

National Aeronautics and Space Administration

Simultaneous Localization and Mapping for Satellite Rendezvous and Proximity Operations Using Random Finite Sets

Lauren Schlenker & Mark Moretto
David Gaylor, Richard Linares



NASA Goddard Space Flight Center
MISSION ENGINEERING AND SYSTEMS ANALYSIS DIVISION

www.nasa.gov



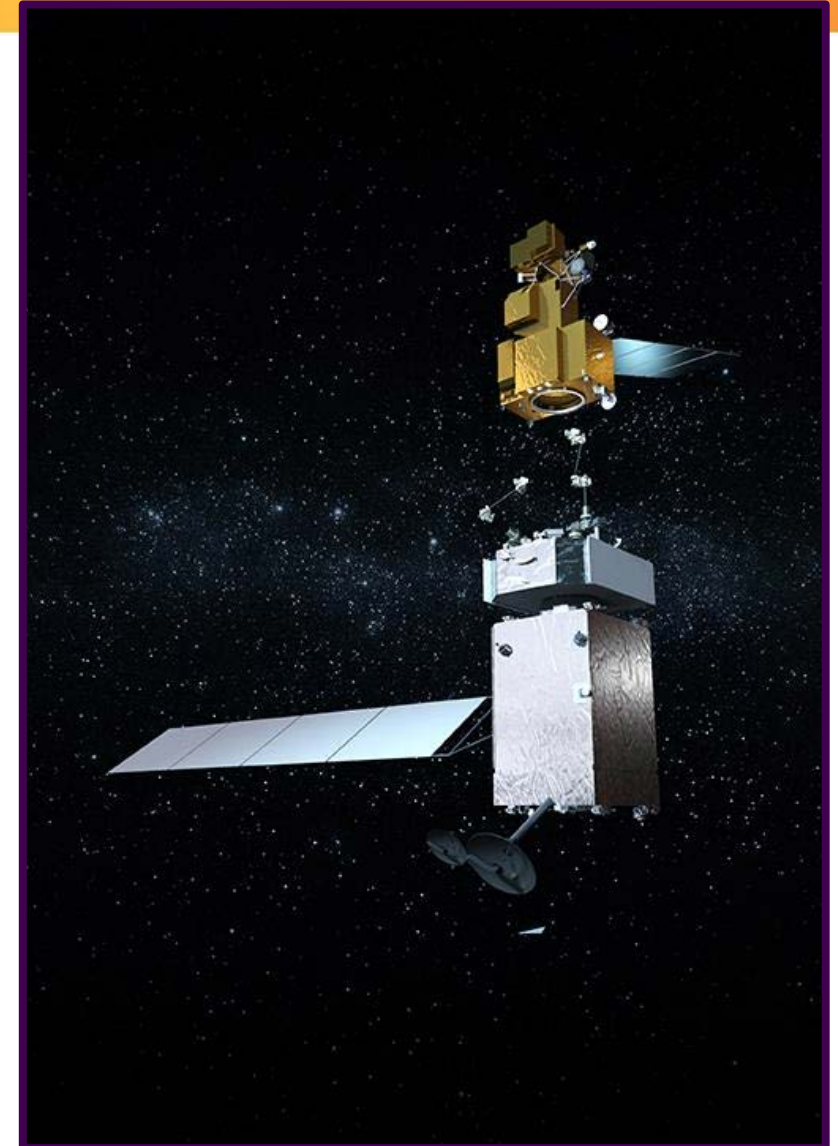
Code 590



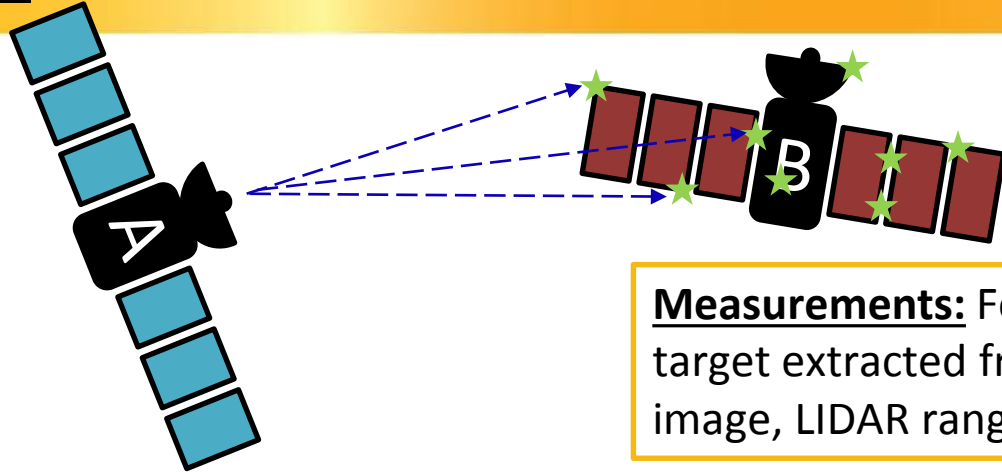
Summary:

- **Autonomous spacecraft relative navigation:** a necessary requirement of future exploration/servicing missions.
- **Challenging environments:** no *a priori* map, need to track multiple features using measurements that can be noisy, have extraneous measurements (clutter), and missed detections.
- **Random Finite Set based filters:** a recent development, specifically formulated for these kinds of problems.
- **Initial simulations:** *first known demonstration* of a Random Finite Set based filter for spacecraft pose estimation and mapping.

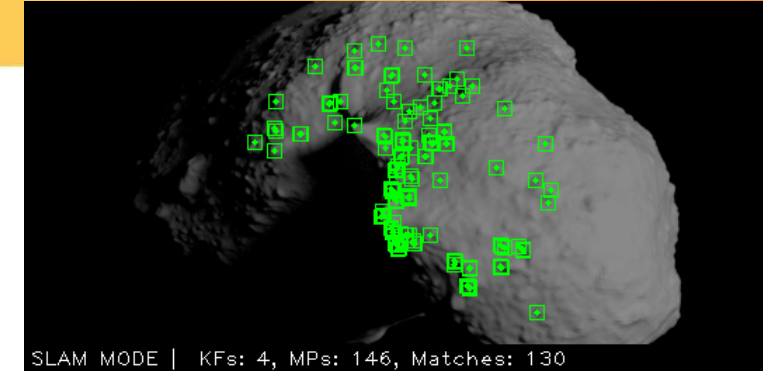
- **Future of space exploration missions:** spacecraft must be able to *autonomously* navigate their environment.
 - Rendezvous with non-cooperative satellite (satellite servicing).
 - Satellite swarms/formation flying.
 - Small body missions (asteroids, comets).
- **This is a hard problem:**
 - Challenging dynamics
 - Multi-Target Tracking: many things to track
 - Measurement Limitations/Issues:
 - Lighting conditions
 - Features enter/exit FOV
 - Extraneous measurements (clutter)
 - Missed detections
 - Lack of *a priori* information
 - No *a priori* map or *a priori* map has significant uncertainty



- **Problem:** Estimate location and rates of observer relative to client.



