

Design, Fabrication, and Critical Current Testing of No-Insulation Superconducting Rotor Coils for NASA's High-Efficiency Megawatt Motor

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Design & Testing of No-Insulation Superconducting Rotor Coils for NASA's HEMM

Motivation

- Reduced energy consumption, emissions, and noise of commercial transport aircraft [1]
 - Electrified aircraft propulsion (EAP) enables system-level benefits to these metrics
- EAP concepts require advances to electric machines
- NASA's High-Efficiency Megawatt Motor (HEMM) sized as generator for NASA's STARC-ABL concept

		STARC-ABL		
	Electric machines	Current design	With HEMM	
	Specific power, kW/kg	13.2	16	
	Efficiency, %	96	98 to 99	
STARC-ABL	Performance relative to STARC-ABL rev A With HEMM			
		Fuel burn, %	-1 to -2	
	Wast	e heat in generator	½ to ¼ (−30 to −44 kW)	

NASA's High-Efficiency Megawatt Motor (HEMM)

- Sized for generator of NASA's STARC-ABL concept
- Wound-field synchronous machine
 - Tolerant of stator fault
- Superconducting rotor
 - Negligible energy loss
 - Very strong magnetic excitation

Parameter	Value
Rated continuous power	1.4 MW
Nominal speed	6,800 rpm
Tip speed	Mach 0.31
Rated torque	2 kNm
Specific power goal	16 kW/kg
Efficiency goal	>98 %



Outline

Talk 1 (Scheidler, 2018 AIAA P&E)

- Complete preliminary design package for rotor
 - Electromagnetic design & optimization
 - Rotor containment design & stress analysis

This talk

- Overview of current rotor design
- Fabrication & testing of sub-scale superconducting rotor coils

Outline

- Rotor & coil design
- Coil fabrication
- Critical current testing
- Conclusions

Rotor Design

Parameter	Value
Electrical frequency	DC
Number of poles	12
Material	Solid $Fe_{49.15}Co_{48.75}V_2$
Outer diameter	30 cm
Inner diameter	18.9 to 20 cm
Axial length	12.5 cm



Design process (see 2018 AIAA P&E paper)

- Defined current & thermal limits
 - Based on manufacturer data & safety factors
- Parametric studies of back iron's width w and thickness t (2D & 3D, nonlinear FEA)
 - Optimized coil's geometry by numerically maximizing # of turns in coil
 - Custom extrapolation of back iron's *B* vs *H* response
 - Metrics: performance performance/mass performance/cost
- Stress analysis of centrifugal loading (2D & 3D FEA)

Soft magnetic material (back iron)

Region available for containment structure & clearances

Rotor Design



Rotor Design



Coil Design

- 2nd generation high temperature superconductor (REBCO) selected
 - Commercially available in long piece length
 - Sufficient performance at "high" temperatures in moderately strong magnetic environments
- REBCO is a composite conductor in the form of thin tape
 - AC losses will be negligible
- No-insulation (NI) coils selected [9-11]
 - Fault tolerant
 - Higher engineering current density
 - Higher mechanical strength



No-insulation superconducting coils are very promising, but have not been studied for rotating systems

Coil Design

Coil characteristics

Parameter	Value
Turn-to-turn insulation	None
Operating temperature	62.8 K
Operating current	51.5 A
# of layers per coil	4
# of turns per layer	~ 230
Solder	52In 48Sn

Superconductor characteristics

Parameter	Value
Material	REBCO
Width	4 mm
Thickness	65 micron
Min. bend radius	15 mm



Risk reduction testing

- Key risks of the superconducting coils
 - Coils will fail when thermally cycled due to thermal stresses
 - Coils will fail when rotor is spun up due to centrifugal stresses
- Risk reduction tests
 - Thermal cycling
 - Goal: demonstrate coils that are not degraded by thermal cycling
 - <u>Approach</u>: measure superconducting performance subject to thermal shock re-measure superconducting performance
 - Proof: negligible change in critical current & "n-value"
 - Rotation (future work)
 - Goal: demonstrate coils that are not degraded by high-speed rotation
 - <u>Approach</u>: measure superconducting performance spin coils re-measure superconducting performance
 - Proof: negligible change in critical current & "n-value"

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Coil Fabrication

- Methodical development approach: simple, sub-scale realistic, full-scale
- 25-turn sub-scale coils
 - Fewer turns & shorter



dimensions in mm



Coil Fabrication

- 3D printed nylon winding fixture
 - Reduced lead time & cost
 - But, limited temperature
- Accurately establishes width of active region & height
- Fixture inverted for epoxy application



