



## Moon 2020 - 2030

A New Era of Coordinated Human and Robotic Exploration

ESTEC, Noordwijk, The Netherlands, 15-16 December 2015



### DEVELOPMENT OF THE RESOURCE PROSPECTOR PLANETARY ROVER

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The Resource Prospector (RP) is an In-Situ Resource Utilization (ISRU) lunar rover mission under study by NASA. RP is planned to launch in 2020 to prospect for subsurface volatiles and to extract oxygen from lunar regolith. The mission will address several of NASA's "Strategic Knowledge Gaps" for lunar exploration. The mission will also address the Global Exploration Roadmap's strategic goal of using local resources for human exploration.

The distribution of lunar subsurface volatiles drives the mission requirement for mobility. The spatial distribution is hypothesized to be governed by impact cratering with the top 0.5 m being patchy at scales of 100 m. The mixing time scale increases with depth (less frequent larger impacts). Consequently, increased mobility reduces the depth requirement for sampling.

The target RP traverse will extend 1 km radially from the landing site to sample craters of varying sizes. Sampling craters with different ages will reveal possible volatile emplacement history. In 1 Ga, approximately 60-70 craters of 10 m diameter form per km<sup>2</sup>. Thus, the rover will need to sample at least ten of these craters, which may require a total traverse path length of 2-3 km.



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During 2014-2015, we developed an initial prototype rover for RP. The current design is a solar powered, four-wheeled vehicle, with hub motor drive, offset four wheel steering, and active suspension. Active suspension provides capabilities including changing vehicle ride height, traversing comparatively large obstacles, and controlling load on the wheels. All-wheel steering enables the vehicle to point arbitrarily while roving, e.g., to keep the solar array pointed at the sun while in motion. The offset steering combined with active suspension improves driving in soft soil.

The rover's on-board software utilizes NASA's Core Flight Software, which is a reusable flight software environment. During 2015, we completed the initial rover software build, which provides low-level hardware interfaces, basic mobility control, waypoint driving, odometry, basic error checking, and camera services.

Development of the prototype rover has enabled maturation of many of the subsystems to TRL 5. During the next year, we will conduct integrated testing of concepts of operation, navigation, and remote driving tools. In addition, we will perform environmental tests including radiation (avionics), thermal and thermal/vacuum (mechanisms), and gravity offload (mobility).

National Aeronautics and Space Administration

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# Development of the Resource Prospector Planetary Rover

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# Resource Prospector (RP) Overview



## Mission

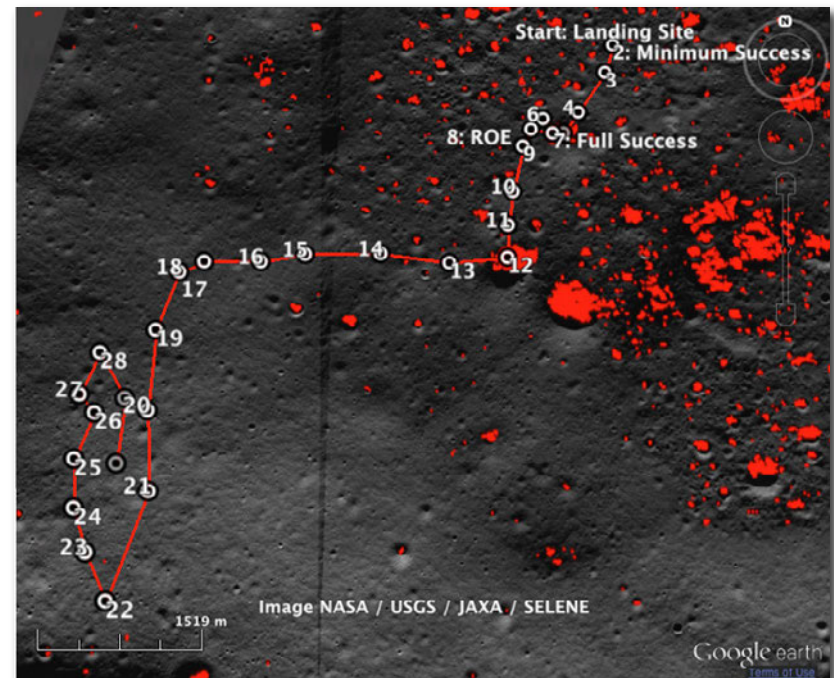
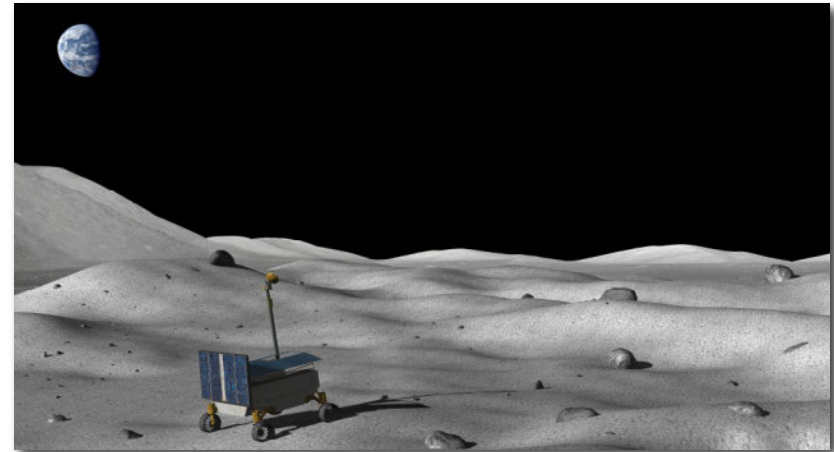
- Characterize the nature and distribution of **lunar polar volatiles**
- Demonstrate **in-situ resource utilization**: process lunar regolith

