              |
| Number of passengers     | 30                               |
| Range                    | 25-65nm (with 20 minute reserve) |

| Sizing Assumptions         | Gas  | Battery | Hybrid | Fuel cell | Fuel Cell Hybrid |
|----------------------------|------|---------|--------|-----------|------------------|
| Structural Weight fraction | 0.42 | 0.331   | 0.37   | 0.42      | 0.37             |
| Fixed Weight               | 3710 | 2660    | 3000   | 3200      | 3200             |



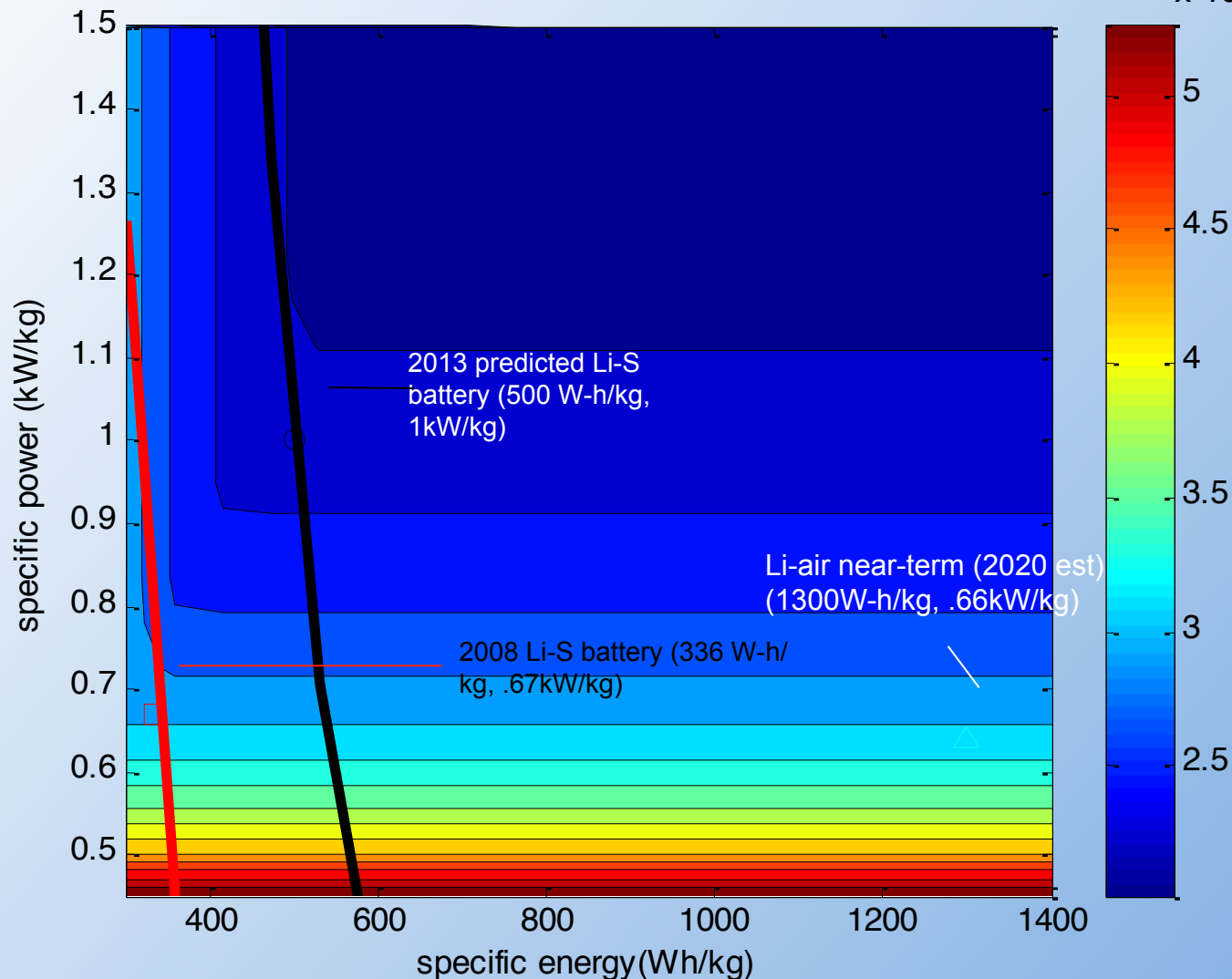
# Sample Battery Electric Design Results (25 nm)



NASA Aeronautics Research Institute

Gross Takeoff Weight(lbs)

$\times 10^4$



- $W_{gross_{2008}} \sim 26,640$  lbs.
- $W_{gross_{2013}} \sim 22,000$  lbs.
- $W_{gross_{2020}} \sim 31,000$  lbs.
- Power and energy improvements make these configurations much more realizable in the near-term
- Sizing loop diverges for Li-ion technology

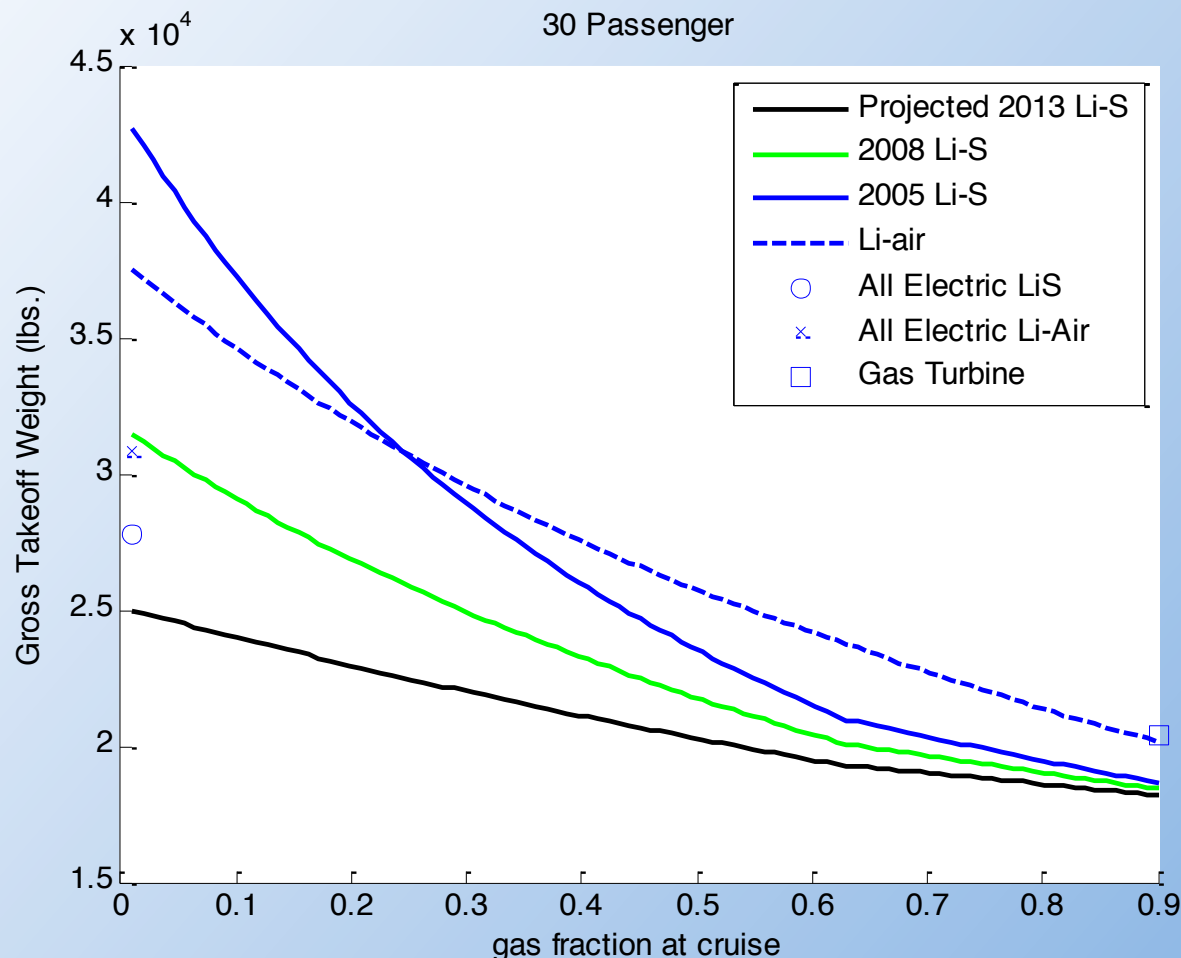




# Hybrid Technology (Turboshaft with Battery)



NASA Aeronautics Research Institute



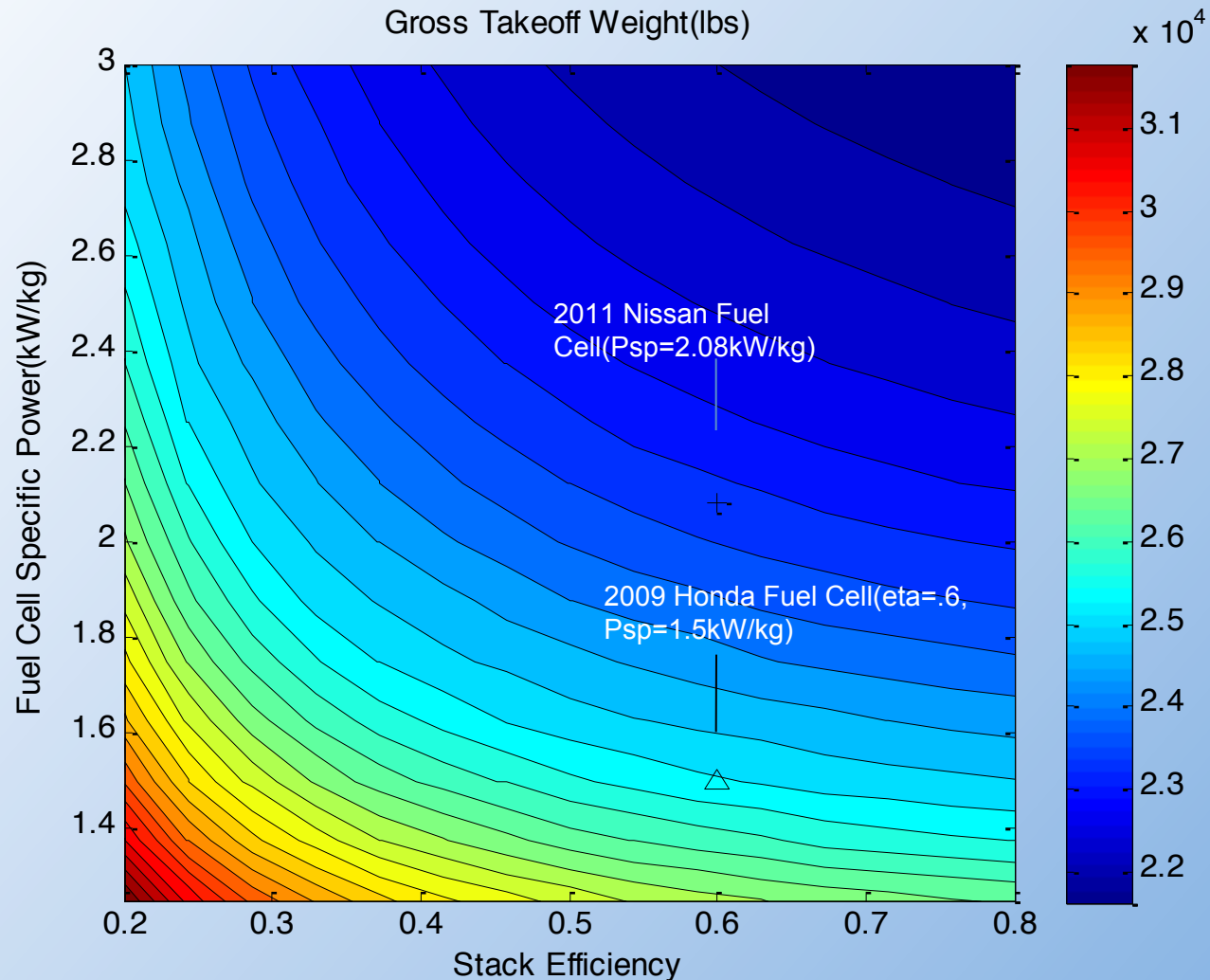
- Battery research tends to focus on improving specific energy, sometimes at cost of specific power
- Current technology outperforms future battery trends for most hybrid configurations, because power constraints tend to dominate these design