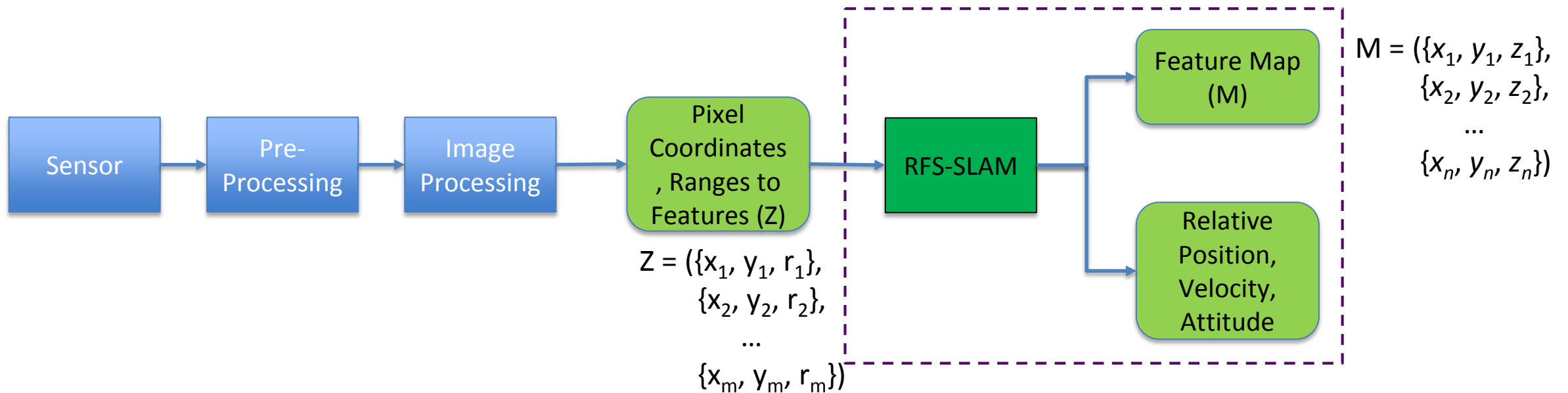
Measurements: Features on target extracted from optical image, LIDAR range to features



Features extracted from an asteroid image.

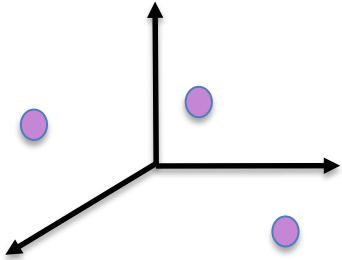
Credit: Kulumani, 2017

- **Solution:** Formulate as a Simultaneous Localization and Mapping (SLAM) problem.
 - Localization = estimating observer's pose (position, velocity, attitude relative to client)
 - Mapping = estimating feature map (points or edges on a client – spacecraft or asteroid)
 - Simultaneously: pose depends on the map and vice versa
 - **More Problems:** Traditional SLAM methods often diverge due to issues with data association, high uncertainty.
 - *False sensor returns (clutter.)*
 - *Missed detection of expected features.*
 - *Map features entering/exiting Field of View.*

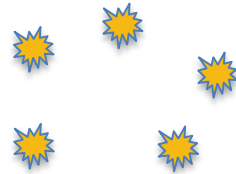
Block Diagram:

Big Problem: Data association is hard.

States: $t = k-1$



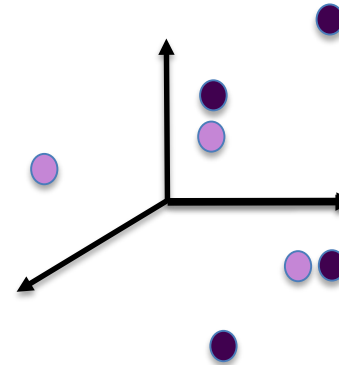
Measurements: $t = k-1$



Extra “clutter”
measurements, so how do
you match measurements
and targets?



States: $t = k$



How do you know what moved where?
What if something new entered? What if
something from before left?

Measurements: $t = k$

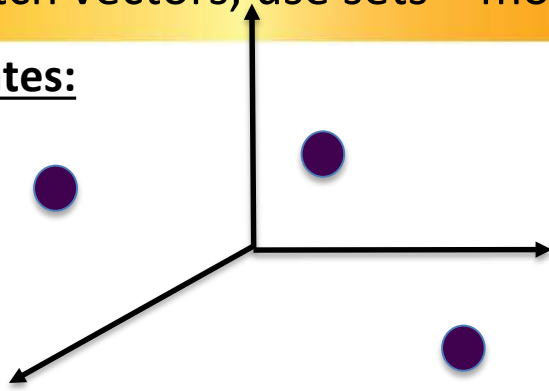


What if you fail to detect
some things?

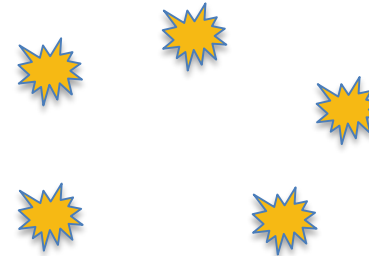
- The Kalman Filter was originally developed for single-target tracking. Must associate measurements with targets outside of the filter.
 - Heuristic methods have been used with the Kalman Filter to handle multi-target tracking.
 - Divergence if association is wrong.
 - Computationally expensive – especially as number of targets/measurements increases.
 - For 12 features and 20 measurements, data association matrix is 43GB, Matlab won't even initialize ☹

Reformulate: Ditch vectors, use sets – more natural, unified framework

True States:



Measurements:



Random Vectors:

$$x_1 = \begin{bmatrix} 4 \\ 5 \\ 6 \end{bmatrix} \quad x_2 = \begin{bmatrix} 1 \\ 3 \\ 2 \end{bmatrix} \quad x_3 = \begin{bmatrix} 2 \\ 1 \\ 5 \end{bmatrix}$$

$$z_1 = \begin{bmatrix} 4 \\ 1 \\ 3 \end{bmatrix} \quad z_2 = \begin{bmatrix} 1 \\ 8 \\ 2 \end{bmatrix} \quad z_3 = \begin{bmatrix} 7 \\ 4 \\ 6 \end{bmatrix} \quad z_4 = \begin{bmatrix} 8 \\ 1 \\ 6 \end{bmatrix} \quad z_5 = \begin{bmatrix} 9 \\ 0 \\ 4 \end{bmatrix}$$

Random Finite Sets:

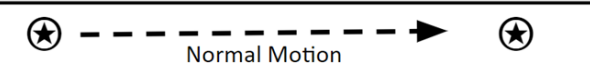
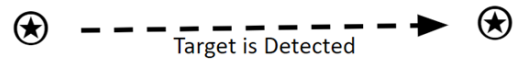
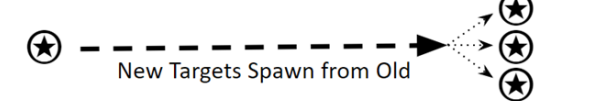
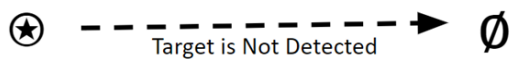
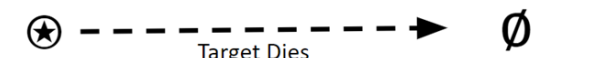
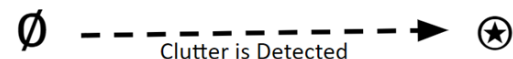
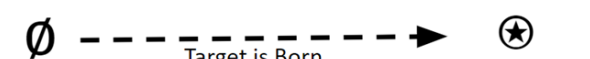
$$\mathcal{X} = \left\{ \begin{bmatrix} 4 \\ 5 \\ 6 \end{bmatrix}, \begin{bmatrix} 1 \\ 3 \\ 2 \end{bmatrix}, \begin{bmatrix} 2 \\ 1 \\ 5 \end{bmatrix} \right\} = \left\{ \begin{bmatrix} 2 \\ 1 \\ 5 \end{bmatrix}, \begin{bmatrix} 4 \\ 5 \\ 6 \end{bmatrix}, \begin{bmatrix} 1 \\ 3 \\ 2 \end{bmatrix} \right\} = \left\{ \begin{bmatrix} 1 \\ 3 \\ 2 \end{bmatrix}, \begin{bmatrix} 4 \\ 5 \\ 6 \end{bmatrix}, \begin{bmatrix} 2 \\ 1 \\ 5 \end{bmatrix} \right\}$$