## Key Points

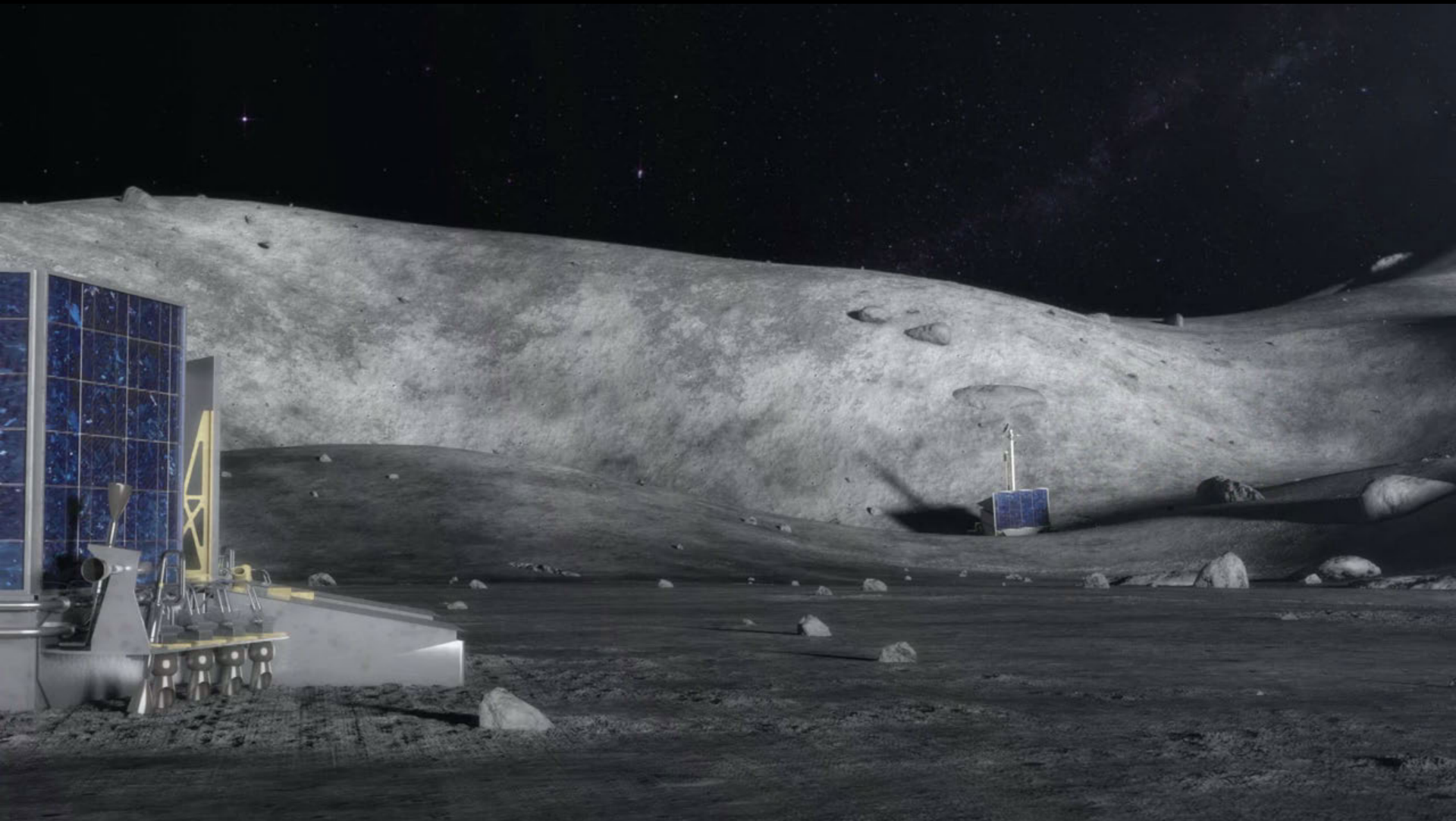
- NASA HEOMD (AES program)
- Class D / Category 3 Mission
- Launch: 2020 (Falcon 9 v1.1)
- Duration: 6-14 Earth days
- Direct-to-Earth communications