# Sample Fuel Cell Results



NASA Aeronautics Research Institute



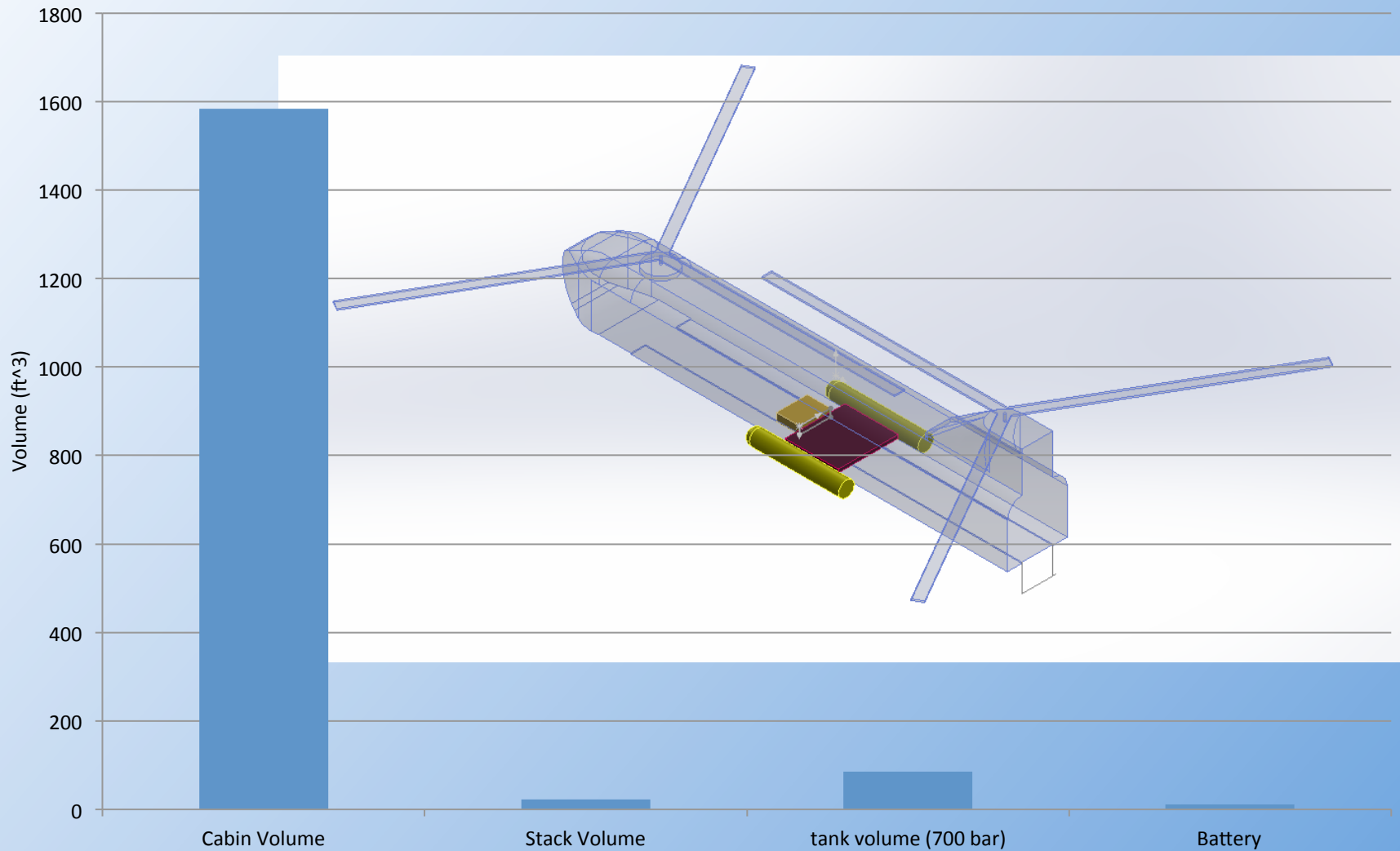
- Similar tradeoff between specific power and specific energy for fuel cell, although harder to quantify
- Efficiencies for  $H_2$  Fuel Cells usually  $\sim 30\text{-}60\%$ .
- Recent technology improvements result in cascading weight benefits
- 1999 technology ( $P_{sp} \sim .4 \text{ kW/kg}$ ) results in gross takeoff weight  $\sim 300,000 \text{ lbs}$ .