$$\mathcal{Z} = \left\{ \begin{bmatrix} 4 \\ 1 \\ 3 \end{bmatrix}, \begin{bmatrix} 1 \\ 8 \\ 2 \end{bmatrix}, \begin{bmatrix} 7 \\ 4 \\ 6 \end{bmatrix}, \begin{bmatrix} 8 \\ 1 \\ 6 \end{bmatrix}, \begin{bmatrix} 9 \\ 0 \\ 4 \end{bmatrix} \right\}$$

*Doesn't matter which order you put elements in, it's still the same set.
This is very powerful, if you have math to manipulate sets (we do.)*

- **Random Finite Sets (RFS) are a more natural way of formulating the general SLAM problem.**

- Vector-based formulation: existing target expected to continue to exist, and expected to be measured.
- RFS formulation: other general propagation and measurement situations can be handled directly.

Generalized Prediction Step Options: $\mathcal{X}_{k-1} \dashrightarrow \mathcal{X}_k$	Generalized Measurement Step Options $\mathcal{X}_k \dashrightarrow \mathcal{Z}_k$
	
	
	
	
$\mathcal{X}_k = \left(\bigcup_{\mathcal{X}_{k-1}} \text{Surviving} \right) \cup \left(\bigcup_{\mathcal{X}_{k-1}} \text{Spawned} \right) \cup \left(\text{Birthed} \right)$	$\mathcal{Z}_k = \left(\bigcup_{\mathcal{X}_k} \text{Detections} \right) \cup \left(\text{Clutter} \right)$

These other “options” usually prevent traditional SLAM approaches from converging.

- **No data association required between measurements and targets!**

- Dramatically reduces computational complexity, no chance of diverging due to incorrect association.

- **No prior knowledge of the environment required!**

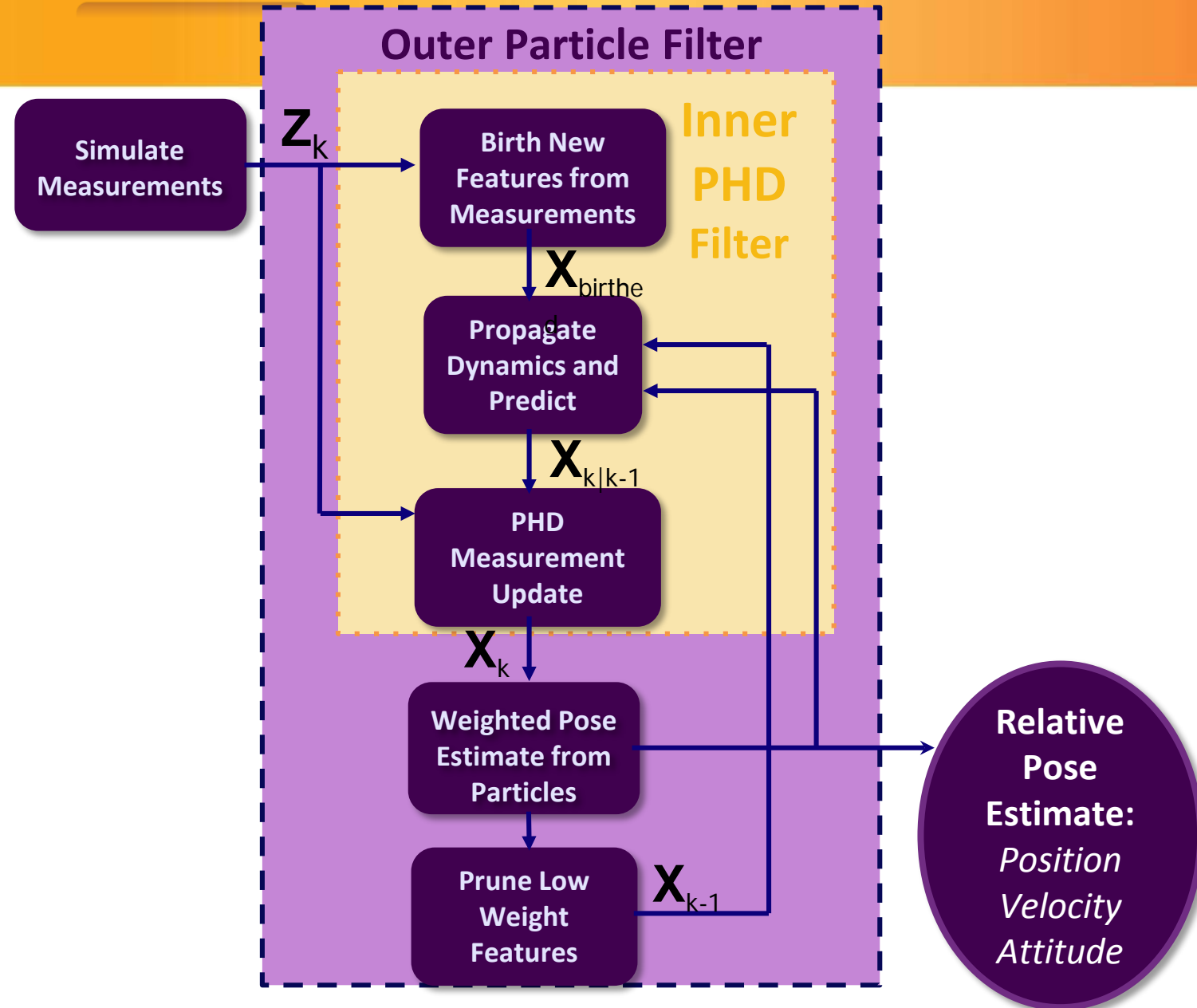
- Ideal for non-cooperative rendezvous scenarios, no map of the target may be available.