## Rover

- Mass: 300 kg (including payload)
- Dimensions: 1.4m x 1.4m x 2m
- Speed made good: 0.5 cm/s
- Power: 300W (solar powered)



# RP Mission Animation



## Minimum Success

- Make measurements from **two places** separated by at least **100 meters** (may require driving 300-400 m path length)
- Surface or subsurface measurements

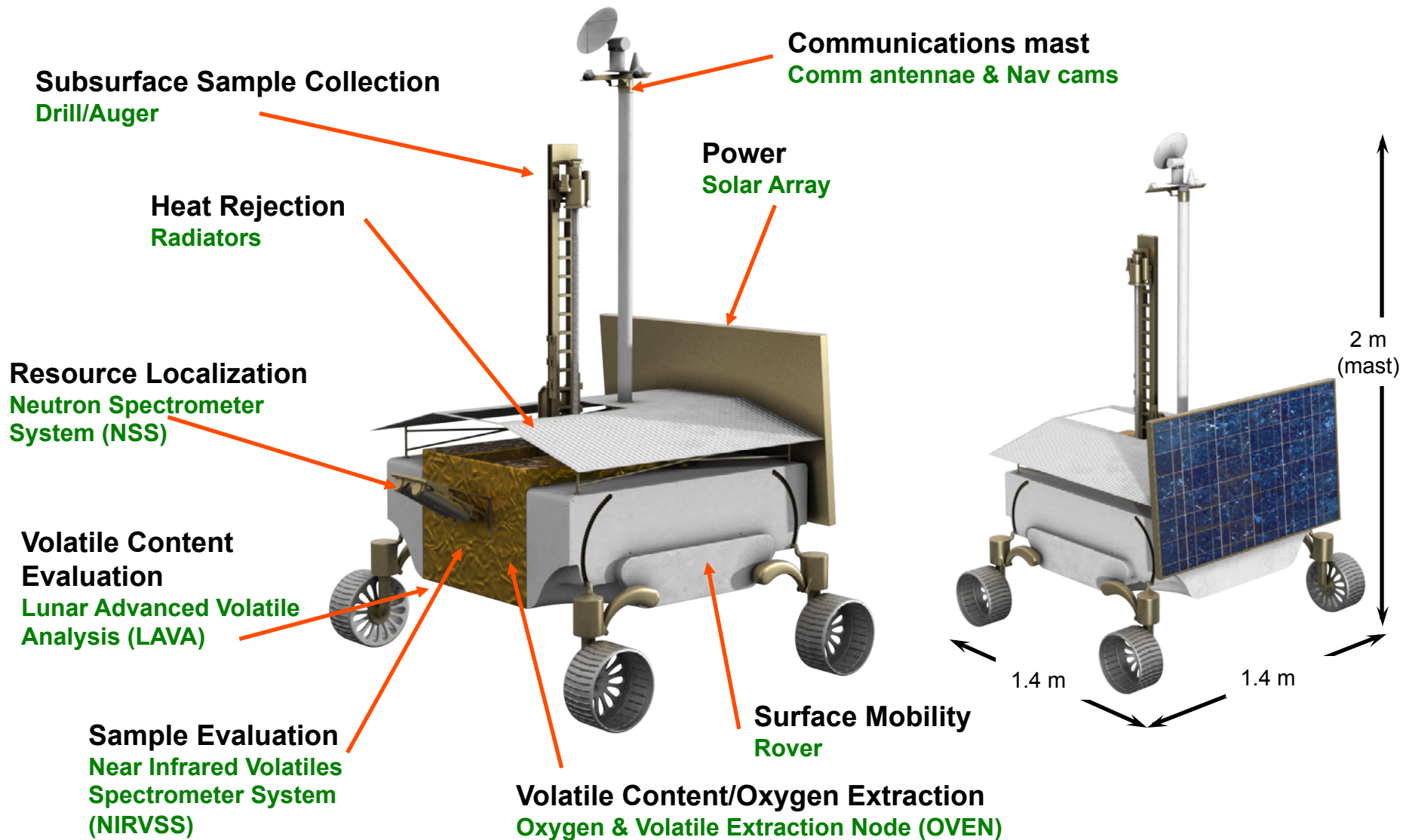
## Full Success

- Measurements from **two places** separated by at least **1000 meters** (may require driving 3000-4000 m path length)
- Surface and subsurface measurements
- Measurements and sample acquired from **permanently shadowed area**
- Demonstrate ISRU