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Critical Current Testing

- Critical current $(I_C) = I_C(T, B, \theta)$
- Coil mounted to G10 plate & suspended in liquid nitrogen
- Measurements: voltage & transport current





Critical Current Testing

Voltage vs current response commonly described by

 $V = V_{\rm c} \left(\frac{I}{I_{\rm c}}\right)^n$ where the critical voltage $V_{\rm c} = 1 \frac{\mu V}{cm} *$ superconductor length

- "n-value" indicates combined quality of superconductor & measurement
- Detect damage via changes in n and/or I_c





Critical Current Testing – 1-layer coils

- Two 1-layer coils tested: V vs I response at 77 K in "self field"
 - Sanity check: measure for increasing & decreasing I
 - Thermal cycling tolerance: measure before & after 2 or 4 thermal shock cycles



Coil 2 (4 thermal cycles)

Critical Current Testing – 1-layer coils

	Coil 1		Coil 2	
	<i>I</i> _c , A	<i>n</i> , –	<i>I</i> _c , A	<i>n</i> , –
Before thermal cycling	76.8	19.8	75.9	23.9
After thermal cycling	76.9	19.7	76.3	21.7

- Averaged results for increasing & decreasing *I*
- Coil 1 (2 thermal cycles)
 - No detectable damage
- Coil 2 (4 thermal cycles)
 - I_c increased by 1%, but *n* decreased by 9%
 - Inconclusive, but at worst only minor degradation of n

Critical Current Testing – 2-layer coils

- 2-layer coil requires superconducting joint \rightarrow solder introduces finite resistance
- After subtracting the linear trend, results analyzed as before
- Coil 3 broke while attempting to demonstrate self-protection feature
 - Damage occurred only in unprotected current lead





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Coil 3

Coil 3

Critical Current Testing – 2-layer coils

<u>Coil 4</u>

- Current lead damaged during coil fabrication
 - *I*_c reduced and *n*-value significantly reduced
- I_c increased by 3%, but *n* decreased by 13%
- Inconclusive, but at worst only modest degradation of n



	<i>I</i> _с , А	<i>n</i> , –	
Before thermal cycling	57.4	5.4	
After 2 thermal cycles	58.3	4.7	
After 6 thermal cycles	59.0	4.7	

Coil 4



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Conclusions

- Discussed the design of the superconducting rotor of NASA's 1.4 MW High Efficiency Megawatt Machine (HEMM)
 - Uninsulated superconducting coils selected to provide fault tolerance and significantly higher engineering current density
 - 2 key risks: resilience to thermal cycling and rotation
- 3D printed winding fixtures work well & allow short lead time
 - But, they prevent the use of some solders while the coil is fixture
- Initial thermal cycling measurements of 1-layer and 2-layer uninsulated coils
 - Tested up to $1.15I_c$ 2 to 6 thermal shock cycles
 - After thermal cycling, I_c increased but *n*-value decreased
 - Results inconclusive, but suggest little to no degradation

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References

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- [7] Welstead, J., and Felder, J. L., "Conceptual design of a single-aisle turboelectric commercial transport with fuselage boundary layer ingestion," 54th AIAA Aerospace Sciences Meeting, AIAA SciTech Forum, San Diego, CA, AIAA 2016-1027, 2016. doi:10.2514/6.2016-1027.





Superconductor current & thermal limits

- Critical current $(I_C) = I_C(T, B, \theta)$
 - Datasheet values $\theta = 0^{\circ}$ and 90° are insufficient
- Datasheet specs de-rated twice: angular dependence & safety factor





Safety factor

±20% Estimate of wire variation

+ ±15% Modeling inaccuracy

±35% (≈1.5 safety factor)

National Aeronautics and Space Administration

Superconductor current & thermal limits

• Measurements at B = 2 T obtained from manufacturer



Optimization of rotor coil's geometry

- Optimized coil's geometry for given iron thickness & width by numerically maximizing # of turns
 - Rectangular coil cross section
 - Also outputs total length & cost of conductor, mass of iron+coil
 - 4 mm is optimal width of superconductor



Soft magnetic material (back iron)



Region available for containment structure and clearances

Preliminary design – double dovetail rotor teeth



Critical Current Testing – 1 layer coils

		Coil 1		Coil 2	
Test		<i>I</i> _с , А	n, –	<i>Ι</i> _c , Α	<i>n</i> , –
Before thermal cycling	I increasing	76.9	18.5	75.8	24.6
	I decreasing	76.6	21.0	75.9	23.2
After thermal cycling	I increasing	76.8	19.7	76.2	21.6
	I decreasing	76.9	19.7	76.3	21.8