- **Probability Hypothesis Density Filter (PHD Filter):**
 - *Proposed by Mahler in the 1990's, has been used for ground and naval robotics since the early 2000's*
 - *Very similar to Kalman Filter:*
 - Optimal Bayes Filter
 - Linear Gaussian dynamics and measurement models
 - Equations are set theoretic analogs to Kalman filter prediction and update steps
 - *The differences make it more general and flexible for multi-target tracking in realistic environments:*
 - Can model probabilities of detection/survival, clutter, etc. directly in the mathematics
 - Add a set of “clutter” to your measurement set
 - Add a set of “birthed targets” to your state set

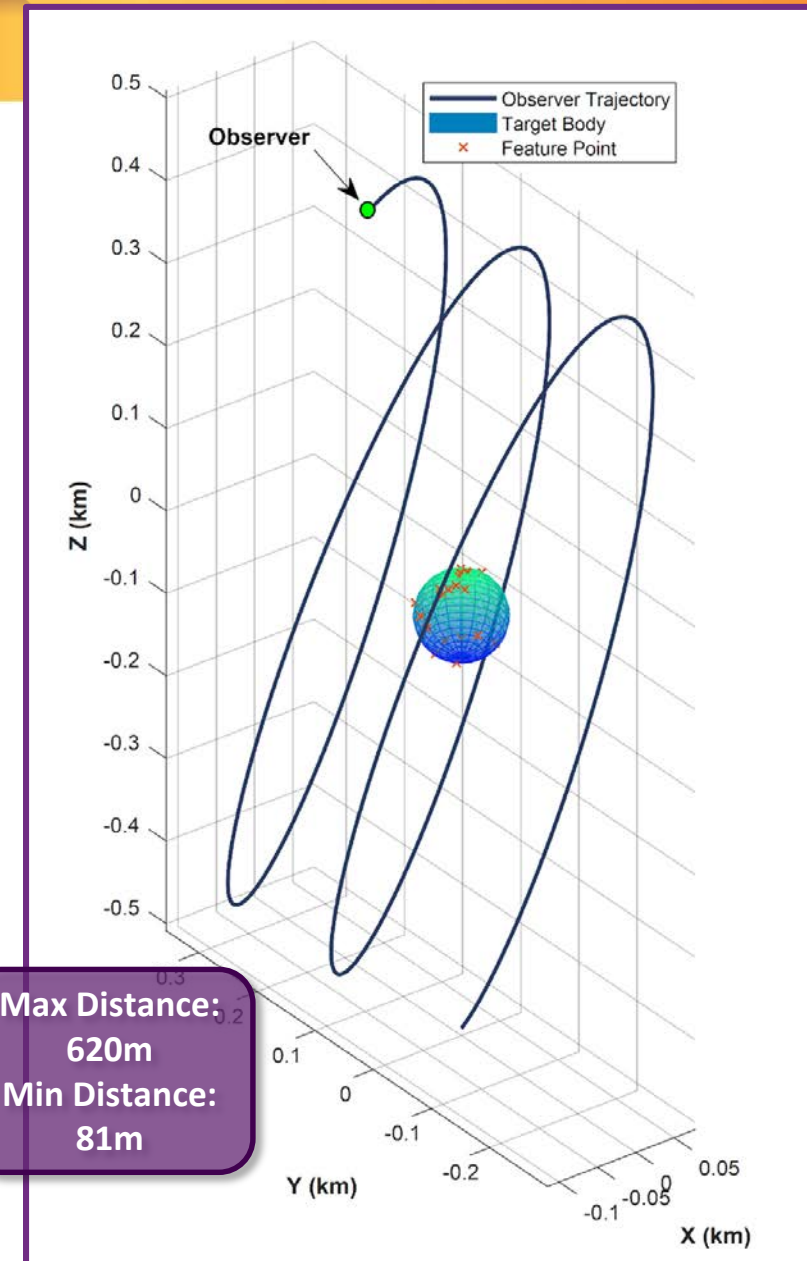
RFS-SLAM Filter Structure:

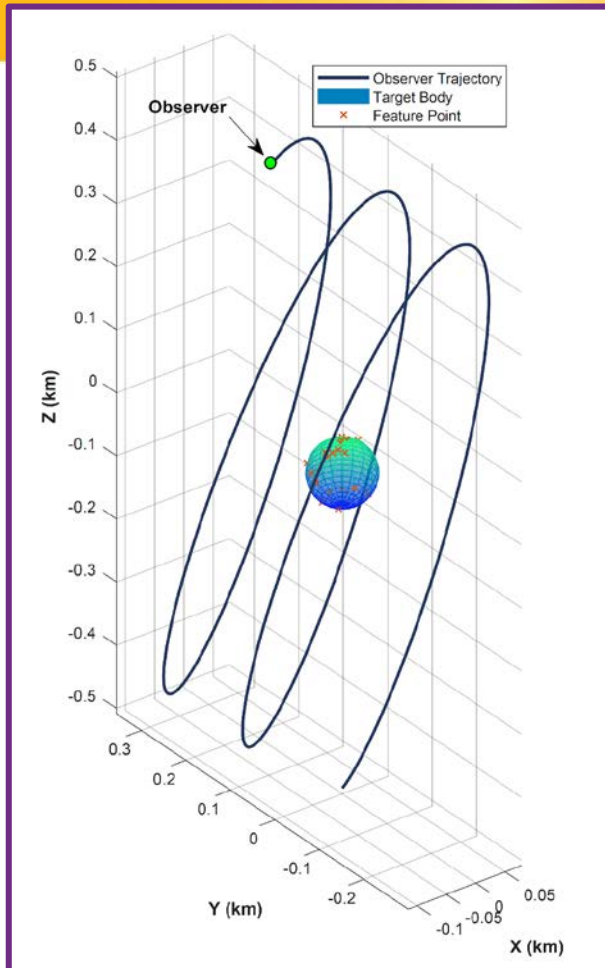
PHD filter wrapped inside a particle filter

- Pose is represented by particles
- **Mapping:** PHD filter determines which cluttered, noisy measurements correspond to actual map features.
 - Each particle's feature map is conditioned on a different pose.
- **Localization:** Quality of each feature map determines the "weight" of each particle (\mathbf{W}_k).
 - Weighted average of pose can be calculated.
 - Alternatively, maximum *a posteriori* estimate.



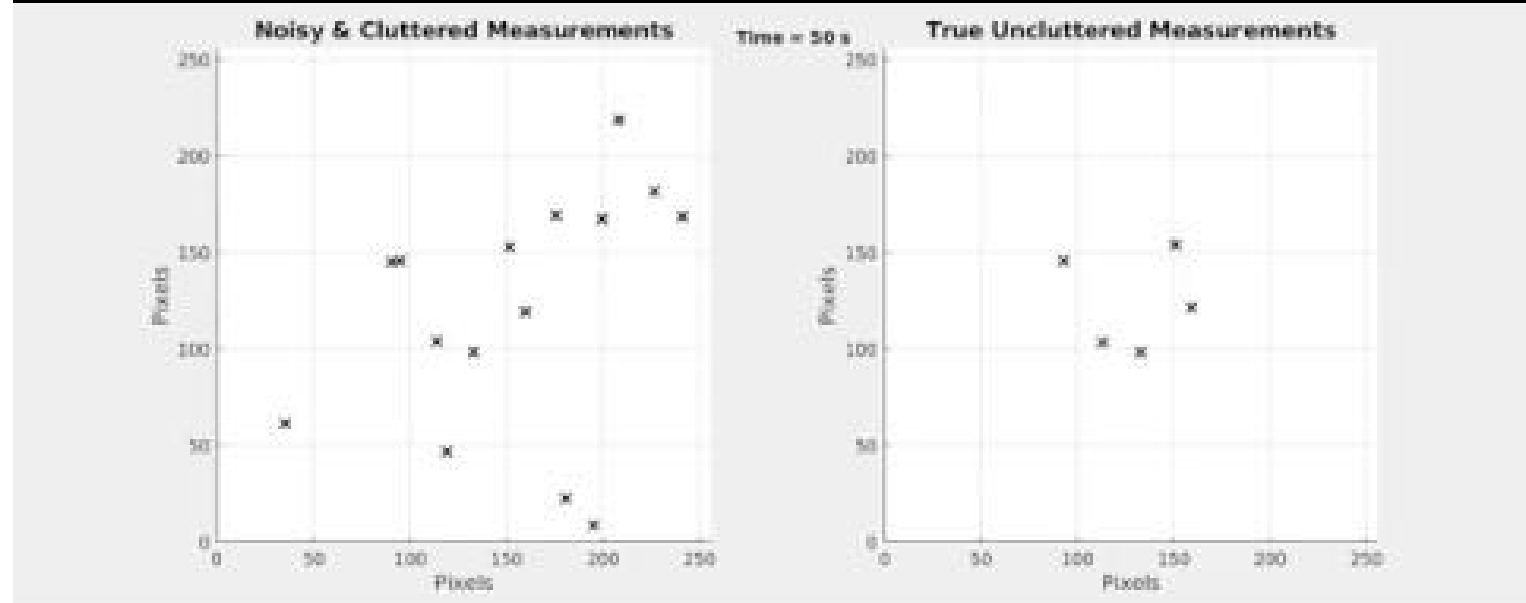
- Target: Sphere, radius 50m, with randomly distributed features
 - *Chosen to simplify simulation of feature occlusion*
 - Simple dot products to determine which simulated measurements are visible
- Observer: Attitude assumed to always point at target
 - Attitude control keeps target centered in sensor field of view (assumed perfect)
- Orbital Dynamics: Clohessy-Wiltshire equations for relative motion
 - Target attitude is constant in the CW frame
- Attitude: Modified Rodrigues Parameters
 - 1 easily avoidable singularity
- Measurements: flash LIDAR (simulated)
 - 256x256 pixel image, 14° Angle of View (similar to Raven)
 - Features = pixel coordinates and range relative to boresight
- Estimation: in target body-fixed frame
 - Map = body-fixed feature locations (static)
 - Pose = observer position, velocity, attitude relative to target