# Serial Fuel Cell Hybrid Volume Breakdown (65 nm)



NASA Aeronautics Research Institute

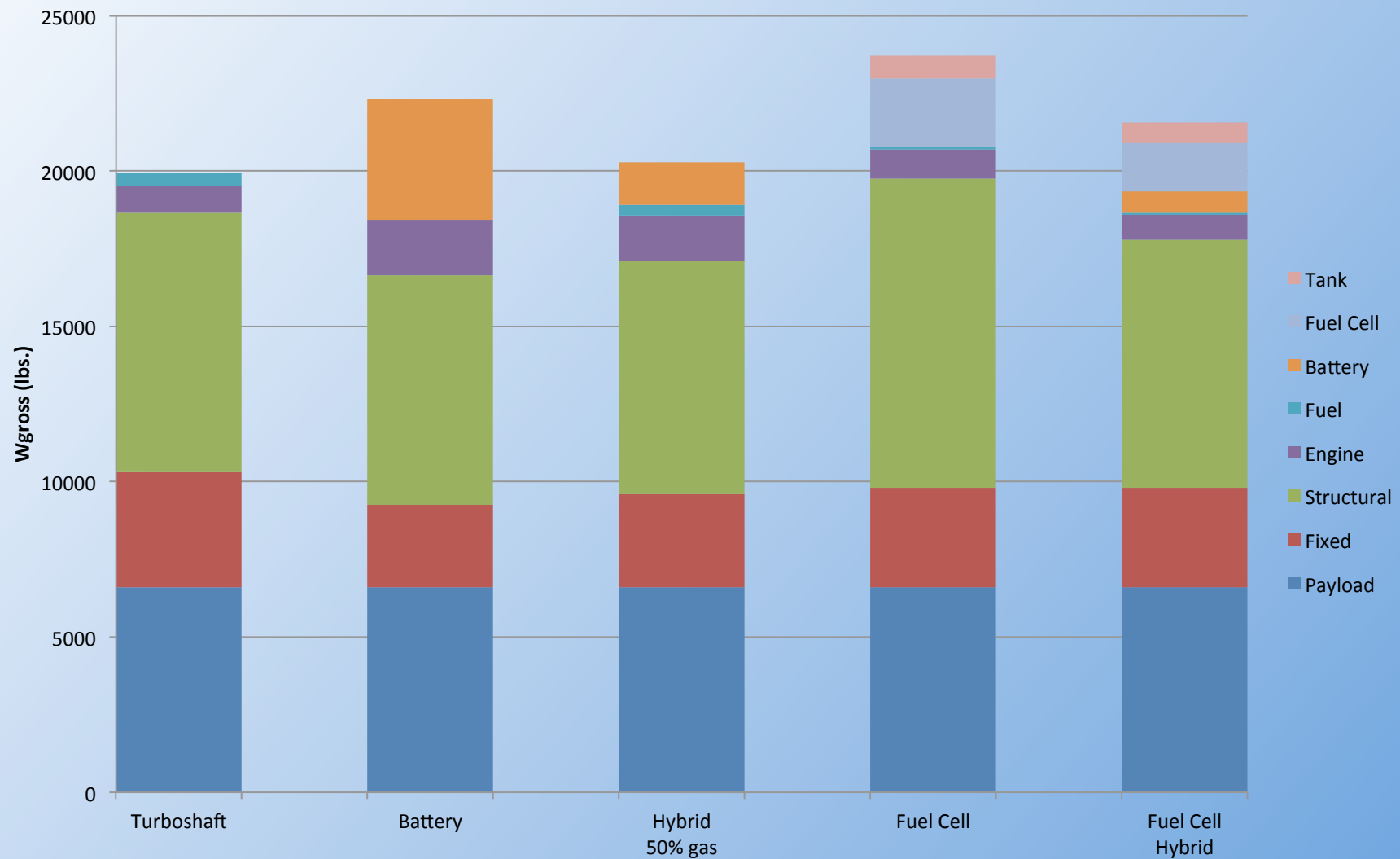




# Weight Breakdown (25 nm)



NASA Aeronautics Research Institute

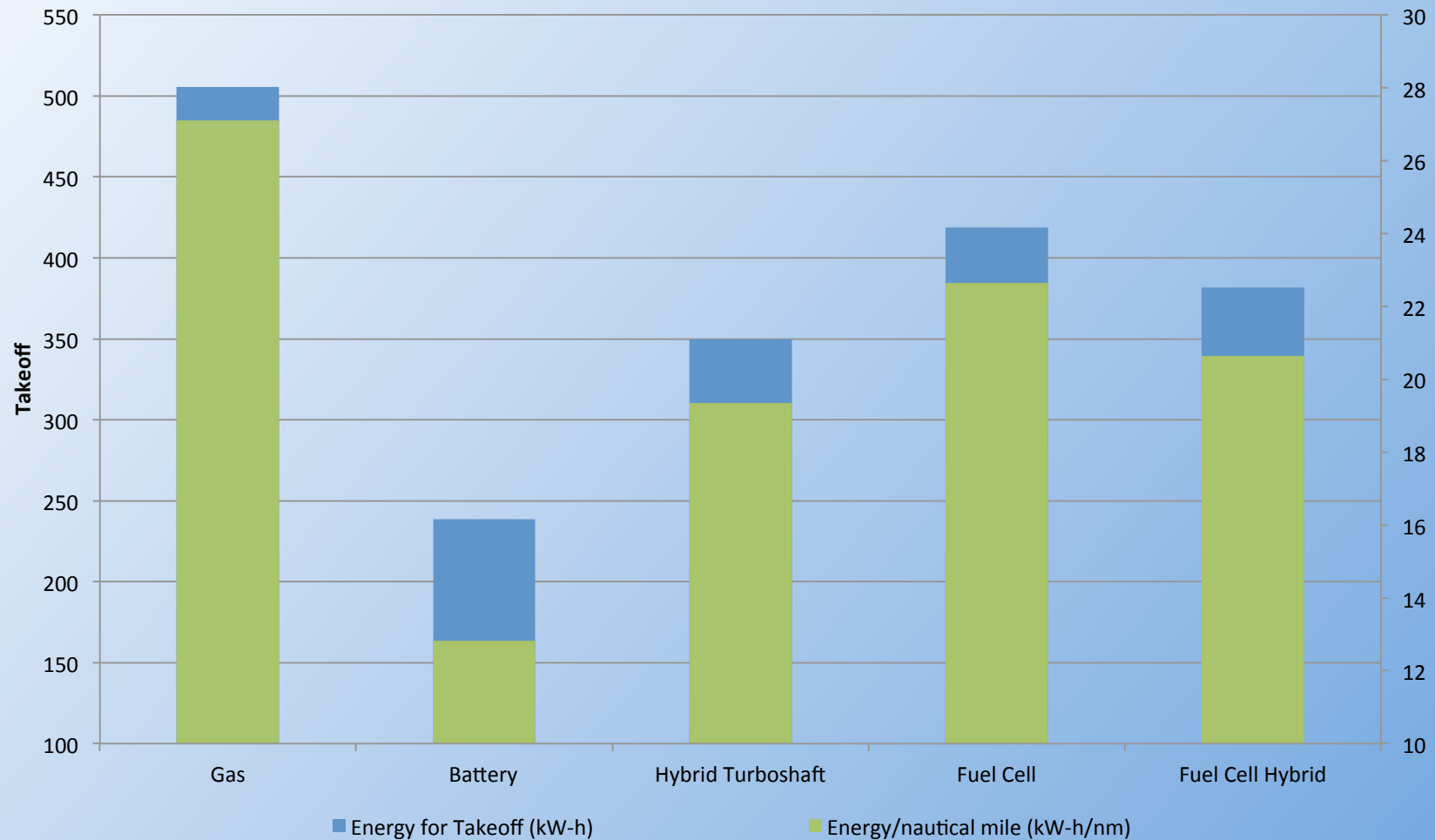




# Energy Usage Comparison (Full)



NASA Aeronautics Research Institute



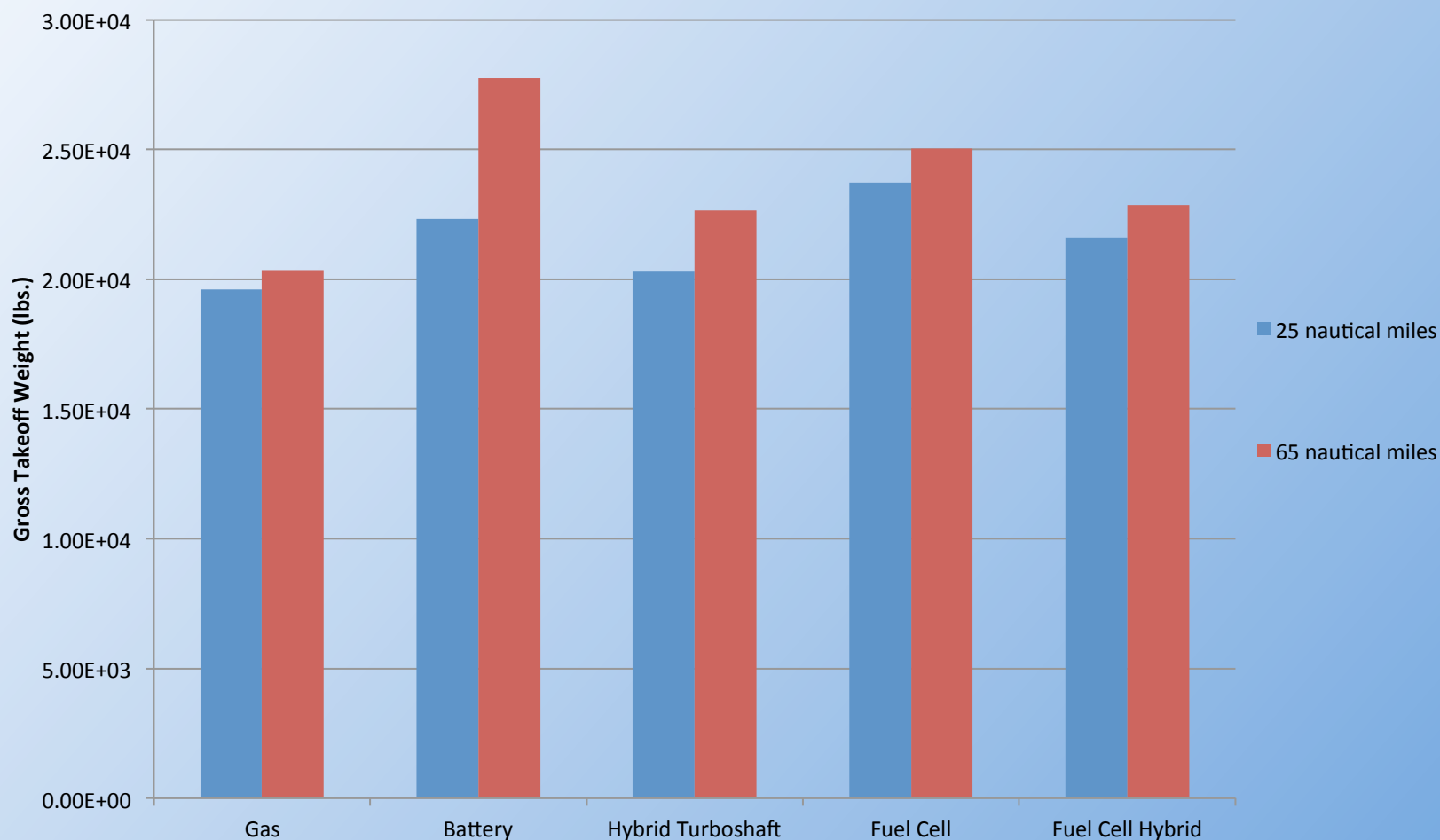




# Gross Takeoff Weight Comparison



NASA Aeronautics Research Institute





NASA Aeronautics Research Institute

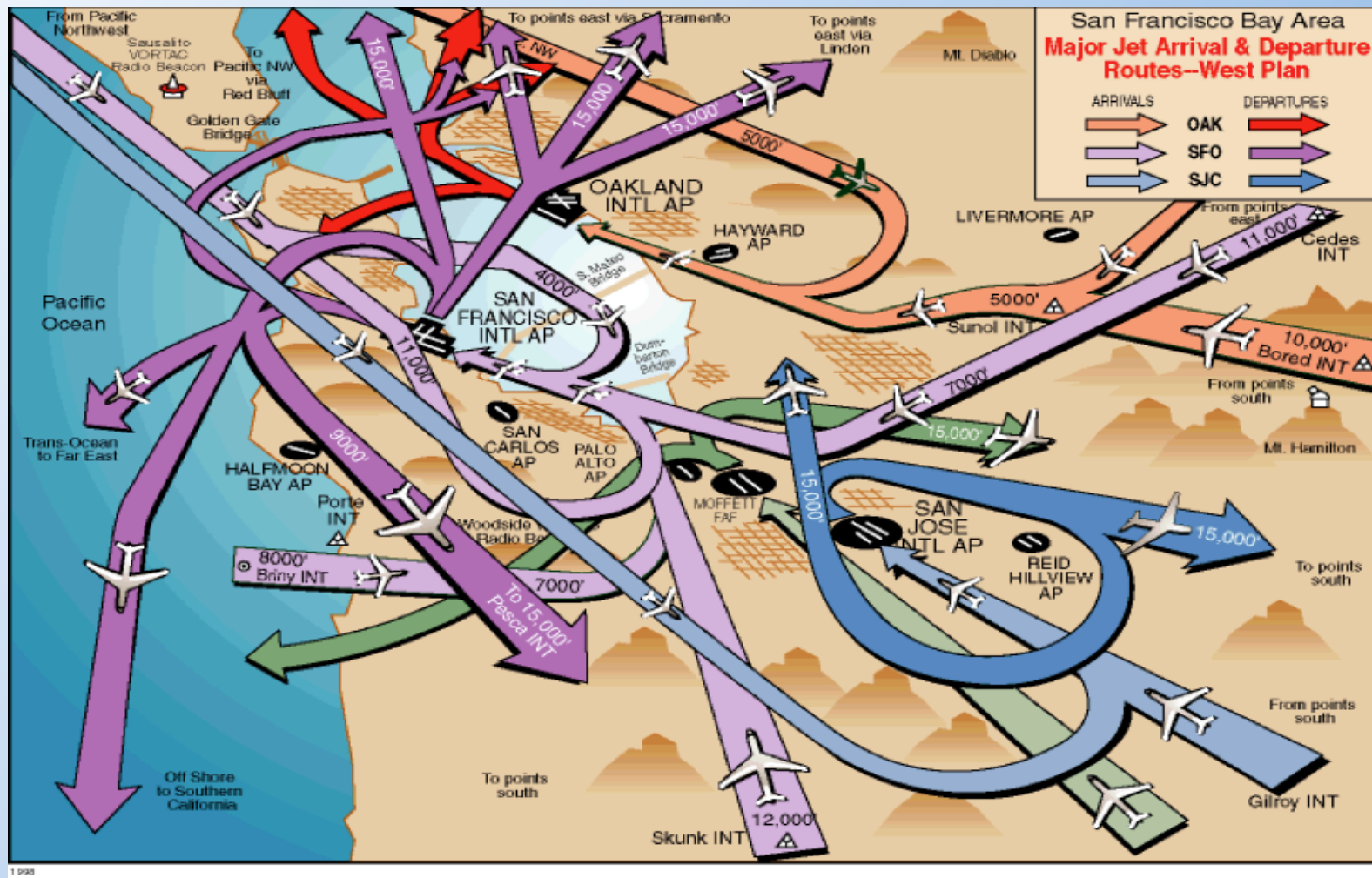