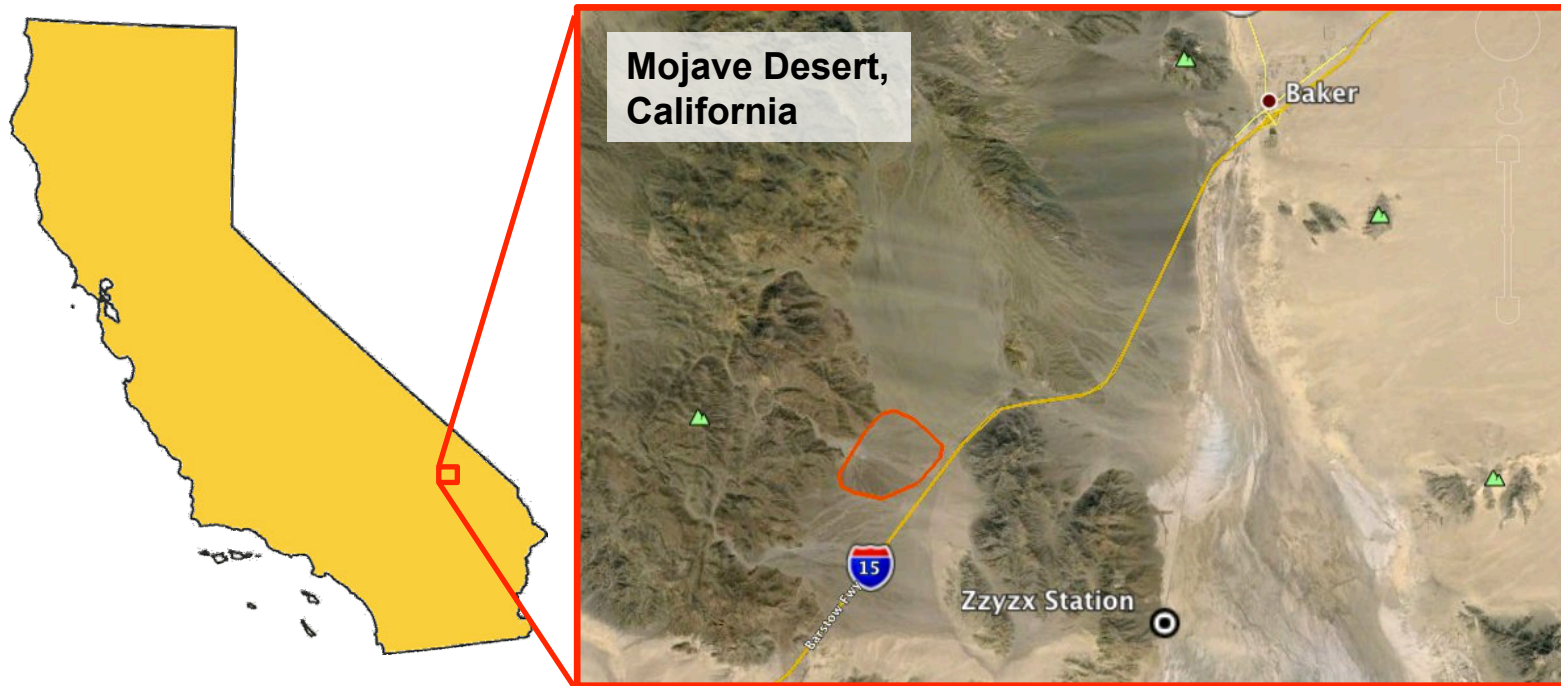
## Stretch Goals

- Make subsurface measurements in at least 8 locations across 1000 m
- Process and analyze subsurface material in at least 4 locations across 1000 m distance
- Provide geologic and thermal context

# RP Rover



# Mojave Volatiles Prospector (October 2014)



## High-Fidelity Analog Field Test

- **Prospecting.** Mature prospecting ops concept for NIRVSS and NSS instruments in a lunar analog field test
- **Real-Time Science Ops.** Improve support software by testing in a setting where the abundance / distribution of water is not known a priori
- **Terrestrial Science.** Understand the emplacement and retention of water in the Mojave Desert by mapping water distribution / variability