Measurement Noise: [1 pixel, 1 pixel, 10 mm]

[Click to Play Video -- newMeasModel.avi](#)

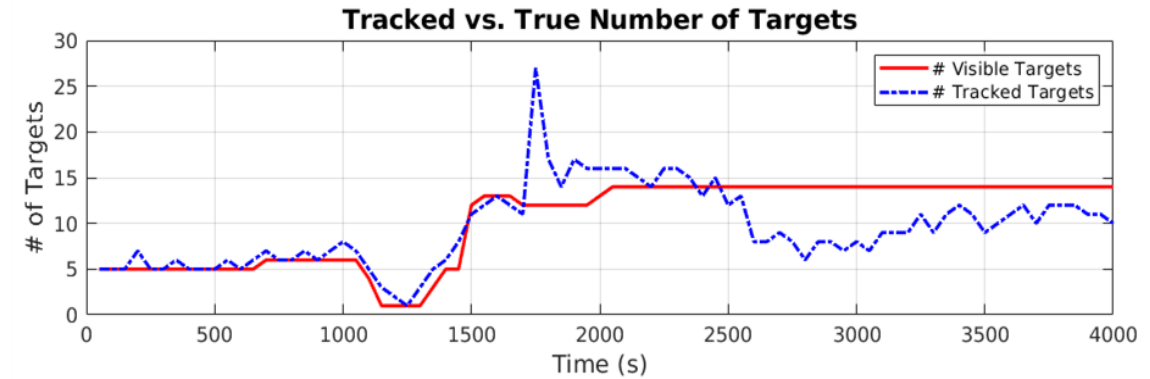


Note: This video shows the image plane; range measurements are also used, but are difficult to visualize.

Case 1: Nominal Orbit

[Click to Play Video -- Case1.avi](#)

Number of Particles Used: 100

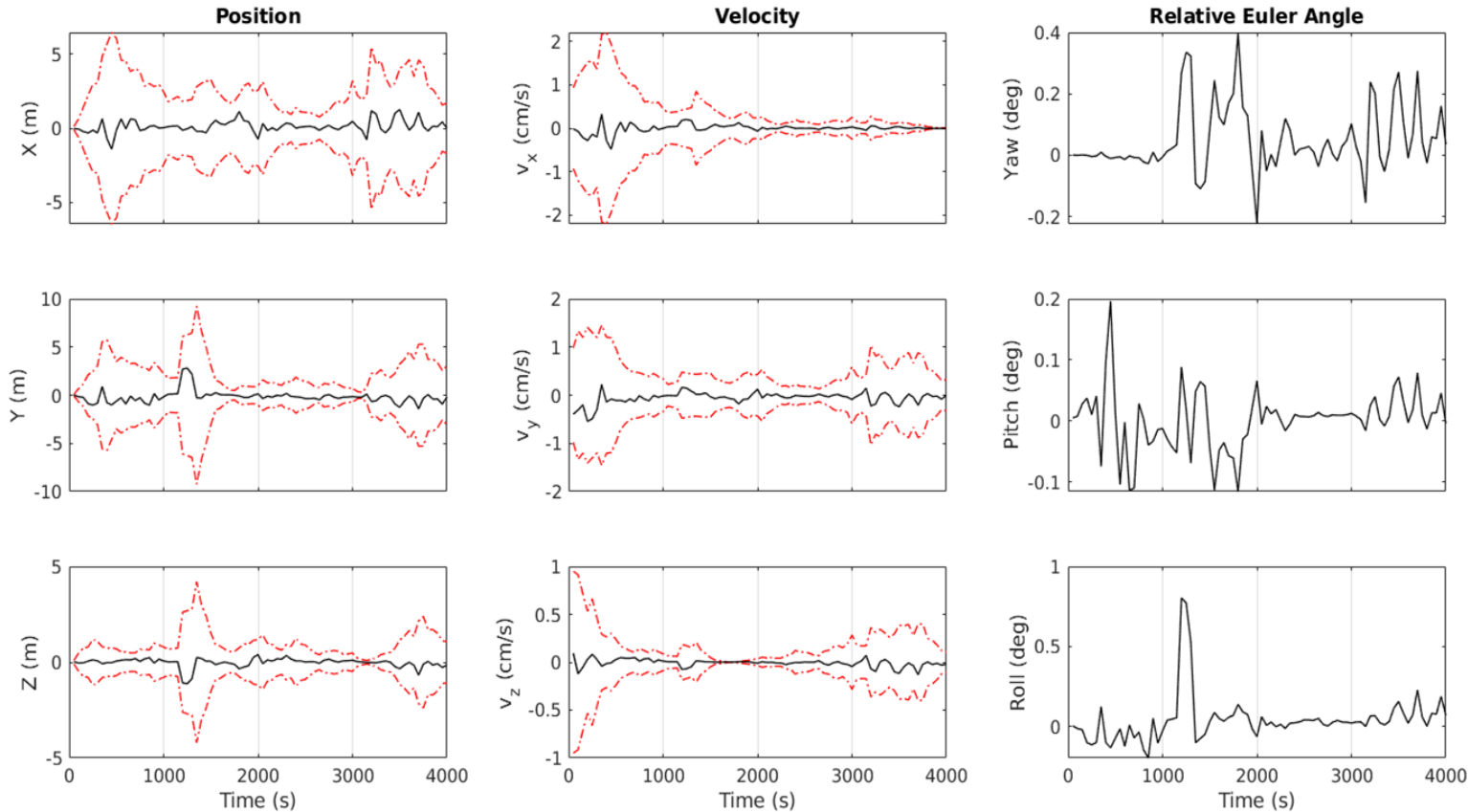


If the filter is well-tuned, the cardinality of the PHD will closely match the number of visible features.