# Airspace Management Issues



# Bay Area Traffic Flows



NASA Aeronautics Research Institute





# ATM Focus



NASA Aeronautics Research Institute

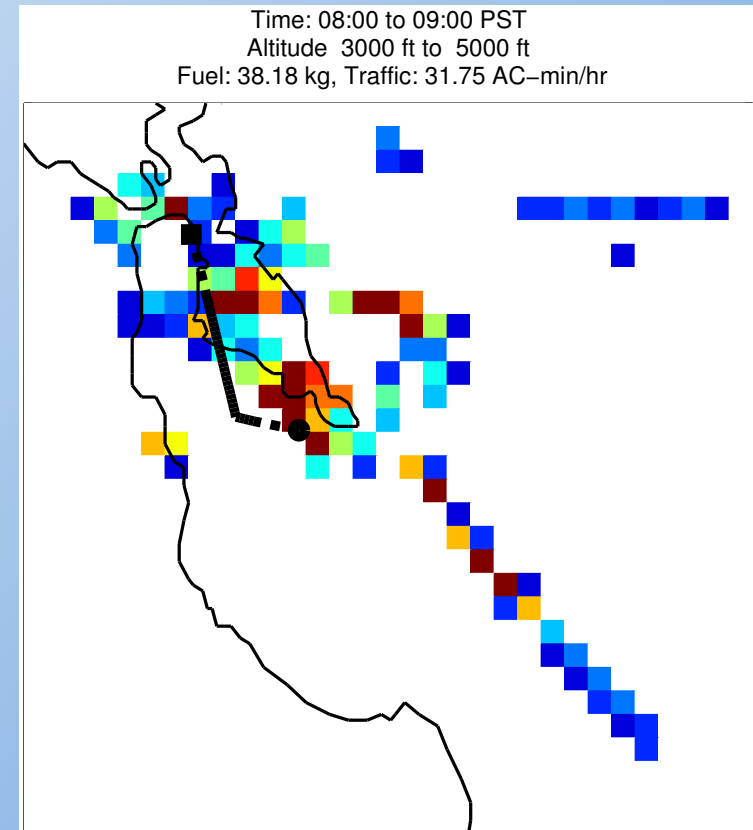
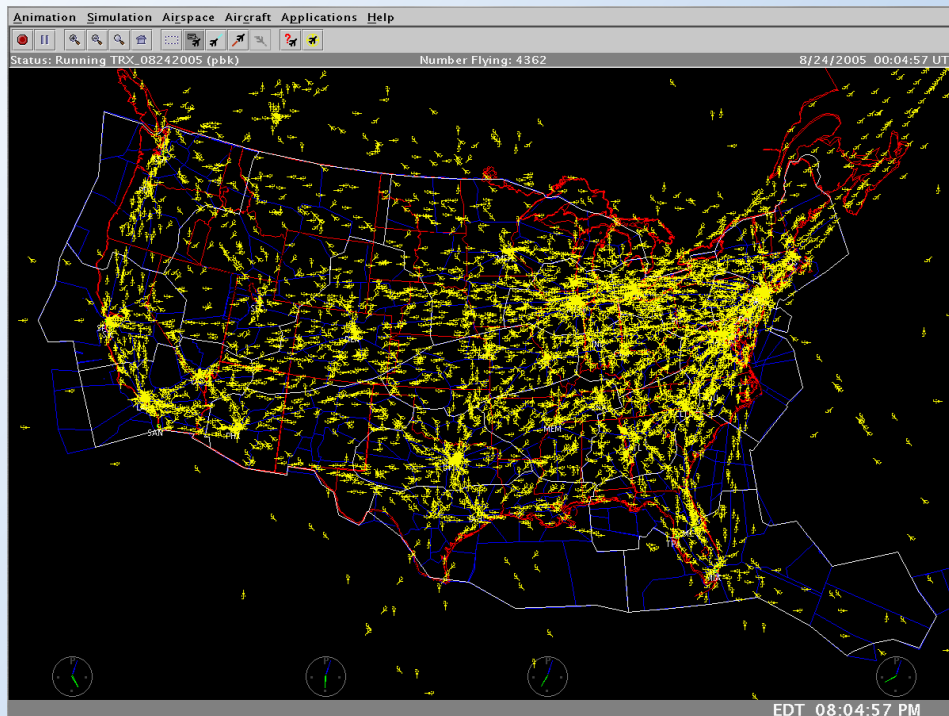
- Generate Hopper flight plans to reduce potential conflicts with background traffic
  - Reduce controller workload by avoiding heavily used airspace
  - Adjust flight plans based on time of day as background traffic shifts
- Investigate emerging technologies which could enable Hopper