# Prospecting Payload on K-REX Rover



**Sample Evaluation**  
Near Infrared Volatiles  
Spectrometer System (NIRVSS)

**Resource Localization**  
Neutron Spectrometer  
System (NSS)

# Science Operations (NASA Ames)



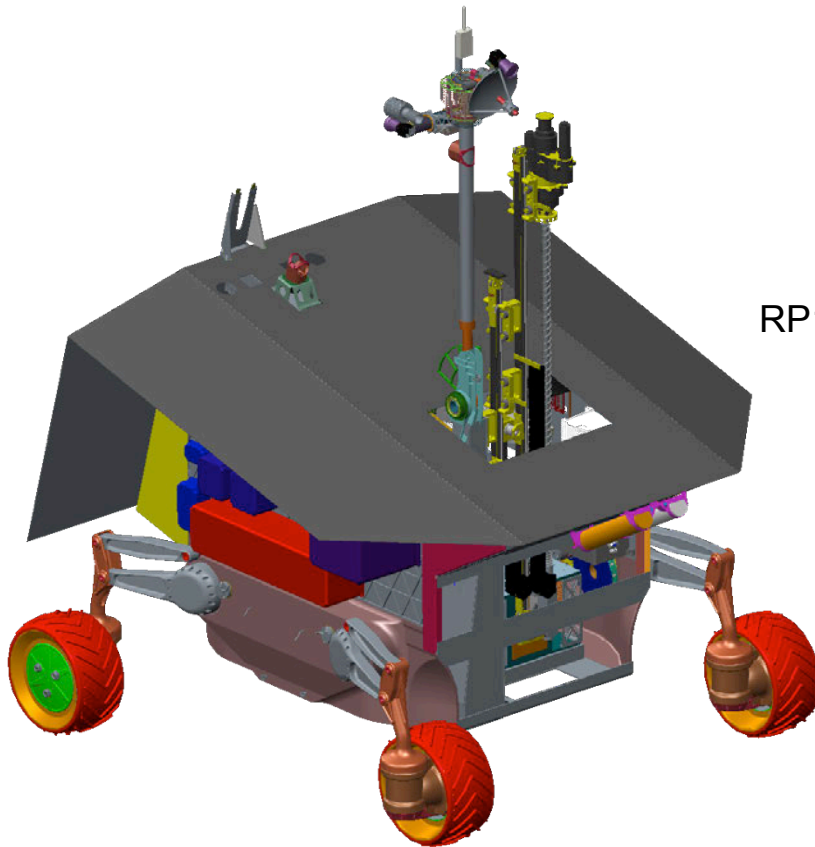
# Mojave Volatiles Prospector



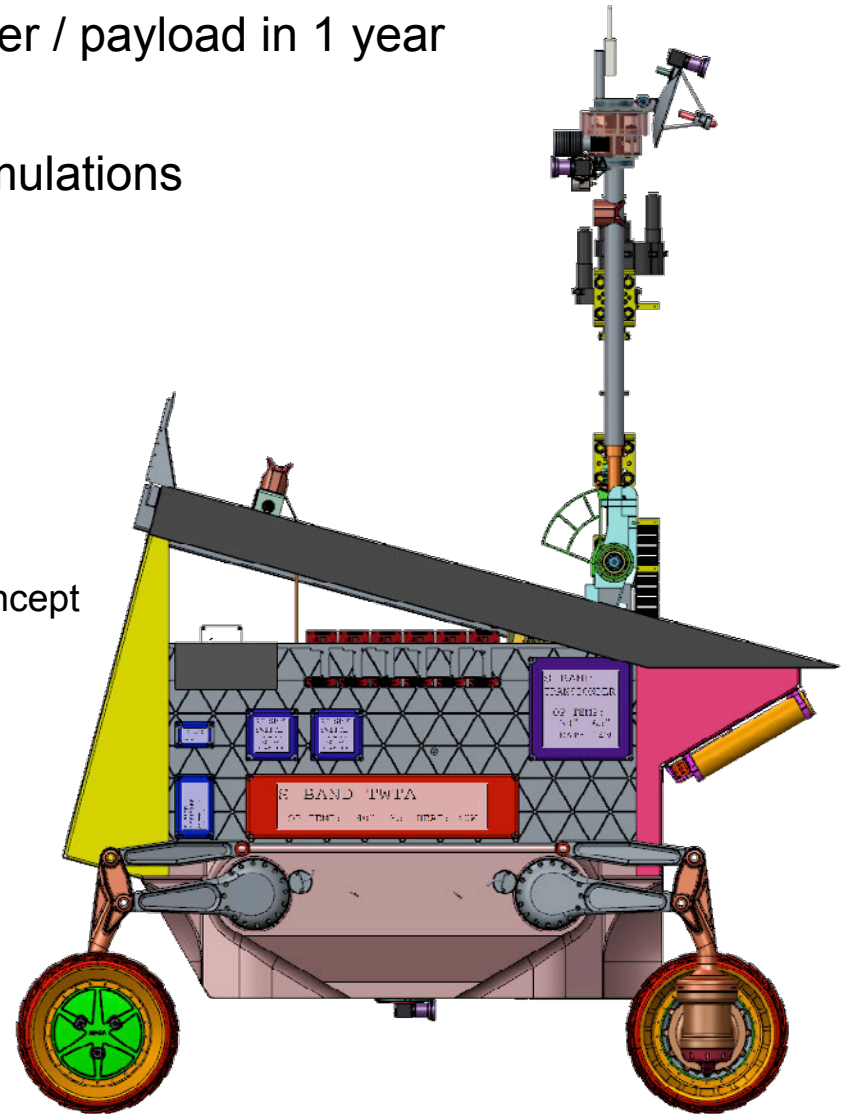
# “RP15” Flight Project in a Year



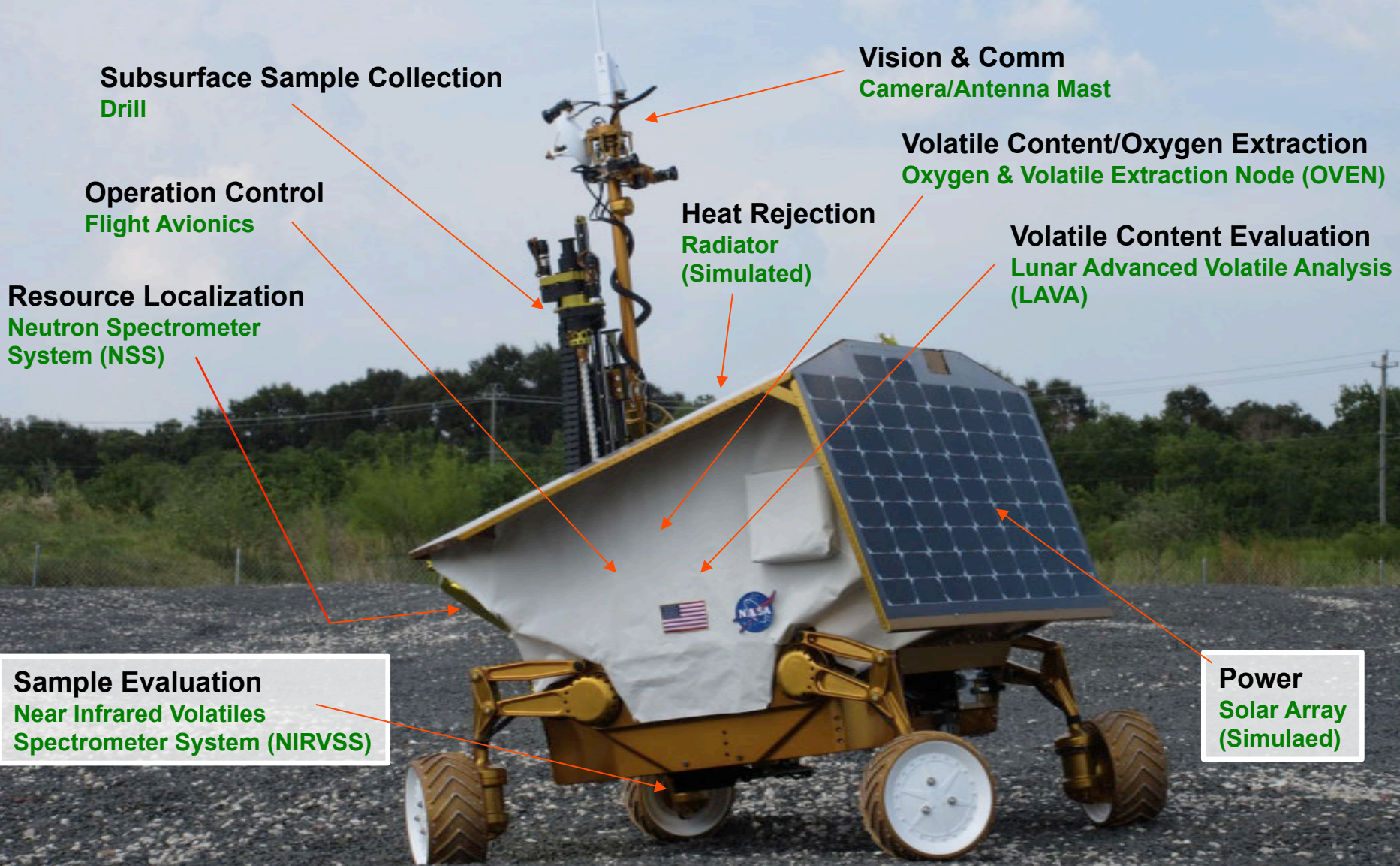
- Build, integrate, and test integrated rover / payload in 1 year
- Mature designs and retire risk for flight
- Perform subsystem and operational simulations



RP15 concept



# RP Engineering Prototype



**Subsurface Sample Collection  
Drill**

**Vision & Comm  
Camera/Antenna Mast**

**Volatile Content/Oxygen Extraction  
Oxygen & Volatile Extraction Node (OVEN)**

**Operation Control  
Flight Avionics**

**Heat Rejection  
Radiator  
(Simulated)**

**Volatile Content Evaluation  
Lunar Advanced Volatile Analysis  
(LAVA)**

**Resource Localization  
Neutron Spectrometer  
System (NSS)**

**Sample Evaluation  
Near Infrared Volatiles  
Spectrometer System (NIRVSS)**

**Power  
Solar Array  
(Simulaed)**

# Driving Tests (August 2015)



# Distributed Operations Testing (August 2015)



NASA ARC Mission Control driving RP15 rover at NASA JSC



NASA JSC Rock Yard from the rover (left) stereo camera



NASA KSC Payload Control

# Rover Operator Interface (VERVE)



The screenshot displays the VERVE for RP15 software interface. The central 3D view shows a rover on a simulated planetary surface with a navigation path of waypoints (0001-0020). A 'MobStates' window indicates the rover is paused at waypoint 0020, with a distance of 0.1 m and a drive speed of 0.10 m/s. The interface includes a 'Waypoints' table, a 'Pose (RAPID)' window, and a 'CMD 4: OK' control panel.