Case 1: Nominal Orbit

Pose Estimate Error vs. Sample Covariance 3σ Bounds

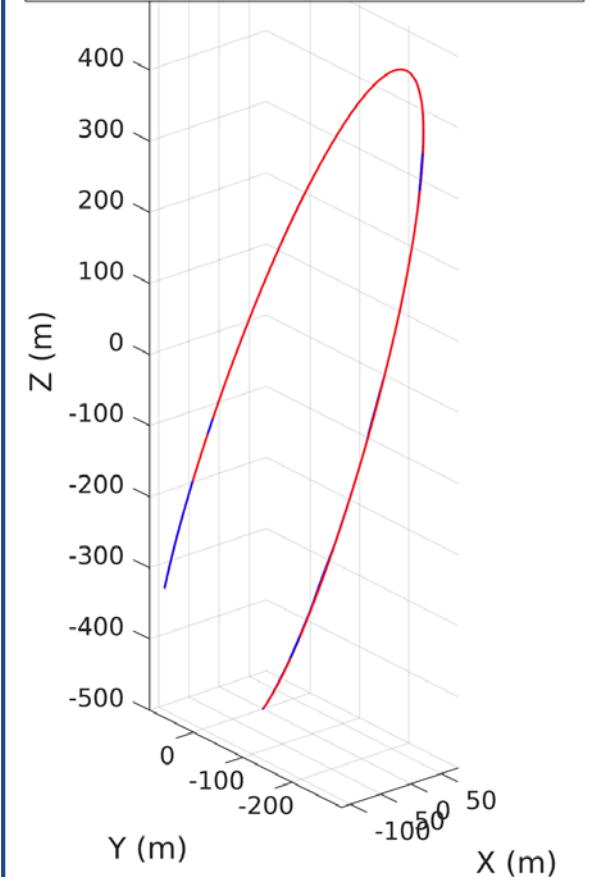
Estimate Error — 3σ Bound



Pose estimate stays well within sample covariance 3σ bounds, and estimated trajectory matches the truth very closely.

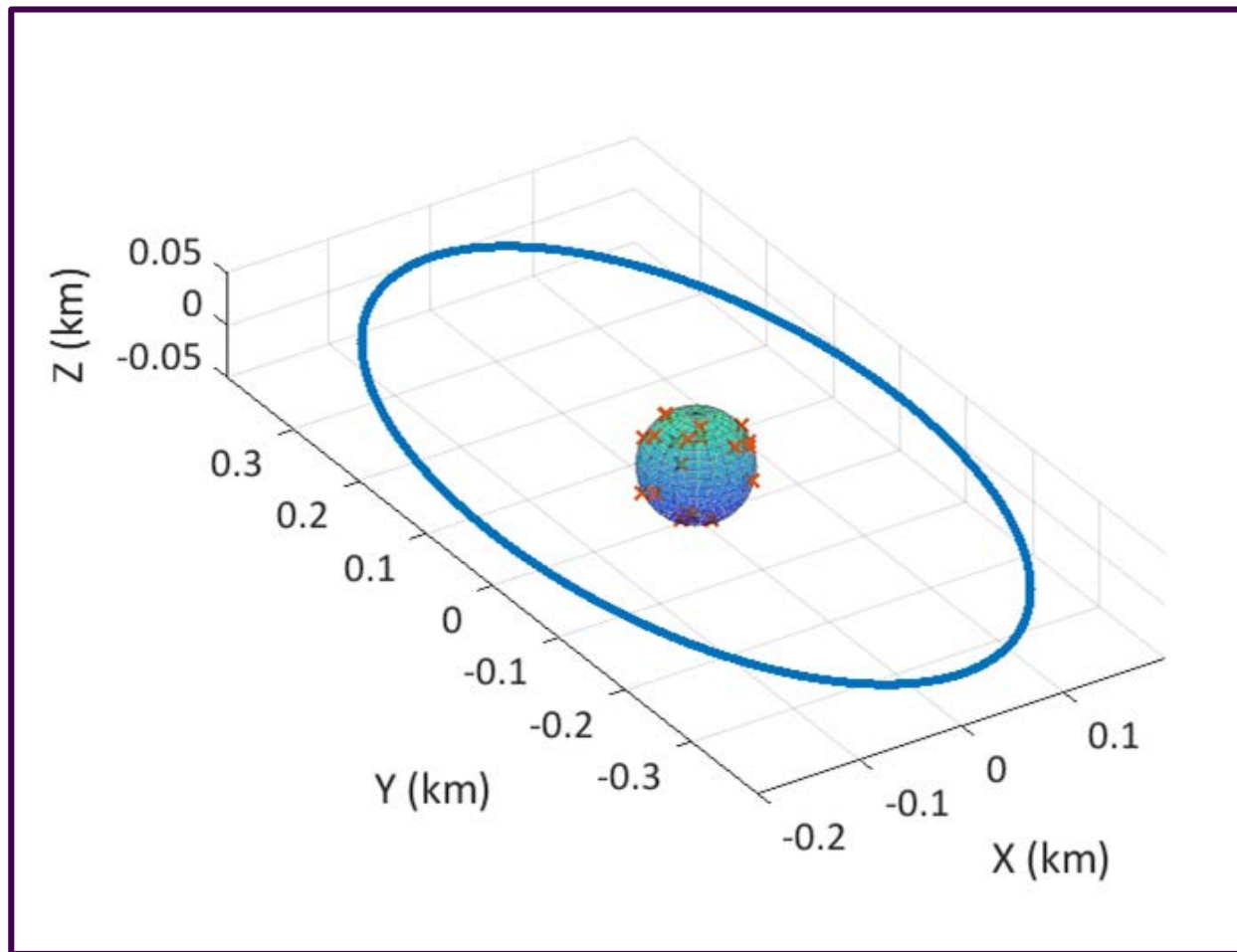
Relative Trajectory Estimate

True Trajectory — Estimated Trajectory



Case 2: Periodic Planar Orbit

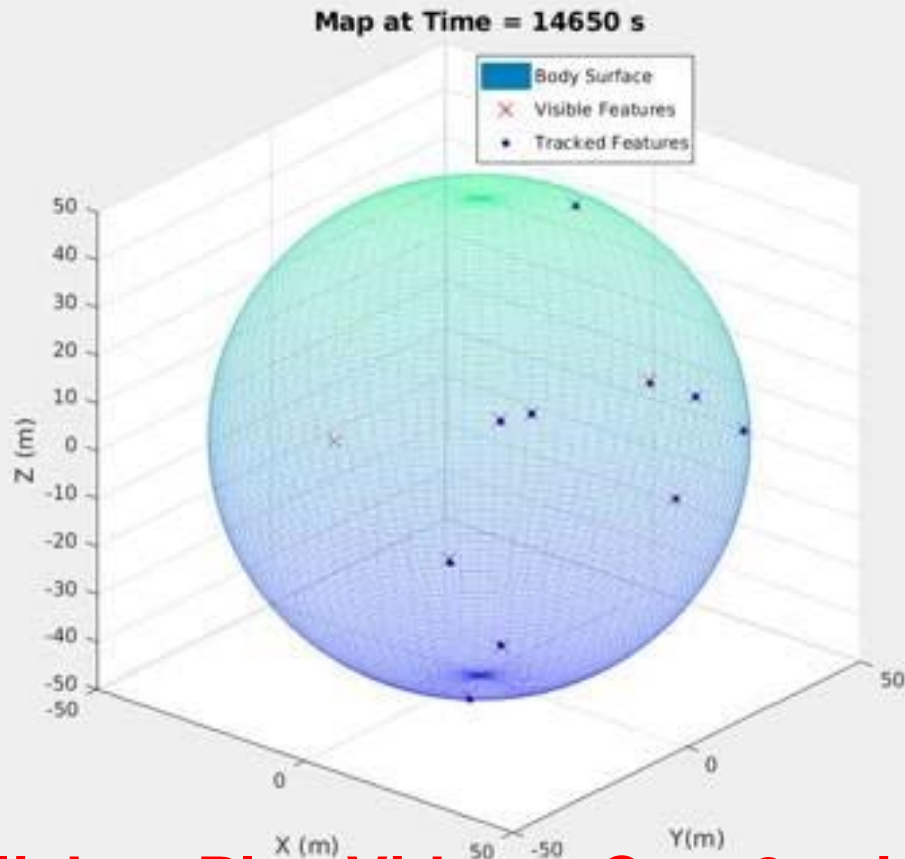
Goal: Test filter stability over a longer period of time.