# Traffic Density Maps



NASA Aeronautics Research Institute



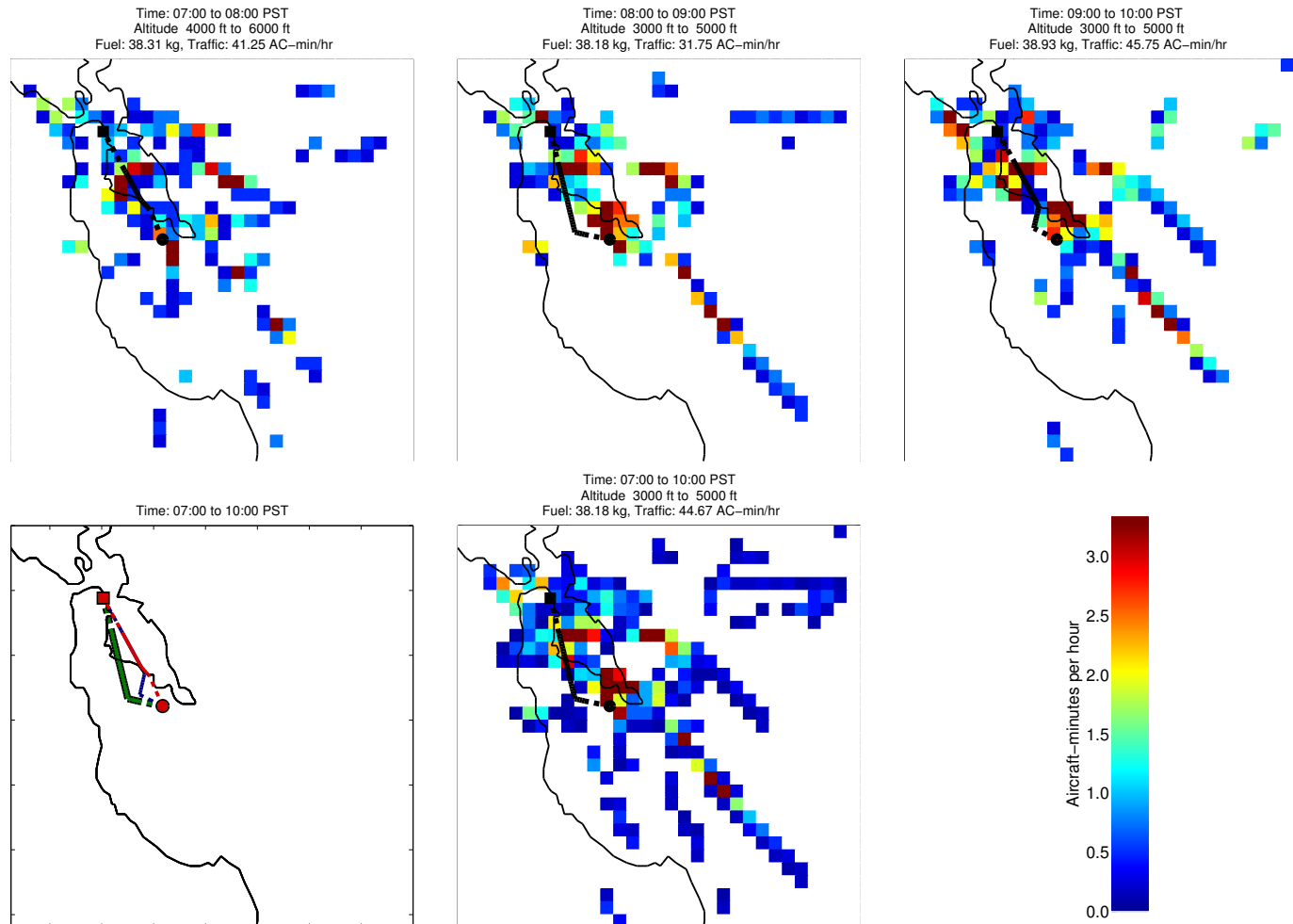




# Time-of-Day Dynamic Flight Path Routing



NASA Aeronautics Research Institute

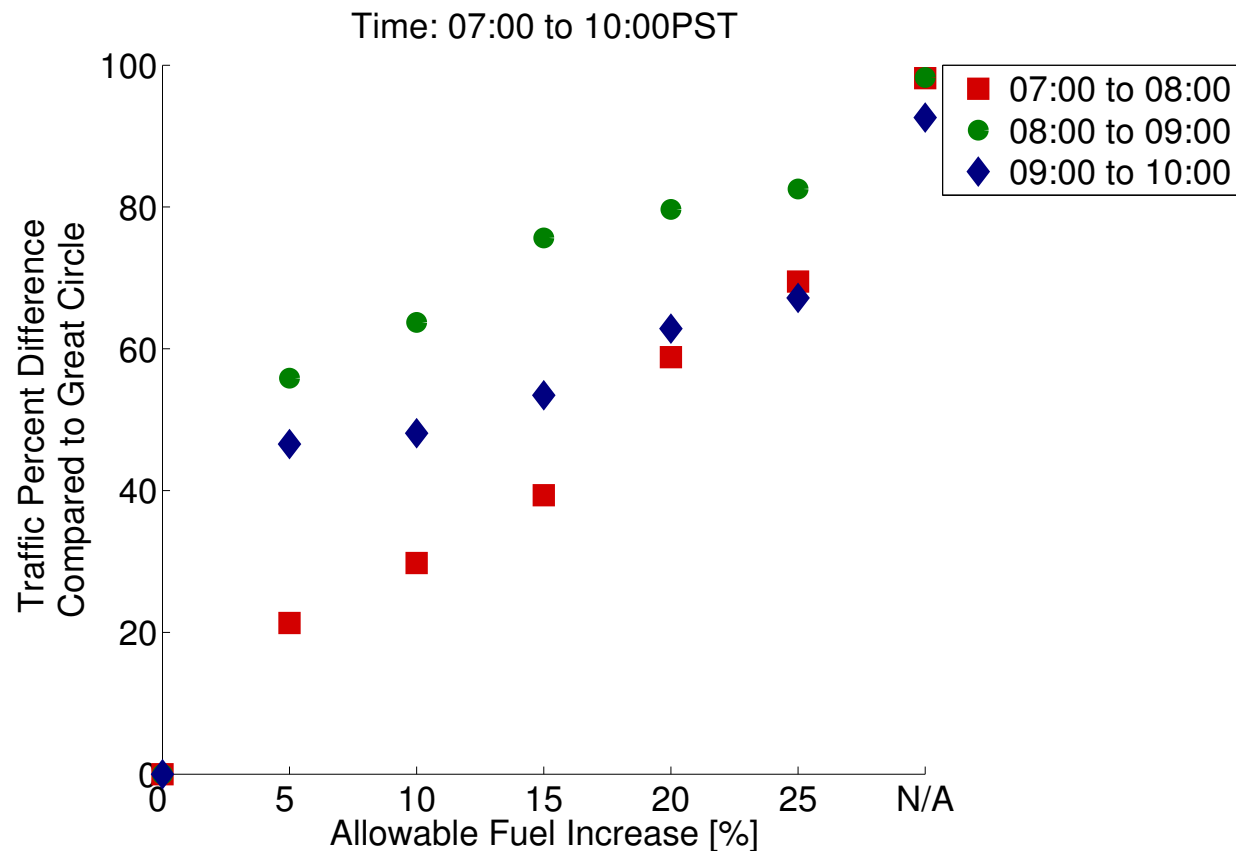




# Traffic Interaction Decrease



NASA Aeronautics Research Institute





# Airspace Emerging Technologies Applicable to Hopper



NASA Aeronautics Research Institute

| Domain                     | Emerging technologies                                                                                                                           |
|----------------------------|-------------------------------------------------------------------------------------------------------------------------------------------------|
| Communication              | Widespread Data Link with limited human oversight                                                                                               |
| Ground operations          | Precision Departure Release Capability (PDRC)                                                                                                   |
| Terminal airspace          | Traffic Management Advisor (TMA)<br>Airborne Dependent Surveillance–Broadcast (ADS-B)                                                           |
| Weather avoidance planning | Dynamic Weather Routes (DWR)                                                                                                                    |
| Enroute                    | Airborne Dependent Surveillance–Broadcast (ADS-B)<br>Strategic ground-based 4D trajectory planning<br>Tactical airborne 4D trajectory execution |



NASA Aeronautics Research Institute

# Network Design and Analysis



# BaySim: Discrete Element Simulation



NASA Aeronautics Research Institute

- Individual passenger and flight agents
- Fast-time simulation of surface and air travel
- PX transition through 12+ daily states from
  - home – work – home
  - Queue up for flights
- Heuristic rules
- Gaussian inputs
- Creates estimates of demand for flights and subsequent inputs for fleet optimization



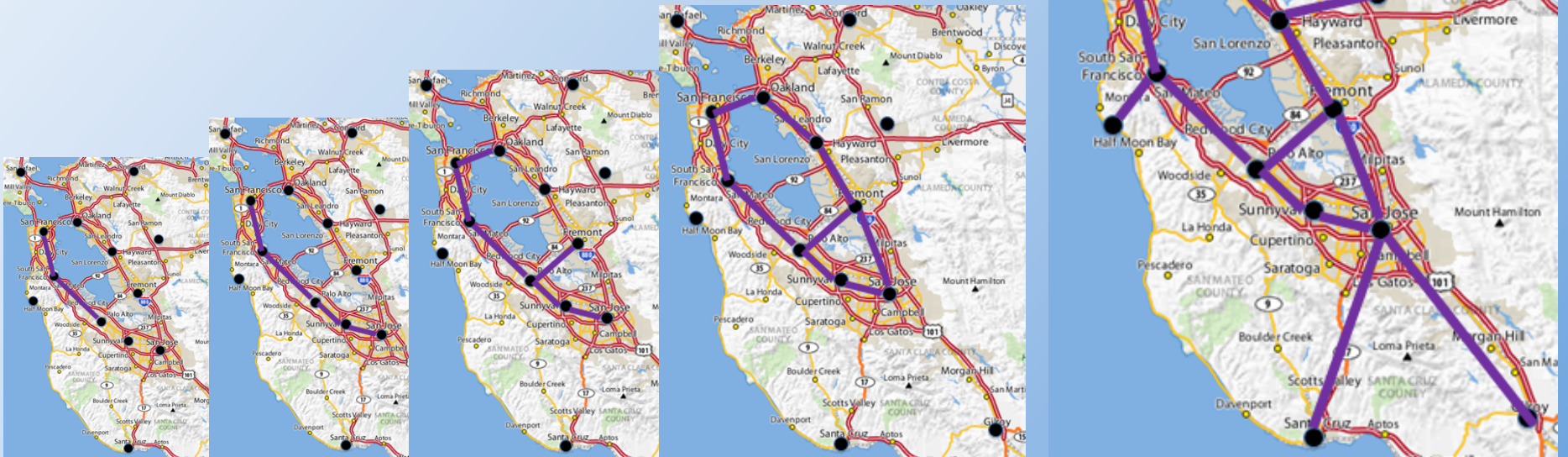


# Network Build-out



NASA Aeronautics Research Institute

- Phase II considers practical network expansion
- Replace Phase I “ $n \times n$ ” network with “ring-spoke”





# BaySim: Passenger Populations



NASA Aeronautics Research Institute

- How many daily passengers?
  - “Tech Industry” employs 386K workers in Bay Area
  - CalTrain serves roughly 42K passengers per day
  - BART serves roughly 370K passengers per day
- Population Sizes: **5K, 10K, 15K, 20, 25K, 30K**
- PX Distribution, starting times, workday length
  - 65% Day            4 to 10 AM       7 to 9 hours
  - 20% Swing        1 to 6 PM        7 to 9 hours
  - 5% Graveyard    9 to 2 AM        7 to 9 hours
  - 10% Other        8 AM to 3 PM    4 to 5 hours



# Population Improvements



NASA Aeronautics Research Institute

- Extensive use of Census and Zip Code databases
  - Zip codes provide appropriate geographic granularity
  - Zip and Census data readily available via websites (i.e., FREE!)
  - 2010 Census data includes commute and business employment demographics
  - Populations within zip codes created via iterative process according to several constraints:
    - Homesite network node  $\neq$  Worksite network node
    - Estimated Flying time < Estimated Driving Time
    - Number of workers at jobsite scaled and limited using local zip code employment statistics
    - Residents and worksites within user-specified distance from station nodes