Status	Id	X
Completed	0004	21.400
Aborted	0005	24.190
Completed	0006	26.170
Completed	0007	28.930
Completed	0008	30.260
Completed	0009	30.550
Completed	0010	30.960
Completed	0011	32.090
Completed	0012	34.240
Completed	0013	36.840
Completed	0014	37.870
Completed	0015	39.290
Completed	0016	40.330
Completed	0017	42.560
Completed	0018	44.880
Completed	0019	45.180
Paused	0020	47.290

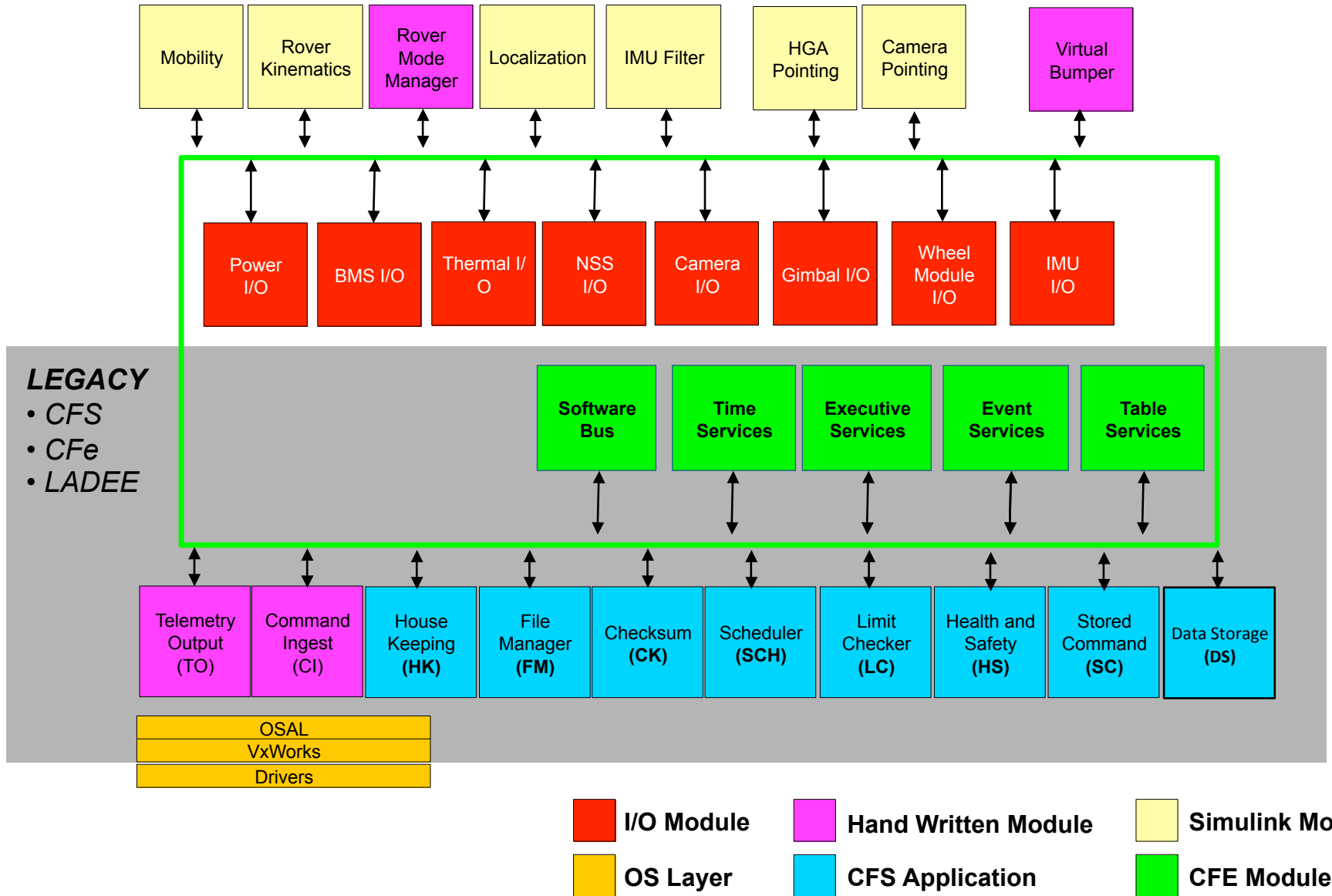
Name	Value
X	45.12
Y	-5.25
Z	0.00
Roll	0.00
Pitch	-0.00
Yaw	20.29