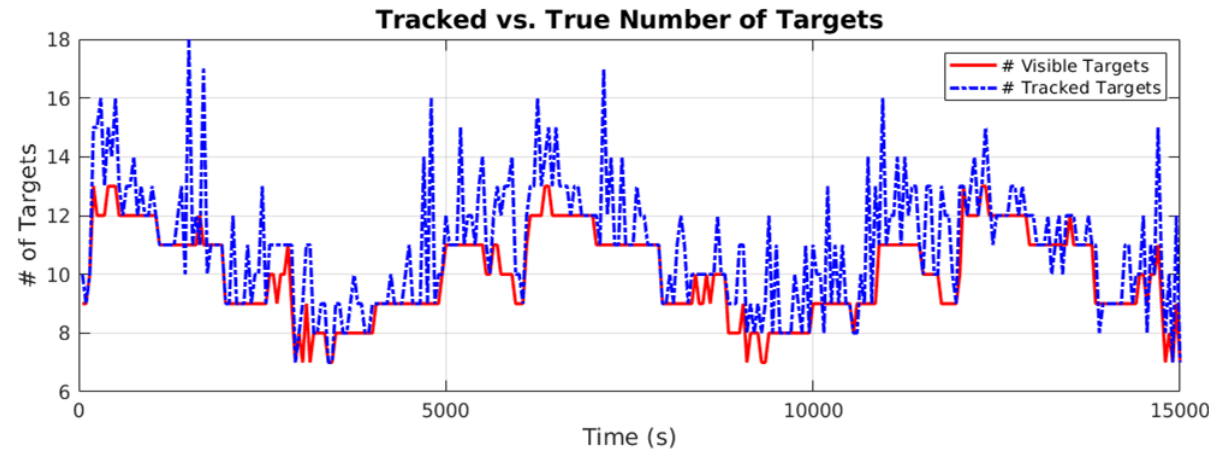
Case 2: Periodic Planar

Orbit

Number of Particles Used: 100

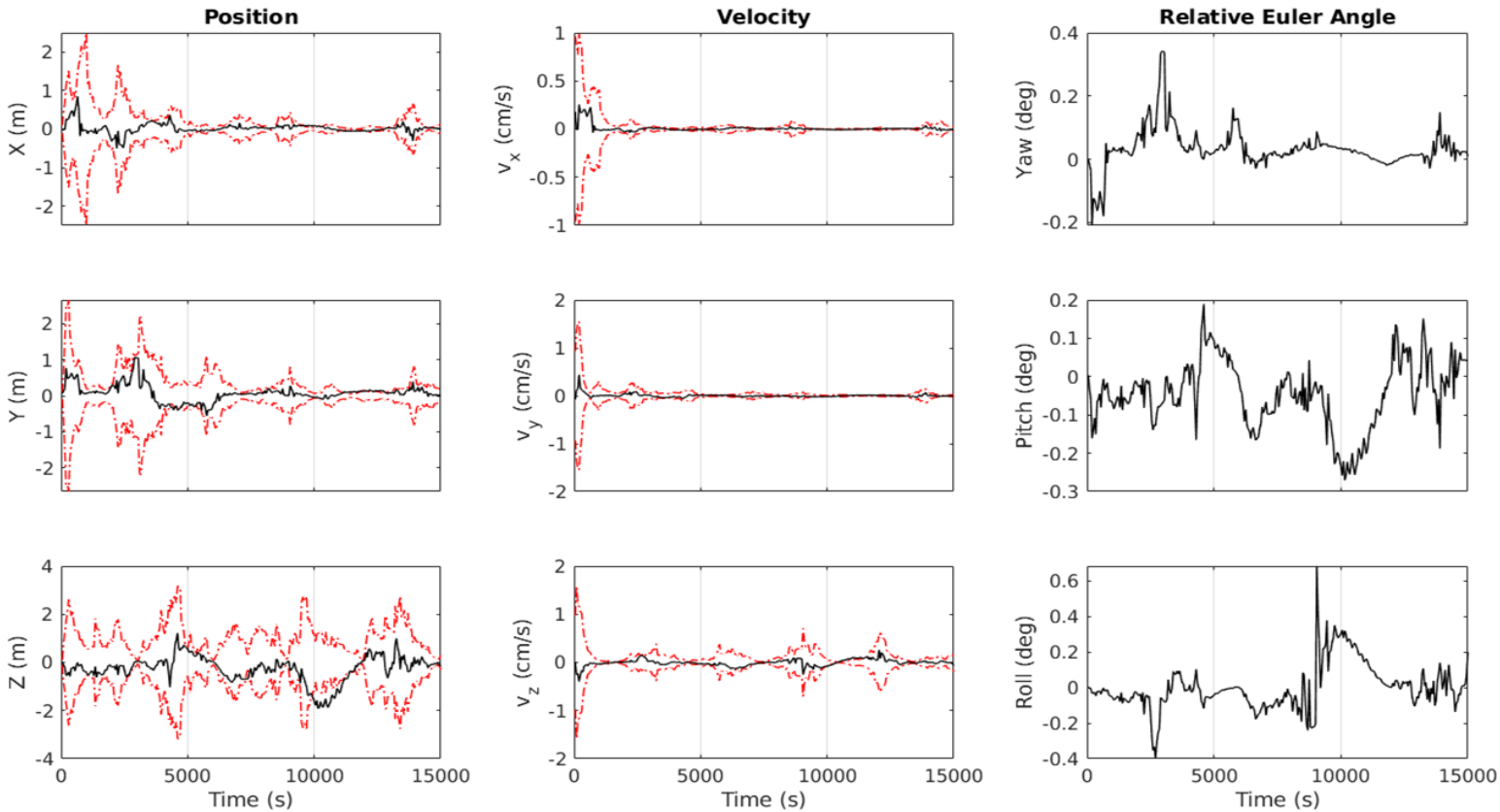


[Click to Play Video -- Case2.avi](#)