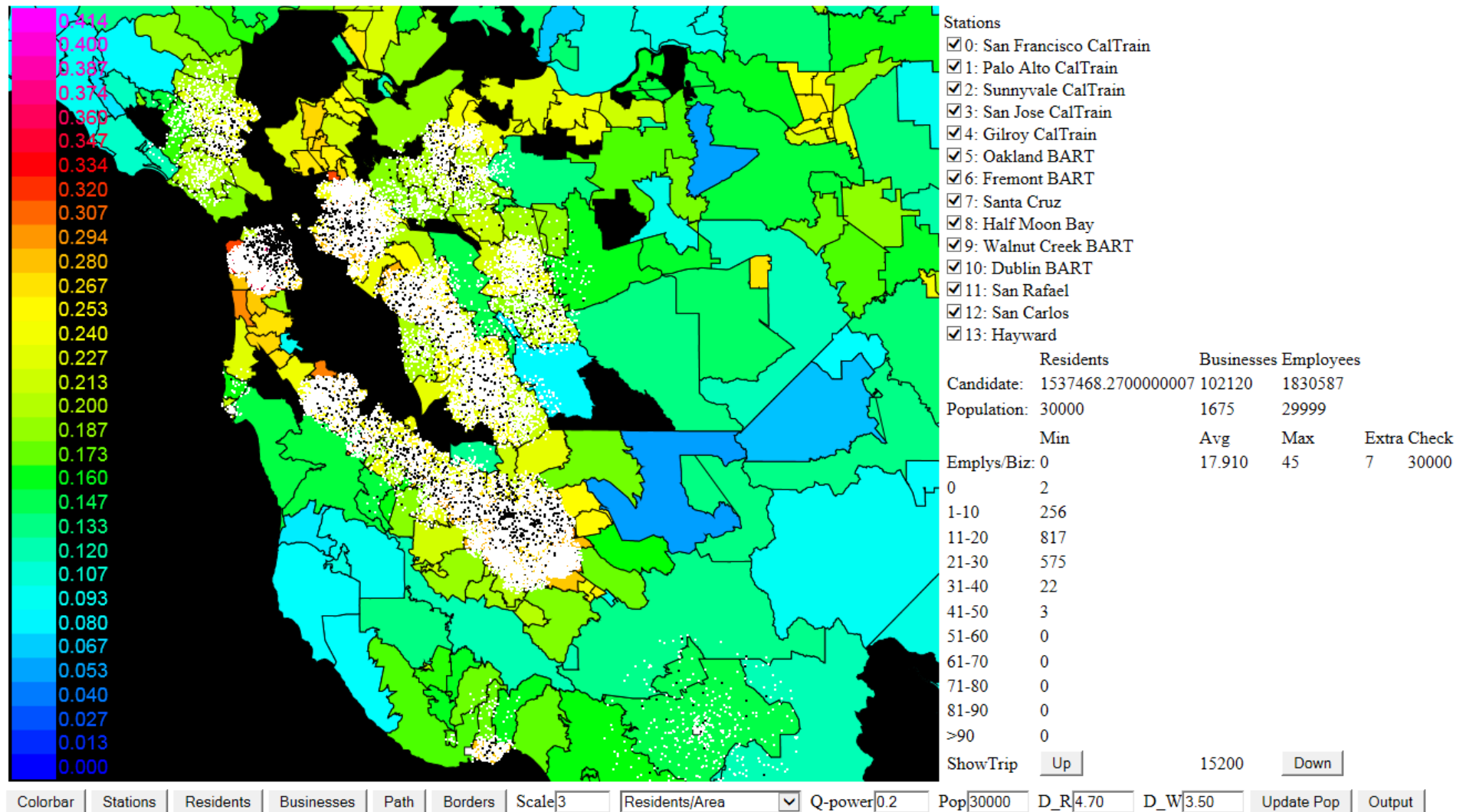




# 30K PX Sample Population



NASA Aeronautics Research Institute





# Dijkstra and Speed Tables



NASA Aeronautics Research Institute

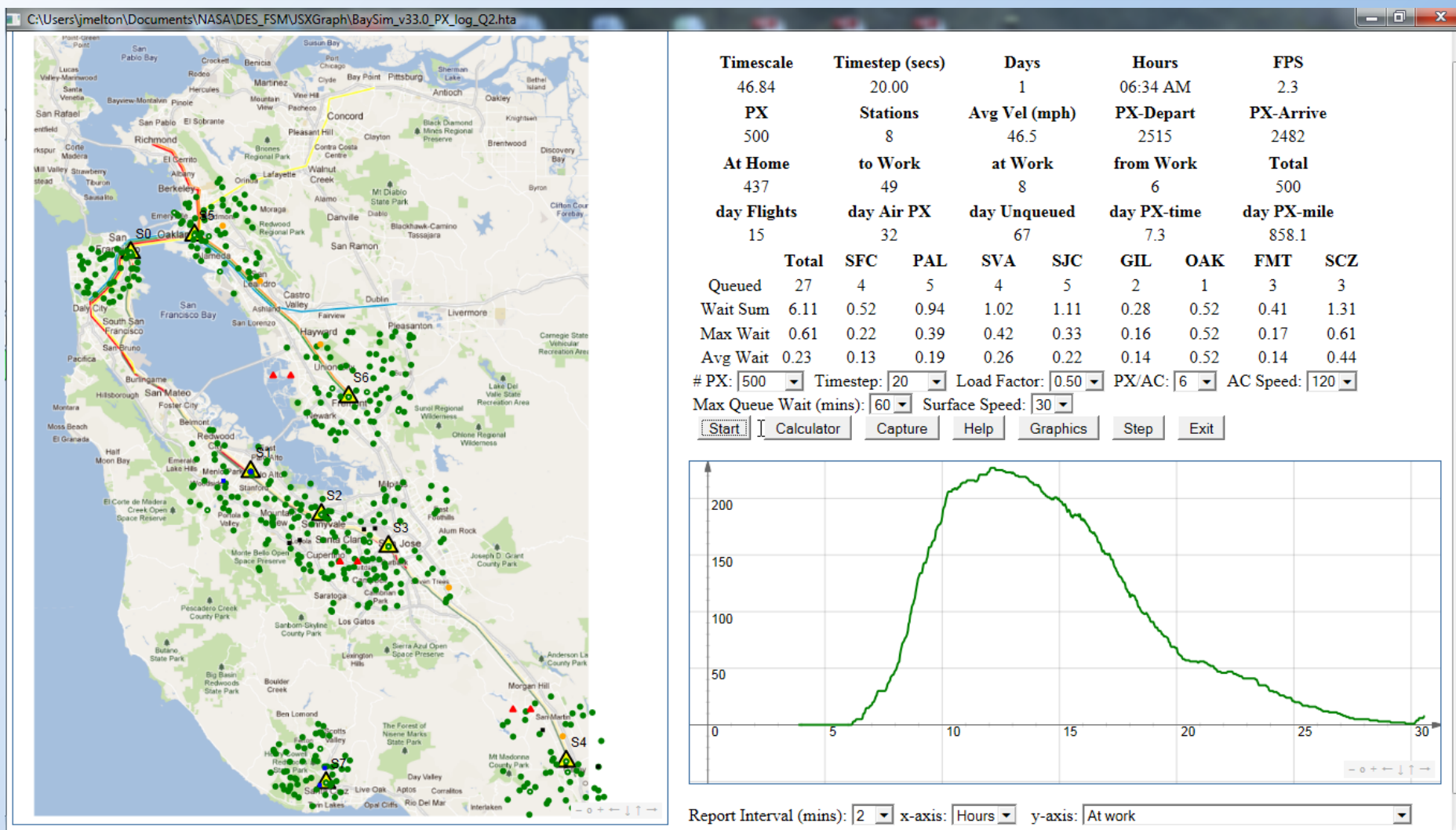
- Each passenger chooses flights attempting to minimize transit time between home and work (including estimates of queue times)
- Dijkstra leg weights (transit times) updated immediately after each flight arrival and best route made available to each passenger before queuing
- Leg transit times from hourly table lookup, based on routes that minimize traffic conflicts with current SJC, SFO, and OAK patterns