# Reduced Gravity Driving (December 2015)

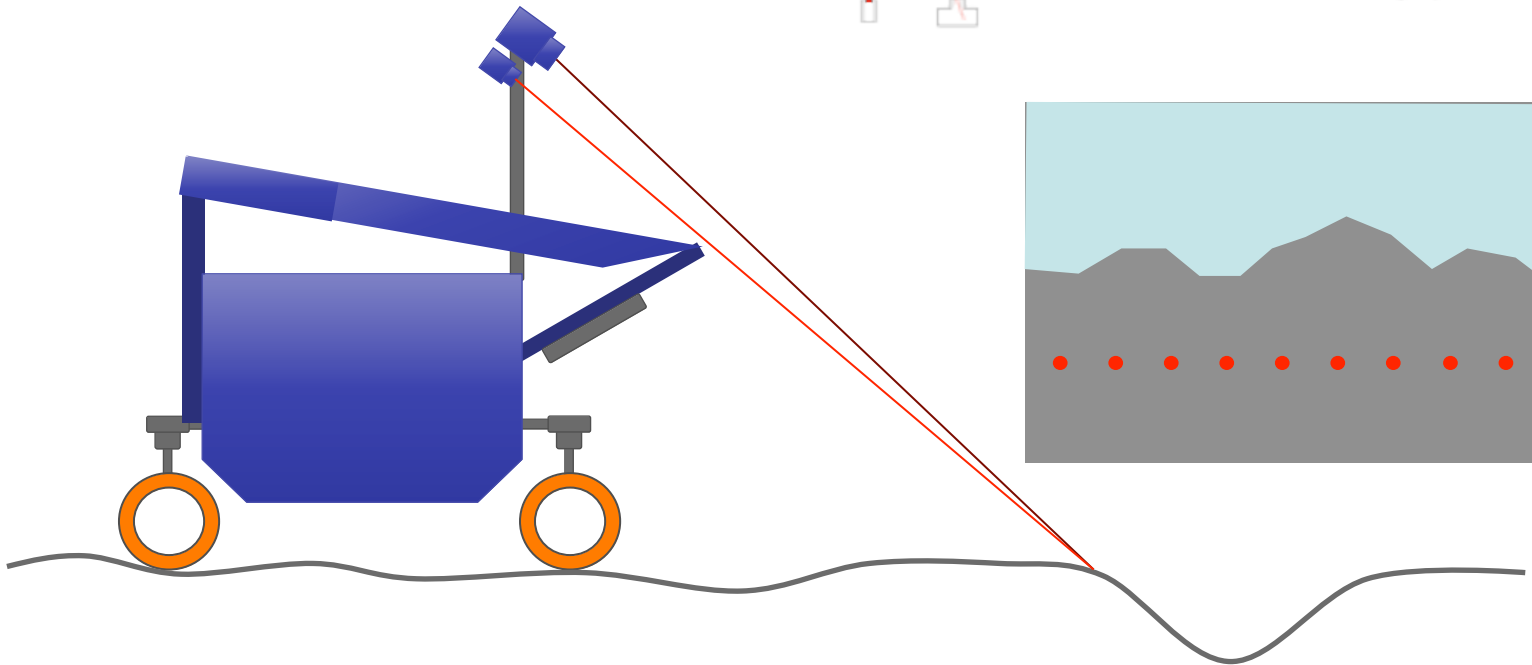
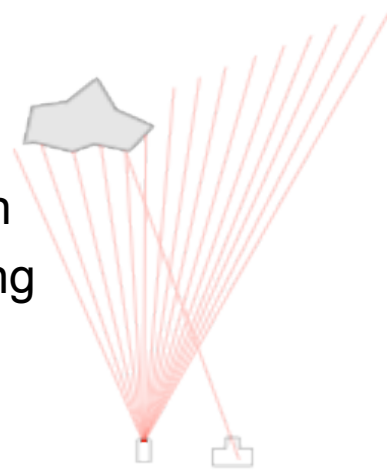


# Flight Software Architecture



## “Virtual Bumper”

- Project laser dots
- Detect dots in HazCam
- Minimal onboard computation
- Enables onboard safeguarding
- 98% success rate



# Hazard Detection Testing (July 2015)



10x

## **Roving**

- Traversing in soft soil – slip and embedding
- Active vs passive suspension
- Sharp thermal gradients across rover
- Variable thermal interface with surface

## **Drilling**

- Slip / unintended motion
- Stances during drilling / stuck drill and options for stuck drill recovery
- Unknown near-surface regolith compaction profile / pre-load requirements

## **Navigation**

- Performance of stereo vision for hazard detection and localization
- Performance of active illumination (flood lighting and laser projection)
- Positive and negative obstacles (size, shape, distribution, composition)
- Regolith / rock optical properties (reflectance, opposition surge, etc)

# Questions ?