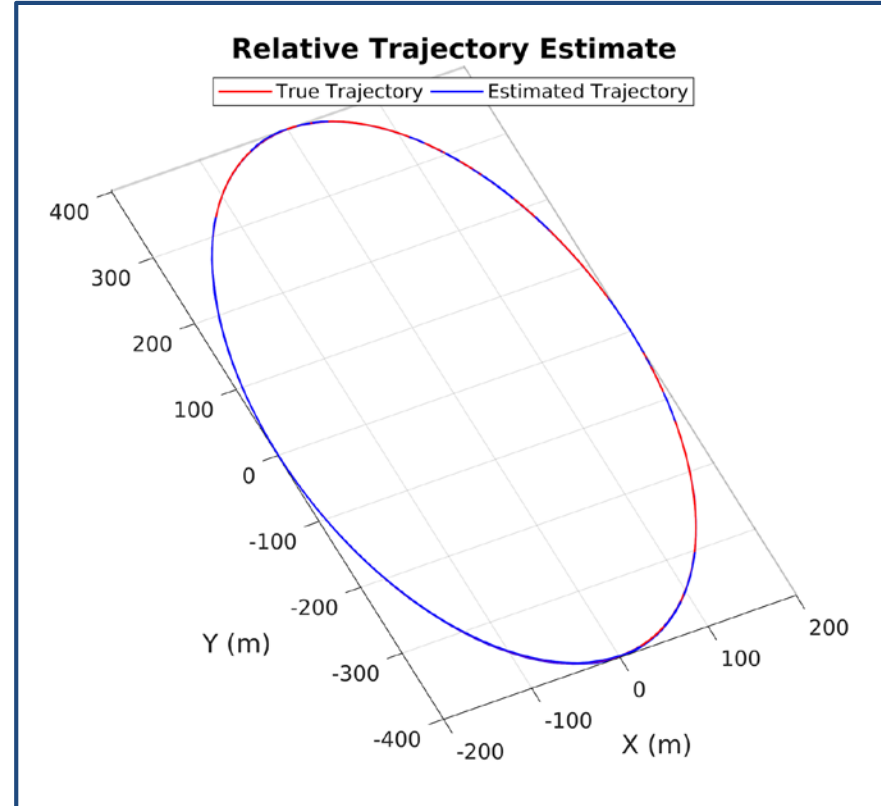
Difference between # of tracked and true targets is periodic with the orbit; filter has been tuned to slightly overestimate # of targets for stability.

Case 2: Periodic Planar Orbit

Pose Estimate Error vs. Sample Covariance 3σ BoundsEstimate Error — 3σ Bound

Relative Trajectory Estimate

True Trajectory — Estimated Trajectory



Pose estimate stay within sample covariance 3σ bounds, and estimated trajectory matches the truth very closely over a long period of time.

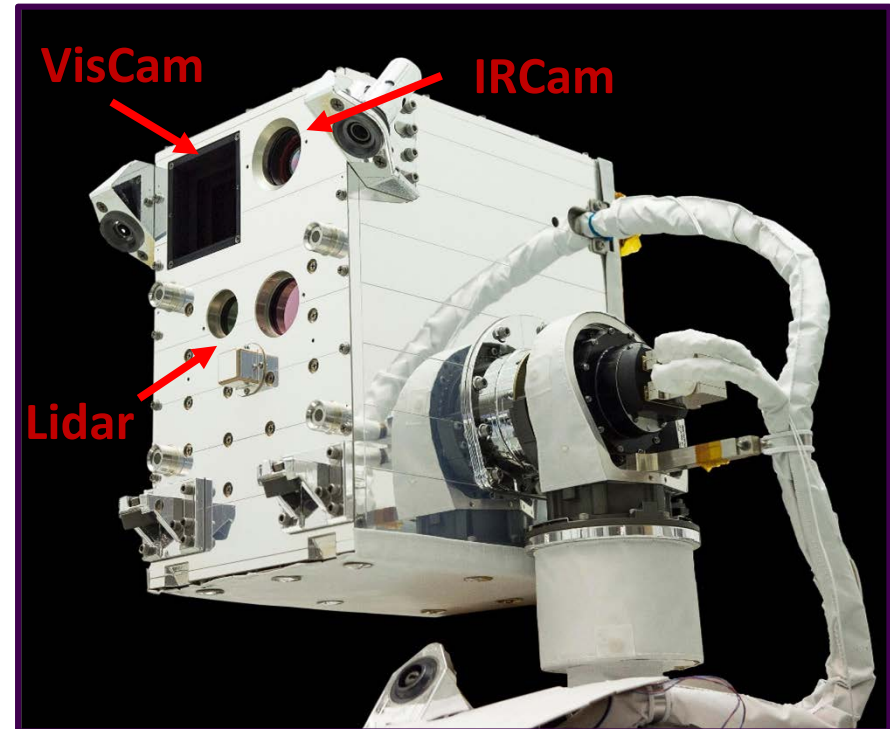
- We have shown using a particle filter that a **Probability Hypothesis Density RFS filter can feasibly perform SLAM in a realistic noisy, non-cooperative space environment.**
 - ✓ First known demonstration of RFS SLAM methods for spacecraft pose estimation and mapping.
 - ✓ Mapping is successful (features are tracked.); Localization is successful (pose is estimated.)
 - ✓ Very little/no *a priori* knowledge of the map given to the filter.
 - ✓ No data association required between states and measurements.
 - ✓ Very little post-processing for feature management.
 - ✓ Heavily cluttered, highly noisy measurements used.
 - ✓ Features move in and out of the sensor field of view.
- **Pose estimate quality is dependent on the number of particles used in the outer particle filter.**
 - More particles is better, but computationally costly.

Moving Forward:

- **Investigate computationally cheaper ways of extracting a pose estimate**
 - e.g. EKF localization from the PHD map, Multi-Bernoulli RFS filter variants.
 - More feasible for onboard computers.
- **More Complicated Dynamics**
 - Higher fidelity dynamics, no constraint on pointing.

- ***Real Life Data***

- Use optical images and LIDAR measurements from existing missions to perform SLAM. (e.g. Raven data from ISS rendezvous)
 - These missions use *a priori* maps, **demonstrate that RFS performance with and without.**



Credit: NASA, Raven

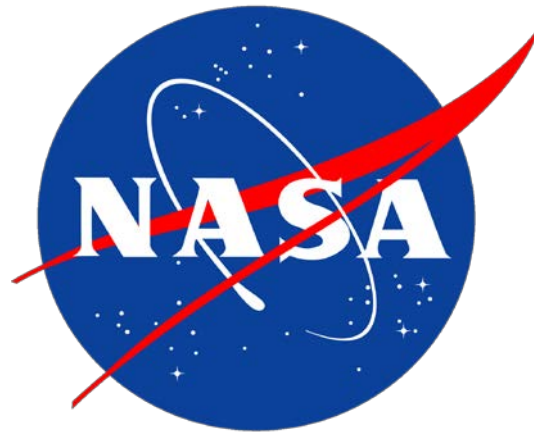
Summary:

- **Autonomous spacecraft navigation:** a necessary requirement of future exploration/servicing missions.
- **Current methods:** ill-suited for challenging measurement environment during rendezvous/prox ops.
- **Random Finite Set based filters:** a more natural choice for these kinds of problems.
- **Initial simulations:** *first known demonstration* of RFS SLAM methods for spacecraft pose estimation and mapping.
- **Moving forward:**
 - Investigate different variants of filter formulations for robustness, real-time computing.
 - Assess performance using flight data.

Acknowledgements:

- Funding from NASA GSFC IRAD Program
- Dr. Richard Linares, Massachusetts Institute of Technology
- Dr. Martin Adams, Universidad de Chile
 - Felipe Inostroza
- Eugene Skelton and Anthony Yu, NASA GSFC

Thanks for listening!



Questions?