# BaySim Phase II Sample Movie



NASA Aeronautics Research Institute



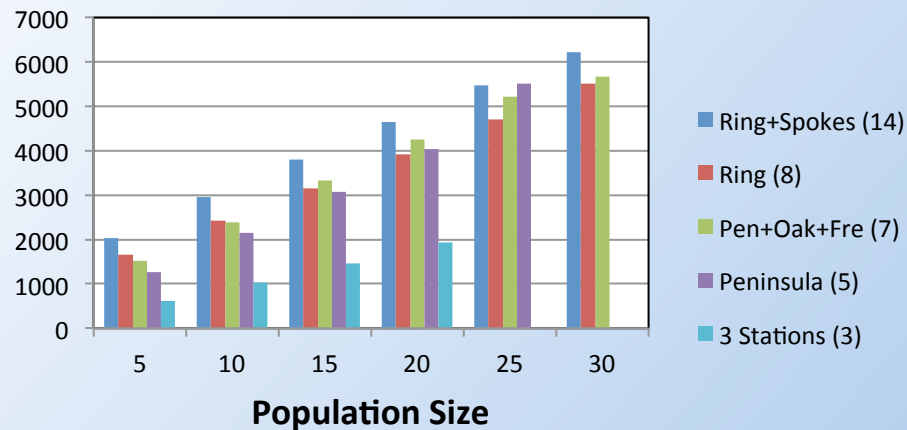




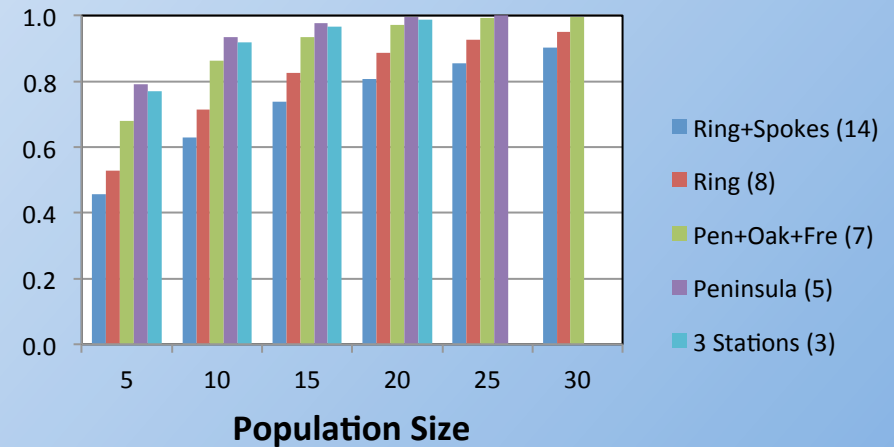
# Network Statistics

NASA Aeronautics Research Institute

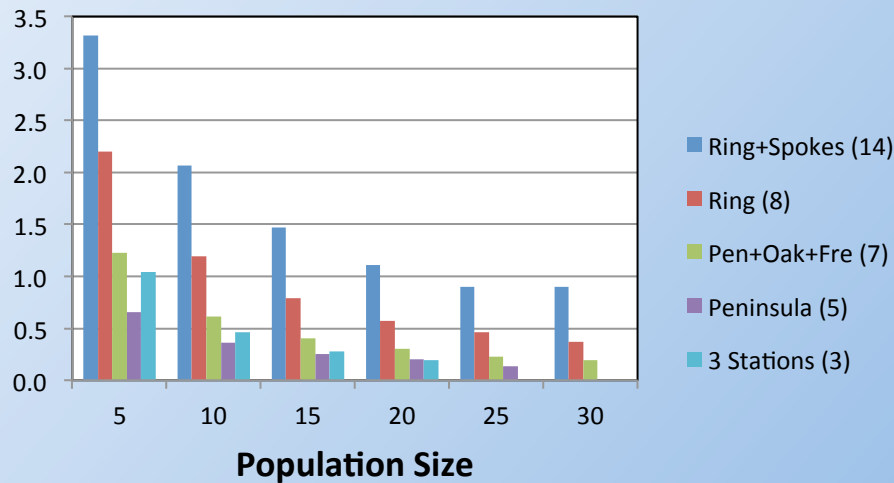
## Number of Flights per Day



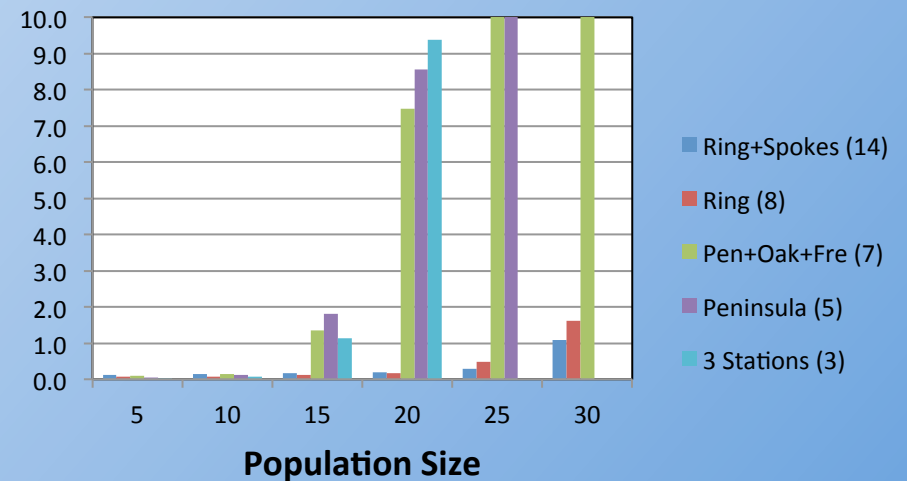
## Average Load Factor



## Average PX Queue Delay in Minutes



## Average Flight Departure Spacing Delay in Minutes



Results with minimum departure spacing of 60 seconds

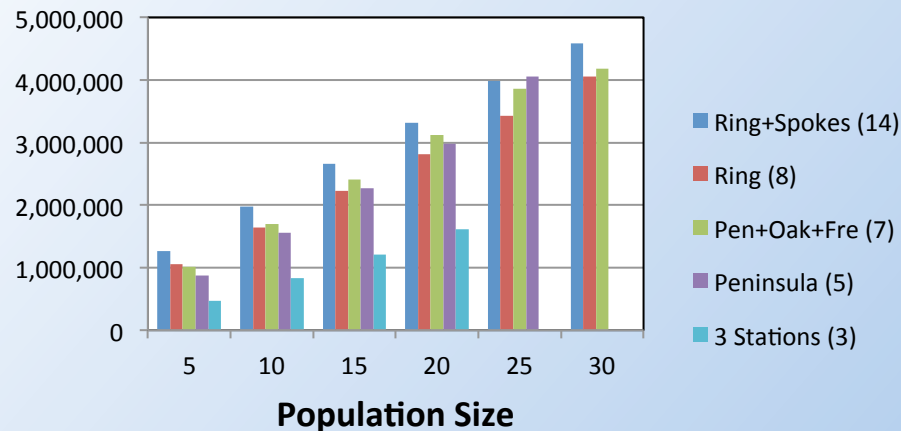


# Network Energy Usage

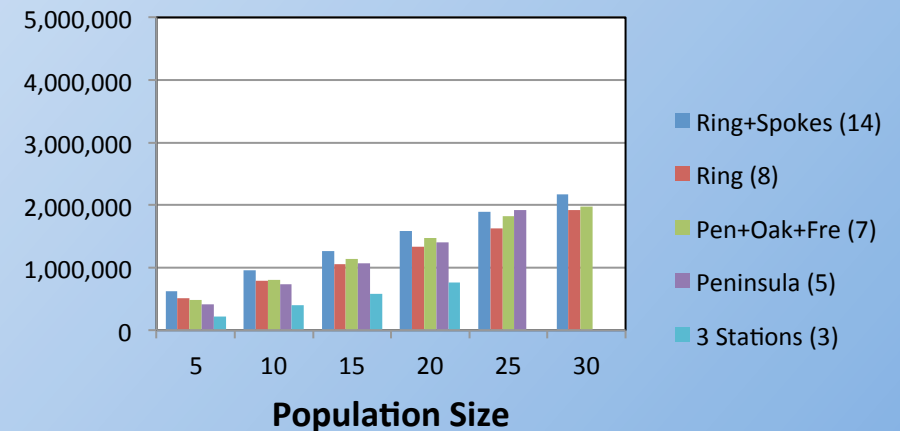


NASA Aeronautics Research Institute

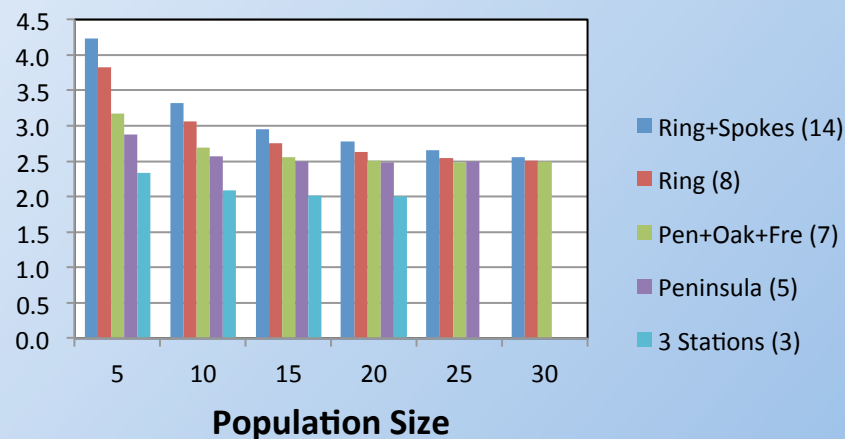
## Total Energy Use in kW-h: Gas



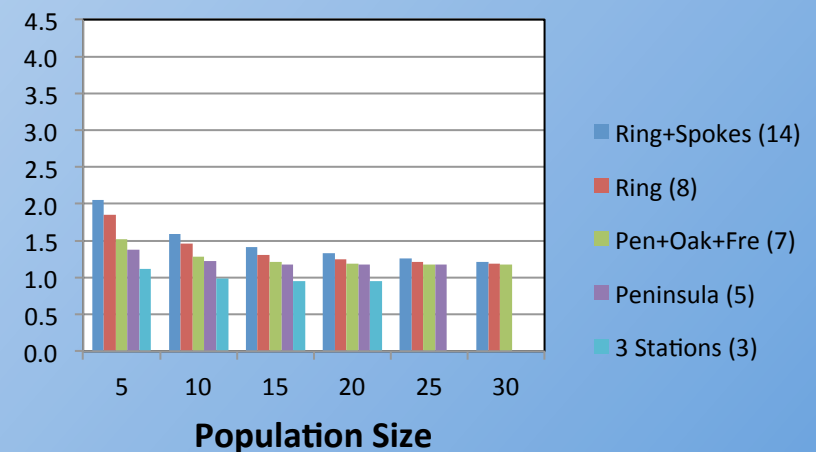
## Total Energy Use in kW-h: Battery



## kW-h/(PX-mile): Gas



## kW-h/(PX-mile): Battery



Results with minimum departure spacing of 60 seconds



NASA Aeronautics Research Institute

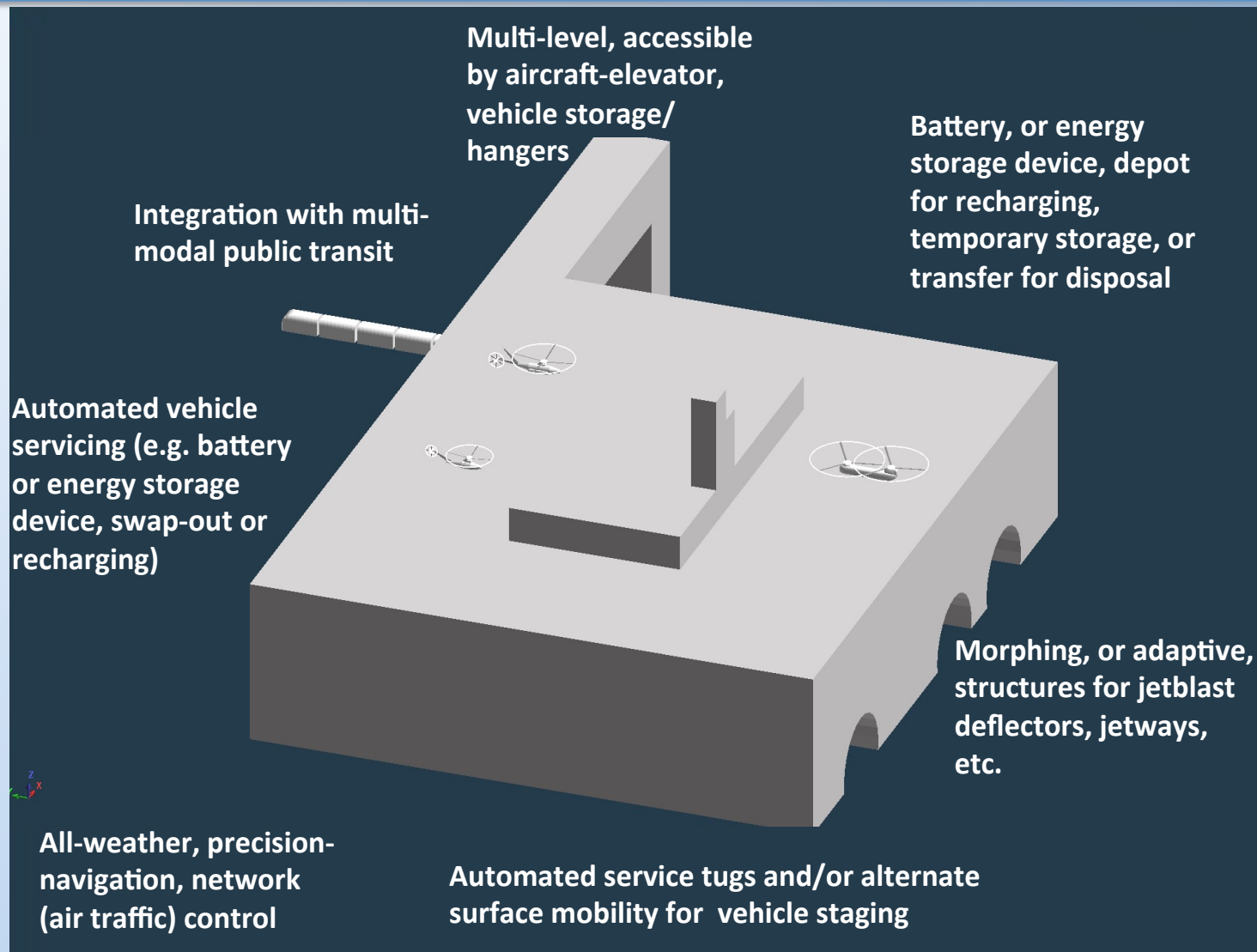
# Operations & Misc. Tech Issues



# Vertiport Station Operations



NASA Aeronautics Research Institute

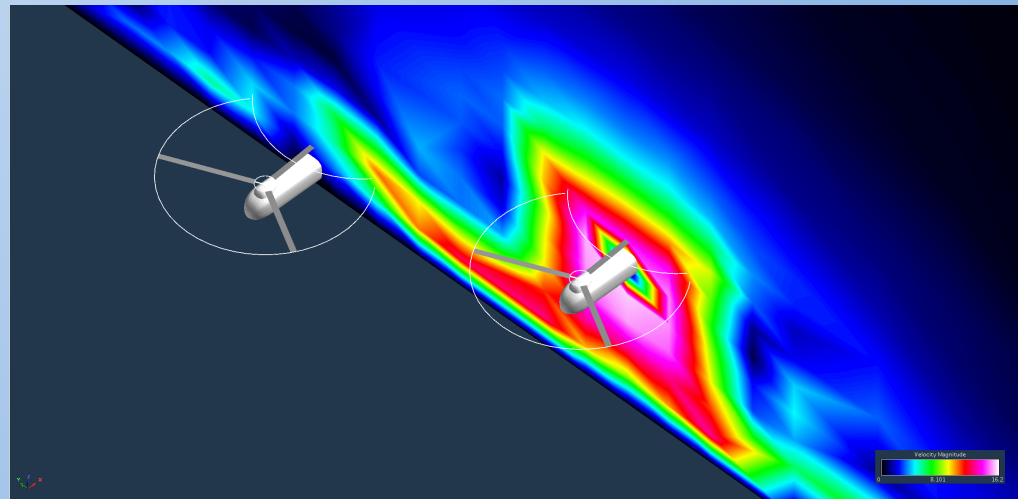
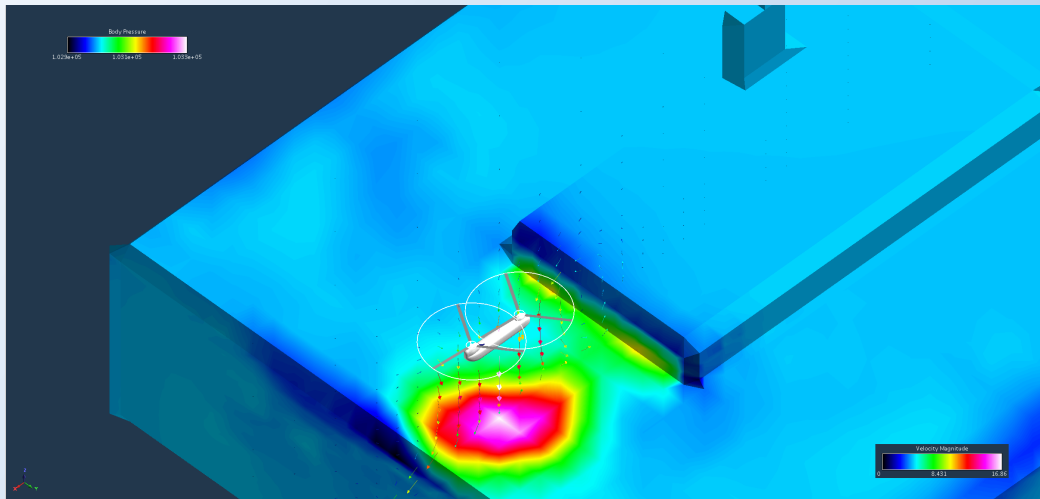




# Rotor Wake Interactions



NASA Aeronautics Research Institute



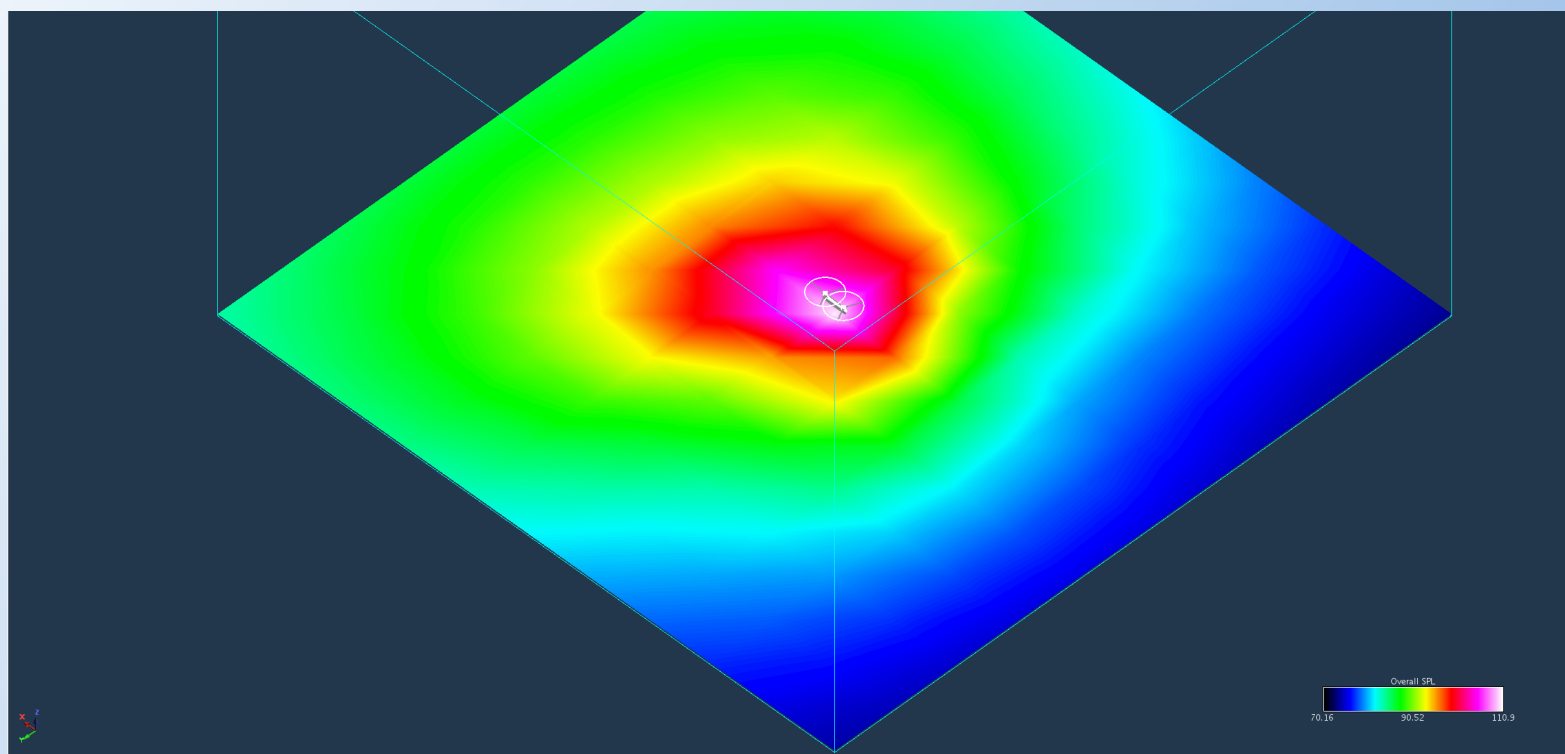




# Preliminary Noise Assessments



NASA Aeronautics Research Institute





# Future Work & Tech Transition



NASA Aeronautics Research Institute

- Some of the electric-propulsion models developed are finding their way into Stanford's SAUVE vehicle conceptual design tool and NASA's NDARC
- Two NASA TM's are being drafted for publication
- Internal electric-propulsion discussion within ARMD Rotary-Wing project
- Hopper helps make a case for vehicle/systems autonomy research

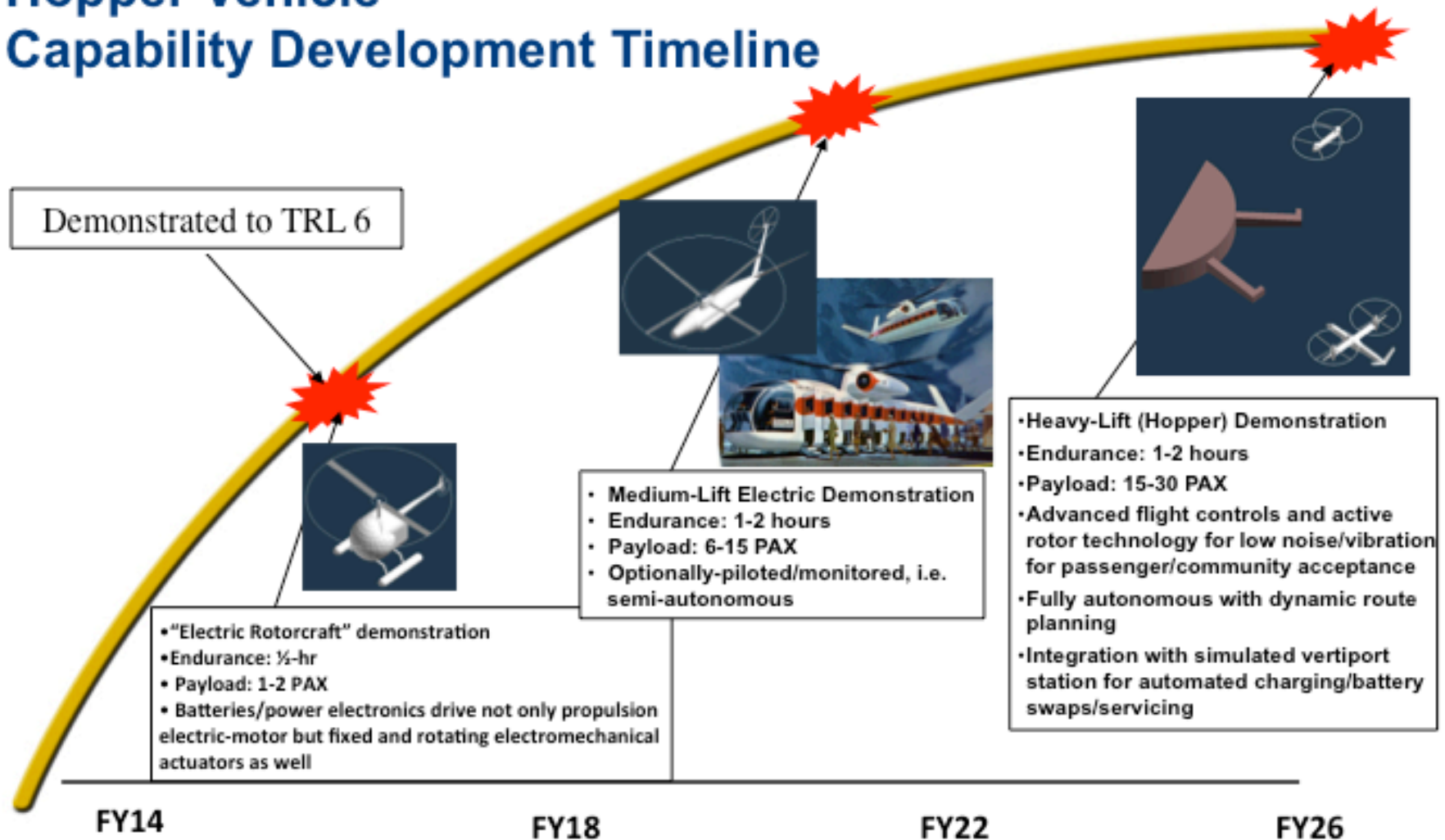


# Hopper Capabilities



NASA Aeronautics Research Institute

## Hopper Vehicle Capability Development Timeline





# Concluding Remarks



NASA Aeronautics Research Institute

- Large numbers of medium- to large- A/C are required to serve assumed passenger levels
- Flying and VTOL are energy intensive transportation modes for short distances and, yet, represent a potential way to bypass urban surface transportation congestion
- While originally conceived as "light infrastructure" (no new roads/rail-lines), the vertiport station power requirements and real estate footprints of ramps and terminals may require require more infrastructure than initially anticipated when large numbers of aircraft are involved
  - Trucks will be required if charged batteries are to be delivered to stations (and discharged batteries removed for offsite recharging);
  - high-power electrical transmission must be provided to stations if charging is onsite.



# Concluding Remarks, Cont.



NASA Aeronautics Research Institute

- Acoustics and other over-flight issues will require technology innovations and public/neighborhood outreach. Will neighborhood populations accept a new type of traffic noise?
- The development of Discrete Event Simulations has value not only in producing quantitative design tradeoff information, but also in 1) uncovering and examining detailed operational policies surrounding transportation networks, and 2) understanding behavioral and demographic issues for the populations they are intended to serve.





# Concluding Remarks, Cont.



NASA Aeronautics Research Institute

- May be best to operate network independently of other air transport that transits outside of the Bay Area, instead of “actively intermeshing”
- Emerging technological and procedural advances, currently being proposed under the Next Generation Air Transportation System (NextGen), could be leveraged to facilitate the realization of the Hopper concept.



# Concluding Remarks, Cont.



NASA Aeronautics Research Institute

- Relatively near-term battery / hybrid / fuel cell technologies could make these short range vehicles realizable within the next 10 years
- Near-term battery technology (500-600 W h / kg at reasonable power densities) will satisfy power and energy requirements for these short range vehicles: more advanced Li-air batteries will not lead to further weight reductions.



# Concluding Remarks, Cont.



NASA Aeronautics Research Institute

- Discrete event simulations and schedule optimizations reveal the possibility of carrying a substantial daily passenger load within the Bay Area.
- Study has identified challenges in integrating Hopper flights and Bay Area commercial traffic that can be partially tackled with Hopper aircraft time-of-day dynamic routing as well as NextGen technologies